

F-16Block 50/52 (GE129)
Checklists - Main Volume

Not suited for Real Operations
Made for FALCON 4 and suitable only for
BMS version

CONTENTS

Check-lists	Cł	nec	k-l	ists
-------------	----	-----	-----	------

1	Contents
2	Verify Checks / Before Engine Start
3	Engine Start / Engine Check AT idle / After Engine Start
4	After Engine Start (Continued) item 2 to 8
5	After Engine Start (Continued) item 9 to 13
6	After Engine Start (Continued) item 14 to 16
7	After Engine Start (Continued) item 17 to 19
8	Before Taxi / Taxi / If checks
9	Before Take Off / Normal Take Off
10	Airborne / Aerial Refuelling
11	Fence In / Initial Point
12	Egress / Fence Out / IF Checks Mnemonics
13	Descent / Approach / Before Landing
14	Final Approach / Landing / After Landing
15	Pre Engine Shut Down / Shut Down
16	Hotpit refuel / Supplemental procedures: ILS

Annex1: Blank page for notes

NOTE:

Refer to Cockpit Interior check Rev 1107 for placing all switches before entering the aircraft

VERIFY CHECK

The following items are important switches that if not correctly positioned, could cause a safety hazard and/or improperly operated systems during engine start.

Please refer to cockpit / interior checklist for a full cockpit check.

FUEL MASTER switch ON - Guard down

2 FNGINF FFFD knob NORM

EPU switch NORM - Guard down 4. ENG CONT switch PRI – Guard down

5. THROTTLE OFF

6 LD GEAR handle Confirm Down and locked

HOOK switch UP

8 MASTER ARM switch SAFE (OFF)

AIR SOURCE knob NORM

BEFORE ENGINE START

1. MAIN PWR switch BATT:

Verify FLCS RLY light ON 2. FLCS PWR TEST switch TEST and hold

Verify lights ON

ACFT BATT TO FLCS FLCS PMG

FLCS PWR (4)

Verify FLCS RLY light OFF

FLCS PWR TEST switch Release 4 MAIN PWR Switch MAIN PWR:

Verify lights ON

ELEC SYS

HYD/OIL PRESS FLCS RLY

SEC

ENGINE

5. EPU GEN & EPU PMG lights Confirm OFF

Communications All set to assigned UHF Backup

7. Canopy Closed - locked - no light

Note:

To prevent possible depletion of battery power, do not allow MAIN PWR switch to remain in BATT or MAIN PWR for more than 5 minutes without engine running.

STARTING ENGINE (GE129)

1. JFS START 2

check JFS light ON

Rev: 1107 BMS 4.32

2. THROTTLE Advance to IDLE at 20% RPM minimum.

3. Idle Detent Toggle (Unless idle/cutoff code enabled in bmsconfig)

4. SEC caution light Check OFF around 20% RPM

5. ENGINE warning light OFF at 60% RPM

6. JFS Switch Confirm OFF (snaps OFF at 55% RPM)

7. HYD/OIL PRESS light OFF between 15 and 70% RPM

Note:

Engine light-off occurs within 10 seconds after throttle advance and is indicated by an airframe vibration and an increase in RPM followed by an increase of FTIT. Without external power connected, only the RPM and FTIT indicators function until the standby generator is online.

ENGINE CHECK AT IDLE

1. FUEL FLOW 700 – 1700 PPH

2. OIL pressure MIN 15 PSI

3. NOZ POS Greater than 94%

4. RPM 62 – 80%

5. FTIT Below 650°C

6. HYD PRESS A&B 2850 - 3250psi - around 12 O'clock position

7. Throttle cutoff release Check – Attempt to retard the throttle to OFF

without depressing the cutoff release.

AFTER ENGINE START

1. TEST switch panel check:

- PROBE HEAT switch: PROBE HEAT: check caution light OFF

TEST: check caution light flashes

OFF

- Fire and Overheat Detect Button: TEST & HOLD

- Check ENG FIRE Warning light ON

- Check OVER HEAT caution light

- Check MASTER CAUTION light ON

- MAL&IND LTS button: DEPRESS and HOLD

Proper VMS operation is verified by the presence of each word in priority sequence.

Rev: 1107 BMS 4.32

AFTER ENGINE START (Continued)

2. AVIONICS POWER Panel

a. MMC (FCC) switch: ON

b. ST STA (SMS) switch: ON c. MFD switch: ON

c. MFD switch: ON d. UFC switch: ON

e. DL (MAP) switch: ON

f. GPS switch: ON

g. INS: Select ALIGN NORM

3. SNSR PWR panel:

a. LEFT HDPT switch: OFF unless required

b. RIGHT HDPT switch: As required

c. FCR switch: FCR d. RDR ALT switch STBY

4. HUD Panel: As desired

Set HUD SYM WHEEL ON

5. CNI (C&I) knob: UFC

6. DTC: Load (always load the DTC prior to setting up

the UFC subpages)

7. UFC radio: Set COM1 & COM2 frequency as briefed.

8. MFL: Reset (MFD TEST page)

9. SEC check:

May be delayed until the BEFORE TAKEOFF

a. THROTTLE: IDLE

b. TOE BRAKE: ENGAGE, no PARKING BRAKES

c. ENG CONT switch: SEC

d. SEC Caution Light: ON - Nozzle: Less than 5%

e. RPM: Stabilized f. THROTTLE: Snap to MIL

Snap to MIL

then snap to IDLE when RPM reaches 85%.

Check for normal indication and smooth operation.

g. NOZ POS: 10% or less within 30 sec after selecting SEC.

h. ENG CONT switch: PRI

i. SEC Caution Light: OFF

j. NOZ POS: Greater than 94%

10. FLIGHT CONTROLS: CYCLE & CHECK

AFTER ENGINE START (Continued)

11. FLCS BIT: Initiate and monitor.

Position BIT switch to BIT. The RUN light ON FLCP illuminates. At successful completion of BIT (approximately 45seconds) the RUN light goes OFF, the BIT switch returns to OFF and the FAIL light and FLCS warning light remain OFF. A BIT pass message appears on the FLCS MFD page

Note:

If the FLCS BIT reports a failure through the FLCS warning light and the FAIL light on the FLCP, the failure cannot be reset. The BIT must be reinitiated. In this case, the RUN light and the FAIL light are simultaneously illuminated for the first steps of the BIT, after which the FAIL light goes OFF unless BIT detects a subsequent failure.

12. SPD BRK switch: Cycle

13. **WHEELS down lights**: Three green

14. FUEL QTY SEL knob Check

The following Values are based on JP-4 or JP5/8

a. Totalizer qty: Check according to flight planning.

b. TEST: FWD/AFT fuel low lights ON

Tot: 6000 lbs

A/L – F/R: 2000 lbs

c. NORM: A/L : 2675/2810 lbs

F/R: 3100/3250 lbs

d. RSVR: both 460/480 lbs e. INT WING: both 525/550 lbs

f. EXT WING: both 2300/2420 lbs (if 370-gallon carried)

both 3750/3925 lbs (if 600-gallon carried)

g. EXT CTR: F/R: 1800/1890 lbs

A/L: 0 lbs

h. FUEL QTY SEL: NORM

15. EPU FUEL QTY: 95 – 102%

AFTER ENGINE START (Continued)

16. AVIONICS (Program as required and verify (manual or DTC)

a. Threat Warning Aux: ON

b. CMDS

RWR switch:

JMR switch:

CHAFF cmds switch:

ON

ON

ON

ON

ON

MODE knob: Set as required PGRM knob: Set as required

c. ECM switch: Set as required (OPR)

d. Threat Warning prime

Handoff Diamond Float mode (short press)

e. MFD

S-Jettison: Preset Jettison and exit S-J mode

Master Mode: Preset SMS as required for each MM

f. AUDIO

COM1&2 Volume SET & check SET & check SET & check ILS Volume knob SET & check

g. DED - UFC

ALOW – MSL – BINGO: Check CRUS – TACAN - IDM: Check

Bullseye: SET & Mode Selected.

17. DBU CHECK (AFTER FLCS BIT completed)

a. DIGITAL BACKUP switch: BACKUPb. DBU ON warning light: Verify ON

c. Operate controls: All surfaces respond normally

d. DIGITAL BACKUP switch: OFF

e. DBU ON warning light: Verify OFF

18. TRIM CHECKS

a. TRIM AP DISC switch: DISC

b. Stick TRIM buttons: Activate in ROLL and PITCH

No control surface, no indicator motion

c. TRIM AP DISC switch: NORM

d: Stick TRIM buttons: Check and centre

Control surface & indicator motion

e. Rudder trim check: YAW TRIM knob:

Check and centre

AFTER ENGINE START (Continued)

19. AIR REFUEL CHECKS

a. AIR REFUEL switch: OPEN

CHECK RDY light ON, DSC light OFF

b. A/R DISC button: Depress

DSC light ON; RDY Light OFF

then 3sec later, RDY light ON, DSC light OFF

Rev: 1107 BMS 4.32

c. AIR REFUEL switch: CLOSE

20. EPU CHECK

a. EPU GEN and EPU PMG lights: Confirm OFF

b. O²: 100%
c. Toe brakes: Engage
d. EPU switch: OFF
e. EPU switch: NORM
f. THROTTLE: 80%

g. EPU/GEN TEST switch: EPU/GEN and hold.

Check lights: EPU AIR light ON

EPU GEN and EPU PMG light OFF FLCS PWR lights ON

EPU RUN light ON within 5 seconds

h. EPU/GEN TEST switch: Release (OFF)

i. THROTTLE IDLE i. O²: NORMAL

If no run light within 10 sec, reinitiate test with throttle at IDLE +15%

21. OBOGS CHECK (At least 2 minutes after engine start)

a. OBOGS BIT switch: BIT

b. VERIFY LIGHT: OXY LOW (right brow) ON for 10sec then OFF

c. Pressure: CHECK 25-40 PSI d. Mode Lever: PBG/ON (as required)

e. Diluter lever: NORM f. EMERGENCY lever g. FLOW indicator Check

BEFORE TAXI

1. Landing Lights ON

Drift Co Switch Set Norm

3. INS Check Check Stage 8.3
Check ALIGN flashes in HUD

4. INS switch NAV position

5. Aircraft Lights As SOP (AC ON – Wing/fus: ON – FLASH)

6. QNH Confirm QNH received from lead or tower

7. Radio Tower Remove chocks

Note 1: Beware of spending excessive time checking the aircraft. Always refer to your next TOS.

Note 2: Be sure the AUX flag disappears from the ADI before scrambling.

As long as GPS switch is ON, the Falcon INS will be accurate from 90 seconds after initial alignment (AUX flag going OFF)

Note 3: Excessive use of wheel brakes and/or differential braking is to be avoided Maximum safe taxi speed on ramps is 20Kts. (15kts in turns)

Max 80% RPM

TAXI

NoseWheel Steering
 Parking Brake
 Engage
 Release

Seat Armed – Caution light OFF

4. Wheelbrakes Test

5. IDM Check in sequence

IF CHECKS

- 1. Pressure Instruments
 - AIRSPEED: Zero
 - ALTIMETER: Set
 - VVI: Zero Remember possible errors.
- 2. Gyroscopic Instruments
 - TURNS: Needle/balls HSI Following
- 3. Navigation Instruments
 - NAV: Check correct bearings for WAYPOINTS
 - TACAN: Set TCN channel and Course for Departure
- 4. Miscellaneous:
 - HUD Compass tape Track heading change
 - HSD Compass tape Track heading change
 - HSI Compass tape Track heading change
 - STDBY Compass Track heading change
 - Clock and Chrono: Check and Reset
 - Engine instruments: Check

5 FNG CONT switch

BEFORE TAKE OFF

PROBE HEAT switch PROBE HEAT

ALT FLAPS switch. NORM 3. MANUAL TF FLY UP switch **ENABLE**

4. Trims Check PITCH and YAW centred.

ROLL as required

Rev: 1107 BMS 4.32

PRI

6. Speedbrake Check closed

7. Departure Clearance Received

8. Radar Altimeter Set ON

9. Stores Config Switch Cat1/Cat3 as required

10. GND JET ENABLE switch As required

Check feeding then NORM 11. External Tanks

12. Flight Controls Cvcle

13. OIL pressure Check PSI

14. All warning & caution lights Check OFF

15. Tacan Verify reading if available

16. Review Speeds Commit to memory Rotation, T/O, Climb speeds

NORMAL TAKE OFF

1 HSI Check on Runway heading

Toe brakes HOLD

3. RPM 90% Check gauges & lights

Oil pressure increase - nozzle closing Engine instruments in the green

NO CAUTION / NO WARNING

4. Brakes Release

Full MIL, AB as required 5. Throttle

NWS Disengage at 70 kts 7. Rotation

As computed Positive Climb

(VSI + Alt) Brakes, Gear Up

- Normal engine operation during MIL takeoff is indicated by an exhaust nozzle position of 15% or less after 5 seconds at MIL.
- Normal engine operation during an AB takeoff is indicated by the nozzle preopening up to 10% more than MIL when AB is first selected. AB is indicated by an increasing fuel flow and nozzle position.
- Apply Power smoothly, note computed speeds for 8-12 degrees pitch rotation as briefed.
- Do not exceed 14 degrees pitch in rotation.
- Insure LG is up and locked before exceeding 300 knots.
- Since TEF and LG retract at the same time, do not rush LG retraction after takeoff, a significant loss of lift may occur.

AIRBORNE / CLIMB

Rev: 1107 BMS 4.32

1. Landing Light Off

2 U/C Check Retracted - handle light Off

3. Engine Gauges in the Green

4. FUEL Verify Tank feeding and set NORM

5 Radio Call airborne or visual

6. DED STP mode. Select NXT

Cycle - As Required 7. MFD

Set Drift 8. DRIFT CO Switch Set Formation and Route Wingman

AERIAL REFUELLING

Tanker rejoin :

 Radio Reguest Refuelling (within 10Nm)

Select TCN Channel (Texaco) 2. TCN

3. TCN Mode

SET A/A TR

Course to Intercept (HSI) 4. Heading Altitude Tanker ALT - 1000 Ft

Before Precontact:

12. ANTI COLLISION light

Master ARM Check Safe

7 Sensors Check Nose Cold EW Mode knob &ECM STBY and OFF

9 FCR STBY

10. RDR ALT STBY

11. EXT Lights DIM (night) - STEADY

13. AIR REFUEL switch Open

14. AR status indicator. Check RDY Light On

15. Seat SAFE (As desired)

Contact:

Boom Operator Follow Instructions and Lights

OFF at Night

Decrease power

17. AR status indicator Check AR/NWS Light On Monitor (List - #2) 18. Fuel Transfer

Disconnect:

19. A/R DISC button Depress

20. Throttle

Post Air refuelling: ARM

21. Seat

F-16 checklists main volume

Air Refuel switch CLOSE

Radio Call DONE refuelling.

24. Master Arm / SMS As required

As required 25. Tacan EW Mode knob &ECM As required

27. FCR As required

28. RDR ALT As required

As required 29. EXT Lights

Note: Tanker overtake speed

Rev: 1107 BMS 4.32

Over 1Nm: 100 Kts overtake

6000 Ft : 60Kts 5000 Ft : 50Kts

Decrease overtake speed by 10 Kts for every 1000 Ft closure.

When within 1000 Ft to Tanker: Do not exceed 10Kts overtake.

FENCE IN

1. Master Mode As Required AG or AA

2. Master ARM Set ARM

Radar As Required
 Chaff/ Flares PGM mode As Required

5. ECM Jammer As Required

6. RWR Check On

7. RWR Mode Diamond Float mode or as required

7. PFD Check no Faults

8. Master A/C Lights Check Off

9. MFD Cycle/ Req data

10. A/G Weapons Set release parameters

11. LASER Switch ON if required

12. Volumes Check threat, com, msl vol

13. TGP pod Activate if required – double check
14. AGM65 Missile power Check ON if required – double check

15. AIM-9 Cooling head Check Cool

16. CAT config Check correct

17. Radio Flight Set Defensive Formation

Note:

Avoid Radio Chatter when entering enemy airspace unless in case of emergency. Use A/C or hands signals instead.

INITIAL POINT

Radio Flight Split, Weapons Free, Engage

Master ARM Check ARM
 Weapons Check SET

4. Attitude Check Speed and ALT

5. DED A-LOW SET on Weapon Min release

6. Threat Assume (A/A) - AWACS

7. Master Mode / Radar As Required

8. CounterMeasures Check As Required

9. Radio Call in HOT

EGRESS

Rev: 1107 BMS 4 32

1. Heading Check to friendly airspace

Caution Panel Check for damage
 Master Mode As Required (A/A)

4. Awacs Check Nearest threat5. MFD Cycle As Required

6. Store config7. ECM Jammer8et Cat I (if possible)As Required

ECM Jammer As Required
 EWMS mode+pgr At pilot discretion

9. Flight Rejoin / Cover 10. DED A-LOW Set for Egress

11. Flight Check Status & Fuel - Rejoin

Note:

When engaging an A/A threat, Jettison remaining A/G stores, and select Catl config. If threat is less than 10 Nm, Use Dogfight Mode

FENCE OUT

Threat Assume A/A Threat - AWACS
 Master ARM Set Safe (According to Threat)

3. Laser switch Set Off

3. Master Mode Set Nav

4. Radar Off (According to Threat)5. ECM Jammer Off (According to Threat)

6. RWR Mode As required 7. Chaff/ Flares Auto disp Set Off

8. PFD Check no Faults

9. Radio Flight Fuel Check (Dest or Alt)

IF CHECKS MNEMONIC

	<u> Holding/enroute</u>		Approach setup
W	Weather	М	Minimas
Н	Holding	Α	Altimeter
0	Obtain app clearance	I	Initial descent rate
L	Letdown plate review	L	Letdown plate
D	Descent checks	M	Missed Approach
S	Speeds	Α	Approach speeds
	•	N	Navaids

Combatsimchecklist.net 2403.72

Note:

For Approach use the F4 Letdown plates

DESCENT

Master Mode Set NAV
 Master ARM Set Safe

3. Altimeter Set & Check (transition ALT)

Check altimeter readings vs HUD altitude

Rev: 1107 BMS 4 32

4. Approach plates Reviewed

5 Instr Mode Select switch TCN/II S or NAV/II S

6. TACAN channel Set according to approach plate

7. HSI course and bearings Set according to approach plate

8. GPS Input coordinates of IAF9. Speeds Compute final approach speeds

APPROACH

See Quick Reference charts volume to compute speeds

Radio Tower Call Inbound

2. Fuel Check Quantity/Transfer/Balance

3. At IAF Follow ATC procedures unless Visual

Approach.

BEFORE LANDING

1. Radio Tower (5Nm out) Request Landing

2. A/C Weight Verify/Update Vref

3. A/C LDG/Taxi Lights Set On

4. Gear Check 3green-handle light off

Speed brake Fully Deployed

6. Drift Co switch Set Norm

7. Traffic Announce traffic in sight if required

Radar/EW Check all STBY

Note:

Unless previously cleared aerobatic manoeuvre is not permitted over the airfield.

Pitch and bank should not exceed 70° IAS<250.

FINAL APPROACH

1. Speed brake Extended

Down 3 greens 2. Gear

Speed Vref as computed

Green: 11° 4. AoA 5. Touchdown 11 to 13° AOA

Note:

Final approach speed/13° AOA Cross Check:

136 kts + 4 kts per 1000 Pound of FUEL/STORE weight

LANDING

1. Speed Throttle Idle

2. AOA Maintain Max 13° for aerobraking

3. Speed 100 kts Ride the nosewheel on the ground

Maintain AFT stick

Engage NWS at taxi speed or when reg. Wheel brakes As required

Note:

Smoothly apply moderate to heavy braking to decelerate to taxi speed. Using less than moderate braking increases the likelihood of a hot brake(s)

AFTER LANDING

CLOSE 1. Speedbrake

2. PROBEHEAT switch OFF 3. ILS OFF

4. Landing/ Taxi Lights As required

5. Radar Alt OFF

PRIOR TO ENGINE SHUT DOWN

1. Radio (tower menu) Request chocks in place

2. Ejection Seat Safe3. RWR PWR OFF

4. JMR&ECM PWR OFF
5. Chaff & Flares CMDs OFF

6. HUD ICP SYM knob OFF

7. L/R Hardpoints Power OFF

8. FCR Power OFF

9. MMC (FCC) Power OFF

10. ST STA (SMS) Power OFF 11. MFD Power OFF

12. UFC/DED Power OFF

13. D/Link Power OFF

14. GPS Power OFF 15. INS Power OFF

16. EPU OFF

(No crew chief able to insert the EPU pin)

Rev: 1107 BMS 4.32

17. CNI switch BACKUP

ENGINE SHUT DOWN

AIR Source Set OFF
 Radios & Volume knobs All OFF

3. Throttle - Stabilize at 75-78% RPM for 5-10 sec

- Idle to allow nozzle to open (1 to 2 sec)

4. Throttle (Idle Detent) Cut OFF position

5. JFS RUN light Check

After Main GEN drops offline:

6. EPU Light check EPU GEN / EPU PMG lights OFF

7. Engine FEED switch Set OFF8. Master LIGHT switch OFF

9. Canopy Open 10. Main Power OFF -2 clicks when RPM < 20%

11. Oxygen regulator OFF & 100%

HOTPIT REFUEL

Rev: 1107 BMS 4.32

Prior to HOTPIT Entry

1. AFTER LANDING CHECKS Complete

2. Radio Frequency Check proper tower frequency tuned

3. AIR REFUEL switch Open: RDY light ON

Power OFF 4. TACAN power knob

5. GND JETT ENABLE switch OFF

Prior to Hot Refuelling

1. EPU switch (safety pin in) OFF

2. Canopy As desired

3. Radio request Hot Refuelling

During Hot Refuelling

Monitor Tower freq & guard 2. Radio freq

3. Flight controls Do not touch - Ensure hands are visible

Hot Refuelling complete

1. AIR REFUEL switch CLOSE

2. EPU GEN & EPU PMG lights Confirm OFF

3. EPU switch (safety pin out) NORM

4 Taxi Taxi clear of the hotpit area and contact

Tower

Note:

Hotpit refuelling requires ground crew to establish intercom communication. inspect tires and install the EPU safety pin. This last action is simulated by

switching the EPU OFF during hotpit refuel

SUPPLEMENTAL PROCEDURE: ILS

1. DED Verify CNI display

Depress and Release 2. T-ILS button

Key in and ENTR 3. ILS frequency

4. DCS Position asterisks about selectable items.

5. HSI Set Inbound localizer course

6. INSTR Mode knob ILS/TCN or ILS/NAV

PAGE INTENTIONALLY LEFT BLANK

USE FOR NOTES



F-16 Block 50/52 (GE129/PW229) Checklists - Quick Reference Charts

Not suited for Real Operations Made for FALCON 4 and suitable only for **BMS** version

CONTENTS

Quick reference charts

- 1. Contents.
- 2. Takeoff Roll Trim with Asymmetric stores.
- 3. Takeoff and landing Crosswind limits.
- 4. Takeoff and Landing datacard.
- Takeoff Factor.
- Takeoff Speed and Distance.
- 7. Takeoff Speed and Distance (Notes)
- 8. Refusal Speed.
- 9. Landing Distance.
- 10. Short Field Landing Distance.
- 11. Short Field Landing Distance in SEC.
- 12. Climb / Optimum Cruise (Drag factor = 1.0)
- 13. Climb / Optimum Cruise (Drag factor = 18.0)
- 14. Climb / Optimum Cruise (Drag factor = 69.0)
- 15. Climb / Optimum Cruise (Drag factor = 86.0)
- 16. Ambient Air Temperature

Annex1: Blank page for notes

NOTE:

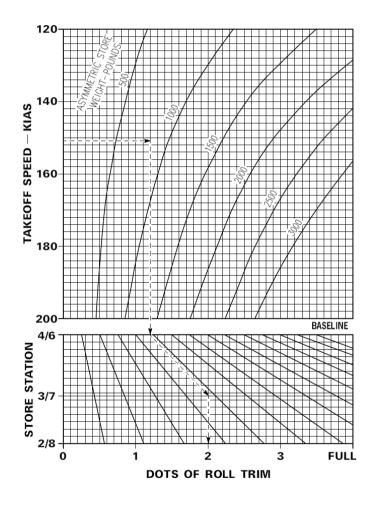
Charts are from the F-16 block 52 PW 229 engine but should suit F16 block 50 GE 129 engine Allow 5% variation

TAKEOFF ROLL TRIM WITH ASYMMETRIC STORES

CONFIGURATION: LEF SCHEDULED & TEF at 20°

Notes:

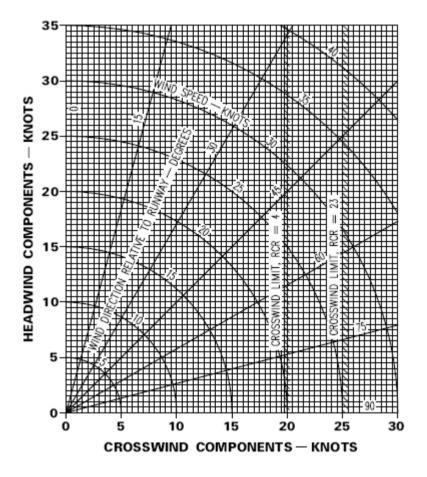
 Increase Takeoff speed 2kts for each dot of roll trim applied to compensate for reduced lift. Takeoff distance increases proportionately to the speed increase.



TAKEOFF & LANDING CROSSWIND LIMITS

Notes:

- Crosswind limits for RCR values 4-23 may be obtained by interpolating between the limits shown.
- Enter chart with steady wind to determine headwind component and with maximum gust velocity to determine crosswind component.



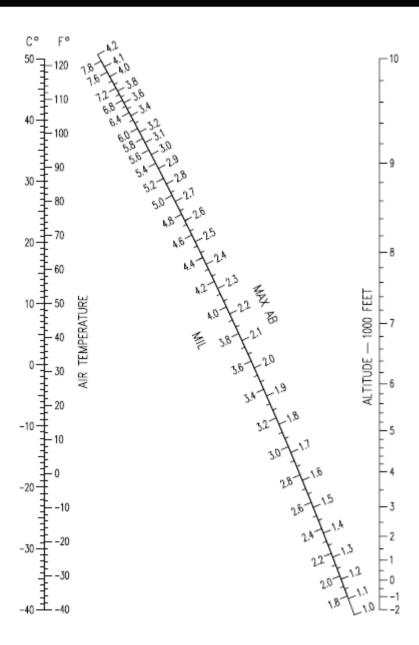
Rev: 1107 BMS 4.32

TAKEOFF & LANDING DATA CARD

CONDITIONS:

	TAKE O	FF	LANDING	i
G.W	·			
Runway Conditions	·			
Runway Temp		 		
Pressure Altitude				
Wind				
Runway Lenght				
Runway Slope				
	TAKEOFF			
Rotation Speed		_ KIAS		
TAKEOFF Speed/Dist		_KIAS		Feet
Refusal Speed		_KIAS		
Max Brake Speed		_KIAS		
	LANDING Immediately after Takeoff		Final Landing	
G.W.		G.W.		
Approach Speed				
Touchdown Speed				-
Landing Distance				_

TAKEOFF FACTOR



TAKEOFF SPEED & DISTANCE

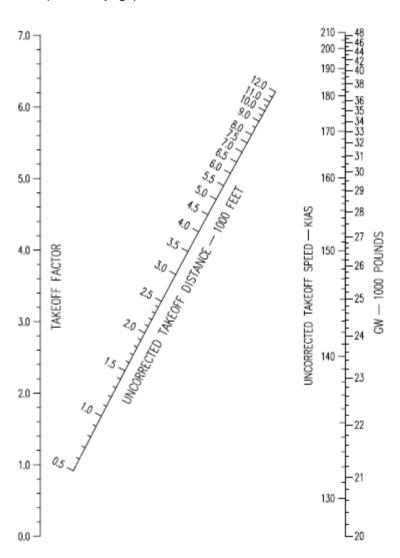
CONFIGURATION:

- . All DRAG Indexes
- . CG = 35% MAC
- . ZERO ROLL Trim

CONDITIONS:

- . All Altitudes
- . All temperatures
- . 10° PITCH attitude

Notes (see next page)



TAKEOFF SPEED & DISTANCE (NOTES)

CONFIGURATION: CONDITIONS:

. All DRAG Indexes

. CG = 35% MAC . ZERO ROLL Trim . All Altitudes

. All temperatures

. 10° PITCH attitude

Rev: 1107 BMS 4.32

- Rotate at 10 KIAS (Non-AB) or 15 KIAS (AB) less than Takeoff speed
- Compute % increase/decrease changes individually.
- Increase takeoff speed 8% and distance 18% for a 8° pitch attitude rotation
- Increase/decrease takeoff speed 0.8 KIAS for each 1% Forward/AFT of 35% MAC
- Increase distance 2% per 100 DRAG INDEX.
- Increase distance 4% per 1% Upslope.
- Increase distance 3.5% per 1% Downslope
- Increase distance 11% per 10 KTS Tailwind.
- Decrease distance 10% per 10 KTS Headwind.
- For takeoff speed correction with ROLL Trim other than zero, refer to Takeoff roll trim with asymmetric stores.

REFUSAL SPEED

CONFIGURATION:

- . All DRAG Indexes
- . Speedbrakes OPEN
- . GW = 32,000 Lb

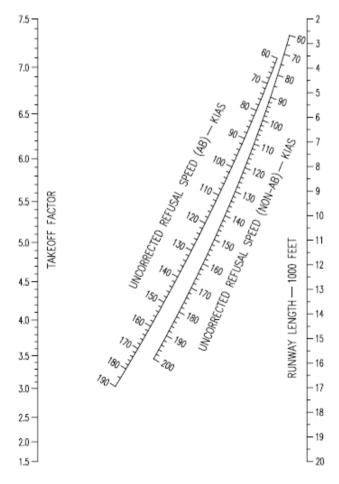
CONDITIONS:

. IDLE selected at refusal speed

Rev: 1107 BMS 4.32

- . Max effort Braking
- . DRY concrete (RCR = 23)

- Compute % increase/decrease changes individually.
- For RCR = 16 (DRY) decrease non-AB / AB refusal speed by 4 / 5 KIAS
- Increase/decrease refusal speed 1.1% / 0.9%with non-AB and 0.7% / 0.7% with AB per 1000 Lb less/additional GW.
- Increase / decrease refusal speed 5 / 5 KIAS with Non-AB and 6 / 6 KIAS with AB per 5KTS headwind /tailwind.



APPROACH SPEEDS

CONFIGURATION: CONDITIONS: . All DRAG Indexes . All temperatures

. All altitudes

. 13° AOA (On speed indexer)

Rev: 1107 BMS 4.32

NOTES:

 Actual approach airspeed at 11/13° AOA may differ by 5 knots due to variation in aircraft CG.

GROSS WEIGHT (LB)	AIRSPEED (KIAS)
19,000	132
20,000	136
21,000	139
22,000	142
23,000	146
24,000	149
25,000	152
26,000	155
27,000	158
28,000	161
29,000	164
30,000	166
31,000	169
32,000	172
33,000	174
34,000	177
35,000	180
36,000	182
37,000	185
38,000	187
39,000	190
40,000	192
41,000	195
42,000	197
43,000	199
44,000	201
45,000	204
46,000	206
47,000	208
48,000	210
	Note : Add 8 KIAS for an
	11° AOA approach
	- F F

SHORT FIELD LANDING DISTANCE

CONFIGURATION:

- . All DRAG Indexes
- . Speedbrakes OPEN

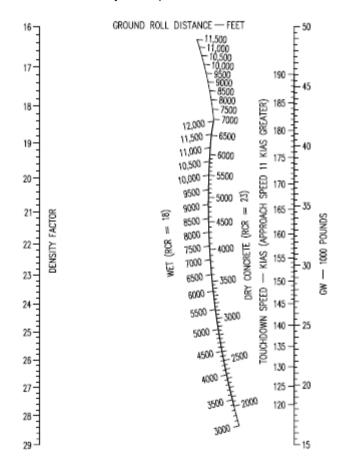
CONDITIONS:

. Touchdown at 13° AOA.

Rev: 1107 BMS 4.32

- . Zero wind & Slope
- . IDLE
- . Max effort braking

- Compute % increase/decrease changes individually.
- Decrease distance 1.5% per 1 KT headwind.
- Increase distance by 2.2% per 1 KT tailwind.



SHORT FIELD LANDING DISTANCE (SEC)

CONFIGURATION:

- . All DRAG Indexes
- . Speedbrakes OPEN

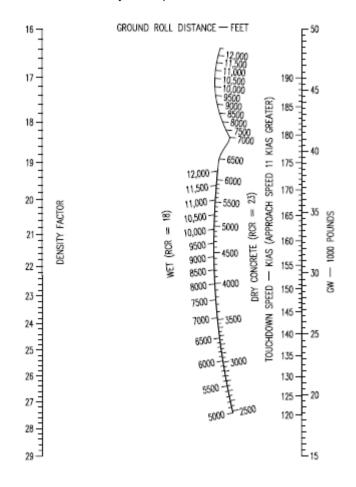
CONDITIONS:

. Touchdown at 13° AOA.

Rev: 1107 BMS 4.32

- . Zero wind & Slope
- . IDLE
- . Max effort braking

- Compute % increase/decrease changes individually.
- Decrease distance 1.5% per 1 KT headwind.
- Increase distance by 2.2% per 1 KT tailwind.



CLIMB / OPTIMUM CRUISE (DRAG FACTOR = 1)

Rev: 1107 BMS 4.32

ENGINE: GE129 FUEL: JP-8 FULL INTERNAL FUEL, NO STORES

- STD day/ Fully serviced Fuel = 7162 LB.
- 800 Lb Fuel allowance for Ground operation and Takeoff/acceleration to MIL climb airspeed. (Assume 30 min ground time)
- Climb at KIAS/MACH NO., whichever is slower

		MIL	CLIMB	OPTIMUM CRUISE		
				N	AT LEVEL (OFF
ALT 1000 FEET	CLIMB @ KIAS MACH	TIME (MIN)	DIST (NM)	FUEL REMAINING AT LEVEL OFF (LB)	MACH/KIAS/KTAS	TOTAL FUEL FLOW (LB/HR)
50						
45	445 0.87	7.6	63.4	5584	0.87/238/502	2622
40	445 0.87	4.5	37.4	5764	0.87/268/502	2454
35	445 0.81	3.2	25.3	5880	0.81/275/466	2327
30	445 0.80	2.4	18.4	5957	0.80/304/471	2541
25	445 0.73	1.8	13.0	6038	0.73/308/442	2655
20	445 0.70	1.3	8.8	6112	0.70/323/430	2889
10	0.59	0.6	3.4	6231	0.59/325/374	3171
0	0.49	0.0	0.0	6362	0.49/326/326	3488

	OPTIMUM CRUISE								
	5000 LB REMA	AINING	3000 LB REMA	AINING	2000 LB REMA	MNING			
ALT 1000 FEET	MACH/KIAS/ KTAS	TOTAL FUEL FLOW (LB/HR)	MACH/KIAS/ KTAS	TOTAL FUEL FLOW (LB/HR)	MACH/KIAS/ KTAS	TOTAL FUEL FLOW (LB/HR)			
50									
45	0.87/238/502	2540	0.85/231/488	2232	0.85/231/488	2128			
40	0.87/268/501	2384	0.84/256/482	2139	0.84/256/482	2069			
35	0.80/272/461	2249	0.80/272/461	2127	0.80/272/461	2070			
30	0.80/304/471	2496	0.76/289/450	2303	0.75/283/442	2220			
25	0.72/301/433	2560	0.70/293/421	2408	0.70/293/421	2369			
20	0.69/318/423	2804	0.66/304/405	2609	0.64/297/396	2514			
10	0.57/316/363	3040	0.54/301/346	2824	0.53/293/338	2715			
0	0.48/316/316	3338	0.46/302/302	3118	0.45/295/295	3008			

CLIMB / OPTIMUM CRUISE (DRAG FACTOR = 18)

Rev: 1107 BMS 4.32

ENGINE: GE129 FUEL: JP-8

FULL INTERNAL FUEL + EXT CENTERLINE TANK, NO STORES

- STD day/ Fully serviced Fuel = 7162 LB + 2040 LB = 9202 LB.
- 800 Lb Fuel allowance for Ground operation and Takeoff/acceleration to MIL climb airspeed. (Assume 30 min ground time)
- Climb at KIAS/MACH NO., whichever is slower

		MIL	CLIME	OPTIMUM CRUISE		
					AT LEVEL (OFF
ALT 1000 FEET	CLIMB @ KIAS MACH	TIME (MIN)	DIST (NM)	FUEL REMAINING AT LEVEL OFF (LB)	MACH/KIAS/KTAS	TOTAL FUEL FLOW (LB/HR)
45						
40	436 0.85	5.8	47.2	7677	0.85/259/488	2750
35	436 0.83	3.8	30.9	7821	0.83/282/476	2668
30	436 0.80	2.8	22.1	7924	0.80/304/471	2823
25	436 0.74	2.1	15.6	8019	0.74/312/447	2947
20	436 0.70	1.5	10.5	8108	0.70/324/430	3126
10	0.60	0.7	3.9	8255	0.60/332/382	3477
0	0.50	0.0	0.0	8402	0.50/330/330	3776

	OPTIMUM CRUISE								
	5000 LB REMA	NINING	3000 LB REMA	AINING	2000 LB REMA	MNING			
ALT 1000 FEET	MACH/KIAS/ KTAS	TOTAL FUEL FLOW (LB/HR)	MACH/KIAS/ KTAS	TOTAL FUEL FLOW (LB/HR)	MACH/KIAS/ KTAS	TOTAL FUEL FLOW (LB/HR)			
45									
40	0.85/259/486	2488	0.84/256/482	2291	0.84/256/482	2214			
35	0.80/272/461	2393	0.80/272/461	2269	0.80/272/461	2210			
30	0.78/295/459	2591	0.75/283/442	2389	0.75/283/442	2343			
25	0.70/293/421	2625	0.70/293/421	2536	0.70/293/421	2495			
20	0.67/311/414	2892	0.65/297/397	2691	0.63/291/388	2594			
10	0.56/309/356	3120	0.53/295/340	2904	0.52/288/331	2789			
0	0.47/310/310	3421	0.45/297/297	3197	0.44/289/289	3081			

CLIMB / OPTIMUM CRUISE (DRAG FACTOR = 69)

ENGINE: GE129 FUEL: JP-8

FULL INTERNAL FUEL + 2 EXT WING TANKS (370GaI), NO STORES

Rev: 1107 BMS 4.32

- STD day/ Fully serviced Fuel = 7162 LB + 5032 LB = 12194 LB.
- 800 Lb Fuel allowance for Ground operation and Takeoff/acceleration to MIL climb airspeed. (Assume 30 min ground time)
- Climb at KIAS/MACH NO., whichever is slower

		MIL	CLIME	OPTIMUM CRUISE		
					AT LEVEL (OFF
ALT 1000 FEET	CLIMB @ KIAS MACH	(MIN)	DIST (NM)	FUEL REMAINING AT LEVEL OFF (LB)	MACH/KIAS/KTAS	TOTAL FUEL FLOW (LB/HR)
45						
40	424 0.85	9.1	74.1	10,363	0.85/259/488	3322
35	424 0.84	5.0	40.7	10,650	0.84/287/484	3170
30	424 0.80	3.6	27.9	10,798	0.80/304/471	3229
25	424 0.75	2.6	19.4	10,921	0.75/315/451	3361
20	424 0.70	1.8	13.0	11,034	0.70/324/430	3482
10	0.60	8.0	4.7	11,222	0.60/333/383	3822
0	0.50	0.0	0.0	11,394	0.50/331/331	4116

	OPTIMUM CRUISE								
	8000 LB REMA	AINING	5000 LB REMA	INING	2000 LB REMA	INING			
ALT 1000 FEET	MACH/KIAS/ KTAS	TOTAL FUEL FLOW (LB/HR)	MACH/KIAS/ KTAS	TOTAL FUEL FLOW (LB/HR)	MACH/KIAS/ KTAS	TOTAL FUEL FLOW (LB,/HR)			
45									
40	0.85/259/488	3012	0.84/256/482	2655	0.84/256/482	2395			
35	0.82/281/475	2891	0.80/272/461	2577	0.78/265/451	2335			
30	0.80/304/471	3051	0.76/286/446	2718	0.73/276/431	2460			
25	0.72/302/434	3070	0.70/293/421	2815	0.69/288/415	2624			
20	0.70/323/429	3322	0.65/301/402	2982	0.61/282/377	2677			
10	0.57/319/367	3532	0.54/299/345	3193	0.50/279/322	2858			
0	0.48/318/318	3814	0.45/301/301	3493	0.43/282/282	3161			

CLIMB / OPTIMUM CRUISE (DRAG FACTOR = 86)

ENGINE: GE129 FUEL: JP-8

FULL INTERNAL FUEL + 2 EXT WING TANKS (370GaI) + CENTERLINE TANK (300GaI), NO STORES

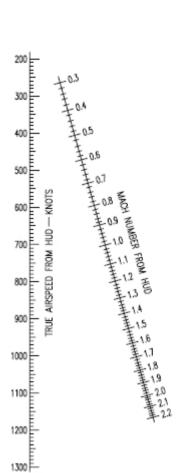
Rev: 1107 BMS 4.32

- STD day/ Fully serviced Fuel = 7162LB + 5032LB + 2040LB = 14234LB.
- 800 Lb Fuel allowance for Ground operation and Takeoff/acceleration to MIL climb airspeed. (Assume 30 min ground time)
- Climb at KIAS/MACH NO., whichever is slower

		MIL	CLIME	OPTIMUM CRUISE		
					AT LEVEL (OFF
ALT 1 000 FEET	CLIMB @ KIAS MACH	(MIN)	DIST (NM)	FUEL REMAINING AT LEVEL OFF (LB)	MACH/KIAS/KTAS	TOTAL FUEL FLOW (LB/HR)
45						
40	424 0.85	9.1	74.1	10,363	0.85/259/488	3322
35	424 0.84	5.0	40.7	10,650	0.84/287/484	3170
30	424 0.80	3.6	27.9	10,798	0.80/304/471	3229
25	424 0.75	2.6	19.4	10,921	0.75/315/451	3361
20	424 0.70	1.8	13.0	11,034	0.70/324/430	3482
10	0.60	8.0	4.7	11,222	0.60/333/383	3822
0	0.50	0.0	0.0	11,394	0.50/331/331	41 16

OPTIMUM CRUISE								
	8000 LB REMAINING		5000 LB REMAINING		2000 LB REMAINING			
ALT 1000 FEET	MACH/KIAS/ KTAS	TOTAL FUEL FLOW (LB/HR)	MACH/KIAS/ KTAS	TOTAL FUEL FLOW (LB/HR)	MACH/KIAS/ KTAS	TOTAL FUEL FLOW (LB/HR)		
45								
40	0.85/259/488	3012	0.84/256/482	2655	0.84/256/482	2395		
35	0.82/281/475	2891	0.80/272/461	2577	0.78/265/451	2335		
30	0.80/304/471	3051	0.76/286/446	2718	0.73/276/431	2460		
25	0.72/302/434	3070	0.70/293/421	2815	0.69/288/415	2624		
20	0.70/323/429	3322	0.65/301/402	2982	0.61/282/377	2677		
10	0.57/319/367	3532	0.54/299/345	3193	0.50/279/322	2858		
0	0.48/318/318	3814	0.45/301/301	3493	0.43/282/282	3161		

AMBIANT AIR TEMPERATURE



ALTITUDE	STD TEMP		
—1000 FT	ç	°F	
SL	15	59	
5	5	41	
10	-5	23	
15	-15	6	
20	-25	-12	
25	-35	-30	
30	-44	-48	
35	-54	-66	
40	-56	-70	
45	-56	-70	
50	-56	-70	
55	-56	-70	
60	-56	-70	

Rev: 1107 BMS 4.32

°F = (9/5 °C) + 32° °C = 5/9(°F - 32°)

PAGE INTENTIONALLY LEFT BLANK USE FOR NOTES