DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION

A33SO Revision 10

Amphibian Aircraft Technologies, LLC. (Grumman)

HU-16A HU-16B

HU-16C (UF-1, UF-1G)

HU-16D (UF-2) HU-16E (UF-2G)

May 1, 2016

TYPE CERTIFICATE DATA SHEET A33SO

This data sheet, which is part of Type Certificate No. A33SO, prescribes conditions and limitations under which the product for which type certificate was issued meets the airworthiness requirements of the Federal Aviation Regulations.

Type Certificate Holder: Amphibian Aircraft Technologies, LLC.

c/o Dunlap Bennett & Ludwig, PLLC

211 Church Street

Leesburg, Virginia 20175

Type Certificate Holder Record Amphibian Aircraft International, Inc. transferred type certificate to Amphibian Aircraft

Technologies, LLC. on May 1, 2016

Robinson Air Crane, Inc. transferred type certificate to Amphibian Aircraft International,

Inc. on August 4, 2010.

Air Crane, Inc. transferred type certificate to Robinson Air Crane, Inc. on June 29, 2010.

I - Grumman Model HU-16A (Restricted Category), approved September 29, 1988

Engines 2 Wright F1820-76A, B, C, D

Fuel 100/130 minimum grade aviation gasoline

Engine Limits Low Impeller Ratio HP **RPM** <u>MP</u> ALT.FT. Takeoff (five min.) 1425 2700 51.5 Sea Level Takeoff (five min.) 1425 2700 50.5 2600 Maximum Continuous 1275 2500 46.5 Sea Level Maximum Continuous 1275 2500 45.5 3500

Straight line variation between points given

Propeller and Propeller Limits 3 blade Hamilton Standard Hydromatic constant speed full feathering and

reversible with integral oil systems.

Hub Model No.: 43D50, 43D50-601, 603 or 637

Blade No.: 6601- (), 6601A-7 thru -9, or 6621A-7, Nickel Plated 6999-7

thru -9, or 7005-7, 7007B, 7007B-7

Pitch settings at the 42 in. station: Low 16 degrees

Feather 89 degrees Reverse -12 degrees

Diameter: Maximum 132.00 in.

Minimum 129.36 in.

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Vne (never exceed)	260 KIAS
Severe Turbulence	130 KIAS
Full Aileron Defection	150 KIAS
Landing Gear Extended	130 KIAS
Landing Lights Extended	120 KIAS
	Severe Turbulence Full Aileron Defection Landing Gear Extended

Flaps Down 15 175 KIAS Flaps Down 30 135 KIAS Flaps Full Down 40 115 KIAS De-Icer Boot Operation 200 KIAS

C.G. Range 294" (20.5% MAC) to 304.0" (28% MAC)

Datum 71.5 forward of nose jack pad

Leveling Means Leveled laterally by leveling lugs located in nose wheel well and longitudinally

by a plum bob drop from the small hole located in the shelf of either wheel well.

Maximum Weight 33,500 lbs.

Minimum Crew 2 (pilot & copilot)

Fuel Capacity Right and Left Main Tanks 340 gal. (each), 2040 lbs., (+297.3)

Right and Left Float Tanks 212 gal. (each), 1272 lbs., (+307.5) Right and Left Drop Tanks 295 gal. (each), 1770 lbs., (+295.0)

Oil Capacity Right and Left Tanks 25.6 gals (each), 192 lbs., (+262.0)

Control Surface Movement

Wing Flaps Down 40 ± 3

Other Operating Limitation U.S.A.F. Flight Manual No. T.O. 1U-16(H)A-1, dated Apr. 8, 1965

Serial Nos. Eligible 149836, 142428, 132429, 131266, 131892, 137900, 137926,

137927, 131916, 142361, 141276, 131911, 142361, 137930, 49-075

II - Grumman Model HU-16B, (Restricted Category), approved September 29, 1988

Engines 2 Wright F1820-76A, B, C, D

Fuel 100/130 minimum grade aviation gasoline

Engine Limits

Low Impeller Ratio	<u>HP</u>	<u>RPM</u>	<u>MP</u>	ALT.FT.
Takeoff (five min.)	1425	2700	51.5	Sea Level
Takeoff (five min.)	1425	2700	50.5	2600
Maximum Continuous	1275	2500	46.5	Sea Level
Maximum Continuous	1275	2500	45.5	3500

Straight line variation between points given

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Propeller and Propeller Limits

3 blade Hamilton Standard Hydromatic constant speed full feathering and

reversible with integral oil systems.

Hub Model No.: 43D50, 43D50-601, 603 or 637

Blade No.: 6601- (), 6601A-7 thru -9, or 6621A-7, Nickel Plated 6999-7

thru -9, or 7005-7, 7007B, 7007B-7

Pitch settings at the 42 in. station: Low 16 degrees

Feather 89 degrees Reverse -12 degrees

Diameter: Maximum 132.00 in.

Minimum 129.36 in.

Airspeed Limits

Vne (never exceed) **254 KIAS** Severe Turbulence **130 KIAS** Full Aileron Defection **150 KIAS** Landing Gear Extended **130 KIAS** Landing Lights Extended **120 KIAS** Flaps Down 15 **243 KIAS** Flaps Down 30 **163 KIAS** Flaps Full Down 40 **140 KIAS** De-Icer Boot Operation **175 KIAS**

C.G. Range 294.4" (20.5% MAC) to 304.0" (28% MAC)

Datum 71.5 forward of nose jack pad

Leveling Means Leveled laterally by leveling lugs located in nose wheel well and longitudinally

by a plum bob drop from the small hole located in the shelf of either wheel well.

Maximum Weight 35,400 lbs.

Minimum Crew 2 (pilot & copilot)

Fuel Capacity Right and Left Main Tanks 340 gal. (each), 2040 lbs., (+297.3)

Right and Left Float Tanks 212 gal. (each), 1272 lbs., (+307.5) Right and Left Drop Tanks 295 gal. (each), 1770 lbs., (+295.0)

Oil Capacity Right and Left Tanks 25.6 gals (each), 192 lbs., (+262.0)

Control Surface Movement

Wing Flaps Down 40 ± 3

Other Operating Limitations U.S.A.F. Flight Manual No. T.O. 1U-16(H)B-1, dated Dec. 1, 1967

Serial Nos. Eligible 0126, 1030, 9099, 51-5291, 51-7195A, 51-0043, 86B, 51-0019, 51-7162,

 $49-097,\,99-7213,\,51-7187,\,51-025,\,2132,\,51-7246,\,51-5303,\,146427,\,51-7240$

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III - Grumman Model HU-16C (UF-1, UF-1G) (Restricted Category), approved January 23, 1995

Same as the Model HU-16A, see Section I for information, except as follows:

Serial Nos. Eligible 131904, 131905, 131906, 131910, 131917, 137912, 137921, 137924, 137928,

137932, 137933, 141262, 141265, 141271, 142360, 142362, 17164

IV-Grumman Model HU-16D (UF-2) (Restricted Category), approved January 23, 1995

Same as the Model HU-16B, see Section II for information, except as follows:

Other Operating Limitations: U.S. Navy Flight Manual NAVWPS 01-85AB-1

Serial Nos. Eligible 7188, 7214, 7227, 7245, 7226, 1311, 146426, 141278

V-Grumman Model HU-16E (UF-2G) (Restricted Category), approved January 23, 1995

Same as the Model HU-16B, see Section II for information, except as follows:

Other Operating Limitations: U.S. Navy Flight Manual NAVWPS 01-85AB-1

Serial Nos. Eligible 1265, 1272, 1294, 137904, 137911, 2135

DATA PERTINENT TO ALL MODELS

Certification Basis FAR 21.25(a)(2), effective February 1, 1965, Amendments 21-1 through 21-60.

Date of application July 27, 1988. Type Certificate A33SO, issued

September 29, 1988, for the special purpose of patrolling, agricultural, forest and wildlife conservation, aerial surveying, weather control, and carriage of cargo.

Production Basis None.

Equipment The basic required equipment as prescribed in the applicable Airworthiness

Regulations (see Certification Basis) must be installed in the aircraft for certification. In addition, Airplane Flight Manual, T.O. 1U-16(H) A-1, T.O. 1U-16(H)B-1, or NAVWPS 01-85AB-1 must be included in the airplane.

Note 1. Current weight and balance report including a list of equipment in certificated empty weight and

loading instructions when necessary must be in the aircraft at the time of original certification and

at all times thereafter.

Note 2. The following placard must be installed in clear view of the pilot:

THIS AIRPLANE MUST BE OPERATED AS A RESTRICTED CATEGORY AIRPLANE IN COMPLIANCE WITH THE OPERATING LIMITATIONS STATED IN THE FORM OF

PLACARDS, MARKINGS, AND MANUALS.

No persons may be carried on a restricted category civil aircraft unless they are necessary for the accomplishment of the work activity directly associated with that special purpose. Reference FAR 91.313.

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Note 3. Models HU-16B, HU-16D and HU-16E airplanes are limited to the following flight hours due to the installation of 7075-T6 spar caps in the wing center section:

HU-16B S/N's	Flight Hrs.	HU-16D S/N's	Flight Hrs.	HU-16E S/N's	Flight Hrs.
0126	8906	7188	9656	1265	9430
1030	9505	7214	9855	1272	9723
9099	8307	7227	9043	1294	9388
51-5291	8936	7245	9855	137904	8932
51-7195A	8646	7226	9049	137911	8312
51-0043	8696	1311	8284	2135	9305
86B	9855	146426	7871		
51-0019	9630	141278	8454		
51-7162	9500				
49-097	8073				
99-7213	9593				
51-7187	8254				
51-025	8621				
2132	8785				
51-7246	8922				
51-5303	8433				
146427	8371				
51-7240	9157				

- Note 4. These airplanes must be serviced and maintained in accordance with T.O. 1U-16(H)A/B-2 or NAVWPS 01-85AB-2 and an approved progressive maintenance program.
- Note 5. A. The airplane must satisfactorily pass an inspection for possible hidden damage and for workmanship and materials used in making any repairs and/or alternations. A FAA 8130-9 must be completed.
 - B. The maintenance, overhaul, and modification records must be reviewed for changes made by the military services that may have an effect on the airworthiness of the airplane. Modifications and changes of equipment which effect the safety or performance of the airplane must be approved by the Federal Aviation Administration.
 - C. Conduct an inspection of the engine support structures to verify their structural integrity.
 - D. This aircraft is not equipped or approved for operation with APU.
 - E. The modification data plate in restricted category must be installed next to manufacturer's data plate in cockpit.
- Note 6. Carriage of hazardous materials is prohibited unless compliance is shown with applicable regulations in the Code of Federal Regulations, Title 49, Part 175.
- Note 7. The following note must be placed under "exceptions" on all export certificates of airworthiness for this airplane. "This airplane is type certificated in the restricted category and may not meet the applicable airworthiness code as provided by Annex 8 to the Convention of International Civil Aviation."
- Note 8. Flight into known icing is prohibited.
- Note 9. Approved cargo nets/straps to be FAA TSO'd equipment.
- Note 10. Two (2) oxygen masks providing 100% breathing capacity for both pilots and two (2) smoke masks are required for flight in the aircraft at all times, if in cargo configuration.
- Note 11. Two (2) fire extinguishers installed one in cockpit and one in cabin are required.
- Note 12. Deleted