

Experiment - 6

Drag estimation on cylinder from wake profile measurement

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1. Aim :

- To estimate Drag from wake velocity profile measurement.
- To check relation between Coefficients of Drag and Reynolds no.

2. Apparatus :

Required apparatus for performing this experiment are:

- Manometer
- C15-10 Armfield tunnel
- Pitot-static Probe
- Fan
- Cylinder model

3. Theory :

Drag : Drag is the force which opposes the motion of an object. Drag is calculated by :

$$D = \int_{-\infty}^{\infty} \rho v (V_{\infty} - v) dy$$

Coefficients of Drag : Drag coefficients caused due to skin friction and Drag. Drag coefficients is calculated by :

$$C_d = \frac{D}{(\rho V_{\infty}^2 d)/2}$$

Stagnation and Static Pressure : Stagnation pressure is the pressure at the stagnation points in the fluid flow. Static pressure is the actual thermodynamic pressure of a flow.

$$P_{stag} = P_{\infty} + \frac{\rho v^2}{2}$$

Reynolds Number : The Reynolds number is the ratio of inertial forces to viscous forces within a fluid which is subjected to relative internal moment due to variation of velocities.

$$Re = \frac{\rho V_{\infty} d}{\mu}$$

4. Procedure :

1. In wind tunnel test section is set.
2. Pitot-static probe is connected to manometer.
3. Fan speed is fixed.
4. Required readings are taken.

5. Observation :

5.1 Wake velocity profile with Y distance of Pitot probe

Table 1: **Wake profile velocity with y distance of pitot probe**

| Port No. | Distance from starting point(in mm) | Stagnation Pressure(in mm of water) | Velocity(m/s) |
|----------|-------------------------------------|-------------------------------------|---------------|
| P_1 | 0 | -1.1 | 8.854 |
| P_2 | 6 | -1.4 | 8.579 |
| P_3 | 12 | -2.1 | 7.899 |
| P_4 | 18 | -1.8 | 7.155 |
| P_5 | 24 | -3.1 | 6.812 |
| P_6 | 30 | -3.4 | 6.450 |
| P_7 | 36 | -2.6 | 7.376 |
| P_8 | 42 | -2.5 | 7.483 |
| P_9 | 48 | -1.7 | 8.295 |
| P_{10} | 54 | -1.1 | 8.854 |

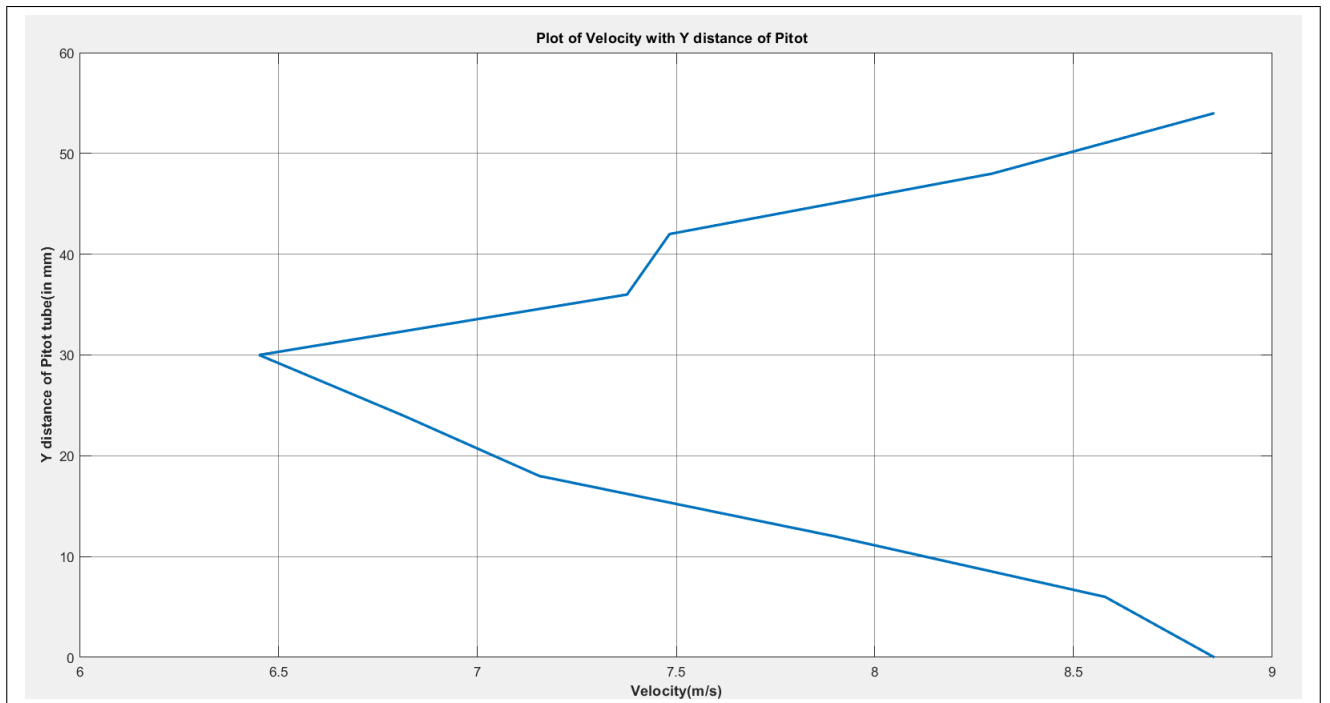


Figure 1: Variation of velocity with y distance of pitot tube.

Drag vs Velocity :

Table 2: **Variation of Drag Coefficients with Reynolds No**

| Sl. No. | Velocity | Drag | Coeff of Drag(C_d) | Reynolds No.(R_e) |
|---------|----------|--------|------------------------|-----------------------|
| 1 | 8 | 1.225 | 1.046 | 16109.98 |
| 2 | 10.2 | 1.3817 | 0.723 | 20709.94 |
| 3 | 12 | 1.458 | 0.553 | 12413.13 |
| 4 | 14.2 | 1.76 | 0.480 | 28595.21 |
| 5 | 16 | 2.23 | 0.476 | 32219.95 |
| 6 | 18 | 3.18 | 0.536 | 36247.44 |

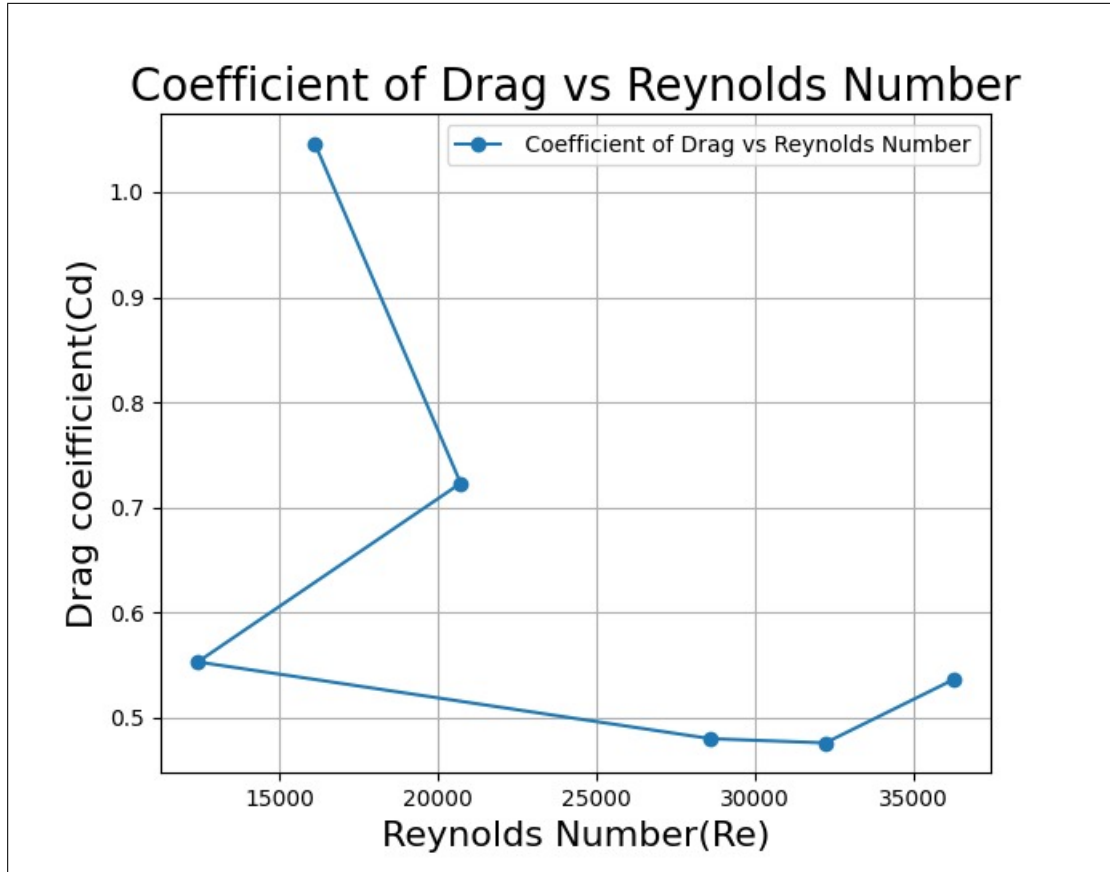


Figure 2: Variation of Coefficients of Drag with Reynolds No.

6. Calculations :

Density of air = 1.225 m^3

Zero error = -0.4 mm of Water

Static Pressure(P_∞) = -(6-0.4) mm of Water = -5.6 mm of water

Density of Water(ρ_w) = 1000 Kg/m^3

6.1 Calculation of wake profile velocity :

For port 1 :

Stagnation Pressure(P_{stag}) = -(1.5-0.4) mm of Water = -1.1 mm of Water

$$P_{stag} = P_\infty + \frac{\rho v^2}{2}$$

$$\begin{aligned}
\Rightarrow P_{stag} - P_{\infty} &= \frac{\rho v^2}{2} \\
\Rightarrow \rho_w g h &= \frac{\rho v^2}{2} \\
\Rightarrow v &= \sqrt{\frac{1000 \times 9.8 \times (5.6 - 1.1) \times 10^{-3} \times 2}{1.225}} \\
\Rightarrow v &= 8.854 \text{ m/s}
\end{aligned}$$

6.2 Calculation of Drag :

Drag(D) is calculated by ,

For 10 ports,

$$D = \int_{-\infty}^{\infty} \rho v (V_{\infty} - v) dy \approx \sum_{i=1}^9 \rho v_{avg} (V_{\infty} - v_{avg}) \Delta y$$

Where, $v_{avg} = \frac{v_i + v_{i+1}}{2}$

$\Delta y = 6 \text{ mm} = 0.006 \text{ m}$

$V_{\infty} = 10.2 \text{ m/s}$

$$D = \sum_{i=1}^9 \rho v_{avg} (V_{\infty} - v_{avg}) \Delta y$$

$$\begin{aligned}
\Rightarrow D &= 1.225 \times 0.006 (8.7165 \times 1.4835) + (8.239 \times 1.961) + (7.527 \times 2.673) + (6.9835 \times 3.2165) + (6.631 \times 3.569) \\
&\quad + (6.913 \times 3.287) + (7.4295 \times 2.7705) + (7.889 \times 2.311) + (8.5745 \times 1.6255)
\end{aligned}$$

$$\therefore D = 1.3817 \text{ N}$$

6.3 Calculation of Coefficients of Drag (C_d) and Reynolds number (R_e):

Diameter(d) = 30 mm

For velocity(V_{∞} 10.2 m/s,

$$C_d = \frac{D}{(\rho V_{\infty}^2 d)/2}$$

$$\therefore C_d = 0.723$$

$$R_e = \frac{\rho V_{\infty} d}{\mu}$$

Where $\mu = 1.81 \times 10^{-5}$

$$\therefore R_e = \frac{1.225 \times 10.2 \times 0.03}{1.81 \times 10^{-5}} = 20709.94$$

7. Sources of Error:

- Error due to instrumental defect.
- Error may occur in taking readings before flow becomes steady.
- Error due to environmental effect like temperature, pressure change.
- Error in measurement due to presence of zero error in parameters.
- Dimensional error may occurs

8. Conclusion :

- Wake velocity profile is symmetrical.
- Drag increases on increasing velocity.
- Drag coefficient increases on increasing Reynolds number.