

# ROS-Based Multi-Agent Systems Control Simulation Testbed (MASCOT)

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# Overview

- 1 Introduction
- 2 Preliminaries
- 3 MASCOT: Structure and Features
- 4 Examples
- 5 Conclusion and Future Work

# Multi-Agent Systems

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A system consists of multiple co-operative agents interacting with each other.

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## Applications:

Robotics, space missions, search and exploration, surveillance, agriculture etc.

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## Existing MAS Simulation:

- MATLAB based simulators.
- Limitation of no. of agents.
- Not readily deployable of hardware.



# MASCOT

- Developed using open source tools.
- ROS and Gazebo.
- Supports low level driver.
- Simple user interface.
- In this version Quadcopter as an agent.
- Multiagent system with double integrator.
- Easy to setup with Docker Support.

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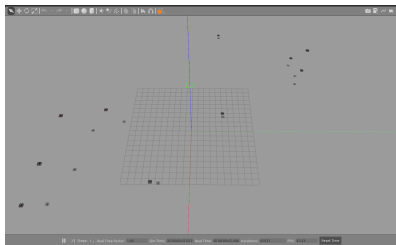


Figure: Initial Position of Drones

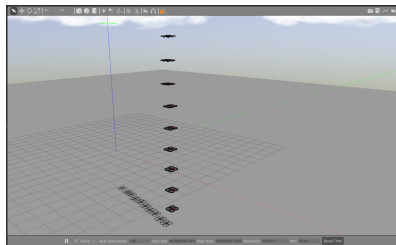


Figure: Final Position

# Preliminaries

# Frame of Reference

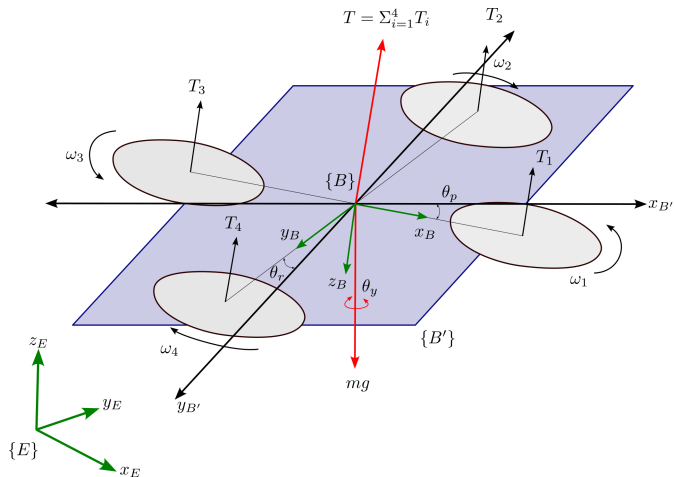


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# Quadcopter Dynamics

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- Overall quadcopter motion equation

$$[\mathbf{T} \quad \boldsymbol{\Gamma}]^T = 4 \begin{bmatrix} \bar{\omega}_1^2 & \bar{\omega}_2^2 & \bar{\omega}_3^2 & \bar{\omega}_4^2 \end{bmatrix}^T \quad (7)$$

# Quadcopter Dynamics as Double Integrator

- Total force on quadcopter

$$\mathbf{f}^{B'} = \mathbf{R}_x(\theta_r) \mathbf{R}_y(\theta_p) \begin{bmatrix} 0 & 0 & T \end{bmatrix}^T$$

- Thus we get  $\mathbf{f}^{B'}$  as

$$\mathbf{f}^{B'} = \begin{bmatrix} T \sin \theta_p \\ T \sin \theta_r \cos \theta_p \\ T \cos \theta_r \cos \theta_p \end{bmatrix}$$

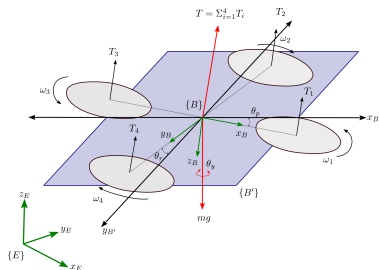


Figure: Quadcopter Dynamics

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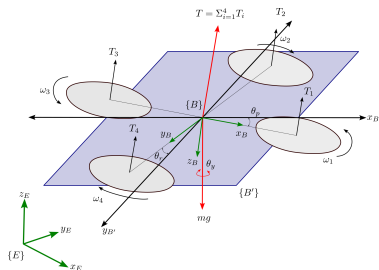


Figure: Quadcopter Dynamics

- For small  $\theta_p$  and  $\theta_r$  the  $\mathbf{f}^{B'}$  can be approximated by

$$\mathbf{f}^{B'} \approx [T\theta_p \quad T\theta_r \quad T]^T$$

- With this assumption the Quadcopter can be assumed as a double integrator system where  $\theta_p$  and  $\theta_r$  are given by

$$\theta_p = \frac{m}{T} a_x^{B'}, \quad \theta_r = \frac{m}{T} a_y^{B'}$$

# MASCOT:Structure and Features

# Tools Used

## ROS:

- Open source robotics framework.
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## TUM Simulator Package:

- Uses the AR Parrot drone model.
- Low level plugin is modified as per the Double integrator dynamics.
- Added the required topics and controls.



# Control Block

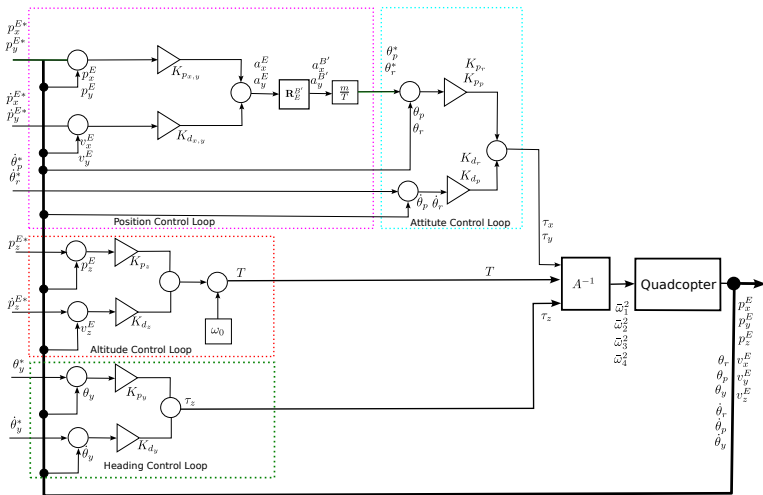


Figure: Control Block

# Architecture

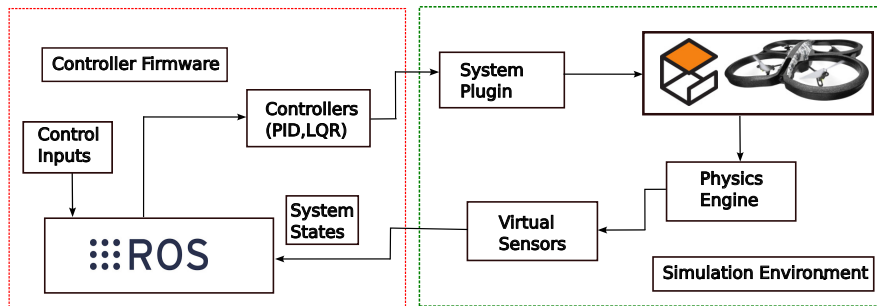


Figure: System Architecture

- Gazebo internal scheduler provides the ROS interface.
- ROS works as middleware which runs independent controller for each agent.
- The intercommunication uses TCPROS protocol.

# Feature and Configuration of Simulation Testbed

## Feature

- Easy Modification.
- Supports multiple languages Python, Cpp, Java.
- Flexibility with no. of agents.

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## Configuration

- **Robot:** Details of the Robots to be simulated
  - **Number:** No. of Agents.
  - **InitialPosition:** Enable initializer.
  - **Position:** Initial Position.
- **Output:** Output config
  - **Velocity :** Generate Vel plot.
  - **Position :** Generate Vel plot.
  - **Save-plot :** Save plots.
  - **Show-plot :** Show plots.
  - **Save-data :** Save Numpy
- **Control:** Controls laws
  - **Custom-Control:**
  - **Tutorial Examples:**
    - \* **Waypoint Navigation:**
      - **P-Gain:** Default = 1.0
      - **D-Gain:** Default = 1.0
    - \* **Consensus:**
      - **Leader:** Robot index to be leader, 0-for leaderless.
      - **Communication Graph:**
      - **L-mat:** Laplacian Matrix.
    - \* **Min-max Consensus:**

# Examples

# Way-Point Navigation

- The position of the quadcopter in  $x^{B'}y^{B'}$  plane is controlled independently by the proportional-derivative controller for each axis.

$$a_x = K_{p_x} (p_x^* - p_x) + K_{d_x} (\dot{p}_x^* - \dot{p}_x)$$

$$a_y = K_{p_y} (p_y^* - p_y) + K_{d_y} (\dot{p}_y^* - \dot{p}_y)$$

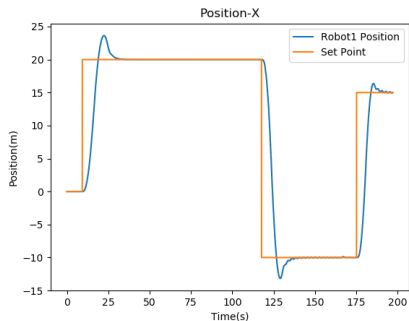


Figure: X-axis Position plot of Waypoint Navigation

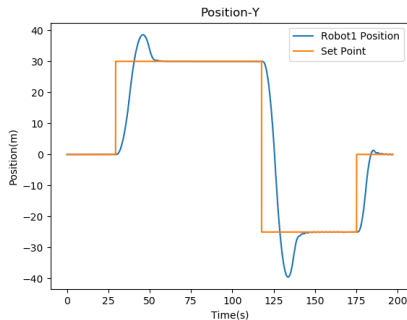


Figure: Y-axis Position plot of Waypoint Navigation

# Consensus Algorithm (Linear)

- A leaderless asymptotic consensus and leader follower is implemented.
- The control algorithms used is as follows:

$$\mathbf{f}_i^E = \begin{cases} \sum_{j=1}^n a_{ij} (\mathbf{p}^j^E - \mathbf{p}^i^E) - \beta \mathbf{v}_i^E & \text{if } \alpha_i \in \mathbf{F} \\ 0 & \text{if } \alpha_i \in \mathbf{L} \end{cases}$$

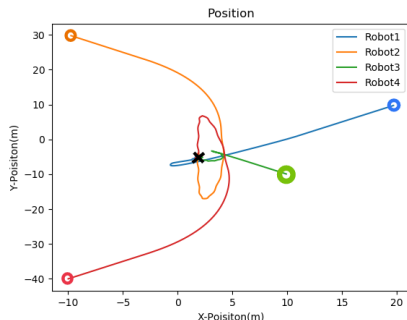


Figure: Leaderless Control plot

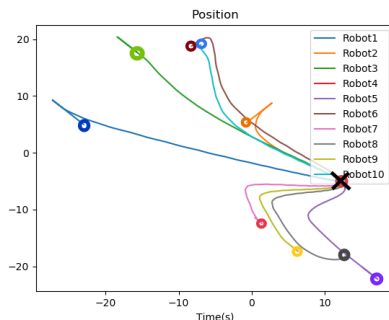


Figure: Leader Follower plot

# Min-Max Time Consensus

- A non linear Min-Max time consensus Algorithm is implemented.
- The Control Law used is

$$\mathbf{f}_c^E = \beta_c \text{sign}(2(\beta_c - \beta_p)(\mathbf{p}_c - \mathbf{p}_p) + (\mathbf{v}_c - \mathbf{v}_p)^2 \text{sign}(\mathbf{v}_c - \mathbf{v}_p))$$

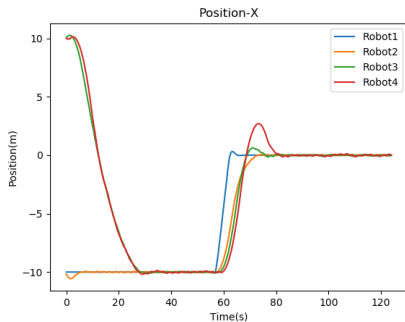


Figure: Position in X axis

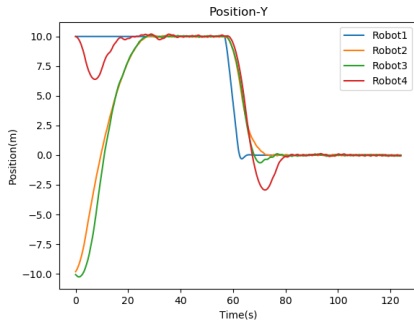


Figure: Position in Y axis



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## Future Work

- System is Open-source and expandable.
- UGVs and different UAVs can be deployed.
- Human In the loop control.
- Deployment on real hardware.

# Thank You



Figure: <https://github.com/Avi241/mascot>