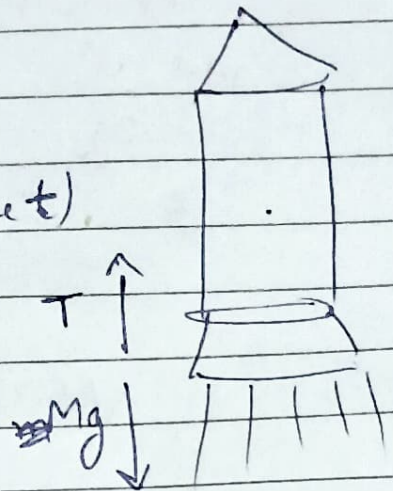


Sol: (i) let us assume initial mass of the rocket is  $M_0$

$$I_{sp} = \frac{T}{\dot{m} g_0} \quad (T = \text{Thrust})$$

$$\Rightarrow \boxed{T = I_{sp} \dot{m} g_0} \quad \text{--- (1)}$$

$$(T - Mg) = Ma \quad \left( \begin{array}{l} M \text{ is the mass at time } t \\ \text{(which is } M_0 - \dot{m}t) \end{array} \right)$$



$$\Rightarrow \frac{T}{M} - g = \frac{dv}{dt}$$

$$\int_0^t \left( \frac{T}{M} - g \right) dt = \int_0^v dv$$

$$\int_0^t \frac{T}{M_0 - \dot{m}t} dt - gt = v$$

$$\int_0^t \frac{I_{sp} \dot{m} g_0}{M_0 - \dot{m}t} dt - gt = v \quad \text{--- (using (1))}$$

$$-I_{sp} g_0 \int_0^t \frac{(-\dot{m})}{M_0 - \dot{m}t} dt - gt = v$$

$$-I_{sp} g_0 \left[ \ln(M_0 - \dot{m}t) \right]_0^t - gt = v$$

$$\boxed{v = I_{sp} g_0 \ln \left( \frac{M_0}{M_0 - \dot{m}t} \right) - gt}$$

Sol: (2)(ii)

Now, we have the eq<sup>n</sup>

$$v = I_{sp} g_0 \ln\left(\frac{M_0}{M_0 - \dot{m}t}\right) - gt$$

Given that ~~at~~  $t = t_b$  ~~fuel~~ is the burnout time

Velocity at that time  $u$  (say) =  $I_{sp} g_0 \ln\left(\frac{M_0}{M_0 - \dot{m}t_b}\right) - gt_b$

①

Height at that point of time,  $H$  (say) can be calculated

~~He~~

$$\frac{dH}{dt} = I_{sp} g_0 \ln\left(\frac{M_0}{M_0 - \dot{m}t}\right) - gt$$

$$\int_0^H dH = \int_0^{t_b} I_{sp} g_0 \ln\left(\frac{M_0}{M_0 - \dot{m}t}\right) dt - \int_0^{t_b} gt dt$$

$$H = I_{sp} g_0 \ln(M_0) t_b + \frac{I_{sp} g_0}{\dot{m}} \left[ \ln(M_0 - \dot{m}t_b) - 1 \right] - \frac{gt_b^2}{2} \quad \text{--- ②}$$

This is the height at burnout time

Maximum height =  $H + \frac{u^2}{2g}$

( $H$  and  $u$  are calculated above)

(By putting these in this eq<sup>n</sup>, we get max height)