

AIR STANDARD CYCLES

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2006

ASSUMPTIONS IN AIR STANDARD CYCLES

- The gas in the engine cylinder is air (a perfect gas), i.e., it obeys the gas laws and has constant specific heat.
- The physical constants of the gas in the engine cylinder are same as those of air at moderate temperatures.
- All the compression and expansion processes are adiabatic and they take place without any internal friction.
- Heat is supplied by bringing a hot body in contact with the cylinder at appropriate points during the process. Similarly, heat is rejected by bringing a cold body in contact with the cylinder at these points.
- The cycle is considered to be closed one and the same air is used again and again to repeat the cycle.
- No chemical reaction takes place in the engine cylinder.

OTTO CYCLE

Thermal Efficiency of Internal Combustion Engine

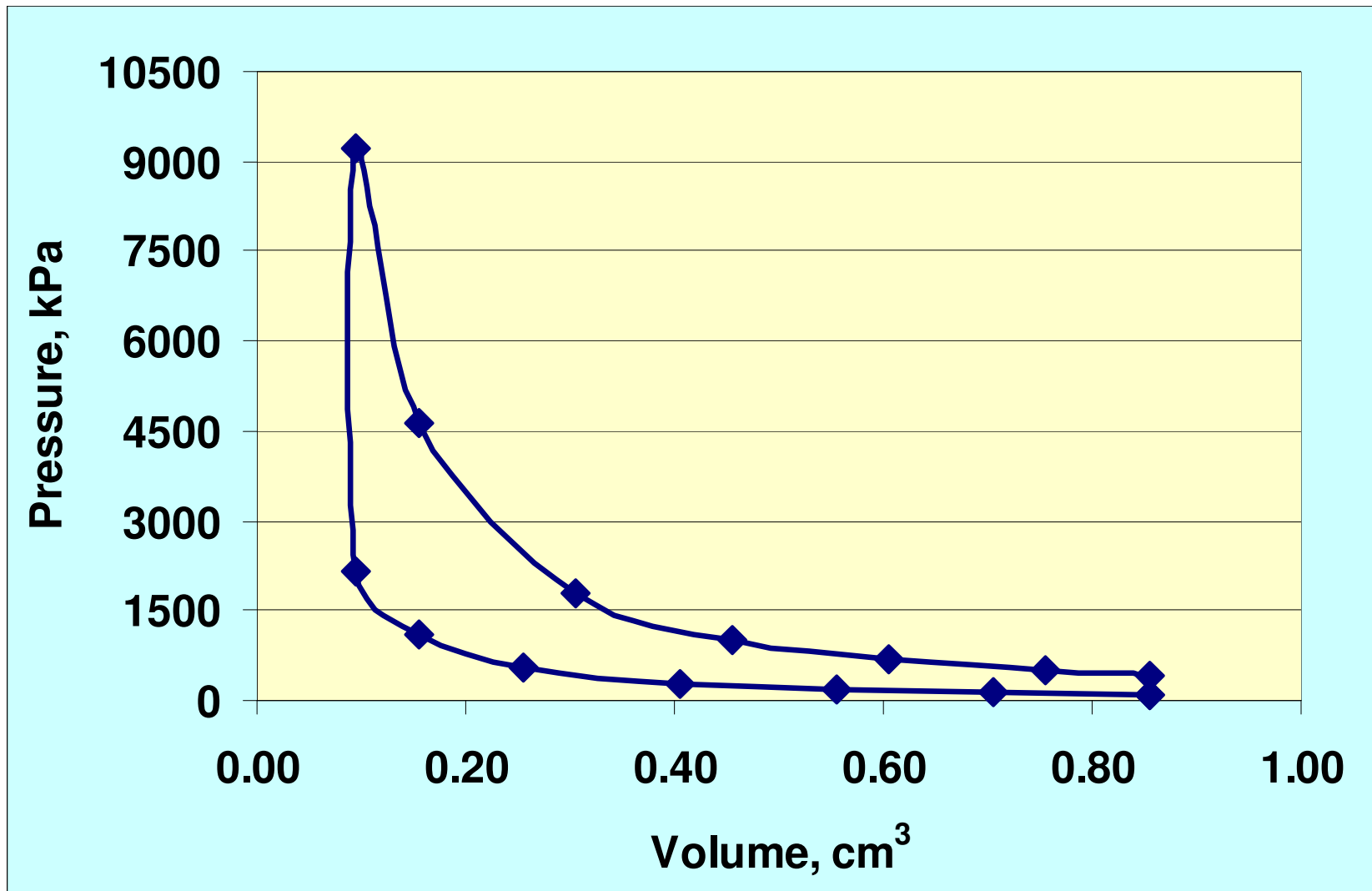
Thermal efficiency for Otto cycle:

$$\eta = 1 - \frac{1}{r_c^{k-1}}$$

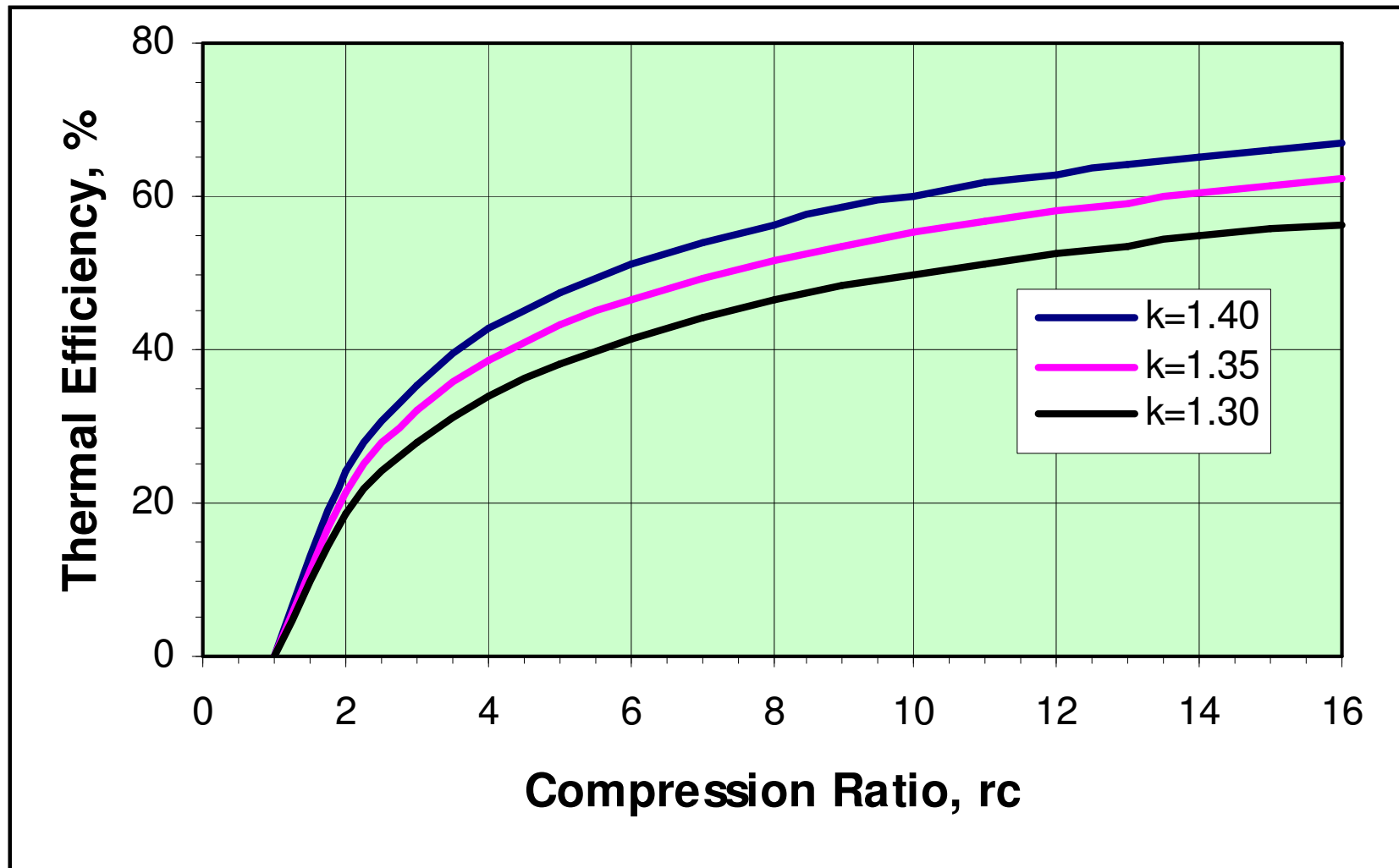
Where, compression ratio, $r_c = v_1/v_2$ and $k = c_p/c_v$

η_{OTTO} increases with the increase of r_c & k

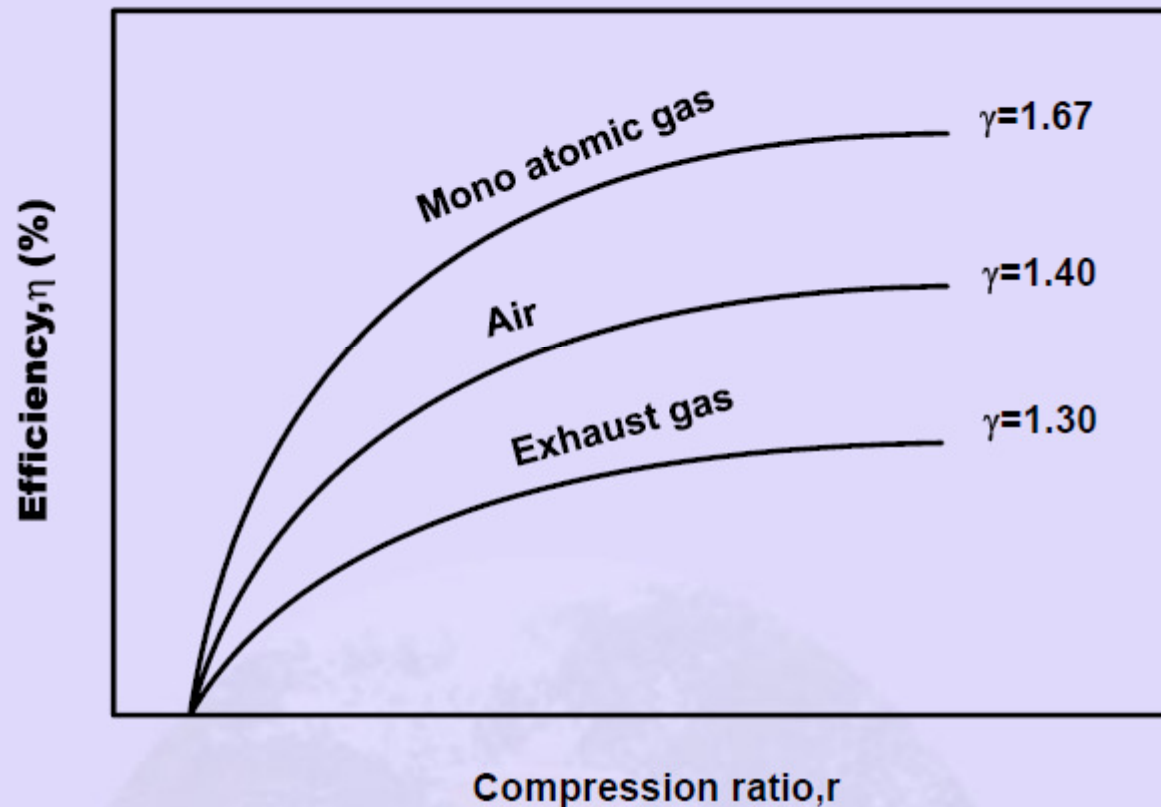
OTTO CYCLE FOR SI ENGINE



Thermal Efficiency of Otto Cycle



VARIATION OF THERMAL EFFICIENCY WITH COMPRESSION RATIO



Effect of CR and γ on efficiency for Otto cycle.

DIESEL CYCLE

Thermal Efficiency of Internal Combustion Engine

Thermal efficiency for Diesel cycle:

$$\eta = 1 - \frac{1}{r_c^{k-1}} \left[\frac{(\beta^k - 1)}{k(\beta - 1)} \right]$$

Where,

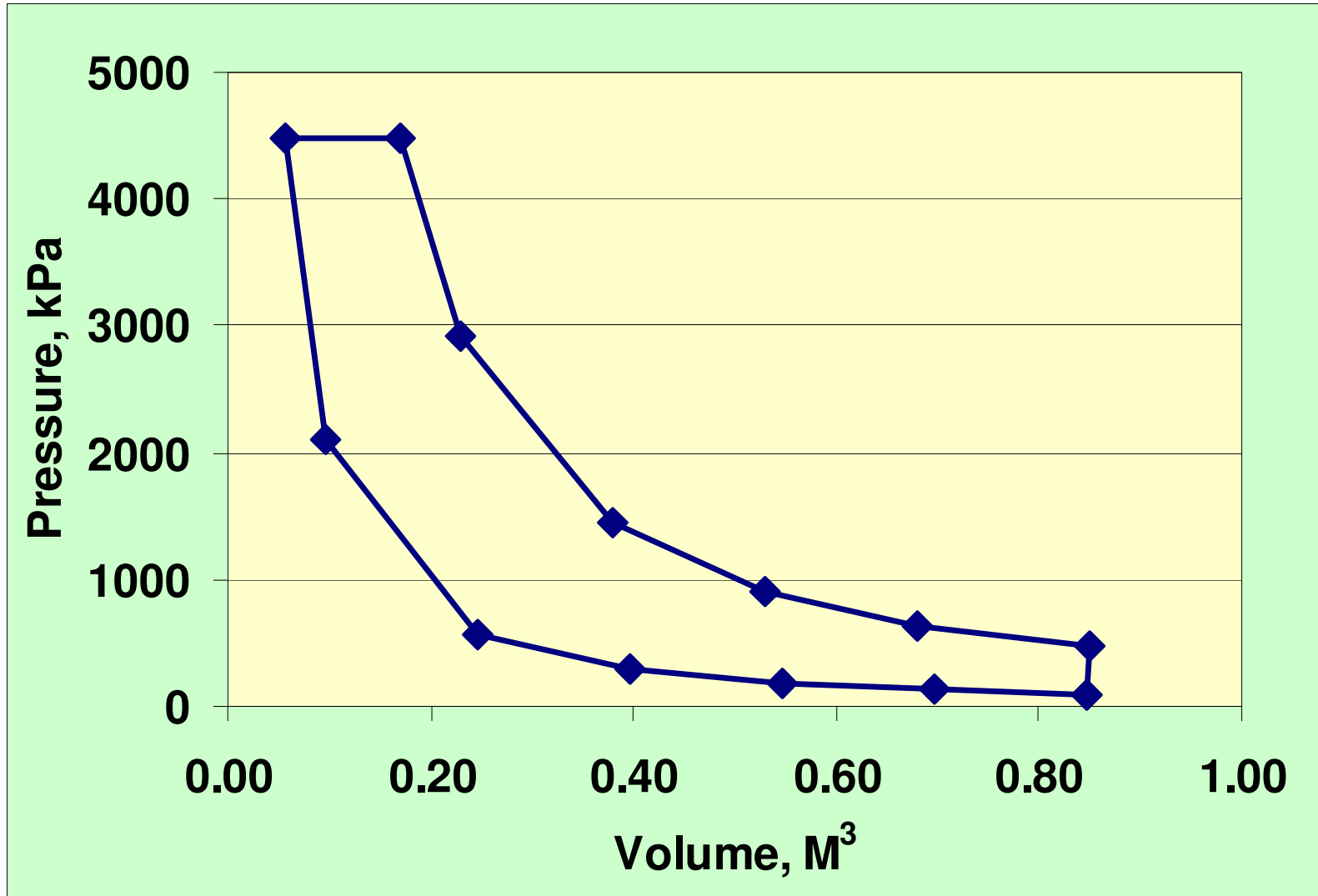
Compression ratio, $r_c = v_1/v_2$ and $k = c_p/c_v$

Cut off ratio, $\beta = v_3/v_2 = T_3/T_2$

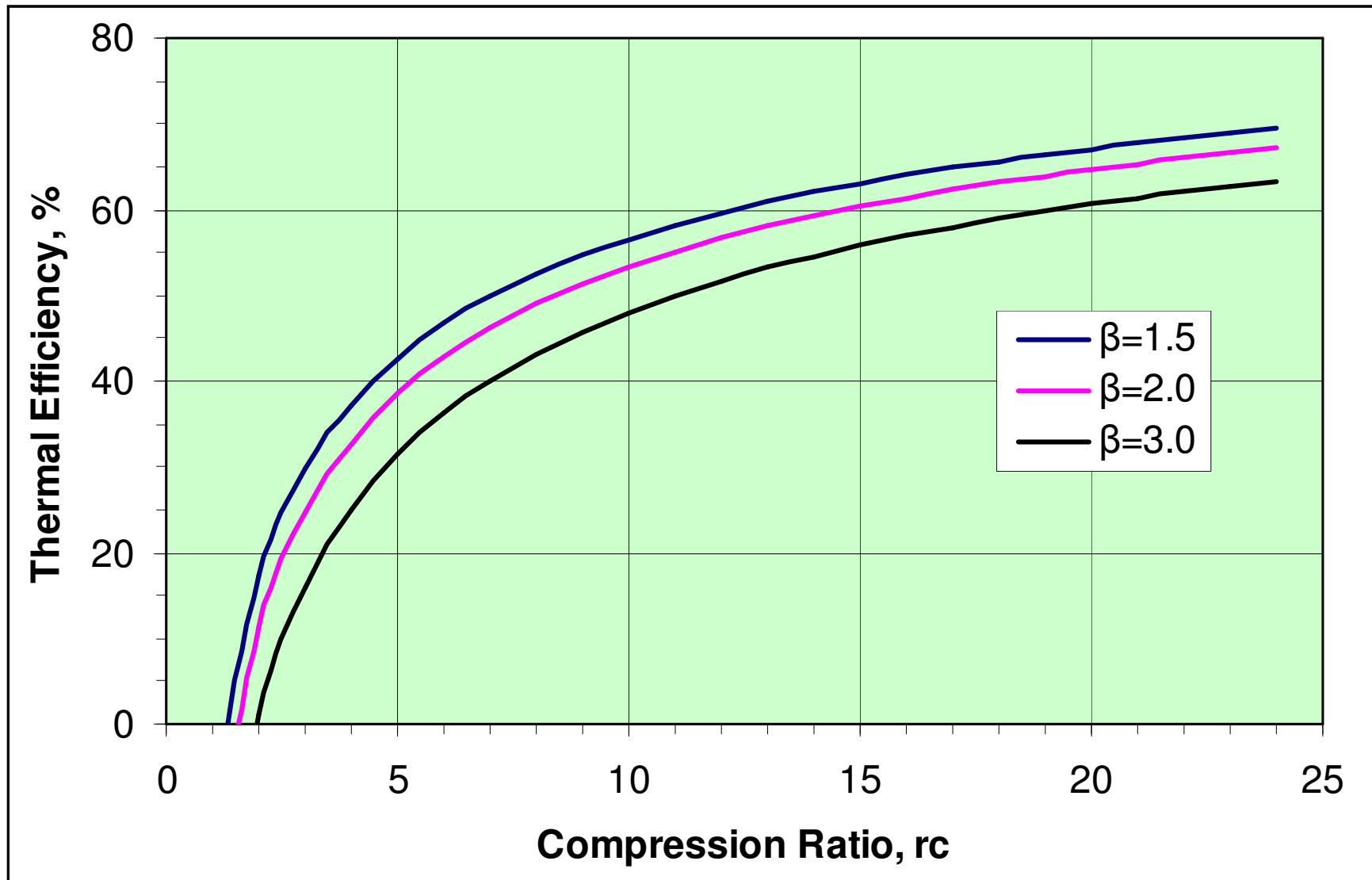
The term in the bracket is always > 1 . So the

$\eta_{\text{DIESEL}} < \eta_{\text{OTTO}}$ for the same r_c

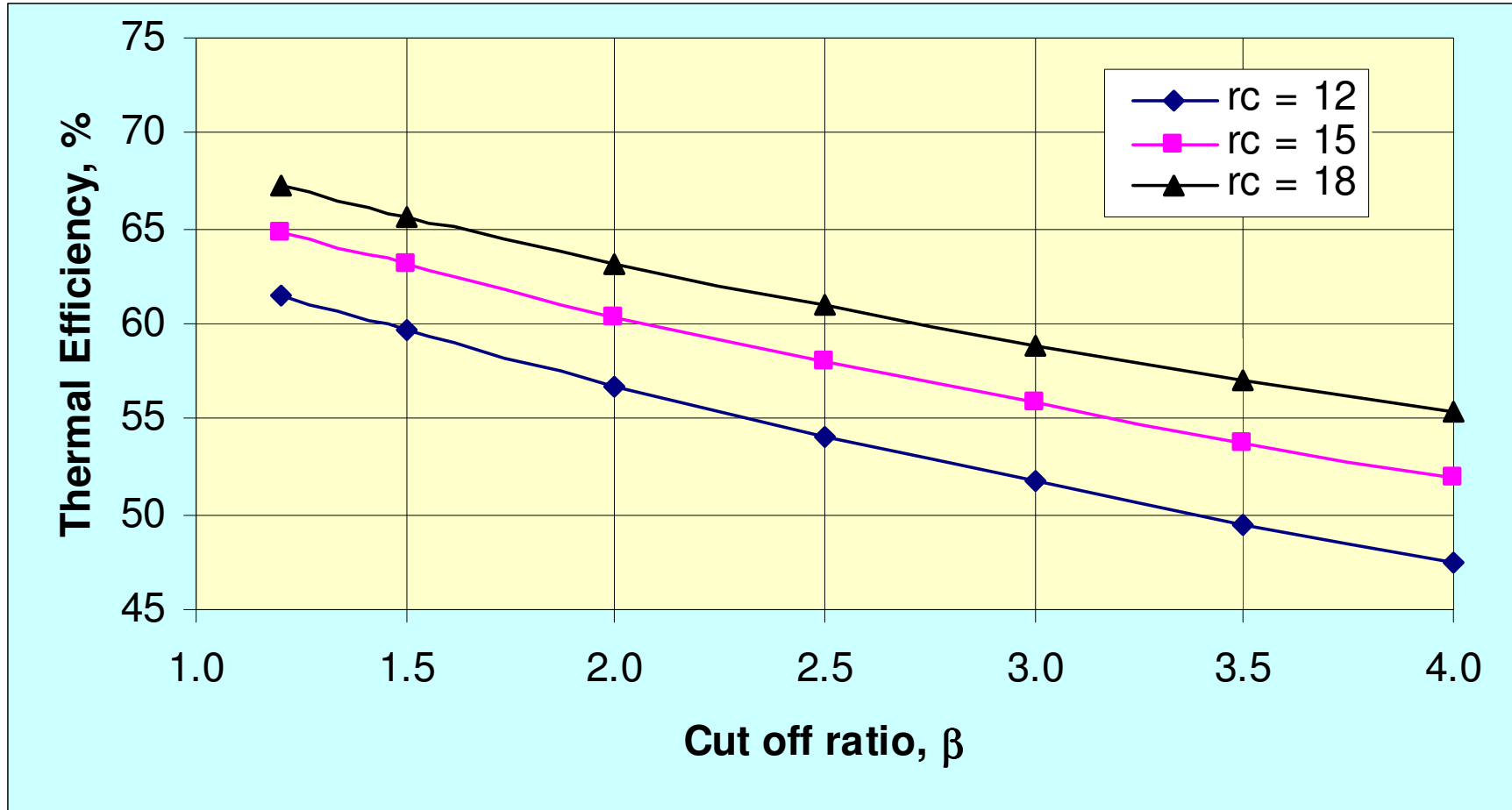
DIESEL CYCLE FOR CI ENGINE



Thermal Efficiency of Diesel Cycle



THERMAL EFFICIENCY VS CUT OFF RATIO FOR DIESEL CYCLE



DIESEL CYCLE

Advantages:

- Very mature and widespread technology, relatively low cost and high reliability, high efficiency (fuel economy of 200-250 g/kWh for modern engines), long life, relatively low thermal signature. Power density of the prime mover (only) is 100 to 400 W/kg over the 1-40 kW power output range. Initial cost is about \$100 to 250 per kW, and operating costs are low. Can meet EPA emission standards.

Disadvantages:

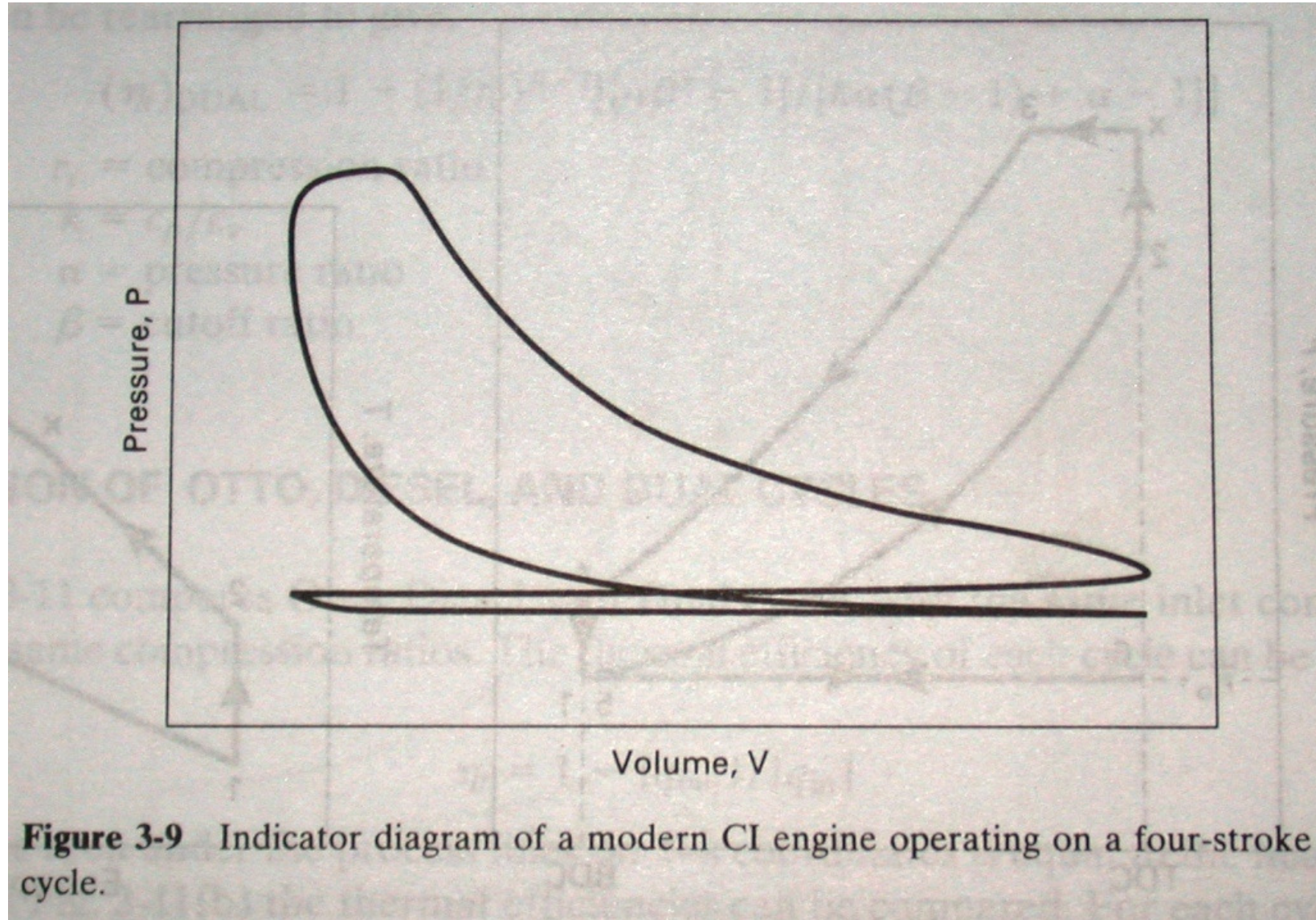
- Can produce high noise and vibration, which increases with rotational speed.

DUAL OR LIMITED PRESSURE CYCLE

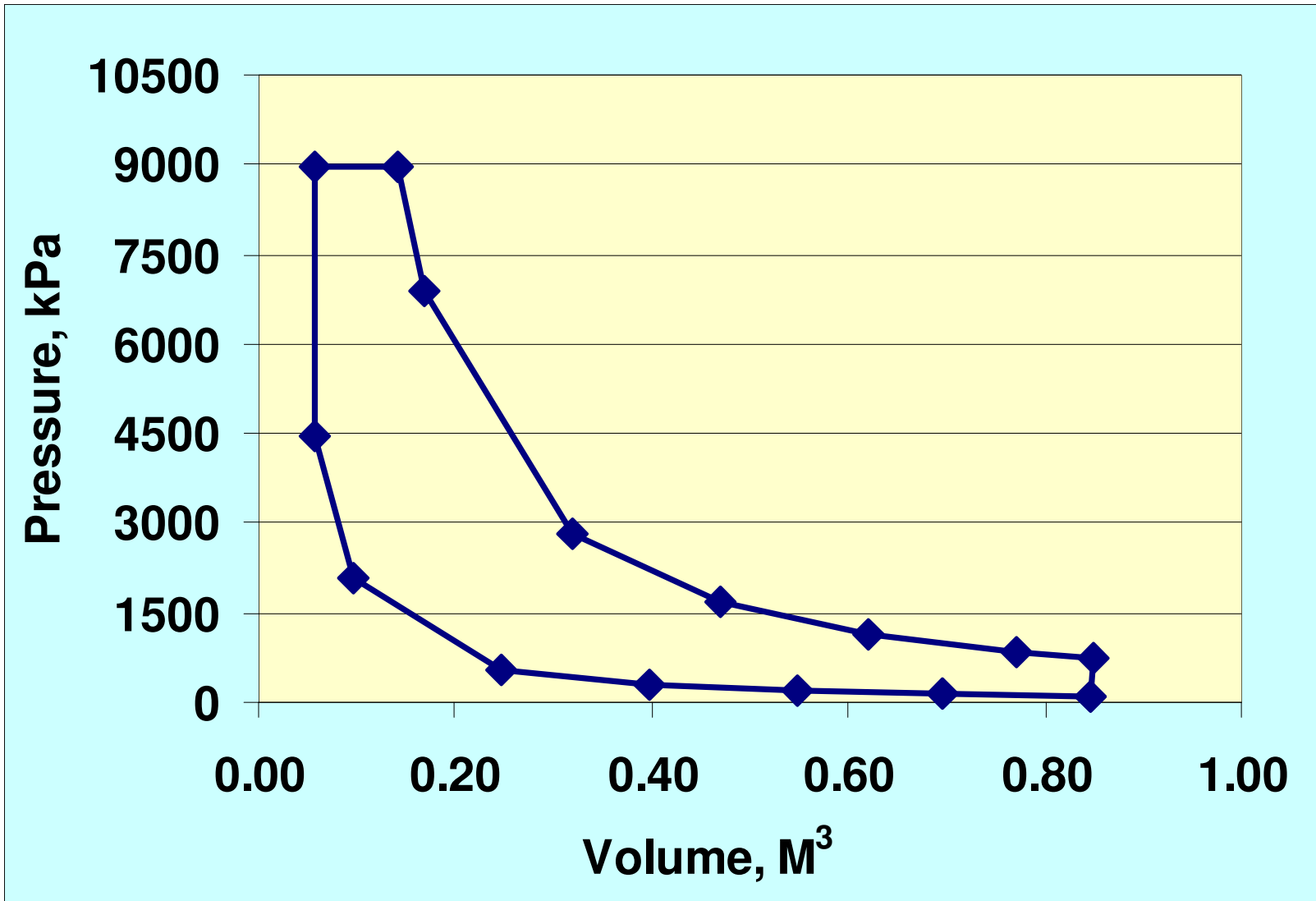
INTRODUCTION TO DUAL CYCLE

- Diesel cycle would operate on the more efficient higher compression ratios, while constant volume combustion of the Otto cycle would give higher efficiency for a given compression ratio.
- The modern CI engine operates to meet this by injecting the fuel much earlier in the cycle, somewhere around 20° bTDC. The first fuel then ignites late in the compression stroke, and some of the combustion occurs almost at constant volume at TDC, much like the Otto cycle.
- The last of the fuel is still being injected at TDC, and combustion of this fuel keeps the pressure high into the expansion stroke.

ACTUAL CYCLE IN CI ENGINE



DUAL CYCLE



Thermal Efficiency of Internal Combustion Engine

Thermal efficiency for Dual cycle:

$$\eta = 1 - \frac{1}{r_c^{k-1}} \left[\frac{(\alpha \beta^k - 1)}{(\alpha - 1) + \alpha k (\beta - 1)} \right]$$

Where,

$r_c = v_1/v_2$ and $k = c_p/c_v$

$\beta = v_3/v_2 = T_3/T_2$ and pressure ratio, $\alpha = P_3/P_2$

Since $\alpha > 1$, the η_{DUAL} is $< \eta_{\text{DIESEL}}$ for the same β and r_c

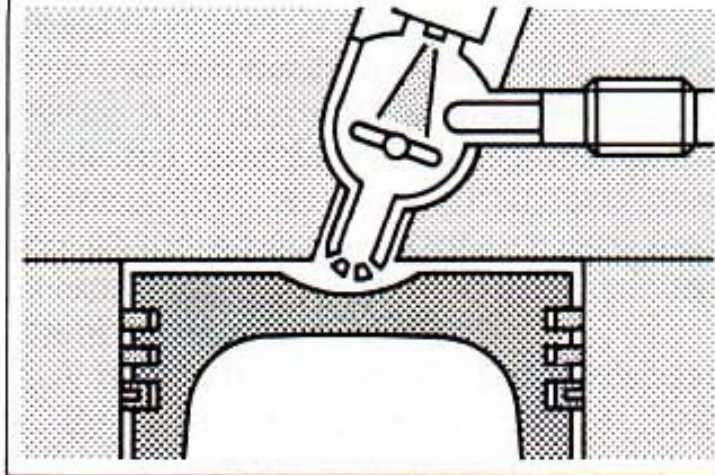
$\alpha = 1$ for Diesel Cycle and $\beta = 1$ for Otto cycle

DUAL/LIMITED PRESSURE CYCLE

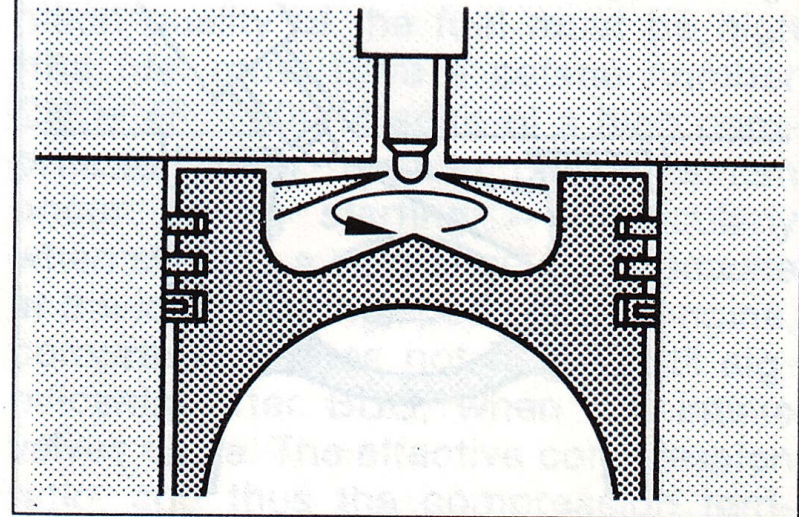
- The theoretical cycle of modern high speed diesel engine is more adequately by dual combustion or limited pressure cycle.
- In this cycle part of the heat addition takes place at constant volume and the remainder at constant pressure.
- Heat addition at constant volume tends to increase the efficiency of the cycle whereas switching over to constant pressure heat addition limits the maximum cylinder pressure.

COMBUSTION CHAMBER FOR DIESEL ENGINE

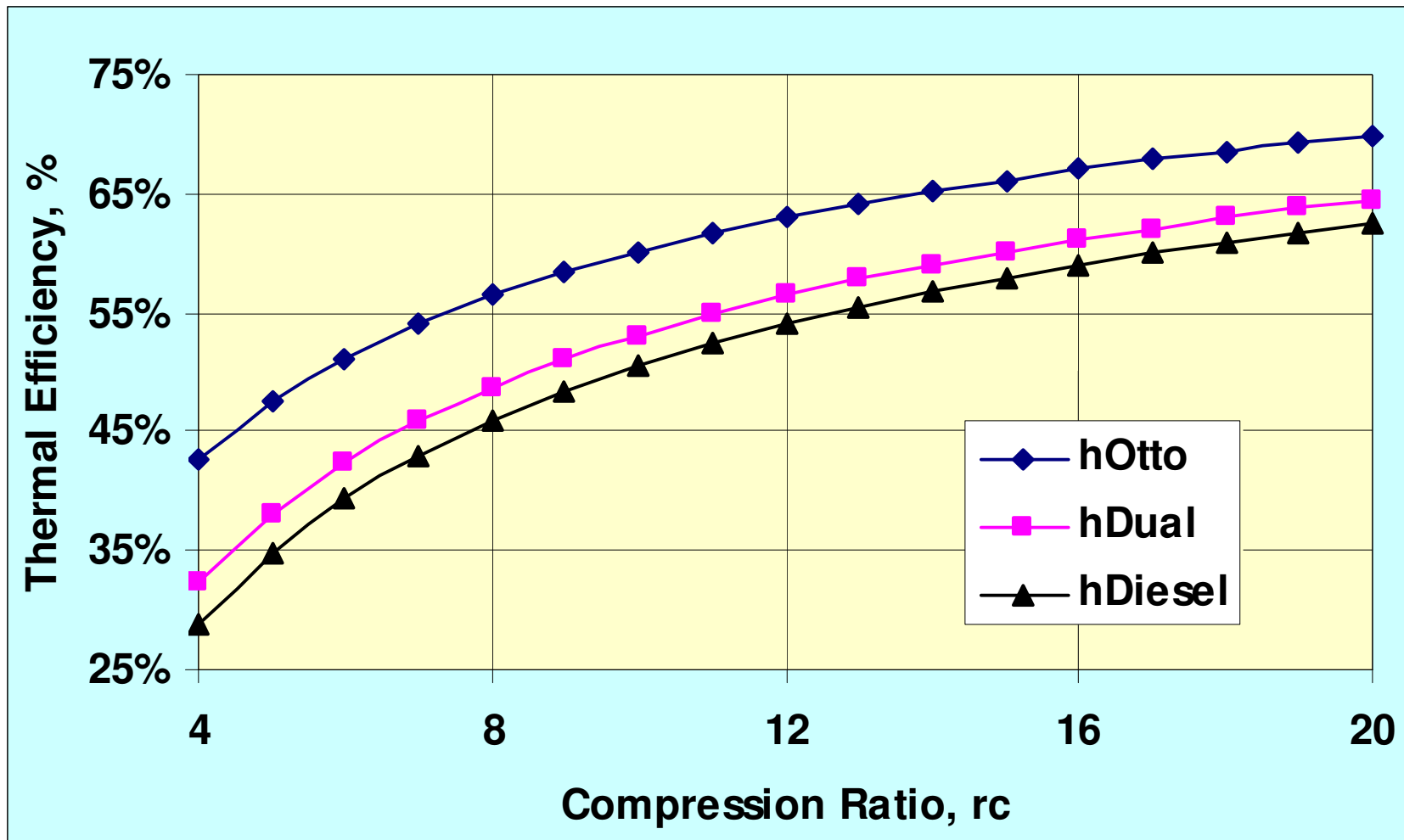
Combustion chamber shape and nozzle location for the prechamber system.



Combustion chamber shape and nozzle location for the multiple-hole process with air swirling.



THERMAL EFFICIENCY VS COMPRESSION RATIO FOR DIFFERENT CYCLES

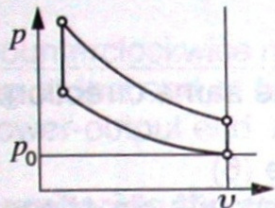
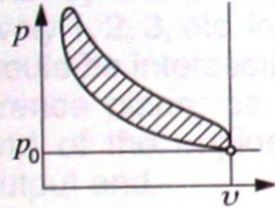
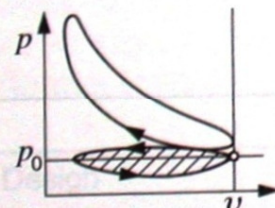


COMPARISON OF CYCLES

- For the same compression ratio and same heat input
 - $\eta_{\text{OTTO}} > \eta_{\text{DUAL}} > \eta_{\text{DIESEL}}$
- For the peak pressure and same heat input
 - $\eta_{\text{OTTO}} < \eta_{\text{DUAL}} < \eta_{\text{DIESEL}}$
- For the peak pressure and maximum temperature
 - $\eta_{\text{OTTO}} < \eta_{\text{DUAL}} < \eta_{\text{DIESEL}}$

OVERALL EFFICIENCY OF SI ENGINE

Table 2. Graphic representations and definitions of the individual and overall efficiencies of the reciprocating-piston engine.

Pressure vs. volume diagram	Designation	Conditions	Definition	Efficiencies
	Theoretical reference constant-volume cycle	Ideal gas, constant specific heat, infinitely rapid heat addition and -dissipation, etc.	$\eta_{th} = 1 - \varepsilon^{1-\kappa}$ Theoretical or thermal efficiency	η_{th}
	Real high pressure working cycle	Wall heat losses, real gas, finitely rapid heat addition and dissipation, variable specific heat	η_{gHD} Efficiency factor of the high-pressure cycle	η_i
	Real charge cycle (4-stroke)	Flow losses, heating of the mixture or the air, etc.	η_{gLW} Charge exchange efficiency	η_g
Mechanical losses	Losses due to friction, cooling, auxiliary units	Real engine	η_m	η_m
				η_e

DEFINITION OF EFFICIENCIES

- η_e – overall efficiency includes the sum of all losses, and can thus be defined as the ratio of effective mechanical work to the mechanical work equivalent of the supplied fuel.
- η_i – indicated efficiency is the ratio of indicated high-pressure work to the calorific content of the supplied fuel.
- η_g – efficiency of cycle factor includes all internal losses occurring in both high-pressure and low-pressure processes.
- η_m – mechanical efficiency defines the relationship between the effective work available at the crankshaft and the indicated work.
- η_{th} – thermal efficiency is the efficiency of the ideal constant volume combustion cycle.

Mean effective pressure =

= work done during the cycle / Stroke volume

$$mep = \frac{Q_s - Q_r}{V_d} = \frac{workdone}{V_d}$$

Where,

Q_s – energy supplied by burning of fuel

Q_r – heat loss through exhaust

TYPICAL VALUES FOR BMEP

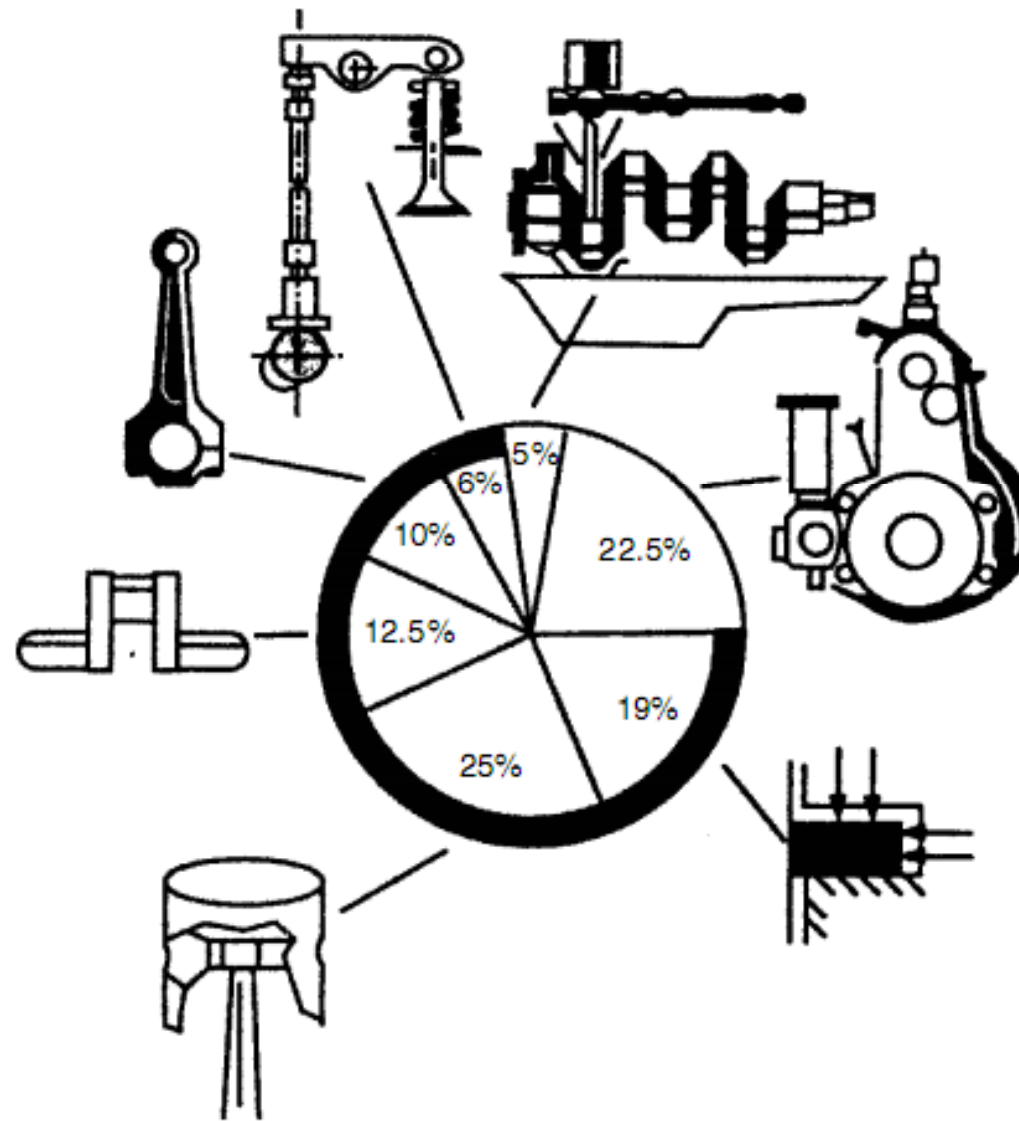
- Naturally aspirated spark-ignition engines, $b_{mep} = 850 - 1050$ kPa at the engine speed where max. torque is obtained and at the rated power, b_{mep} values are 10 to 15 percent lower.
- Turbocharged automotive SI engines the max $b_{mep} = 1250 - 1700$ kPa and at max. rated power b_{mep} is $900 - 1400$ kPa.
- Naturally aspirated 4-stroke diesels the max $b_{mep} = 700 - 900$ kPa and at the max. rated power = 700 kPa.
- Turbocharged 4-stroke diesel, the max $b_{mep} = 1000 - 1200$ kPa and at max. rated power b_{mep} is $850 - 950$ kPa.

ACTUAL CYCLES

Actual cycles differ from ideal cycles because of the following factors:

- Variation of air-fuel ratio
- Valve operation and timing
- Finite combustion time
- The actual combustion process
- Dissociation
- Exhaust blowdown
- Heat transfer
- Fluid friction

Distribution of friction losses in an internal combustion engine



ACTUAL CYCLE IN SI ENGINE

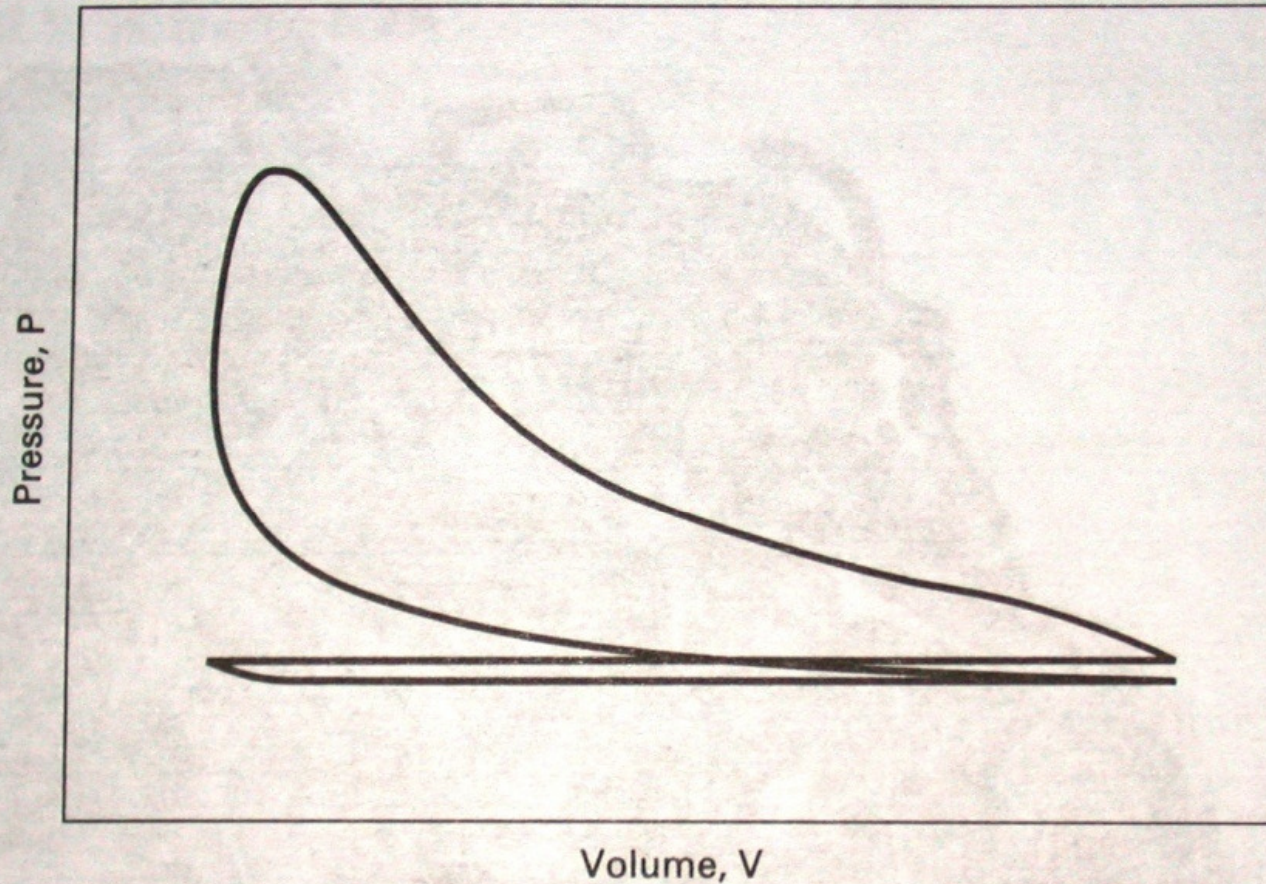
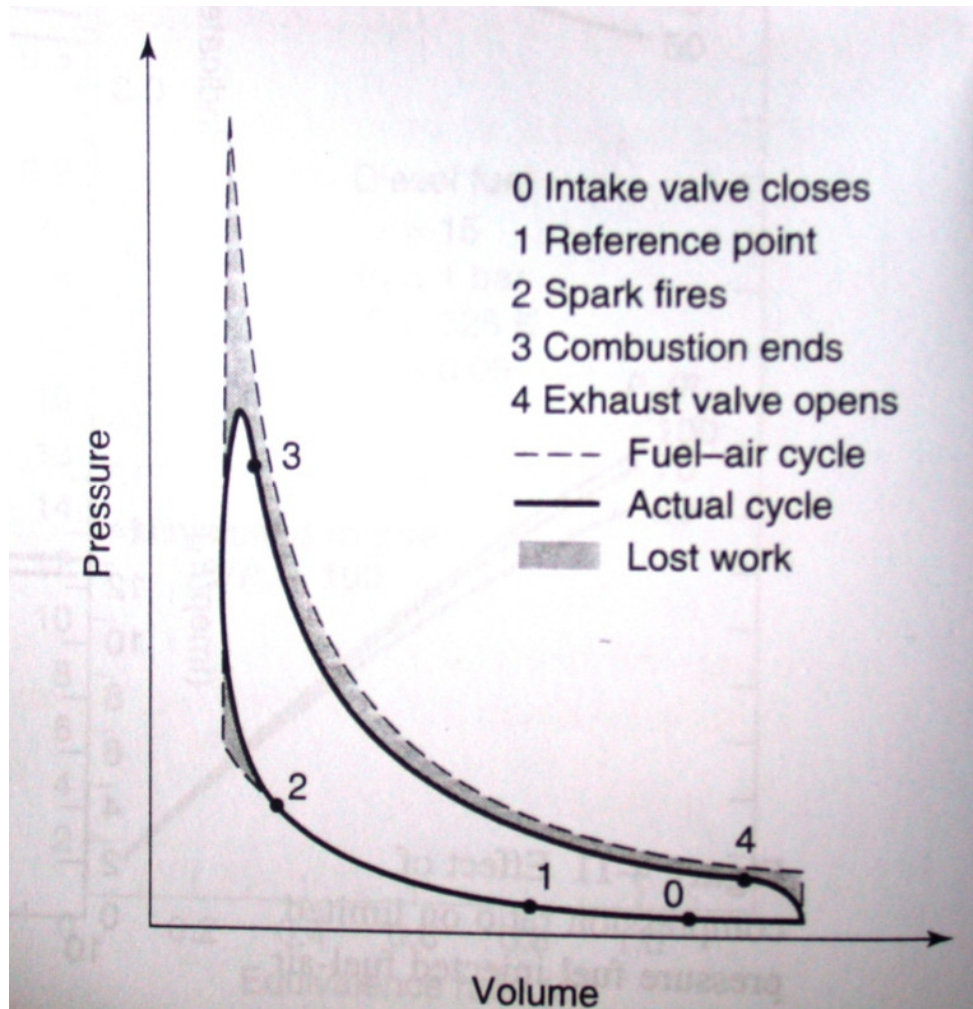


Figure 2-6 Indicator diagram for a typical four stroke cycle SI engine. An indicator diagram plots cylinder pressure as a function of combustion chamber volume over a 720° cycle. The diagram is generated on an oscilloscope using a pressure transducer mounted in the combustion chamber and a position sensor mounted on the piston or crankshaft.

LOST WORK

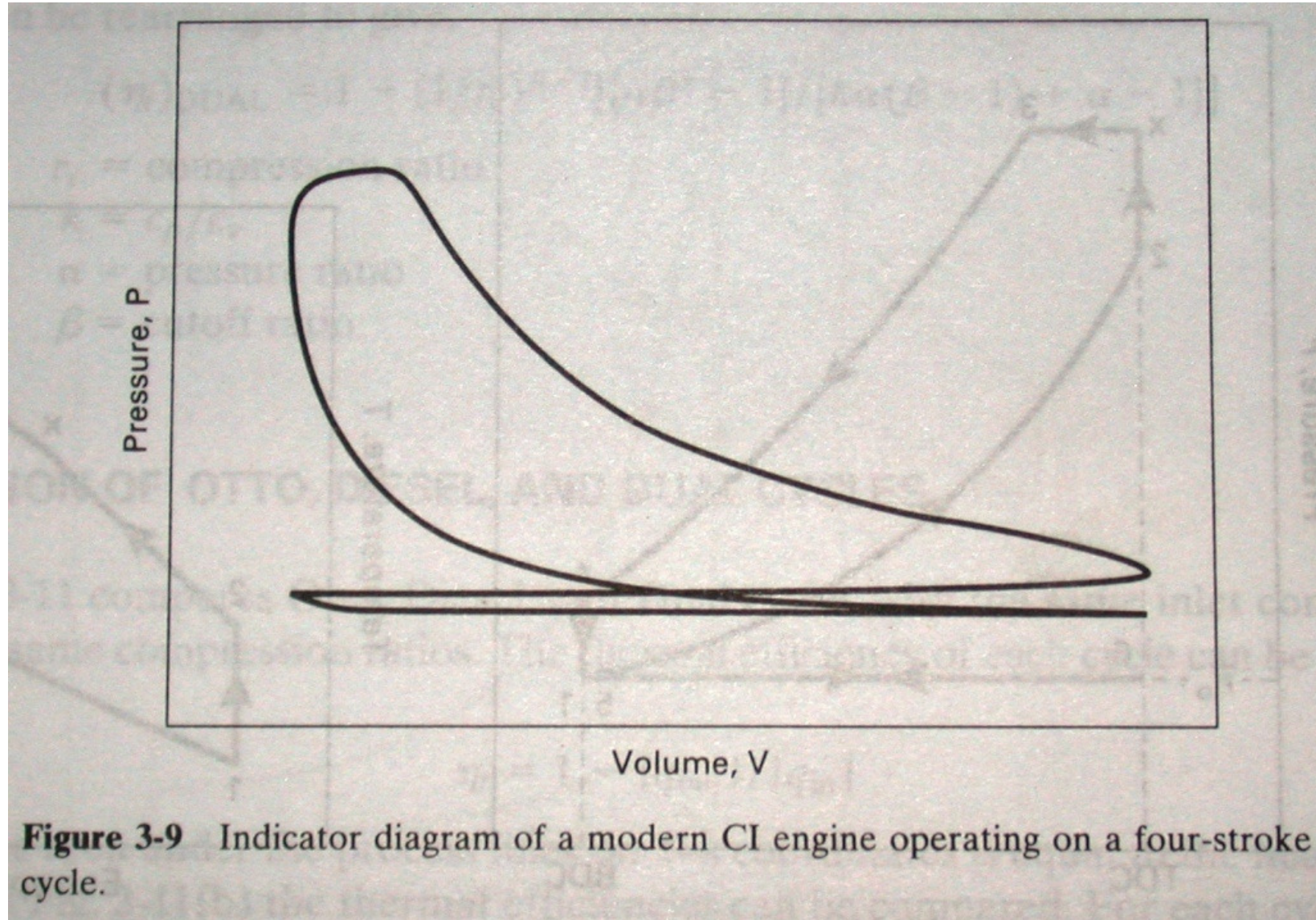


- Heat loss
- Mass loss
- Finite burn rate
- Finite blowdown rate

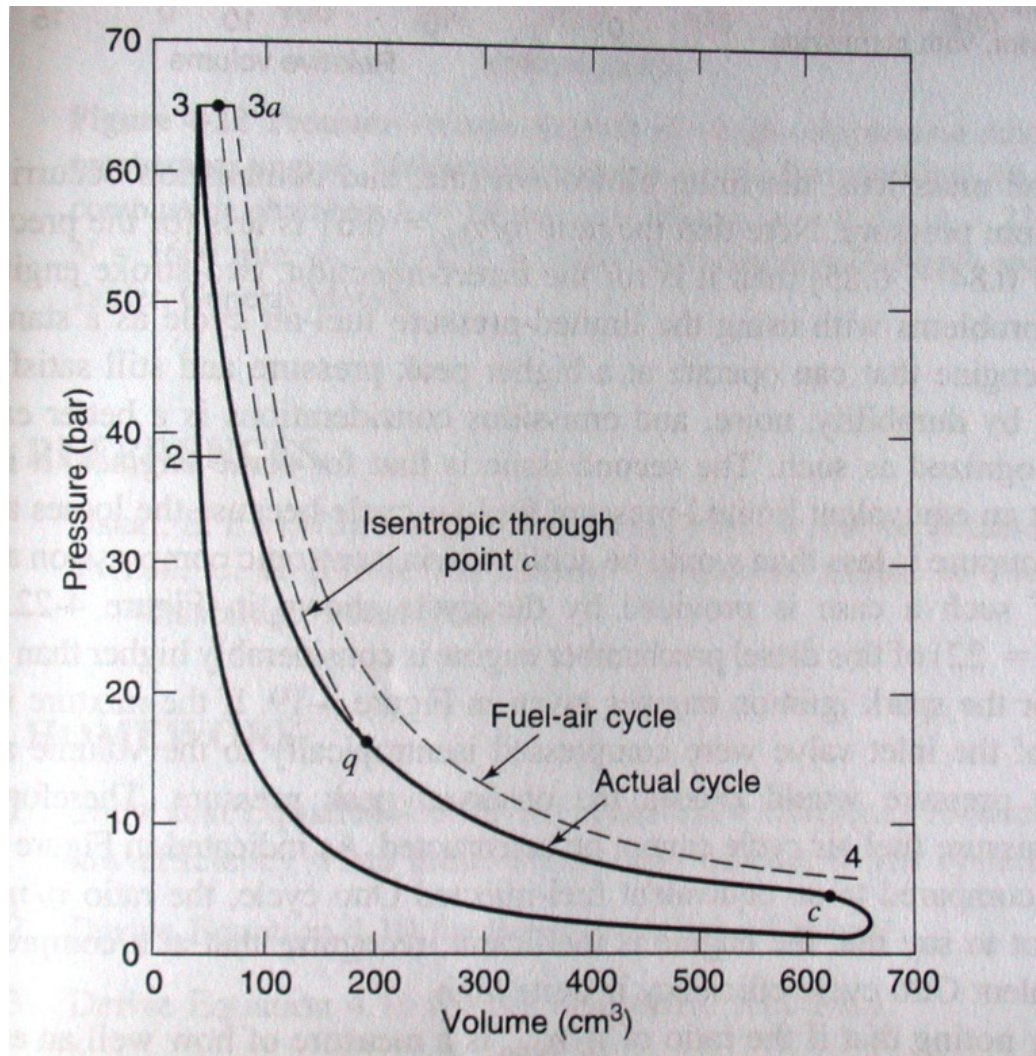
These losses result in the actual efficiency being less than that of equivalent fuel-air cycle by a factor ranging from 0.8 to 0.9 (η/η_{Otto}).

The imep is maximized slightly rich of stoichiometric, and increases with increasing compression ratio and inlet pressure.

ACTUAL CYCLE IN CI ENGINE



LOST WORK



- Heat loss
- Mass loss
- Finite blowdown rate
- Combustion occurring at less than the maximum pressure

The figure shows the cylinder pressures in a diesel pre-chamber engine compared with equivalent limited pressure fuel-air cycle.

MAIN DIFFERENCE BETWEEN ACTUAL AND THEORETICAL CYCLES

- Compression and expansion are not friction less adiabatic processes. A certain amount of friction is always present and there is considerable heat transfer between the gases and cylinder wall.
- Combustion does not occur either at constant volume or at constant pressure.
- The thermodynamics properties of the gases after combustion are different than those of the fuel-air mixture before combustion.
- The combustion may be incomplete.
- The specific heats of the working fluid are not constant but increases with temperature.

MAIN DIFFERENCE BETWEEN ACTUAL AND THEORETICAL CYCLES

- The cylinder pressure during exhaust process is higher than the atmosphere. As a result, more work has to be done by the piston on the gases to expel them out of the cylinder, than work done by the gases on the piston during the intake stroke.
- This difference in work, called pumping work, is represented by the pumping loop shown by hatched area.
- Note that this work is negative and represents loss of work called pumping loss.

CONCLUSIONS

- The thermal efficiency of actual Otto cycle in SI engine is about 25-30%.
- Similarly the thermal efficiency of actual Diesel cycle in CI engine is 30-35%
- Desirable increase in thermal efficiency of IC engine very much depends on material, air-fuel ratio, ignition timing, fuel injection timing, residual gases in the cylinder etc.

REVIEW PROBLEMS

1. An air standard Otto cycle has a compression ratio of 8 and temperature and pressure at the beginning of compression are 20°C and 1 bar respectively. The constant volume heat addition is 1800 kJ/kg . Calculate the maximum temperature and pressure for the cycle and the temperature at the end of the expansion process. What is the efficiency and mean effective pressure of the cycle. Take $c_v = 0.718 \text{ kJ/kg.K}$ and $k = 1.4$
2. The initial condition of air in an Otto cycle is 288K and 1.05 bar. The air compressed isentropically to a pressure 13 bar. Then heat is added at constant volume till the pressure becomes 35 bar. Calculate:
 - a) Compression ratio
 - b) Air standard efficiency
 - c) Mean effective pressure of the cycle

REVIEW PROBLEMS

1. In an air-standard Diesel cycle, the compression ratio is 16, the cylinder bore is 200 mm and the stroke is 300 mm. Compression begins at 1 bar and 27°C. The cut-off takes place at 8 percent of the stroke. Determine:
 - a) The pressure, the volume and the temperature at all salient points [$V_1=0.010048\text{m}^3$, $V_2=0.000628\text{m}^3$, $m=0.01167\text{ kg}$, $p_2=48.5\text{ bar}$, $T_2=909.4\text{K}$]
 - b) The cut-off ratio [2.2]
 - c) The air standard efficiency [60.4%]
 - d) The mean effective pressure [8.21 bar]

Review question on diesel cycle

- Given:

- a) Cylinder bore = 200 mm and stroke = 250 mm
- b) $R_c = 18$ and $\beta = 2$
- c) $P_1 = 0.1 \text{ MPa}$ and $T_1 = 300\text{K}$

- Find:

- a) T and P in state 2,3 & 4
- b) Thermal efficiency (η), work done and mep

REVIEW PROBLEMS

1. An air-standard Dual cycle has a compression ratio of 15, and the compression begins at 1 bar, 27°C. The maximum pressure is limited to 60 bar. The heat transferred to air at constant volume is twice that at constant pressure. Calculate:
 - a) the pressures and temperatures at the cardinal points of the cycle, [$T_2=886\text{K}$, $p_2=44.3\text{ bar}$, $T_3=1200\text{K}$, $T_4=1312\text{K}$, $\beta=1.093$, $T_5=460\text{K}$, $W=223\text{ kJ/kg}$]
 - b) the cycle efficiency, [66%] and
 - c) the mean effective pressure of the cycle. [2.77 bar]



Engine Thermodynamic Analysis

Ideal Otto Cycle

Glenn
Research
Center

C_v = Specific Heat constant volume

γ = Specific Heat Ratio

p = pressure

T = Temperature

V = Volume

f = fuel / air ratio

Q = Fuel heating value

cps = cycles per second

P = Power

$V_2/V_3 = r$ = Compression Ratio

Compression Stroke:

$$T_3/T_2 = r^{\gamma-1}$$

$$p_3/p_2 = r^{\gamma}$$

Combustion:

$$T_4 = T_3 + fQ/c_v$$

$$p_4 = p_3(T_4/T_3)$$

Power Stroke:

$$T_5/T_4 = r^{1-\gamma}$$

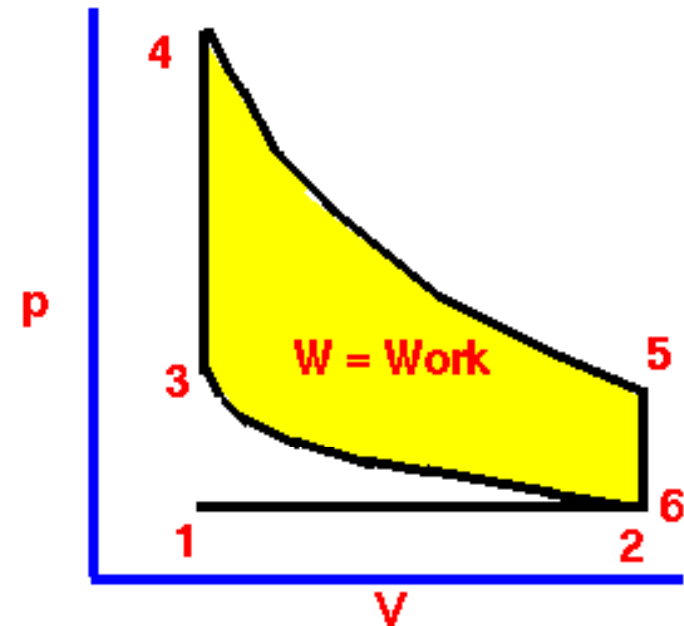
$$p_5/p_4 = r^{-\gamma}$$

Work per cycle:

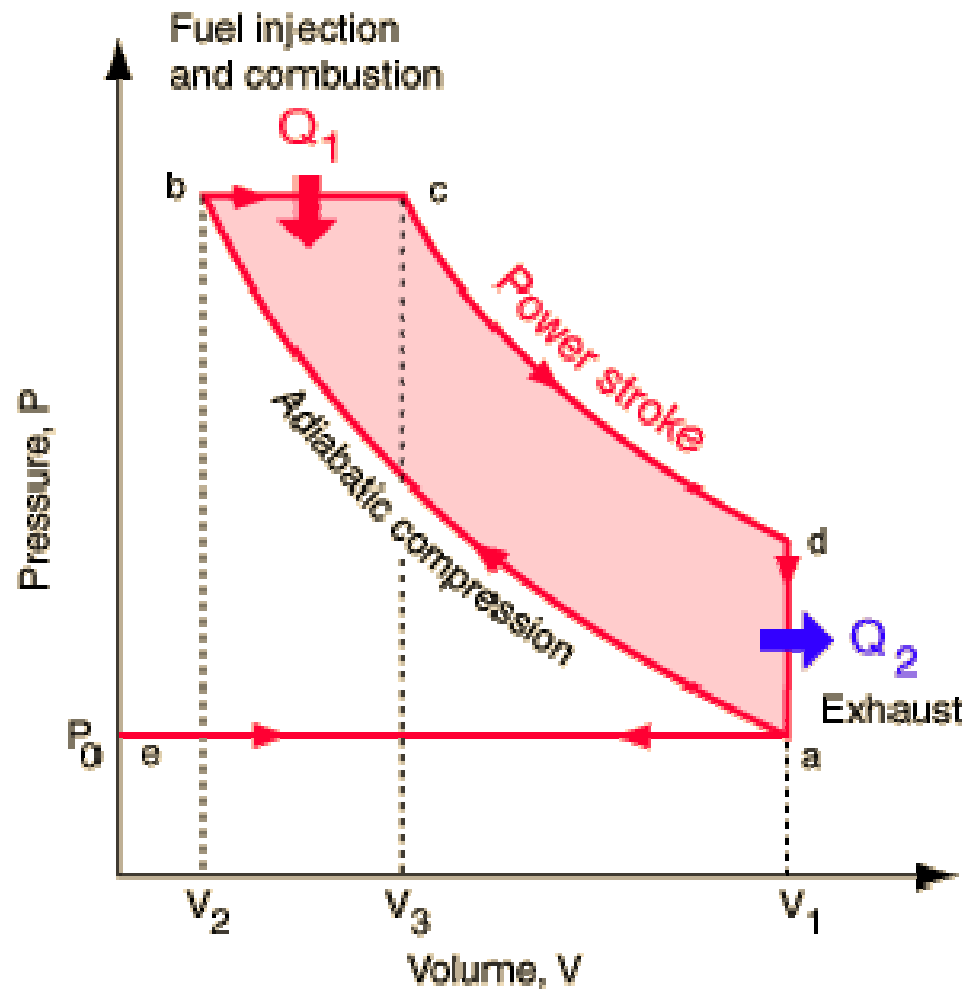
Engine Power:

$$W = c_v[(T_4 - T_3) - (T_5 - T_2)]$$

$$P = W \text{ cps}$$

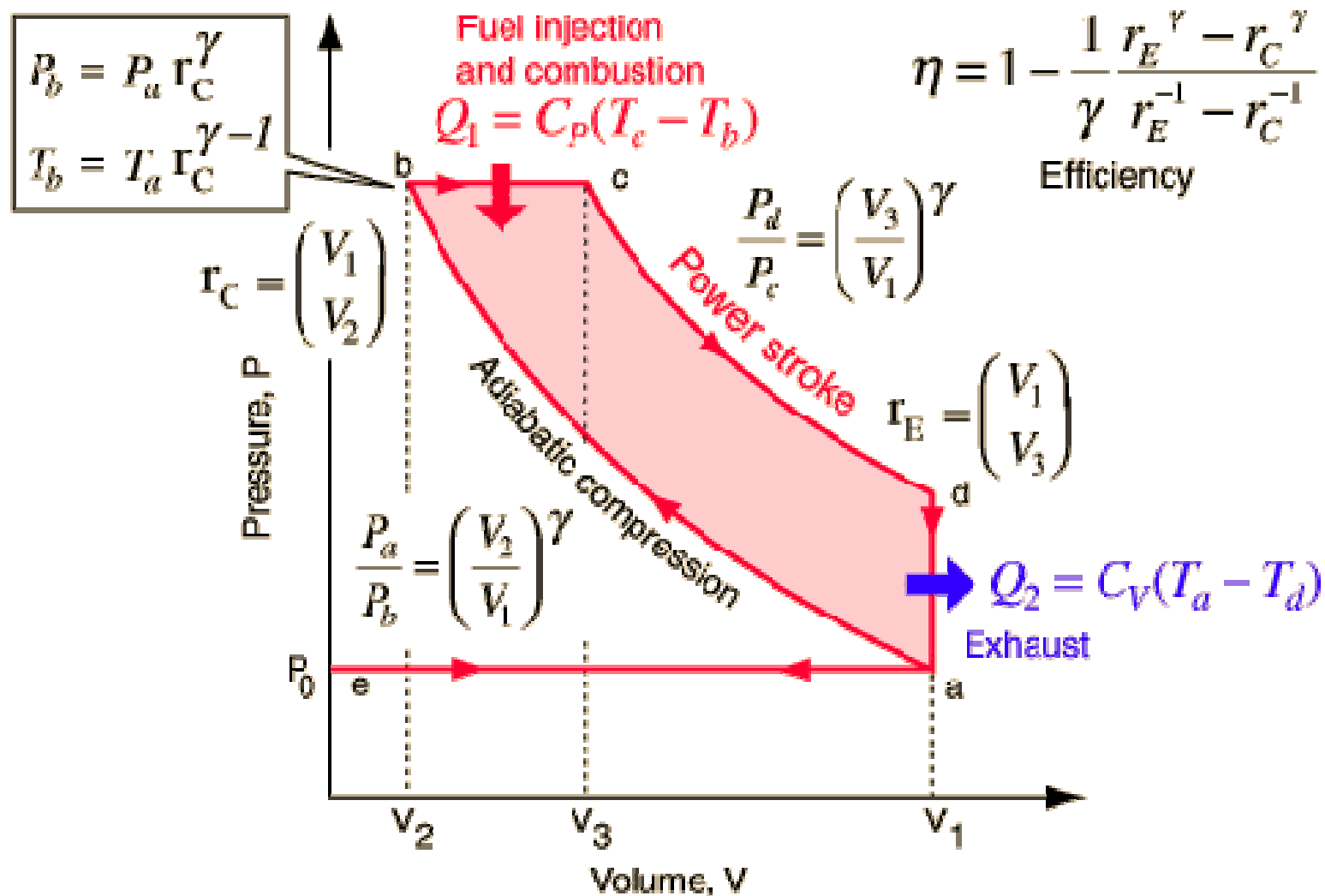


Diesel cycle



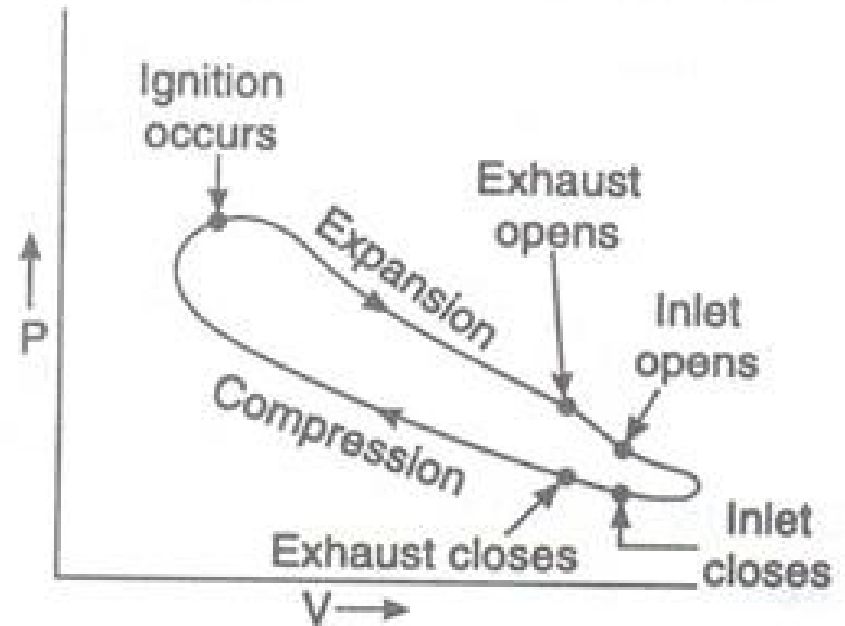
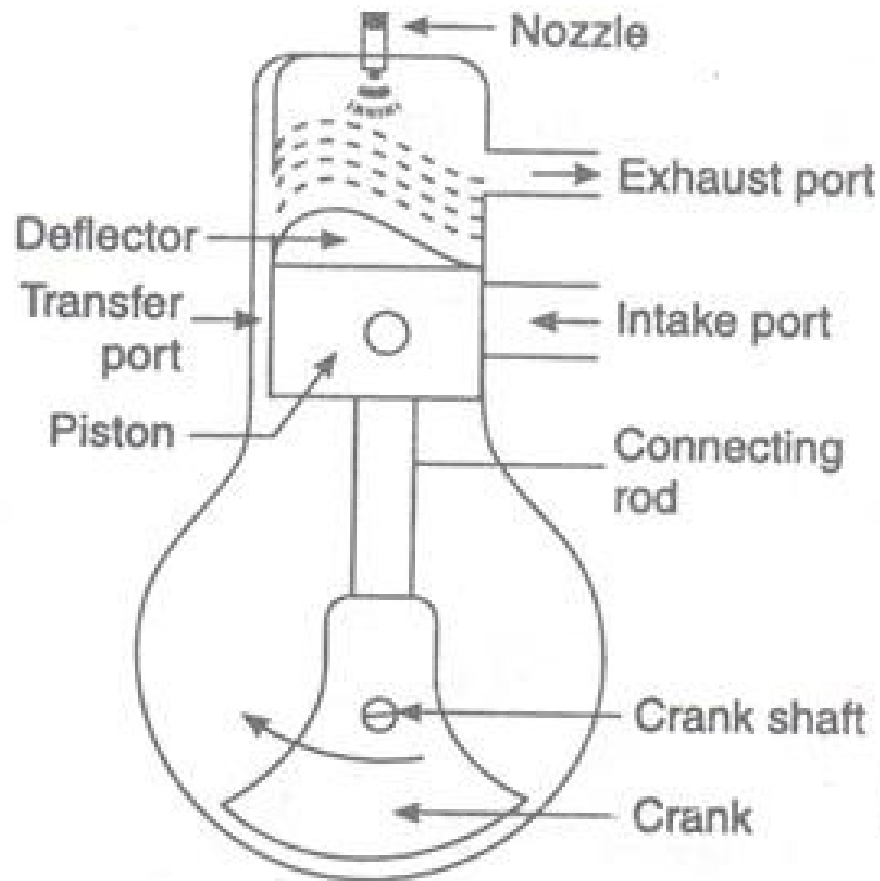
<http://hyperphysics.phy-astr.gsu.edu/hbase/thermo/diesel.html>

DIESEL ENGINE CYCLE



<http://hyperphysics.phy-astr.gsu.edu/hbase/thermo/diesel.html>

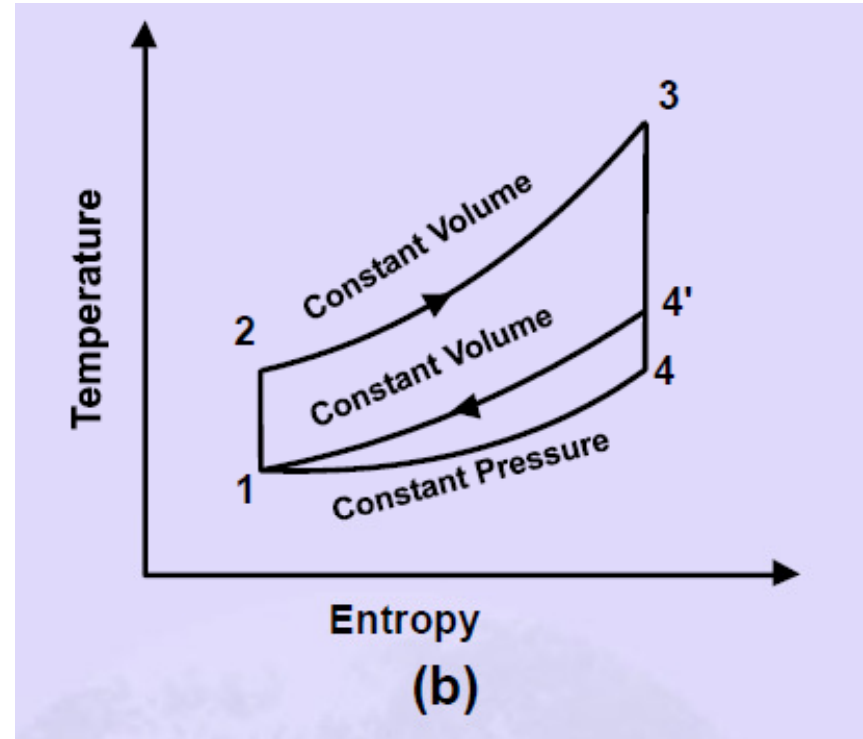
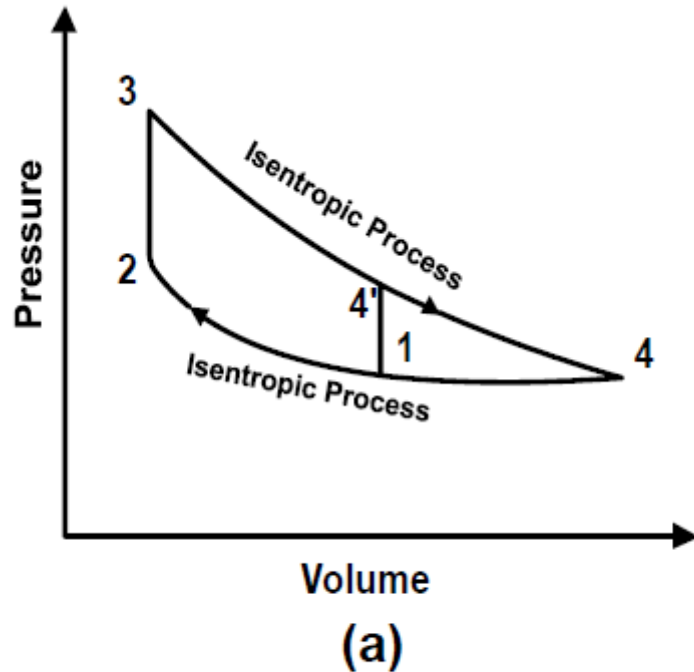
2 stroke working cycle



Indicator diagram for 2-stroke C.I. engine

ATKINSON CYCLE

Atkinson cycle on p-v and T-s diagrams



Process 1-2: Reversible adiabatic compression (v_1 to v_2).

Process 2-3: Constant volume heat addition.

Process 3-4: Reversible adiabatic expansion (v_3 to v_4).

Process 4-1: Constant pressure heat rejection.

$$\eta_{th} = 1 - \frac{1}{r^{\gamma-1}} \frac{\gamma(r_v - 1)}{(r_p - 1)}$$

$$r_v = \frac{T_4}{T_3} = \frac{V_4}{V_1} \quad \& \quad r_p = \frac{T_3}{T_2} = \frac{P_3}{P_2}$$

Thermal efficiency of Atkinson cycle

The [Atkinson cycle](#) is a variation on the Otto cycle which effectively increased the engine's expansion ratio compared with the compression ratio by using a complex crankshaft linkage. This enables the exhaust stroke to be longer than the induction stroke and hence the swept volumes are different. The greater expansion allows more energy to be extracted from the fuel charge and allows the engine to run cooler. It provides better efficiency at the expense of power density.

$$\text{Heat supplied} = C_v (T_3 - T_2)$$

$$\text{Heat rejected} = C_p (T_4 - T_1)$$

$$\text{Net workdone} = C_v (T_3 - T_2) - C_p (T_4 - T_1)$$

$$\eta_{th} = \frac{C_v (T_3 - T_2) - C_p (T_4 - T_1)}{C_v (T_3 - T_2)}$$

$$= 1 - \frac{\gamma (T_4 - T_1)}{(T_3 - T_2)}$$

$$\text{Let, } r = \frac{v_1}{v_2} = CR$$

$$T_2 = T_1 r^{\gamma - 1}$$

$$\frac{T_3}{T_2} = \frac{P_3}{P_2} = r_p = \text{Pressure ratio}$$

$$T_3 = T_2 r_p = T_1 r^{\gamma-1} r_p$$

$$\frac{T_3}{T_4} = \left(\frac{P_3}{P_4} \right)^{\frac{\gamma-1}{\gamma}} = \left(\frac{P_3}{P_1} \right)^{\frac{\gamma-1}{\gamma}} = \left(\frac{P_3}{P_2} \cdot \frac{P_2}{P_1} \right)^{\frac{\gamma-1}{\gamma}} = \left(r_p \cdot \frac{P_2}{P_1} \right)^{\frac{\gamma-1}{\gamma}} = r_p^{\frac{\gamma-1}{\gamma}} \cdot r^{\gamma-1}$$

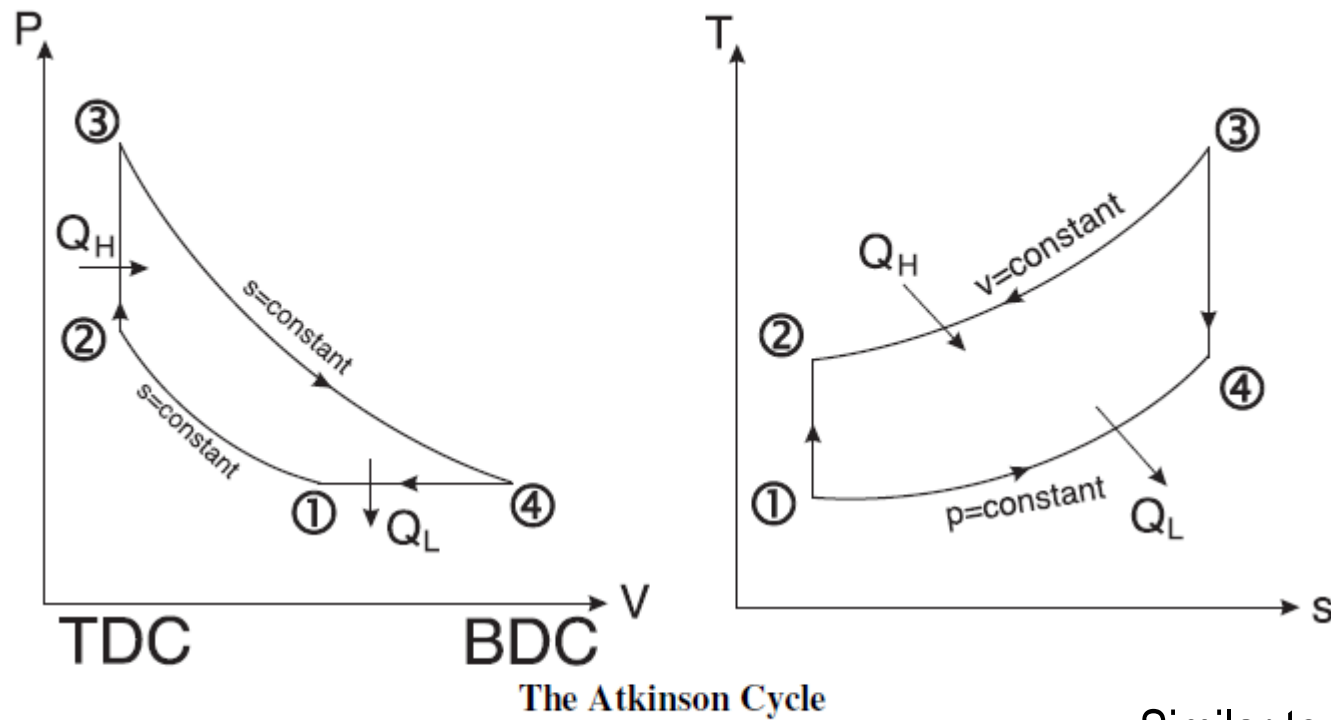
since,

$$\frac{P_2}{P_1} = \left(\frac{v_1}{v_2} \right)^{\gamma} = r^{\gamma}$$

and

$$T_4 = \frac{T_3}{r_p^{\frac{\gamma-1}{\gamma}} r^{\gamma-1}} = \frac{T_1 r_p r^{\gamma-1}}{r_p^{\frac{\gamma-1}{\gamma}} r^{\gamma-1}} = T_1 r_p^{\frac{1}{\gamma}}$$

$$\eta_{th} = 1 - \gamma \left[\frac{\frac{1}{r_p^{\frac{1}{\gamma}}} - 1}{(r_p - 1)r^{\gamma-1}} \right]$$



$$r = \frac{V_1}{V_2} = \text{compression ratio}$$

$$r_\alpha = \frac{V_4}{V_3} = \text{expansion ratio (larger than the compression ratio)}$$

$$\eta_{th} = 1 - \gamma^{1-\gamma} \frac{\left[\frac{r_\alpha}{r} - 1 \right]}{\left[\frac{r_\alpha^\gamma}{r^\gamma} - 1 \right]} = 1 - \gamma \left[\frac{r_\alpha - r}{r_\alpha^\gamma - r^\gamma} \right]$$

Similar to Otto cycle but with constant pressure heat rejection that allows for a higher expansion ratio (more work extraction) compared to the compression ratio and in turn a higher cycle efficiency.

ATKINSON CYCLE

Advantages

- Reduces pumping losses in the engine by using a more efficient combustion cycle
- Reduces the amount of fuel-air to be compressed, resulting in greater fuel efficiency
- Allows the expansion stroke to be longer than the compression stroke, improving thermal efficiency

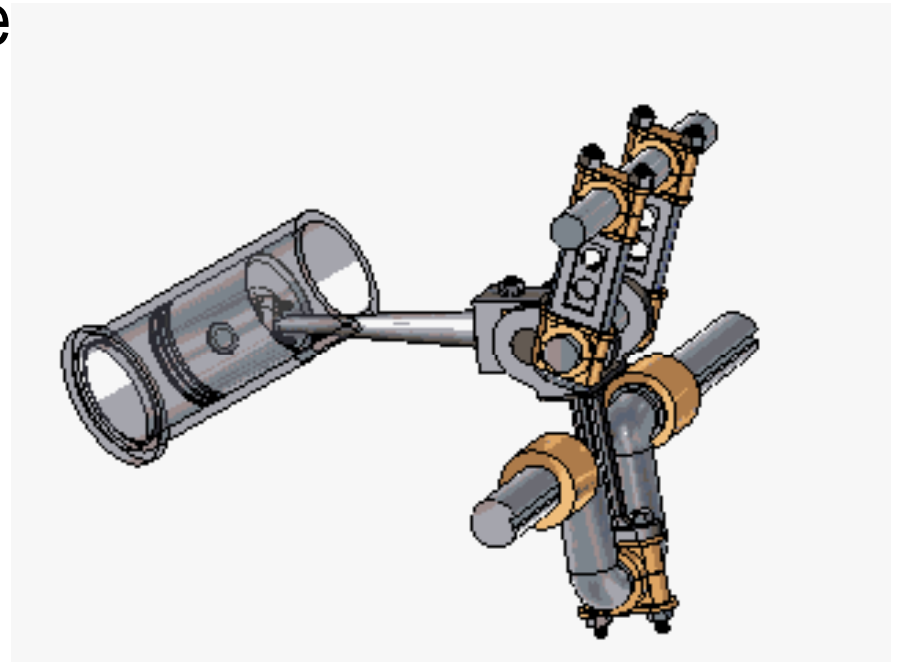
Disadvantages

- Requires a more complex linkage system
- Some engine power is lost due to backflow of fuel-air mixture

Atkinson cycle is used in hybrid vehicles

Atkinson-Cycle Engines

- Operates on a four-stroke cycle
- **Intake, compression, combustion, and exhaust strokes occur in a one revolution of crankshaft**
- Combustion stroke is longer than compression stroke allowing more expansion of combustion gases
- **Greater efficiency when compared to Otto engines**



The Atkinson Cycle Engine

Atkinson-Cycle Engines Continued

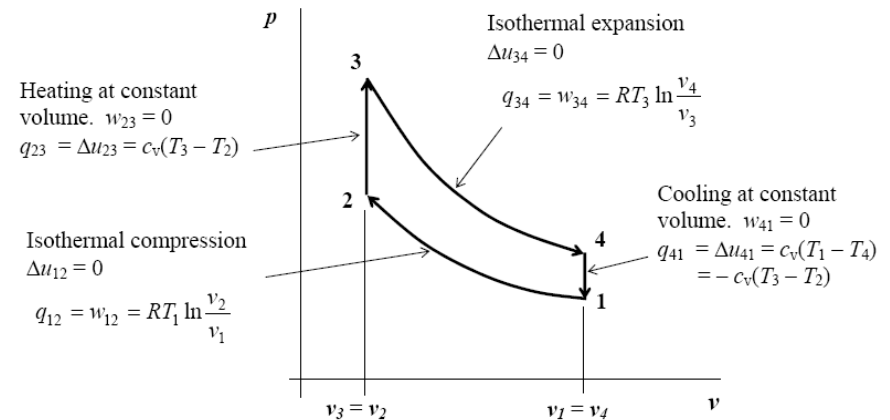
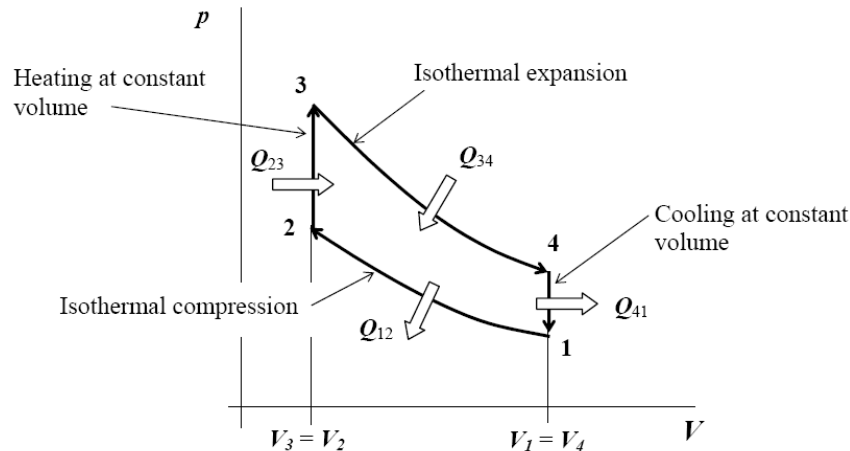
- Biggest disadvantage is reduction in power density (power/unit volume) arising from the reduction in air intake.
- Atkinson cycle engine can be supplemented with electric motor to provide more power if needed.
- Electric motors can be used in combination or independent of Atkinson cycle engines to provide the desired power output most efficiently.

STIRLING CYCLE

The Stirling Cycle

Process breakdown

Stirling cycle.



Recall that $v_3 = v_2$ and $v_4 = v_1$, also $v_1 > v_2$

$$w_{net} = RT_1 \ln \frac{v_2}{v_1} + RT_3 \ln \frac{v_1}{v_2} = -RT_1 \ln \frac{v_1}{v_2} + RT_3 \ln \frac{v_1}{v_2}$$

$$w_{net} = R(T_3 - T_1) \ln \frac{v_1}{v_2} \blacktriangleleft$$

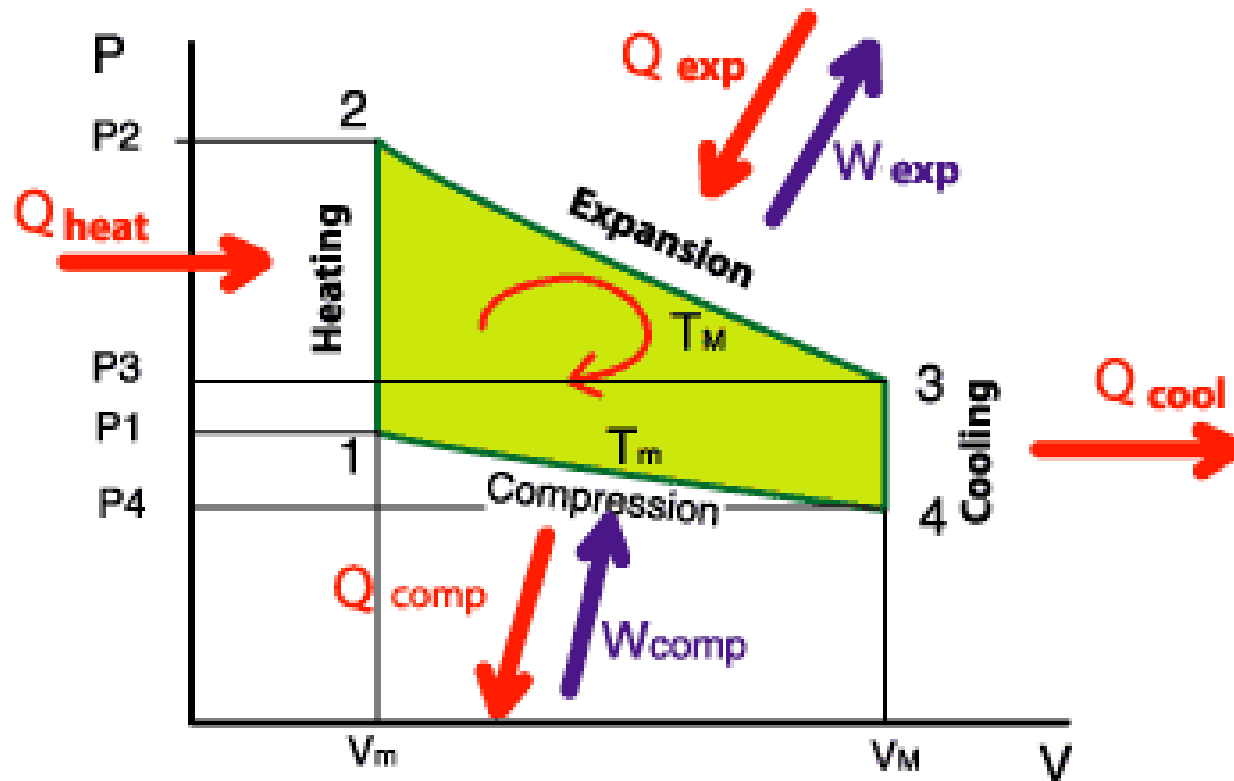
For stirling engine with regeneration

$$\eta = \frac{W_{net}}{Q_{34}} = \frac{R(T_3 - T_1) \ln \frac{V_4}{V_3}}{RT_3 \ln \frac{V_4}{V_3}} = \frac{T_3 - T_1}{T_3} = 1 - \frac{T_1}{T_3} = 1 - \frac{T_C}{T_H}$$

For stirling engine without regeneration

$$\eta = \frac{R(T_3 - T_1) \ln \frac{V_1}{V_2}}{c_V(T_3 - T_1) + RT_3 \ln \frac{V_1}{V_2}}$$

Stirling cycle



Source: <http://www.robertstirlingengine.com/theory.php#thermo>

Stirling cycle

According to the good old principle: "nothing is lost, nothing is created, everything is transformed" we can approach the energy exchanges during a cycle of Stirling: any loss of calorific energy during a cycle is equal to the recovered mechanical energy during this same cycle.

On the diagram opposite, we see that we provide in calorific energy: $Q_{exp} + Q_{heat}$

However, we recover: $Q_{cool} + Q_{comp}$

About the mechanical energy, we recover W_{exp} but we provide W_{comp}

The overall balance becomes, by stipulating that the heat energy lost was fully transformed into mechanical energy:

$$Q_{exp} + Q_{heat} - Q_{cool} - Q_{comp} = W_{exp} - W_{comp}$$

This leads us to speak about mechanical efficiency: it is the ratio of recovered mechanical energy (the engine goal, in fact) to the heat energy that we must provide:

$$\text{Efficiency} = (W_{exp} - W_{comp}) / (Q_{exp} + Q_{heat})$$

NB : as we can see it in the page "[the regenerator](#)", it is not necessary to provide Q_{heat} if a regenerator is installed. Indeed, at this time there, we recover Q_{cool}

If we refer to the page "[the principles](#)", we are able to show that the efficiency may be expressed according to the temperatures (expressed in Kelvin) of the heat source and of the cold source, according to the following formula:

$$\text{Efficiency} = 1 - T_m / T_M$$

STIRLING CYCLE

Advantages:

- Potentially high thermal efficiency. Fairly insensitive to fuel or fuel quality. No lubricants needed; the working fluid can support bearing loads.

Disadvantages:

- High cost, which may decrease as less expensive materials are used and more units are produced. Low power density and low power-to-weight ratio. Working fluid leakage and sealing.

Numerical example

- Consider an ideal Stirling-cycle engine in which the state at the beginning of the isothermal compression process is 100 kPa, 25°C, the compression ratio is 6, and the maximum temperature in the cycle is 1100°C. Calculate the maximum cycle pressure and the thermal efficiency of the cycle with and without regenerators.
- Ans: 2763 kPa, 0.374, 0.783

Source: <http://nptel.iitm.ac.in/courses/101101001/downloads/lec20-problem-solution.pdf>

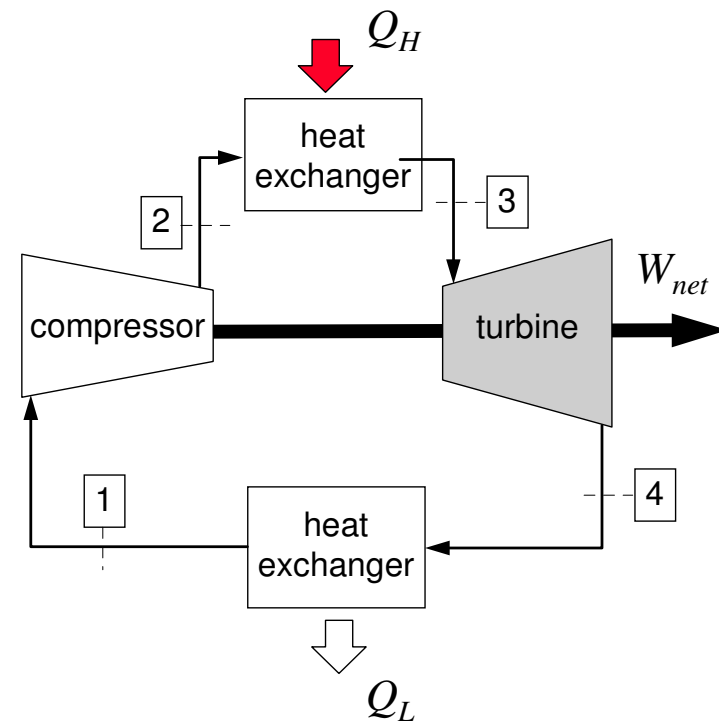
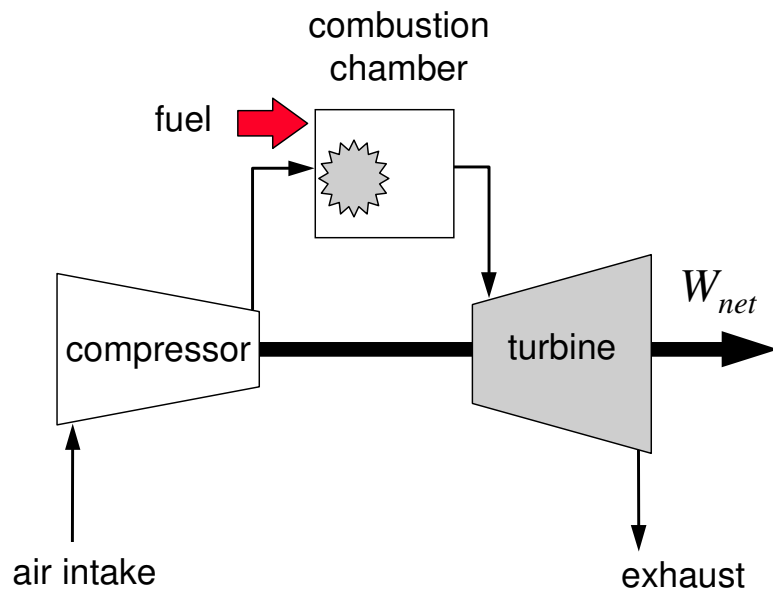
BRAYTON CYCLE

BRAYTON CYCLE

Open Brayton cycle

Closed Brayton cycle

Gas turbine cycle



EFFICIENCY OF A BRAYTON CYCLE

Efficiency of a Brayton cycle

1st law for this cycle: $W = Q_H - Q_L$

energy conversion efficiency is:

$$\eta = \frac{\text{useful work}}{\text{heat input}} = \frac{W}{Q_H} = \frac{Q_H - Q_L}{Q_H}$$

$$\eta = 1 - \frac{Q_L}{Q_H} = 1 - \frac{mC_p(T_4 - T_1)}{mC_p(T_3 - T_2)}$$

$$\eta = 1 - \frac{T_1(T_4/T_1 - 1)}{T_2(T_3/T_2 - 1)}$$

Efficiency of a Brayton cycle (cont.)

for an isentropic process:

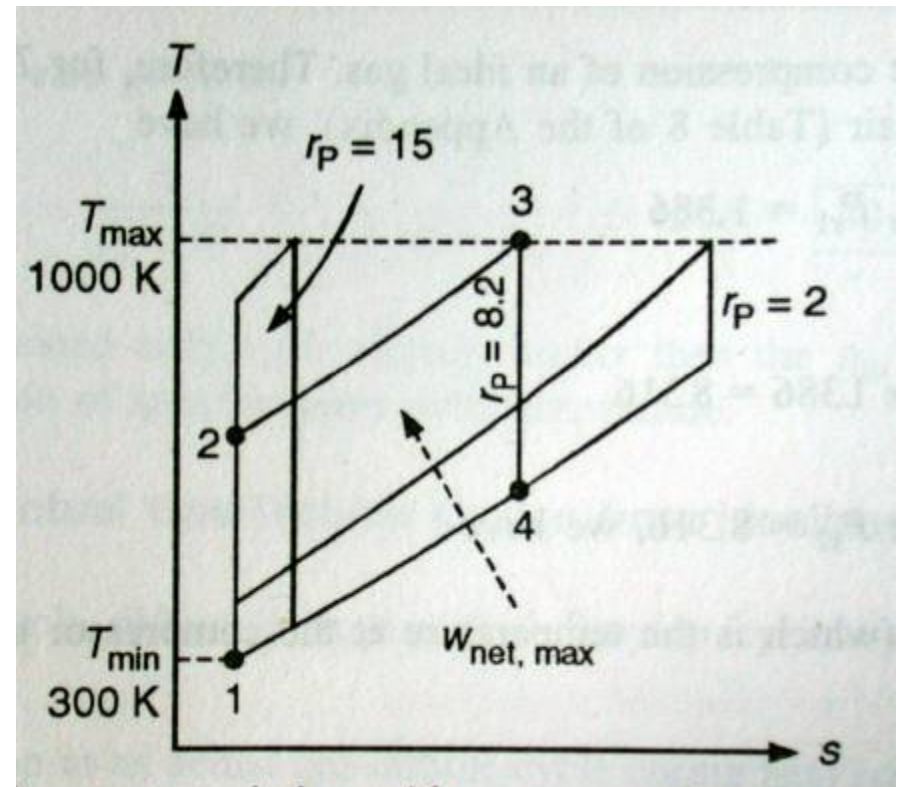
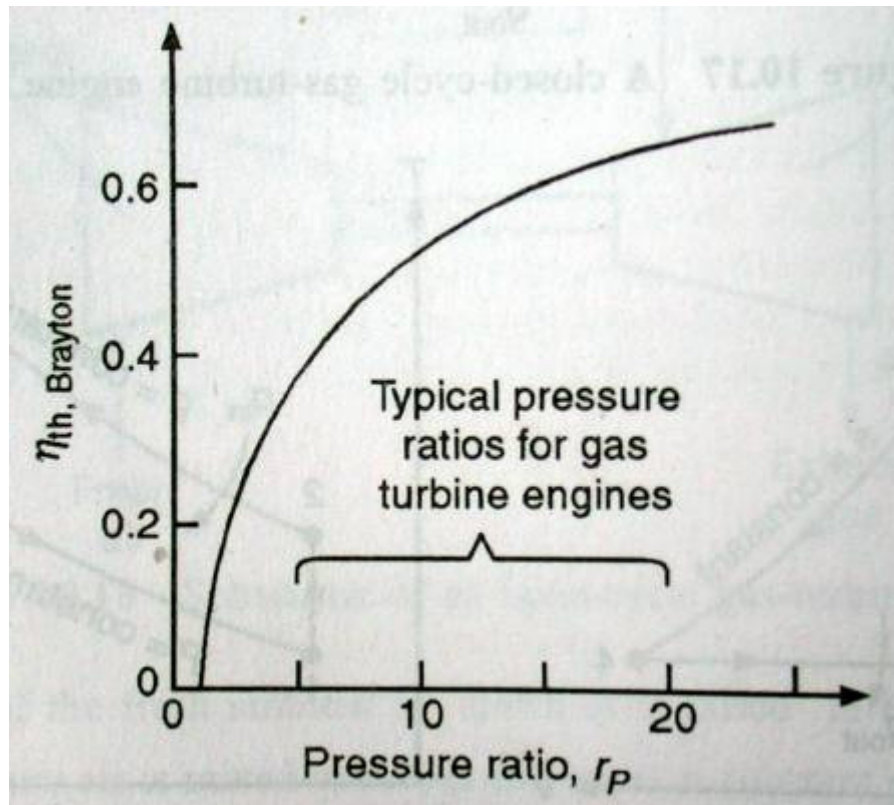
$$P_1/P_2 = (V_2/V_1)^k$$

in case of an ideal gas: $P_1V_1/P_2V_2 = T_1/T_2$

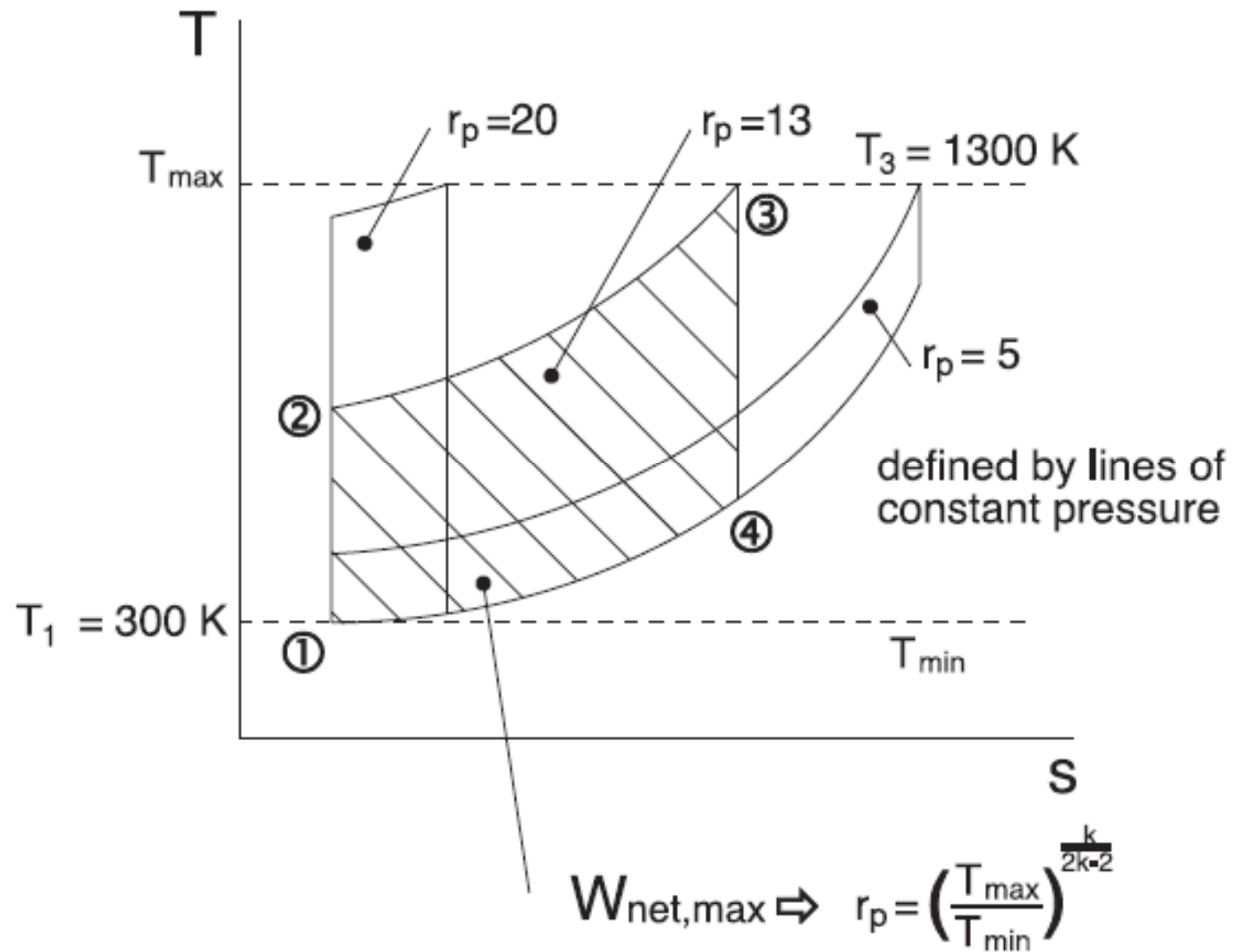
$$\left(\frac{T_2}{T_1}\right)^{\frac{k}{k-1}} = \frac{P_2}{P_1} = \frac{P_3}{P_4} = \left(\frac{T_3}{T_4}\right)^{\frac{k}{k-1}}$$
$$\frac{T_3}{T_2} = \frac{T_4}{T_1}$$

$$\eta = 1 - \frac{T_1}{T_2} = 1 - \left(\frac{P_1}{P_2}\right)^{\frac{k-1}{k}}$$

Efficiency vs pressure ratio



Max net work

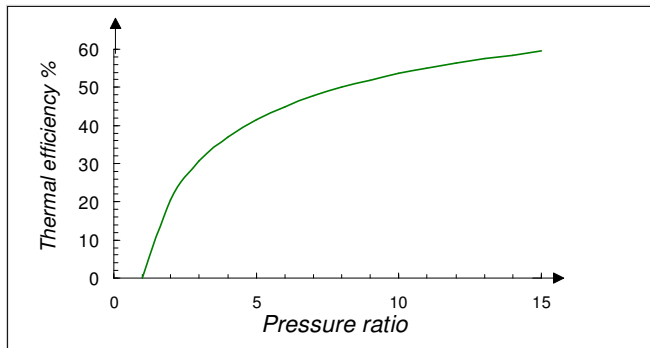


EFFICIENCY (cont.)

Efficiency of a Brayton cycle (cont.)

$$\eta = 1 - \frac{1}{\left(P_2/P_1\right)^{\frac{k-1}{k}}}$$

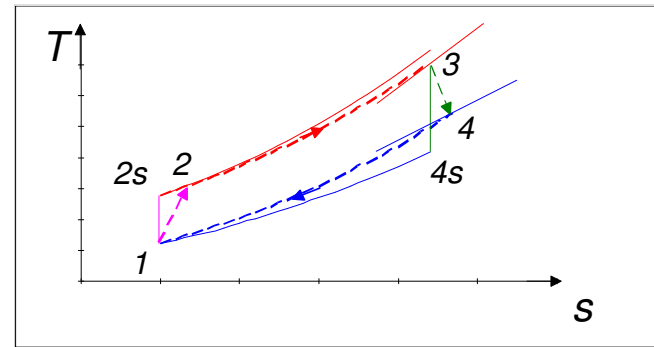
↖ isentropic pressure ratio



The actual gas turbine process

$$\eta_{comp} = \frac{h_{2s} - h_1}{h_2 - h_1}$$

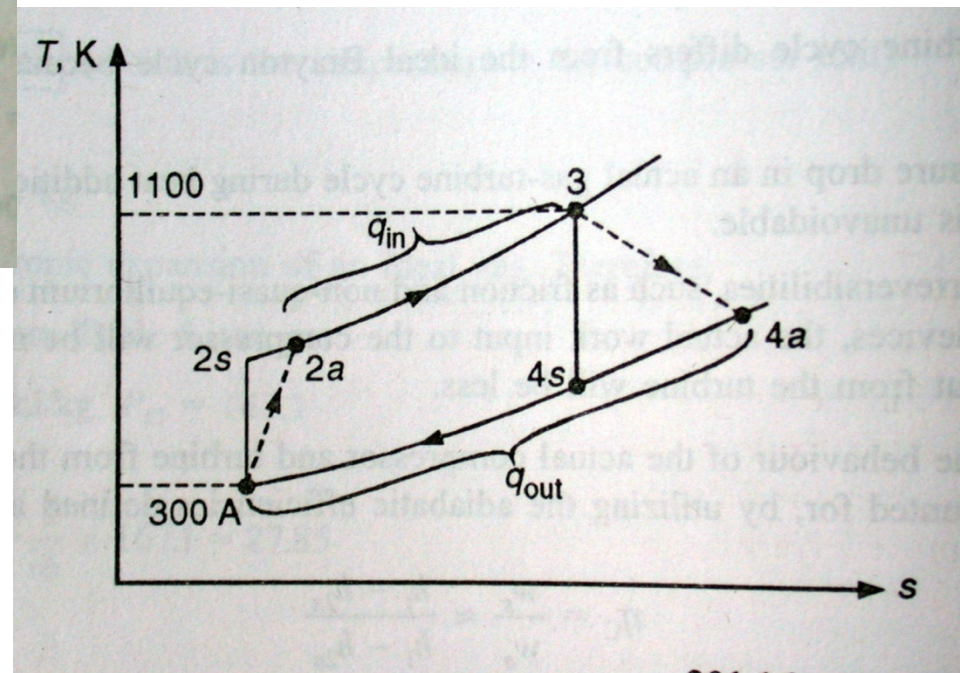
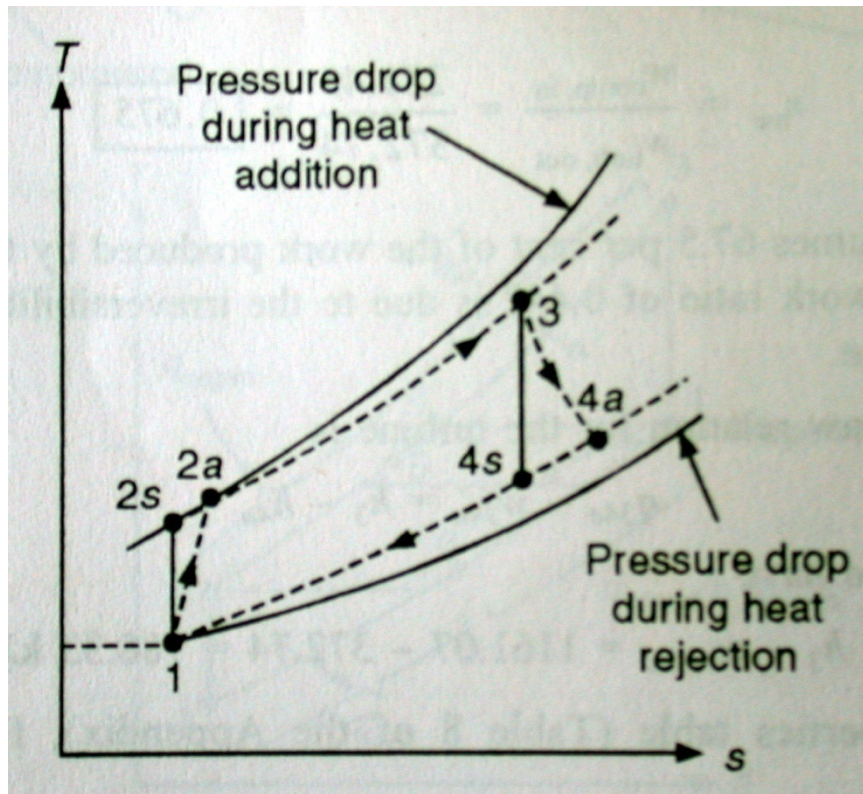
$$\eta_{turb} = \frac{h_3 - h_4}{h_3 - h_{4s}}$$



DEVIATION OF ACTUAL GAS-TURBINE CYCLE FROM IDEALIZED CYCLE

The actual gas turbine cycle differs from the ideal Brayton cycle because of the following reasons:

- Some pressure drop in an actual gas turbine cycle during heat addition and heat rejection processes is unavoidable.
- Owing to irreversibility, such as friction and non-quasi-equilibrium operation conditions of actual devices, the actual work input of the compressor will be more, and the actual work output from the turbine will be less.



BRAYTON CYCLE

Advantages:

- These continuous flow machines offer a high power density, and a respectable efficiency at full load on the order of 30% for the open cycle, and even higher for the closed. Brayton cycles are fairly insensitive to fuel quality. The sizes considered for this study also have a distinct advantage over other methods in that the rotating components can be supported by air bearings in any orientation. Thus, no auxiliary lubricating fluid or oil is needed.

Disadvantage:

- The primary disadvantage is cost. Although expensive alloys are now used, the cost will decrease as less expensive materials are used and more units are produced.