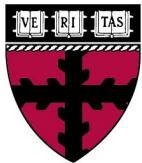
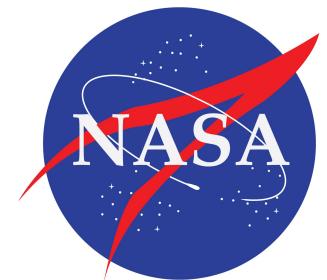
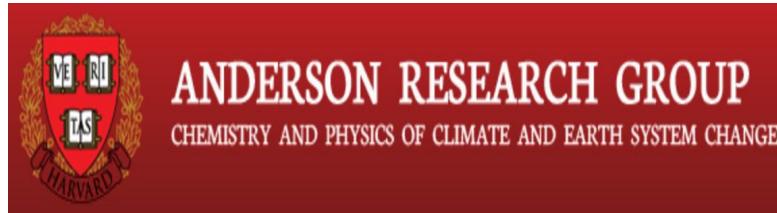


# **Linking Climate Change with Ozone Loss over the US in Summer: New Instrument for the Detection of HCl in the Stratosphere**

ES96 Spring 2016



**HARVARD**  
John A. Paulson  
School of Engineering  
and Applied Sciences





# HCl Instrument

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- Measure HCl concentrations in the lower stratosphere:
  - 50ppt accuracy
  - Precision of 10%
- Integrated Cavity Output Spectroscopy (ICOS)
- High altitude operation





# Overview

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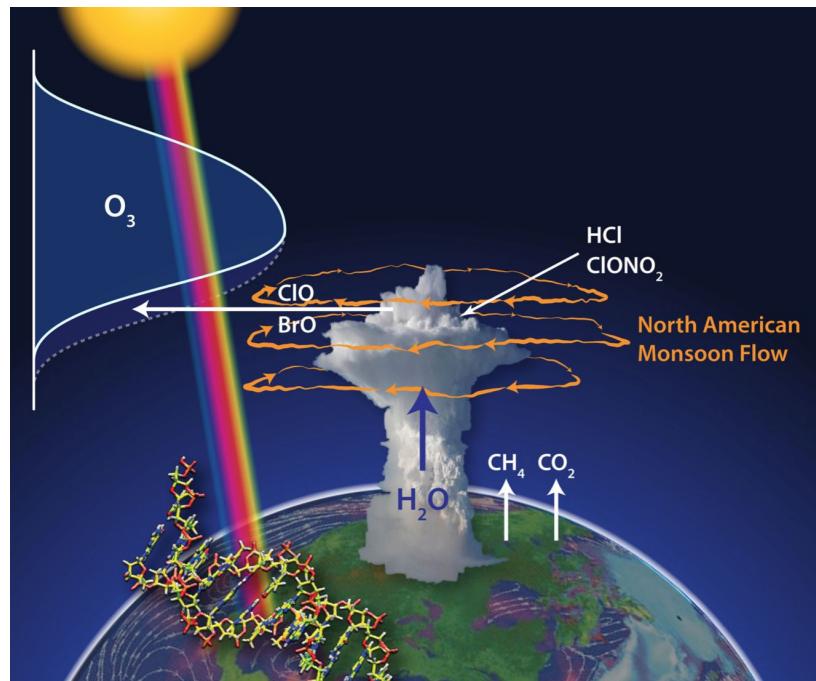
1. Scientific Context
2. System Overview
3. Design Details
  - a. Laser
  - b. Cavity
  - c. Detector
  - d. Pump
  - e. System Wide Design
4. Next Steps





# Scientific Context

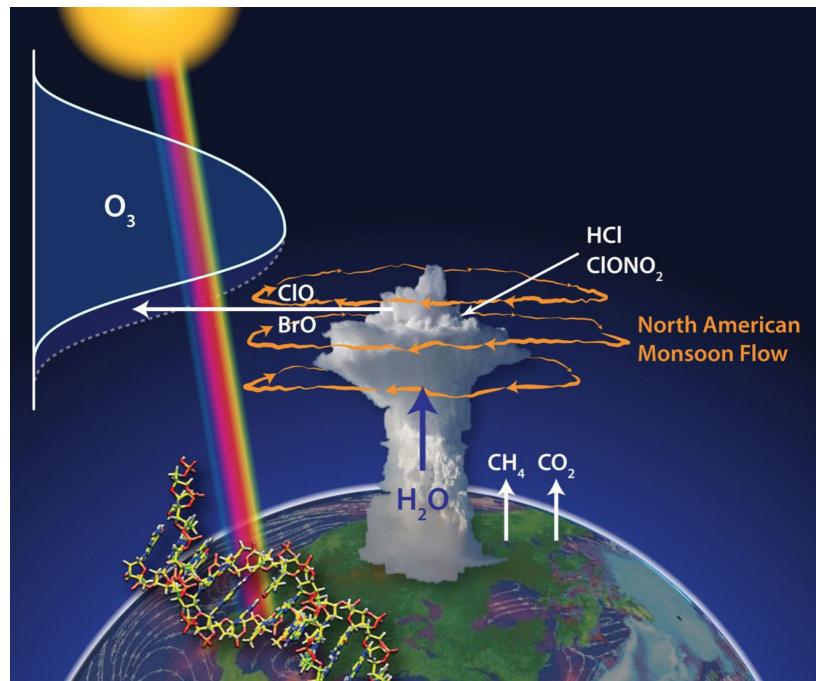
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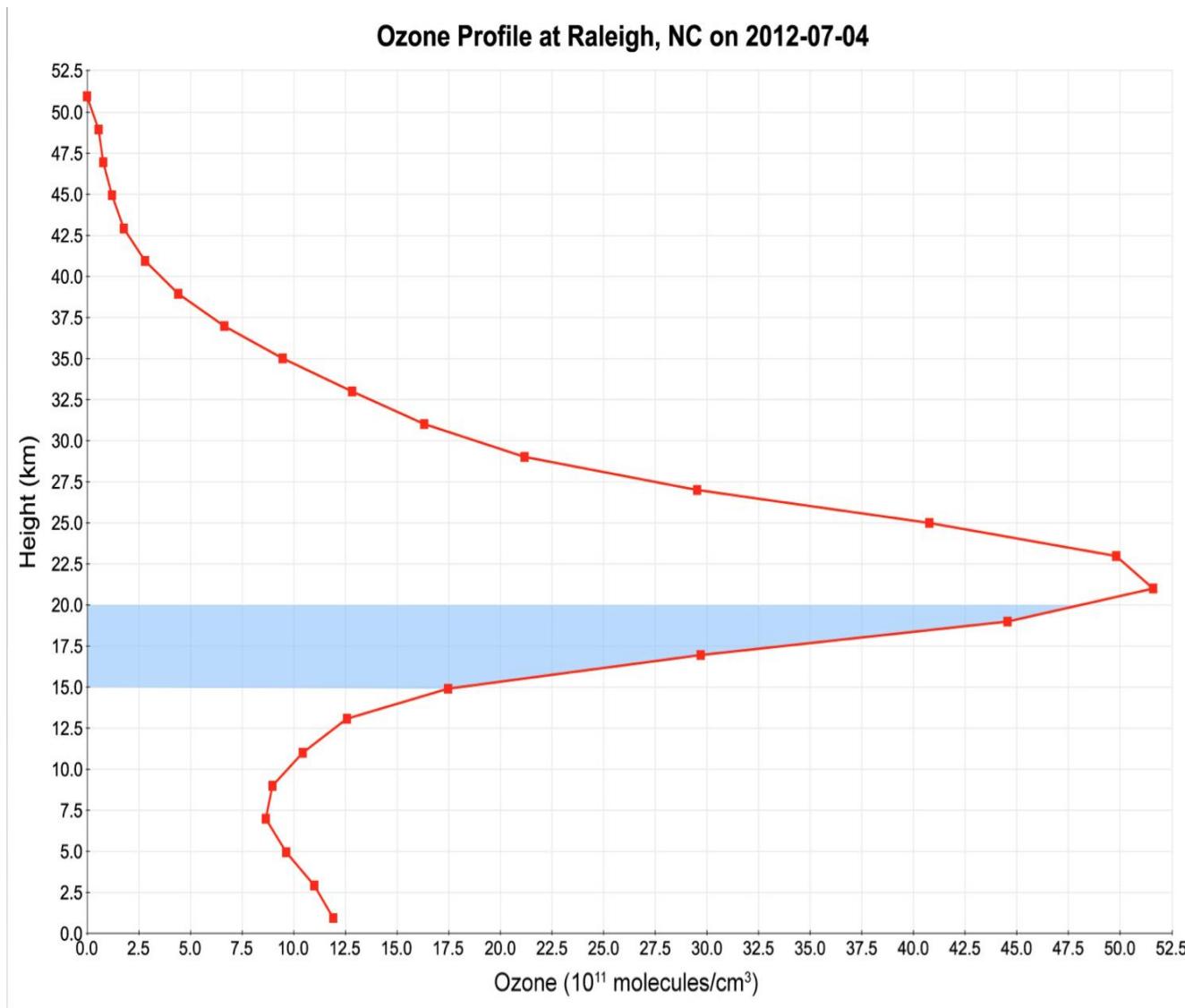
# Scientific Context

1. Ozone
2. HCl Effect on Ozone
3. Stratospheric Ozone Depletion in Central US
4. ASC<sup>5</sup>ENA Proposal



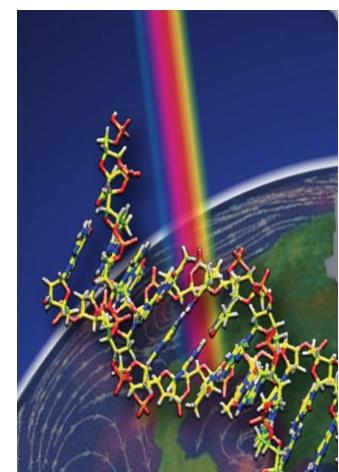
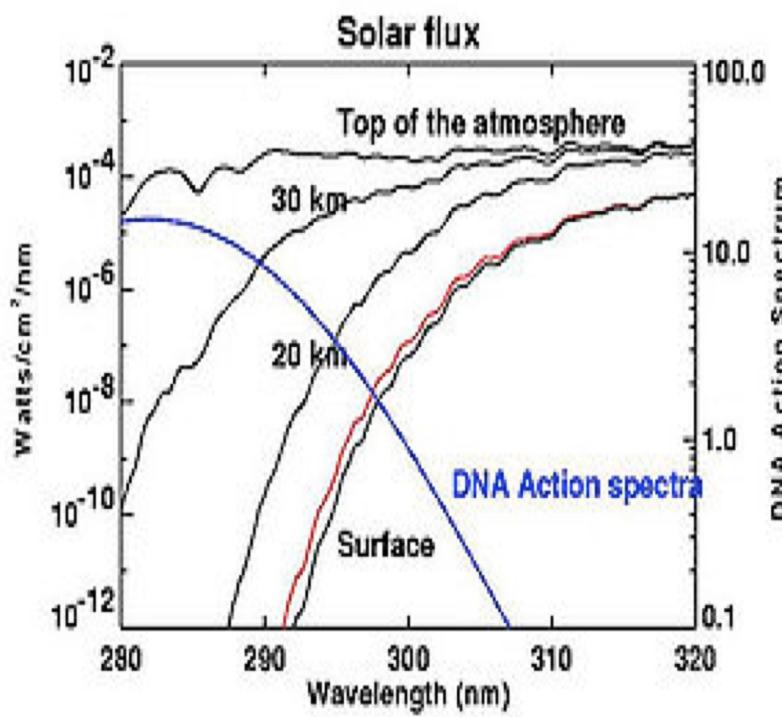
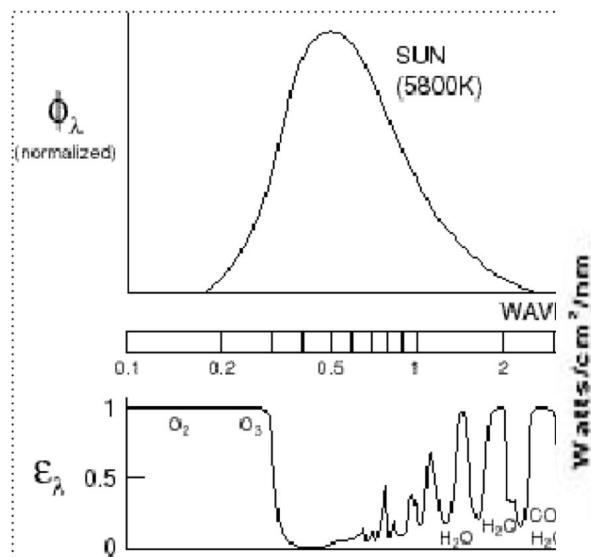


# Location of Ozone over the US





# Effect of Ozone

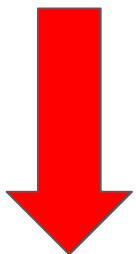




# Why do we care?

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**1% decrease in ozone column**



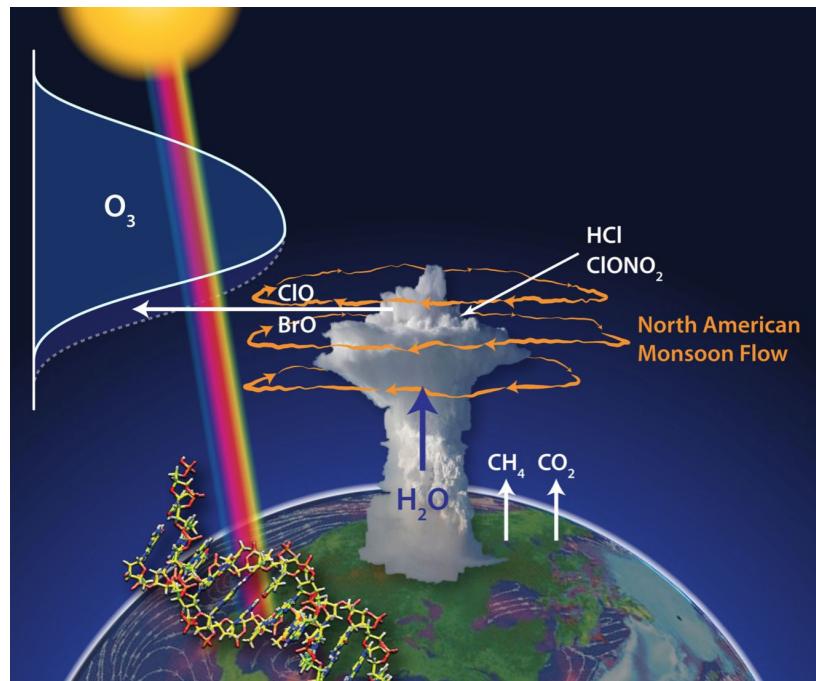
**3% increase in skin cancer  
rates. 3,500,000 new skin  
cancer cases per year in the US**





# Scientific Context

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2. HCl Effect on Ozone
3. Stratospheric Ozone Depletion in Central US
4. ASC<sup>5</sup>ENA Proposal



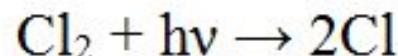
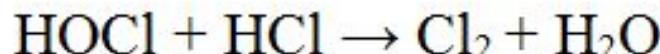
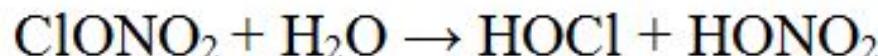
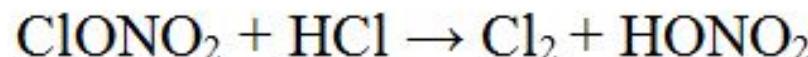


# HCl Effect on Ozone

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- HCl is a reservoir of inorganic chlorine in the atmosphere
- Presence of HCl in the atmosphere creates the potential for chlorine activation
- Chlorine activation occurs on particle surfaces

## Activation of Chlorine Catalytic Cycle

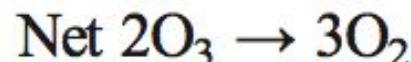
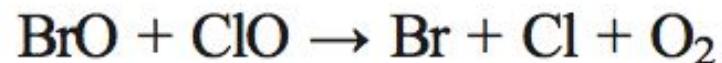
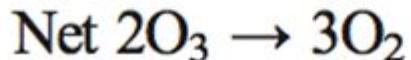
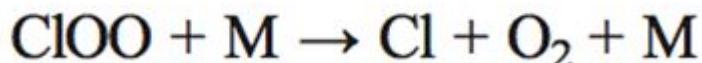
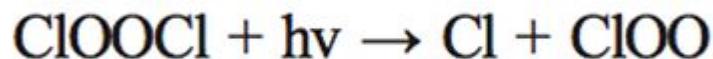
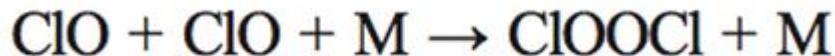




# HCl Effect on Ozone

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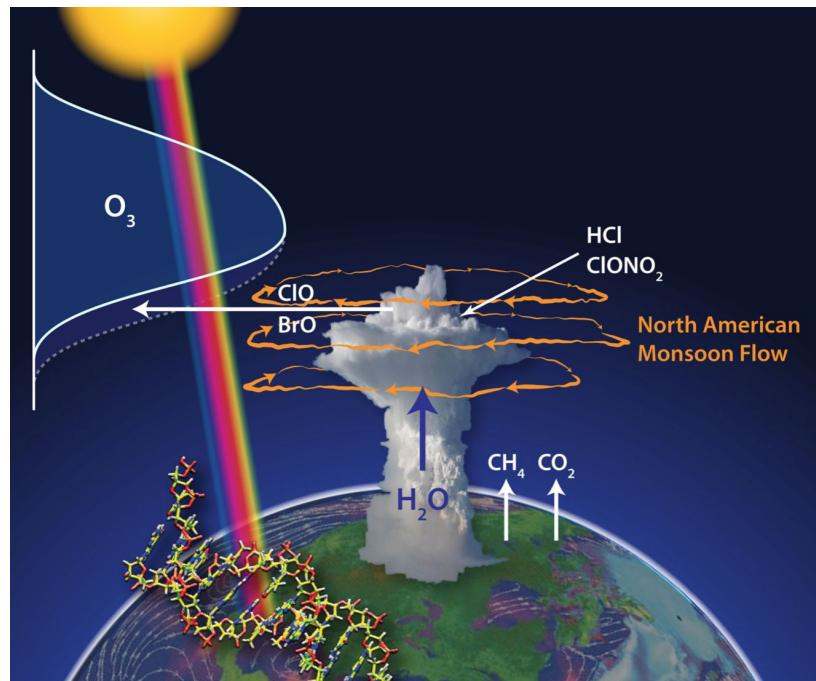
- Cl and Br act in catalytic cycles for the degradation of  $O_3$





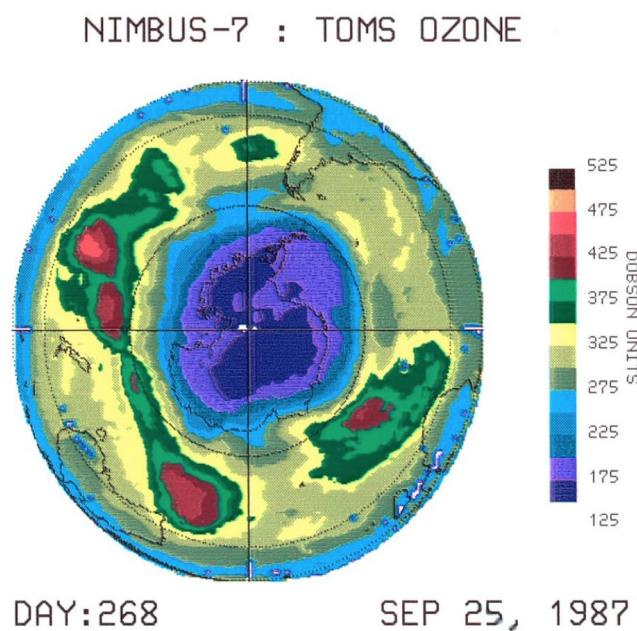
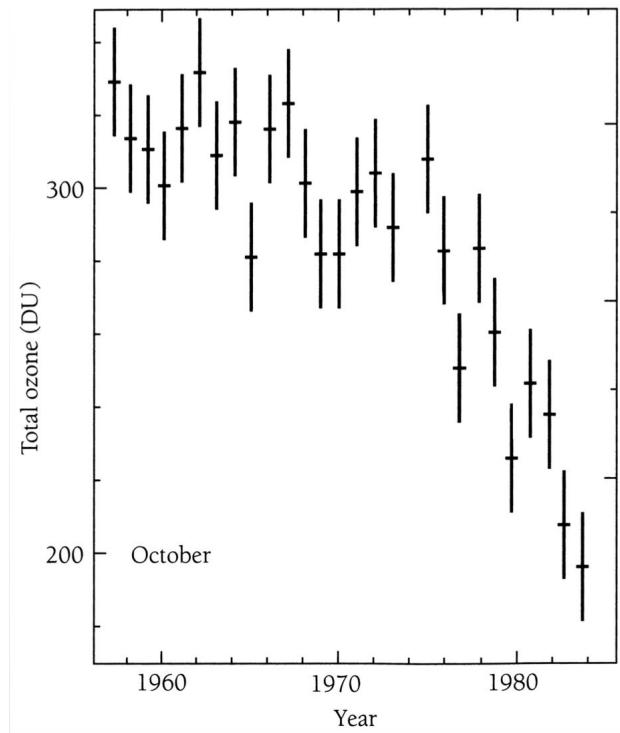
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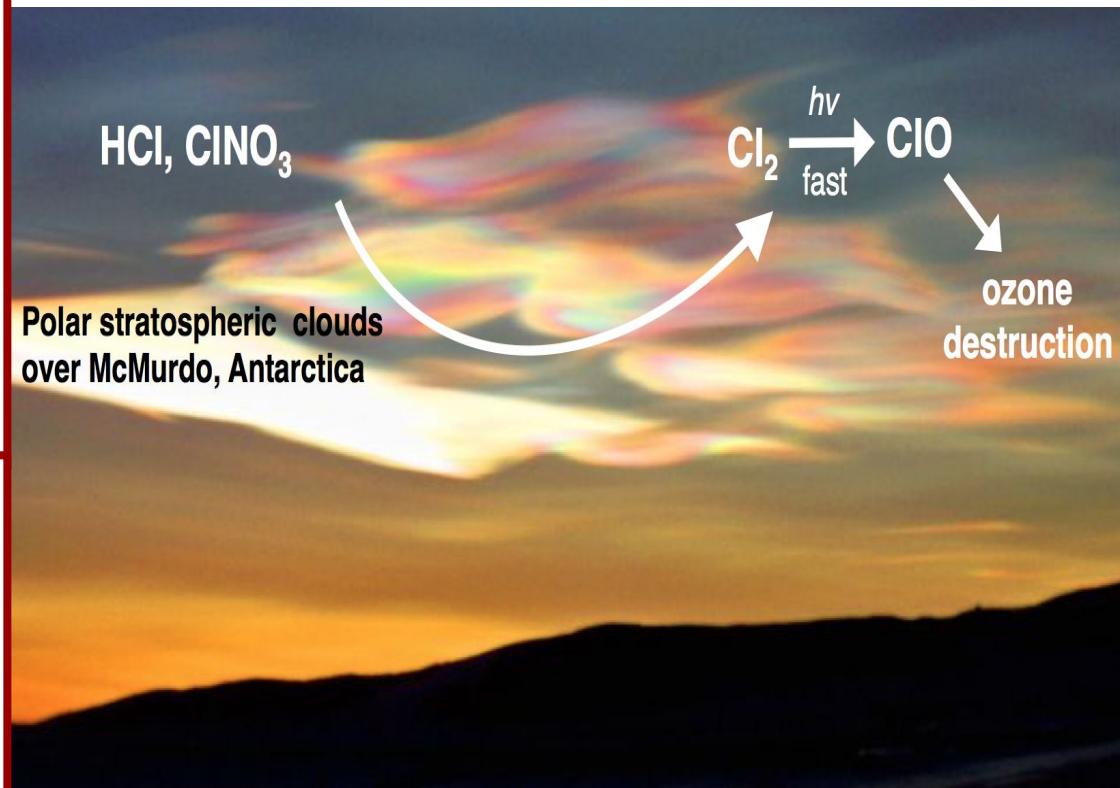
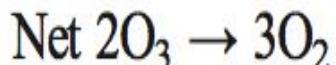
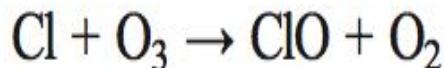
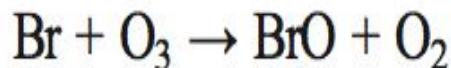
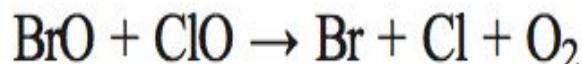
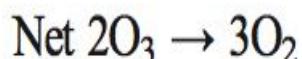
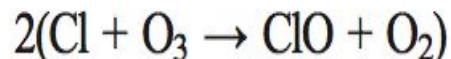
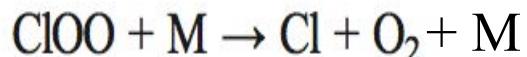
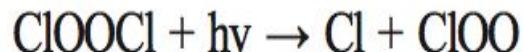
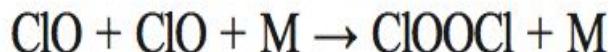
# Antarctic Ozone Hole (1st noticed 1985)



- Ground based observations clarified by NASA satellite data

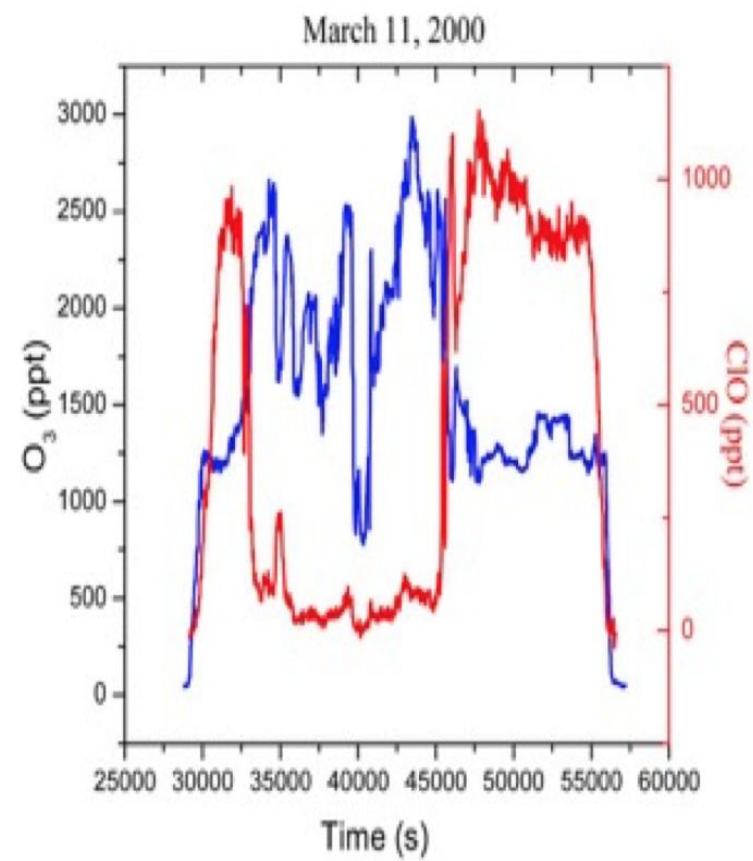


# Antarctic Ozone Hole (1st noticed 1985)





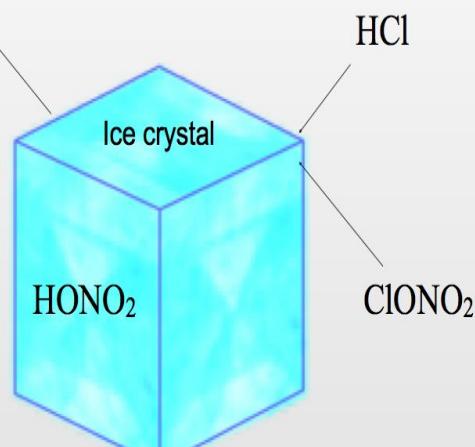
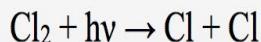
# Arctic Ozone Depletion



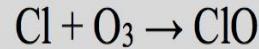
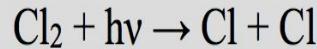
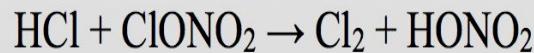


# Arctic Ozone Depletion

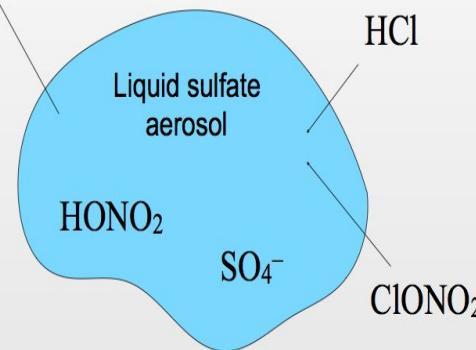
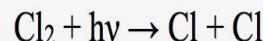
## Frozen “Polar Stratospheric Clouds”



**Result:**



## Liquid Sulfate-Water Aerosols

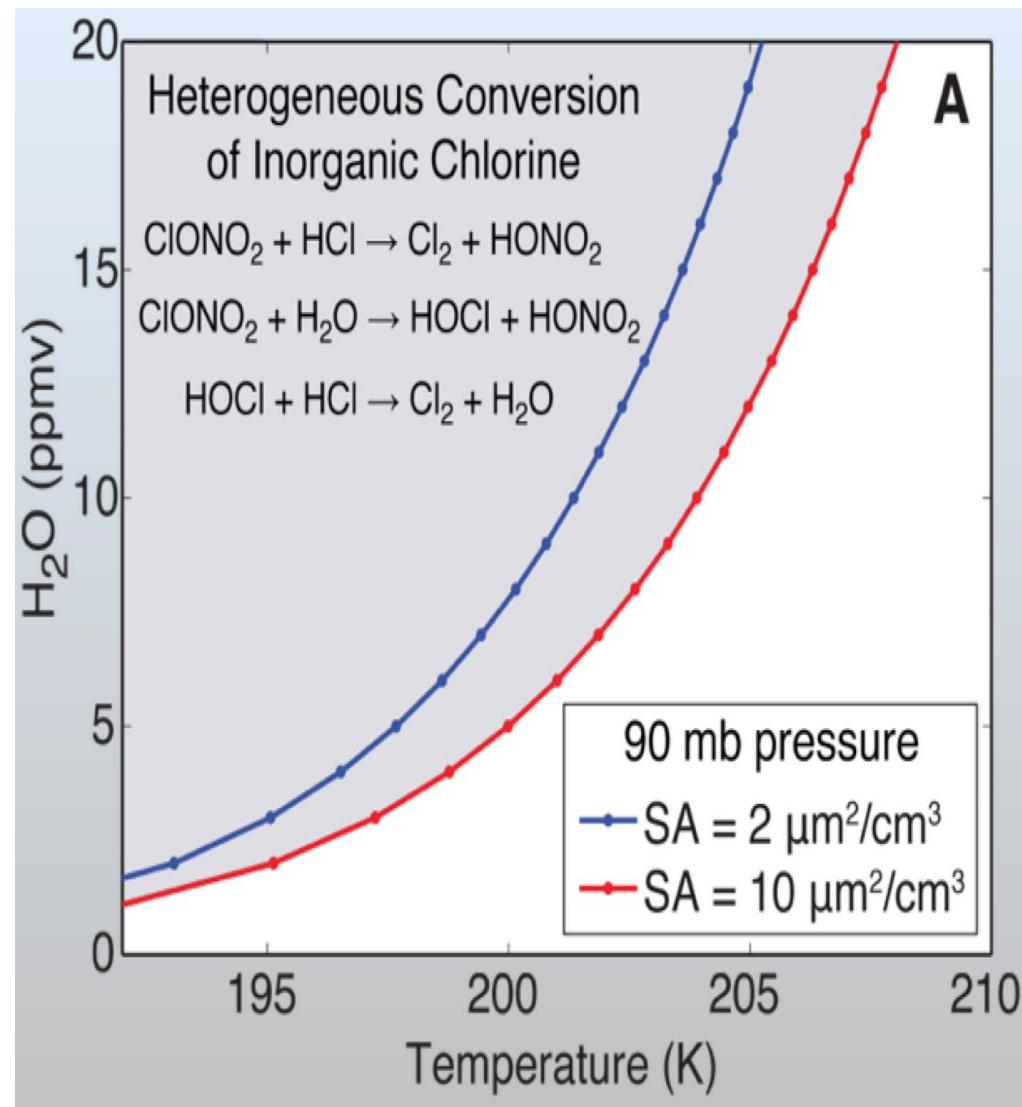


- Reaction also occurs on sulfate aerosols
- Sulfate aerosols are ubiquitous in the stratosphere



# Conditions for ClO production

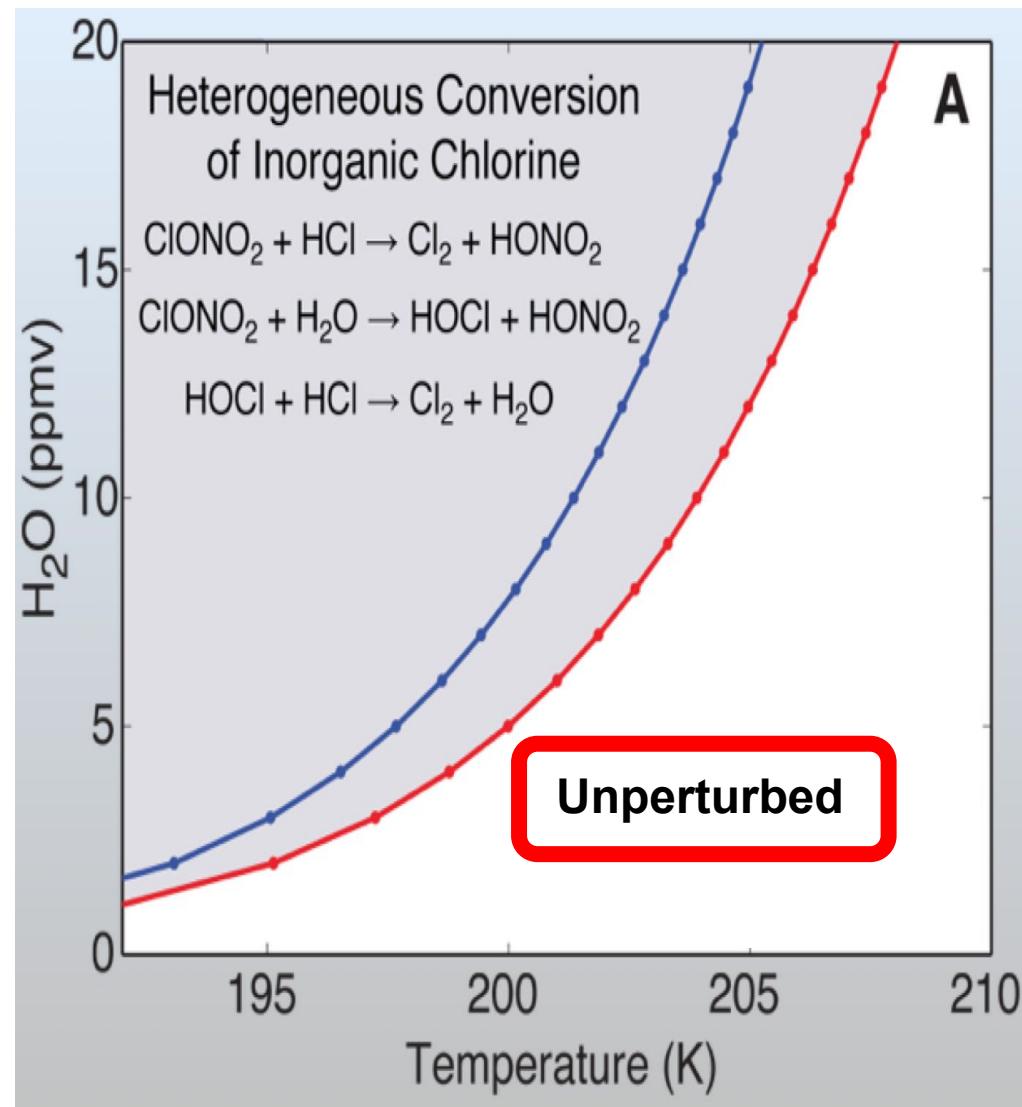
1. Sulfate Aerosols
  2. Temperature
  3. Water Vapour Concentration
- 
- Solar Radiation Management
  - Volcanoes





# Conditions for ClO production

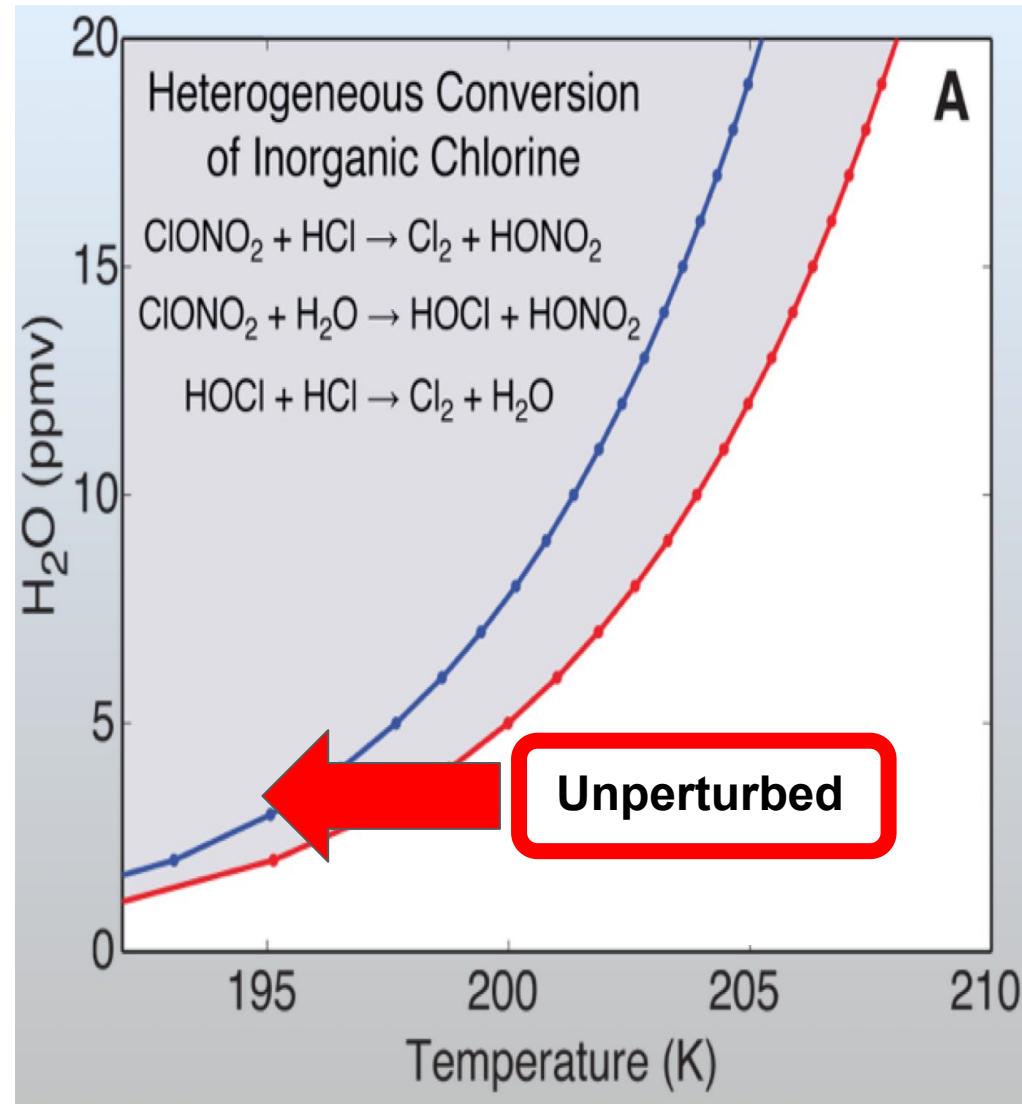
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# Conditions for ClO production

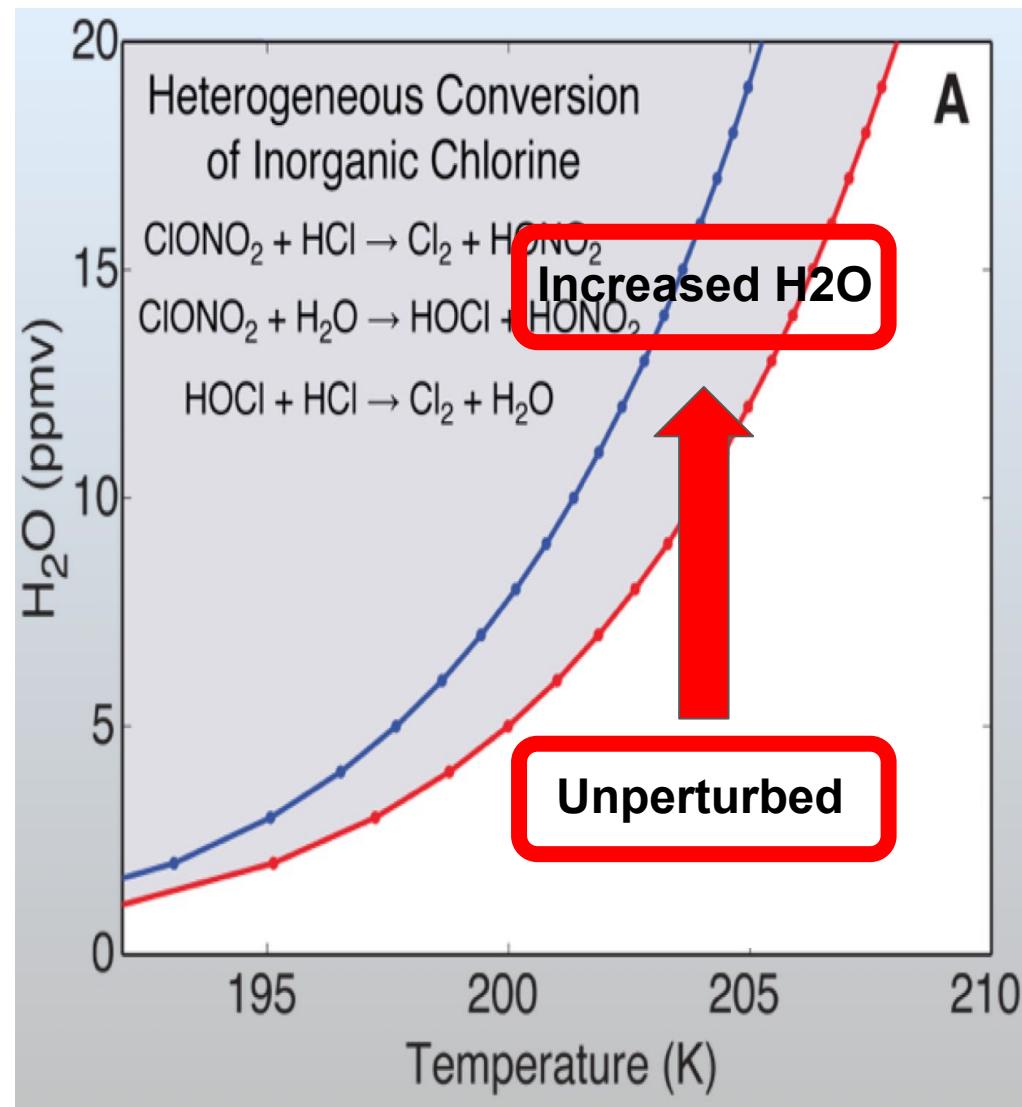
1. Sulfate Aerosols
2. Temperature
3. Water Vapour Concentration





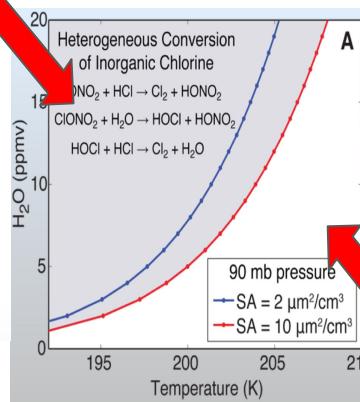
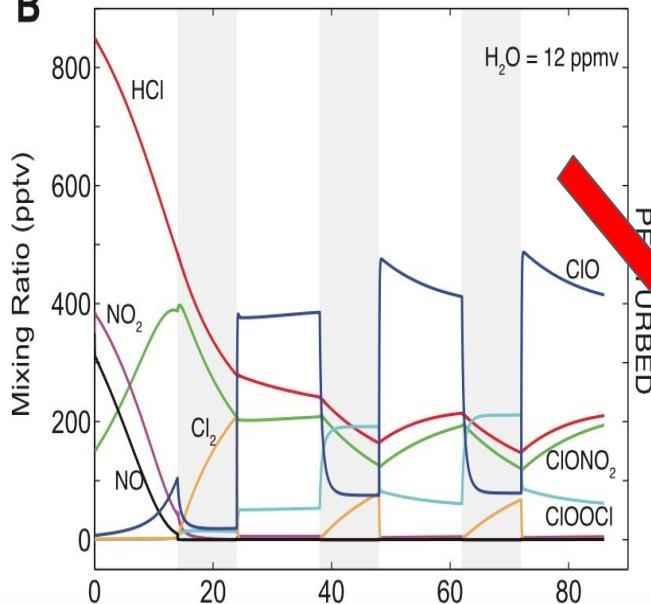
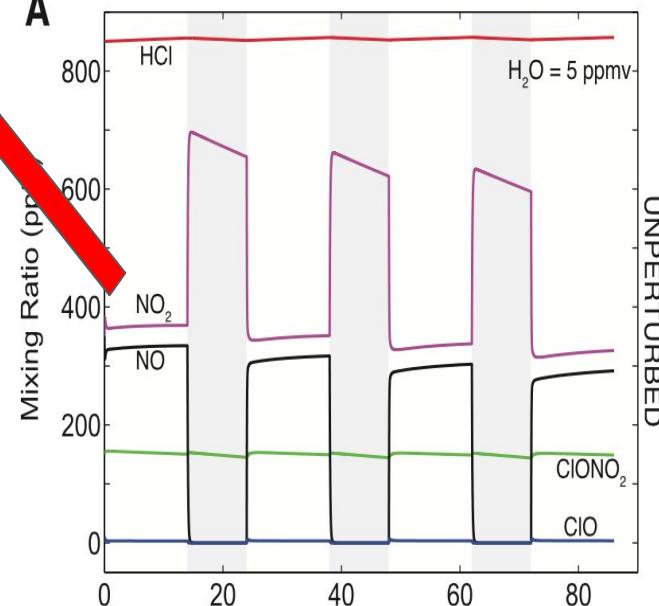
# Conditions for ClO production

1. Sulfate Aerosols
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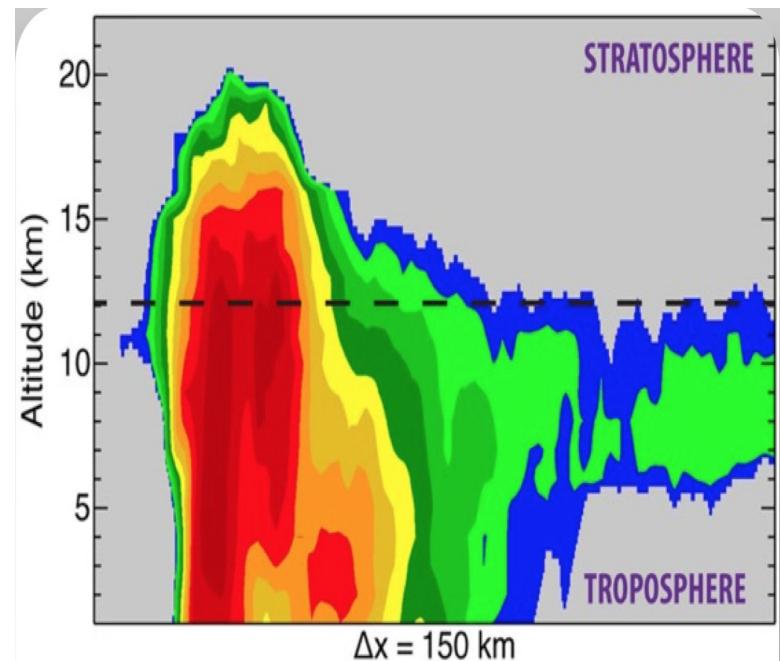
# Conditions for ClO production

**B****A**



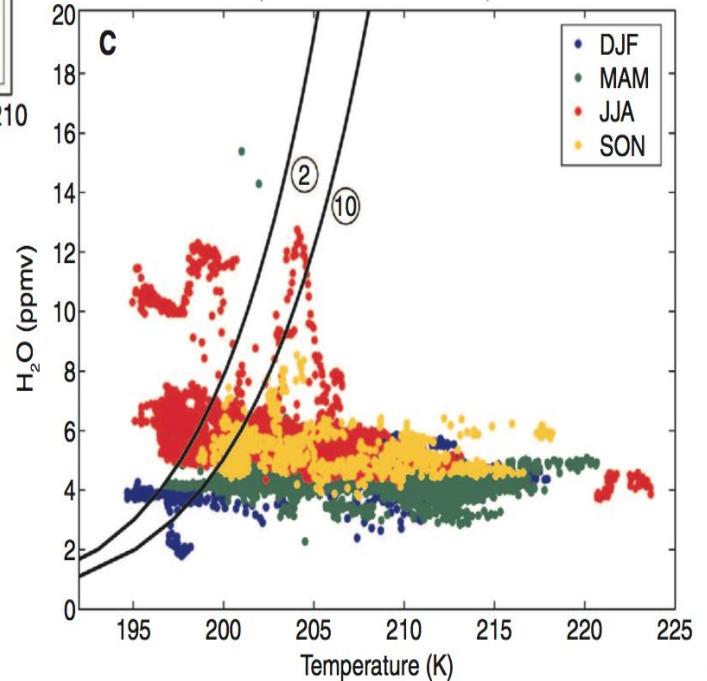
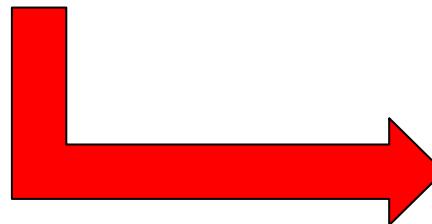
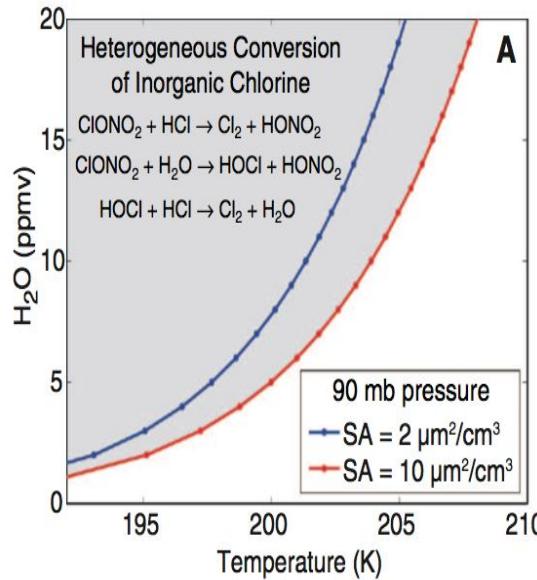
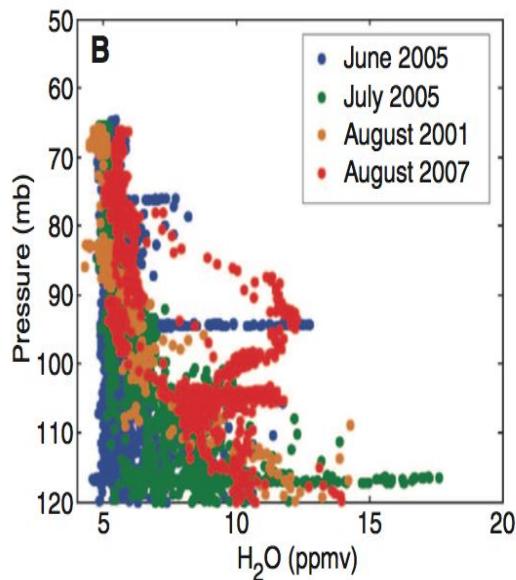
# Conditions for ClO production

1. Sulfate Aerosols
  2. Temperature
  3. Water Vapour Concentration
- 
- **Convective Storms**
  - **Anticyclonic Monsoon Flow**



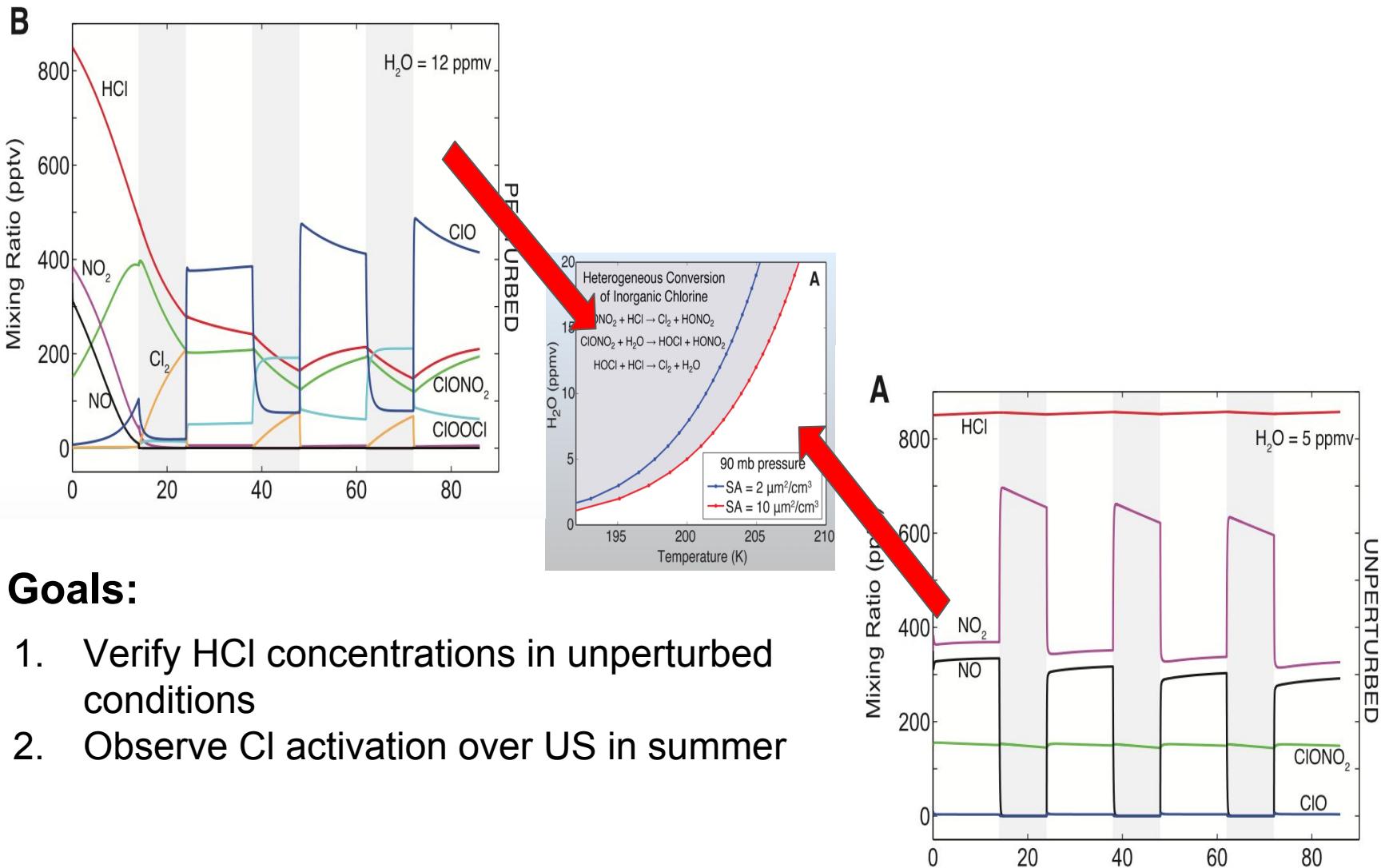


# Observations





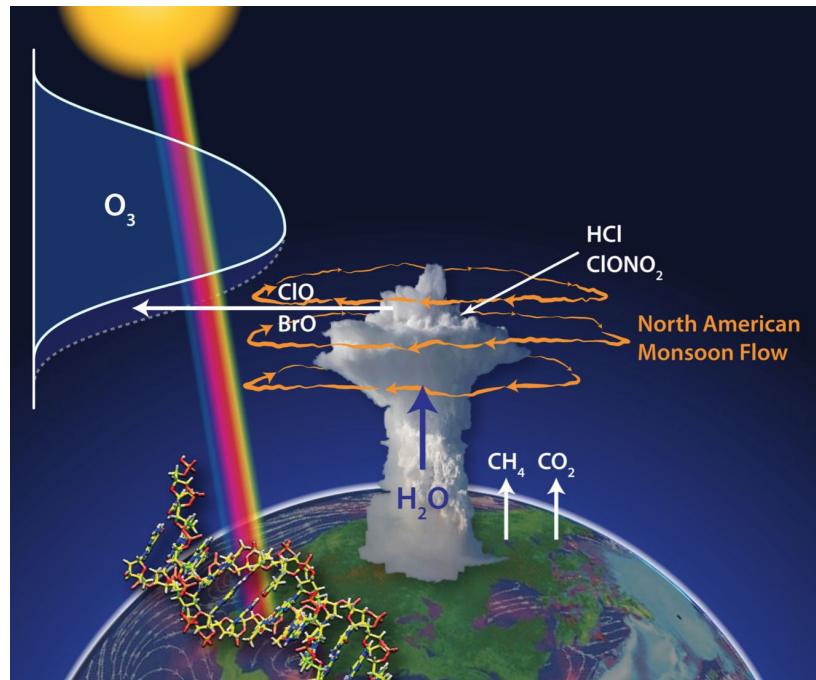
# Conditions for ClO production





# Scientific Context

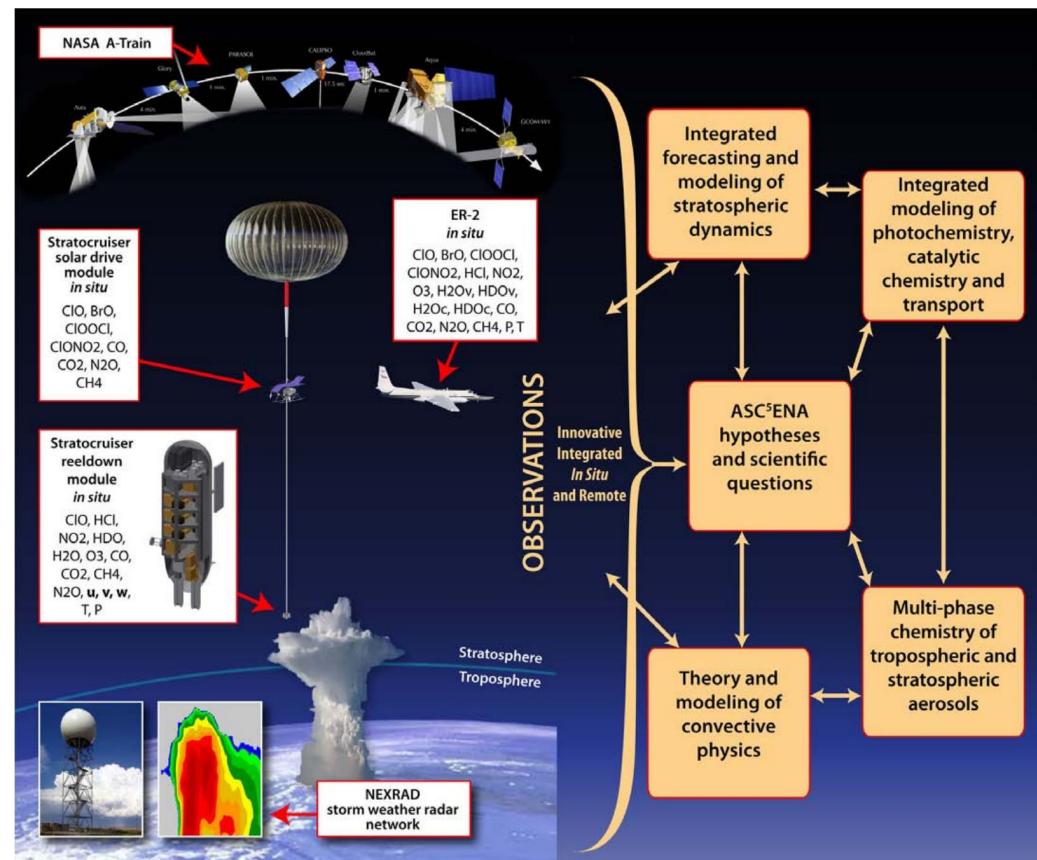
1. Ozone
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# The Proposed Mission Outline

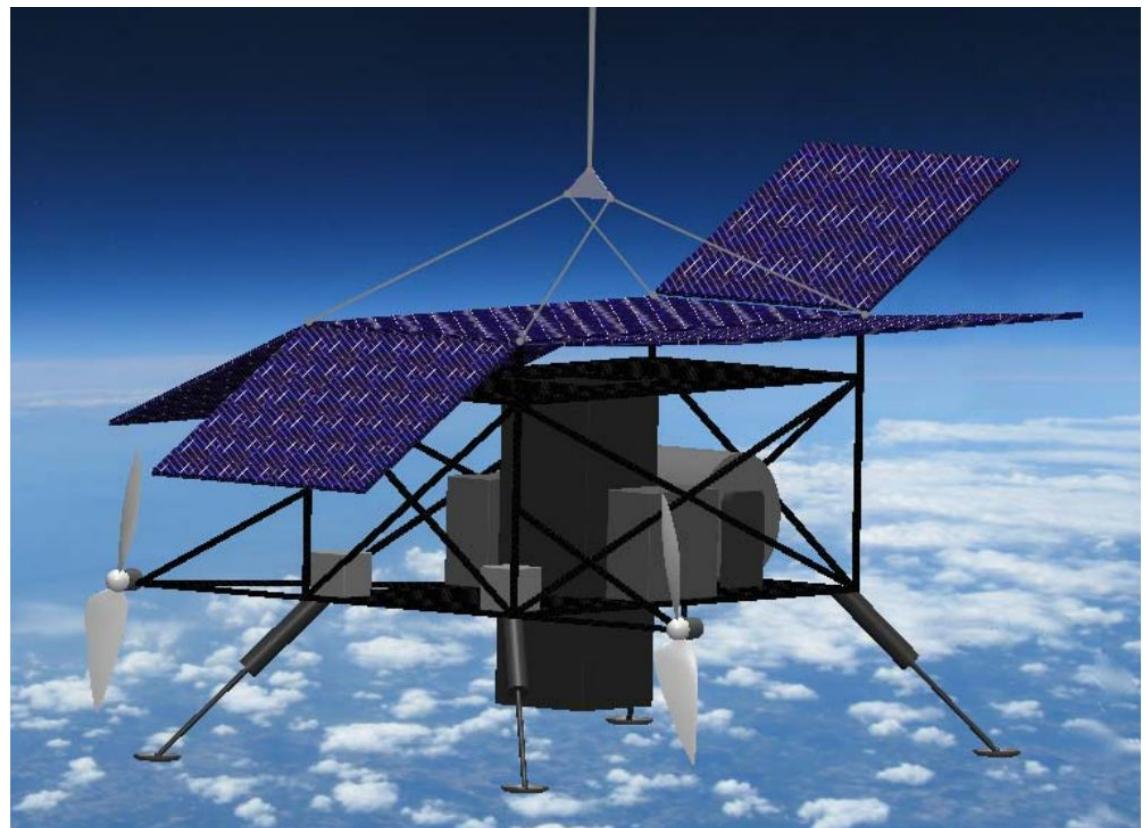
1. Long duration pumpkin balloon
2. Drive module
3. Suspended Payload
  - 4-6 week mission
  - Altitude of 20km





# Stratocruiser Drive Module

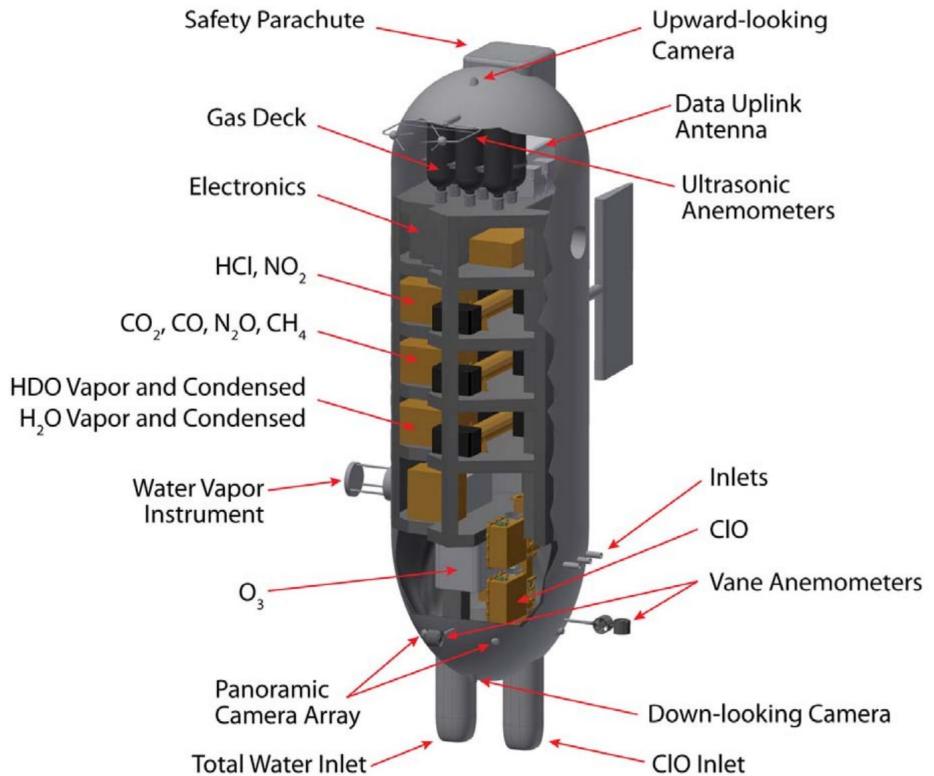
- Waypoint or heading navigation
- Range of 500km/day
- Speeds up to 8 m/s
- 5.3kWh / 2.0kWh (Day/Night) of continuous power





# Stratocruiser Suspended Payload

- **Winch system lowers payload at 5-10 m/s for up to 10km**
- **200kg mass limit**
- **Data is sent via Iridium satellite data link to the Science Control Center**





# System Overview

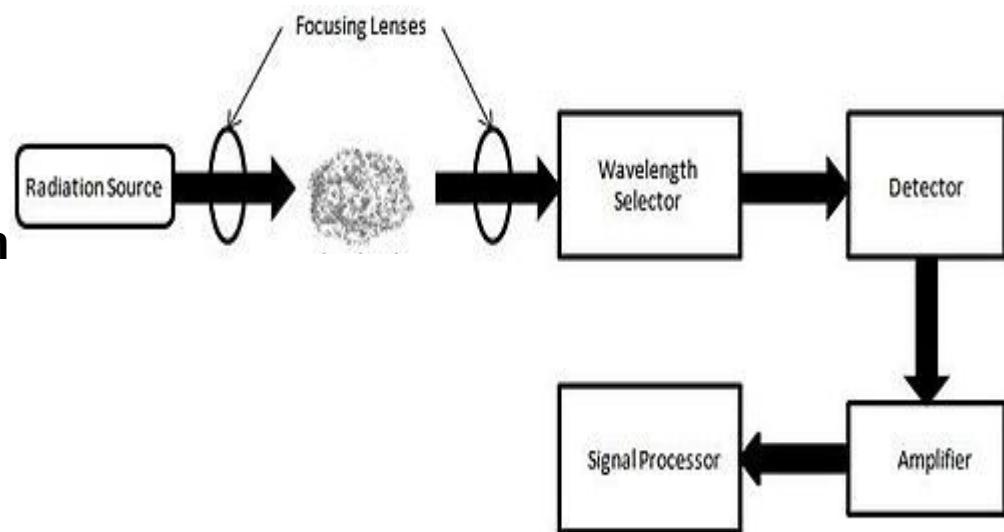
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1. Scientific Context
2. **System Overview**
3. Design Details
  - a. Laser
  - b. Cavity
  - c. Detector
  - d. Pump
  - e. System Wide Design
4. Next Steps



# Integrated Cavity Output Spectroscopy (ICOS)

- Specialized form of absorption spectroscopy
- Beer's Law:  $\ln(I/I_0) = -n\sigma\ell$ 
  - $n$  : concentration
  - $\sigma$  : absorption cross-section
  - $\ell$  : path length
- $\therefore \uparrow$  path length  $\rightarrow \uparrow$  absorption
- ICOS increases effective path length from 0.5 - 1m to >5km
- Highly sensitive - pptv



A general example of an absorption spectroscopy system



# System Diagram

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§

**Detector**



# Design Details

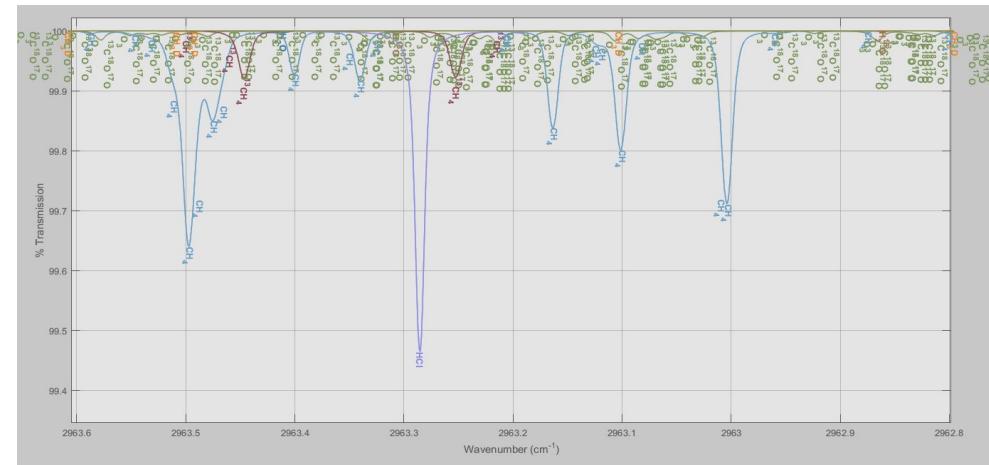
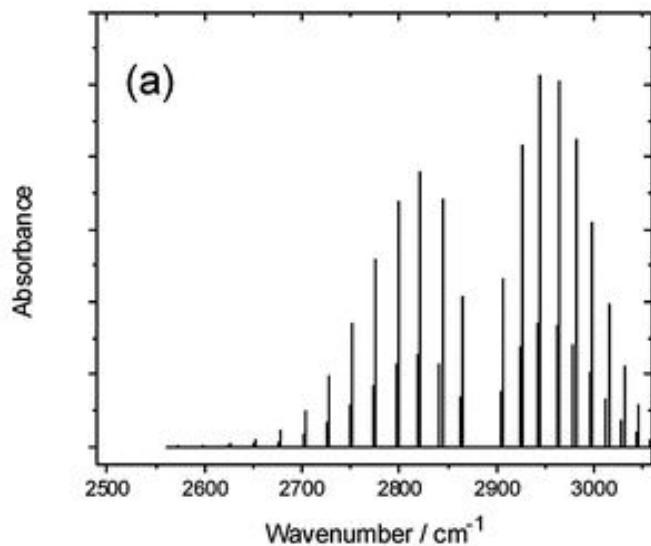
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# Laser System - Laser Specifications

- **Wavelength Choice**
  - Requires high resolution and separation from other peaks
  - Final decision:  $3.375 \mu\text{m}$  ( $2963 \text{ cm}^{-1}$ )



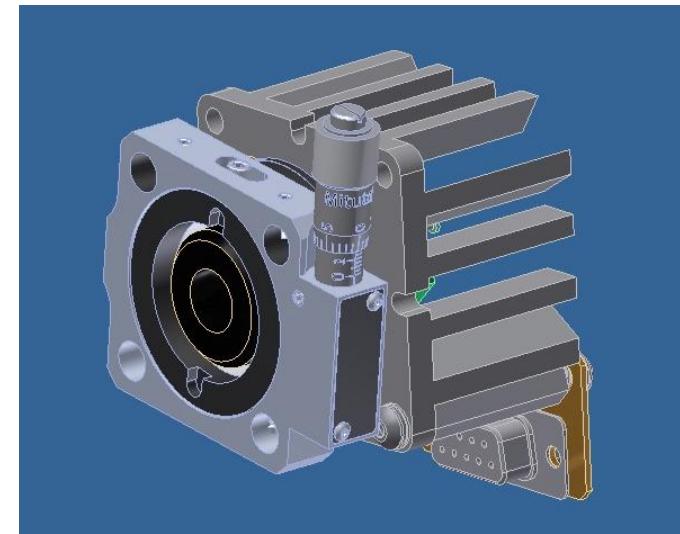
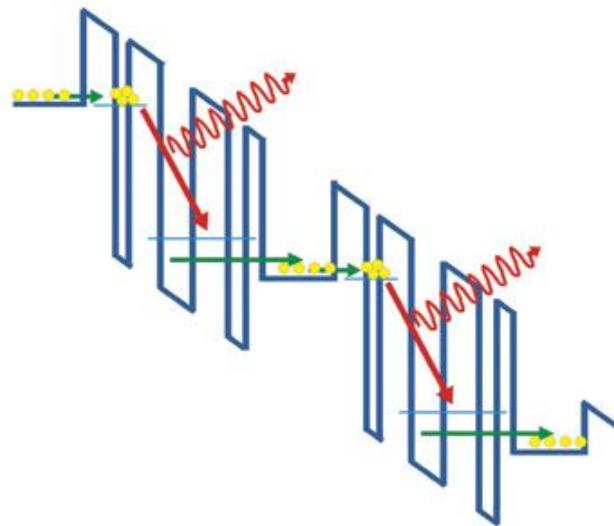


# Laser System - Laser Specifications

- **Laser Choice - Interband**

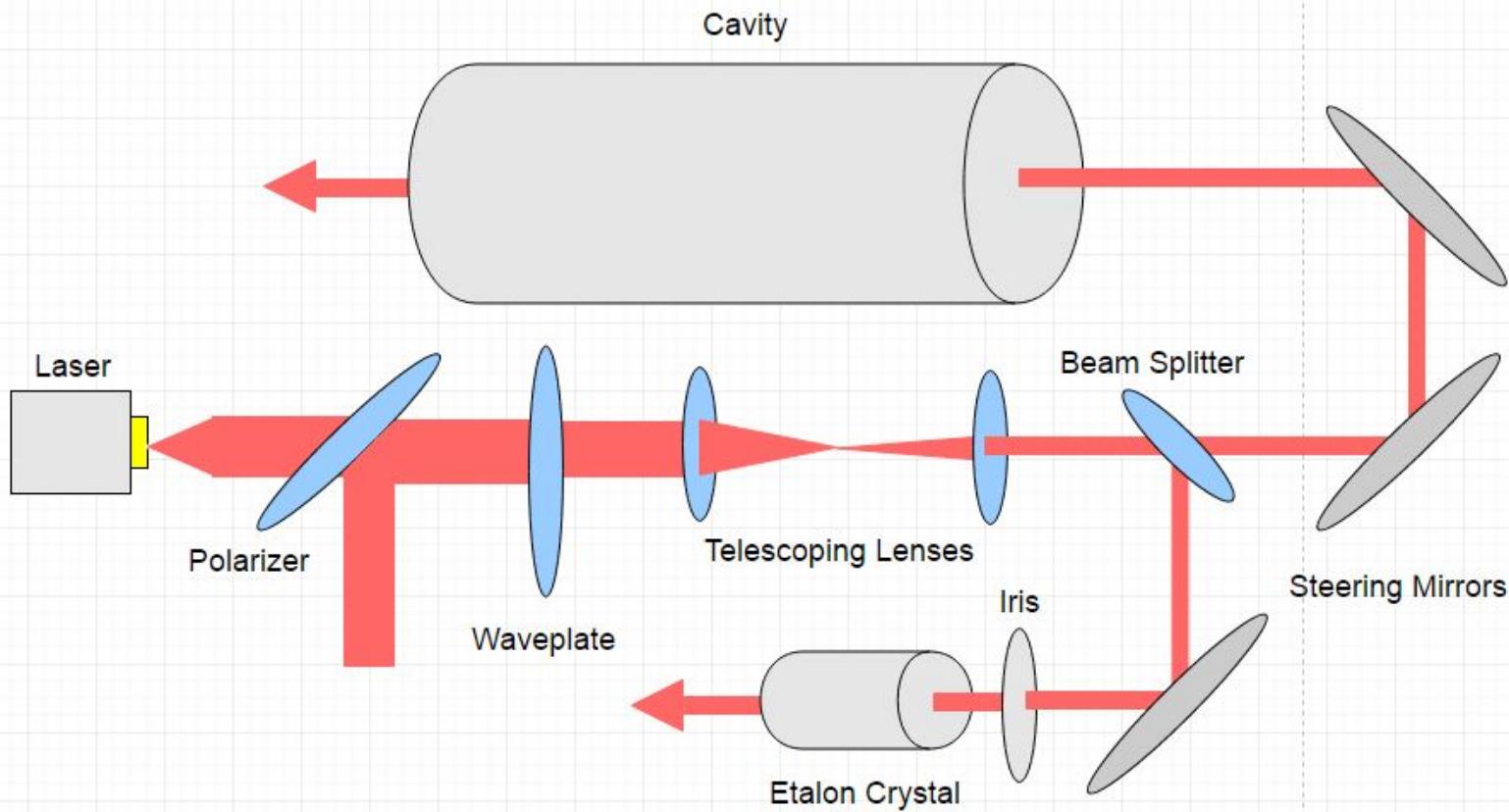
## Cascade Laser

- JPL-ICL57 for final instrument
- JPL-ICL58 for testing





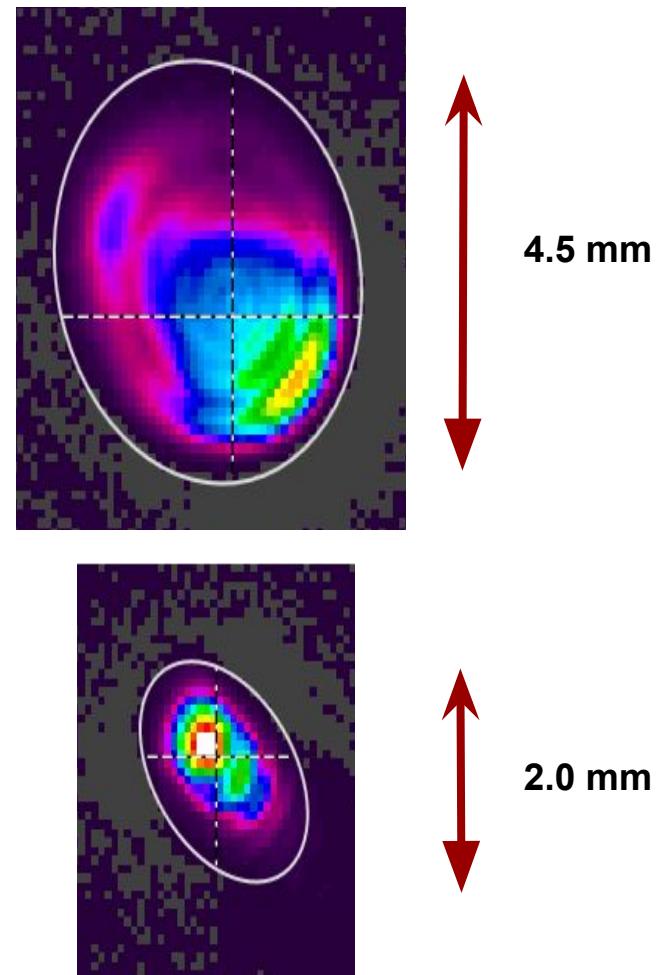
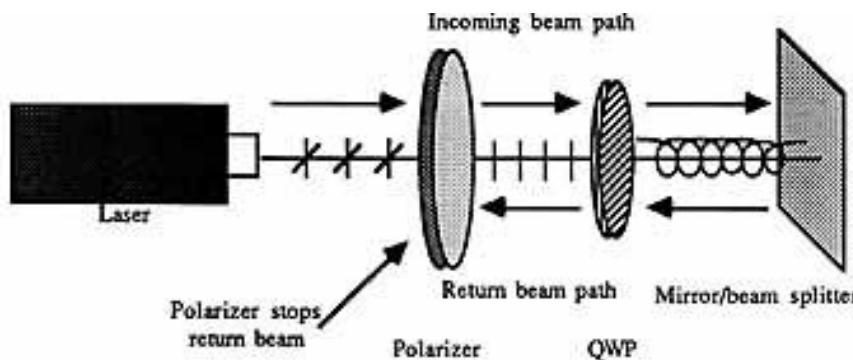
# Laser System - Fore-Optics Diagram





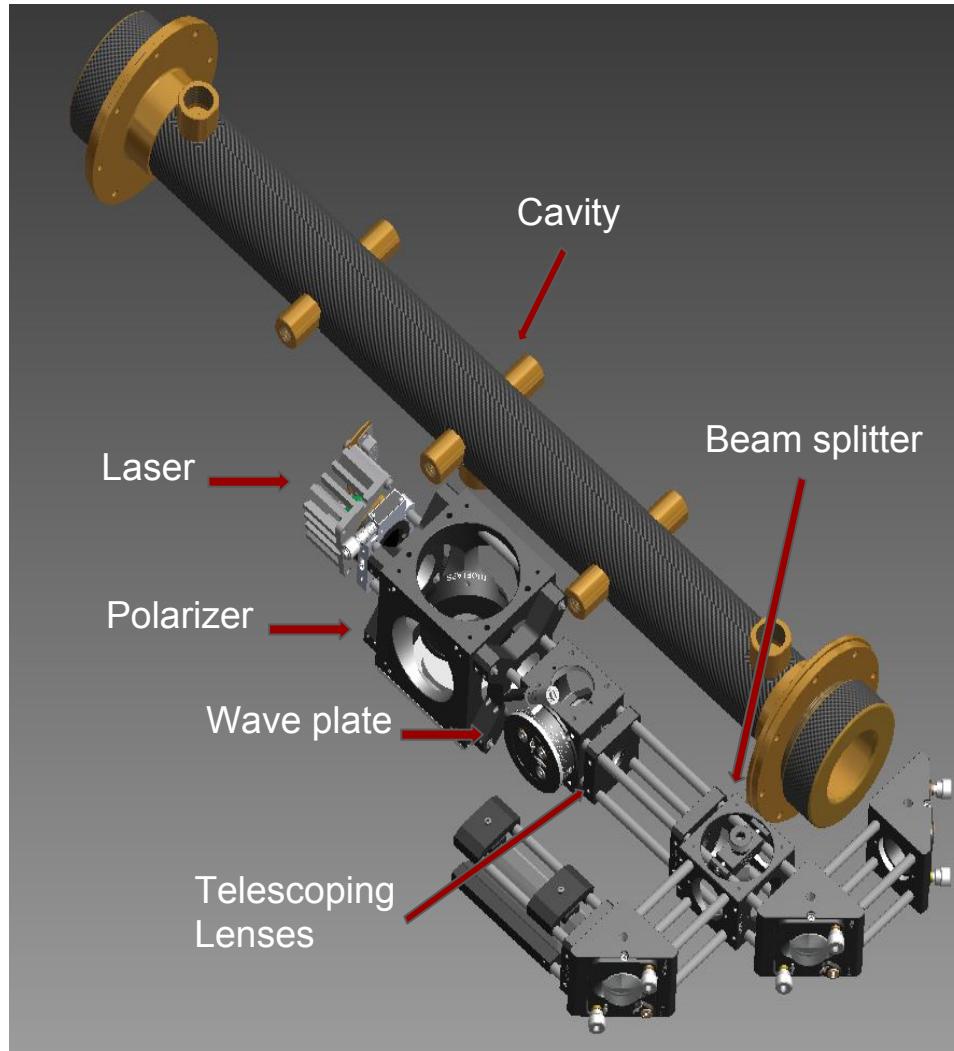
# Laser System - Fore-Optics Specifications

- Reduced interference into laser with wave plate and polarizer
- Reduced beam diameter for greater resolution in cavity

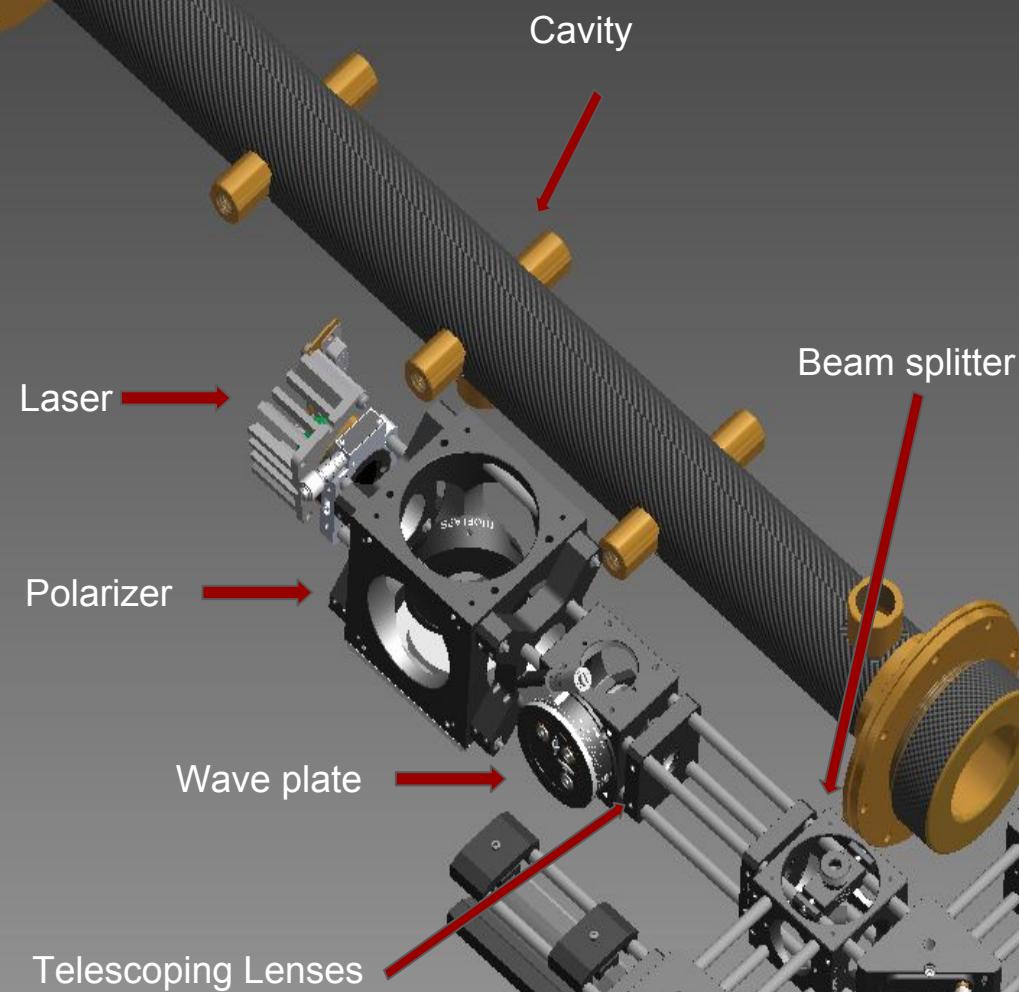




# Laser System - Fore-Optics CAD Model



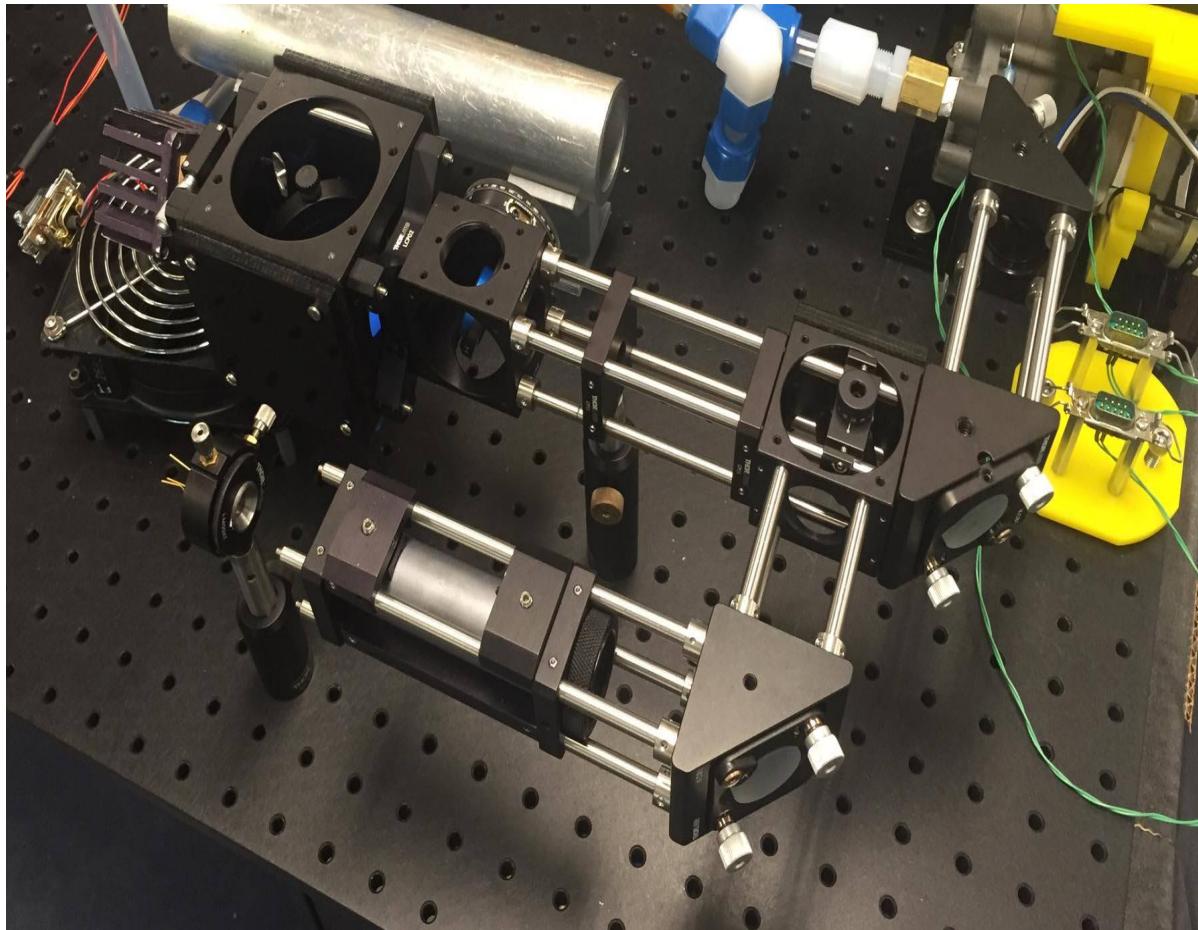
- **Modular cage system for easy assembly and change of components**
- **Individual rotation mounts for precision alignment control**
- **Stainless steel cage and single-point attachment to cavity for in-flight durability**





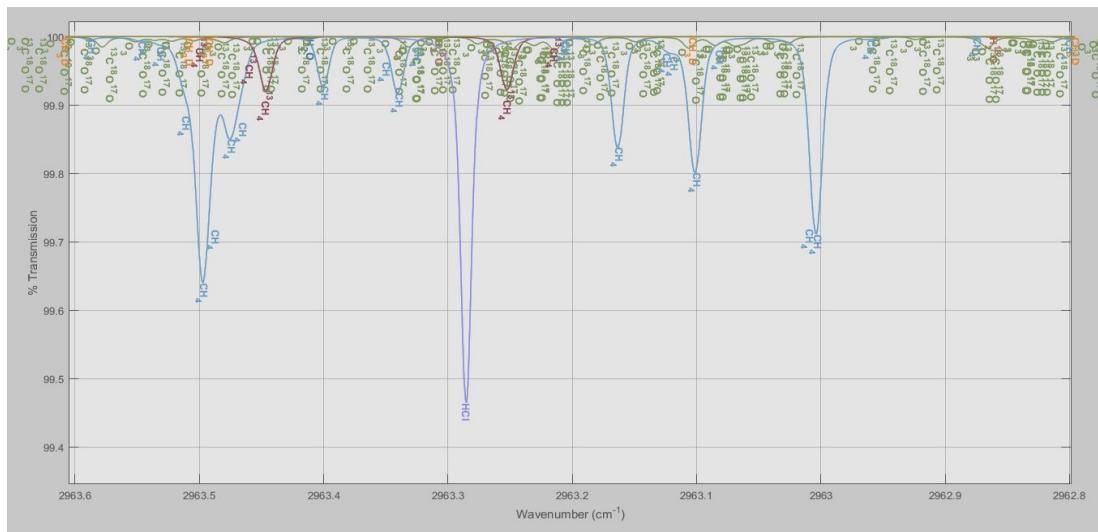
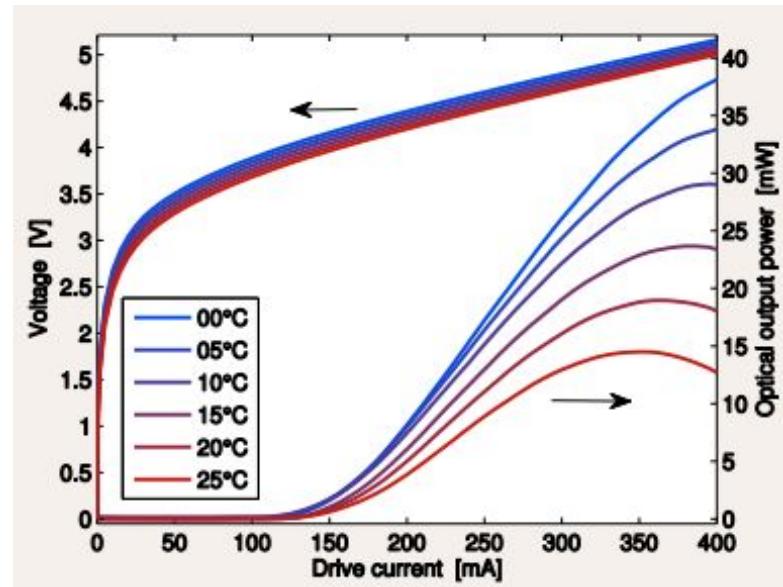
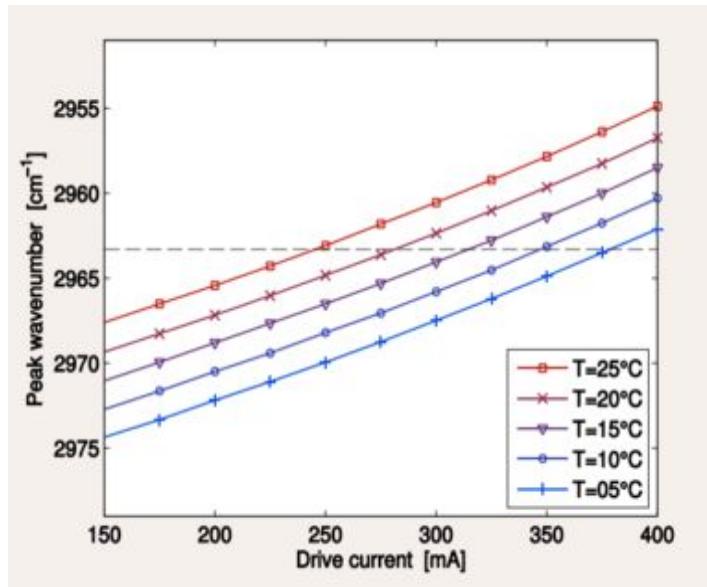
# Laser System - Fore-Optics Assembly

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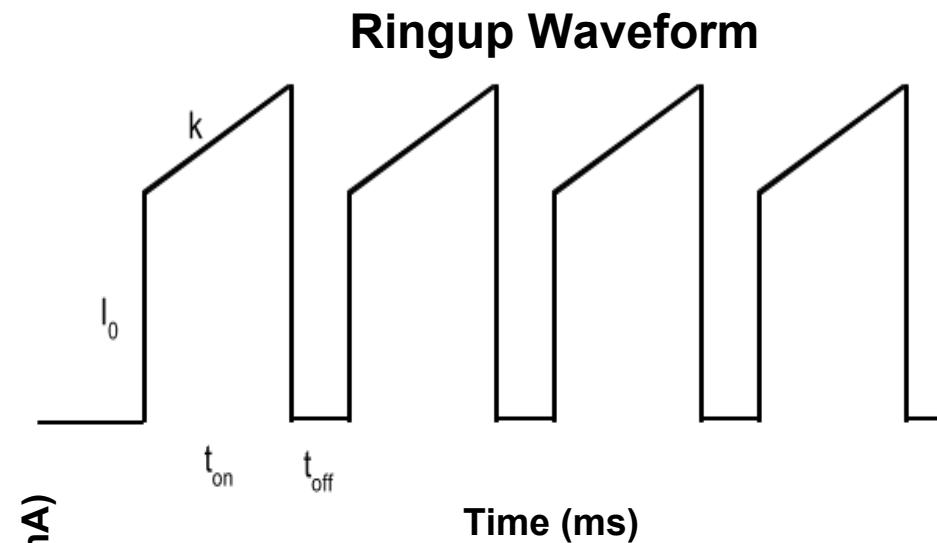
# Laser System - Laser Driver Overview



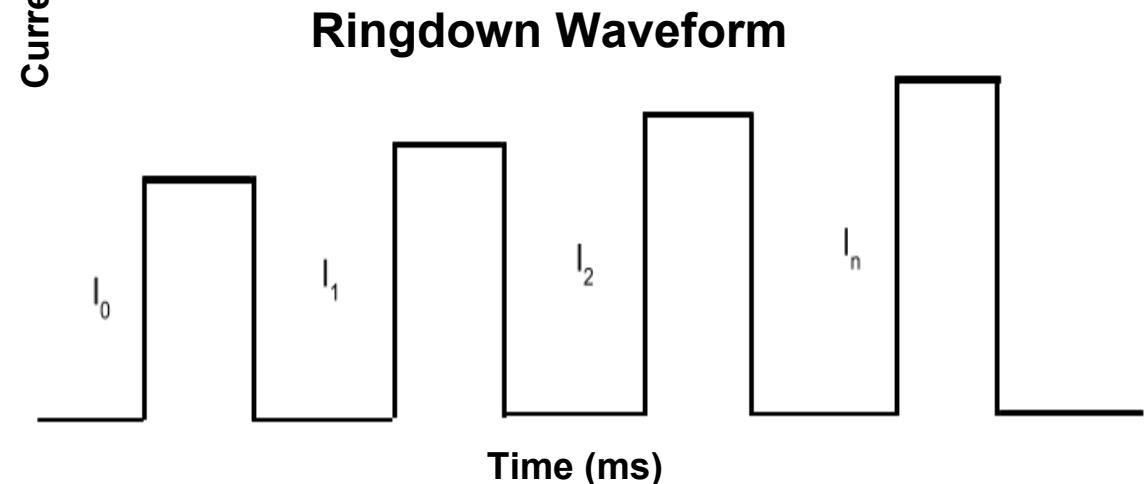


# Laser System - Driver Waveforms

- Programmed by FPGA, and generated by laser driver.
- Current waveform sweeps across laser wavelengths.
- Ringdown waveform operates laser at single wavelengths.

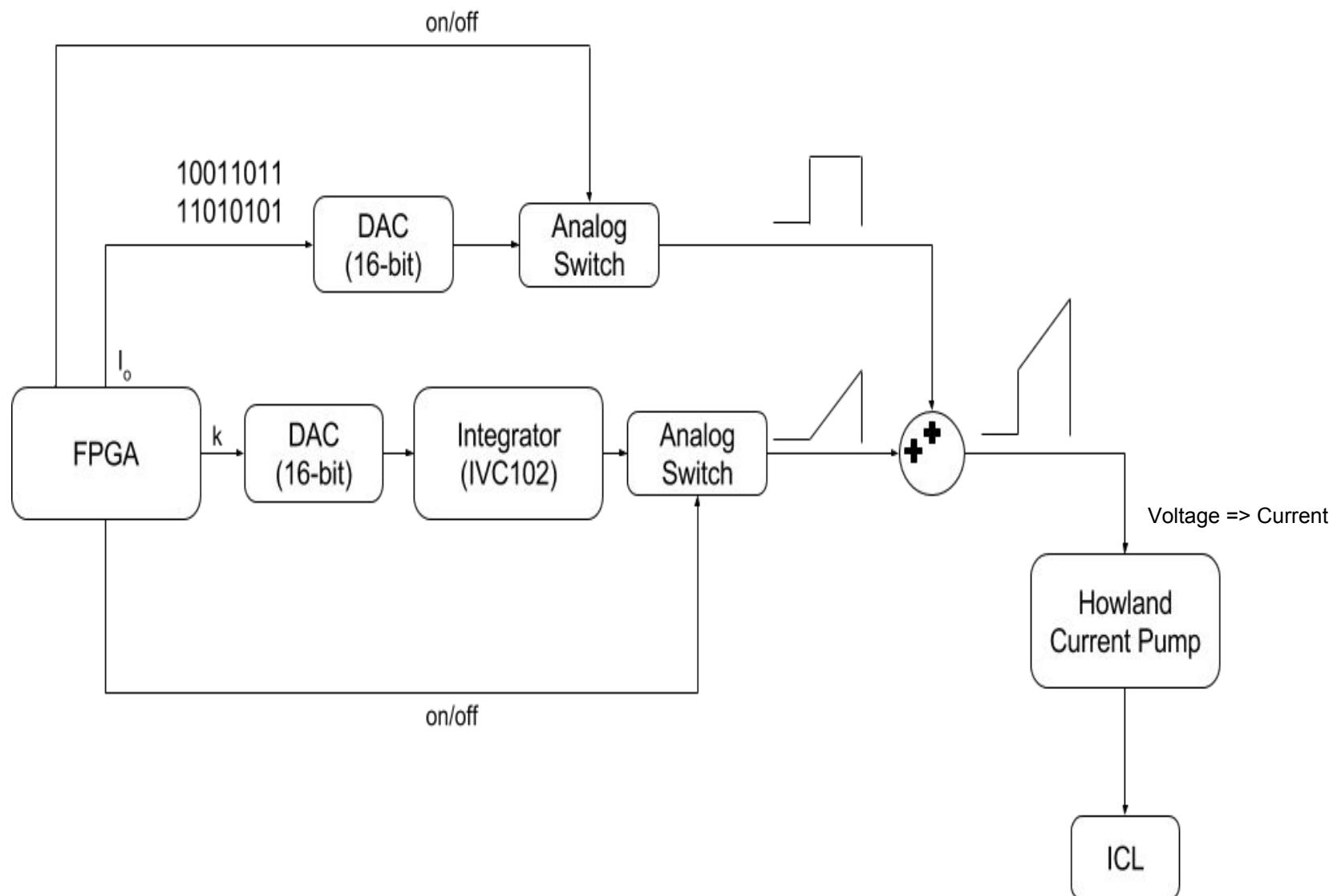


- Frequency: 1kHz
- Rise/Fall Time: <1us
- ~50% duty cycle - ringdown waveform



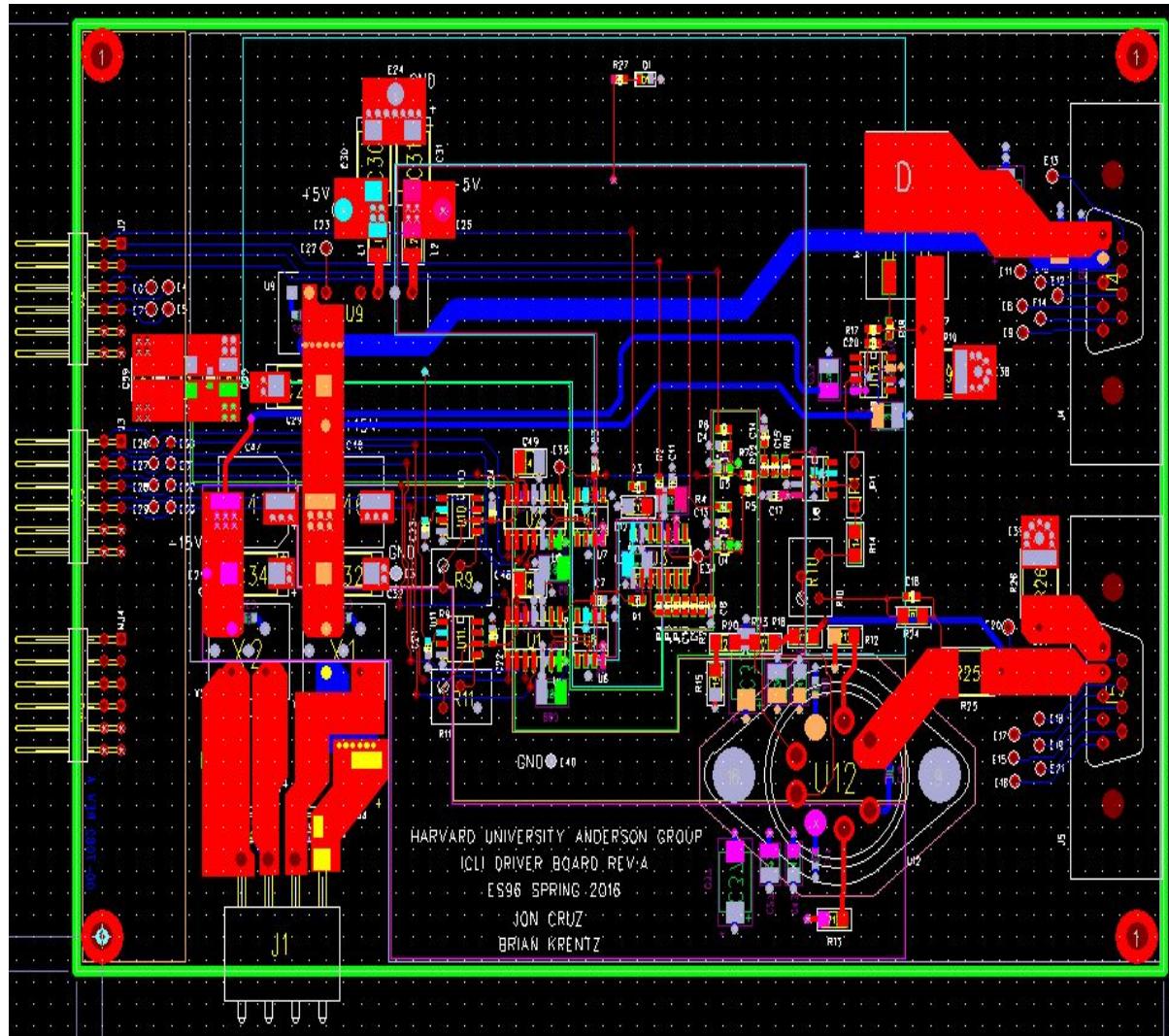


# Laser System - Driver Block Diagram





# Laser System - Driver Board





# Cavity

---

1. Scientific Context
2. System Overview
3. Design Details
  - a. Laser
  - b. Cavity**
  - c. Detector
  - d. Pump
  - e. System Wide Design
4. Next Steps



# System Diagram - Cavity

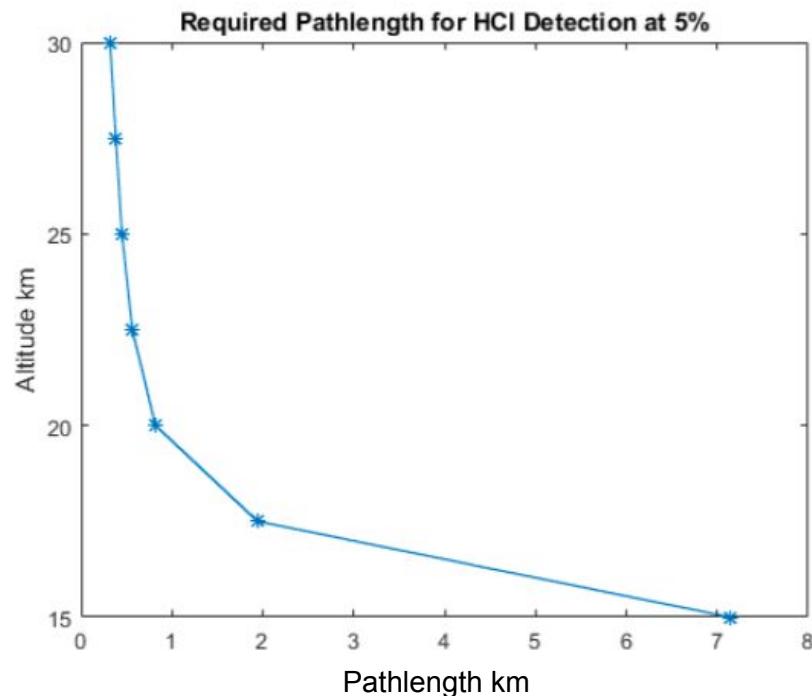
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# Cavity - Objectives

- **House the atmosphere sample in an airtight, thermally uniform environment**
- **Maintain relatively low cell volume**
- **Generate effective pathlength of several kilometers**



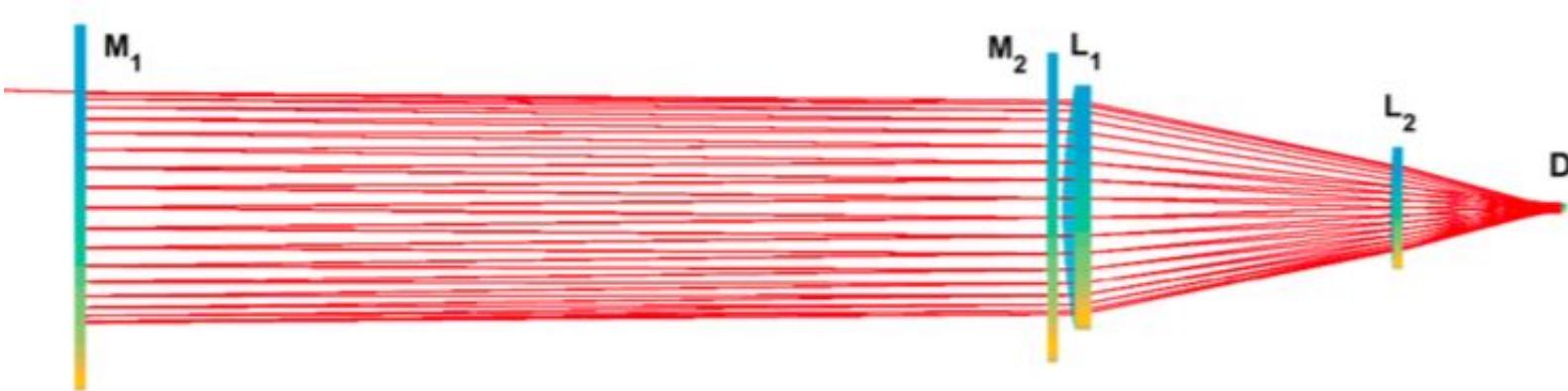


# Cavity - Optical configuration

- Integrated cavity output spectroscopy (ICOS)
- Two highly reflective mirrors
- Series of post cavity lenses focused onto a detector

$$\ell' \cong \frac{\ell}{L + \alpha}$$

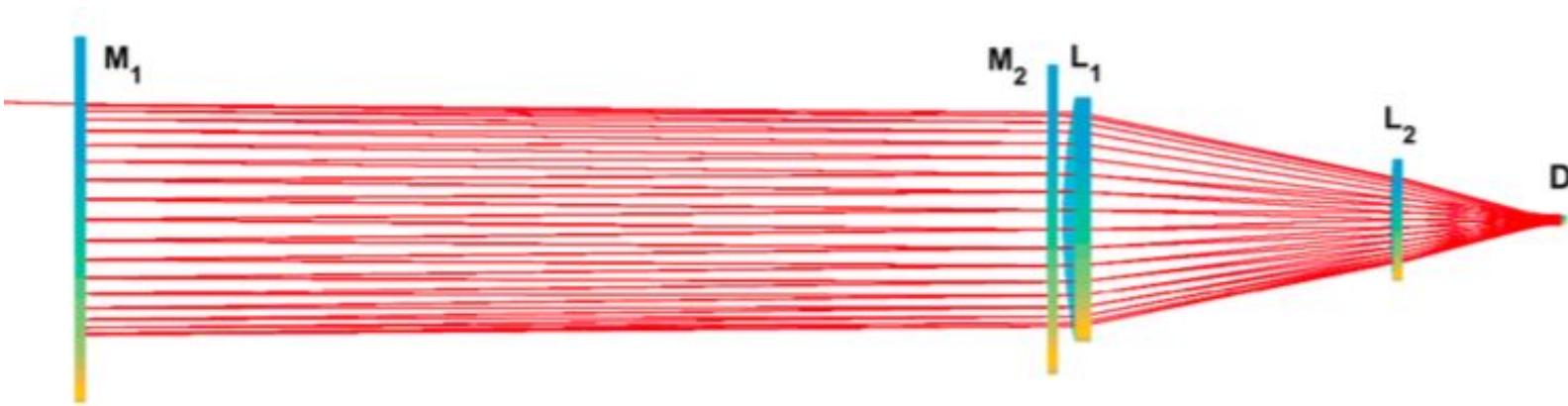
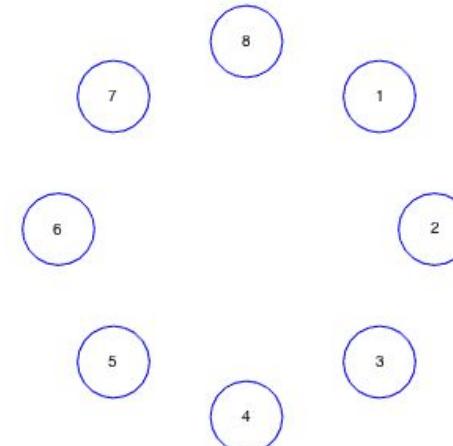
$$L = 1 - e^{-\ell/c\tau} \cong \ell/c\tau$$





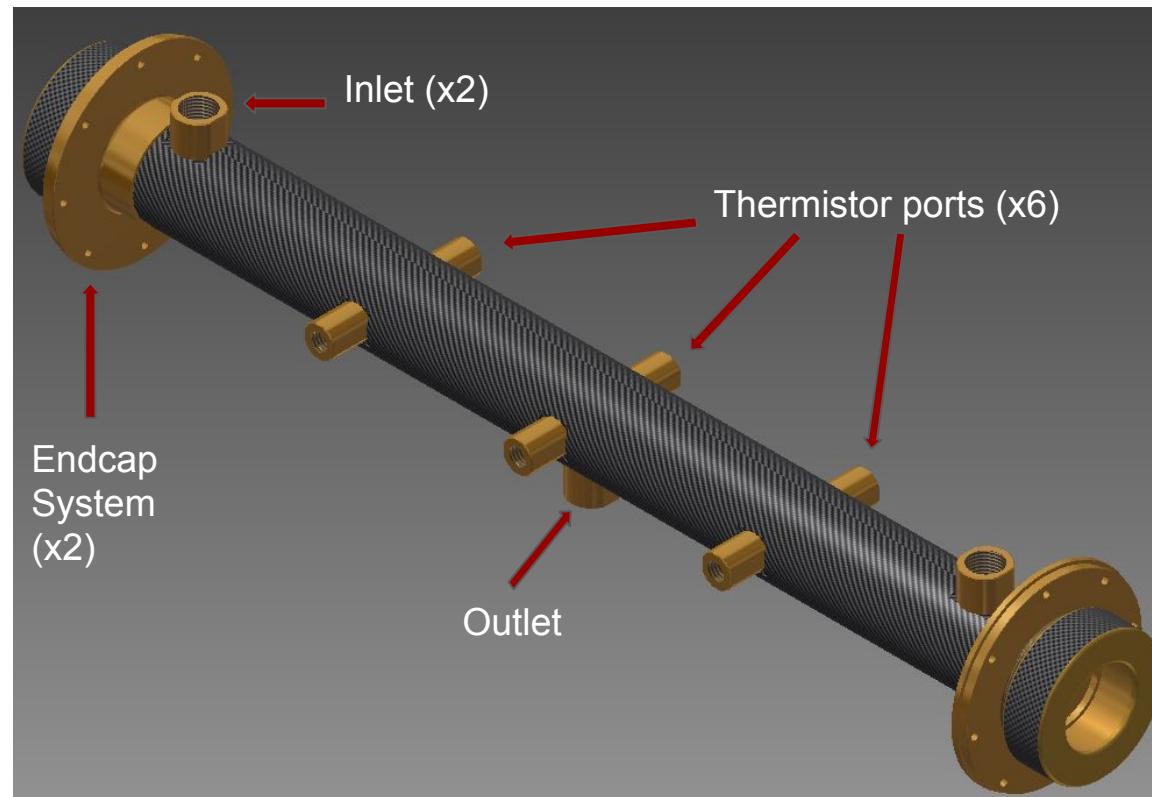
# Cavity - Optical configuration

- Material: Zinc Selenide
- Mirror Loss: 60 - 1000 ppm
- $M_1$ : 200 cm; 300cm
- $M_2$ : 1000 cm; 400 cm
- Mirror diameter: 1.5in; 2in
- Volume: < 1 liter





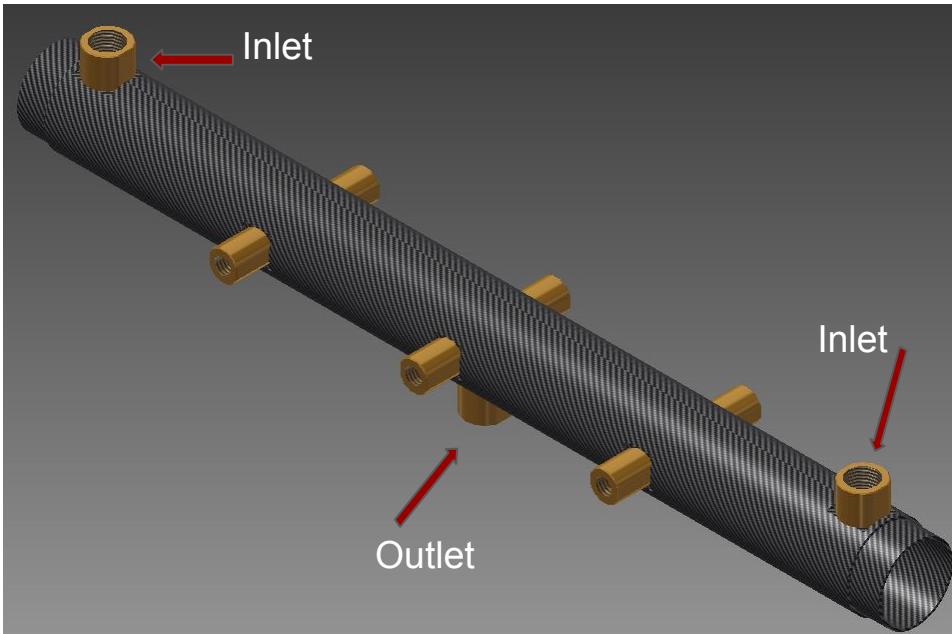
# Cavity - Structure



- **Carbon-Fiber tube**
  - **Low mass**
  - **Low volume**
- **Cryogenic Epoxy**
  - **Non-reactive with HCl**
  - **-120° F to 120° F range**
- **T and P data for sample**



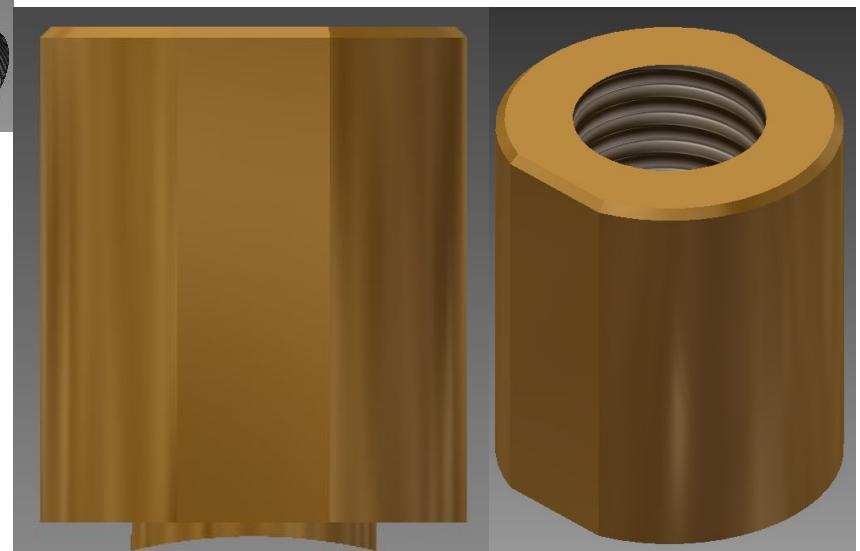
# Cavity - Sensors & Ports



Thermistor boss

## Cavity assembly:

- 2 inlets
- 1 outlet
- 6 in-cavity thermistor ports

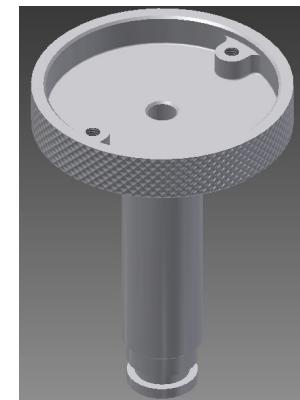




# Temperature & Pressure Sensor

- **9 total thermistors in cavity**
  - NTC  $100\text{K}\Omega$  1% bead
  - 6 along length of cavity
  - 1 at each inlet and outlet
- **1 thermistor on surface of cavity**
  - Acts as temperature control for 2 thermofoil heaters
- **1 pressure sensor located at outlet**
  - Absolute pressure transducer, 0-5Vdc output, cable connection
  - Supply voltage: 10-30Vdc @ 10mA

Thermistor housing



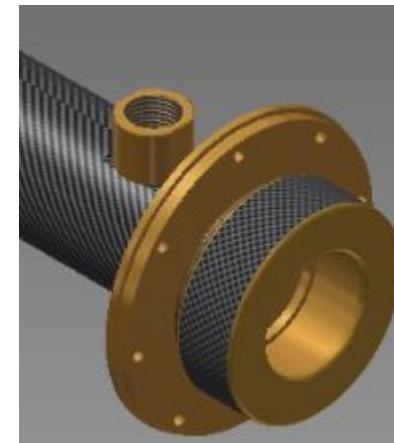
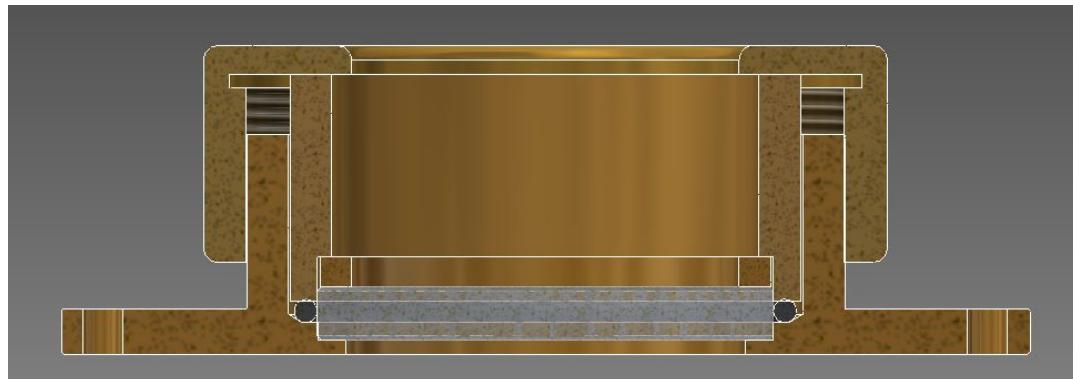
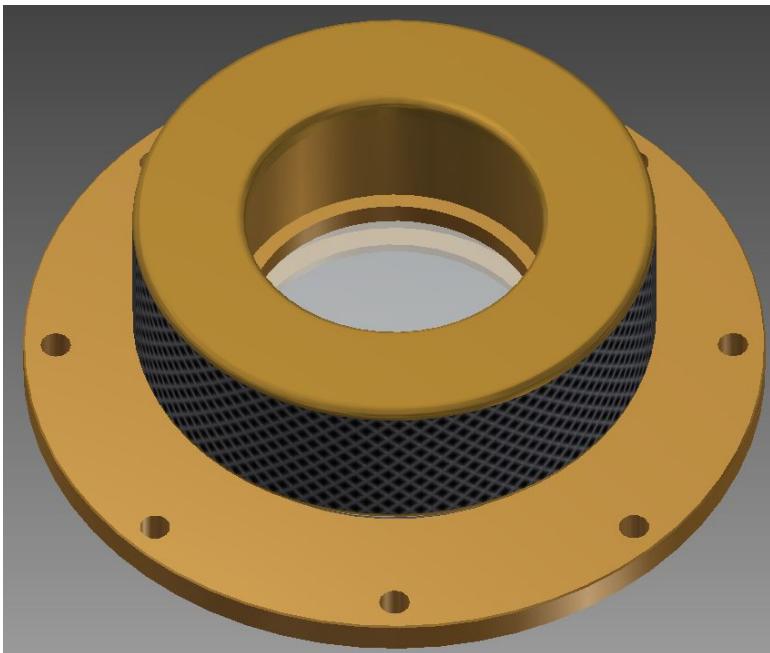
Pressure probe





# End Cap System

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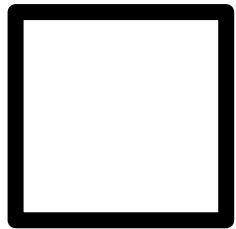
# Detector

---

1. Scientific Context
2. System Overview
3. Design Details
  - a. Laser
  - b. Cavity
  - c. **Detector**
  - d. Pump
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4. Next Steps



# System Diagram - Detector(s)





# ICOS Detector Subsystem Diagram

Cooled Detector

- Cooled Detector subsystems
  - Lenses
  - Detector + Cooling (integrated)
  - Pre-Amp
  - Heat sink/fan

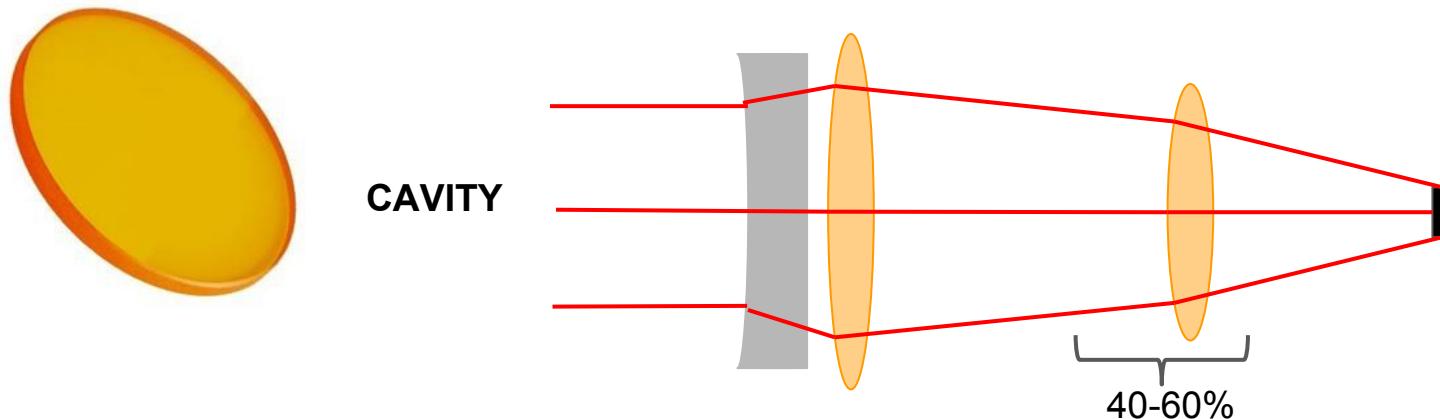
- Etalon Detector subsystems
  - Etalon Crystal
  - Detector (uncooled)
  - Pre-Amp



# Post-Cavity Lens Selection

We selected three post-cavity lenses for more than a dozen potential configurations.

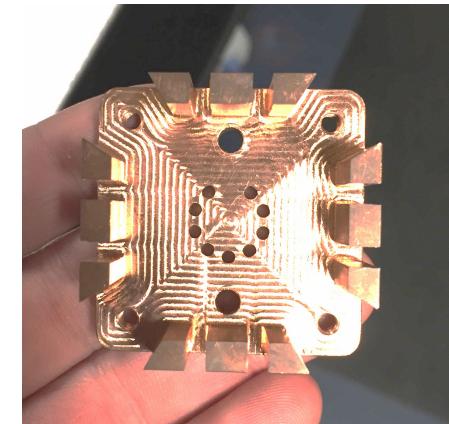
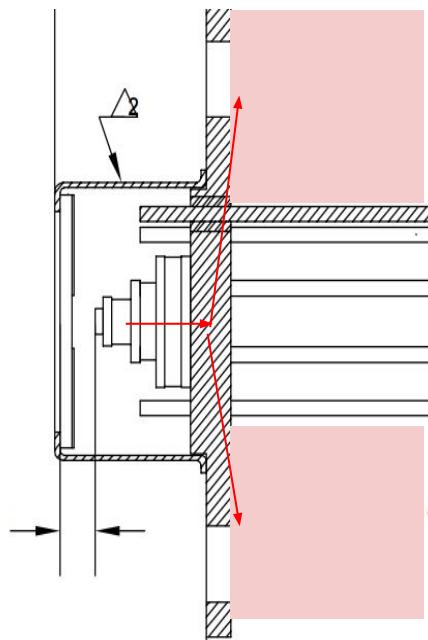
- ZC-PM-25-150
  - Zinc Selenide
  - Positive Meniscus
  - 25.4 mm diameter
  - 150 mm focal length
- ZC-PM-38-200
  - Zinc Selenide
  - Positive Meniscus
  - 38.1 mm diameter
  - 203 mm focal length
- ZC-PM-38-63
  - Zinc Selenide
  - Positive Meniscus
  - 38.1 mm diameter
  - 63.5 mm focal length





# Detector Thermal Control

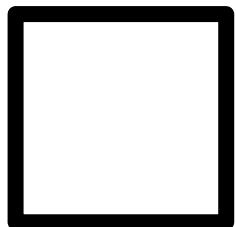
- Material
  - Copper
    - High Thermal Conductivity
    - Machinability
- Heat exchange
  - TEC
  - Overall Container
  - Detector Operating Temp
    - -75 °C





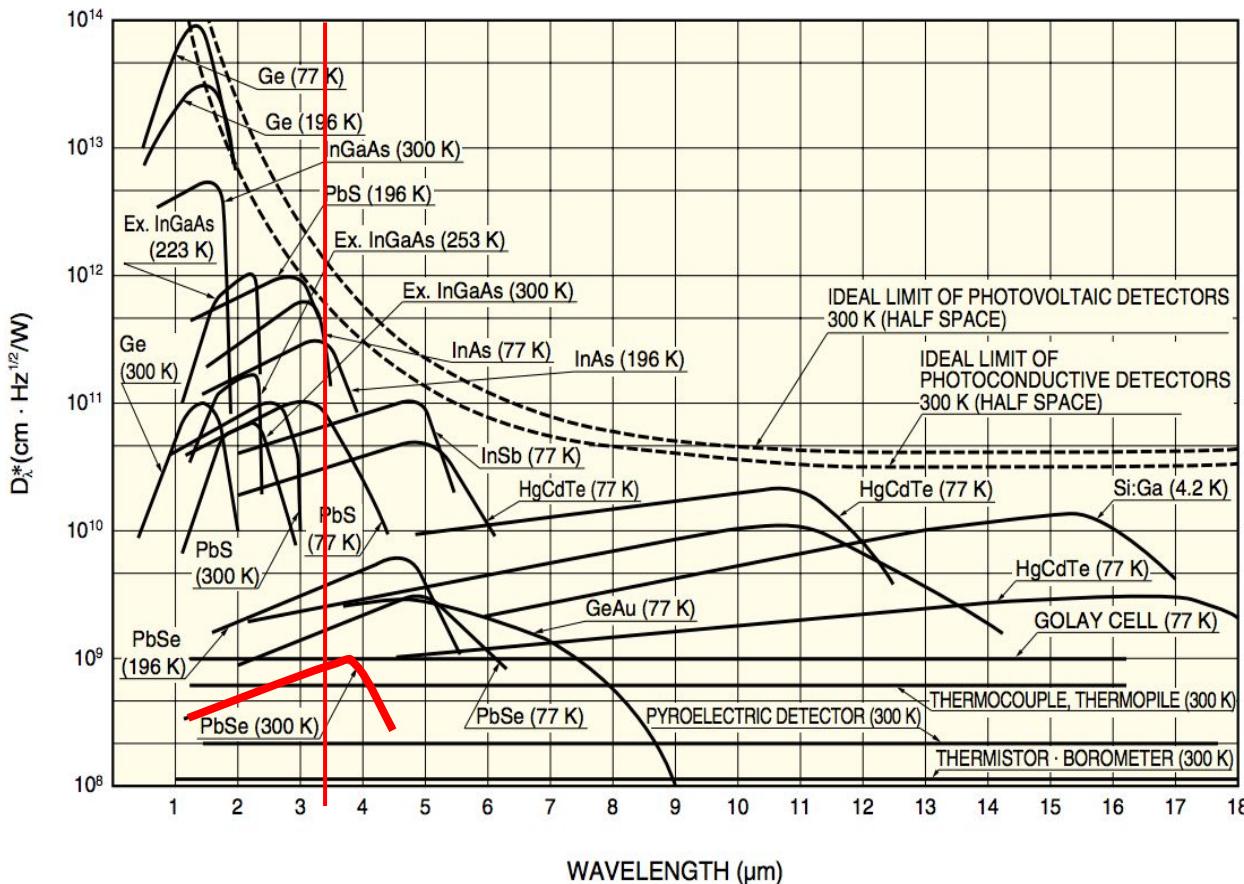
# System Diagram - Etalon Detector (ED)

---





# ED Design Criteria - Wavelength



## Final Candidate:

- PbSe uncooled
- 2mmx2mm
- Photoconductive detector from OptoDiode
- TO-5 Package,
- 3-pin pinout
- BXP-25





# ED Design Criteria - Time Constant

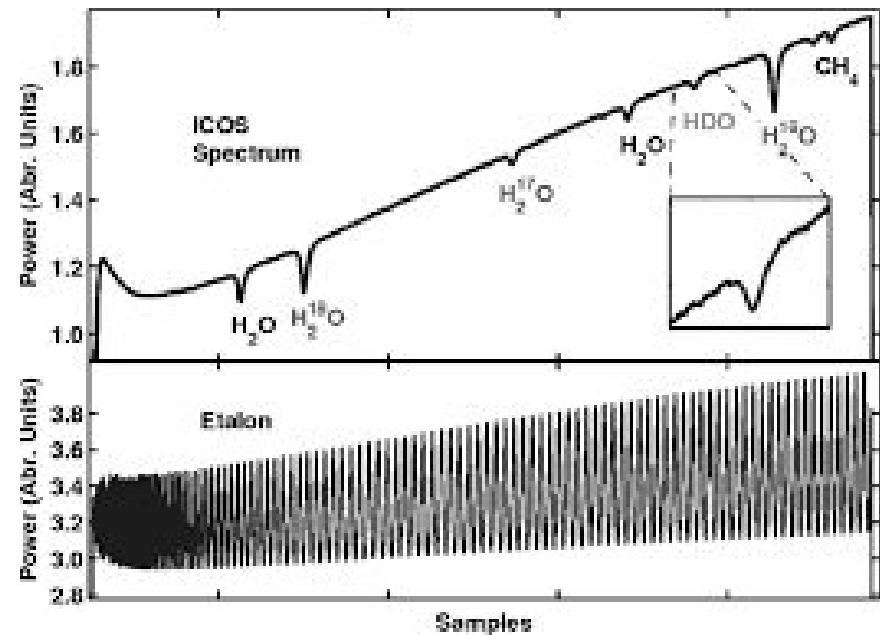
Time for 1 etalon fringe:

$$\frac{t_{full-sweep}}{\Delta v_{wavenumbers}} * FSR = \frac{t_{full-sweep}}{\Delta v_{wavenumbers}} * \frac{1}{2nL}$$

Assuming  $1\text{cm}^{-1}$  in 1ms, L = 7.62cm, n = 4,

$$t_{fringe} = 16\mu\text{s}$$

**61 fringes per scan**



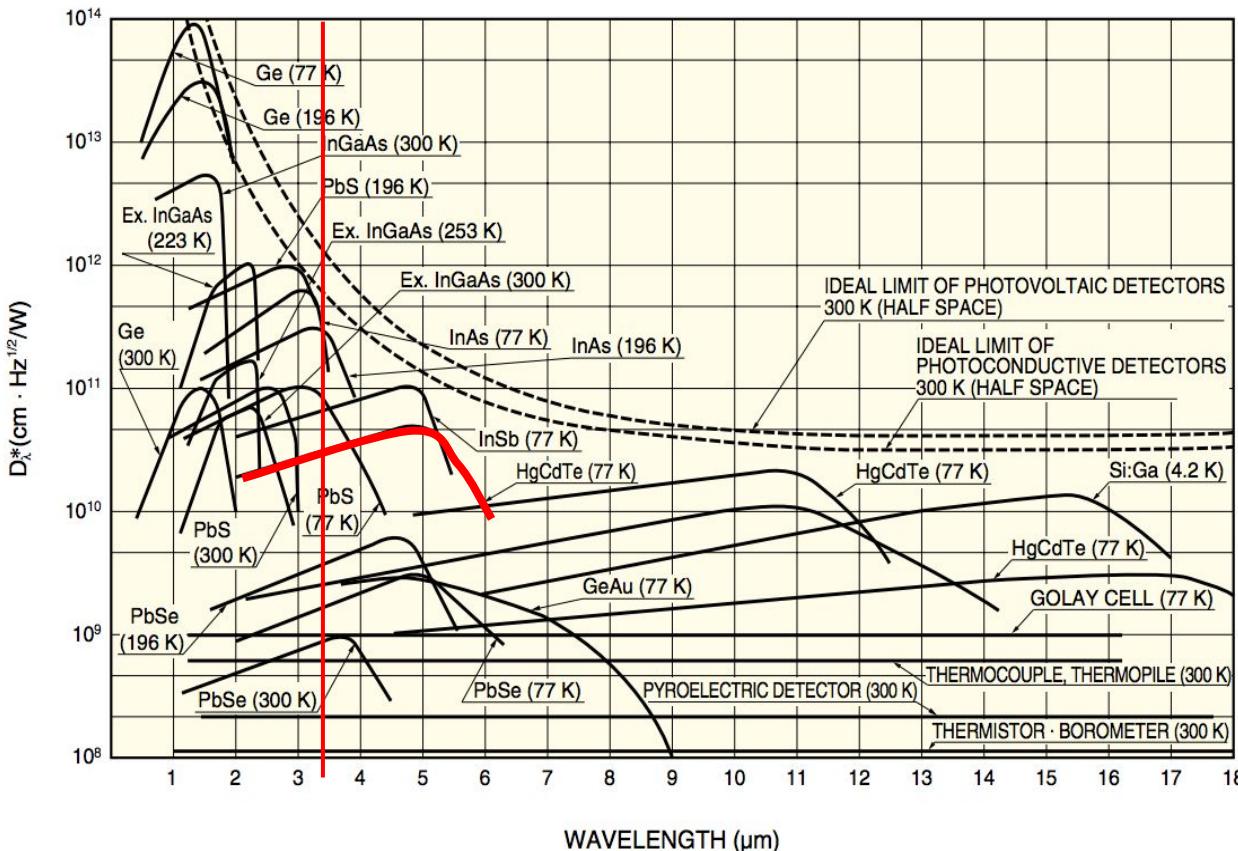


# System Diagram - Cooled Detector (CD)



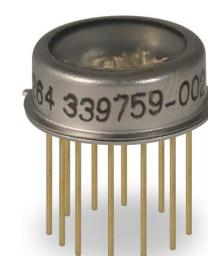


# CD Design Criteria - Wavelength



Final Candidate:

- **HgCdTe 4-stage TEC-cooled 2mmx2mm**
- **Photoconductive detector from Infrared Associates**
- **TO-66 package, 8-pin pinout**
- **MCT-4.5-TE4-2.00**





# CD Design Criteria - Time Constant

**Ringdown Time** - required time for laser beam to complete the full path length.

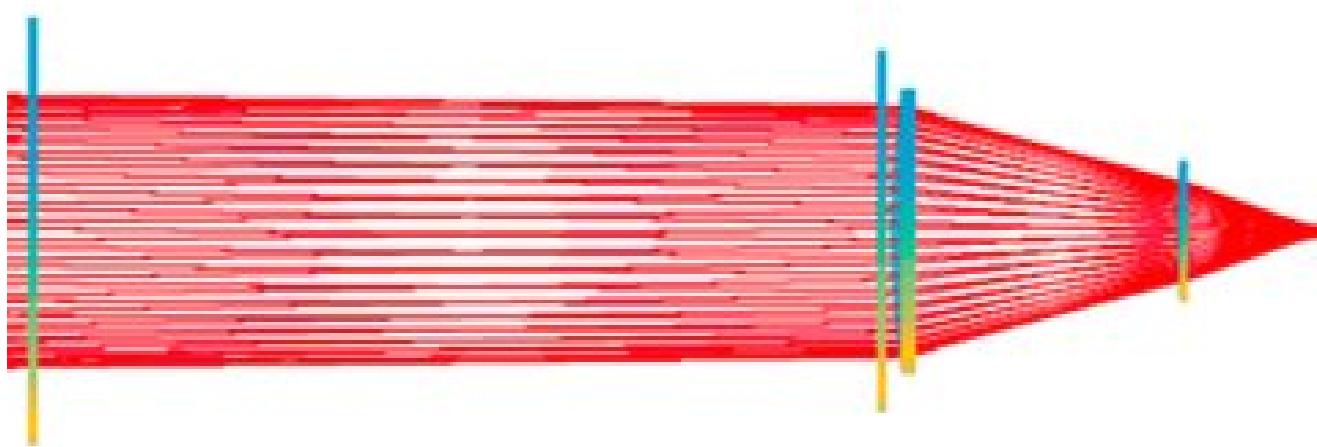
**Effective Path Length** - length of cell multiplied by number of passes of laser.

$$\ell_{\text{path}} = \ell_{\text{cell}} / L_{\text{loss}}$$

$$t_{\text{ringdown}} = \ell_{\text{path}} / c$$

$\ell_{\text{path}}$  = effective path length  
 $\ell_{\text{cell}}$  = cell length  
 $L_{\text{loss}}$  = 1 - Reflectivity

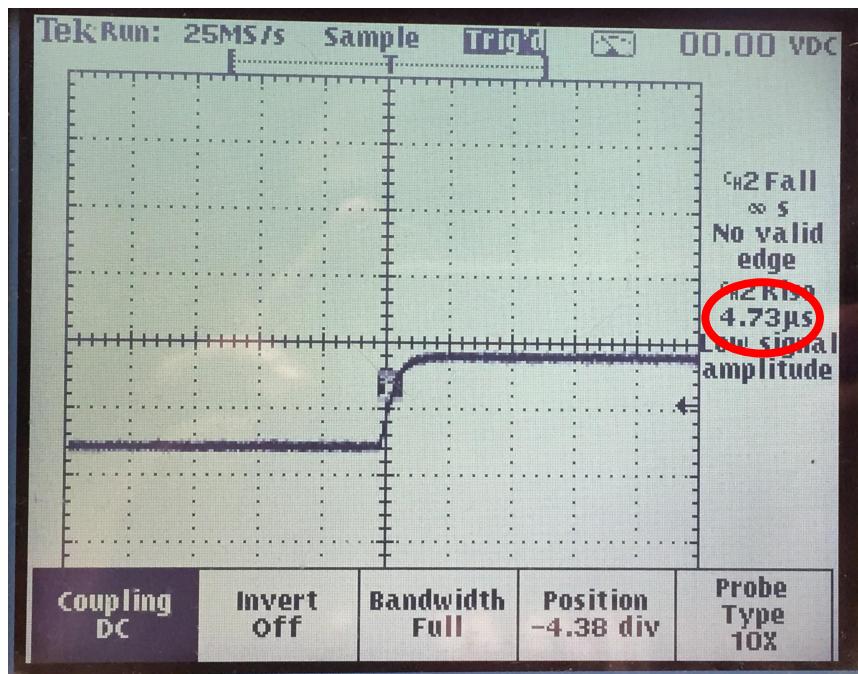
$$17\mu\text{s} < t_{\text{ringdown}} < 21\mu\text{s}$$



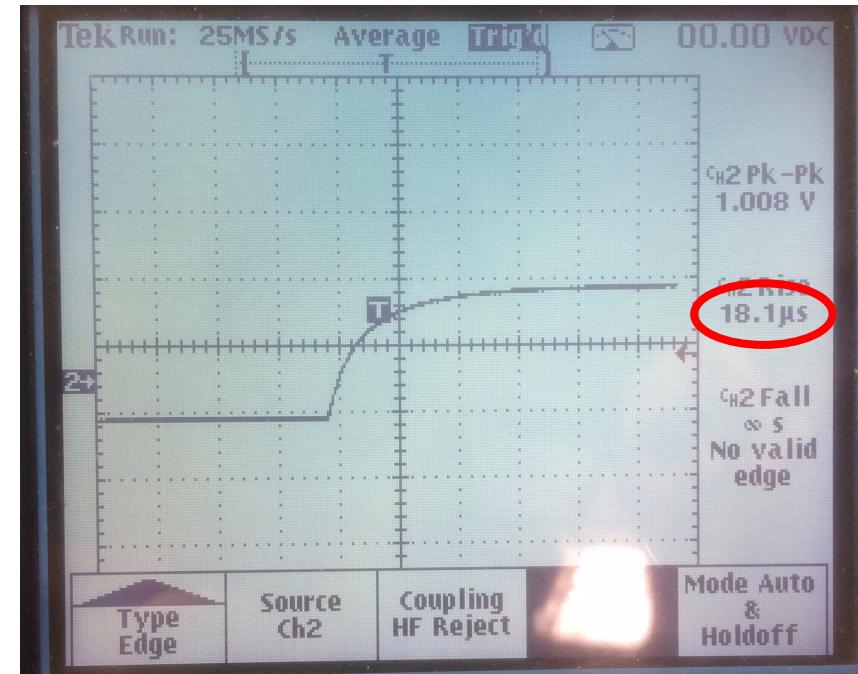


# Detector Testing

Etalon Detector



Cooled Detector





# Pre-Amp

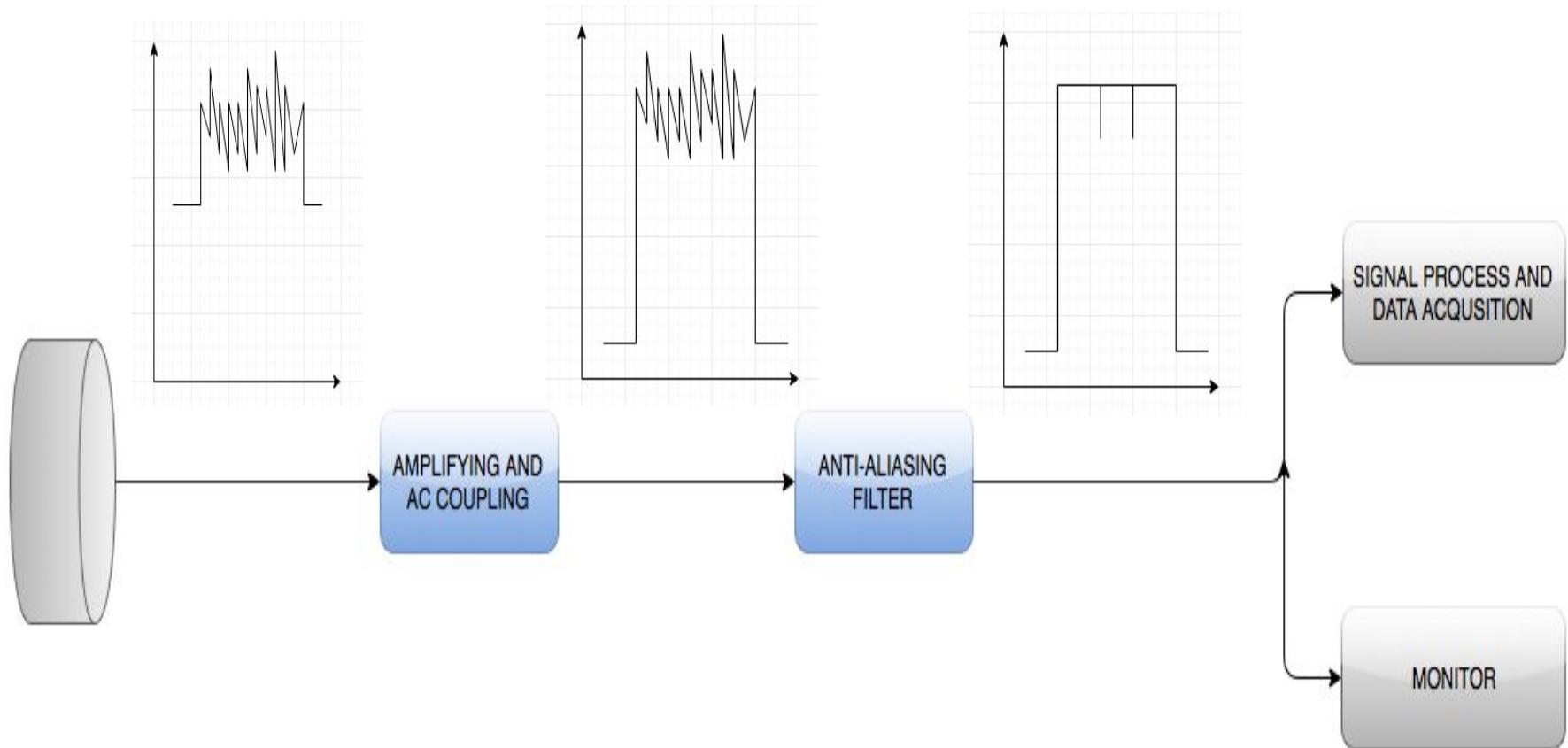
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## Why a pre-amp?

- Amplify detector signal
- Remove DC bias
- Remove noise picked up



# Pre-amp Block Diagram





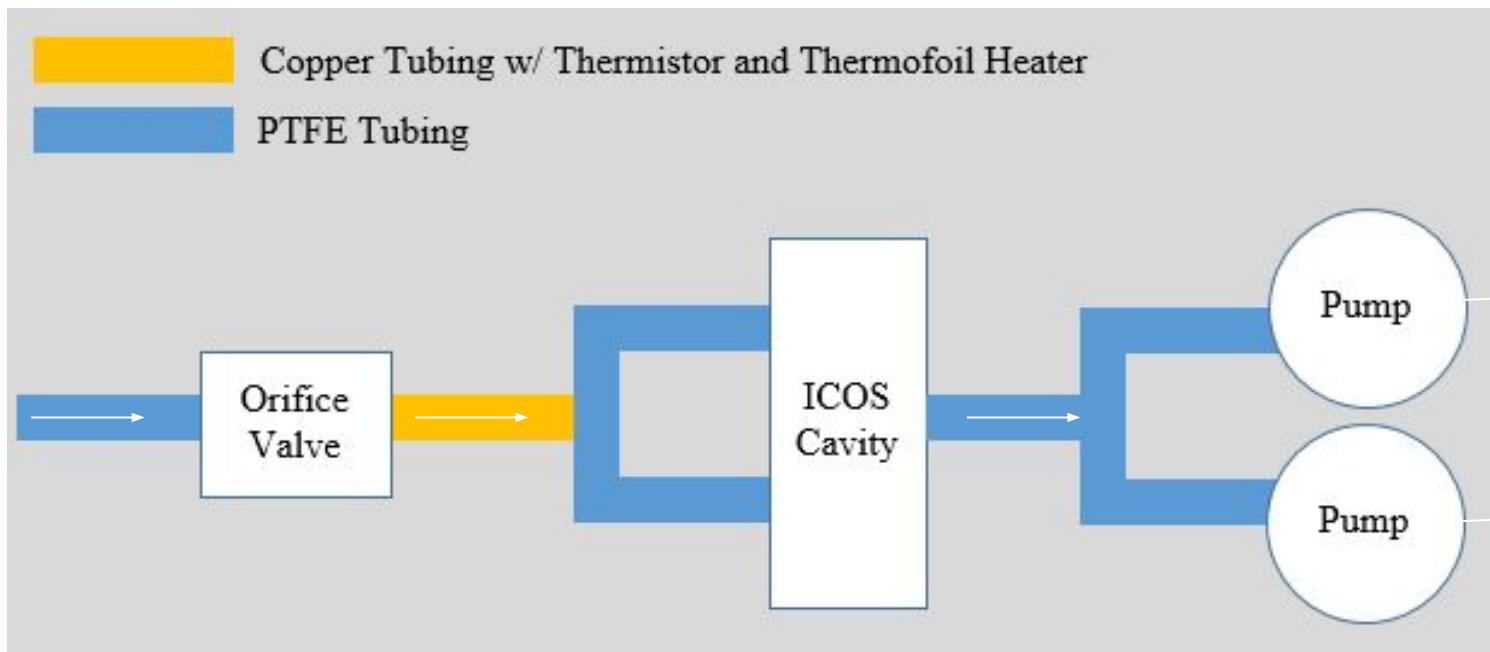
# Pump

---

1. Scientific Context
2. System Overview
3. Design Details
  - a. Laser
  - b. Cavity
  - c. Detector
  - d. Pump
  - e. System Wide Design
4. Next Steps

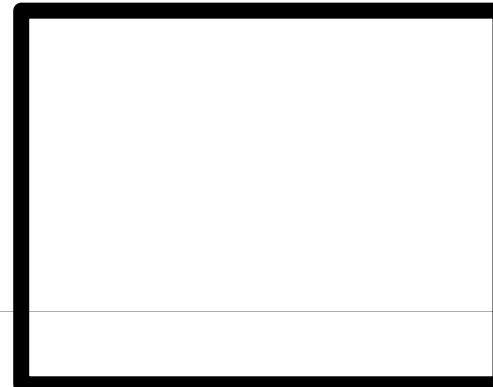


# Pump - Flow Setup





# System Diagrams - Pumps





# Pump - Requirements

<b>Minimum Flow Rate</b>	90 L/min
<b>Vacuum Pressure</b>	40 torr or 53 mbar
<b>Max Pressure Differential</b>	165 torr or 220 mbar
<b>Max Power</b>	500 W
<b>Max Weight</b>	10 kg

- **Flow Rate  $> 1.5 \text{ L/s} = 90 \text{ L/min}$** 
  - **Cavity volume of  $\sim 1\text{L}$**
  - **1Hz sampling rate (at the minimum)**
- **Brushless DC Motor**
  - **prevents arcing**
- **Continuous flow for steady data sampling**



# Pump - Selection

**SVF-x-50**

Without Controller



OEM1

With Controller

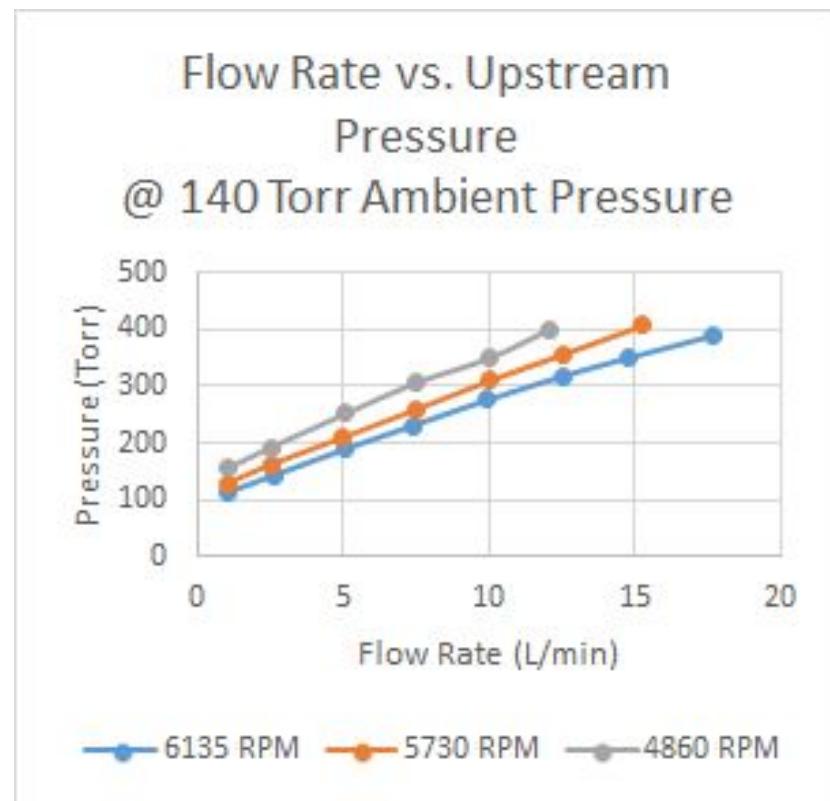
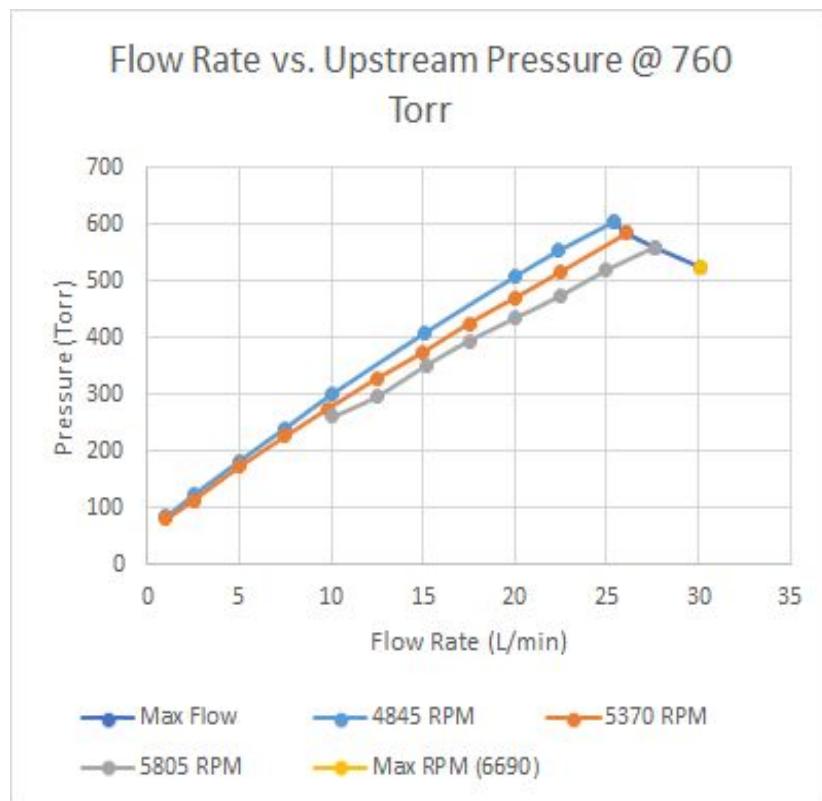


OEM2

Pump	SVF-E0-50 (Scroll Labs)	T2-01 (Parker)	KNF N838 (Old Pump)
<b>Pump Type</b>	Scroll	Diaphragm	Diaphragm
<b># of pumps</b>	2	2	1
<b>Weight (kg)</b>	1.0 (2.0 total)	1.38 (2.76 total)	3
<b>Volume of Unit</b>	0.968 (1.936 total)	1.78 (3.56 total)	1.5
<b>Max Power Usage (W)</b>	95±3 (190±6 total)	69 (138 total)	101
<b>Vacuum Pressure (mbar)</b>	≤1.0 (ultimate)	406 (differential)	90
<b>Flow Rate/ Unit (S.L/min)</b>	45	66	60
<b>Price per Unit (\$)</b>	2590.00 (5180)	809.00 (1618.00)	--



# Pump - Pump Characterization



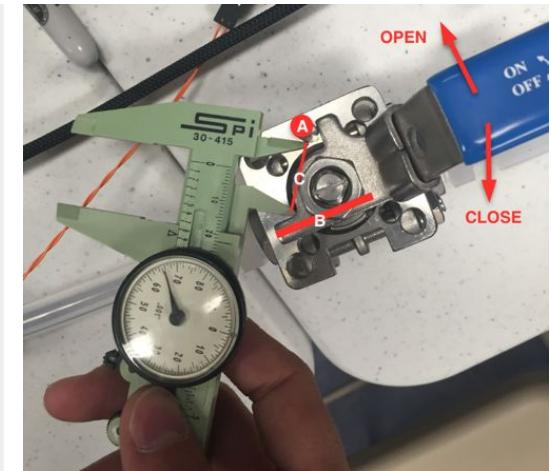
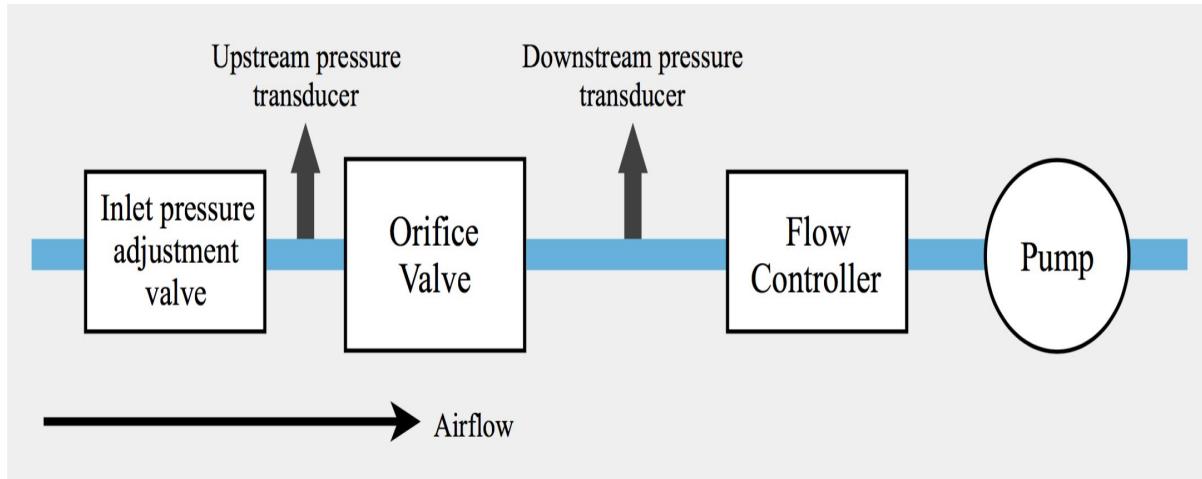


# Pressure and Temperature Control





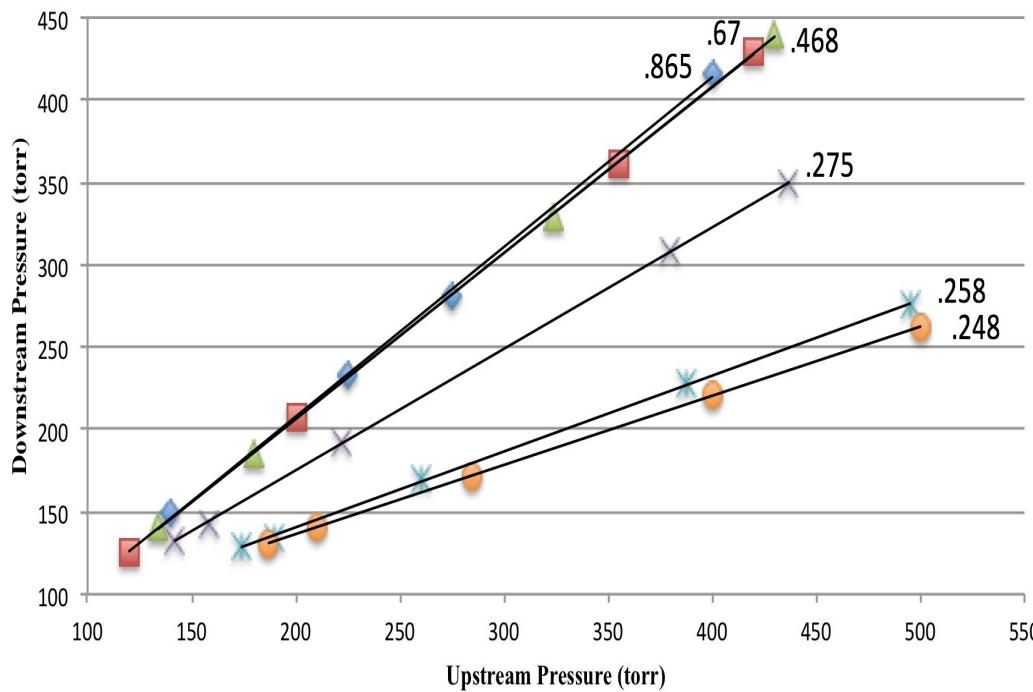
# Pump - Orifice Valve Testing





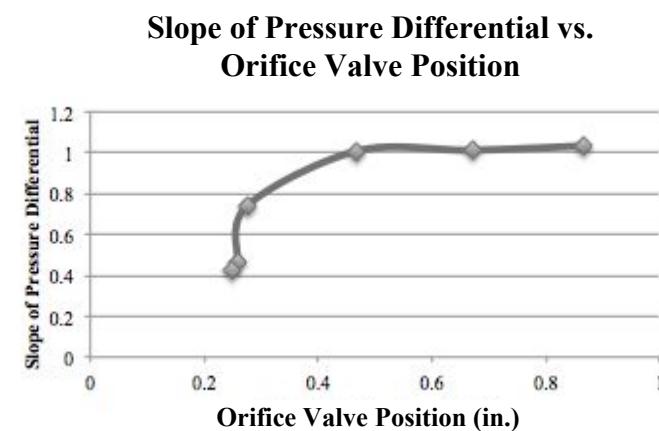
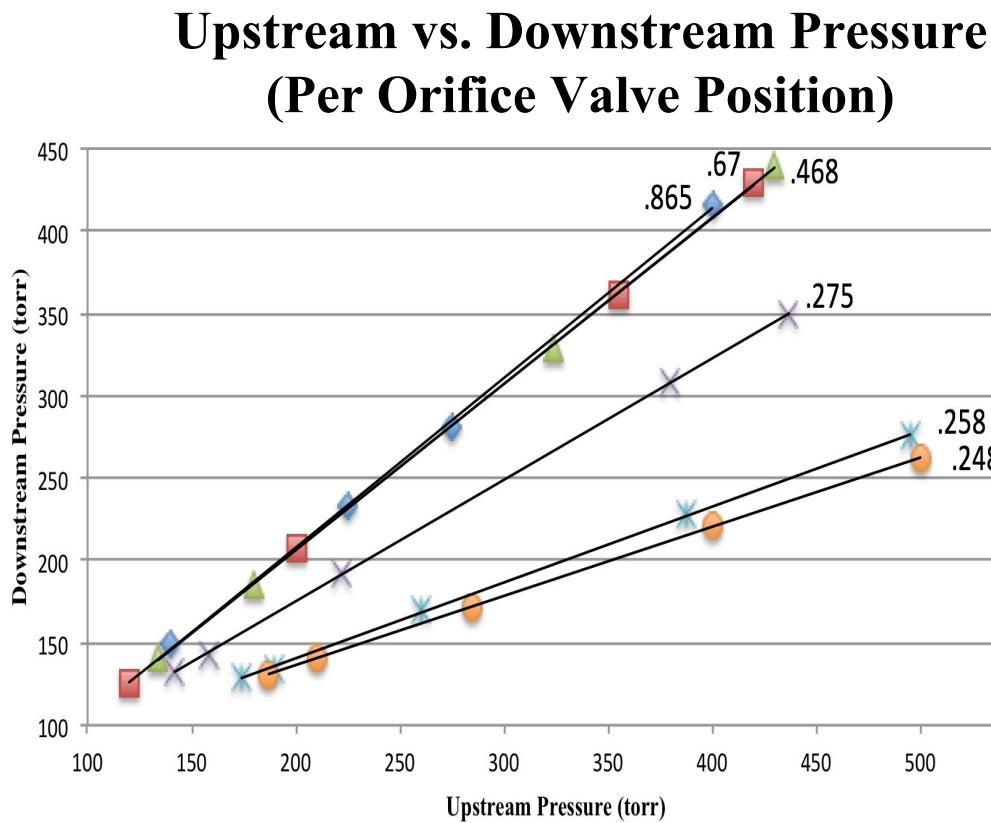
# Pump - Orifice Valve Testing

Upstream vs. Downstream Pressure  
(Per Orifice Valve Position)



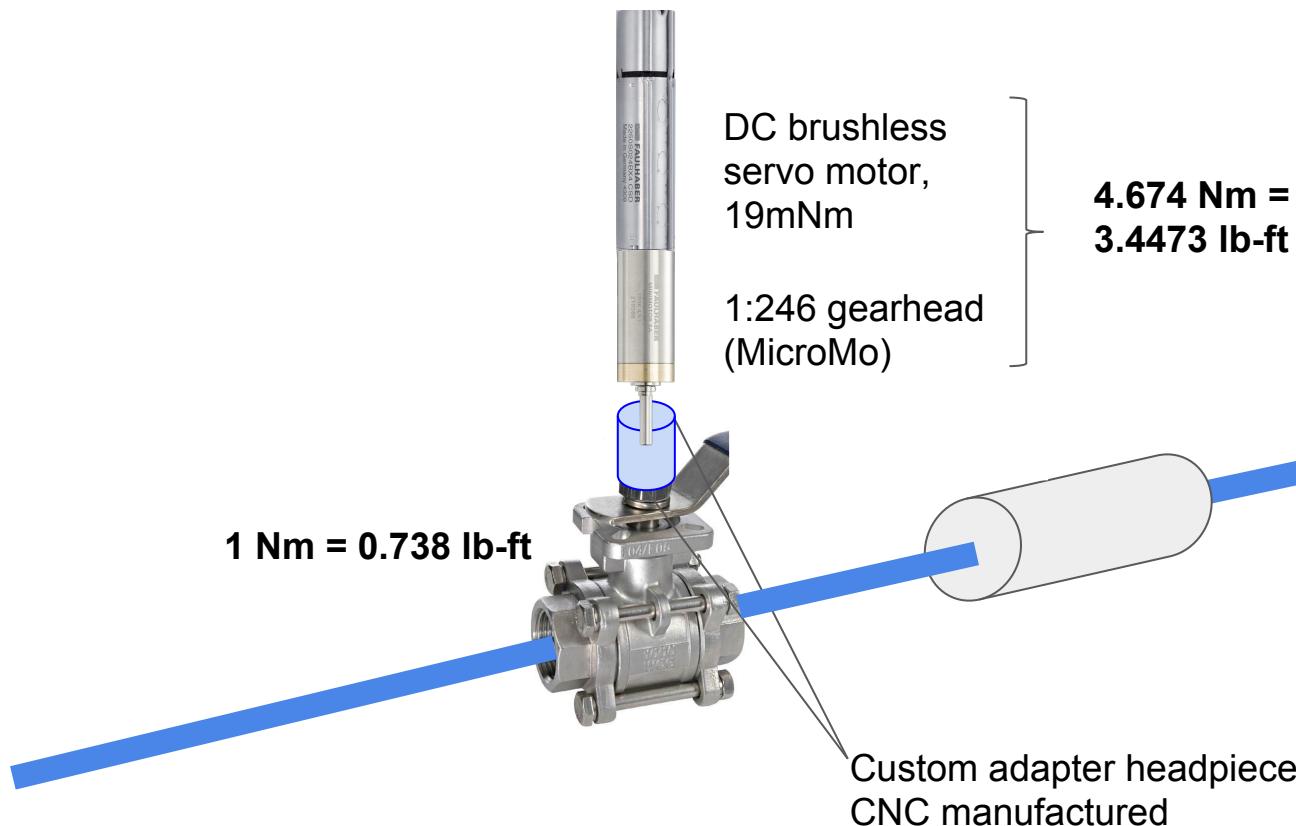


# Pump - Orifice Valve Testing



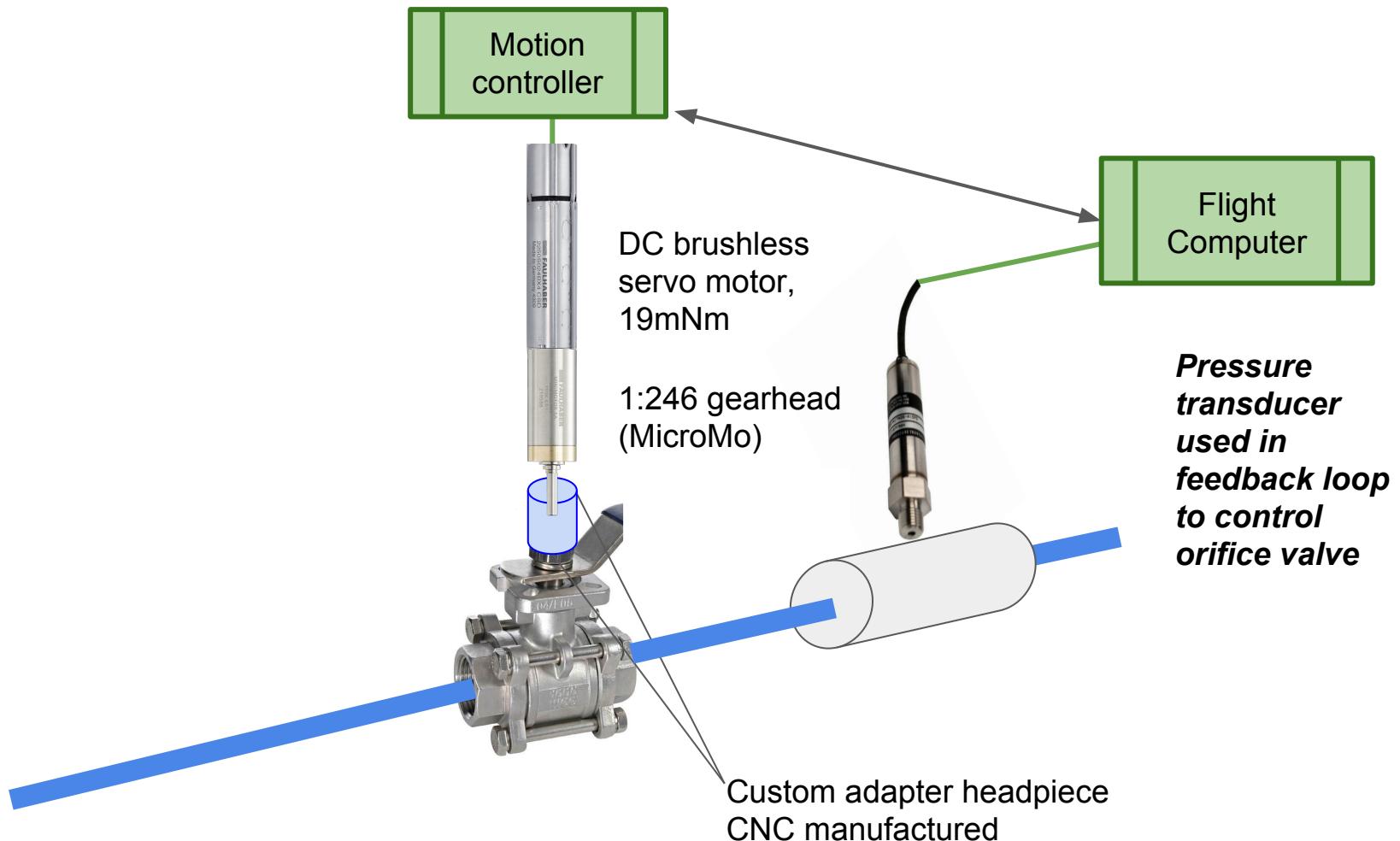


# Pump - Orifice Valve Feedback





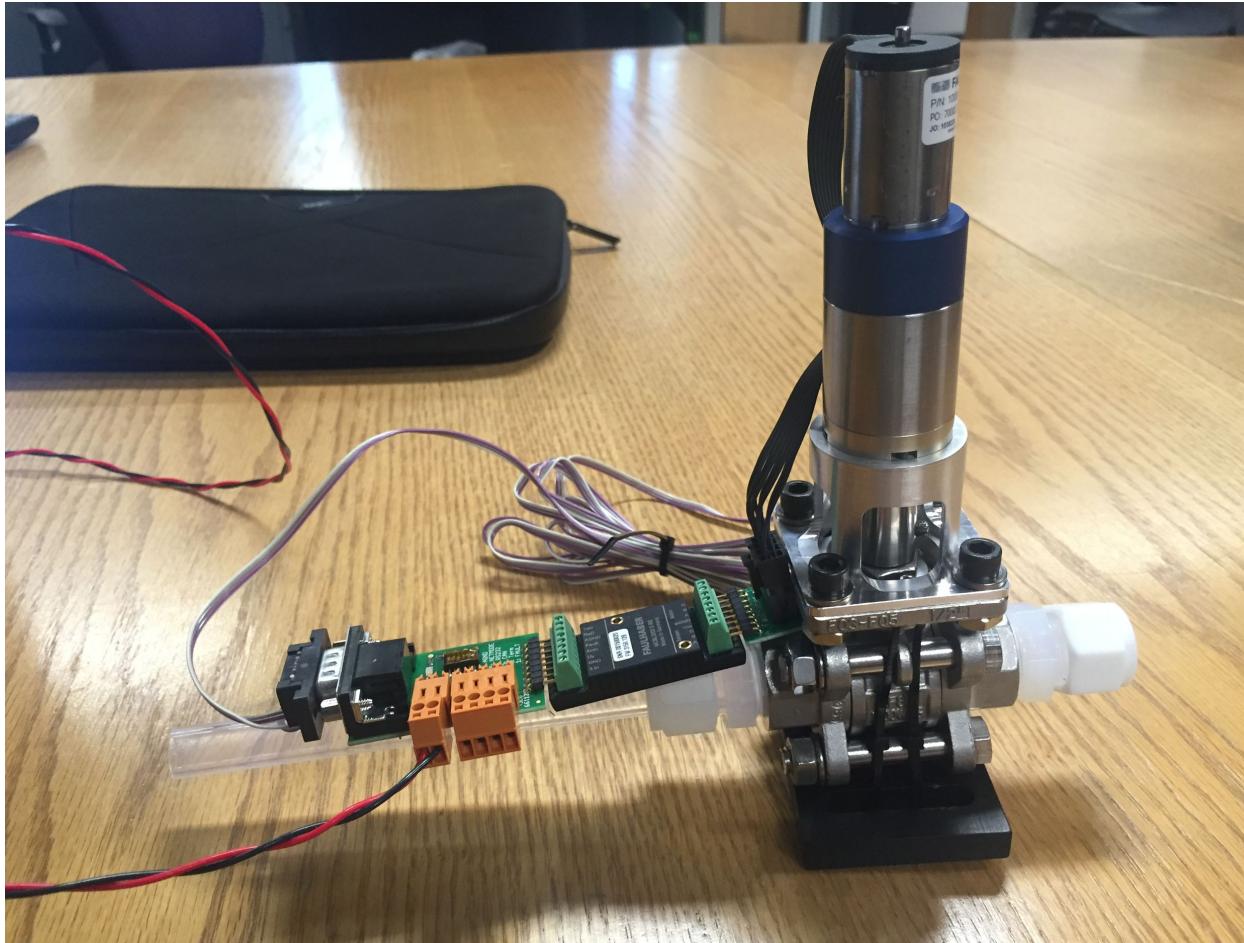
# Pump - Orifice Valve Feedback





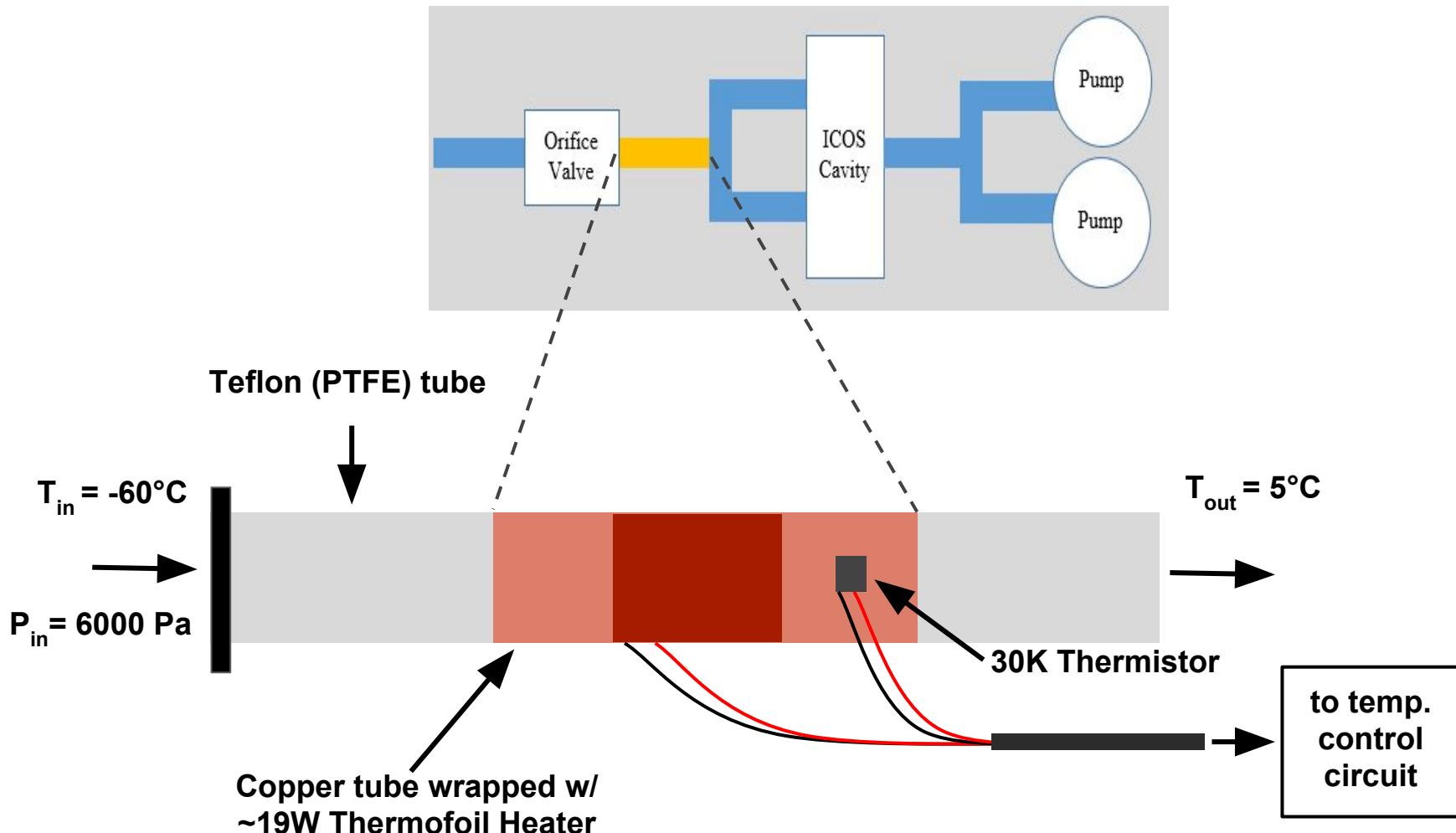
# Pump - Orifice Valve Feedback

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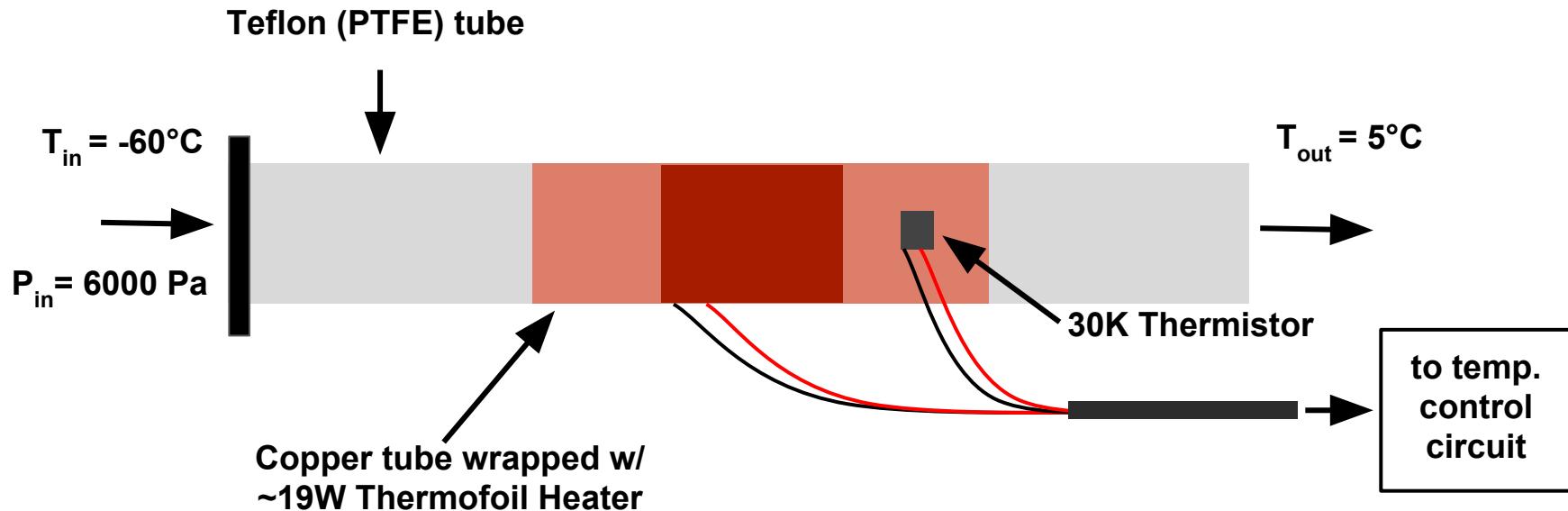


# Pump - Air Temperature Control





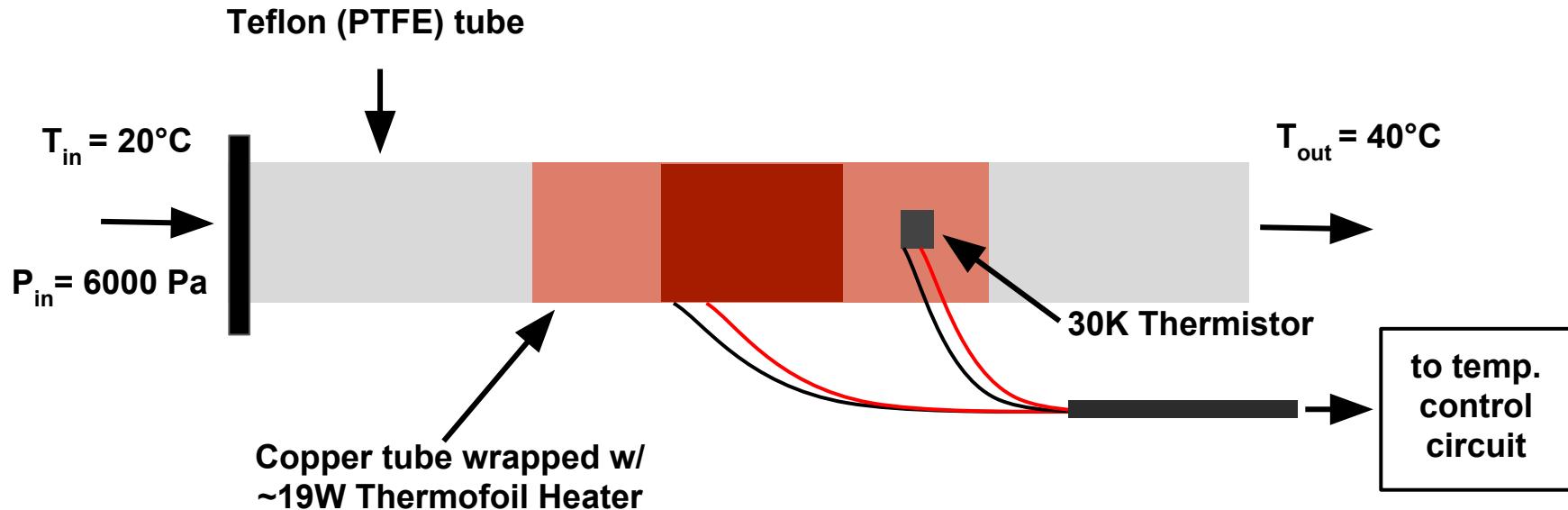
# Pump - Air Temperature Control



2.0x0.5 in  
@ 25 W/in<sup>2</sup>



# Pump - Air Temperature Control



Testing indicates:

- The heater is powerful enough for our needs
- Cu tube  $\Delta T = 101^{\circ}\text{C}$
- Downstream air  $\Delta T$  is not pressure-dependent

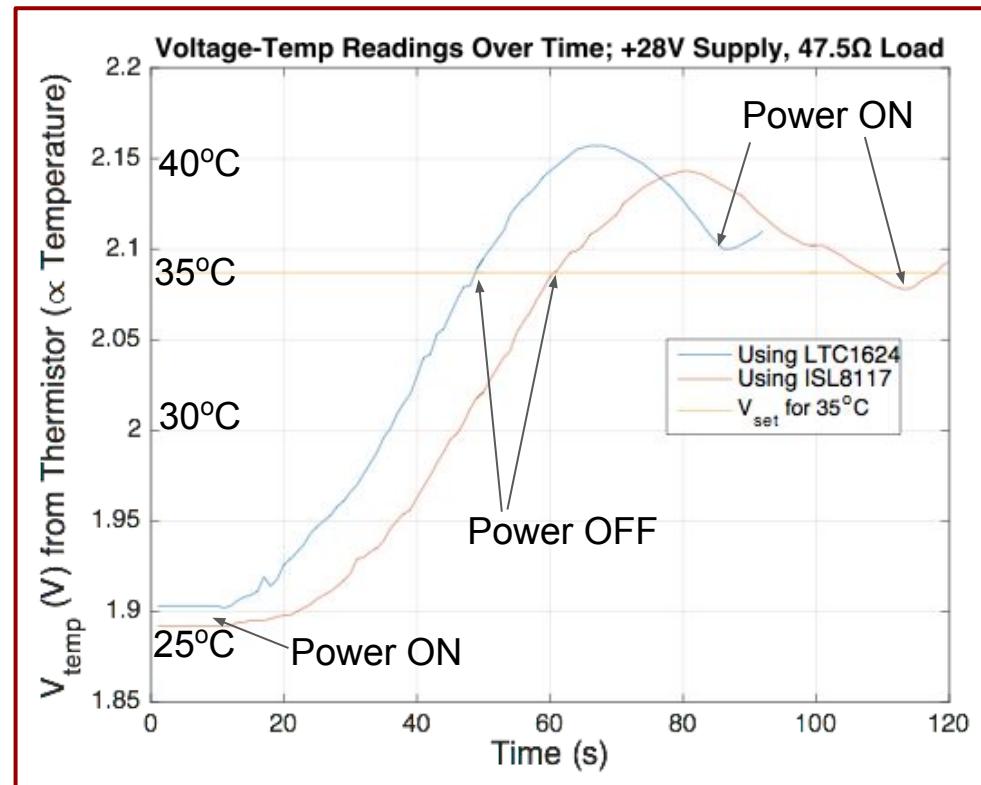


2.0x0.5 in  
@ 25 W/in<sup>2</sup>



# Pump - Inlet Temperature Control

- Tested switching voltage regulators on the PI board using  $47.5\Omega$  power resistor load with thermistor attached
- ISL8117 showed improved performance over LTC1624 (older version)
  - Low power output when reading near set point, improved timing for ON
- Next steps: redesign board for more seamless integration of new chip, tune PID for full system





# System Wide Design

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1. Scientific Context
2. System Overview
3. Design Details
  - a. Laser
  - b. Cavity
  - c. Detector
  - d. Pump
  - e. **System Wide Design**
4. Next Steps

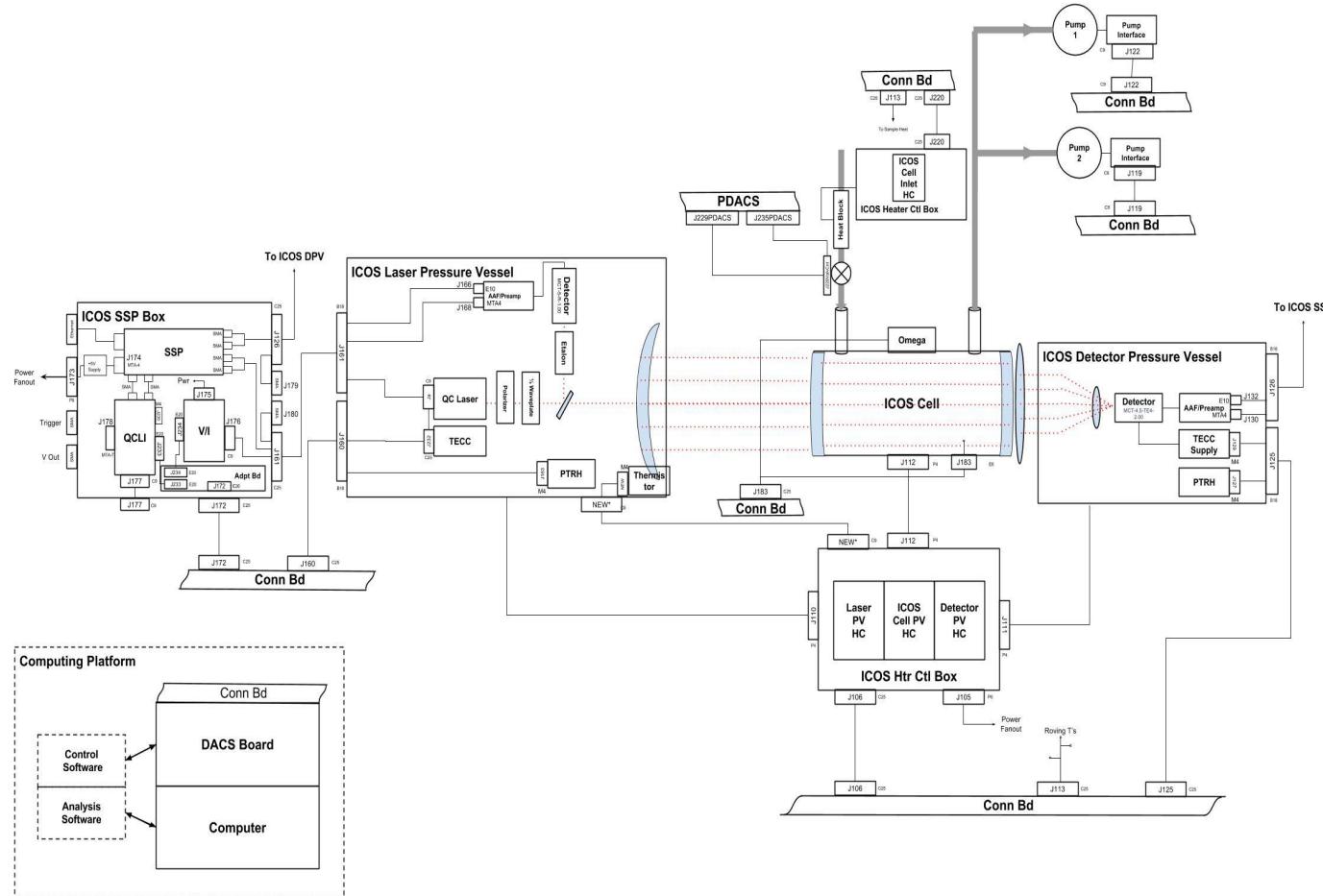


# System Wide - Budgets

System	Power Est (W)	Mass Est (kg)	Volume Est (L)
Pump	215	3.10	2.26
Cavity	39	1.20	2.50
Laser (Driver)	1	0.23	0.02
Laser (Fore-optics)	0	2.27	8.85
Detector	1	2.60	4.56
Additional Housings	0	10.00	2.00
Total	256 (41 not incl pump)	19.40	20.19
Design Goals	<500 (60 not incl. pump)	<25 kg	<25 L

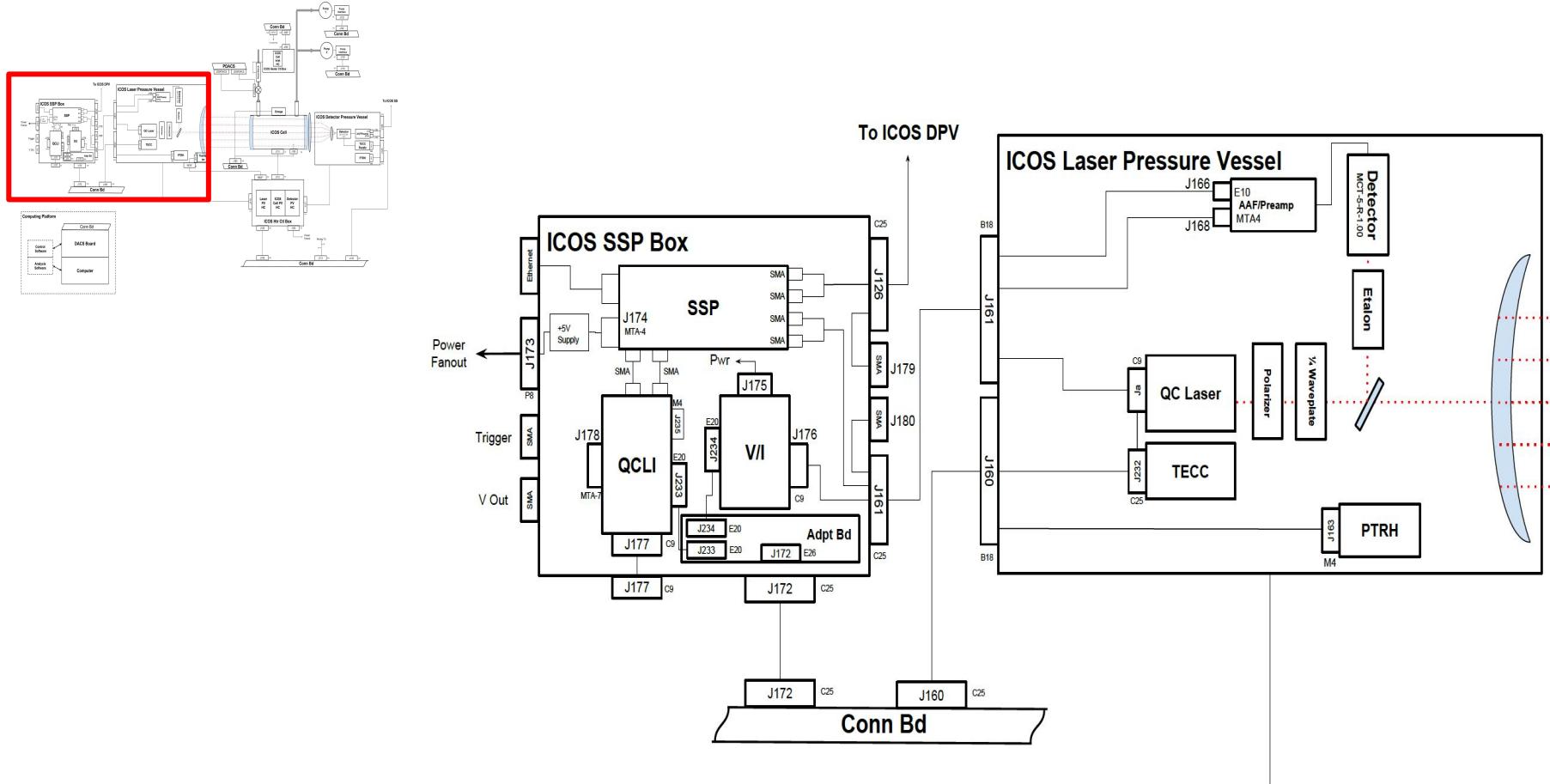


# System Wide Interconnect Diagram



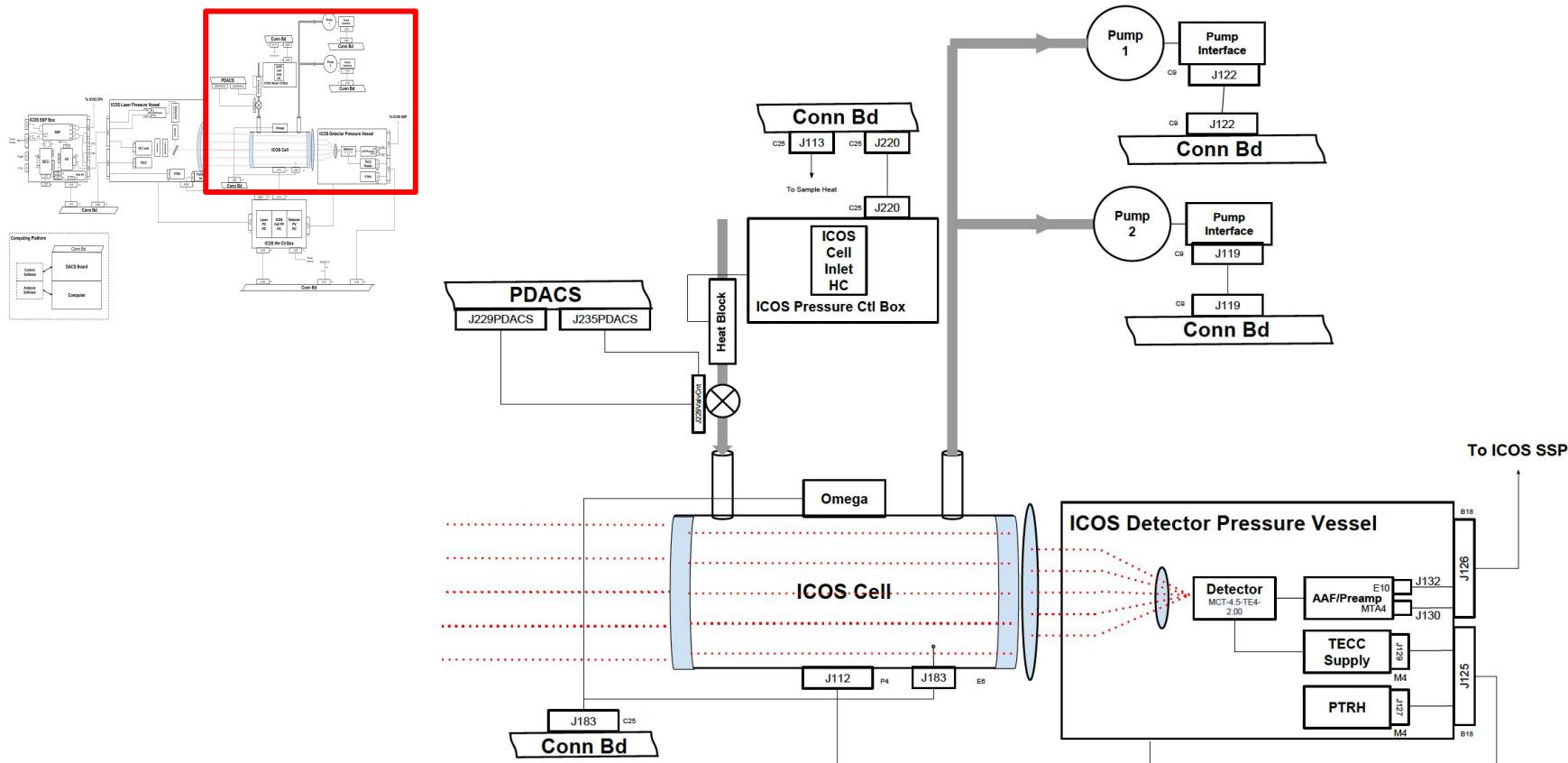


# System Wide Interconnect Diagram





# System Wide Interconnect Diagram





# System Wide - Flight Computer

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**Hardware**  **Software**

- Wires
- Connectors
- Connector Board
- DACS Board – Data Acquisition and Control System Board
- Computer
- Data Collector
- Telemetry Buffer
- Data Clients
  - TMCalgo – Data Analyser
  - Computer Displays
- Commands



# Next Steps

---

1. Scientific Context
2. System Overview
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  - a. Laser
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4. Next Steps



# Improvements

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- **Pump:** lighter, smaller pump. Higher flow rate than older ICOS systems
- **Fore optics:** Now use independent mounting for all components, telescope to decrease mirror spotting, and a fully incorporated, minimized cage system for modularity and ease of assembly
- **Laser Driver:** simplified to one board, faster response time to voltage changes
- **Cavity:** smaller, reduced weight (hardware), updated locations of temperature probes
- **Simulation:** New MATLAB cavity script, refined ray tracing model
- **Sensing:** Pressure feedback loop for flow control
- **Detector:** Cheaper, faster etalon detector



# Next Steps

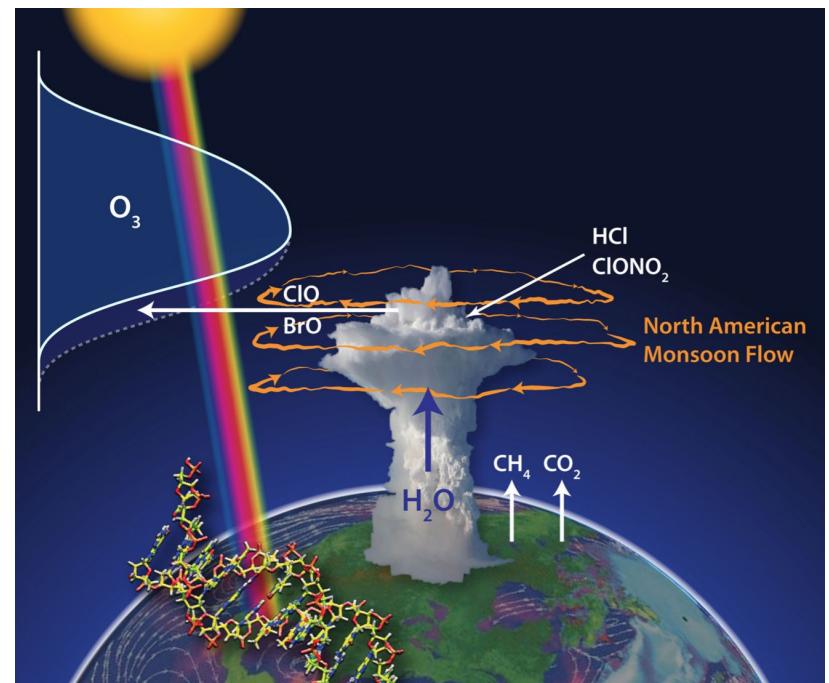
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- **NASA Undergraduate Student Instrument Project (USIP)**
- **Summer 2016: Independent research (5 students)**
- **ES100 - Theses (8 students)**
- **ES96 - Spring 2017**
- **Spring/Summer 2017: NASA Balloon Flight for HCl and Ozone**





# HCl Detection Instrument





# Acknowledgements

---

## I. ES96 Teaching Staff

- A. Professor Jim Anderson, Karena McKinney, Marco Rivero, Mike Greenberg, Norton Allen, Terry Martin

## II. SEAS Teaching Labs Staff

- A. Maddie Hickman, Joe Huggard, Andreas Haggerty, Elaine Kristant

## III. Machine Shop

- A. Mike McKenna