

CONTROL SYSTEM

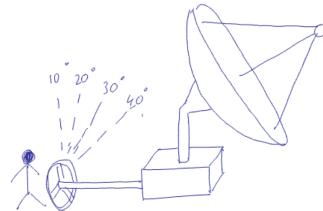
18m ISRO antenna used to track Chandrayaan
(part of Indian Deep Space Network)



This antenna needed to track Chandrayaan.

Q. How to change the azimuth angle precisely according to the motion of Chandrayaan ?

1) Impossible to do manually

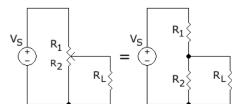


Impossible: Too slow and imprecise

Human inputs can be converted to electrical signals

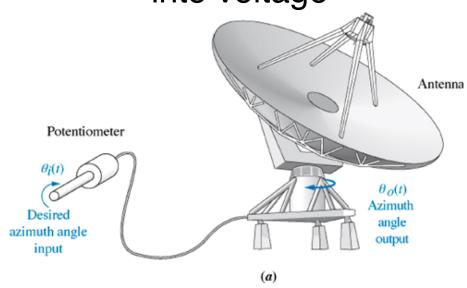
Potentiometer

can help.



Converts angle command into voltage

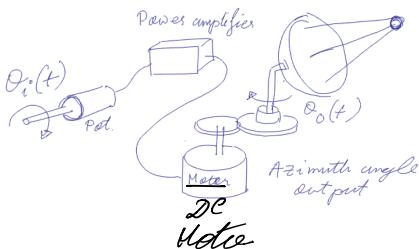
Use POT to convert angle command into voltage



Power Amplification Required

Unrealistic arrangement! WHY?
Answer: Large motor requires a lot of power.

Use Power amplifier to magnify POT output voltage



STILL SOMETHING MISSING! WHAT?

Answer: Error Correction

→ A simplified model of this setup is given to you.

Problem : 1) Design $\theta_o(t)$ to move the Antenna from any initial angle to 5 rad and hold it there automatically.

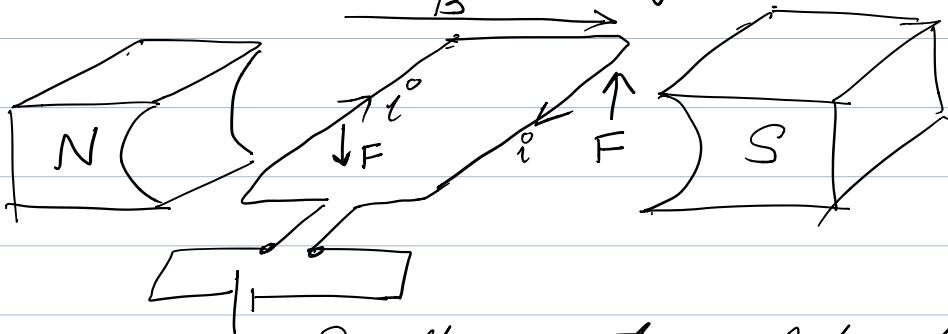
2) how quickly can your control system reach target?

3) how accurately can your control system hold target after sufficient time has passed?

4) What is the maximum deviation from the target that your system is experiencing?

5) The answers to 2,3,4 depend on what parameters? How?

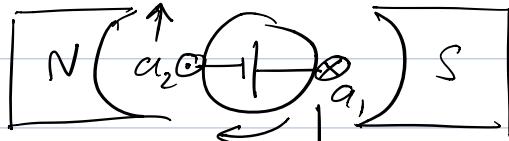
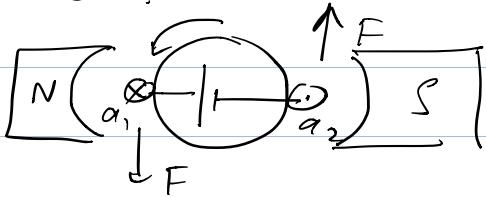
DC - Machine (Principle of Operation)



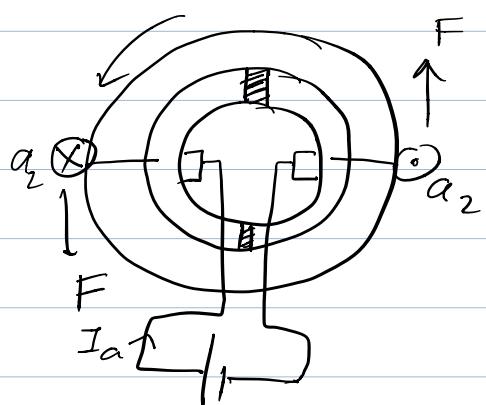
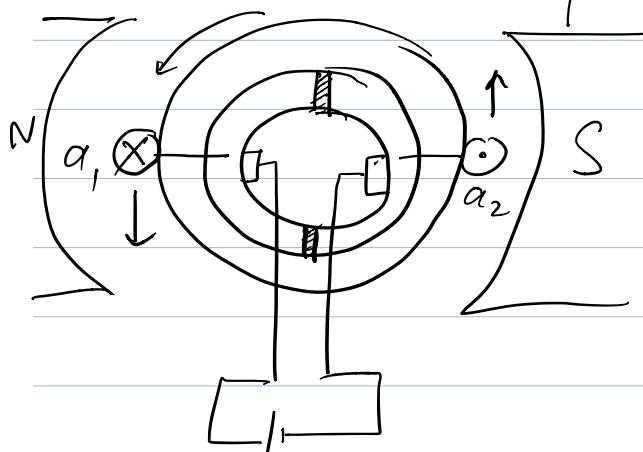
Resulting motion - Anti-clockwise rotation.

Flemings Left Hand rule:

Thrust / Thrust
Field / Fore finger
Current / Centre finger



Commutator trick | See-saw motion - no continuous rotation!



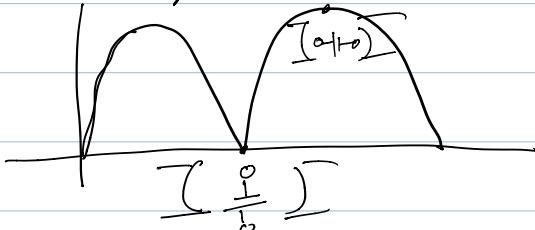
Force Formula: $F = BIL \sin \phi$

$\phi \rightarrow$ angle between B & $I \approx 90^\circ$

See - $F = BIL$ / Torque $\propto F \sin \theta$



Torque $\propto I_a$

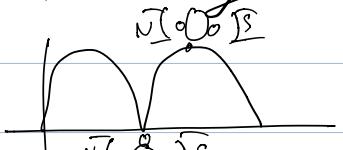


Q. where did the sin theta term go?

Q. So decrease armature resistance $\Rightarrow I_a \uparrow$

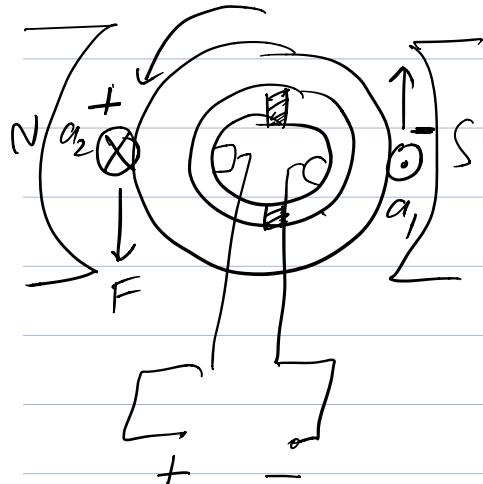
\Rightarrow Torque goes up ??

\rightarrow Is that possible? \rightarrow Faraday's law does not agree!



The motion of the coil in B induces

"Back EMF" $E_b = 2BLV \sin \theta$



single

coil

$E_b = - \frac{d\Phi}{dt} = - \frac{d(BA \cos \theta)}{dt}$

$= - BA \sin \theta \frac{d\theta}{dt} = - BA \omega \sin \theta$

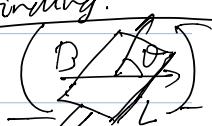
Faraday's right hand rule

Field

Motion

Current

Real motor
get rid of the
 $\sin \theta$ term by
using multiple
poles & clever
winding.



$$E_b \propto \omega$$

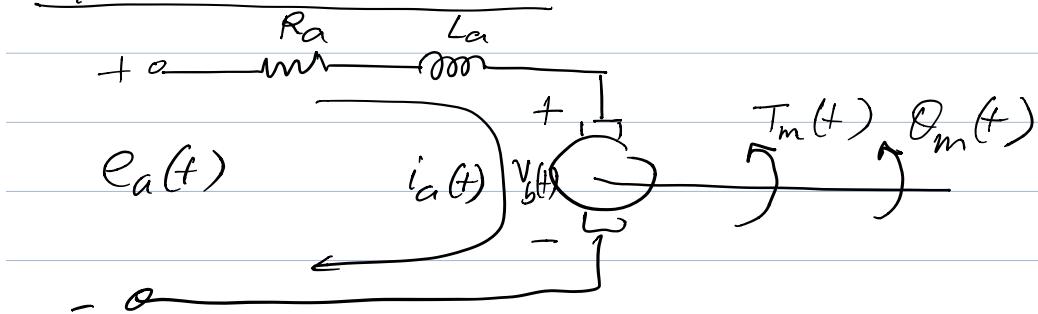
Show Motor Animation

For a real DC machine, poles & conductors are arranged carefully to hold the torque nearly constant as the armature rotates.

$$T_m = K_t I_a \quad \text{--- (1)}$$

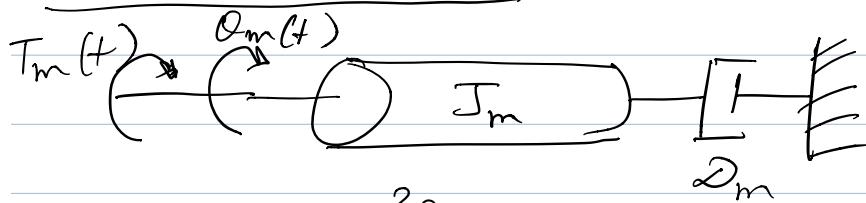
$$V_b = K_b \frac{d\theta_m}{dt} = K_b \omega_m \quad \text{--- (2)}$$

Equivalent Circuit:



$$R_a i_a(t) + L_a \frac{di_a(t)}{dt} + V_b(t) = e_a(t) \quad \text{--- (3)}$$

Mechanical load



$$T_m(t) = J_m \frac{d^2\theta_m}{dt^2} + D_m \frac{d\theta_m}{dt} \quad \text{--- (4)}$$

Using (1) and (2) in (3),

$$\frac{Ra}{Kt} T_m(t) + \frac{La}{Kt} \frac{d}{dt} [T_m(t)] + K_b \frac{d\theta_m(t)}{dt} = e_a(t)$$

Assume La to be small,

$$\frac{Ra}{Kt} T_m(t) + K_b \frac{d\theta_m(t)}{dt} = e_a(t) \quad \text{--- (5)}$$

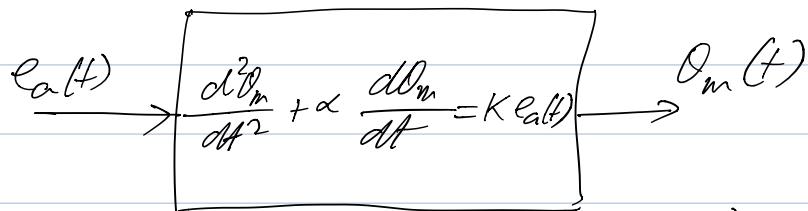
Using (4) in (5),

$$\frac{Ra}{Kt} \left[J_m \frac{d^2 \theta_m}{dt^2} + \theta_m \frac{d\theta_m}{dt} \right] + K_b \frac{d\theta_m}{dt} = e_a(t)$$

$$\frac{Ra J_m}{Kt} \frac{d^2 \theta_m}{dt^2} + \left[\frac{Ra \theta_m}{Kt} + K_b \right] \frac{d\theta_m}{dt} = e_a(t)$$

$$\text{Let } K = \frac{Kt}{Ra J_m}; \quad \alpha = \frac{1}{J_m} \left(\theta_m + \frac{Kt K_b}{Ra} \right)$$

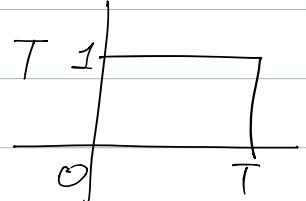
$$\text{Then } \frac{d^2 \theta_m}{dt^2} + \alpha \frac{d\theta_m}{dt} = K e_a(t)$$



1) For a step voltage, $e_a(t) = \begin{cases} 1 & t=0 \\ 0 & t>0 \end{cases}$

$$\theta_m(t) = A + Bt + C e^{-\alpha t}$$

2) For a square pulse of duration T
 - Check in the simulator

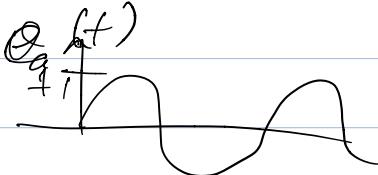


$$\theta_m(t) = At - B u(t-T) \left[\frac{A}{B} (t-T) + D e^{-\alpha(t-T)} + E \right] + BE + Ce^{-\alpha t}$$

so $\theta_m(t) \approx \theta_m(T)$ for $t \gg T$.

3) For sinusoidal input:

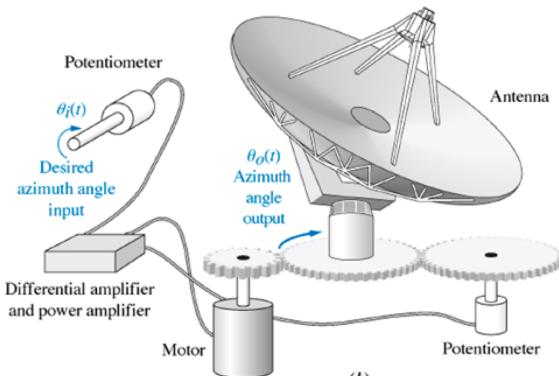
$$\theta_m(t) = A + B \sin \omega t + C \cos \omega t - D e^{-\alpha t}$$



FACT: These calculations are not enough to successfully "point" the antenna at 5 rad.
- especially if there are disturbances.

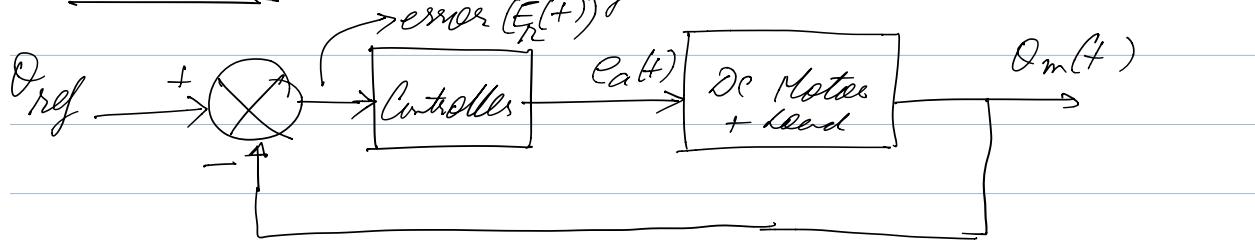
- Must know where we are at any pt
- Then only we can keep correcting

The central theme: feedback

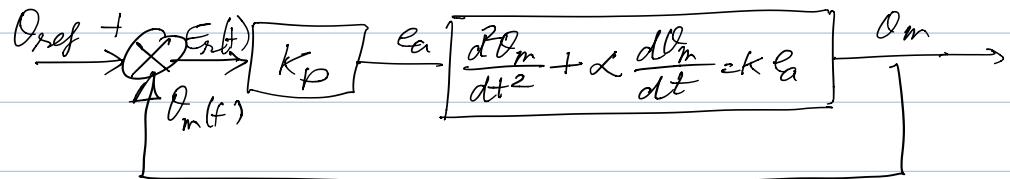


- Use another gear/pot to check whether actually the output is following the input.
- If not then use the difference/error to drive the system

Feedback: The basis of automatic control



Let's try with the simplest possible controller:
Just a gain $\rightarrow K_p$

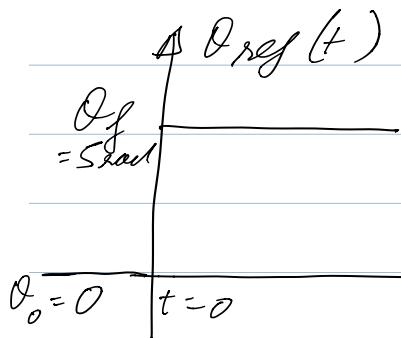


$$\begin{aligned} E_r(t) &= \theta_{ref} - \theta_m \\ e_a(t) &= K_p E_r(t) \end{aligned}$$

$$\Rightarrow \frac{1}{KK_p} \left[\frac{d^2\theta_m}{dt^2} + \alpha \frac{d\theta_m}{dt} \right] = \theta_{ref} - \theta_m$$

$$\Leftrightarrow \boxed{\frac{d^2\theta_m}{dt^2} + \alpha \frac{d\theta_m}{dt} + (KK_p)\theta_m = KK_p \theta_{ref}}$$

Various $\theta_{ref}(t)$ are possible. Let's try with a step input (solv. to one problem of starting from θ_0 & bringing θ_m to θ_f as quickly as possible)

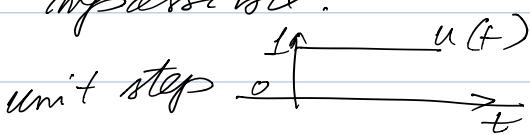


Q. What will the response $O_m(t)$ look like?

A: We would like $O_m(t)$ to be exactly like $O_{ref}(t)$.
— But that is impossible.

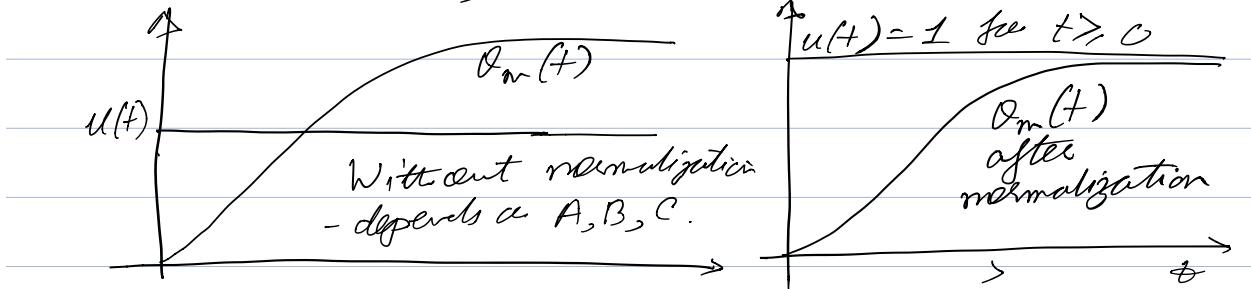
For $O_{ref} = u(t)$, solution depends on roots of $D^2 + \alpha D + K_k p = 0$

$$D_{1,2} = -\frac{\alpha \pm \sqrt{\alpha^2 - 4K_k p}}{2}$$



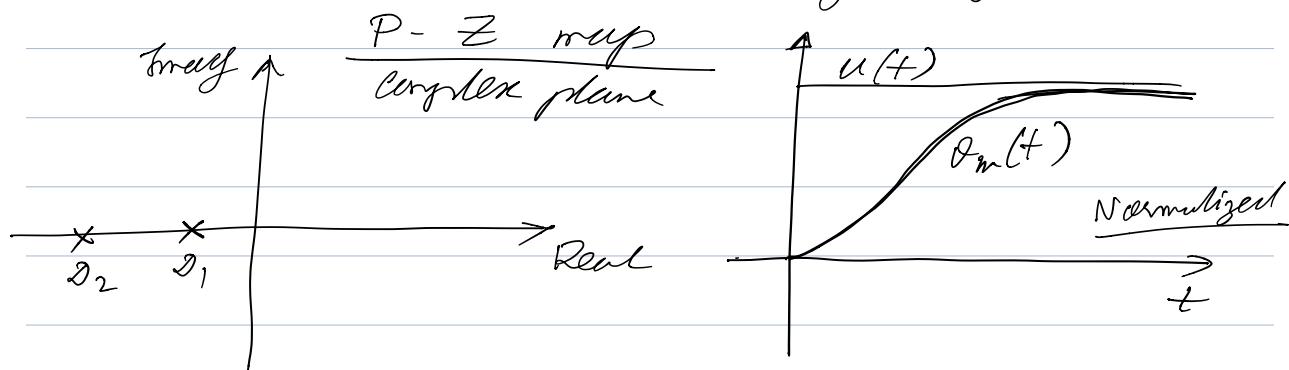
Depending on what K_p you choose, the roots $D_{1,2}$ can either be

1) Real & distinct (D_1, D_2) $D_1, D_2 < 0$
- then $O_m(t) = A e^{D_1 t} + B e^{D_2 t}$



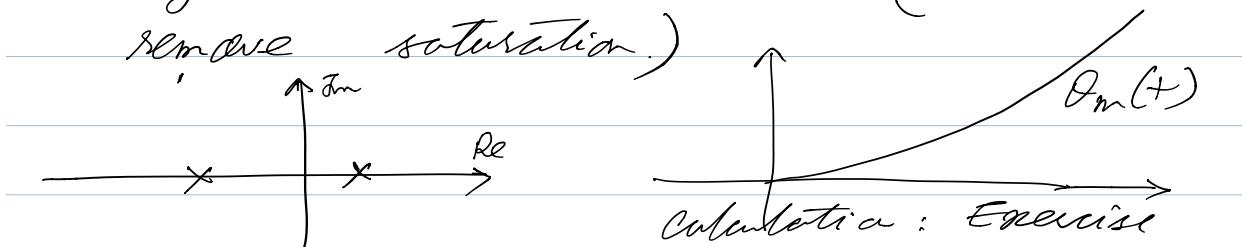
Such a response is called "OVERDAMPED"

In controls, we try to make a direct correspondence between the location of D_1, D_2 & the nature of response.



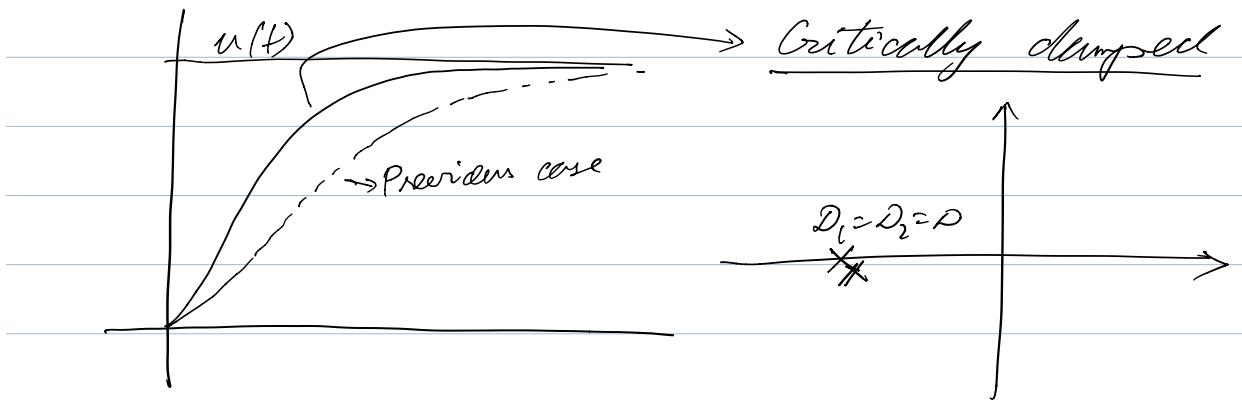
Q. Can any K_p make D_1/D_2 positive?

- Try it on the simulation (Have to remove saturation.)



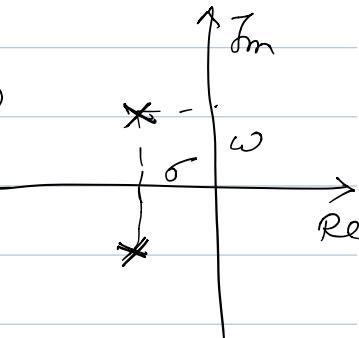
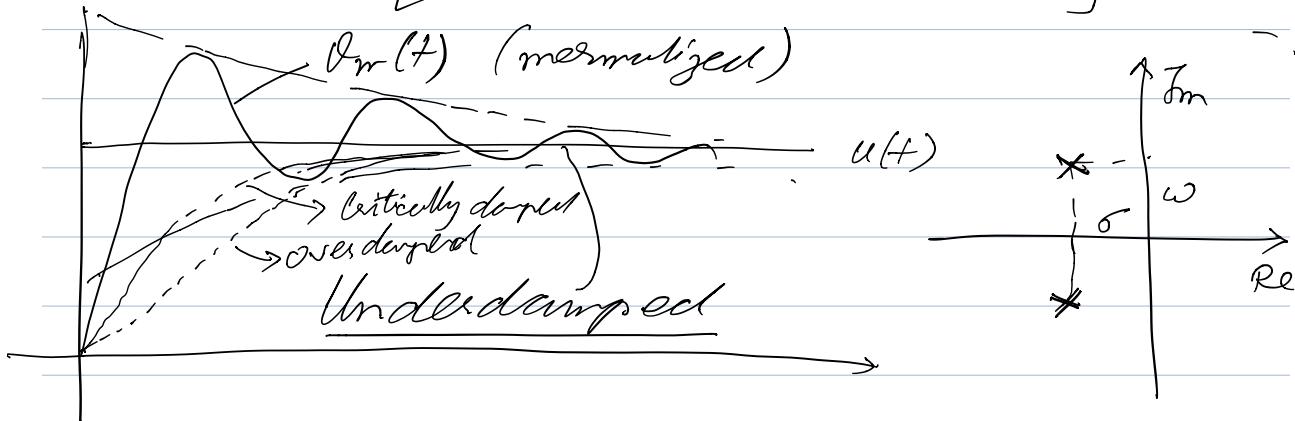
2) Real & repeated $D_1 = D_2 = D$

- then $\Omega_m(t) = A + B e^{Dt} + C t e^{Dt}$



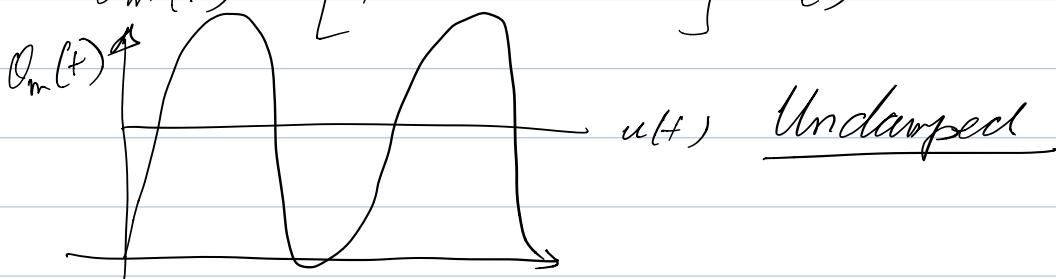
3) Complex Poles: $\sigma_{1,2} = \sigma \pm j\omega$ ($\sigma < 0$)

Then $\theta_m(t) = [A + Be^{\sigma t} \cos(\omega t - c)] u(t)$



4) Purely Imaginary Poles: $\sigma_{1,2} = \pm j\omega$

$$\theta_m(t) = [A + \cos \omega t] u(t)$$



Q. Can you get this response in the simulation?
 → Try to justify.

IMPORTANT QUESTION: You could do
 (see) all these by trial & error.

Why do you need the math?

→ Cannot tune (in most cases) on the real device

→ We will see systems where manual tuning is too hard.

Q. What can K_p do?

Recall : $\rho_{1,2} = -\frac{\alpha}{2} \pm \frac{1}{2}\sqrt{\alpha^2 - 4KkP}$ $(\alpha, K, kP > 0)$

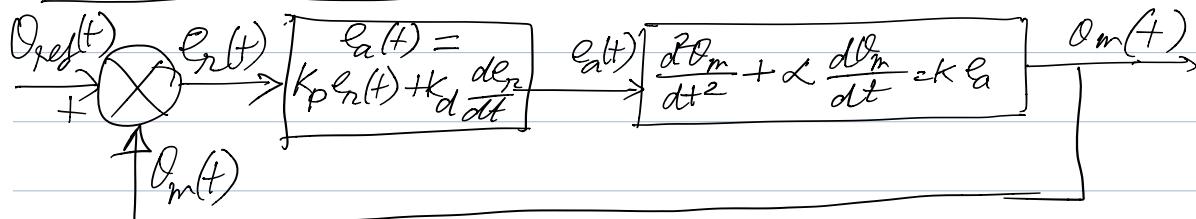
- $K_p < \frac{\alpha^2}{4K}$ → real distinct roots (overdamped)
→ Over damped — Very slow response
- $K_p = \frac{\alpha^2}{4K}$ → real repeated roots
still slow — (critically damped)

→ $K_p > \frac{\alpha^2}{4K}$ → complex conjugate — faster but oscillatory — oscillations not good.

Show Antenna animation

we want fast but no oscillation
— Can we add damping? — Remember that system cannot be changed!

P - D Controllers



$$e_r(t) = \theta_{ref}(t) - \theta_m(t) \quad \text{--- (1)}$$

$$e_a(t) = K_p e_r(t) + K_d \frac{d\theta_r(t)}{dt} \quad \text{--- (2)}$$

Using (1) in (2), $e_a(t) = K_p (\theta_{ref}(t) - \theta_m(t)) + K_d \frac{d}{dt} (\theta_{ref}(t) - \theta_m(t))$

Then

$$e_a(t) = K_p \theta_{ref} - K_p \theta_m(t) - K_d \frac{d\theta_m(t)}{dt} + K_d \frac{d\theta_{ref}}{dt}$$

Using (3) in the system eqn:

$$\frac{d^2\theta_m}{dt^2} + \zeta \frac{d\theta_m}{dt} = K \left[K_p \theta_{ref} - K_p \theta_m(t) - K_d \frac{d\theta_m(t)}{dt} + K_d \frac{d\theta_{ref}}{dt} \right]$$

$$\Leftrightarrow \frac{d^2\theta_m}{dt^2} + (\zeta + KK_d) \frac{d\theta_m(t)}{dt} + KK_p \theta_m(t) = KK_p \theta_{ref} + KK_d \frac{d\theta_{ref}}{dt}$$

Q. How to analyse such diff. eqns? \rightarrow Will learn later
 For the moment assume $\theta_{ref}(t) = \text{constant}$ (Step - input
 sorry about the lack of rigour)

$$\Leftrightarrow \frac{d^2\theta_m}{dt^2} + (\zeta + KK_d) \frac{d\theta_m(t)}{dt} + KK_p \theta_m(t) = KK_p \theta_{ref}$$

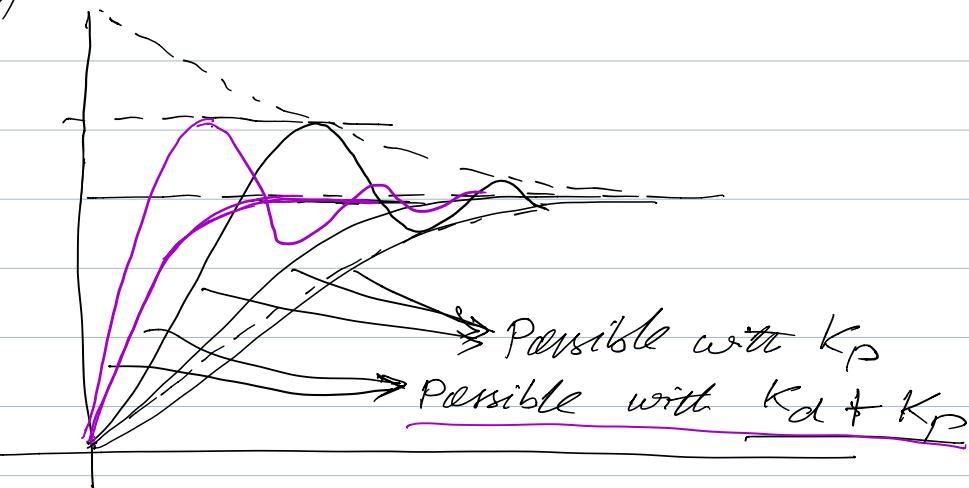
We successfully added damping without changing the original system.

The solution will depend on the roots of

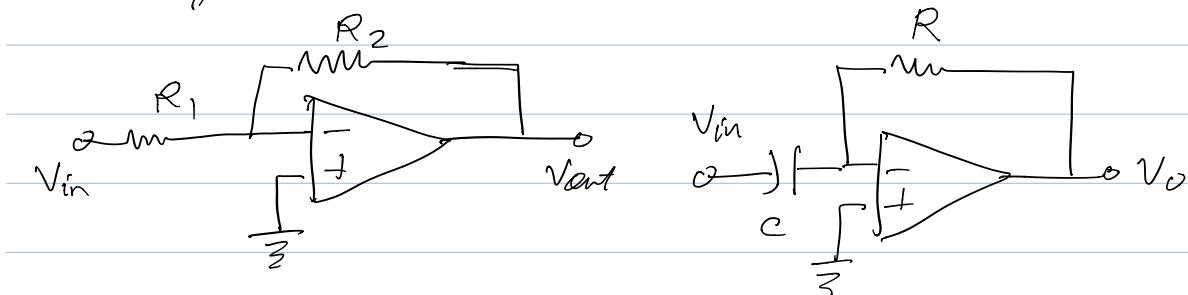
$$\mathcal{D}^2 + (\alpha + KK_d)\mathcal{D} + KK_p = 0$$

$$\mathcal{D}_{1,2} = -\frac{(\alpha + KK_d)}{2} \pm \frac{1}{2}\sqrt{(\alpha + KK_d)^2 - 4KK_p}$$

Note that we can change the real part of the roots



Analog circuit implementation (Ideal)

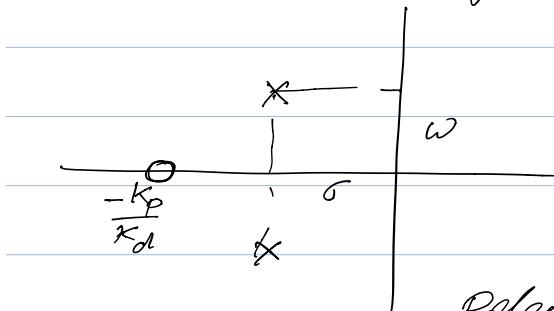


$$V_o = -\frac{R_2}{R_1} V_{in}$$

$$V_o = -R C \frac{dV_{in}}{dt}$$

Very rarely used — usually digitally implemented in micro-controllers (?) .

Q. Where is the pole zero map?



$$\frac{d^2\theta_m}{dt^2} + (\zeta + KK_d)\frac{d\theta_m}{dt} + KK_p$$

$$= KK_d \frac{d\theta_{ref}(t)}{dt} + KK_p \theta_{ref}(t)$$

Poles \rightarrow Roots of the LHS = 0

Zeros \rightarrow Roots of the RHS = 0

$$\sigma^2 + (\zeta + KK_d)\sigma + KK_p = 0 \rightarrow \sigma_{1,2} = \sigma \pm j\omega$$

$$KK_d\sigma + K_p = 0 \rightarrow \sigma_1 = -\frac{K_p}{KK_d}$$

Fundamental question: How to place poles & zeros to have "good" response?

\Leftrightarrow How to choose K_p & K_d to have "good" response.

Q. What is good response?

— Control specifications

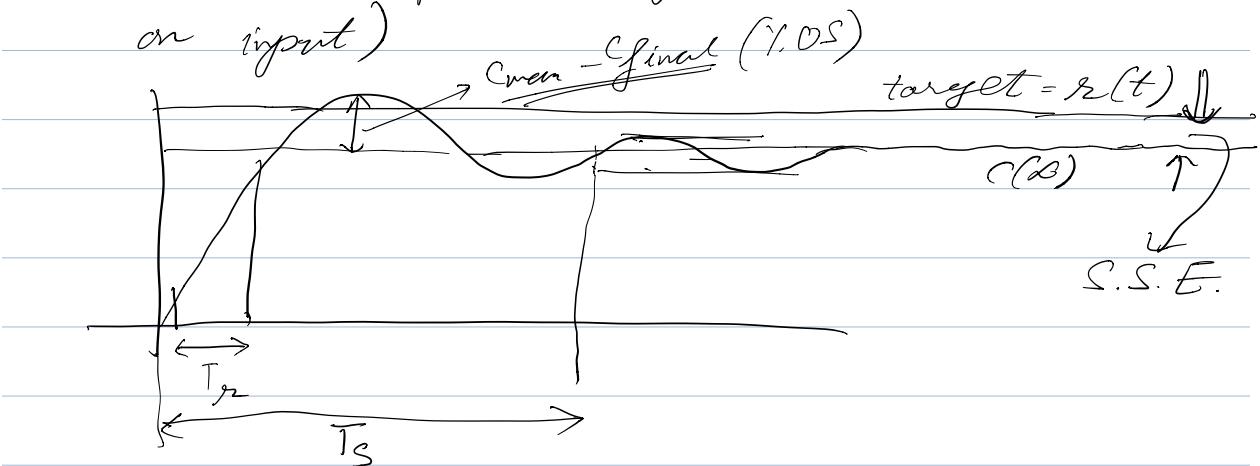
1) Rise time \rightarrow time for the response to rise from 10% to 90% of its final value

2) Settling Time \rightarrow time taken by response to reach & stay within 2% of final value

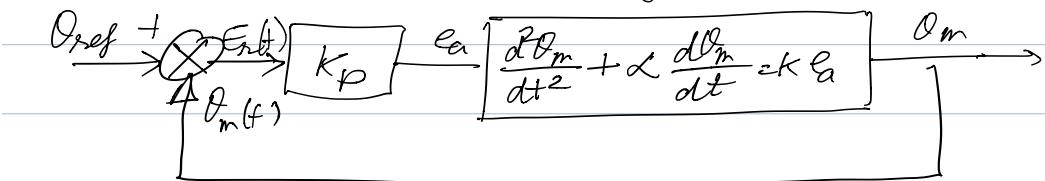
3) Percentage Overshoot: $\%OS = \frac{C_{max} - C_{final}}{C_{final}} \times 100\%$

(1, 2, 3 are for step input)

4) Steady state error: SSE is the difference between output & input as $t \rightarrow \infty$ (depends on input)



Q. Did we have SSE in our example?
 → Proportional feedback case



$$E_r(t) = R_{ref} - \theta_m \quad \frac{d\theta_m}{dt^2} + \zeta \frac{d\theta_m}{dt} = k_e e_r$$

$$e_a(t) = K_p E_r(t)$$

$$\frac{d^2(\theta_{ref} - e_r)}{dt^2} + \zeta \frac{d(\theta_{ref} - e_r)}{dt} = K K_p e_r$$

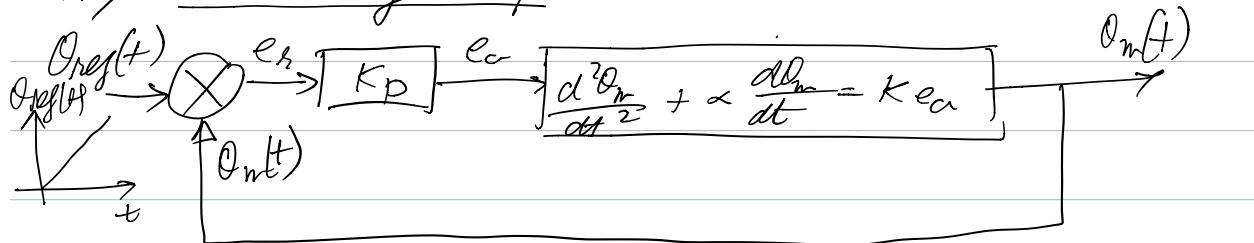
$$\frac{d^2 e_r}{dt^2} + \zeta \frac{de_r}{dt} + K K_p e_r = 0$$

$$\Rightarrow e_r(t) \rightarrow 0 \text{ as } t \rightarrow \infty$$

So no error at steady state.

Speed Control - Need for PI

A) With only K_p : θ



$$\Omega_{ref}(t) = \omega_{ref} t$$

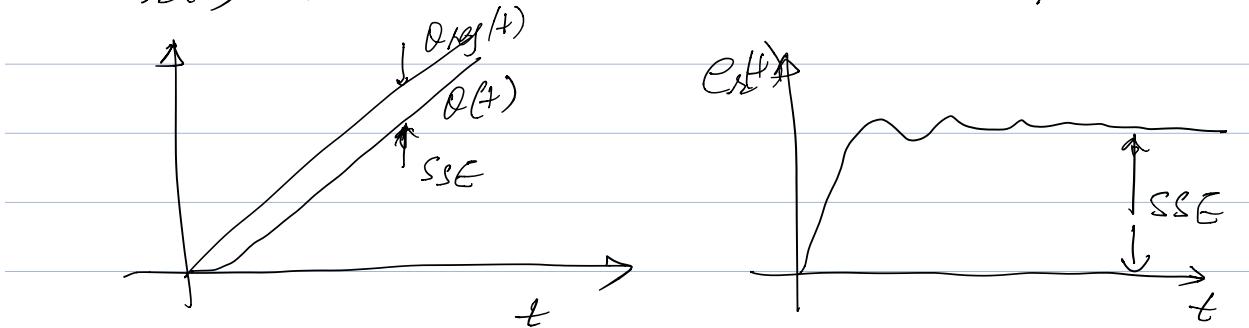
$$\text{Same steps as above: } \frac{d^2(\Omega_{ref} - \Omega_r)}{dt^2} + \alpha \frac{d(\Omega_{ref} - \Omega_r)}{dt} = K_p e_r$$

$$\Leftrightarrow -\frac{d^2 \Omega_r}{dt^2} + \alpha \omega_{ref} - \alpha \frac{d \Omega_r}{dt} = K_p e_r$$

$$\Leftrightarrow \frac{d^2 \Omega_r}{dt^2} + \alpha \frac{d \Omega_r}{dt} + K_p e_r = \alpha \omega_{ref}$$

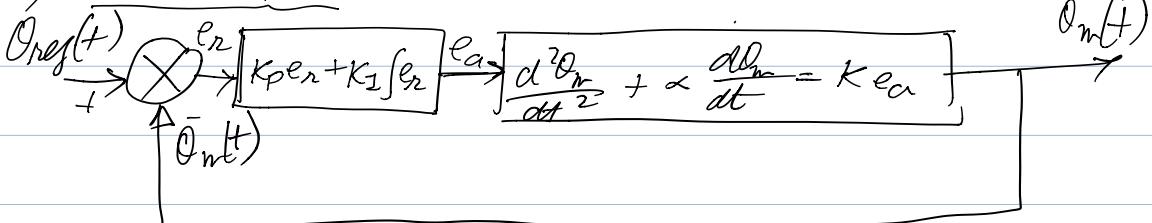
Solution : $e_r(t) = \begin{cases} A + Be^{\sigma_1 t} + Ce^{\sigma_2 t} & (\sigma_1 \neq \sigma_2 \text{ real}) \\ A + Be^{\sigma t} + Cte^{\sigma t} & (\sigma_1 = \sigma_2 = \sigma \text{ real}) \\ A + Be^{\sigma t} \cos(\omega t + c) & (\sigma \pm j\omega) \end{cases}$

$e_r(t) \rightarrow A$ as $t \rightarrow \infty \Rightarrow \text{SSE} \neq 0$.



Q. How to get SSE to zero?
 → Use PI.

B) With PI



$$\frac{d^2\theta_m}{dt^2} + \alpha \frac{d\theta_m}{dt} = K \left(K_p e_r + K_I \int_0^t e_r dt \right)$$

$$\frac{d^2(\theta_{ref}(t) - e_r)}{dt^2} + \alpha \frac{d(\theta_{ref}(t) - e_r)}{dt} = K \left(K_p e_r + K_I \int e_r dt \right)$$

For $\theta_{ref}(t) = \omega_{ref} t$,

$$-\frac{d^2 e_r}{dt^2} + \alpha \omega_{ref} - \alpha \frac{de_r}{dt} = K K_p e_r + K K_I \int_0^t e_r dt$$

Dif. again:

$$\frac{d^3 e_r}{dt^3} + \alpha \frac{d^2 e_r}{dt^2} + \underbrace{KK_p}_{\beta} \frac{de_r}{dt} + \underbrace{KK_I}_{\gamma} e_r = 0$$

→ 3rd order diff. eqn.: But no forcing function.

We guess that a solution is of the form:

$$e_r(t) = e^{\lambda t}$$

$$\text{Then } D^3 e^{\lambda t} + \alpha D^2 e^{\lambda t} + \beta D e^{\lambda t} + \gamma e^{\lambda t} = 0$$

$$\Leftrightarrow D^3 + \alpha D^2 + \beta D + \gamma = 0$$

Let the roots be $\lambda_1, \lambda_2, \lambda_3$.

$$e_r(t) = A e^{\lambda_1 t} + B e^{\lambda_2 t} + C e^{\lambda_3 t} \quad (\text{real distinct})$$

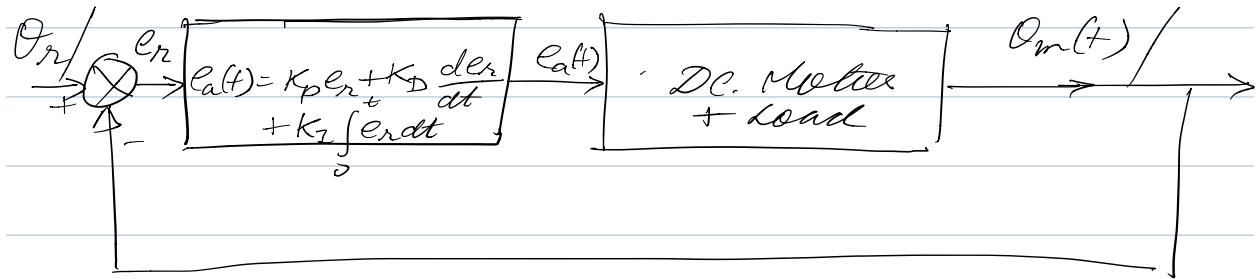
$$e_r(t) = A e^{\lambda_1 t} + B e^{\lambda_2 t} + C t e^{\lambda_3 t} \quad (\text{real, } \lambda_1 \neq \lambda_2 = \lambda_3)$$

$$e_r(t) = A e^{\lambda_1 t} + B t e^{\lambda_1 t} + C t^2 e^{\lambda_1 t} \quad (\lambda_1 = \lambda_2 = \lambda_3 \text{ real})$$

$$e_r(t) = A e^{2\sigma t} + B e^{\sigma t} \cos(\omega t + C) \left(\begin{matrix} \sigma_1 \text{ real} \\ \sigma \pm j\omega \end{matrix} \right)$$

In each case, $e_r(t) \rightarrow 0$ as $t \rightarrow \infty$.

So: final feedback controller:



PID controllers are used in most Control Systems

Q. How to tune? 3 parameters \rightarrow many possibilities.

Q. How to ensure that your design will work in the real world? \rightarrow for inaccurate models & arbitrary disturbances.

\rightarrow Better answers in 3rd year - EE302 / EE324.