

Homework Assignment #06 (Sun and the Orbit)

[Answer, Code]

```
%constants
mu = 3.986004415*10^14;
ae = 6378136.3;
we = 7.292115*10^-5;
g=9.81;
j2=1.082*10^-3;
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%%Preallocations
n = zeros(1,4);
bigWbar = zeros(1,4);
wbar = zeros(1,4);
Mbar = zeros(1,4);
T_kepler = zeros(1,4);
T_anomalistic = zeros(1,4);
u = zeros(1,4);
T_nodal = zeros(1,4);
bigW = zeros(4,361);
beta = zeros(4,361);
```

Problem 1.

Index: [Topex Grace ERS-1 Lageos]

a. Precession Rates:

Node_Precession_Rate =

$[-2.0912e+000 \quad -134.7252e-003 \quad 995.4267e-003 \quad 342.0030e-003]$

Perigree_Precession_Rate =

$[-442.4410e-003 \quad -3.9328e+000 \quad -2.9563e+000 \quad -214.0469e-003]$

Mean_Anomaly_Precession_Rate =

$[4.6198e+003 \quad 5.5452e+003 \quad 5.1599e+003 \quad 2.2989e+003]$

```
%% Problem 1
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% a. Calculate Precession rates in units of degrees/day
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%      1      2      3      4
% [Topex  Grace  ERS-1  Lageos]
a = [7705  6820  7156  12271]*1000; % m
e = [0.001  0.0016  0.001  0.004]; %
i = [65.99  89.02  98.6  109.83]*pi/180; % rad

for j=1:4
n(j) = sqrt(mu/a(j)^3);
bigWbar(j) = -1.5*n(j)*(ae/a(j))^2*j2/sqrt(1-e(j)^2)*cos(i(j));
wbar(j) = -0.75*n(j)*(ae/a(j))^2*j2/(1-e(j)^2)^2*(1-5*(cos(i(j)))^2);
Mbar(j) = n(j)*(1-(0.75*(ae/a(j))^2*j2/(1-e(j)^2)^(3/2)*(1-3*(cos(i(j)))^2)));
end
%convert from rad/s to deg/day
Node_Precession_Rate = (bigWbar)*180/pi*60*60*24
Perigree_Precession_Rate = (wbar)*180/pi*60*60*24
Mean_Anomaly_Precession_Rate = (Mbar)*180/pi*60*60*24
b. Periods:
    Keplerian_Period =

        [112.1808e+000  93.4192e+000  100.4073e+000  225.4653e+000]

    Anomalistic_Period =

        [112.2122e+000  93.4855e+000  100.4677e+000  225.4976e+000]

    Nodal_Period =

        [112.2230e+000  93.5519e+000  100.5253e+000  225.5186e+000]

% b. Calculate the Kepler, Anomalistic, and Draconitic/Nodal Periods
for j=1:4
    T_kepler(j) = 2*pi/n(j);
    T_anomalistic(j) = 2*pi/Mbar(j);
    u(j) = wbar(j) + Mbar(j);
    T_nodal(j) = 2*pi/u(j);
end
Keplerian_Period = T_kepler/60
Anomalistic_Period = T_anomalistic/60

```

Nodal_Period = T_nodal/60

c. Sun-cycle duration:

SunCycle_Duration =

-117.0038e+000 -321.3220e+000 36.8082e+003 -559.3160e+000

% c. Calculate the Sun-Cycle Duration in units of days

bigWs = (360/365.2426); %deg/day

for j=1:4

Cs(j) = 360/(Node_Precession_Rate(j) - bigWs);

end

SunCycle_Duration = Cs

d. Identify Sun-synchronous orbit:

Only the satellite ERS-1 (Satellite #3) is nearly sun-synchronous because its mean sun and nodal precession rate difference is near zero.

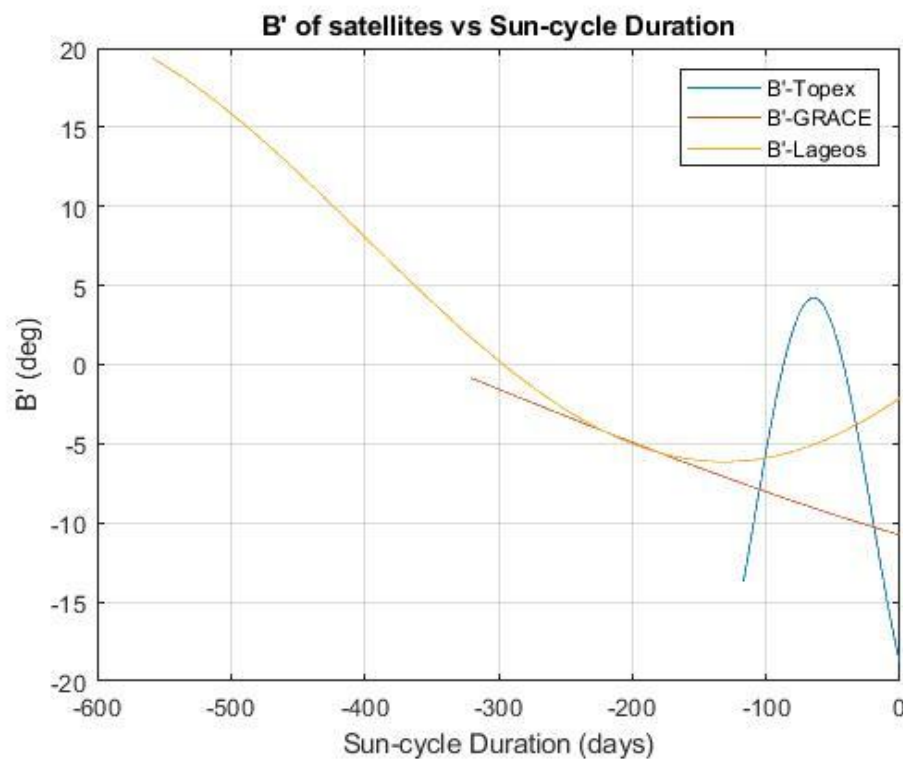
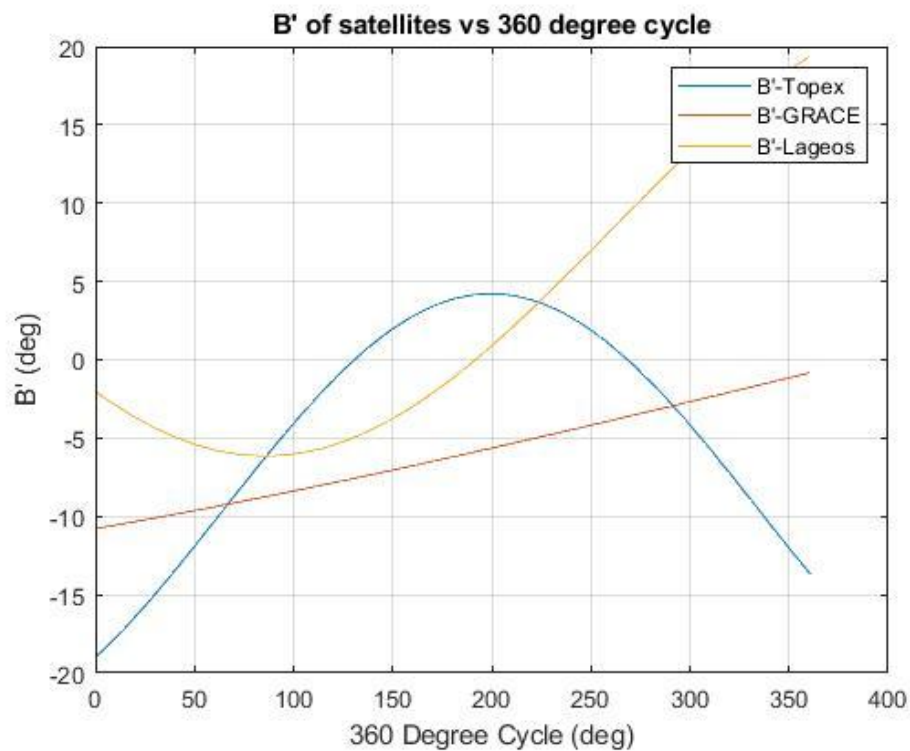
SunSyncTest =

-3.0768e+000 -1.1204e+000 9.7804e-003 -643.6433e-003

% d. Identify if Satellite is in sun-synchronous orbit

SunSyncTest = Node_Precession_Rate - bigWs

Problem 2. Calculate and plot B' for all satellites not in sun-synchronous orbit



Where “negative” days indicates westward propagation, assuming eastward direction is positive.

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%% Problem 2.
% Calculate the B' angle at 1-day intervals and plot its variation within
% one complete cycle of 360 degrees.
% Epoch Noon Jan 1, 2000
P2epoch = [2000 1 1 12 0 0];
%A. (10) Calculate the mean longitude of the sun (L)
JD = juliandate(P2epoch);
D = JD - 2451545.0; %in units of days
%Mean Longitude of Sun:
L = (pi/180)*(280.460 + 0.9856474*D); % rad
Ldeg = rad2deg(L);
%B. Calculate the RA and Dec of the Sun
%Mean Anomaly:
g = pi/180*(357.528 + 0.9856003*D); % rad
%ecliptic longitude:
lambda = L + deg2rad(1.915)*sin(g) + deg2rad(0.020)*sin(2*g);
%Ecliptic Obliquity:
eps = (pi/180)*(23.439 - (4e-7)*D);
%Right Ascension:
RA = atan(cos(eps)*tan(lambda));
RAdeg = rad2deg(RA)
Dec = asin(sin(eps)*sin(lambda));
Decdeg = rad2deg(Dec)
%C. Calculate GMST
GMST = 15*(pi/180)*(18.697374458 + 24.06570982*D); %in units of rad

% Use RA and Dec to calculate e_sun (remains constant)
e_sun = [sin(pi/2 - Dec)*cos(RA),
        sin(pi/2 - Dec)*cos(RA),
        cos(pi/2 - Dec)];

% To calculate e_h, need the change in RAAN
% Change in time = sun cycle duration
bigWo = 0;
%Topex B'
dt1 = 0:(Cs(1)/360):Cs(1);
bigW(1,:) = bigWo + Node_Precision_Rate(1)*(dt1); % degrees
for j=1:361
eh_Topex = [sind(bigW(1,j))*sin(i(1)),
            -1*cosd(bigW(1,j))*sin(i(1)),

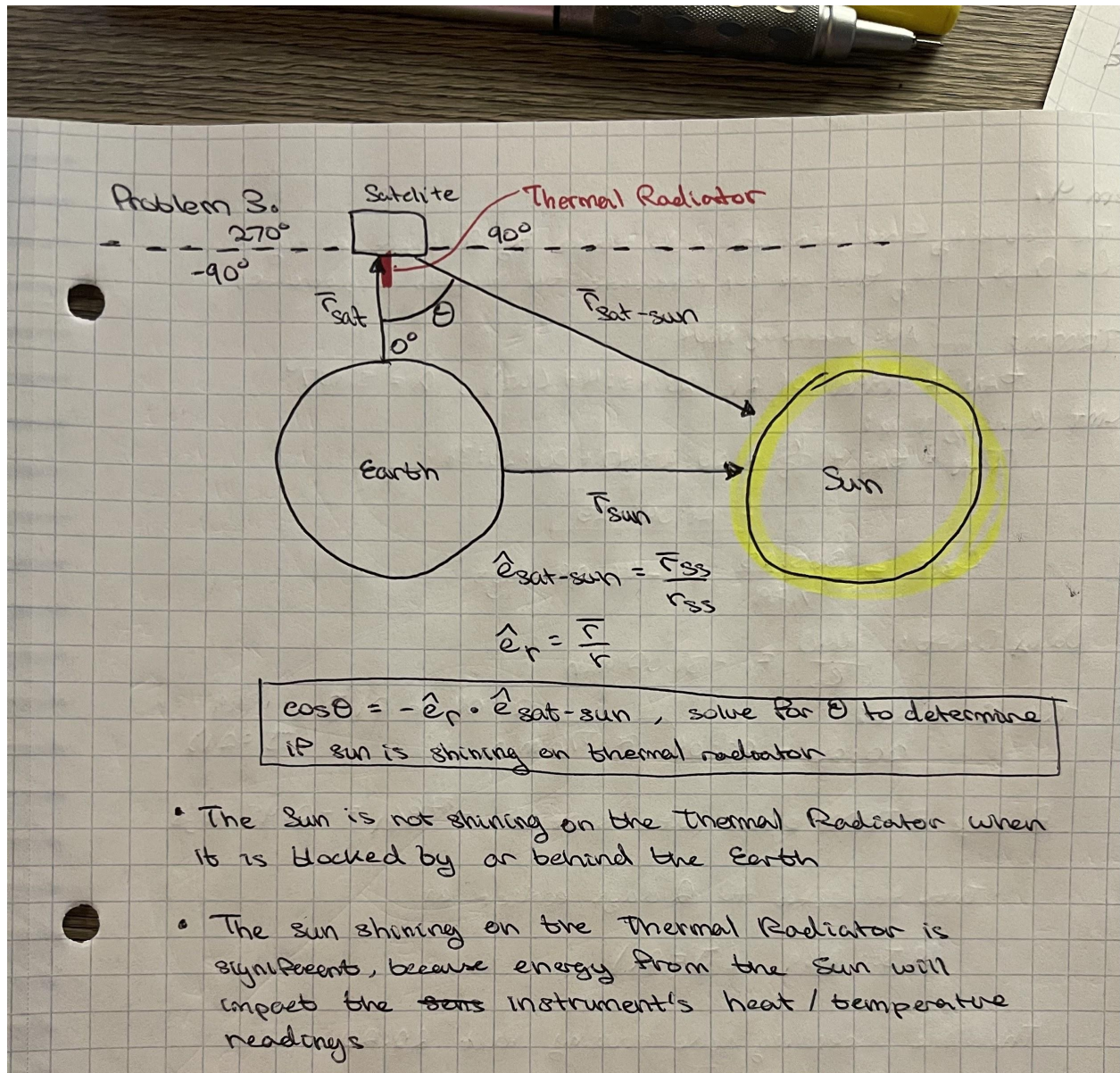
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        cos(i(1))];
beta(1,j) = 90 - acosd(dot(e_sun,eh_Topex)); % in degrees
end
%GRACE B'
dt2 = 0:(Cs(2)/360):Cs(2);
bigW(2,:) = bigWo + Node_Precision_Rate(2)*(dt2); % degrees
for j=1:361
eh_Grace = [sind(bigW(2,j))*sin(i(2)),
            -1*cosd(bigW(2,j))*sin(i(2)),
            cos(i(2))];
beta(2,j) = 90 - acosd(dot(e_sun,eh_Grace)); % in degrees
end
%Lageos B'
dt4 = 0:(Cs(4)/360):Cs(4);
bigW(4,:) = bigWo + Node_Precision_Rate(4)*(dt4); % degrees
for j=1:361
eh_Lageos = [sind(bigW(4,j))*sin(i(4)),
            -1*cosd(bigW(4,j))*sin(i(4)),
            cos(i(4))];
beta(4,j) = 90 - acosd(dot(e_sun,eh_Lageos)); % in degrees
end
Cycleof360 = 1:361;
figure(1)
plot(Cycleof360,beta(1,:),Cycleof360,beta(2,:),Cycleof360,beta(4,:))
title("B' of satellites vs 360 degree cycle")
legend("B'-Topex","B'-GRACE","B'-Lageos")
xlabel("360 Degree Cycle (deg)")
ylabel("B' (deg)")
grid on
figure(2)
plot(dt1,beta(1,:),dt2,beta(2,:),dt4,beta(4,:))
title("B' of satellites vs Sun-cycle Duration")
legend("B'-Topex","B'-GRACE","B'-Lageos")
xlabel("Sun-cycle Duration (days)")
ylabel("B' (deg)")
grid on

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Problem 3.



Problem 4.

Problem 4.

- The Crossing time increases or decreases by 24 hours every C_3 Days, using this one can calculate the increases or decreases

TRMM: AN: Ascending Node

$$t_{AN} = 1999 \cdot 01 \cdot 21 \quad 20:43:47 \text{ (UT)}, \quad \lambda = 5.157^\circ$$

LMT Crossing Time:

$$T_{AN} = t_{AN} + \frac{\lambda}{15} = 20:43:47 + 00:20:38 \\ = 21:04:25$$

$$C_3 = -\frac{365.25}{7.29} = -46.29 \text{ days}$$

Daily Drift: $\frac{1440}{C_3} = -31.02 \text{ min}$

Time Difference: $22:20 - 21:04 \approx 76 \text{ min}$ 21:04 on 21 Jan 1999 (J=21)

Passage of TRMM occurs with shift of $-76/31 = -2.45 \text{ days}$

- This translates to 2 days earlier on

19 Jan 1999 (J=21)

Therefore ascending node crossing about 22:20 occurs on the days J_k given by...

$$J_k = 19 + k |C_3| \dots \\ \text{where } C_3 = -46.29 \text{ days}$$