Midterm presentation

The Quadrocopters

Technische Universität München

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Model

Newton-Euler Equations

Forces

$$F_{\text{ext}} = F_g + \sum_{i=1}^4 F_i$$

Torques

$$au_{\mathsf{ext}} = \sum_{i=1}^{4} au_i + (au_\phi + au_ heta)$$

Quaternions

$$q=a+\mathrm{i} b+\mathrm{j} c+\mathrm{k} d \qquad a,b,c,d\in\mathbb{R}$$
 representing rotation $\Leftrightarrow \|q\|=1$ Advantage \to no singularities
$$\mathsf{Problem}\to \|q\|=1 \text{ additional coontraint}$$

Dynamics

$$T(x, u) = M \cdot \begin{pmatrix} \dot{x}_8 \\ \vdots \\ \dot{x}_{13} \end{pmatrix} + \Theta(x)$$

$$\frac{d}{dt} \begin{pmatrix} x_1 \\ \vdots \\ x_7 \\ x_8 \\ \vdots \\ x_{13} \end{pmatrix} = \begin{pmatrix} \dot{x}_1 \\ \vdots \\ \dot{x}_7 \\ M^{-1}(T(x, u) - \Theta(x)) \end{pmatrix}$$

Prospect

Refinement of Model wind aerodynamical forces