

Automotive Simulation Models & Control Problem Formulation

Faculty of «Electronic Engineering for Intelligent Vehicles» and «Advanced Automotive Engineering»

Nicola Mimmo

Department of Electrical, Electronics and Information Engineering «G. Marconi» - University of Bologna

Contents

- Active Suspensions
- Adaptive Cruise Control
- Wheel Speed Controls
- Electronic Stability Program / Torque Vectoring
- Self-Park Assist



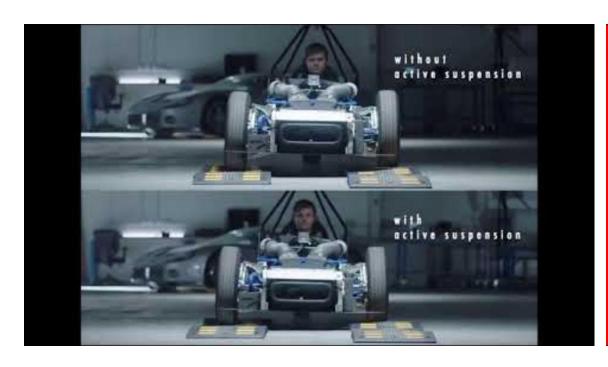
Contents

- Active Suspensions
- Adaptive Cruise Control
- Wheel Speed Controls
- Electronic Stability Program / Torque Vectoring
- Self-Park Assist



Motivations & Goals

Improve the ride quality and change the setup.







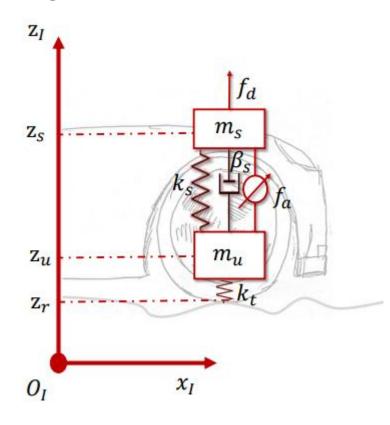
Motivations & Goals

Improve the ride quality and change the setup.





Single-corner Model



Sprung mass
$$m_s \ddot{z}_s = -m_s g + f_s + f_d$$

Unsprung mass
$$m_u \ddot{z}_u = -m_u g - f_s + f_t (z_u - z_r)$$

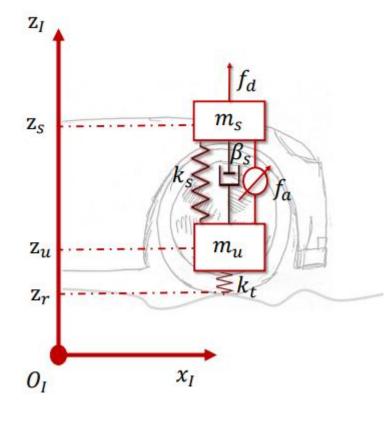
Tire
$$f_t(z_u - z_r) = \begin{cases} 0 & z_u - z_r > \ell_t \\ -k_t(z_u - z_r - \ell_t) & z_u - z_r \le \ell_t \end{cases}$$

Suspension
$$f_s := -k_s(z_s-z_u-\ell_s)-eta_s(\dot{z}_s-\dot{z}_u)+f_a$$

Aerodynamics (downforce)
$$f_d = \frac{1}{2} \rho S v^2 C_z$$



Single-corner Model



Sensors

Accelerometer (z)

MEMS Based Accelorometer

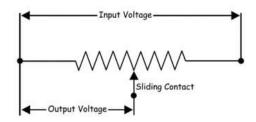
Accelerometer Sensor MEM Mechanism

ElectronicWines

$$y_a = -g - \frac{k_s}{m_s}(z_s - z_u - \ell_s) - \frac{\beta_s}{m_s}(\dot{z}_s - \dot{z}_u) + \frac{f_a + f_d}{m_s} + \nu_a$$

Potentiometer

$$y_p = z_s - z_u + \nu_p$$





Regulated Output

$$e = y_p - r$$



let
$$\dot{z}_s := v_s$$
 and $\dot{z}_u := v_u$

$$\dot{z}_s = v_s$$

$$m_s \dot{v}_s = -m_s g - k_s (z_s - z_u - \ell_s) - \beta_s (v_s - v_u) + f_a + f_d$$

$$\dot{z}_u = v_u$$

$$m_u \dot{v}_u = -m_u g + k_s (z_s - z_u - \ell_s) + \beta_s (v_s - v_u) - f_a + f_t (z_u - z_r)$$



Define

$$\begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} := \begin{bmatrix} z_s - z_u \\ v_s - v_u \\ z_u - z_r \\ v_u - \dot{z}_r \end{bmatrix}$$

Define
$$\begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} := \begin{bmatrix} z_s - z_u \\ v_s - v_u \\ z_u - z_r \\ v_u - \dot{z}_r \end{bmatrix} \quad \begin{array}{l} \dot{z}_s = v_s \\ m_s \dot{v}_s = -m_s g - k_s (z_s - z_u - \ell_s) - \beta_s (v_s - v_u) + f_a + f_d \\ \dot{z}_u = v_u \\ m_u \dot{v}_u = -m_u g + k_s (z_s - z_u - \ell_s) + \beta_s (v_s - v_u) - f_a + f_t (z_u - z_r) \end{array}$$



$$\begin{split} \dot{x}_1 &= x_2 \\ \dot{x}_2 &= -\frac{m_s + m_u}{m_s m_u} k_s (x_1 - \ell_s) - \frac{m_s + m_u}{m_s m_u} \beta_s x_2 + \frac{m_s + m_u}{m_s m_u} f_a + \frac{1}{m_s} f_d - \frac{1}{m_u} f_t (x_3) \\ \dot{x}_3 &= x_4 \\ \dot{x}_4 &= -g + \frac{k_s}{m_u} (x_1 - \ell_s) + \frac{\beta_s}{m_u} x_2 - \frac{f_a}{m_u} + \frac{1}{m_u} f_t (x_3) - \ddot{z}_r \end{split}$$

Define

$$\begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} := \begin{bmatrix} z_s - z_u \\ v_s - v_u \\ z_u - z_r \\ v_u - \dot{z}_r \end{bmatrix}$$

Define
$$\begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} := \begin{bmatrix} z_s - z_u \\ v_s - v_u \\ z_u - z_r \\ v_u - \dot{z}_r \end{bmatrix} = \begin{bmatrix} \dot{z}_s = v_s \\ m_s \dot{v}_s = -m_s g - k_s (z_s - z_u - \ell_s) - \beta_s (v_s - v_u) + f_a + f_d \\ \dot{z}_u = v_u \\ m_u \dot{v}_u = -m_u g + k_s (z_s - z_u - \ell_s) + \beta_s (v_s - v_u) - f_a + f_t (z_u - z_r) \end{bmatrix}$$



$$\begin{split} \dot{x}_1 &= x_2 \\ \dot{x}_2 &= -\frac{m_s + m_u}{m_s m_u} k_s (x_1 - \ell_s) - \frac{m_s + m_u}{m_s m_u} \beta_s x_2 + \frac{m_s + m_u}{m_s m_u} f_a + \frac{1}{m_s} f_d - \frac{1}{m_u} f_t (x_3) \\ \dot{x}_3 &= x_4 \\ \dot{x}_4 &= -g + \frac{k_s}{m_u} (x_1 - \ell_s) + \frac{\beta_s}{m_u} x_2 - \frac{f_a}{m_u} + \frac{1}{m_u} f_t (x_3) - \ddot{z}_r \end{split}$$

$$y_p = x_1 + \nu_p$$

$$y_a = -g - \frac{k_s}{m_s}(x_1 - \ell_s) - \frac{\beta_s}{m_s}x_2 + \frac{f_a + f_d}{m_s} + \nu_a$$

$$e = x_1 + \nu_p - r.$$



$$\dot{x}_1 = x_2$$

$$\dot{x}_2 = -\frac{m_s + m_u}{m_s m_u} k_s (x_1 - \ell_s) - \frac{m_s + m_u}{m_s m_u} \beta_s x_2 + \frac{m_s + m_u}{m_s m_u} f_a + \frac{1}{m_s} f_d - \frac{1}{m_u} f_t (x_3)$$

$$\dot{x}_3 = x_4$$

$$\dot{x}_4 = -g + \frac{k_s}{m_u} (x_1 - \ell_s) + \frac{\beta_s}{m_u} x_2 - \frac{f_a}{m_u} + \frac{1}{m_u} f_t (x_3) - \ddot{z}_r$$

$$y_{p} = x_{1} + \nu_{p}$$

$$y_{a} = -g - \frac{k_{s}}{m_{s}}(x_{1} - \ell_{s}) - \frac{\beta_{s}}{m_{s}}x_{2} + \frac{f_{a} + f_{d}}{m_{s}} + \nu_{a}$$

$$\mathbf{x} = \operatorname{col}(x_{1}, x_{2}, x_{3}, x_{4}), \quad u = f_{a},$$

$$\mathbf{d} = \operatorname{col}(\ddot{z}_{r}, f_{d}),$$

$$\mathbf{v} = \operatorname{col}(\nu_{p}, \nu_{a}),$$

$$\mathbf{w} = \operatorname{col}(\mathbf{d}, \boldsymbol{\nu}, r),$$

$$e = x_1 + \nu_p - r.$$
 $\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, u, \mathbf{w})$ $\mathbf{x}(t_0) = \mathbf{x}_0$ $\mathbf{y} = \mathbf{h}(\mathbf{x}, u, \mathbf{w})$ $e = h_e(\mathbf{x}, u, \mathbf{w}),$



$$\dot{x}_1 = x_2$$

$$\dot{x}_2 = -\frac{m_s + m_u}{m_s m_u} k_s (x_1 - \ell_s) - \frac{m_s + m_u}{m_s m_u} \beta_s x_2 + \frac{m_s + m_u}{m_s m_u} f_a + \frac{1}{m_s} f_d - \frac{1}{m_u} f_t (x_3)$$

$$\dot{x}_3 = x_4$$

$$\dot{x}_4 = -g + \frac{k_s}{m_u} (x_1 - \ell_s) + \frac{\beta_s}{m_u} x_2 - \frac{f_a}{m_u} + \frac{1}{m_u} f_t (x_3) - \ddot{z}_r$$

$$y_p = x_1 + \nu_p$$

$$y_a = -g - \frac{k_s}{m_s}(x_1 - \ell_s) - \frac{\beta_s}{m_s}x_2 + \frac{f_a + f_d}{m_s} + \nu_a$$

$$\mathbf{x} = \operatorname{col}(x_1, x_2, x_3, x_4), \quad u = f_a,$$

$$\mathbf{d} = \operatorname{col}(\ddot{z}_r, f_d),$$

$$\mathbf{v} = \operatorname{col}(\nu_p, \nu_a),$$

$$\mathbf{w} = \operatorname{col}(\mathbf{d}, \boldsymbol{\nu}, r),$$

$$\mathbf{d} = \operatorname{col}(\ddot{z}_r, f_d),$$

$$\mathbf{v} = \operatorname{col}(\nu_p, \nu_a),$$

$$\mathbf{v} = \operatorname{col}(\mathbf{d}, \boldsymbol{\nu}, r),$$

$$\mathbf{d} = \operatorname{col}(\ddot{z}_r, f_d),$$

$$\mathbf{v} = \operatorname{col}(u_p, v_a),$$

$$\mathbf{v} = \operatorname{col}(u_p$$

$$e = x_1 + \nu_p - r.$$

$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, u, \mathbf{w}) \quad \mathbf{x}(t_0) = \mathbf{x}_0$$

$$\mathbf{y} = \mathbf{h}(\mathbf{x}, u, \mathbf{w})$$

$$e = h_e(\mathbf{x}, u, \mathbf{w}),$$

$\mathbf{x} = \text{col}(x_1, x_2, x_3, x_4), \quad u = f_a,$

- bounded
- 2) Asymptotically steer the regulated output to zero assuming constant disturbances.

Contents

- Active Suspensions
- Adaptive Cruise Control
- Wheel Speed Controls
- Electronic Stability Program / Torque Vectoring
- Self-Park Assist



Adaptive Cruise Control

Motivations & Goals

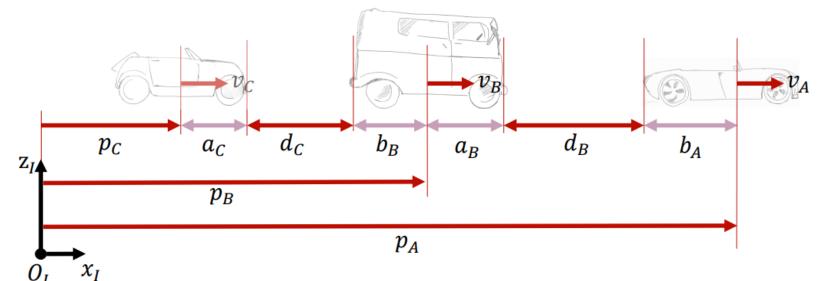
Improve safety and comfort, reduce driver's workload.





Adaptive Cruise Control

1D Model



Platoon dynamics

$$\begin{split} \dot{p}_A &= v_A \\ \dot{p}_B &= v_B \\ \dot{v}_B &= g \sin \theta + \frac{1}{m_B} \left(f_B - D_B (v_B - w) \right) \\ \dot{p}_C &= v_C \\ \dot{v}_C &= g \sin \theta + \frac{1}{m_C} \left(f_C - D_C (v_C - w) \right) \end{split}$$

Intervehicle distance

$$d_B := p_A - p_B - b_A - a_B$$

$$d_C := p_B - p_C - b_B - a_C$$

Aerodynamic Drag

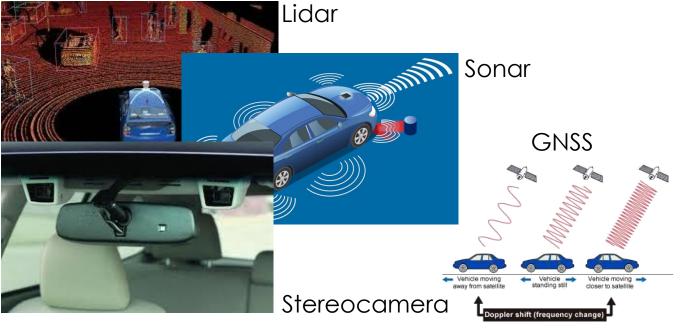
$$D_B(v_B - w) = \frac{1}{2}\rho S_B C_{D_B}(v_B - w)^2$$
$$D_C(v_C - w) = \frac{1}{2}\rho S_C C_{D_C}(v_C - w)^2$$



Adaptive Cruise Control

1D Model

Sensors



Regulated output

$$e_{dB} = y_{dB} - d_B^{\star}$$

$$e_{dC} = y_{dC} - d_C^{\star}$$

Safety distance

$$d_B^{\star} = d_{\min} + \frac{1}{k}v_B^2 \qquad k > 0$$

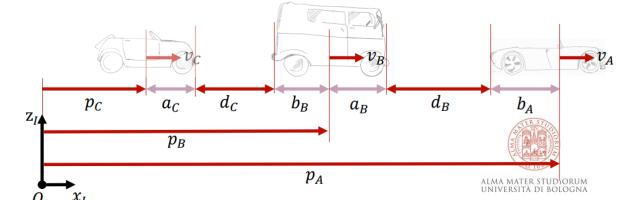
$$d_C^{\star} = d_{\min} + \frac{1}{k}v_C^2,$$

$$y_{dB} = d_B + \nu_{dB}$$
 range sensor vehicle B

$$y_{vB} = v_B + \nu_{vB}$$
 GNSS vehicle B

$$y_{dC} = d_C + \nu_{dC}$$
 range sensor vehicle C

$$y_{vC} = v_C + \nu_{vC}$$
 GNSS vehicle C,



Adaptive Cruise Control Problem Formulation

let
$$\mathbf{x} := \text{col}(d_B, v_B, d_C, v_C), \ \mathbf{u} := \text{col}(f_B, f_C), \ \mathbf{d} := \text{col}(v_A, \sin \theta, w),$$

 $\boldsymbol{\nu} := \text{col}(\nu_{dB}, \nu_{vB}, \nu_{dC}, \nu_{vC}), \ \mathbf{w} := \text{col}(\mathbf{d}, \boldsymbol{\nu}), \ \mathbf{y} := \text{col}(y_{dB}, y_{vB}, y_{dC}, y_{vC}), \ \text{and} \ \mathbf{e} := \text{col}(e_{dB}, e_{dC}).$

Then

$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, \mathbf{u}, \mathbf{w}) \quad \mathbf{x}(t_0) = \mathbf{x}_0$$

 $\mathbf{y} = \mathbf{h}(\mathbf{x}, \mathbf{u}, \mathbf{w})$
 $\mathbf{e} = \mathbf{h}_e(\mathbf{x}, \mathbf{u}, \mathbf{w})$

where

where
$$\mathbf{f}(\mathbf{x}, \mathbf{u}, \mathbf{w}) = egin{bmatrix} v_A - v_B & & & & & & \\ g \sin \theta + (f_B - 1/2\rho S_B C_{D_B} (v_B - w)^2) \ /m_B & & & & & & \\ v_B - v_C & & & & & \\ g \sin \theta + (f_C - 1/2\rho S_C C_{D_C} (v_C - w)^2) \ /m_C \end{bmatrix}$$

$$\mathbf{h}(\mathbf{x}, \mathbf{u}, \mathbf{w}) = \begin{bmatrix} d_B + \nu_{dB} \\ v_B + \nu_{vB} \\ d_C + \nu_{dC} \\ v_C + \nu_{vC} \end{bmatrix}$$

$$\mathbf{h}_e(\mathbf{x}, \mathbf{u}, \mathbf{w}) = \begin{bmatrix} y_{dB} - d_{\min} - y_{vB}^2/k \\ y_{dC} - d_{\min} - y_{vC}^2/k \end{bmatrix}.$$

Control Goals

- 1) Keep all the signals bounded
- 2) Asymptotically steer the regulated output to zero assuming constant disturbances.



Contents

- Active Suspensions
- Adaptive Cruise Control
- Wheel Speed Controls
- Electronic Stability Program / Torque Vectoring
- Self-Park Assist



Motivations & Goals How do ground vehicles move?





Motivations & Goals

How do ground vehicles move?

- Ground vehicle performance is related to the tire-road friction
- The tire-road friction depends on the wheel hub speed and the wheel rotational speed (among others such as temperature, road conditions, inflation pressure, tire wearing, vertical load, ...)
- Therefore, controlling the wheel rotational speed let us control the vehicle traction/braking forces



Goals

We can regulate the wheel speed to



Goals

We can regulate the wheel speed to

Aim at the desired traction performance (Traction Control)





Goals

We can regulate the wheel speed to

Aim at the desired traction performance (Traction Control)



Aim at the best traction performance (Launch Control)





Goals

We can regulate the wheel speed to

Aim at the desired traction performance (Traction Control)



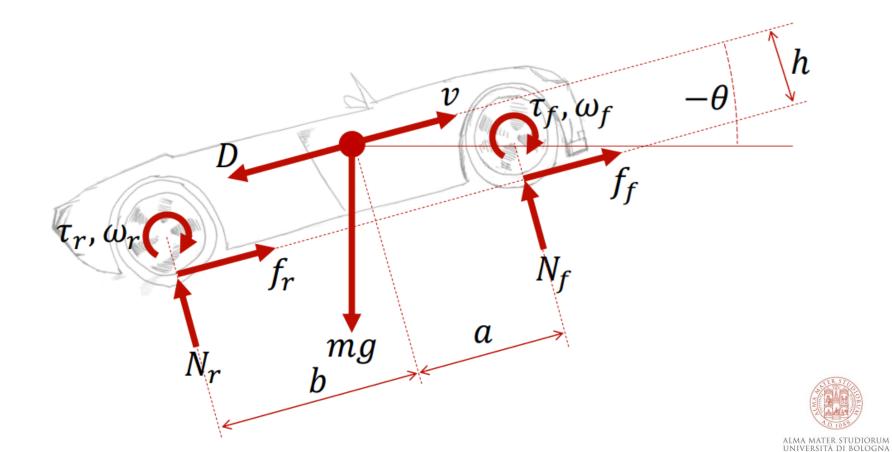
Aim at the best traction performance (Launch Control)



Aim at the best braking performance (Anti-lock Braking System)



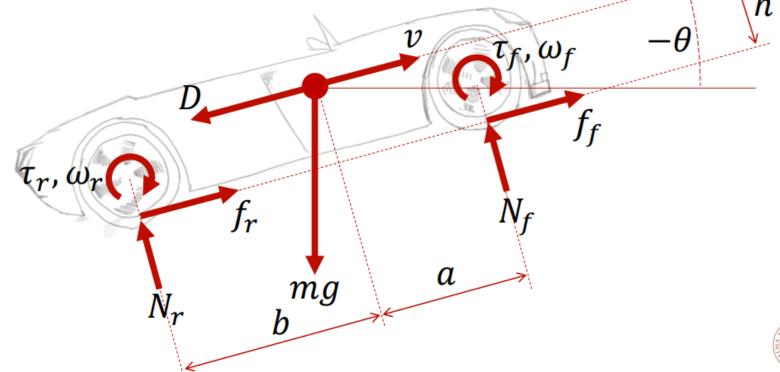




$$m\dot{v} = -mg\sin\theta + f_f + f_r - D$$
 longitudinal
$$0 = -mg\cos\theta + N_f + N_r$$
 vertical
$$0 = -(f_f + f_r)h - N_f a + N_r b$$
 rotational.

$$J_r \dot{\omega}_r = \tau_r - f_r r_r$$

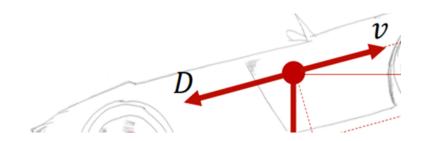
$$J_f \dot{\omega}_f = \tau_f - f_f r_f$$



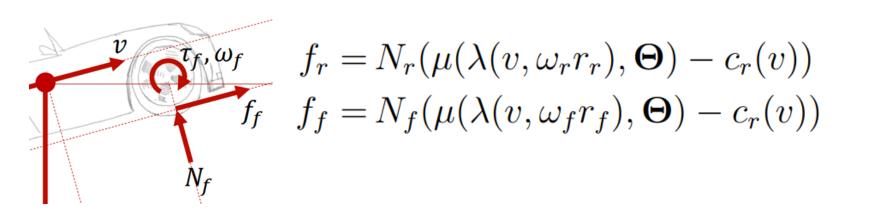


Aerodynamic Drag

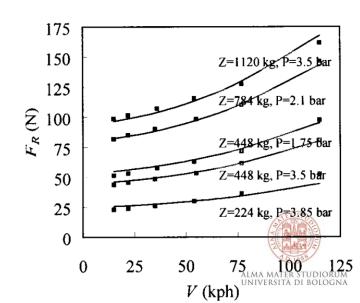
$$D(v - w) = \frac{1}{2}\rho S \frac{(v - w)^3}{|v - w|} C_D$$

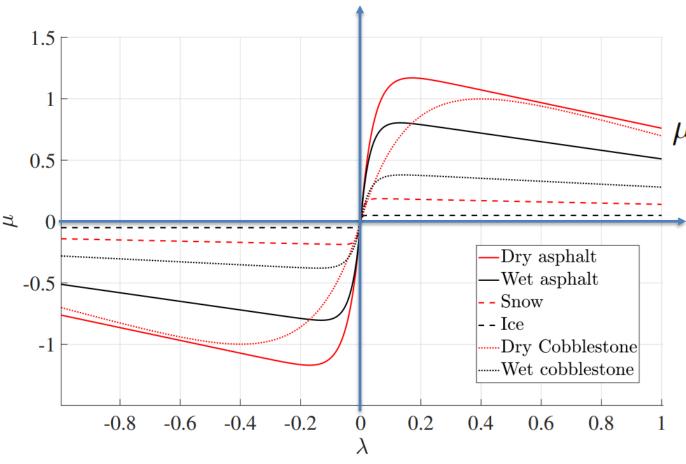


Tire Forces



Rolling Resistance
$$c_r(v) = c_{r0} + c_{r1}v + c_{r2}v^2$$





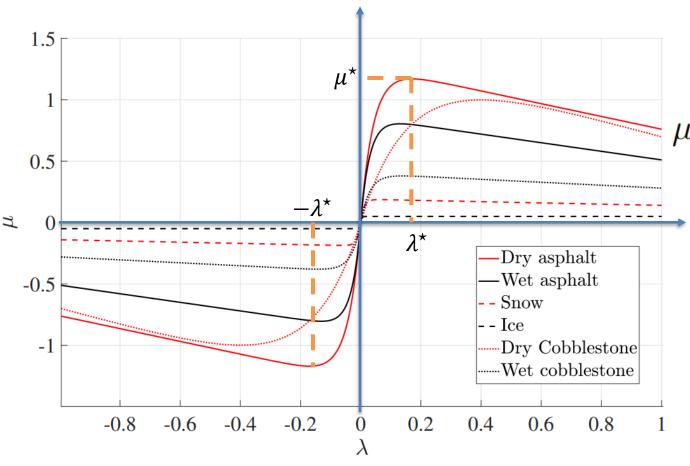
Traction Coefficient (Burckhardt)

$$\mu(\lambda, \mathbf{\Theta}) = \operatorname{sign}(\lambda)\theta_1 \left(1 - e^{-|\lambda|\theta_2}\right) - \lambda\theta_3$$
 $\mathbf{\Theta} := \operatorname{col}(\theta_1, \theta_2, \theta_3) \in \mathbb{R}^3$

Longitudinal Slip Ratio

$$\lambda(v, \omega r) = \frac{\omega r - v}{\epsilon + \max\{|v|, |\omega r|, |\omega_r - v|\}}$$





Traction Coefficient (Burckhardt)

$$\mu(\lambda, \mathbf{\Theta}) = \operatorname{sign}(\lambda)\theta_1 \left(1 - e^{-|\lambda|\theta_2}\right) - \lambda\theta_3$$
 $\mathbf{\Theta} := \operatorname{col}(\theta_1, \theta_2, \theta_3) \in \mathbb{R}^3$

Longitudinal Slip Ratio

$$\lambda(v, \omega r) = \frac{\omega r - v}{\epsilon + \max\{|v|, |\omega r|, |\omega_r - v|\}}$$



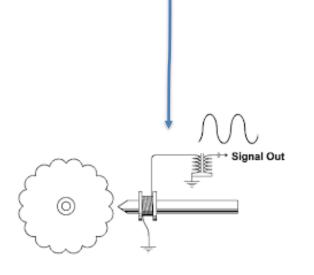
Sensors

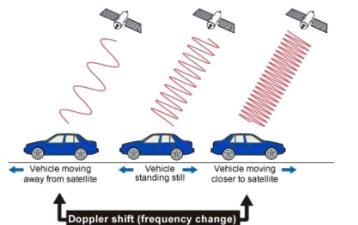
$$y_v = v + \nu_i$$

$$y_r = \omega_r (1 + \nu_r)$$

 $y_f = \omega_f (1 + \nu_f)$ front tonewheel

 $y_v = v + \nu_v$ GNSS receiver $y_r = \omega_r (1 + \nu_r)$ rear tonewheel





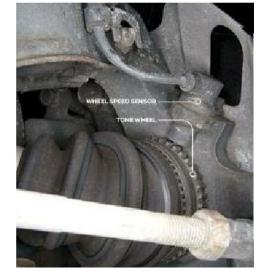
Velocity via Doppler Effect





Position via Trilateration





$$m\dot{v} = -mg\sin\theta + f_f + f_r - D$$
 longitudinal
 $0 = -mg\cos\theta + N_f + N_r$ vertical
 $0 = -(f_f + f_r)h - N_f a + N_r b$ rotational.



$$m\dot{v} = -mg\sin\theta + f_f + f_r - D \quad \text{longitudinal}$$

$$0 = -mg\cos\theta + N_f + N_r \quad \text{vertical}$$

$$0 = -(f_f + f_r)h - N_f a + N_r b \quad \text{rotational.}$$

$$\begin{bmatrix} N_r \\ N_f \end{bmatrix} = \frac{mg\cos\theta}{1 - (\mu_r - \mu_f)\bar{h}} \begin{bmatrix} (\mu_f - c_r)\bar{h} + \bar{a} \\ -(\mu_r - c_r)\bar{h} + \bar{b} \end{bmatrix}$$



$$m\dot{v} = -mg\sin\theta + f_f + f_r - D \quad \text{longitudinal}$$

$$0 = -mg\cos\theta + N_f + N_r \quad \text{vertical}$$

$$0 = -(f_f + f_r)h - N_f a + N_r b \quad \text{rotational.}$$

$$\begin{bmatrix} N_r \\ N_f \end{bmatrix} = \frac{mg\cos\theta}{1 - (\mu_r - \mu_f)\bar{h}} \begin{bmatrix} (\mu_f - c_r)\bar{h} + \bar{a} \\ -(\mu_r - c_r)\bar{h} + \bar{b} \end{bmatrix}$$

$$\dot{v} = -g\sin\theta + \frac{N_f}{m}(\mu_f - c_r) + \frac{N_r}{m}(\mu_r - c_r) - \frac{D}{m}$$



$$m\dot{v} = -mg\sin\theta + f_f + f_r - D$$
 longitudinal $0 = -mg\cos\theta + N_f + N_r$ vertical $0 = -(f_f + f_r)h - N_f a + N_r b$ rotational.

$$\begin{bmatrix} N_r \\ N_f \end{bmatrix} = \frac{mg\cos\theta}{1 - (\mu_r - \mu_f)\bar{h}} \begin{bmatrix} (\mu_f - c_r)\bar{h} + \bar{a} \\ -(\mu_r - c_r)\bar{h} + \bar{b} \end{bmatrix}$$

$$\dot{v} = -g\sin\theta + \frac{N_f}{m}(\mu_f - c_r) + \frac{N_r}{m}(\mu_r - c_r) - \frac{D}{m}$$

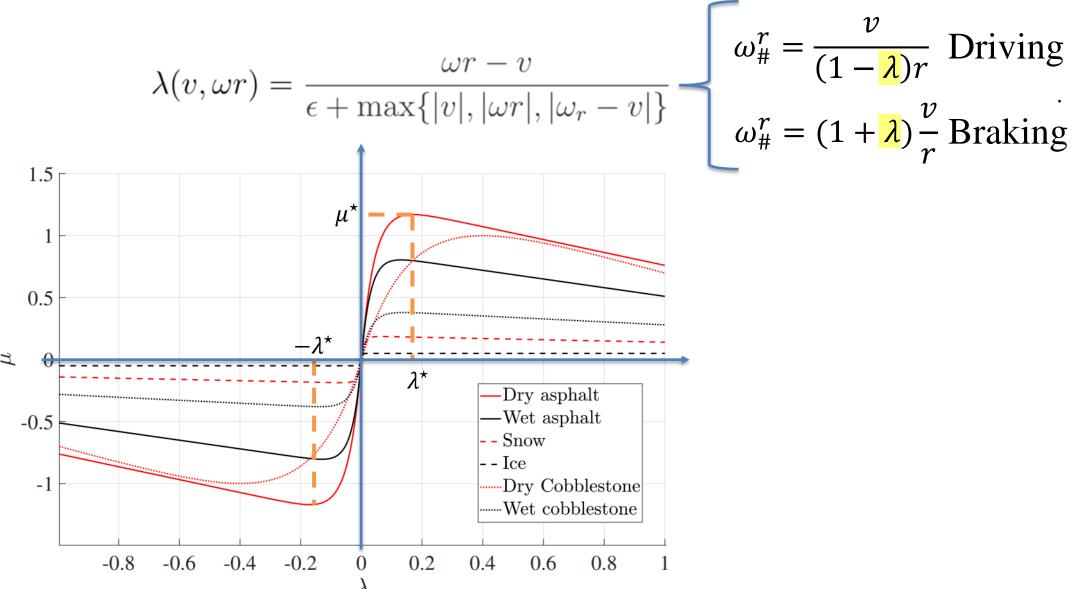
$$\dot{\omega}_r = (\tau_r - f_r r_r)/J_r$$

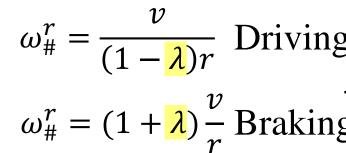
$$\dot{\omega}_f = (\tau_f - f_f r_f)/J_f$$

$$f_r = N_r(\mu(\lambda(v, \omega_r r_r), \mathbf{\Theta}) - c_r(v))$$

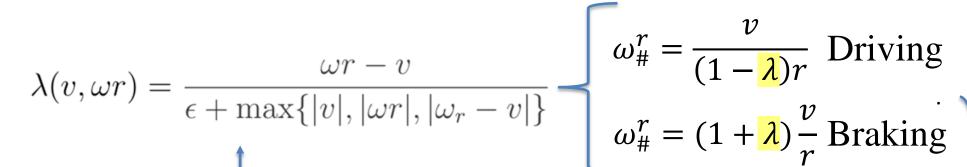
$$f_f = N_f(\mu(\lambda(v, \omega_f r_f), \mathbf{\Theta}) - c_r(v))$$

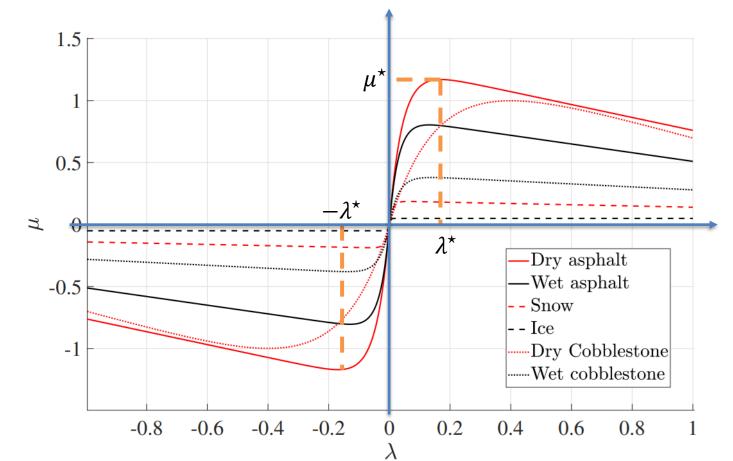












$$\omega_{\#}^{r} = \frac{v}{(1-\lambda)r}$$
 Driving

$$\omega_{\#}^{r} = (1 + \lambda) \frac{v}{r}$$
 Braking

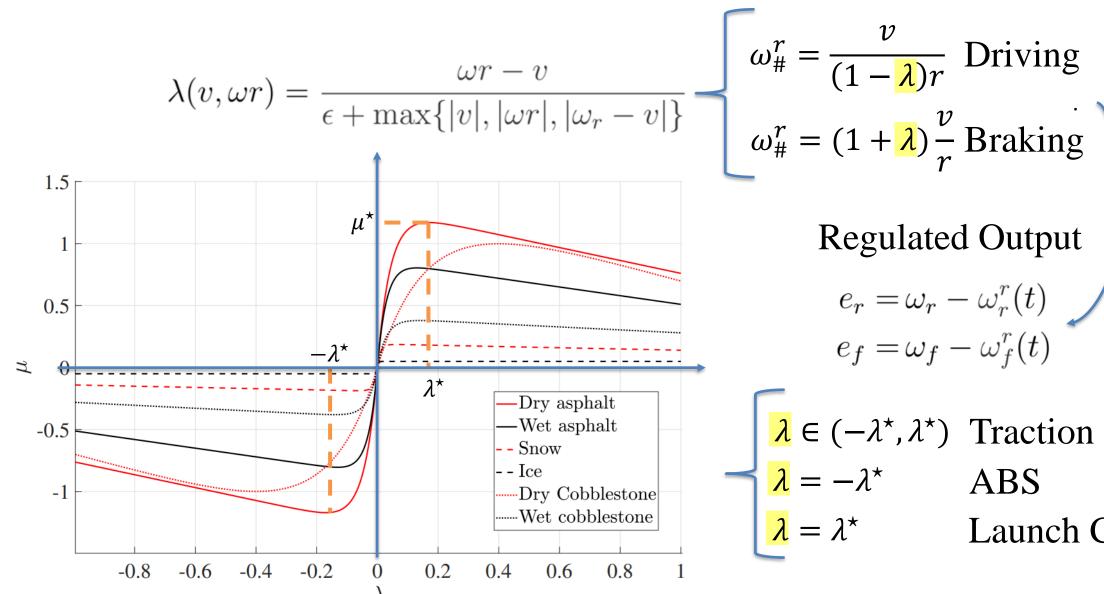
Regulated Output

$$e_r = \omega_r - \omega_r^r(t)$$

$$e_r = \omega_r - \omega_r^r(t)$$

$$e_f = \omega_f - \omega_f^r(t)$$





$$\omega_{\#}^{r} = \frac{v}{(1-\lambda)r}$$
 Driving

$$\omega_{\#}^{r} = (1 + \lambda) \frac{v}{r}$$
Braking

Regulated Output

$$e_r = \omega_r - \omega_r^r(t)$$

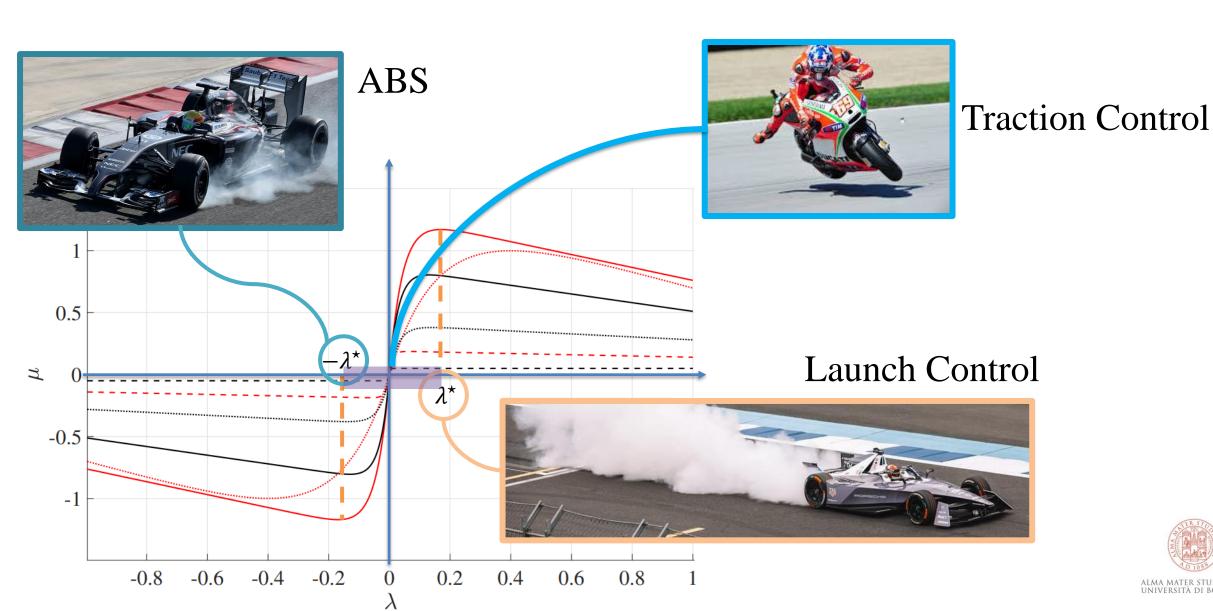
$$e_f = \omega_f - \omega_f^r(t)$$

$$\lambda \in (-\lambda^*, \lambda^*)$$
 Traction Control

$$\lambda = -\lambda^*$$

$$\lambda = \lambda^*$$

 $\lambda = -\lambda^*$ ABS $\lambda = \lambda^*$ Launch Control





let
$$\mathbf{x} := \operatorname{col}(v, \omega_r, \omega_f)$$
, $\mathbf{u} := \operatorname{col}(\tau_r, \tau_f)$, $\mathbf{w} := \operatorname{col}(\theta, w, \nu_v, \nu_r, \nu_f, \omega_r^r, \omega_f^r)$
 $\mathbf{y} := \operatorname{col}(v, \omega_r, \omega_f)$, and $\mathbf{e} := \operatorname{col}(e_r, e_f)$
 $\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, \mathbf{u}, \mathbf{w})$ $\mathbf{x}(t_0) = \mathbf{x}_0$
 $\mathbf{y} = \mathbf{h}(\mathbf{x}, \mathbf{u}, \mathbf{w})$
 $\mathbf{e} = \mathbf{h}_e(\mathbf{x}, \mathbf{u}, \mathbf{w})$

$$\mathbf{f}(\mathbf{x}, \mathbf{u}, \mathbf{w}) = \begin{bmatrix} \frac{N_f(\mu_f - c_r) + N_r(\mu_r - c_r)}{m} - g\sin\theta - \frac{D}{m} \\ J_r^{-1}(\tau_r - N_r r_r(\mu_r - c_r)) \\ J_f^{-1}(\tau_f - N_f r_f(\mu_f - c_r)) \end{bmatrix}$$

$$\mathbf{h}(\mathbf{x}, \mathbf{u}, \mathbf{w}) = \begin{bmatrix} v + \nu_v \\ \omega_r (1 + \nu_r) \\ \omega_f (1 + \nu_f) \end{bmatrix}, \quad \mathbf{h}_e(\mathbf{x}, \mathbf{u}, \mathbf{w}) = \begin{bmatrix} \omega_r (1 + \nu_r) - \omega_r^r \\ \omega_f (1 + \nu_f) - \omega_f^r \end{bmatrix}$$



let
$$\mathbf{x} := \operatorname{col}(v, \omega_r, \omega_f)$$
, $\mathbf{u} := \operatorname{col}(\tau_r, \tau_f)$, $\mathbf{w} := \operatorname{col}(\theta, w, \nu_v, \nu_r, \nu_f, \omega_r^r, \omega_f^r)$
 $\mathbf{y} := \operatorname{col}(v, \omega_r, \omega_f)$, and $\mathbf{e} := \operatorname{col}(e_r, e_f)$
 $\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, \mathbf{u}, \mathbf{w}) \quad \mathbf{x}(t_0) = \mathbf{x}_0$

$$\mathbf{f}(\mathbf{x},\mathbf{u},\mathbf{w}) = \begin{bmatrix} \frac{N_f(\mu_f - c_r) + N_r(\mu_r - c_r)}{m} - g\sin\theta - \frac{D}{m} \\ J_r^{-1} \left(\tau_r - N_r r_r(\mu_r - c_r)\right) \\ J_f^{-1} \left(\tau_f - N_f r_f(\mu_f - c_r)\right) \end{bmatrix}$$
 the regulated output to zero assuming constant disturbances.

y = h(x, u, w)

 $e = h_e(x, u, w)$

Control Goals

- 1) Keep all the signals bounded
- 2) Asymptotically steer

$$\mathbf{h}(\mathbf{x}, \mathbf{u}, \mathbf{w}) = \begin{bmatrix} v + \nu_v \\ \omega_r (1 + \nu_r) \\ \omega_f (1 + \nu_f) \end{bmatrix}, \quad \mathbf{h}_e(\mathbf{x}, \mathbf{u}, \mathbf{w}) = \begin{bmatrix} \omega_r (1 + \nu_r) - \omega_r^r \\ \omega_f (1 + \nu_f) - \omega_f^r \end{bmatrix}$$



Contents

- Active Suspensions
- Adaptive Cruise Control
- Wheel Speed Controls
- Electronic Stability Program / Torque Vectoring
- Self-Park Assist



ESP/TV

Motivations & Goals
Improve safety (through stabilisation: ESP)





ESP/TV

Motivations & Goals

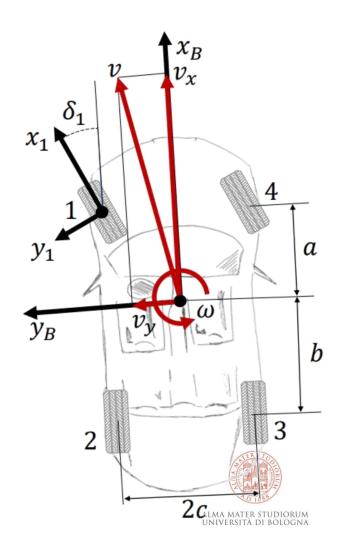
Improve safety (through reference tracking: Torque Vectoring)





let $v_x, v_y \in \mathbb{R}$ be the vehicle inertial speed expressed in the body axes, let $\omega \in \mathbb{R}$ be the yaw rate,

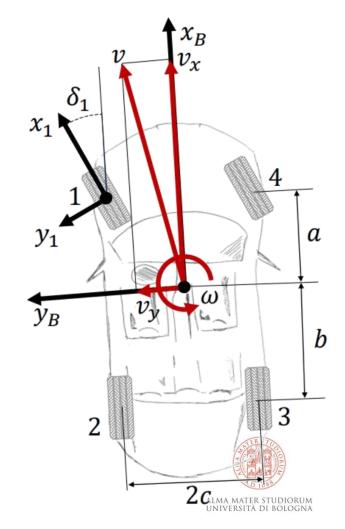
$$\begin{bmatrix} \dot{v}_x \\ \dot{v}_y \\ \dot{\omega} \end{bmatrix} = \begin{bmatrix} \omega v_y \\ -\omega v_x \\ 0 \end{bmatrix} + \begin{bmatrix} (f_x - D(v_x - w))/m \\ f_y/m \\ \tau/J \end{bmatrix}$$



let $v_x, v_y \in \mathbb{R}$ be the vehicle inertial speed expressed in the body axes, let $\omega \in \mathbb{R}$ be the yaw rate,

$$\begin{bmatrix} \dot{v}_x \\ \dot{v}_y \\ \dot{\omega} \end{bmatrix} = \begin{bmatrix} \omega v_y \\ -\omega v_x \\ 0 \end{bmatrix} + \begin{bmatrix} (f_x - D(v_x - w))/m \\ f_y/m \\ \tau/J \end{bmatrix}$$

Aerodynamic Drag $D(v-w) = \frac{1}{2}\rho S \frac{(v-w)^3}{|v-w|} C_D$

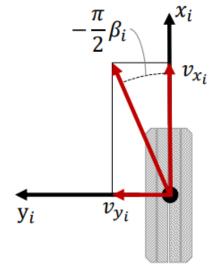


let $v_x, v_y \in \mathbb{R}$ be the vehicle inertial speed expressed in the body axes, let $\omega \in \mathbb{R}$ be the yaw rate,

$$\begin{bmatrix} \dot{v}_x \\ \dot{v}_y \\ \dot{\omega} \end{bmatrix} = \begin{bmatrix} \omega v_y \\ -\omega v_x \\ 0 \end{bmatrix} + \begin{bmatrix} (f_x - D(v_x - w))/m \\ f_y/m \\ \tau/J \end{bmatrix} - \frac{\pi}{2} \beta_i$$

$$\begin{array}{c} \chi_i \\ v_{\chi_i} \end{array}$$

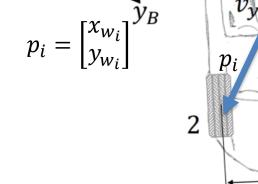
$$\begin{array}{c} \lambda_B \\ v_{\chi} \end{array}$$
Aerodynamic Drag
$$D(v - w) = \frac{1}{2} \rho S \frac{(v - w)^3}{|v - w|} C_D$$



 \boldsymbol{a}

$$\begin{bmatrix} f_x \\ f_y \\ \tau \end{bmatrix} = \sum_{i=1}^4 N_i \mu(\lambda_i, \mathbf{\Theta}_i) \begin{bmatrix} \cos \delta_i \\ \sin \delta_i \\ x_{w_i} \sin \delta_i - y_{w_i} \cos \delta_i \end{bmatrix} \qquad p_i = \begin{bmatrix} x_{w_i} \\ y_{w_i} \end{bmatrix}^{y_B}$$

$$+ \sum_{i=1}^4 N_i \mu(\beta_i, \mathbf{\Theta}_i) \begin{bmatrix} -\sin \delta_i \\ \cos \delta_i \\ x_{w_i} \cos \delta_i + y_{w_i} \sin \delta_i \end{bmatrix} \qquad 2$$

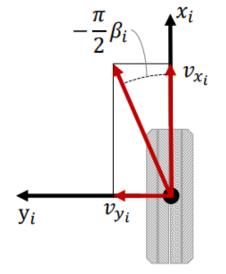


let $v_x, v_y \in \mathbb{R}$ be the vehicle inertial speed expressed in the body axes, let $\omega \in \mathbb{R}$ be the yaw rate,

$$\begin{bmatrix} \dot{v}_x \\ \dot{v}_y \\ \dot{\omega} \end{bmatrix} = \begin{bmatrix} \omega v_y \\ -\omega v_x \\ 0 \end{bmatrix} + \begin{bmatrix} (f_x - D(v_x - w))/m \\ f_y/m \\ \tau/J \end{bmatrix} - \frac{\pi}{2} \beta_i$$

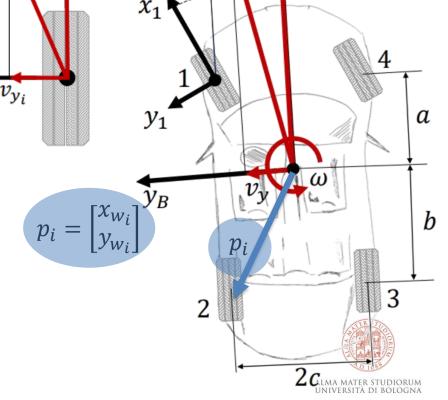
$$\begin{array}{c} \chi_i \\ v_{\chi_i} \end{array}$$

$$\begin{array}{c} \lambda_B \\ v_{\chi} \end{array}$$
Aerodynamic Drag
$$D(v - w) = \frac{1}{2} \rho S \frac{(v - w)^3}{|v - w|} C_D$$



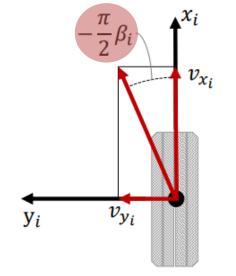
$$\begin{bmatrix} f_x \\ f_y \\ \tau \end{bmatrix} = \sum_{i=1}^4 N_i \mu(\lambda_i, \mathbf{\Theta}_i) \begin{bmatrix} \cos \delta_i \\ \sin \delta_i \\ x_{w_i} \sin \delta_i - y_{w_i} \cos \delta_i \end{bmatrix} \qquad p_i = \begin{bmatrix} x_{w_i} \\ y_{w_i} \end{bmatrix}^{y_B}$$

$$+ \sum_{i=1}^4 N_i \mu(\beta_i, \mathbf{\Theta}_i) \begin{bmatrix} -\sin \delta_i \\ \cos \delta_i \\ x_{w_i} \cos \delta_i + y_{w_i} \sin \delta_i \end{bmatrix}$$



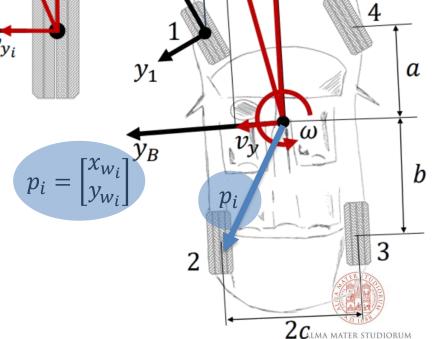
let $v_x, v_y \in \mathbb{R}$ be the vehicle inertial speed expressed in the body axes, let $\omega \in \mathbb{R}$ be the yaw rate,

$$\begin{bmatrix} \dot{v}_x \\ \dot{v}_y \\ \dot{\omega} \end{bmatrix} = \begin{bmatrix} \omega v_y \\ -\omega v_x \\ 0 \end{bmatrix} + \begin{bmatrix} (f_x - D(v_x - w))/m \\ f_y/m \\ \tau/J \end{bmatrix}$$
Aerodynamic Drag
$$D(v - w) = \frac{1}{2}\rho S \frac{(v - w)^3}{|v - w|} C_D$$



$$\begin{bmatrix} f_x \\ f_y \\ \tau \end{bmatrix} = \sum_{i=1}^4 N_i \mu(\lambda_i, \mathbf{\Theta}_i) \begin{bmatrix} \cos \delta_i \\ \sin \delta_i \\ x_{w_i} \sin \delta_i - y_{w_i} \cos \delta_i \end{bmatrix} \qquad p_i = \begin{bmatrix} x_{w_i} \\ y_{w_i} \end{bmatrix}$$

$$+ \sum_{i=1}^4 N_i \mu(\boldsymbol{\beta}_i, \mathbf{\Theta}_i) \begin{bmatrix} -\sin \delta_i \\ \cos \delta_i \\ \cos \delta_i \end{bmatrix}$$



$$\begin{bmatrix} f_x \\ f_y \\ \tau \end{bmatrix} = \sum_{i=1}^4 N_i \mu(\lambda_i, \mathbf{\Theta}_i) \begin{bmatrix} \cos \delta_i \\ \sin \delta_i \\ x_{w_i} \sin \delta_i - y_{w_i} \cos \delta_i \end{bmatrix} + \sum_{i=1}^4 N_i \mu(\beta_i, \mathbf{\Theta}_i) \begin{bmatrix} -\sin \delta_i \\ \cos \delta_i \\ x_{w_i} \cos \delta_i + y_{w_i} \sin \delta_i \end{bmatrix}$$



$$\sum_{i=1}^{4} \begin{bmatrix} 1 & 1 \\ y_{w_i} + h(\mu(\beta_i, \mathbf{\Theta}_i)\cos\delta_i + \mu(\lambda_i, \mathbf{\Theta}_i)\sin\delta_i) \\ h(\mu(\lambda_i, \mathbf{\Theta}_i)\cos\delta_i - \mu(\beta_i, \mathbf{\Theta}_i)\sin\delta_i) + x_{w_i} \end{bmatrix} N_i = \begin{bmatrix} mg \\ 0 \\ 0 \end{bmatrix}$$

$$\begin{bmatrix} f_x \\ f_y \\ \tau \end{bmatrix} = \sum_{i=1}^4 N_i \mu(\lambda_i, \mathbf{\Theta}_i) \begin{bmatrix} \cos \delta_i \\ \sin \delta_i \\ x_{w_i} \sin \delta_i - y_{w_i} \cos \delta_i \end{bmatrix} + \sum_{i=1}^4 N_i \mu(\beta_i, \mathbf{\Theta}_i) \begin{bmatrix} -\sin \delta_i \\ \cos \delta_i \\ x_{w_i} \cos \delta_i + y_{w_i} \sin \delta_i \end{bmatrix}$$



$$\mathbf{v}_{i} := \begin{bmatrix} 1 \\ y_{w_{i}} + h(\mu(\beta_{i})\cos\delta_{i} + \mu(\lambda_{i})\sin\delta_{i}) \\ h(\mu(\lambda_{i})\cos\delta_{i} - \mu(\beta_{i})\sin\delta_{i}) + x_{w_{i}} \end{bmatrix} \quad \mathbf{H} := \begin{bmatrix} \mathbf{v}_{1} & \dots & \mathbf{v}_{n} \end{bmatrix}$$

$$\sum_{i=1}^{4} \begin{bmatrix} 1 & 1 \\ y_{w_i} + h(\mu(\beta_i, \mathbf{\Theta}_i) \cos \delta_i + \mu(\lambda_i, \mathbf{\Theta}_i) \sin \delta_i) \\ h(\mu(\lambda_i, \mathbf{\Theta}_i) \cos \delta_i - \mu(\beta_i, \mathbf{\Theta}_i) \sin \delta_i) + x_{w_i} \end{bmatrix} N_i = \begin{bmatrix} mg \\ 0 \\ 0 \end{bmatrix}$$

$$\begin{bmatrix} f_x \\ f_y \\ \tau \end{bmatrix} = \sum_{i=1}^4 N_i \mu(\lambda_i, \mathbf{\Theta}_i) \begin{bmatrix} \cos \delta_i \\ \sin \delta_i \\ x_{w_i} \sin \delta_i - y_{w_i} \cos \delta_i \end{bmatrix} + \sum_{i=1}^4 N_i \mu(\beta_i, \mathbf{\Theta}_i) \begin{bmatrix} -\sin \delta_i \\ \cos \delta_i \\ x_{w_i} \cos \delta_i + y_{w_i} \sin \delta_i \end{bmatrix}$$



$$\mathbf{v}_{i} := \begin{bmatrix} 1 \\ y_{w_{i}} + h(\mu(\beta_{i})\cos\delta_{i} + \mu(\lambda_{i})\sin\delta_{i}) \\ h(\mu(\lambda_{i})\cos\delta_{i} - \mu(\beta_{i})\sin\delta_{i}) + x_{w_{i}} \end{bmatrix} \quad \mathbf{H} := \begin{bmatrix} \mathbf{v}_{1} & \dots & \mathbf{v}_{n} \end{bmatrix} \quad \mathbf{H} \begin{bmatrix} N_{1} \\ \vdots \\ N_{4} \end{bmatrix} = \begin{bmatrix} mg \\ 0 \\ 0 \end{bmatrix}$$

$$\sum_{i=1}^{4} \begin{bmatrix} 1 \\ y_{w_i} + h(\mu(\beta_i, \mathbf{\Theta}_i)\cos\delta_i + \mu(\lambda_i, \mathbf{\Theta}_i)\sin\delta_i) \\ h(\mu(\lambda_i, \mathbf{\Theta}_i)\cos\delta_i - \mu(\beta_i, \mathbf{\Theta}_i)\sin\delta_i) + x_{w_i} \end{bmatrix} N_i = \begin{bmatrix} mg \\ 0 \\ 0 \end{bmatrix}$$
Vertical translation Roll Pitch

$$\begin{bmatrix} f_x \\ f_y \\ \tau \end{bmatrix} = \sum_{i=1}^4 N_i \mu(\lambda_i, \mathbf{\Theta}_i) \begin{bmatrix} \cos \delta_i \\ \sin \delta_i \\ x_{w_i} \sin \delta_i - y_{w_i} \cos \delta_i \end{bmatrix} + \sum_{i=1}^4 N_i \mu(\beta_i, \mathbf{\Theta}_i) \begin{bmatrix} -\sin \delta_i \\ \cos \delta_i \\ x_{w_i} \cos \delta_i + y_{w_i} \sin \delta_i \end{bmatrix}$$



$$\mathbf{v}_{i} := \begin{bmatrix} 1 \\ y_{w_{i}} + h(\mu(\beta_{i})\cos\delta_{i} + \mu(\lambda_{i})\sin\delta_{i}) \\ h(\mu(\lambda_{i})\cos\delta_{i} - \mu(\beta_{i})\sin\delta_{i}) + x_{w_{i}} \end{bmatrix} \quad \mathbf{H} := \begin{bmatrix} \mathbf{v}_{1} & \dots & \mathbf{v}_{n} \end{bmatrix} \quad \mathbf{H} \begin{bmatrix} N_{1} \\ \vdots \\ N_{4} \end{bmatrix} = \begin{bmatrix} mg \\ 0 \\ 0 \end{bmatrix}$$

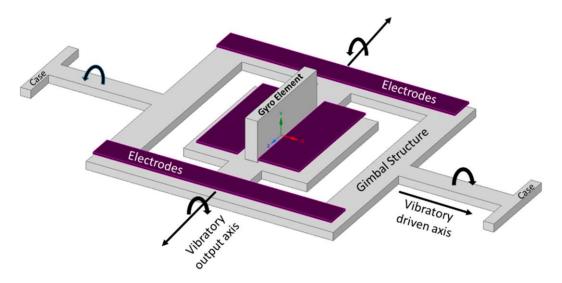
$$\begin{bmatrix} N_1 \\ \vdots \\ N_4 \end{bmatrix} = \mathbf{H}^\top \begin{bmatrix} \mathbf{H} \mathbf{H}^\top \end{bmatrix}^{-1} \begin{bmatrix} mg \\ 0 \\ 0 \end{bmatrix} \longleftarrow$$

$$\begin{bmatrix} f_x \\ f_y \\ \tau \end{bmatrix} = \sum_{i=1}^4 N_i \mu(\lambda_i, \mathbf{\Theta}_i) \begin{bmatrix} \cos \delta_i \\ \sin \delta_i \\ x_{w_i} \sin \delta_i - y_{w_i} \cos \delta_i \end{bmatrix} + \sum_{i=1}^4 N_i \mu(\beta_i, \mathbf{\Theta}_i) \begin{bmatrix} -\sin \delta_i \\ \cos \delta_i \\ x_{w_i} \cos \delta_i + y_{w_i} \sin \delta_i \end{bmatrix}$$



Sensor (Gyroscope)

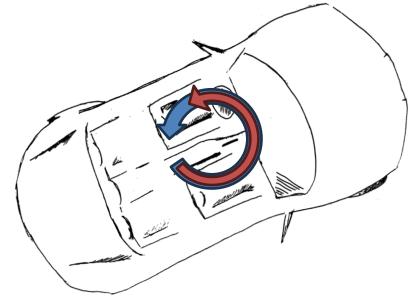
$$y = \omega + \nu$$
.



Micro-ElectroMechanical Systems (MEMS)

Regulated output

$$e := \omega - r(t)$$
.





ESP/TV Problem Formulation

Let
$$\mathbf{x} := \operatorname{col}(v_x, v_y, \omega), \ \mathbf{u} := \operatorname{col}(\delta_1, \dots, \delta_4, \omega_1, \dots, \omega_4)$$

 $\mathbf{d} := \operatorname{col}(w, \mathbf{\Theta}_1, \mathbf{\Theta}_2, \mathbf{\Theta}_3, \mathbf{\Theta}_4), \ \mathbf{w} := \operatorname{col}(\mathbf{d}, \nu, r)$

Then

$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, \mathbf{u}, \mathbf{w}) \quad \mathbf{x}(t_0) = \mathbf{x}_0$$
 $y = h(\mathbf{x}, \mathbf{u}, \mathbf{w})$
 $e = h_e(\mathbf{x}, \mathbf{u}, \mathbf{w})$

with

$$\mathbf{f}(\mathbf{x}, \mathbf{u}, \mathbf{w}) := \begin{bmatrix} \omega v_y \\ -\omega v_x \\ 0 \end{bmatrix} + \begin{bmatrix} (f_x - D(v_x - w))/m \\ f_y/m \\ \tau/J \end{bmatrix}$$

$$h(\mathbf{x}, \mathbf{u}, \mathbf{w}) := \omega + \nu$$

$$h_e(\mathbf{x}, \mathbf{u}, \mathbf{w}) := y - r(t).$$

Control Goals

- 1) Keep all the signals bounded
- 2) Asymptotically steer the regulated output to zero assuming constant disturbances.



Contents

- Active Suspensions
- Adaptive Cruise Control
- Wheel Speed Controls
- Electronic Stability Program / Torque Vectoring
- Self-Park Assist



Self-Park Assist

Motivations & Goals Facilitate Parking





Self-Park Assist

Motivations & Goals Facilitate Parking

Jokes apart! Self-Park Assist is fundamental for impaired people



Self-Park Assist 2D Model

Bicycle model with Ackerman Steering

$$v_x = V \cos \delta$$

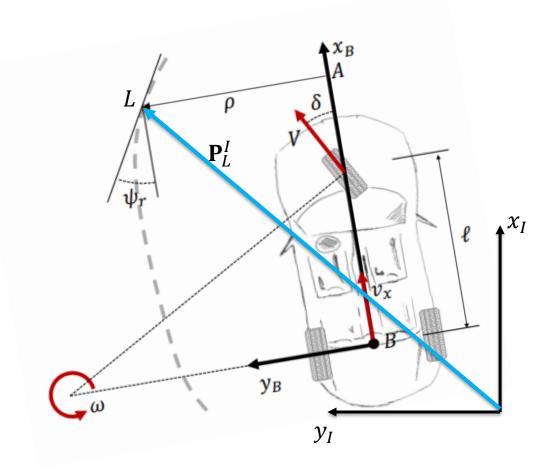
$$v_y = 0$$

$$\omega = \frac{V}{\ell} \sin \delta$$

Let $\chi:\mathbb{R}^2 \to \mathbb{R}$ be the line curvature. Then

$$\dot{\rho} = V \left(\tan \psi_r \left(\frac{\rho}{\ell} \sin \delta - \cos \delta \right) - \frac{x_A}{\ell} \sin \delta \right)$$

$$\dot{\psi}_r = V \left(\frac{\sin \delta}{\ell} \left(1 + \rho \frac{\chi(\mathbf{p}_L^I)}{\cos \psi_r} \right) - \chi(\mathbf{p}_L^I) \frac{\cos \delta}{\cos \psi_r} \right)$$





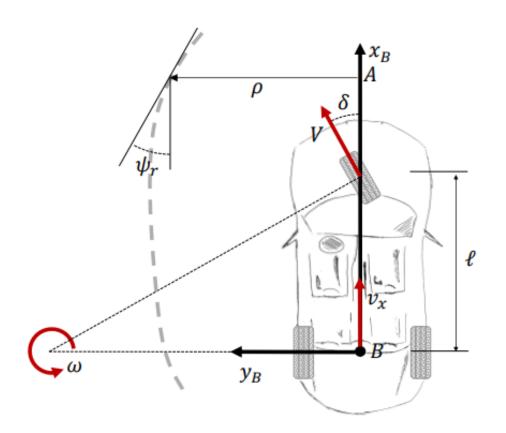
Self-Park Assist 2D Model

We assume ho and ψ_r are estimated and χ is known

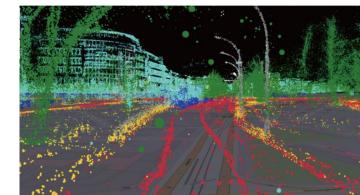
$$\mathbf{y} = \operatorname{col}(\rho + \nu_{\rho}, \psi_r + \nu_{\psi}, \chi)$$



We assume the reference path is generated by a planner



We assume \mathbf{p}_B^I and \mathbf{p}_A^I are obtained via SLAM



Self-Park Assist Problem Formulation

Let

$$\mathbf{x} \coloneqq \operatorname{col}(\rho, \psi_r)$$

$$u \coloneqq \delta$$

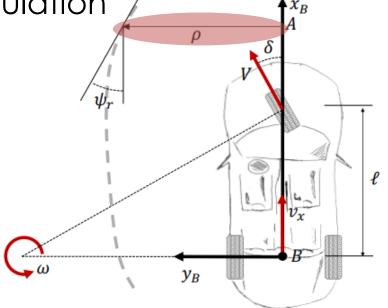
$$d \coloneqq \chi$$

$$\mathbf{v} \coloneqq \operatorname{col}(\nu_\rho, \nu_\psi)$$

$$\mathbf{w} \coloneqq \operatorname{col}(d, \mathbf{v})$$

$$\mathbf{y} = \operatorname{col}(\rho, \psi_r, d) + \mathbf{v}$$

$$e \coloneqq \rho$$



Control Goals

- 1) Keep all the signals bounded
- 2) Asymptotically steer the regulated output to zero assuming constant disturbances.

$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, u, \mathbf{w}) \quad \mathbf{x}(t_0) = \mathbf{x}_0$$

 $\mathbf{y} = \mathbf{h}(\mathbf{x}, u, \mathbf{w})$
 $e = h_e(\mathbf{x}, u, \mathbf{w})$

$$\mathbf{f}(\mathbf{x}, u, \mathbf{w}) = V \begin{bmatrix} \tan \psi_r \left(\frac{\rho}{\ell} \sin \delta - \cos \delta \right) - \frac{x_A}{\ell} \sin \delta \\ \frac{\sin \delta}{\ell} \left(1 + \rho \frac{\chi(\mathbf{p}_L^I)}{\cos \psi_r} \right) - \chi(\mathbf{p}_L^I) \frac{\cos \delta}{\cos \psi_r} \end{bmatrix}$$

$$\mathbf{h}(\mathbf{x}, u, \mathbf{w}) := \operatorname{col}(\rho + \nu_{\rho}, \psi_{r} + \nu_{\psi}, d)$$

$$h_e(\mathbf{x}, u, \mathbf{w}) := \rho + \nu_{\rho}$$





Nicola Mimmo

Department of Electrical, Electronics and Information Engineering «G. Marconi»

nicola.mimmo2@unibo.it