

# VISION

**INSPIRING ROCKETRY AMONG THE YOUNGSTERS** 

•ADIT JAIN VASU PALIWAL

### HISTORY OF ROCKETS



COMBUSTION CYCLES

Explander cycle

Gas Generator cycle

Staged combustion

Full flow staged combustion

PROPELLANT, ENGINES AND POWER CYCLE

### COORDINATE ROCKET **EQUATIONS** SYSTEMS

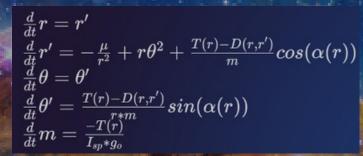
 $1.\Delta M/M = 1 - \exp(\Delta v/v_{ex})$ 2.  $\Delta v = Isp g0 ln(Mo/Mf)$ 

1. ALT-AZ System 2. RA-DEC System

## TRAGECTORY PLANNING

### The five parameters-

- Thrust at five different altitude
- Gravity turn altitude of start and end
- Cone Half Angle
- Rocket Radius
- Initial Wet Mas **ROCKET EQUATIONS**





### BASIC ANATOMY

ORBITAL DYNAMICS

1.Six Keplerian Elements 5. Hohmann Transfer

6. Graveyard Orbit

7. Gravitational Assist

The main parts of a rocket are

2. Lagrange points

3. J2 Nodal Regression

4. J2 Apsidal Rotation

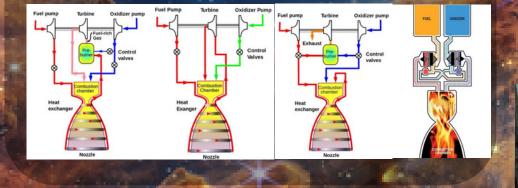
- Engine
- For stability, Cg should always be higher than Cp



**PASSION FOR** 

# Mentors

SHASHANK SINHA

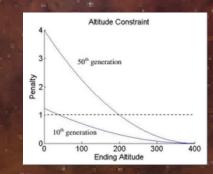




### SSTO RE-ENTRY HEAT SHIELD

- 1. SSTO
- 2. DSTO
- 3. Re-Entry Maneuver
- 4. Heat Shields

### **MOGA TO SSTO**

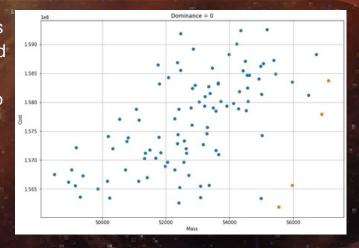


Penalty function is defined as a function of the max altitude of the rocket and it's generation.

The fitness of the model is then dependent on the penalty and the levels of dominance of the model.

## GENETIC ALGORITHM & ROCKET OPTIMIZATION

Genetic algorithms are commonly used to generate highquality solutions to optimization and search problems (with the aid of fitness function & pareto fronts).



PROPELLANTS

Hypergols

**ENGINE PARTS** 

Solid Monopropellant

Combustion Chanber

Liquid

Nozzle

Injector

Abhay singh, Aryan Bharadwaj, Eesh Naugraiya, Manjusree Nayak, Pooja Kumari, Amber Singh, Shubhangi, Sanyam Pasricha, Ashish Patidar, Harshit Bhushan, Mokshagna, Anushree Shukla, Shraman Das, Chitwan Goel, Princy Chauhan, Shrivallabh Pol