## DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION

H8WE Revision 2 SCHNECK Vertol (Army) CH-21C

5 March 1979

## TYPE CERTIFICATE DATA SHEET NO. H8WE

This data sheet which is a part of type certificate No. H8WE prescribes conditions and limitations under which the product for which the type certificate was issued meets the airworthiness requirements of the Federal Aviation Regulations.

Type Certificate Holder

Jack A. Schneck Route 1, Box 114

Williams, California 95987

## I - Model U.S. Army CH-21C (Restricted Category) approved 19 November 1971

Engine Wright R-1820-103

Fuel 115/145 or 100/130 min. grade

**Aviation Gasoline** 

Engine limits

			MP		
Low Impeller Gear Ratio	HP	RPM	IN HG	ALT.	
Maximum Continuous	1275	2500	46.5	S.L.	
Maximum Continuous	1275	2500	45.5	2200	
Take-Off (5 min)	1425	2700	51.5	S.L.	
Take-Off (5 min)	1425	2700	50.5	1300	
(Straight line manifold pressure variation with altitude).					

NOTE: Operation at 2600 RPM prohibited.

High Impeller Gear Ratio

980	2500	43.0	10,800
980	2500	42.0	15,600
1100	2600	50.0	9,800
1100	2600	49.0	13,000
	980 1100	980 2500 1100 2600	980 2500 42.0 1100 2600 50.0

(Straight line manifold pressure variation with altitude).

NOTE: Operation at 2700 RPM prohibited.

Rotor Limits and Engine Operating Speeds Power Off (Rotor Tach.) Maximum 258 RPM Minimum 233 RPM Power On (Engine Tach.) Maximum 2500 RPM Minimum 2300 RPM

Maximum 2700 RPM; 30 minute limit

Airspeed limits Vne (Never exceed) 102 knots IAS Sea Level

(See FAA approved Rotorcraft Flight Manual Supplement).

C.G. range 35 in. fwd. (Sta. 319) to 6.5 in. aft

(Sta. 360.5) of datum

Datum Point midway between rotors (Sta. 354)

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Maximum weight 13,500 pounds (See FAA approved Rotorcraft Flight Manual Supplement for

variations with altitude and External Loads).

Minimum Crew 1 (pilot)

Fuel capacity Main Fuel Tank 304 Gal. (Sta. 395.0)

(300 Gal. usable in forward flight)

Oil capacity 21.5 Gal. (Sta. 540.3)

Other Operating

Department of the Army Technical Manual TM 55-1520-205-10 "Operator's Manual CH-21 Helicopter" dated 1964 and Schneck FAA approved Flight Limitations

Manual Supplement dated 19 November 1971 or Subsequent FAA Approved

Revision.

Serial Nos. Eligible All Army Serial Nos.

Certification basis FAR 21.25(a)(2) effective 1 February 1965 Type Certificate No. H8WE issued

19 November 1971 for special purposes of: Carriage of external loads per

FAR 133 and Agriculture Frost Control by air agitation. Date of Application for Type Certificate 2 April 1971.

None. No helicopter may be produced under this type approval. Production basis

Prior to original Certification of each aircraft, an FAA representative must perform an inspection for workmanship, materials, and conformity with

the approved type design.

Equipment: The basic required equipment as prescribed in the applicable airworthiness

regulations (see certification basis) must be installed in the helicopter for

certification. In addition, equipment necessary for the particular special purpose

must be installed.

NOTE 1. Current weight and balance report including list of equipment included in certificated empty weight, and loading instructions when necessary, must be in each helicopter at the time of original

airworthiness certification and at all times thereafter.

NOTE 2. The following placard must be installed in clear view of the pilot.

> "This helicopter must be operated in compliance with the operating limitations specified in Department of the Army Technical Manual TM 55-1520-205-10 and the FAA approved supplement

thereto."

For additional placards see flight manual supplement.

NOTE 3. Prior to civil airworthiness certification, the following must be accomplished:

Modifications in accordance with FAA Sealed Jack A. Schneck Drawings AJS, AJS-A, 1JS,

2JS, 3JS, 4JS and 5JS.

The provisions of AD 60-7-8.

This model helicopter must be serviced and maintained in compliance with Department of the Army NOTE 4.

Technical Manual TM 55-1520-205-20, "Organizational Maintenance Manual CH-21C helicopter"

dated 1964.