

SNS 419

Maneuver-Hohmann Transfer

By:

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Our spacecraft has a semi major axis of 7000 km and is in a circular orbit. We wish to put it at a 20000 km semi major axis circular orbit. Calculate the total amount of  $\Delta v$  Required to transfer to the new orbit using a Hohmann transfer.

SemiMajorAxis-Transfer orbit: 13500.000000000 km

dv1: 1.638710281 km/s

dv2: 1.249638081 km/s

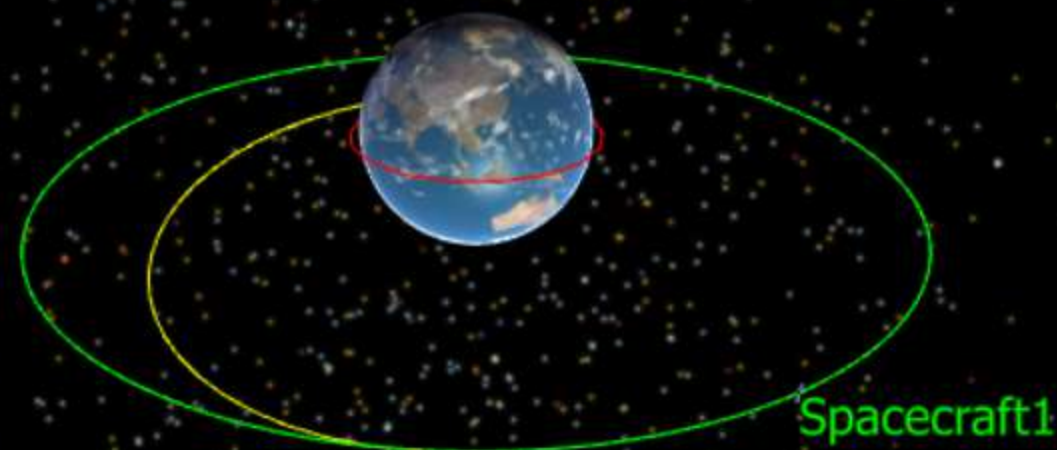
totalDV: 2.888348362 km/s

Report 10

dV1	dV2	totalDV
1.638710281	1.249638081	2.888348362

MissionView 9

Jan 02 2020 04:10:00.273848134 UTC  
Target: Earth  
Source: Earth(270° RA, 20° Dec, 30000 km Radius)  
FOV: 70°



Output Properties

Selected Component:

MissionView 9

Print

Export

All Output Properties...

Viewpoint

Current: None

View Mode

3D View

Reference Frame

MJ2000

Source

Earth

Source Offsets

Right Ascension

270 deg

Declination

20 deg

Translation

30000 km

Field Of View

70 deg

Target

Earth

Tail Reference

Earth



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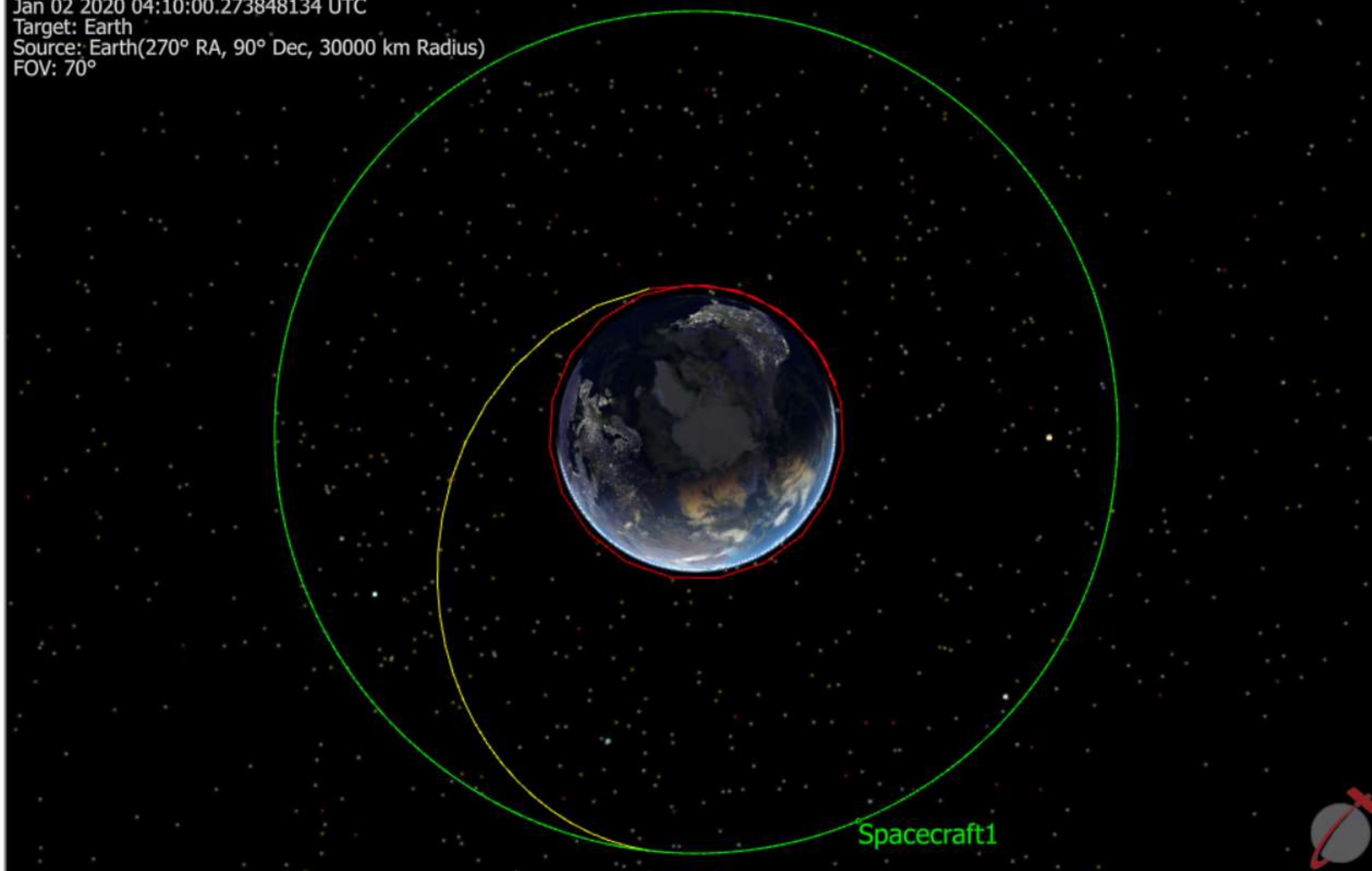
totalDV: 2.888348362 km/s

Jan 02 2020 04:10:00.273848134 UTC

Target: Earth

Source: Earth(270° RA, 90° Dec, 30000 km Radius)

FOV: 70°



Selected Component:

MissionView 9

Print

Export

All Output Properties...

Viewpoint

Current: None

View Mode

3D View

Reference Frame

MJ2000

Source

Earth

Source Offsets

Right Ascension

270 deg

Declination

90 deg

Translation

30000 km

Field Of View

70 deg

Target

Earth

Tail Reference

Earth





```
1  
2 Variable InitialRadius = 7000 ;  
3  
4 Variable FinalRadius = 20000 ;
```

Block...End  
FreeForm  
Assign(=)  
Call  
Close  
For...End  
Get  
If...End  
Include  
Maneuver  
Open  
Pause  
Plot  
Put  
Receive  
Report  
Restore  
Save  
Send  
Show  
Step  
Stop  
Target...End  
Try...End  
Update  
View  
Watch  
While...End



```
1 // initial Speed Orbit
2
3 Variable vInitial = sqrt(Earth.Mu * ( (2/InitialRadius) - (1/InitialRadius) ));
4
5
6 // Semi-Major Axis of the transfer trajectory
7
8 Variable SemiMajorTransfer = (FinalRadius + InitialRadius)/2;
9
10
11 // Velocity at periapsis of the transfer trajectory
12
13 Variable vTransfPeri = sqrt(Earth.Mu * ( (2/InitialRadius) - (1/SemiMajorTransfer) ));
14
15
16 // Delta V of the first maneuver
17
18 Variable dV1 = vTransfPeri - vInitial;
19
20
21 // Velocity at apoapsis of the transfer trajectory
22
23 Variable vTransfApog = sqrt(Earth.Mu * ( (2/FinalRadius) - (1/SemiMajorTransfer) ));
24
25
26 // Velocity of the final orbit
27
28 Variable vFinal = sqrt(Earth.Mu * ( (2/FinalRadius) - (1/FinalRadius) ));
29
30
31 // Delta V of the second Maneuver
32
33 Variable dV2 = vFinal - vTransfApog;
34
35
36 // Total Delta V required
37
```

Block...End  
FreeForm  
Assign(=)  
Call  
Close  
For...End  
Get  
If...End  
Include  
Maneuver  
Open  
Pause  
Plot  
Put  
Receive  
Report  
Restore  
Save  
Send  
Show  
Step  
Stop  
Target...End  
Try...End  
Update  
View  
Watch  
While...End

```
17
18 Variable dV1 = vTransfPeri - vInitial;
19
20
21 // Velocity at apoapsis of the transfer trajectory
22
23 Variable vTransfApog = sqrt(Earth.Mu * ( (2/FinalRadius) - (1/SemiMajorTransfer) ));
24
25
26 // Velocity of the final orbit
27
28 Variable vFinal = sqrt(Earth.Mu * ( (2/FinalRadius) - (1/FinalRadius) ));
29
30
31 // Delta V of the second Maneuver
32
33 Variable dV2 = vFinal - vTransfApog;
34
35
36 // Total Delta V required
37
38 Variable totalDV = dV1 + dV2;
39
40
41
42
43 // Assigns the InitialRadius to the spacecraft
44 // to ensure the Spacecraft SMA is the same as the one the user defined
45 Spacecraft1.A = InitialRadius;
46
```

Block...End  
FreeForm  
Assign(=)  
Call  
Close  
For...End  
Get  
If...End  
Include  
Maneuver  
Open  
Pause  
Plot  
Put  
Receive  
Report  
Restore  
Save  
Send  
Show  
Step  
Stop  
Target...End  
Try...End  
Update  
View  
Watch  
While...End



Object Browser

Add Edit View

ConsoleWindow

Console (G)

ImpulsiveBurn

ImpulsiveBurn1

ImpulsiveBurn2

SolarSystem

FF\_SolarSystem (G)

Spacecraft

Spacecraft1

ViewWindow

ViewWindow1

Mission Sequence 8 - perform maneuver 1

FreeForm Label

perform maneuver 1

Print

```
1 // Changes color of SpaceCraft tail
2
3 Spacecraft1.Color = ColorTools.Yellow;
4
5
6 // Assigns the calculated delta v value to the Impulsive Burn
7
8 ImpulsiveBurn1.BurnDirection[0] = dV1;
9
10
11 Maneuver Spacecraft1 using ImpulsiveBurn1;
12
13
14 // Steps the Spacecraft to apoapsis and visualizes the Spacecraft
15
16 WhileStepping Spacecraft1 to (Spacecraft1.OrbitApoapsis);
17     Update ViewWindow1;
18
19
20 End;
21
```

Script Elements

- Block...End
- FreeForm
- Assign(=)
- Call
- Close
- For...End
- Get
- If...End
- Include
- Maneuver
- Open
- Pause
- Plot
- Put
- Receive
- Report
- Restore
- Save
- Send
- Show
- Step
- Stop
- Target...End
- Try...End
- Update
- View
- Watch
- While...End

Object Browser Externals

Line: 1 Column: 1 Offset: 1

Status Message: The Mission Plan executed successfully.

Status: Stopped

Run Duration: 1 s

Object Browser

Add Edit View

ConsoleWindow

Console (G)

ImpulsiveBurn

ImpulsiveBurn1

ImpulsiveBurn2

SolarSystem

FF\_SolarSystem (G)

Spacecraft

Spacecraft1

ViewWindow

ViewWindow1

Mission Sequence 8 - perform maneuver 1 9 - preform maneuver 2

FreeForm Label

preform maneuver 2

Print

```
1 // Changes the color of SpaceCraft tail again
2
3 Spacecraft1.Color = ColorTools.Lime;
4
5
6 // Sets the calculated delta v to the Impulsive Burn
7
8 ImpulsiveBurn2.BurnDirection[0] = dV2;
9
10
11 Maneuver Spacecraft1 using ImpulsiveBurn2;
12
13
14 // Reports the delta v values
15
16 Report dV1, dV2, totalDV;
17
```

Script Elements

- Block...End
- FreeForm
- Assign(=)
- Call
- Close
- For...End
- Get
- If...End
- Include
- Maneuver
- Open
- Pause
- Plot
- Put
- Receive
- Report
- Restore
- Save
- Send
- Show
- Step
- Stop
- Target...End
- Try...End
- Update
- View
- Watch
- While...End

Object Browser Externals

Line: 1 Column: 1 Offset: 1

Status Message: The Mission Plan executed successfully.

Status: Stopped

Run Duration: 1 s



```
1 Report " " to Console;
2
3 Report "SemiMajorAxis-Transfer orbit: ",SemiMajorTransfer, " km" to Console;
4
5 Report " " to Console;
6
7 Report "dv1: ",dV1, " km/s" to Console;
8
9 Report "dv2: ",dV2, " km/s" to Console;
10
11 Report "totalDV: ",totalDV, " km/s" to Console;
12
```

Print