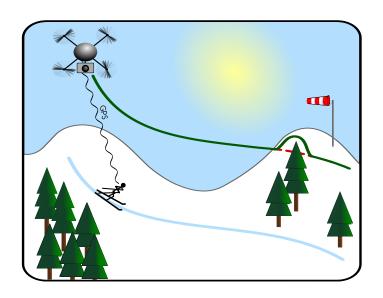
Real Time Control of a Quadcopter

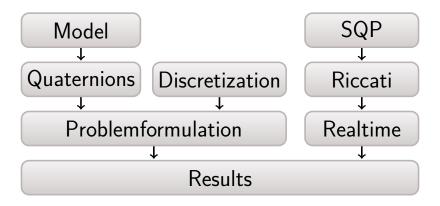
Simon Kick, Philipp Fröhlich, Benedikt König, Annika Stegie

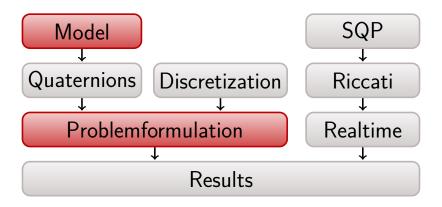
Technische Universität München

11 July 2015

Motivation





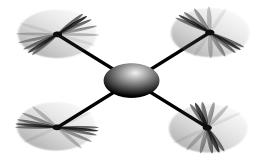


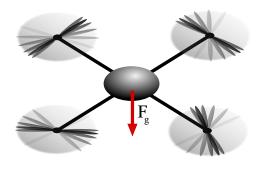
Optimal Control Formulation

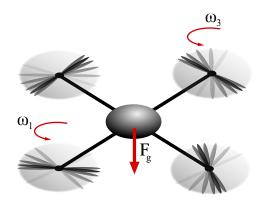
```
\min_{x,u} J(x,u) \quad \text{s.t.} \quad \begin{array}{c} \tilde{h}(x,u) = 0 \\ \dot{x}(t) = f(x(t),u(t)) \end{array}
```

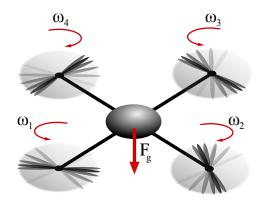
x: state

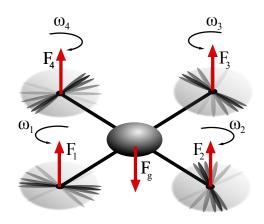
u: control



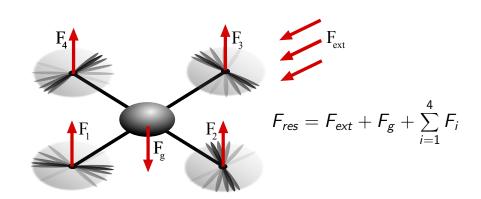


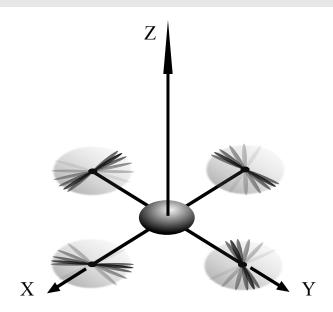


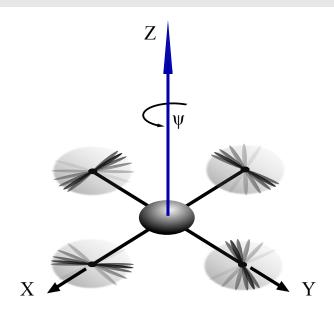


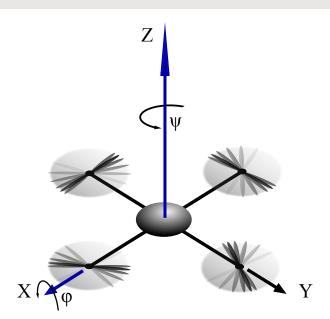


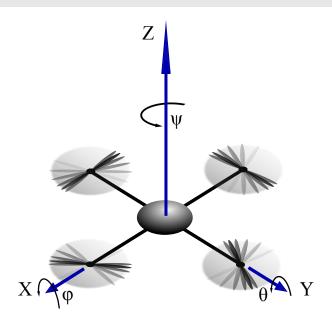
Forces

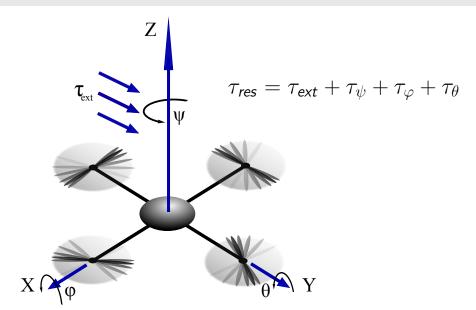








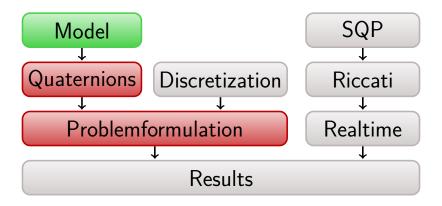




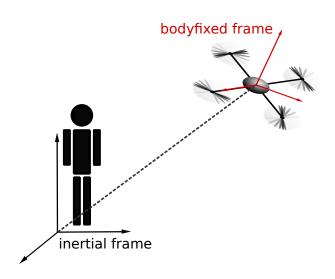
Obtain ODE

$$\left. egin{aligned} F_{res} &= F_{ext} + F_g + \sum_{i=1}^4 F_i \ au_{res} &= au_{ext} + au_\psi + au_\varphi + au_ heta \end{aligned}
ight. \Rightarrow \quad \dot{x}(t) = f(x(t), u(t))$$

$$\begin{cases}
\tilde{h}(x, u) = 0 \\
\dot{x}(t) = f(x(t), u(t))
\end{cases} \Rightarrow h(x, u) = 0$$



Copter Rotation



Quaternions

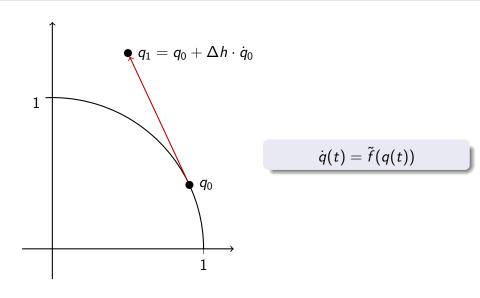
$$q = a + ib + jc + kd$$
 $a, b, c, d \in \mathbb{R}$ \Leftrightarrow $q = \begin{pmatrix} a \\ b \\ c \\ d \end{pmatrix} \in \mathbb{R}^4$

Quaternions

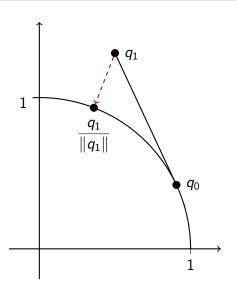
$$q = a + ib + jc + kd$$
 $a, b, c, d \in \mathbb{R}$ \Leftrightarrow $q = \begin{pmatrix} a \\ b \\ c \\ d \end{pmatrix} \in \mathbb{R}^4$

represent rotation
$$\Leftrightarrow$$
 $\|q\|=1$ \Leftrightarrow $q\in\mathcal{S}^3$

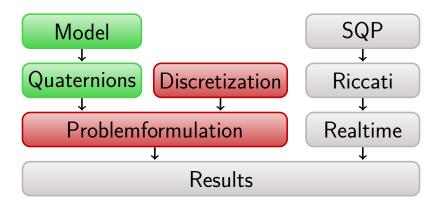
Drift Correction

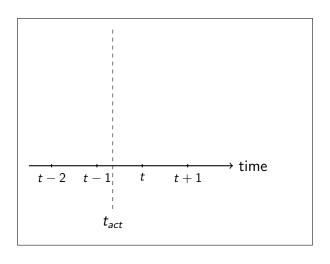


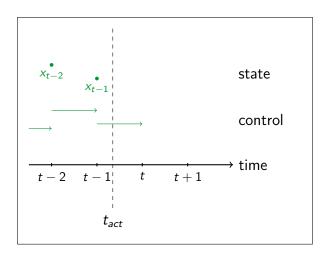
Drift Correction

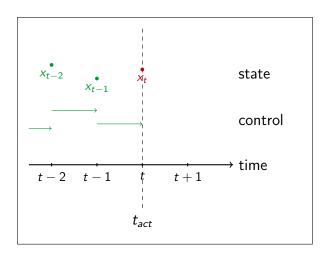


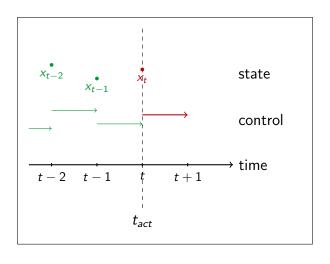
$$\dot{q}(t) = ilde{f}(q(t)) - \lambda(q(t))$$







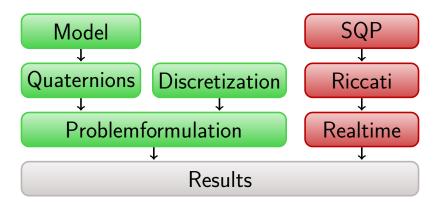




Discrete Problem

$$\min_{x,u} \sum_{i=t}^{t+N} J_i(x_i, u_i)$$
 s.t. $h_i(x_i, u_i) = 0$ $i = t, ..., t + N$

 $J_i(x_i, u_i)$ discretized goal function $h_i(x_i, u_i)$ equality constraints at time i



The Lagrangian

$$L(y) = \sum_{i=t}^{t+N} J_i(x_i, u_i) + \sum_{i=t}^{t+N} \lambda_i^T h_i(x_i, u_i)$$

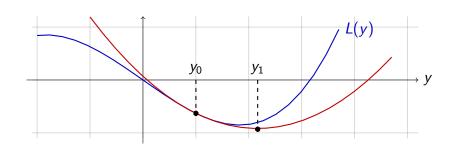
$$y := (\lambda, x, u)$$
 $y^* ext{ optimal } \Leftrightarrow \nabla_y L(y^*) = 0$

The SQP Method

Find y^* :

$$y_1 = y_0 + s$$

$$\min_{s} \frac{1}{2} s^T \nabla^2 L(y_0) s + \nabla L(y_0)^T s$$



Quasi Newton-Method

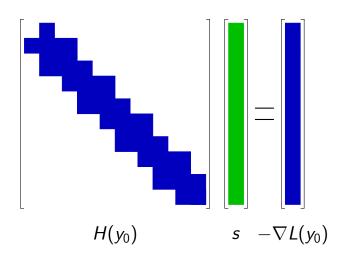
Find s with:

$$\nabla L(y_0) + \nabla^2 L(y_0)s = 0$$

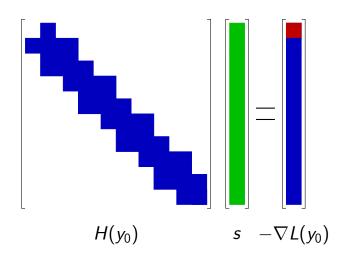
Approximate $\nabla^2 L(y_0)$ and solve:

$$H(y_0)s = -\nabla L(y_0)$$

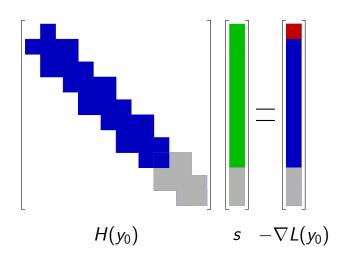
Riccati Recursion



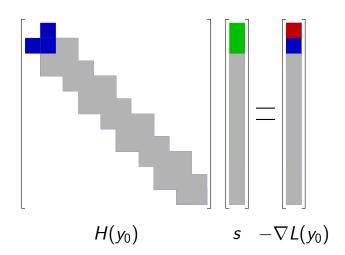
Riccati Recursion



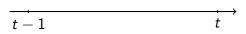
Riccati Recursion



Riccati Recursion



What happens in interval [t-1,t] ?



What happens in interval [t-1, t] ?



• calculate control u_{t-1} (Riccati Part II)

What happens in interval [t-1,t] ?



- calculate control u_{t-1} (Riccati Part II)
- calculate y (Riccati Part II)

What happens in interval [t-1, t]?



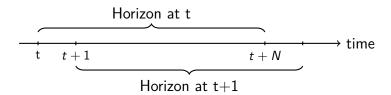
- calculate control u_{t-1} (Riccati Part II)
- calculate y (Riccati Part II)
- prepare u_t (Newton & Riccati Part I)

What happens in interval [t-1, t]?

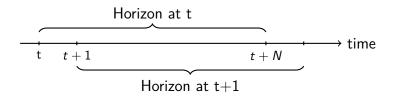


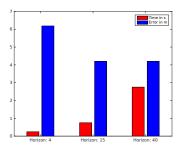
- calculate control u_{t-1} (Riccati Part II)
- calculate y (Riccati Part II)
- \bullet prepare u_t (Newton & Riccati Part I)
- \Rightarrow with horizon 18 this is 28% faster.

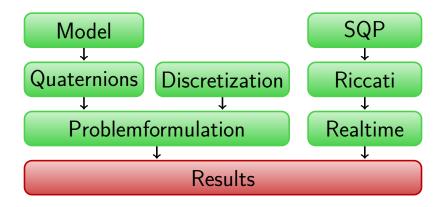
Finite Horizon



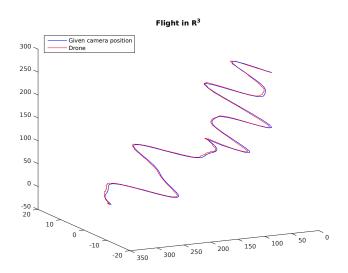
Finite Horizon



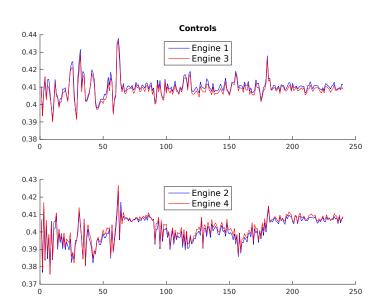




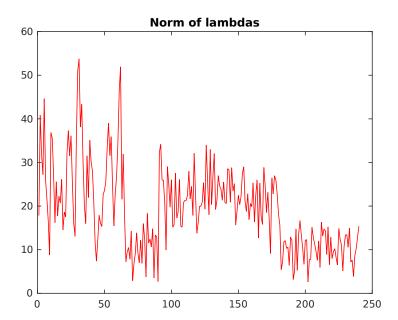
Following a Skier



Following a Skier



Following a Skier



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What we have learned from the Project:

- •
- MATLAB[®] is special
- write tests!!!
- ..