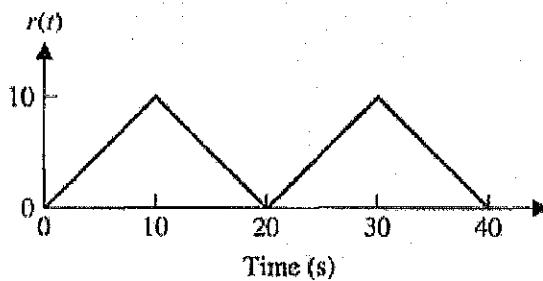


ME 5200/6200 and ECE 5615/6615 Classical Control

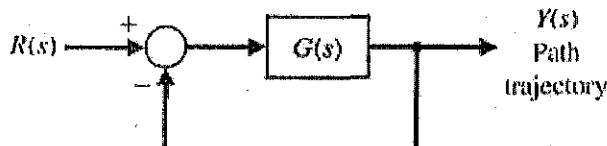
Homework 06 Solutions

Problem 1

Consider the following reference input and block diagram below:



(a)



(b)

Let the transfer function for the system be:

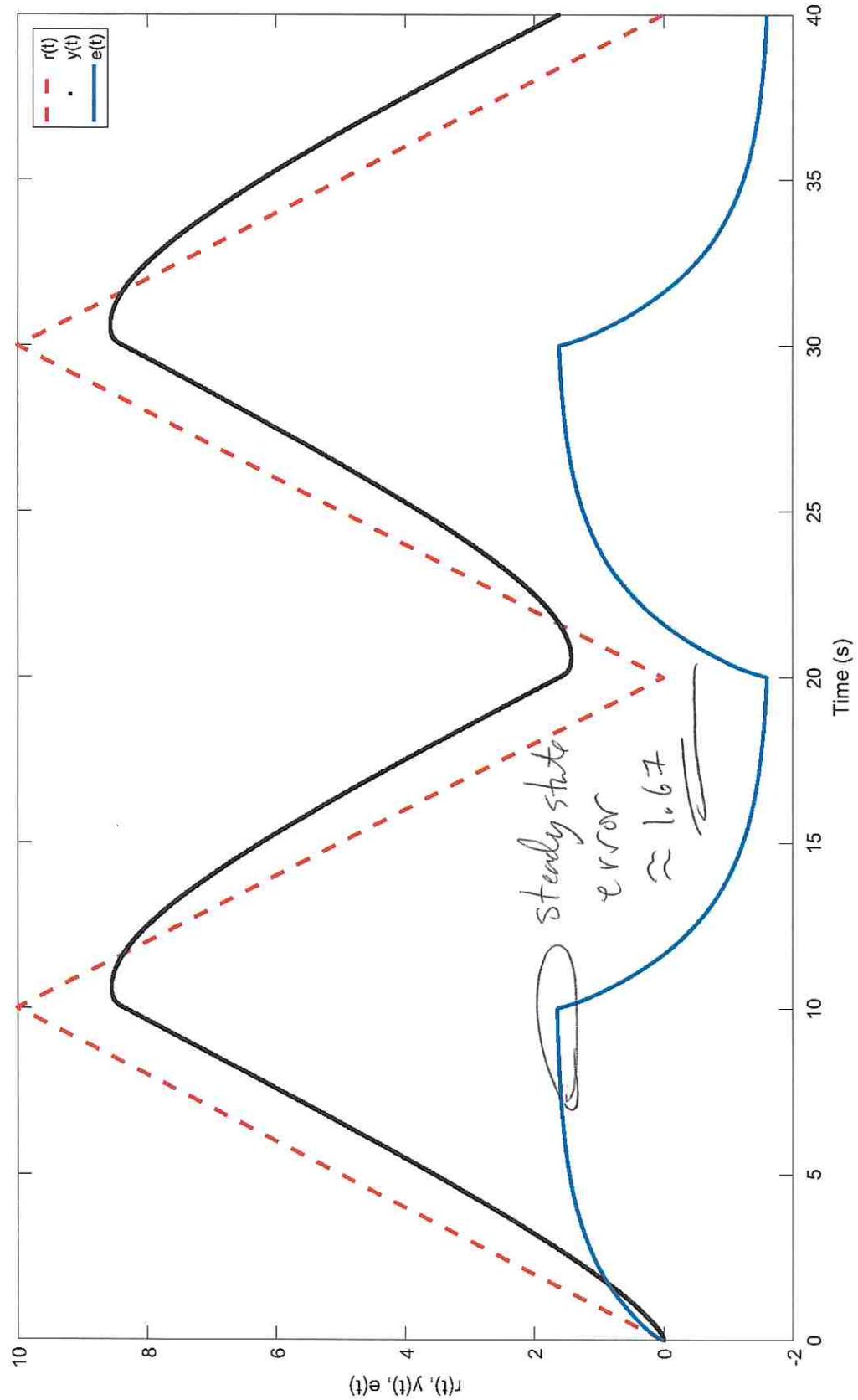
$$G(s) = \frac{75(s+1)}{s(s+5)(s+25)}$$

- (a) Use Matlab to plot the time response of the output of the closed-loop system.
- (b) Use Matlab to make a plot of the tracking error as a function of time.
- (c) From the error plot, what is the steady state error when the input $r(t)$ is shown above.

Make sure to submit a PDF print out of your Matlab code (m-file or Simulink diagram).

Can solve by hand as well, we have a ramp from $0 \rightarrow 10s$ and same from $(10 \rightarrow 20)s$, thus:

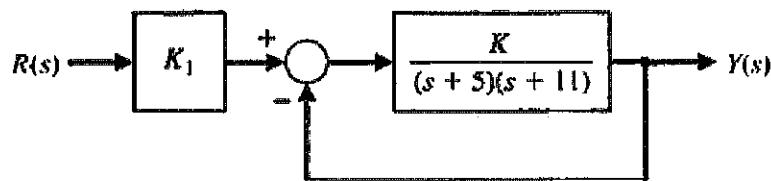
$$\begin{aligned} K_V &= \lim_{s \rightarrow 0} s G(s) = \lim_{s \rightarrow 0} s \left(\frac{75(s+1)}{s(s+5)(s+25)} \right) \\ &= \frac{75}{125} = 0.6 \Rightarrow \rho_{ss} = \frac{1}{K_V} = 1.67 \end{aligned}$$



Problem 2

For the closed-loop system below,

- Determine the steady-state error for a unit step input in terms of K and K_1 , where $E(s) = R(s) - Y(s)$.
- Select K_1 so that the steady-state error is zero.



(a) Steady-state error is:

$$e_{ss} = \lim_{s \rightarrow 0} \frac{(s+5)(s+11) + K(1-K)}{(s+5)(s+11) + K}$$

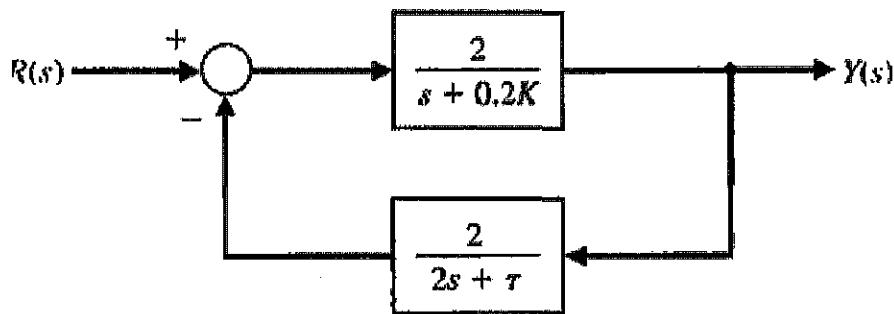
$$e_{ss} = \frac{55 + K(1-K)}{\cancel{55 + K}}$$

(b) To achieve zero s.s., Pick K_1 as:

$$K_1 = 1 + \frac{55}{K}$$

Problem 3

Consider the system below:



$$G_c(s)G(s) = \frac{2}{s + 0.2K} \quad \text{and} \quad H(s) = \frac{2}{2s + \tau}.$$

If $\tau = 2.43$, determine the value of K such that the steady-state error of the closed-loop system response to a unit step input, $R(s) = 1/s$, is zero.

Closed-loop T.F. is :

$$T(s) = \frac{2(2s + \tau)}{(s + 0.2K)(2s + \tau) + 4}$$

If $R(s) = 1/s$, then output is

$$Y(s) = \frac{2(2s + \tau)}{(s + 0.2K)(2s + \tau) + 4} * \frac{1}{s}$$

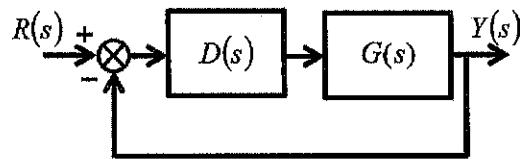
and $y_{ss} = \lim_{s \rightarrow 0} sY(s) = \frac{2\tau}{0.2K + 4}$, and $e_{ss} = 1 - y_{ss}$

and we want $y_{ss} = 1$, so we pick K as

$$K = 10 - 20/\tau \Rightarrow \boxed{e_{ss} = 0}$$

Problem 4

Consider the closed-loop block diagram shown below. Let $G(s) = \frac{1}{(s+3)(s+4)}$. For each controller given below, find the steady-state error for: (i) a unit step input, (ii) a unit ramp input, and (iii) a unit parabolic input. State the system type and calculate the appropriate error constant for each of the five controllers. Assume, in each case, that the closed-loop system is stable.



- (a) $D(s) = 2$ (a P controller)
- (b) $D(s) = \frac{2(s+5)}{s}$ (a PI controller)
- (c) $D(s) = 2(s+5)$ (a PD controller)
- (d) $D(s) = \frac{2(s+5)(s+1)}{s(rs+1)}$, $\tau = \frac{1}{30}$ (a PID controller with a low-pass filter)
- (e) $D(s) = \frac{2(s+5)(s+1)}{s^2}$ (a PID controller with a second integrator)

Problem 4

(4) (a) $T(s) = \frac{2}{(s+3)(s+4)}$

No integrators

Type 0 (1 pt)

$$K_p = \lim_{s \rightarrow 0} sT(s) = \boxed{\frac{1}{6}} \quad (1 \text{ pt})$$

$$(i) e_{ss, \text{step}} = \frac{1}{1 + K_p} = \frac{1}{1 + \frac{1}{6}} = \boxed{\frac{6}{7}} \quad (1 \text{ pt})$$

$$(ii) e_{ss, \text{ramp}} = \infty$$

Wrong
Type
for finite
error!

$$\boxed{e_{ss, \text{step}} = \frac{6}{7}} \quad \leftarrow (1 \text{ pt}) \rightarrow (iii) e_{ss, \text{para}} = \infty$$

$$(iii) e_{ss, \text{para}} = \infty$$

(b) $T(s) = \frac{2(s+5)}{(s+3)(s+4)s}$

1 Integrator

Type 1 (1 pt)

$$K_v = \lim_{s \rightarrow 0} sT(s) = \frac{10}{12} = \boxed{\frac{5}{6}} \quad (1 \text{ pt})$$

$$(i) e_{ss, \text{step}} = 0$$

From

$$(iii) e_{ss, \text{para}} = \infty$$

Type
(1 pt)

$$(ii) e_{ss, \text{ramp}} = \frac{1}{K_v}$$

$$\boxed{e_{ss, \text{ramp}} = \frac{6}{5}}$$

(c) $T(s) = \frac{2(s+5)}{(s+3)(s+4)s}$

No integrators

Type 0 (1 pt)

$$K_p = \lim_{s \rightarrow 0} T(s) = \frac{10}{12} = \boxed{\frac{5}{6}} \quad (1 \text{ pt})$$

$$(i) e_{ss, \text{step}} = \frac{1}{1 + K_p} = \frac{1}{1 + \frac{5}{6}} = \boxed{\frac{6}{11}} \quad (1 \text{ pt})$$

$$(ii) e_{ss, \text{ramp}} = \infty$$

$$\boxed{e_{ss, \text{step}} = \frac{6}{11}} \quad \leftarrow (1 \text{ pt}) \rightarrow (iii) e_{ss, \text{para}} = \infty$$

$$(iii) e_{ss, \text{para}} = \infty$$

$$(d) T(s) = \frac{2(s+5)(s+1)}{s(2s+1)(s+3)(s+4)}$$

1 integrator
Type 1 (1 pt)

$$K_v = \lim_{s \rightarrow 0} sT(s) = \frac{10}{12}$$

$$\boxed{K_v = \frac{5}{6}} \quad (1 \text{ pt})$$

$$(i) e_{ss, \text{step}} = 0$$

$$(ii) e_{ss, \text{ramp}} = \frac{1}{K_v} = \boxed{\frac{6}{5}} \quad (1 \text{ pt})$$

$$(iii) e_{ss, \text{para}} = \infty$$

$$(e) T(s) = \frac{2(s+5)(s+1)}{s^2(s+3)(s+4)}$$

2 integrators
Type 2 (1 pt)

$$K_a = \lim_{s \rightarrow 0} s^2 T(s) = \frac{10}{12}$$

$$\boxed{K_a = \frac{5}{6}} \quad (1 \text{ pt})$$

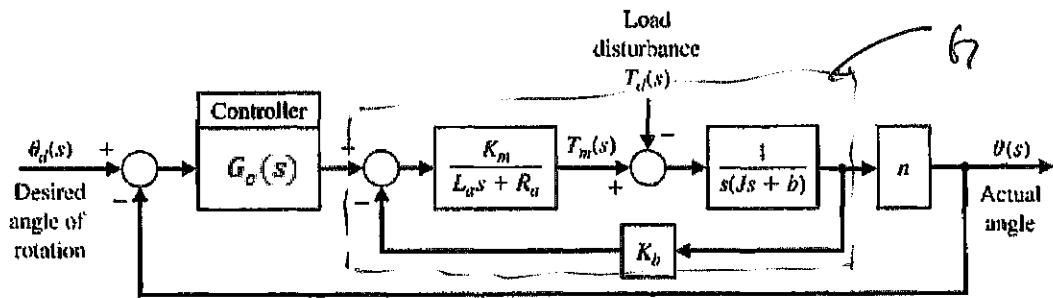
$$(i) e_{ss, \text{step}} = 0$$

$$(ii) e_{ss, \text{ramp}} = 0$$

$$(iii) e_{ss, \text{para}} = \frac{1}{K_a} = \boxed{\frac{6}{5}}$$

Problem 5

A robot arm has a shoulder joint where the model of the joint is given by the block diagram shown below. The joint uses a DC motor with armature control and has gears on the output shaft, so n is a constant. In the block diagram, there is a load disturbance $T_d(s)$ which represents the effect of the load. The entire closed-loop system shown below has two inputs, one for the desired angle of rotation and the other is the load disturbance. For simplicity, assume that K_m , L_a , R_a , J , b , n , and K_b are all equal to 1.



- Determine the transfer function that relates the desired angle of rotation $\theta_d(s)$ to the actual angle $\theta(s)$. What is the system type for this case?
- Determine the transfer function that relates the load disturbance $T_d(s)$ to the actual angle $\theta(s)$. What is the system type for this case?
- Determine the steady-state error when the desired angle $\theta_d(s)$ is a ramp input with slope B and the controller is $G_c(s) = 1/s$ (integral control). You can assume the disturbance is zero.
- Suppose the desired angle is zero and the load disturbance is a step of magnitude D. Determine the steady-state error if $G_c(s) = K$ (proportional control).

(a) T.F. from θ_d to θ :

$$\frac{\underline{\theta(s)}}{\underline{\theta_d(s)}} = \frac{G_{rc}(s) G_r n}{1 + G_{rc}(s) G_r n}$$

$$G_r = \frac{\left(\frac{K_m}{L_a s + R_a} \right) \left(\frac{1}{s(J_s + b)} \right)}{1 + \left(\frac{K_m}{L_a s + R_a} \right) \left(\frac{1}{s(J_s + b)} \right) K_b}$$

$$= \frac{K_m}{(L_a s + R_a)s(J_s + b) + K_m K_b}$$

Hence:

$$\frac{\underline{\theta(s)}}{\underline{\theta_d(s)}} = \frac{G_{rc}(s) \left(\frac{K_m}{(L_a s + R_a)s(J_s + b) + K_m K_b} \right)^n}{1 + G_{rc}(s) \left(\frac{K_m}{(L_a s + R_a)s(J_s + b) + K_m K_b} \right)^n}$$

looking at $G_{rc}(s) G_r(s) n$, $G_r(s)$ is type 0,
 n is a constant, so system type will
be determined by $G_{rc}(s)$!

$$(b) \quad \frac{\Theta(s)}{T_d(s)} = \frac{\frac{n}{s(s\beta+b)}}{1 + G_c(s) \left(\frac{K_m}{(L_a s + R_a)(s)(J_s + b) + K_m K_b} \right)^n}$$

$$(c) \quad \Theta_d(s) = \frac{B}{s^2} \quad G_c(s) = \frac{1}{s}$$

$$\rho_{ss} = \frac{1}{K_v} \quad \text{and} \quad K_v = \lim_{s \rightarrow 0} s G_c(s) G_d(s) n$$

$$K_v = \lim_{s \rightarrow 0} s \left(\frac{1}{s} \right) \left(\frac{K_m}{(L_a s + R_a)s(J_s + b) + K_m K_b} \right)^n$$

$$K_v = \frac{K_m n}{K_m K_b} = \frac{n}{K_b} = \frac{1}{1} = 1$$

$$\Rightarrow \boxed{\rho_{ss} = 1}$$

$$(d) \quad G_c(s) = K$$

~~$$\Theta_{ss} = \lim_{s \rightarrow 0} s \Theta(s) = \lim_{s \rightarrow 0} s \left(\frac{D}{B} \right) \left(\frac{\frac{K_m n}{K_m K_b}}{1 + K \left(\frac{K_m}{K_m K_b} \right) n} \right)$$~~

$$\Theta_{ss} = \frac{DK}{2}$$

$$\Rightarrow \rho_{ss} = \Theta_d - \Theta_{ss} = 0 - \frac{DK}{2} \Rightarrow \boxed{\rho_{ss} = -\frac{DK}{2}}$$