A study on the correction of azimuthal angle by fusion of geomagnetic sensor and inertial sensor

Jongtaek Oh
Department of Electronics & Information
Hansung University
Seoul, South Korea
jtoh@hansung.ac.kr

Abstract— For indoor navigation systems that use low-cost sensors such as those built-in to smartphones, azimuth estimation is critical to location accuracy. However, the accuracy of the measured azimuth in many locations is very bad due to the influence of the magnetic field on the geomagnetic sensor. In this paper, the geomagnetic sensor data are fused with the acceleration sensor and gyro sensor data, to correct the erroneous azimuthal angle and it shows improved performance compared to the conventional method.

Keywords— azimuth, geomagnetic sensor, acceleration sensor, gyro sensor, fitting

I. INTRODUCTION

The navigation device tracking the trajectory of the moving object is based on the moving direction and distance. As a method of estimating the moving direction, there is a method of estimating a rotation angle using a gyro sensor, and an azimuthal angle estimation using a geomagnetic sensor. For the gyro sensor, the rotation angle estimation is accurate in a short time, but there is a problem that the error accumulates as the time passes. The acceleration sensor has no cumulative error, but it cannot measure the yaw (φ) angle in the two-dimensional plane. Therefore, we can estimate the rotation angle yaw by fusion of these two sensors [1]. In this case, however, there is a problem of not knowing the azimuthal angle at the initial stage of starting, and there is a problem that the rotation angle error becomes cumulative as the moving object moves.

On the other hand, in the geomagnetic sensor, the measurement values of the geomagnetic field intensity in the x, y, and z directions of the navigation frame are measured as m_x , m_y , and m_z , respectively. The azimuthal angle α is computed using m_x and m_y , because the azimuthal angle on the two-dimensional plane is only required for the user walking with the smartphone. However, since there are scattered iron devices around the user, these devices affect the geomagnetic sensor, resulting in distortions in measured m_x and m_y [2]. As a result, the azimuthal angle using the geomagnetic sensor becomes incorrect in many locations. A variety of studies have been conducted to correct this

[3] used many true azimuthal angle data and measured data in order to learn the neural network, and an inertial sensor and a geomagnetic sensor were used to correct the sensor frame error in [4]. Also it was proposed a maximum likelihood estimator for iteratively correcting geomagnetic distortion [5], and a method of using the geomagnetic sensor parameters after first calibrating them in a place without distortion is also proposed in [6]. However, it is difficult to apply to general equipment such as smartphones and accuracy is not appropriate.

In this paper, I apply the acceleration sensor and the gyro sensor to the Extended Kalman Filter (EKF) to estimate the rotation angle of the smartphone, and measure the geomagnetic sensor value at the same time. By fusion of these two data, the geomagnetic sensor values m_x and m_y with mitigated distortion are estimated. Experiments have been carried out by the proposed method and the improved performance is confirmed as compared with the conventional method.

II. MASUREMENTS OF GEOMAGNETIC SENSOR

In order to fuse rotation angle measured by inertial sensors and azimuthal angle estimated by geomagnetic sensor, the angle of rotation must be precisely estimated using the EKF. The gyro sensor data was applied to the state transition equations and the acceleration sensor data was used as the measurement value.

A. Rotation Angle Measurements

The equation for approximately calculating the roll angle ϕ and pitch angle θ of the smartphone using the acceleration sensor is (1), where g is the gravity, and f_x and f_y are the acceleration values of x and y axis in the navigation frame, respectively [1].

$$\theta = \sin^{-1}\left(\frac{-f_x}{g}\right), \quad \phi = \sin^{-1}\left(\frac{-f_y}{g \cdot \cos \theta}\right) \tag{1}$$

The following EKF model equations using gyro sensor values as state variables $\overline{x} = [\phi \ \theta \ \varphi]^T$ are as follows,

$$\dot{\overline{x}} = f(\omega_x, \omega_y, \omega_z, \overline{x}) + w = [f_1 f_2 f_3]^T + w$$
 (2)

where ω_i is the angular velocity measured from gyro sensor, w is system noise, φ is the rotation angle, and

$$\begin{split} f_{_{1}} &= \omega_{_{x}} + \omega_{_{y}} \sin \phi \tan \theta + \omega_{_{z}} \cos \phi \tan \theta \,, \\ f_{_{2}} &= \omega_{_{y}} \cos \phi - \omega_{_{z}} \sin \phi \,, \\ f_{_{3}} &= \omega_{_{y}} \sin \phi / \cos \theta + \omega_{_{z}} \cos \phi / \cos \theta \,. \end{split}$$

The discrete time system model equation is as follows and **A** is Jacobian of f_i . The measurement equation is (4) and v is the measurement noise, where \overline{z} is the measured ϕ and θ .

$$\overline{X}_{k+1} = \mathbf{A} \, \overline{X}_k + W_k \tag{3}$$

$$\overline{z} = \begin{bmatrix} 100 \\ 010 \end{bmatrix} \begin{bmatrix} \phi \\ \theta \\ \varphi \end{bmatrix} + v \tag{4}$$

B. Conventional Geomagnetic Sensor Correction Method

Geomagnetic distortion is classified into hard iron and soft iron. In the absence of distortion, the geomagnetic field strengths m_x and m_y could be plotted on a two-dimensional plane as a circle centered at the origin. However, in the case of distortion, the central axis moves and becomes a rotated ellipse. In [7], in order to correct the sensor data error due to geomagnetic distortion, the ellipse was rotated in the opposite direction and the shape was adjusted to form a circle. The following figure depicts m_x and m_y with and without distortion.

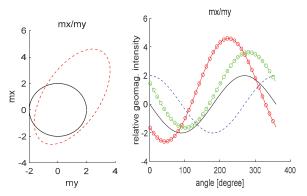


Fig. 1. In the left, the solid line is for the case without distortion, and the dotted line is with distortion. In the right, the solid lines are for m_{χ} and the dotted lines are for m_{χ} without distortion, and the marked lines are with distortion, respectively.

III. NOVEL DISTORTION CORRECTION

There would be three methods to correct the geomagnetic sensor value in the graph domain. The first method of the conventional method, is to transform the center-point moving and distorted circle-shaped geomagnetic sensor data into circle-like data with the center point at the origin. In this paper, I propose a novel method of fitting the rotation angle and the azimuthal angle to the linear model and a method of least square fitting the m_x and m_y intensity to the sinusoidal function.

A. Linear Fitting of Azimuthal Angle

Accelerometer sensor and gyro sensor can be used to measure the angle of rotation comparing geomagnetic sensor data. Therefore, when the azimuthal angle according to geomagnetic sensor data is plotted according to the rotation angle of equidistance, it should become a straight line when there is no distortion. Using this characteristic, the measured azimuthal angle data was fitted to the linear model by Least Mean Square (LMS) method, as below

$$\alpha(\varphi) = a \cdot \varphi + b \tag{5}$$

where α : azimuthal angle, φ : rotation angle, a: slope, and b: intercept. (5) is expressed in the form of matrices for the measured azimuthal data as follows for each rotation angle,

$$\begin{bmatrix} 1 \\ 1 \\ 1 \\ 1 \end{bmatrix} b = \begin{bmatrix} \alpha(0) - a \cdot 0 \\ \alpha(1) - a \cdot 1 \\ \vdots \\ \alpha(359) - a \cdot 359 \end{bmatrix}.$$
 (6)

(6) can be denoted as $\mathbf{A}_{L}x = \overline{b}$ and, initial azimuthal angle b can be obtained as $x = \left[\mathbf{A}_{L}^{\mathsf{T}}\mathbf{A}_{L}^{\mathsf{T}}\overline{b}\right]^{-1}\mathbf{A}_{L}^{\mathsf{T}}\overline{b}$ by LMS. This method differs from the conventional method in statistically correcting the entire geomagnetic sensor data.

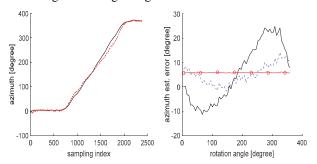


Fig. 2. In the left, the solid line is the rotation angle obtained by the inertial sensors on the time axis, and the dotted line is the azimuthal angle calculated by geomagenetic sensor data. The right figure shows the estimated error corresponing to the rotation angle. Solid line is uncorrected, dotted line is by conventional method, and marked line is result by proposed linear fitting method.

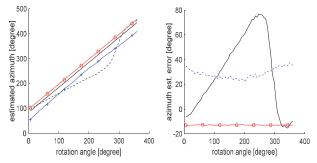


Fig. 3. In the left, the solid line is the rotation angle to which the initial azimuth angle is added, the dotted line is not corrected, the '+' marked line is the conventional method, and the 'o' marked line is the linear fitting result. The right figure shows the estimated error corresponing to the rotation angle.

These are for the place where the geomagnetic distortion is more severe than the place of Fig. 2.

As shown in Fig. 2 and 3, linear fitting using LMS shows much more accurate results than the conventional method.

B. Sinusoidal Fitting of m_x and m_y

The azimuth angle is calculated with the geomagnetic field intensity m_x and m_y which could be distorted. Thus, it could be fitted to the original form of m_x and m_y without distortion to improve accuracy. The geomagnetic field intensity measured by the geomagnetic sensor could be modeled as follows,

$$m_{x} = x_{offset} - k_{x} \cdot \sin(\varphi + \alpha_{1})$$

$$m_{y} = y_{offset} + k_{y} \cdot \cos(\varphi + \alpha_{2})$$
(7)

where $x_{\rm offset}$ and $y_{\rm offset}$ are the distances from the center of the figure to the origin in the m_x - m_y plane, respectively, k_x and k_y are amplitude coefficients, and α_1 and α_2 are phase shifts in the geomagnetic intensity-angle plane. When the geomagnetic sensor data is circle, α_1 and α_2 should be the same, and it stands for the azimuthal angle for which the smartphone is aiming. (7) is non-linear equation of α , so the parameters are estimated using least square function of lsqcurvefit() in MATLAB.

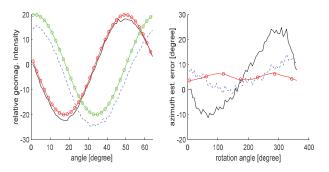


Fig. 4. In the left, the solid line and the dotted line are measured m_x and m_y , respectively, and the marked lines are fitted results to the sinusoidal functions by the least square method. The right figure shows the estimated azimuthal angle corresponing to the rotation angle.

Fig. 4 is the result of fitting in the same place as Fig. 2. In smaller geomagnetic distortion locations, the sinusoidal fitting works better, but in larger locations the linear fitting performs better.

IV. CONCLUSION

The conventional method is to correct the distorted circle on the m_x - m_y plane with a circle geometrically, but in the proposed method, the geomagnetic sensor data sampled according to the rotation angle is corrected by linear or

sinusoidal fitting method. Both methods showed much better performance than the conventional method and confirmed that the performance varies depending on the degree of geomagnetic distortion.

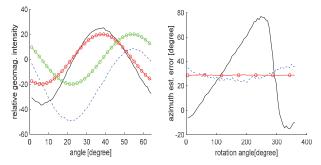


Fig. 5. Fitting results in a place with greater distortion than in Fig. 4.

TABLE I. GEOMAGNETIC SENSOR CORRECTION PERFORMANCE

place	RMS estimation error of azimuthal angle [degree]		
method	Place A	Place B	Place C
No correction	13.3	18.5	42.5
Conventional method	6.5	13.4	29.0
Linear fitting	5.8	12.2	12.9
Sinusoidal fitting	5.1	12.5	28.8

ACKNOWLEDGMENT

This work was supported by the 2018 Research Fund of the NRF (No. 2017R 1D 1A 1B03031244).

REFERENCES

- Phil Kim, Kalman Filter for Beginners with MATLAB Examples. Seoul, A-JIN Publishing, 2011.
- [2] W.H.K. Vries, H.E.J. Veeger, C.T.M. Baten, and F.C.T. Helm, "Magnetic distortion in motion labs, implications for validating inertial magnetic sensors," Gait & Posture, no. 29, 2009, pp. 535-541.
- [3] J.H. Wang and Y. Gao, "A new magnetic compass calibration algorithm using neural networks," Measurement Scienc and Technology, no. 17, 2006, pp. 153-160.
- [4] S. Bonnet, C. Bassompierre, C. Godin, S. Lesecq, and A. Barraud, "Calibration methods for inertial and magnetic sensors," Sensors and Actuators, no. 156, 2009, pp. 302-311.
- [5] J.F. Vasconcelos, G. Elkaim, C. Silvestre, P. Oliveira, and B. Cardeira, "Geometric approach to strapdown magnetometer calibration in sensor frame," IEEE Trans. Aero. Elec. Sys., vol. 47, no. 2, 2011, pp. 1293-1306.
- [6] J. Li, Q. Zhang, D. Chen, M. Pan, and F. Luo, "Magnetic interferential field compensation in geomagnetic measurement," Trans. Inst. Meas. Cont., vol. 36, no. 2, 2014, pp. 244-251.
- [7] C.C. Foster and G.H. Elkaim, "Extension of a two-step calibration methodology to include nonorthogonal sensor axes," IEEE Trans. Aero. Elec. Sys., vol. 44, no. 3, 2008, pp. 1070-1078.