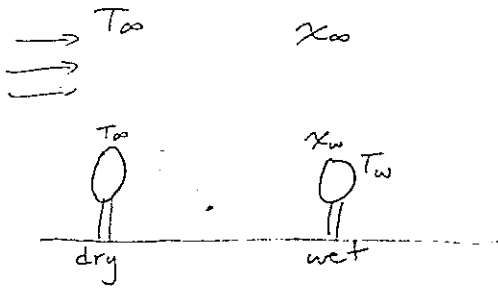


Transport #2(a.)

Wet-bulb thermometers



Water evaporates from wet bulb; ΔH^{vap} lowers wet bulb's temperature w.r.t. T_{∞} .

Fick's Law
 $q = n(T_{\infty} - T_w) =$

$$N_{H_2O} = kC(x_w - x_{\infty})$$

$$k(T_{\infty} - T_w) = (N_{H_2O})(MW_{H_2O})(\Delta H_{H_2O}^{vap})$$

$$\frac{k(T_{\infty} - T_w)}{k} = C(x_w - x_{\infty})(MW_{H_2O})(\Delta H^{vap})$$

— Chilton - Colburn Analogy (betw. Mass, Heat, Momentum)

$$\frac{f}{2} \approx \frac{Nu}{Re Pr^{1/3}} \approx \frac{Sh}{Re Sc^{1/3}} \quad k \Leftrightarrow \frac{h}{\rho \hat{C}_p} \Leftrightarrow \frac{f v}{2}$$

$$\frac{f}{2} = \frac{Nu}{Re Pr^{1/3}} \quad \frac{h}{v} \left(\frac{v}{D}\right)^{2/3} = \frac{h}{\rho \hat{C}_p v} \left(\frac{v}{\alpha}\right)^{2/3} \approx \frac{f}{2}$$

$$\left(\frac{v}{D}\right) \approx \left(\frac{v}{\alpha}\right) \approx 1 \quad \text{for gases}$$

$$\text{so } \frac{k}{v} \approx \frac{h}{\rho \hat{C}_p v} \Rightarrow \frac{h}{k} \approx \rho \hat{C}_p$$

$$\rho \hat{C}_p (T_{\infty} - T_w) = C(MW)(x_w - x_{\infty})(\Delta H^{vap})$$

$$x_{\infty} = x_w - \left(\frac{\hat{C}_p}{\Delta H^{vap}} \right) (T_{\infty} - T_w)$$

\uparrow x^{sat} at T_w from VLE relation \uparrow known values $\nwarrow \nearrow$ measured values

\Rightarrow can get mole fraction of water in air
 \Downarrow

for relative humidity, divide by x^{sat} at T_{∞}

Transport #2(b.)

Iceberg being towed in the ocean.

- Don't want detailed velocity profiles

↳ friction factor (convective) treatment

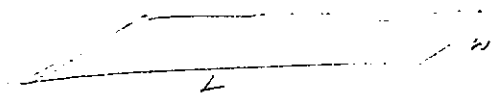
Momentum:

$$F_{\text{towing}} = \left(\frac{1}{2} \rho v^2 \right) (LW) f$$

Wetted area;
assume major
surface contacting
water is bottom of
iceberg, modeled
as a flat, rectangular
plate

Mass (?)

Iceberg is pure ice water; Sea water is salty
↳ $x_s = 0$ ↳ x_{sao}



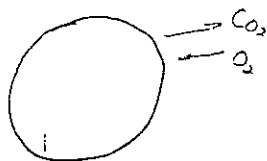
SO?
(how to relate to
momentum?)

Heat (?)

ΔH_{fusion}

Transport #2 (C.)

Burning Carbon Particle



Assumptions / B.C.'s

$$N_{O_2} = -N_{CO_2} \text{ at steady state}$$

$$X_{O_2} \approx 0 \text{ @ surface of particle (at } r=R) \text{ (process diffusion limited)}$$

$$X_{O_2} = 0.2 \text{ far from surface (at } r \rightarrow \infty) \text{ (composition of air)}$$

Diffusion? (Molecular Mechanism)
→ concentration gradient

Overall transfer

Coefficients?

→ assume that outside of particle Boundary Layer, air is well mixed

is this an accurate B.C.?

Should it be $X_{CO_2} \approx 0.2$?

↑ N_2 still present, no reason for it to be lacking near particle surface.

$$X_{CO_2} \approx 1.0 \text{ @ } r=R$$

$$X_{CO_2} \approx 0 \text{ @ } r \rightarrow \infty \text{ (0.03% in air)}$$

Assume size of particle doesn't change

How to incorporate energy balance?

$(T_{particle\ surface} - T_{\infty})$ driving force for convective heat transfer?

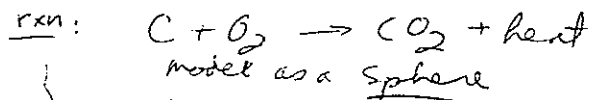
$$= (\Delta H_{rxn})(N_{O_2})_{\dot{m}}?$$

Transport

(#2 c.)

(Heat & mass transfer)

Burning carbon particle:



Issues:

Heat generation (ΔH_{rxn})

diffusion of reactant (O_2) in, $N_{CO_2} = -N_{O_2}$
product (CO_2) out

Changing particle size?

(#2 b.) (Heat & Mass transfer)

ΔH_{ice} fusion

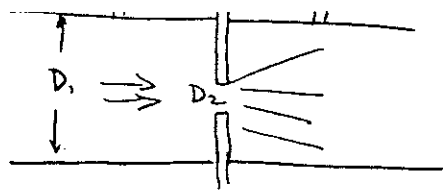
Iceberg being towed in ocean.

model as fluid flowing over dissolving (melting)
flat plate?

(#2 a.) \rightarrow see Cussler, Diffusion

Transport #6

$$\text{Sherwood \#} = Sh = \frac{k_l \overset{\substack{\text{mass transfer coeff.} \\ \text{char. length}}}{l}}{D \rightarrow \text{diffusion coeff.}} = \frac{\text{mass transfer velocity}}{\text{diffusion velocity}}$$



large permanent pressure drop
cheap, easy to maintain

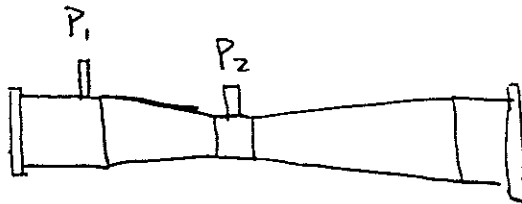
$$Q_f \left(\frac{\text{ft}^3}{\text{sec}} \right) = C_d S_c \sqrt{\frac{2p(p_1 - p_2)}{1 - \left(\frac{S_c}{S_1} \right)^2}}$$

C_d = coefficient of discharge

S_c = cross-sectional area at point of minimum

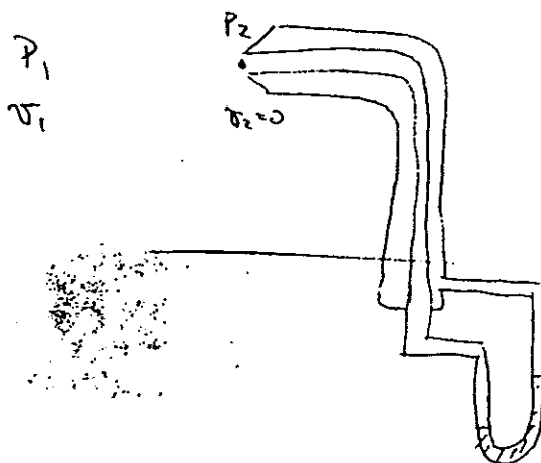
S_1 = cross-sectional area in upstream duct.

1) Venturi Meter



$$Q_1 = \frac{K Y A_2}{P_1} \sqrt{\frac{2(p_1 - p_2) p_1}{1 - \left(\frac{A_2}{A_1} \right)^2}}$$

2) Pitot Tube



$$(V_1)^2 = 2 \left(-\frac{\Delta P}{\rho} + \Sigma F \right)$$

$$V_1 = \sqrt{\frac{2(-\Delta P)}{\rho}}$$

Transport #10

(a) Bernoulli's Equation (Denn p 90, BSL p 211)

↳ (for viscous isothermal fluids)

Assumes: - Steady state

- single phase, uniform properties (ρ constant)

- uniform equivalent pressure (the same over entire cross-section)

$$\frac{\alpha_2}{2} \langle V \rangle_2^2 + gh_2 = \frac{\alpha_1}{2} \langle V \rangle_1^2 + gh_1 - \int_{P_1}^{P_2} \frac{dP}{\rho} + \delta W_s - l_v$$

Differential form:

$$\frac{1}{2} d(\alpha \langle V \rangle^2) + g dh + \frac{1}{\rho} dP = \delta W_s - \underbrace{(T ds - dQ_H)}_{d l_v (\geq 0)}$$

Notation:

V = component of velocity vector normal to surface.

$$\alpha \equiv \frac{\langle V^3 \rangle}{\langle V \rangle^3}$$

↳ = 0 only for reversible processes.

W_s, Q_H = work, heat added to system

l_v = "viscous losses" per unit mass (rate at which mech. energy converted to heat)

Bernoulli equation comes from (general equation): (cons. of energy)

$$\frac{d}{dt} \int_{z_1}^{z_2} \langle \rho(e + \frac{1}{2} v^2 + gh) \rangle A dz$$

$$= \underbrace{\langle \rho(e + \frac{1}{2} v^2 + gh) V \rangle_1 A_1}_{\text{internal kinetic potential}} - \langle \rho(e + \frac{1}{2} v^2 + gh) V \rangle_2 A_2 \quad \left. \vphantom{\frac{d}{dt} \int_{z_1}^{z_2} \langle \rho(e + \frac{1}{2} v^2 + gh) \rangle A dz} \right\} \text{Total energy variation}$$

$$+ \underbrace{\langle P V \rangle_1 A_1 - \langle P V \rangle_2 A_2}_{\text{rate of doing work to move fluid into and out of control volume.}} + \underbrace{\dot{Q}_H + \dot{W}_s}_{\text{heat + shaft work added to system.}}$$

Transport #10

(b) Hagen - Poiseuille Law



flow through cylinder
Laminar, $Re < 2100$

Derivation:

Assumptions: $v_z = v_z(r)$ only

$$v_\theta, v_r = 0$$

$$N-S: 0 = -\frac{\partial P}{\partial z} = \eta \mu \frac{1}{r} \frac{d}{dr} \left(r \frac{dv_z}{dr} \right), \quad P = P(z) \text{ only}$$

$$\frac{1}{\mu} \frac{dP}{dz} = \frac{1}{r} \frac{d}{dr} \left(r \frac{dv_z}{dr} \right) = \text{const} = \frac{1}{\mu} \frac{\Delta P}{L}$$

$$\frac{d}{dr} \left(r \frac{dv_z}{dr} \right) = \frac{1}{\mu} \frac{\Delta P}{L} r$$

$$r \frac{dv_z}{dr} = \frac{1}{2\mu} \frac{\Delta P}{L} r^2 + C_1$$

$$\frac{dv_z}{dr} = \frac{1}{2\mu} \frac{\Delta P}{L} r + \frac{C_1}{r}$$

$\Rightarrow C_1 \Rightarrow 0$ since $\frac{dv_z}{dr}$ finite @ $r=0$

$$v_z = \frac{1}{4\mu} \frac{\Delta P}{L} r^2 + C_2$$

\Rightarrow no-slip B.C.: @ $r=R$
 $v_z = 0$

$$v_z(r) = \frac{R^2}{4\mu} \left(-\frac{\Delta P}{L} \right) \left[1 - \left(\frac{r}{R} \right)^2 \right]$$

$$\langle v_z \rangle = \frac{1}{\pi R^2} \int_{\text{area}} v_z dA = \frac{R^2}{8\mu} \left(-\frac{\Delta P}{L} \right) \quad (\Delta P > 0)$$

$$\int_{\text{area}} v_z dA = \boxed{Q = \frac{\pi R^4}{8\mu} \left(-\frac{\Delta P}{L} \right) = \frac{\pi D^4}{128\mu} \left(-\frac{\Delta P}{L} \right)}$$

$$\text{Note: } f = \left(-\frac{\Delta P}{L} \right) \frac{D}{2\rho v^2} \Rightarrow \frac{D}{2} \left(-\frac{\Delta P}{L} \right) = \frac{8\mu \langle v_z \rangle}{R} = \frac{8\mu \langle v_z \rangle}{D/2}$$

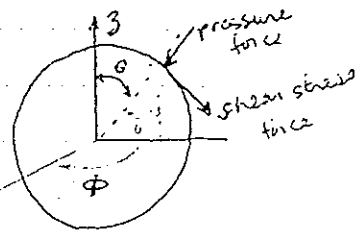
$$f = \frac{16\mu \langle v_z \rangle}{\rho \langle v_z \rangle^2 D} = \frac{16\mu}{\rho \langle v_z \rangle D} = \frac{16}{Re} \quad \leftarrow \text{Fanning friction factor from 1st principles}$$

Transport #10

(c) Stokes's Law - flow around a sphere at $Re < 1$
(inertialless regime)

drag coefficient, $C_D = \frac{24}{Re}$

drag force, $F_D = 3\pi\mu v_\infty D = 6\pi\mu v_\infty R$



In Denn p248-254 \Rightarrow derivation (cont, N.S.,
sol'n of DE's to get $v_r, v_\theta,$
 $T_{r\theta}$, and pressure)

In BSL p56 \rightarrow Above expressions given
w/o derivation.

Force acting
tangentially
in θ -dir.

$\tau_{r\theta} = \frac{3}{5} \frac{\mu v_\infty}{R} \left(\frac{R}{r}\right)^4 \sin\theta$

$p = p_0 - \rho g z - \frac{3}{5} \frac{\mu v_\infty}{R} \left(\frac{R}{r}\right)^2 \cos\theta$

$v_r = v_\infty \left[1 - \frac{3}{2} \left(\frac{R}{r}\right) + \frac{1}{2} \left(\frac{R}{r}\right)^3\right] \cos\theta$

$v_\theta = -v_\infty \left[1 - \frac{3}{4} \left(\frac{R}{r}\right) - \frac{1}{4} \left(\frac{R}{r}\right)^3\right] \sin\theta$

B.C.'s / assumptions

no slip: $v_r, v_\theta = 0$ @ $r = R$

no ϕ dependence: $\frac{\partial}{\partial \phi} = 0, v_\phi = 0$

Creeping flow (low $Re, < 1$)

z-component of pressure force: $-p \cos\theta$
per unit area

unit area = $(R \sin\theta d\phi)(R d\theta) = R^2 \sin\theta d\phi d\theta$

normal force $F_n = \int_0^{2\pi} \int_0^\pi (p|_{r=R} \cos\theta) R^2 \sin\theta d\theta d\phi$
 \nearrow plug in for this

$F_n = \frac{4}{3} \pi R^3 \rho g + 2\pi \mu R v_\infty$

friction force $F_t = \int_0^{2\pi} \int_0^\pi (+T_{r\theta}|_{r=R} \sin\theta) R^2 \sin\theta d\theta d\phi$
 \nearrow plug in for this

$F_t = 4\pi \mu R v_\infty$

3-component of shear stress
 $= (-T_{r\theta})(-\sin\theta)$

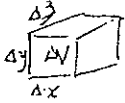
\nearrow
 $(-\cos(\frac{\pi}{2} - \theta))$

$F = \underbrace{\frac{4}{3} \pi R^3 \rho g}_{\text{buoyant force}} + \underbrace{2\pi \mu R v_\infty}_{\text{form drag}} + \underbrace{4\pi \mu R v_\infty}_{\text{friction drag}}$

Stokes's Law: $F_i = F_r = 6\pi \mu R v_\infty$ (force assoc. w/ fluid movmt.)

Transport #10

(d) Continuity equation (conservation of mass in control volume)



$$\begin{aligned} \text{rate of change of mass in } \Delta V &= \frac{d}{dt} (\underbrace{\rho \Delta x \Delta y \Delta z}_{\text{density}}) \\ &= (\text{mass in} - \text{mass out}) @ \text{ each face} \end{aligned}$$

$$\frac{\partial \rho}{\partial t} = - \frac{\partial(\rho v_x)}{\partial x} - \frac{\partial(\rho v_y)}{\partial y} - \frac{\partial(\rho v_z)}{\partial z}$$

$$\downarrow$$

$$\underbrace{\frac{\partial \rho}{\partial t} + v_x \frac{\partial \rho}{\partial x} + v_y \frac{\partial \rho}{\partial y} + v_z \frac{\partial \rho}{\partial z}}_{\frac{D\rho}{Dt}} = -\rho \left(\frac{\partial v_x}{\partial x} + \frac{\partial v_y}{\partial y} + \frac{\partial v_z}{\partial z} \right)$$

(rectangular)

(Cylindrical):

$$-\frac{\partial \rho}{\partial t} = \frac{1}{r} \frac{\partial}{\partial r} (\rho r v_r) + \frac{1}{r} \frac{\partial}{\partial \theta} (\rho v_\theta) + \frac{\partial}{\partial z} (\rho v_z)$$

(Spherical):

$$-\frac{\partial \rho}{\partial t} = \frac{1}{r^2} \frac{\partial}{\partial r} (\rho r^2 v_r) + \frac{1}{r \sin \theta} \frac{\partial}{\partial \theta} (\rho v_\theta \sin \theta) + \frac{1}{r \sin \theta} \frac{\partial}{\partial \phi} (\rho v_\phi)$$

Transport #10

(c) Momentum Equation (Cauchy), Navier-Stokes Eqs

Basis: conservation of momentum in control volume

rate of change of momentum = mom. entering - mom. leaving + \sum forces acting on control volume.

(x-dir) $\frac{\partial}{\partial t} (\underbrace{\rho v_x}_{x\text{-mom./volume}}) \underbrace{\Delta x \Delta y \Delta z}_{\text{volume}} = (\rho v_x) (\underbrace{\sum v_i \cdot \text{area}}_{\substack{\text{in - out} \\ \text{+ faces}}}) \Big|_x - (\rho v_x) \Big|_{x+\Delta x}$

+ stresses (area) + body force

\downarrow

in x-dir $\left\{ \begin{array}{l} \tau_{xx}(\Delta y \Delta z) \\ \tau_{yx}(\Delta x \Delta z) \\ \tau_{zx}(\Delta y \Delta x) \end{array} \right. = \sigma_x$ stress acting on x-face

body force \downarrow (usu. only gravity, sometimes electrical)

$\rho g_x \Delta x \Delta y \Delta z$

stresses (shear, body symmetry, pressure \rightarrow normal forces, + body forces (gravity))

- \rightarrow Stresses are symmetric: $\tau_{xy} = \tau_{yx}$ (from conservation of angular momentum)
- \rightarrow $\sigma_{xx} = \tau_{xx} - P \Rightarrow \tau_{xx} = \sigma_{xx} + P$ \rightarrow no internal couples arising from fluid structure.

Cauchy momentum equation: (in rectangular coords.)

(x-dir) $\rho \left(\frac{\partial v_x}{\partial t} + v_x \frac{\partial v_x}{\partial x} + v_y \frac{\partial v_x}{\partial y} + v_z \frac{\partial v_x}{\partial z} \right) = -\frac{\partial P}{\partial x} + \frac{\partial \tau_{xx}}{\partial x} + \frac{\partial \tau_{yx}}{\partial y} + \frac{\partial \tau_{zx}}{\partial z} + \rho g_x$

Stress constitutive equation

$\tau_{xx} = \mu \left[2 \frac{\partial v_x}{\partial x} - \frac{2}{3} (\nabla \cdot \underline{V}) \right]$ $\left(\nabla \cdot \underline{V} = \frac{\partial v_x}{\partial x} + \frac{\partial v_y}{\partial y} + \frac{\partial v_z}{\partial z} \right)$

$\tau_{xy} = \tau_{yx} = \mu \left[\frac{\partial v_x}{\partial y} + \frac{\partial v_y}{\partial x} \right]$

$P = p + \rho g h - \frac{1}{3} \mu \nabla \cdot \underline{V}$

Navier-Stokes:

$\rho \frac{D\underline{V}}{Dt} = -\nabla P + \mu \nabla^2 \underline{V}$ (general)

(x-dir; rectangular) $\Rightarrow \rho \left(\frac{\partial v_x}{\partial t} + v_x \frac{\partial v_x}{\partial x} + v_y \frac{\partial v_x}{\partial y} + v_z \frac{\partial v_x}{\partial z} \right) = -\frac{\partial P}{\partial x} + \mu \left(\frac{\partial^2 v_x}{\partial x^2} + \frac{\partial^2 v_x}{\partial y^2} + \frac{\partial^2 v_x}{\partial z^2} \right)$

Other coords, other components \Rightarrow pp 159-162 in Dean

Transport #14 (and # 42)

$$b \approx \frac{k}{v} = \frac{h \Delta T}{v} = \frac{1}{2} f v$$

$$= \frac{h}{\rho \hat{C}_p v}$$

$$N_i (\text{mass flux}) = k \Delta c_i$$

$$= (a + b v) \Delta c_i$$

$$q (\text{heat flux}) = h \Delta T = \frac{h}{\rho \hat{C}_p} \Delta (\rho \hat{C}_p T)$$

$$= (a' + b' v) \Delta (\rho \hat{C}_p T)$$

$$\tau (\text{momentum flux}) = f \left(\frac{1}{2} \rho v^2 \right) = \left(\frac{f v}{2} \right) (\rho v)$$

$$= (a'' + b'' v) (\rho v)$$

$$\frac{k}{v} = \frac{h}{\rho \hat{C}_p v} = \frac{f}{2}$$

$$\frac{k}{v} = \frac{h}{\rho \hat{C}_p v} = \frac{f}{2}$$

$a, a', a'' \rightarrow$ due to diffusion (molecular process, indep of velocity)
 $b, b', b'' \rightarrow$ due to convection (effect of eddies)

When eddies dominate (very rapid turbulent flow) —

Reynolds Analogy: $b = b' = b''$

so:

$$\frac{k}{v} = \frac{h}{\rho \hat{C}_p v} = \frac{f}{2} \quad \left(\text{only good for gases} \right)$$

(where $D \approx \alpha \approx \nu \approx 0.1 \text{ cm}^2/\text{sec.}$)

Chilton-Colburn Analogy

(Changes in liquids best represented as Pr & Sc numbers,)

so:

$$b = \frac{k}{v} \left(\frac{\nu}{D} \right)^{2/3}$$

$$b' = \frac{h}{\rho \hat{C}_p v} \left(\frac{\nu}{\alpha} \right)^{2/3}$$

$$b'' = \frac{f}{2} \left(\frac{\nu}{v} \right)^{2/3} = \frac{f}{2}$$

\Downarrow

$$\frac{k}{v} \left(\frac{\nu}{D} \right)^{2/3} = \frac{h}{\rho \hat{C}_p v} \left(\frac{\nu}{\alpha} \right)^{2/3} = \frac{f}{2}$$

$Sc \qquad Pr$

$$\left(\alpha = \frac{k}{\rho \hat{C}_p} \right) \leftarrow \text{thermal conductivity}$$

(18) Friction factor \rightarrow flow in conduits
 (Transport) Drag coefficient \rightarrow flow around submerged objects

$$F_{\text{kinetic}} = (A_{\text{exposed}})(\text{Kinetic energy per volume}) f$$

$$F_k = A K f \rightarrow f = \text{friction factor, drag coefficient.}$$

flow in conduits

(ρ = constant density of fluid)

A = wetted surface area

$$K = \frac{1}{2} \rho \langle v \rangle^2$$

$$F_k = [(P_o - P_L) + \rho g(h_o - h_L)] (\text{cross-sec. area})$$

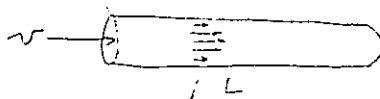
$\nearrow A_{cs}$

(a) for pipe (length L , radius R , diam. D)

$$A = 2\pi R L$$

$$K = \frac{1}{2} \rho \langle v \rangle^2$$

$$A_{cs} = \pi R^2$$



$$F_k = (P_o - P_L) + \rho g(h_o - h_L) (\pi R^2) = (P_o - P_L) \pi R^2 = (2\pi R L) \left(\frac{1}{2} \rho \langle v \rangle^2 \right) f$$

$$\left\langle f = \frac{(P_o - P_L) (D)}{4 L \left(\frac{1}{2} \rho \langle v \rangle^2 \right)} \right\rangle \text{ getting } f \text{ from experimental data}$$

$$F_k = \int_0^L \int_0^{2\pi} \left(-\mu \frac{\partial v_z}{\partial r} \right)_{r=R} R d\theta dz$$

for fully developed flow, $\frac{\partial v_z}{\partial r} \neq f(z)$; $\frac{\partial v_z}{\partial r} \neq f(r)$

$$F_k = (2\pi R L) \left(-\mu \right) \left(\frac{\partial v_z}{\partial r} \right)_{r=R} = (2\pi R L) \left(\frac{1}{2} \rho \langle v \rangle^2 \right) f$$

$$f = \frac{-2\mu}{\rho \langle v \rangle^2} \left(\frac{\partial v_z}{\partial r} \right)_{r=R} = \left(\frac{\mu}{\rho \langle v \rangle D} \right) \left(-\frac{2D}{\langle v \rangle} \left(\frac{\partial v_z}{\partial r} \right)_{r=R} \right)$$

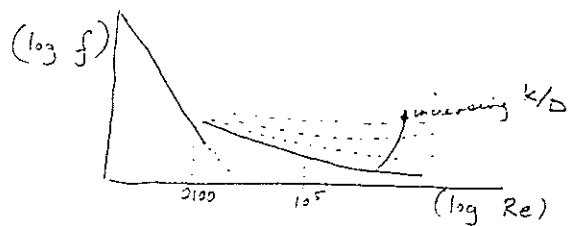
contributions to f :
 friction drag only
 - no form drag -

$$\boxed{f = \left(\frac{1}{Re} \right) \left(-\frac{2D}{\langle v \rangle} \left(\frac{\partial v_z}{\partial r} \right)_{r=R} \right)} \Rightarrow f = f(Re)$$

for pipe flow.

T, 18, cont.

T, 18, cont.

pipe

$$f = \frac{16}{Re} \quad \text{for } Re < 2100 \quad (\text{Hagen-Poiseuille})$$

$$f = \frac{0.0791}{(Re)^{1/4}} \quad \text{for } 2100 < Re < 10^5 \quad (\text{Blasius})$$

also to consider: - tube wall roughness $\Rightarrow \frac{k}{D}$ $\begin{matrix} \text{height of roughness} \\ \text{D = diameter} \end{matrix}$

- non-circular cross section

$$\downarrow \text{mean hydraulic radius } R_h = \frac{S}{Z} \quad \begin{matrix} S = \text{cross section area} \\ Z = \text{wetted perimeter} \end{matrix}$$

$$D = 4 R_h$$

(useful for turbulent flow only.)

flow around submerged objects ($\rho = \text{constant}$)

A = area of solid projected onto plane \perp to v_{∞}

$$K = \frac{1}{2} \rho v_{\infty}^2$$

$$F_k = \text{gravitational - buoyant forces} = (\text{Volume}) \rho_{\text{fluid}} g - (\text{Volume}) \rho g$$

(at terminal velocity.)

(b) for flow around a sphere (radius R , diam D)

$$A = \pi R^2$$

$$K = \frac{1}{2} \rho v_{\infty}^2$$

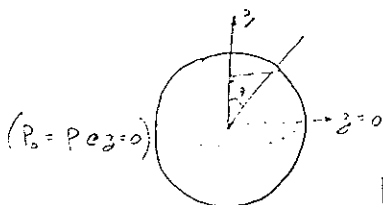
$$V_d = \frac{4}{3} \pi R^3$$

$$F_k = \left(\frac{4}{3} \pi R^3 \right) (g) (\rho_{\text{sph}} - \rho) = (\pi R^2) \left(\frac{1}{2} \rho v_{\infty}^2 \right) f$$

$$f = \frac{4 D g (\rho_{\text{sph}} - \rho)}{3 v_{\infty}^2 (\rho)}$$

v_{∞} is terminal velocity
(quantity measured for
experimental determination of f)

contributions to $f \Rightarrow$ friction drag and form drag.



$$F_k = \overbrace{F_n - F_s}^{\text{form}} + \underbrace{F_{\text{friction}}}_{\text{friction}}$$

$$F_n = \int_0^{2\pi} \int_0^{\pi} (-P|_{r=R} \cos \theta) R^2 \sin \theta d\theta d\phi$$

$$F_s = \int_0^{2\pi} \int_0^{\pi} \left\{ -(P_0 - \rho g z) |_{r=R} \cos \theta \right\} R^2 \sin \theta d\theta d\phi$$

$$F_{\text{friction}} = \int_0^{2\pi} \int_0^{\pi} \left\{ \mu \left[r^2 \left(\frac{\partial v_{\theta}}{\partial r} \right) + \frac{1}{r} \frac{\partial v_r}{\partial \theta} \right] \right\}_{r=R} \sin \theta R^2 \sin \theta d\theta d\phi$$

$$\Rightarrow f = f(Re)$$

$$\text{Viscosity} = \frac{\text{shear stress}}{\text{shear rate}} = \frac{\tau_{rz}}{\left(\frac{\partial v_z}{\partial r}\right)}$$

Viscometry → capillary, cone and plate, coaxial cylinders

- ⇒ Capillary Viscometer → need to measure:
- flow rate Q
 - pressure drop Δp
 - know: radius R
 - length L

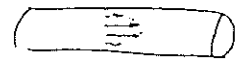
- laminar flow ($Re < 2100$)

- steady state, fully-developed flow

Solve N-S equation, get velocity profile:

$$v_z(r) = \frac{R^2}{4\mu} \left(-\frac{\Delta p}{L}\right) \left[1 - \left(\frac{r}{R}\right)^2\right]$$

$$\langle v_z \rangle = \frac{1}{\pi R^2} \int_{\text{area}} v_z dA = \frac{R^2}{8\mu} \left(-\frac{\Delta p}{L}\right)$$



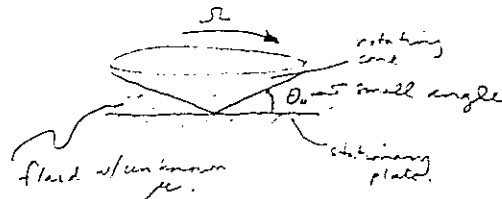
$$\mu = \frac{\pi R^4}{8} \left(-\frac{\Delta p}{L}\right) \frac{1}{Q} \rightarrow \text{volume flow rate}$$

⇒ Cone and Plate Viscometer

cone rotated with

known angular velocity Ω ;

Torque required measured



only $\tau_{\theta\phi}$ important

Rigorous derivation → BSL pp 98-101

⇒ Coaxial cylinder Viscometer (Couette-Hatschek, MacMichael viscometers)

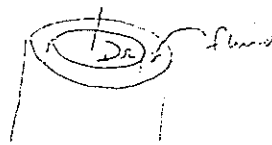
either inner or outer cylinder

rotated while other remains

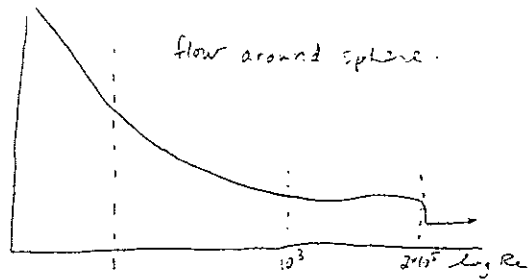
stationary → essentially Couette flow if $R_o - R_i$ small

Known: Ω (angular velocity)

Measured: torque required to turn @ Ω .



T, 18, cont

Sphere $\log f$ 

$$C_D = f \approx \frac{24}{Re}$$

for $Re < 0.1$ (1.0) Stokes's Law / Creeping flow
10% error

$$C_D = f \approx \frac{18.5}{Re^{3/5}}$$

for $2 < Re < 5 \times 10^2$ (10³) intermediate region

$$C_D = f \approx 0.44$$

for $5 \times 10^2 < Re < 2 \times 10^5$ (10³) Newton's "Law" region

(c) flow around submerged flat plate (p. 203 BSL problem 6.6)
 width W , length L , velocity on both sides



$$(i) F_k = 1.328 \sqrt{\rho \mu L W^3 v_\infty^3}$$

friction drag only (plate is flat)

$$= (WL \rho v_\infty^2) \bar{f}$$

$$F_k = (2WL)(\frac{1}{2} \rho v_\infty^2) \bar{f} \rightarrow \text{definition of } \bar{f}$$

(width area)

$$= 1.328 \sqrt{\mu W v_\infty} \sqrt{\rho L W v_\infty^3}$$

$$\bar{f} = \frac{1.328 \sqrt{\mu W v_\infty}}{\sqrt{\rho L W v_\infty^3}} = 1.328 \sqrt{\frac{\mu}{\rho L v_\infty}} \quad \boxed{f = 1.328 Re^{-1/2}}$$

Laminar flow.

$$(ii) F_k = 0.074 \rho v_\infty^2 WL \left(\frac{L v_\infty \rho}{\mu} \right)^{-1/5}$$

\rightarrow accurate for $5 \times 10^5 < \frac{L \rho v_\infty}{\mu} < 2 \times 10^7$
 (turbulent flow)

$$= \rho v_\infty^2 WL f$$

$$\bar{f} = 0.074 \left(\frac{L v_\infty \rho}{\mu} \right)^{-1/5}$$

$$\boxed{f = 0.074 Re^{-1/5}}$$

Turbulent flow

$5 \times 10^5 < Re < 2 \times 10^7$

Flow around other objects:

$$f = \frac{F_k}{(\frac{1}{2} \rho v_\infty^2) A_{\text{projected}}}$$

e.g. flow over very long cylinder ($L \gg D$)



$$f = \frac{F_k}{(\frac{1}{2} \rho v_\infty^2 (LD))} = \frac{(F_k/L)}{\frac{1}{2} \rho v_\infty^2 D}$$

drag force
 per unit
 cylinder length.

Transport #22

How are diffusivity and viscosity of a mixture determined?

math 230 notes

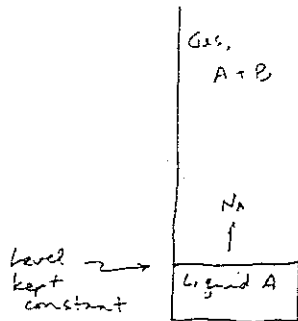
Diffusivity:

from BSL p523-25

one possibility:

@ steady state, $N_B = 0$

→ Gas stream of A & B



$$N_{A_z} = -C D_{AB} \frac{\partial x_A}{\partial z} + x_A (N_{A_z} + N_{B_z})$$

$$N_{A_z} = -\frac{C D_{AB}}{1-x_A} \frac{\partial x_A}{\partial z}$$

$$-\frac{\partial N_{A_z}}{\partial z} = 0$$

(mole fraction)

if know concentration profile of A,

C (overall gas concentration), can get

D_{AB} .

from solution of D.E.

another possibility: discussed in CHE 230

1. know rate of Evaporation

$$N_{\text{coly}} = (\Delta \text{Vol}) \rho (\text{mw}) A \left(\frac{1}{\text{time}} \right) \rightarrow \text{Air}$$

$$N_A = \frac{P D_{AB}}{R T (z_2 - z_1)} \ln \left(\frac{P_{B,z_2}}{P_{B,z_1}} \right)$$

μ :

Camelotens (BSL 18)

based on P_e', T_e' ,

$$\mu_e = \sigma x_i \mu_c$$

Viscosity → (corr)

Transport #26

What is the angular dependence of Nu for a falling drop?

$$Nu = \frac{hD}{k}$$

dimensionless
temperature gradient
at the surface.

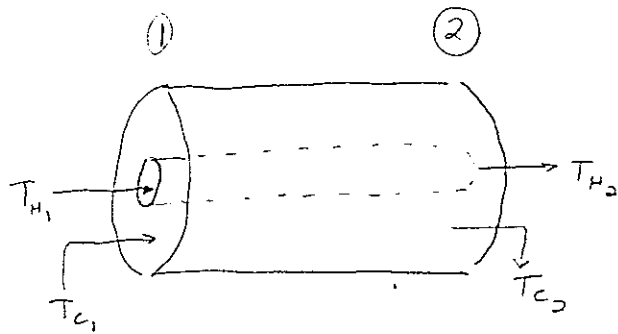
Transport # 20

$$dQ = U dA (T_H - T_C)$$

$$\begin{aligned} &= -(WC_p)_H dT_H \\ &= (WC_p)_C dT_C \end{aligned}$$

$$dT_H = \frac{-dQ}{(WC_p)_H}$$

$$dT_C = \frac{dQ}{(WC_p)_C}$$



$$d(T_H - T_C) = dT_H - dT_C = -\left(\frac{1}{(WC_p)_H} + \frac{1}{(WC_p)_C}\right) dQ$$

$$= -\left(\frac{1}{(WC_p)_H} + \frac{1}{(WC_p)_C}\right) (U dA) (T_H - T_C)$$

$$\frac{d(T_H - T_C)}{T_H - T_C} = \int_1^2 d \ln(T_H - T_C) = \int_A U dA \left(\frac{1}{(WC_p)_H} + \frac{1}{(WC_p)_C}\right)$$

$$\ln\left(\frac{(T_H - T_C)_2}{(T_H - T_C)_1}\right) = \ln\left(\frac{\Delta T_2}{\Delta T_1}\right) = -UA \left(\frac{1}{(WC_p)_H} + \frac{1}{(WC_p)_C}\right)$$

$$Q = -(WC_p)_H (T_{H2} - T_{H1}) \Rightarrow \frac{1}{(WC_p)_H} = \frac{T_{H1} - T_{H2}}{Q}$$

$$= (WC_p)_C (T_{C2} - T_{C1}) \Rightarrow \frac{1}{(WC_p)_C} = \frac{T_{C2} - T_{C1}}{Q}$$

$$-UA \left(\frac{1}{(WC_p)_H} + \frac{1}{(WC_p)_C}\right) = -UA \left(\frac{T_{H1} - T_{H2}}{Q} + \frac{T_{C2} - T_{C1}}{Q}\right)$$

$$= -\frac{UA}{Q} ((T_H - T_C)_2 + (T_H - T_C)_1)$$

$$\ln\left(\frac{\Delta T_2}{\Delta T_1}\right) = \frac{UA}{Q} (\Delta T_2 - \Delta T_1)$$

$$Q = UA \left(\frac{\Delta T_2 - \Delta T_1}{\ln\left(\frac{\Delta T_2}{\Delta T_1}\right)} \right)$$

overall heat transfer coefficient for
co-current or counter-current
tube-shell heat exchangers.

Transport #34

from Reynolds Analogy

$$j_H = \frac{h}{\rho \hat{c}_p v} \left(\frac{v}{\alpha} \right)^{2/3}$$

from experimental data, approximating $j_H \propto \frac{1}{Re}$

$$= \frac{h}{\rho \hat{c}_p v} \left(\frac{\hat{c}_p \mu}{k} \right)^{2/3}$$

$$= \left(\frac{h}{k} \right) \left(\frac{\mu}{\rho v} \right) \left(\frac{k}{\hat{c}_p \mu} \right)^{1/3}$$

$$= \left(\frac{hD}{k} \right) \left(\frac{\mu}{\rho v D} \right) \left(\frac{\alpha}{v} \right)^{1/3}$$

$$= Nu Re^{-1} Pr^{-1/3}$$

$$Re = \frac{\text{inertial forces}}{\text{viscous forces}}$$

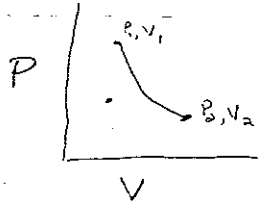
#38

Lewis # $\frac{\alpha}{D} = \frac{k}{\rho \hat{c}_p D}$ is independent of gas-phase velocity (same effect on heat and mass transfer, effects cancel.)

$$\frac{k}{\rho \hat{c}_p D}$$

Transport #50

Derive equations for gas undergoing isentropic expansion.



1st Law for non-flow: $dU = dQ - dW$

isentropic, so $dQ = 0$

Assume ideal gas. Equation reduces to:

$$dU = -dW$$

$$C_v dT = -P dV = -\frac{RT}{V} dV$$

$$C_v \int_{T_1}^{T_2} \frac{dT}{T} = -R \int_{V_1}^{V_2} \frac{dV}{V}$$

$$C_v \ln\left(\frac{T_2}{T_1}\right) = -R \ln\left(\frac{V_2}{V_1}\right)$$

$$\ln \frac{V_2}{V_1} = \ln\left(\frac{T_2}{T_1}\right)^{-\frac{C_v}{R}} \Rightarrow \left(\frac{V_2}{V_1}\right) = \left(\frac{T_2}{T_1}\right)^{-\frac{C_v}{R}}$$

for ideal gas,

$$H = U + PV = U + RT \Rightarrow dH = dU + R dT \Rightarrow C_p dT = C_v dT + R dT$$

$$\text{So: } C_p = C_v + R$$

$$\text{also (notation) } \frac{C_p}{C_v} = \gamma$$

$$\gamma = \frac{C_v + R}{C_v} = 1 + \frac{R}{C_v}$$

$$1 - \gamma = -\frac{R}{C_v}$$

$$-\frac{C_v}{R} = \frac{1}{1 - \gamma}$$

$$\Rightarrow \left(\frac{V_2}{V_1}\right) = \left(\frac{T_2}{T_1}\right)^{\left(\frac{1}{1-\gamma}\right)} \Rightarrow \left(\frac{T_2}{T_1}\right) = \left(\frac{V_2}{V_1}\right)^{(1-\gamma)}$$

$$\frac{V_2}{V_1} = \frac{\frac{RT_2}{P_2}}{\frac{RT_1}{P_1}} = \frac{P_1 T_2}{P_2 T_1} \Rightarrow \left(\frac{P_1}{P_2}\right) = \left(\frac{T_1}{T_2}\right) \left(\frac{T_2}{T_1}\right)^{\left(\frac{1}{1-\gamma}\right)} = \left(\frac{T_2}{T_1}\right)^{\left(\frac{1-(1-\gamma)}{1-\gamma}\right)}$$

$$\left(\frac{P_1}{P_2}\right) = \left(\frac{T_2}{T_1}\right)^{\left(\frac{\gamma}{1-\gamma}\right)}$$

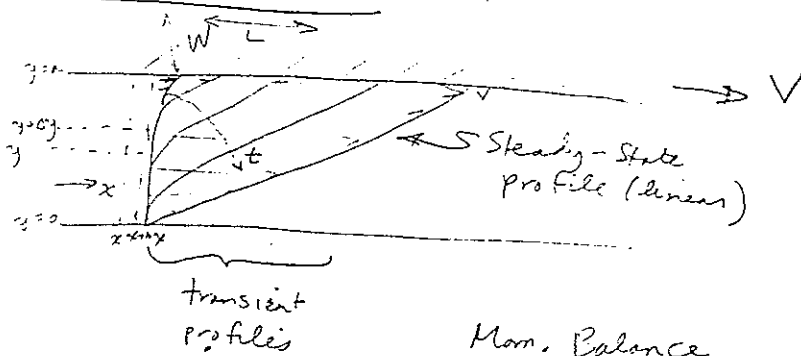
$$\frac{T_2}{T_1} = \frac{\frac{P_2 V_2}{R}}{\frac{P_1 V_1}{R}} = \frac{P_2 V_2}{P_1 V_1} \Rightarrow \left(\frac{P_2}{P_1}\right) = \left(\frac{V_1}{V_2}\right) \left(\frac{V_2}{V_1}\right)^{(1-\gamma)} = \left(\frac{V_2}{V_1}\right)^{(1-\gamma-1)}$$

$$\left(\frac{P_2}{P_1}\right) = \left(\frac{V_1}{V_2}\right)^{\gamma} \Rightarrow P_2 V_2^{\gamma} = P_1 V_1^{\gamma} = PV^{\gamma} = \text{const.}$$

Transport #58

push

+1



Why is S.S. profile linear?

Mom. Balance

$$-\mu \left(\frac{\partial v_x}{\partial y} \right) \Big|_y + \mu \left(\frac{\partial v_x}{\partial y} \right) \Big|_{y+\Delta y} = 0$$

$$\lim_{\Delta y \rightarrow 0} \mu \left(\frac{\partial^2 v_x}{\partial y^2} \right) \Delta y = 0$$

Integrate ↓

$$\frac{\partial v_x}{\partial y} = C_1$$

Integrate ↓

$$v_x = C_1 y + C_2$$

Apply B.C.'s:

$$0 = C_1(0) + C_2 \Rightarrow C_2 = 0$$

$$V = C_1(a) \Rightarrow C_1 = \frac{V}{a}$$

So we see that $v_x = \frac{V}{a} y$ is linear,

because shear stress $-\mu \frac{\partial v_x}{\partial y} = -\frac{\mu V}{a}$ is constant.

Part 2 driving force for fluid flow in a pipe: pressure drop + gravity (if not horizontal)

driving force here: Force applied to keep top plate moving at V .

Momentum balance above.

More generalized:

$$\text{force} = \left(\tau_{yx} \Big|_{y=a} \right) (\text{Area} = LW) \\ \Rightarrow \frac{\text{force}}{\text{unit area of plate}} = -\frac{\mu V}{a}$$

$$\frac{\partial}{\partial t}(\rho v) = -[\nabla \cdot \rho \underline{v} \underline{v}] - \nabla p - [\nabla \cdot \underline{\tau}] + \rho g \Rightarrow \text{momentum transport in } y\text{-direction only?} \\ -\frac{\partial \tau_{yx}}{\partial y} = \mu \frac{\partial^2 v_x}{\partial y^2} = 0$$