

# Broad Trajectory Searches Using Monte Carlo Tree Search with the Inclusion of $\Delta$ VEGA Trajectories



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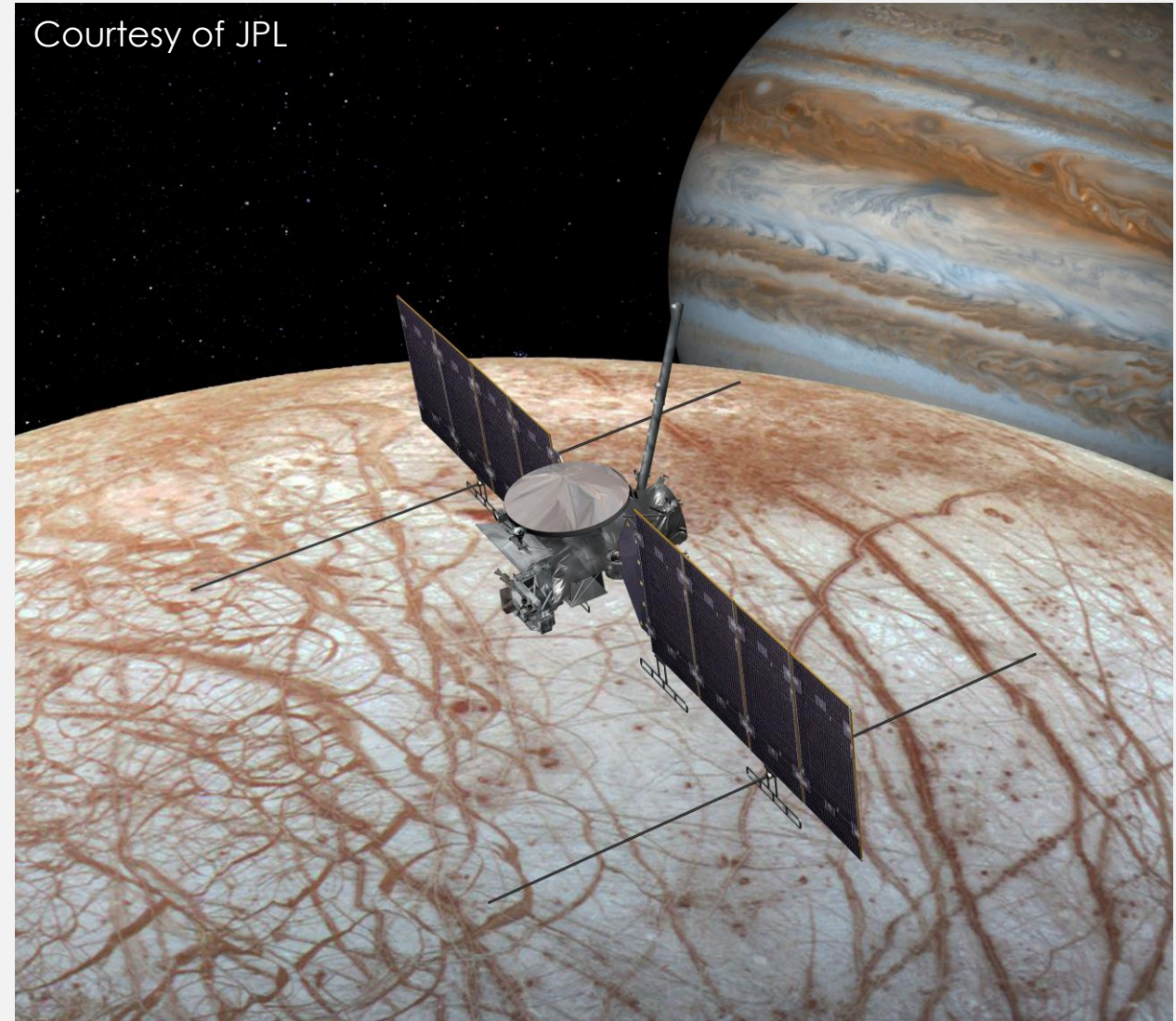
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# Introduction

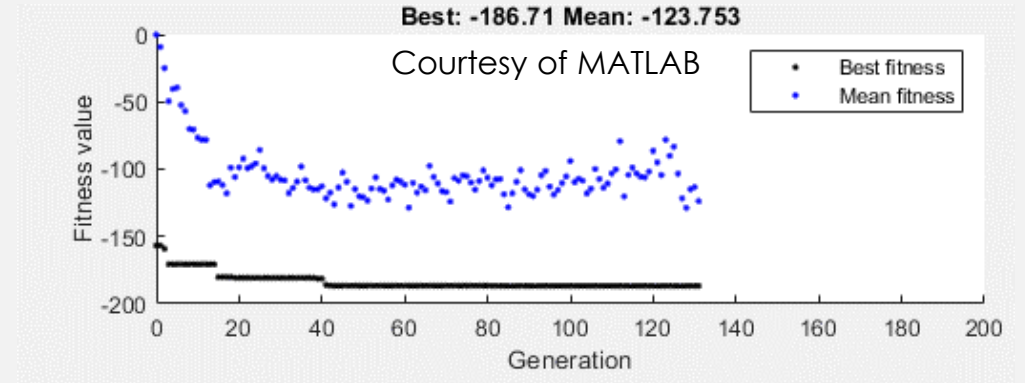
- Solution Space Complexity
  - With each additional flyby, another dimension is added to the solution space
- Broad Search Role
  - Reduce solution space to only areas of interest
  - **Solve for sequences** with rough timings to be converged in later steps



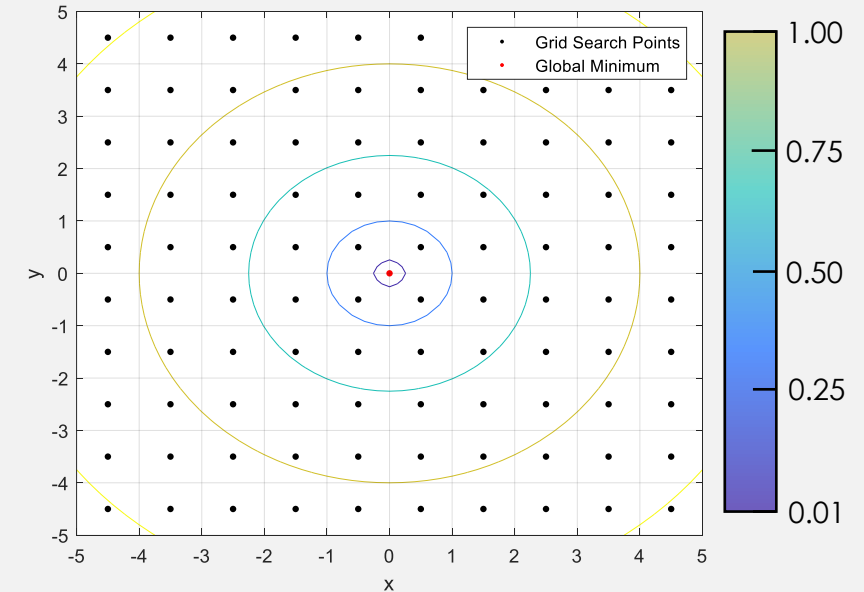
Broad searches are one of the first steps in trajectory design, by creating regions of interest

# Previous Broad Search Algorithms

- Evolutionary Algorithms
  - Improves ability to search after each generation
  - Examples:
    - Particle Swarm Optimization
    - Differential Evolution
    - Genetic Algorithms
- Grid Search
  - Evenly Searches Solution Space
  - Examples:
    - Beam Search
    - Breadth-First Search
    - Depth-First Search
    - Lazy Race Tree Search



Genetic Algorithm Training



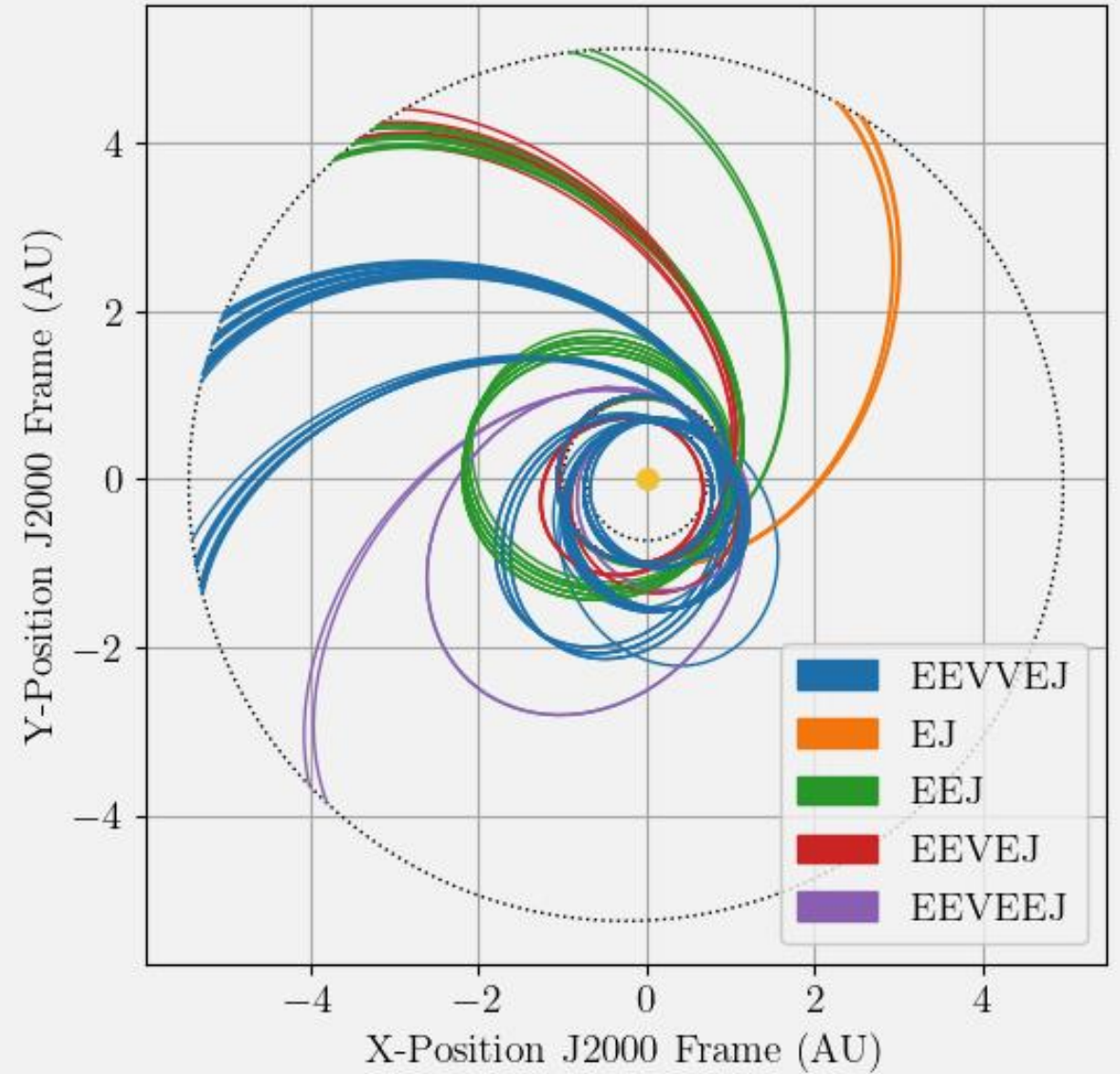
Grid Search Dispersion

Grid searches are computationally expensive, but using heuristics reduces the cost



# Objectives

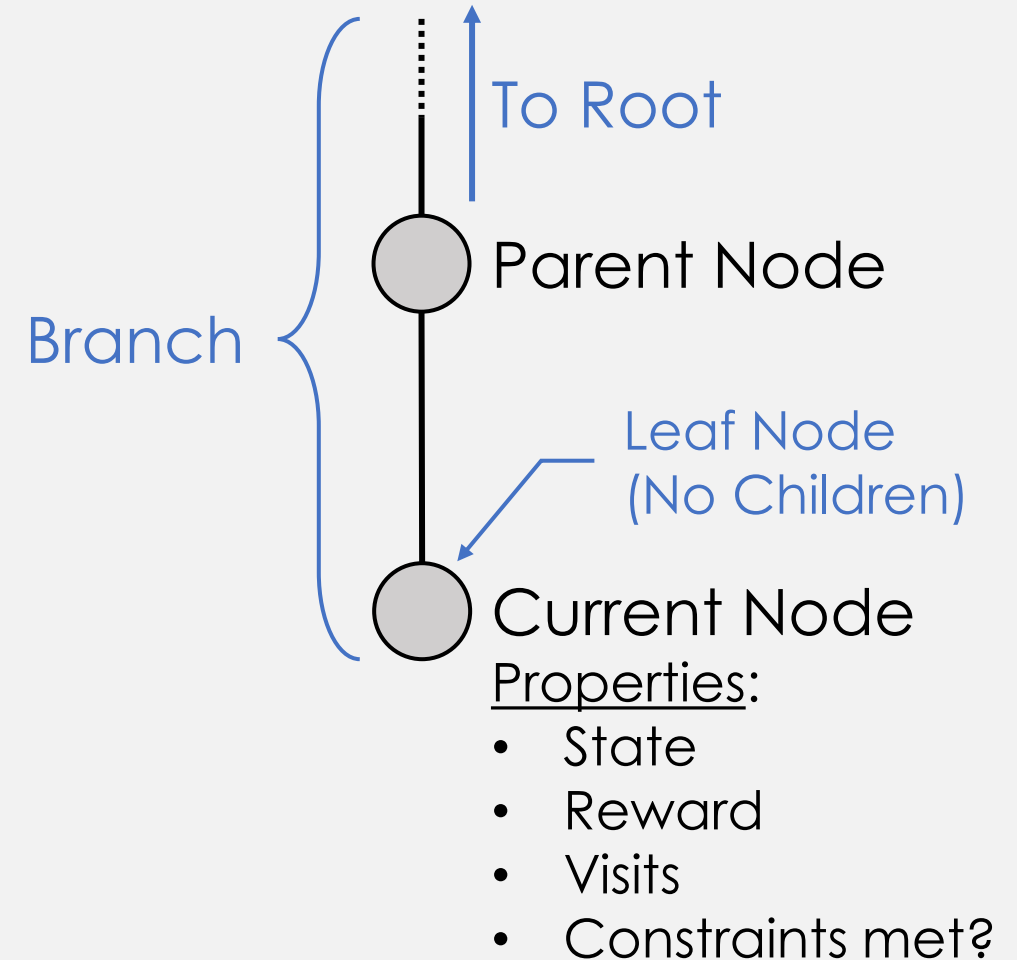
- Create a tool to **find multi-flyby sequences** given solution space
- Rank solutions based off their viability for optimization
- Include the option for a  $\Delta$ VEGA orbit in the solution space



The inclusion of  $\Delta$ VEGA orbits provides a diverse set of possible trajectories to explore

# Intro to **M**onte **C**arlo **T**ree **S**earches (MCTS)

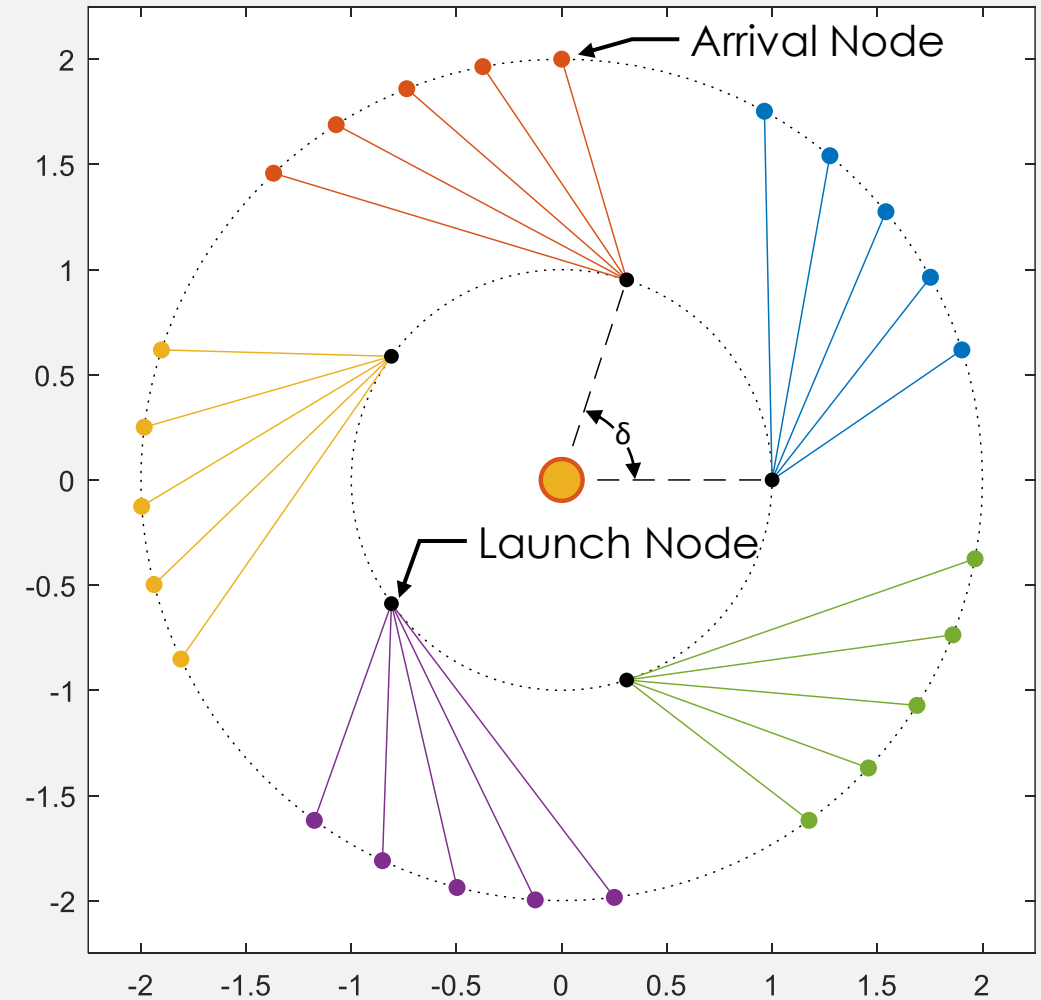
- Used in competitive game AI (GO, Chess, etc.)
- Each node, ○, is one possible state
- Lines are transitions between states
- Uses **heuristics to narrow search options** to optimal candidates



MCTS works best in environments with random behavior and minimal observability

# Application to Broach Trajectory Searches

- Goal
  - Explore only the sequences that show promise while ignoring sequences that break constraints
- Angular Grid
  - Angular grids patterns better suit orbit conics than Cartesian grid patterns
- Node Properties
  - State: (Planetary NAIF ID, Encounter Epoch)
  - $\Delta V$  used to reach point in trajectory
- Constraints
  - Flyby Altitude/Bending Angle
  - $\Delta V$  Budget
  - Maximum C3
- Tree is **built using a loop of four steps**  
Selection → Expansion → Simulation → Backprop



Each node in a sequence is connected by a Lambert arc

# Application to Broach Trajectory Searches

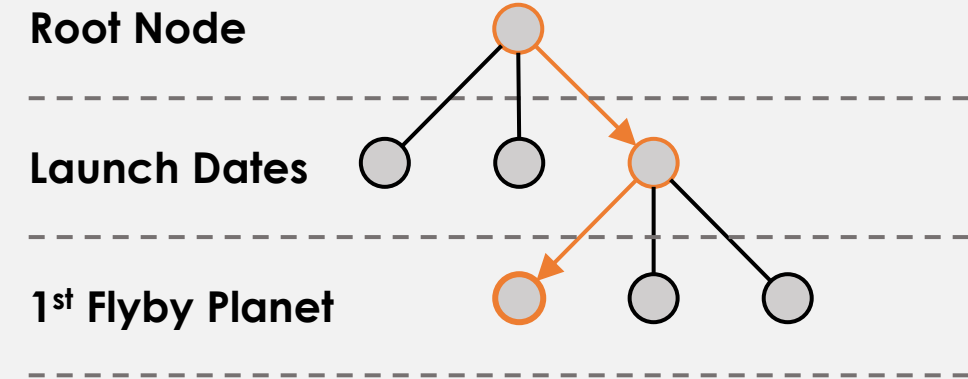
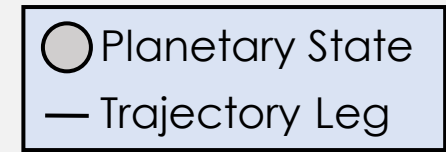
- Paths down the tree to **find leaf a node**
- Estimates value of node through UCB1 function: <sup>[1]</sup>

$$X + C_p \sqrt{\ln n / N}$$

- $X$ : Future reward from child node
- $C_p$ : Exploration-Exploitation Parameter
- $n$ : Number of visits to current node
- $N$ : Number of visits to child node

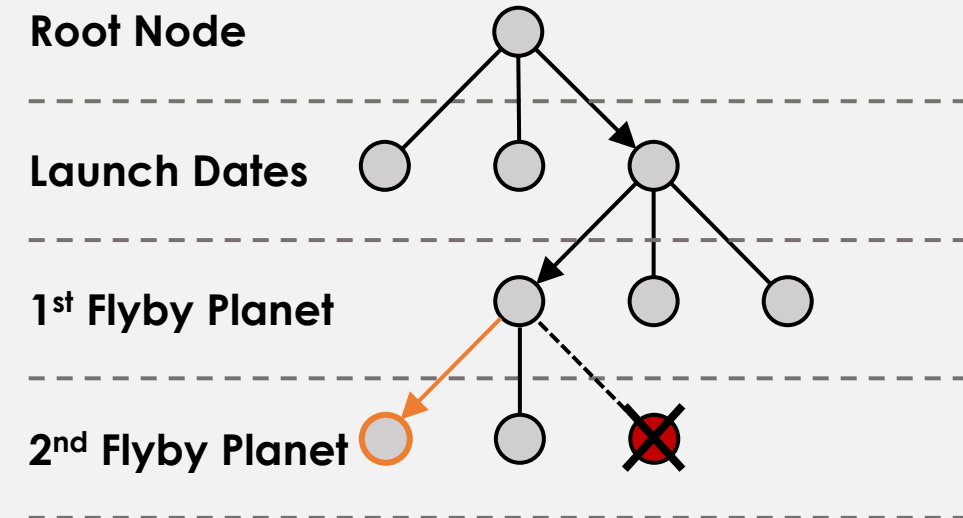
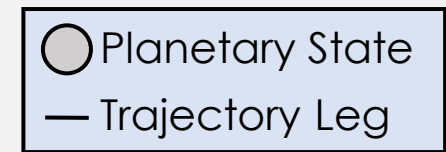
Selection ——— Expansion ——— Simulation ——— Backprop

The  $C_p$  parameter was found to have the best results when set to  $1/\sqrt{2}$  <sup>[1]</sup>



# Application to Broach Trajectory Searches

- Creates a **new layer of nodes** to explore
- Checks each child for feasibility
  - Calculates unoptimized  $\Delta V$  using powered flyby assumption [2]
  - If node exceeds  $\Delta V$  budget, it is made terminal, ~~✗~~
- Selects lowest  $\Delta V$  child by convention



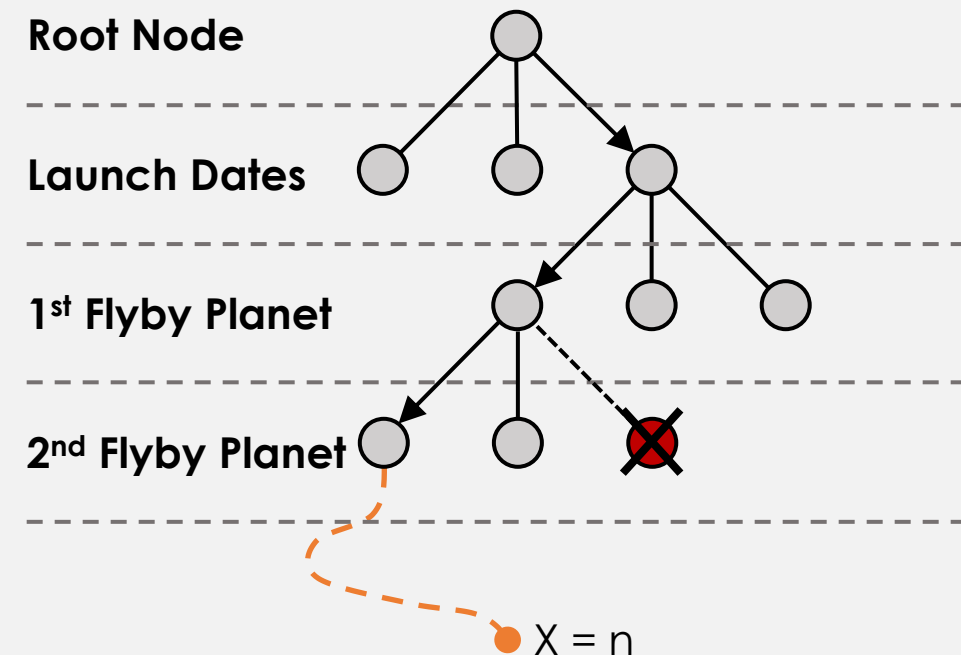
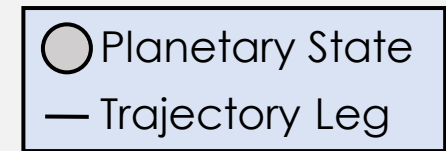
Selection — Expansion — Simulation — Backprop

Powered flyby approximations are used to mitigate the coarse nature of the grid search



# Application to Broach Trajectory Searches

- Run Monte Carlo simulations from leaf node to determine future reward,  $X$
- Takes **random steps** through possible decision and returns reward once terminal condition reached
  - Either the  $\Delta V$  budget is exceeded or destination planet reached

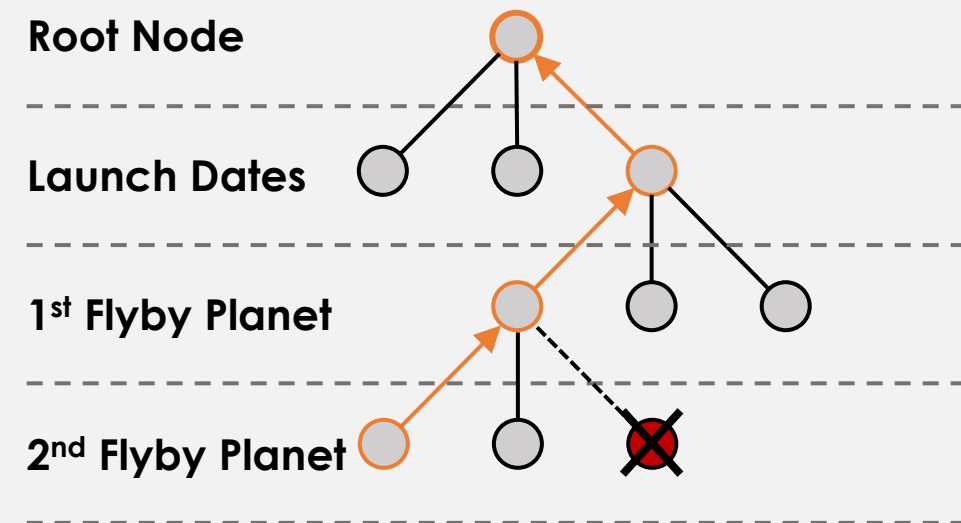
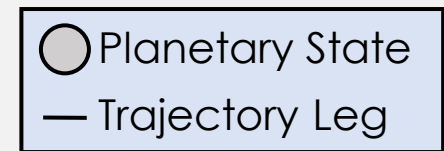


Selection — Expansion — Simulation — Backprop

A small bonus is provided to simulations that terminate early, corresponding to depth

# Application to Broach Trajectory Searches

- Propagates simulated reward back through branch
- New rewards are weighted against previously received rewards
- **Creates a more accurate picture** of the branches as the tree builds

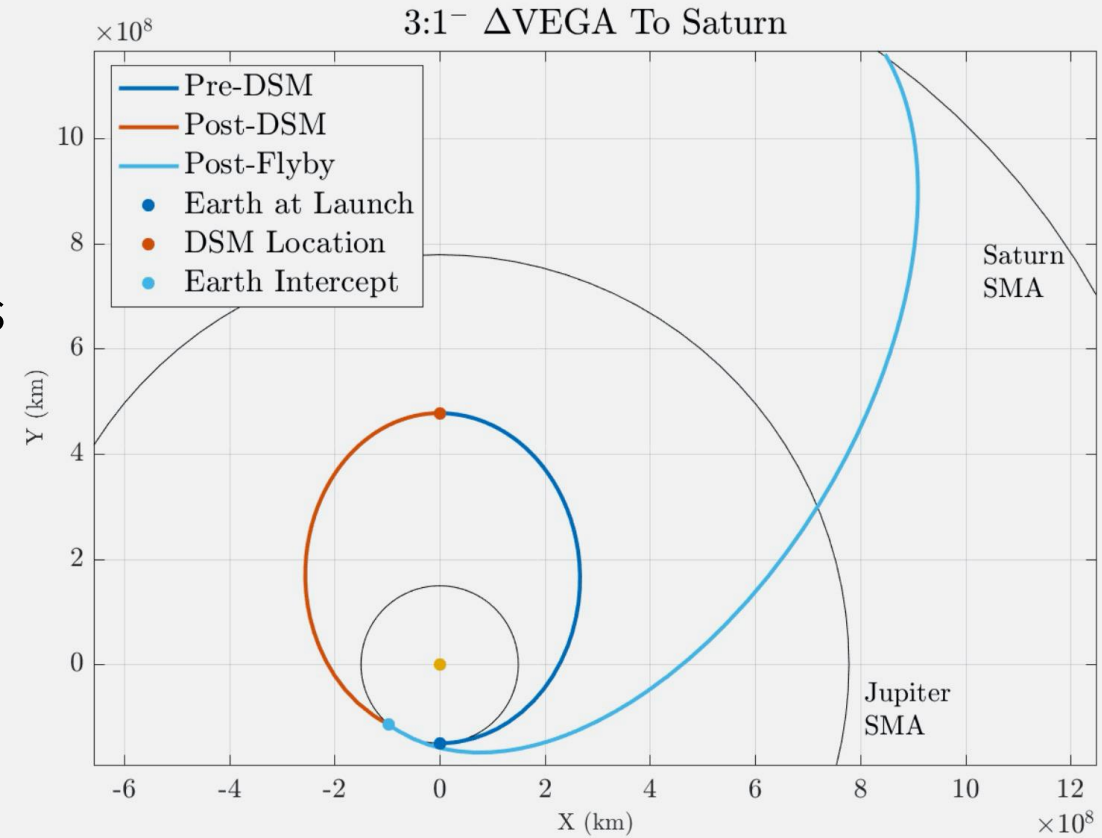


Selection — Expansion — Simulation — Backprop

At the end of backpropagation, the loop continues until the iteration budget is gone

# Implementation of $\Delta$ VEGA

- $k:1$  +/- resonance of Earth (2, 3, and 4)
- Pre-DSM orbit properties determined from  $k$
- Earth intercept point held fixed and a Lambert arc is used to determine DSM  $\Delta V$  and EGA incoming relative velocity.
- Minimizer used to reduce normal (in-plane) component of  $\Delta V$  by adjusting the pre-DSM orbit.<sup>[3]</sup>
- Solutions stored in lookup table and evenly spaced set of MCTS nodes correspond to each  $k$  and intercept true anomaly.



A lookup table method prevents the need of calculating the DSM during the tree search. This estimate serves as an initial condition for the optimization process.

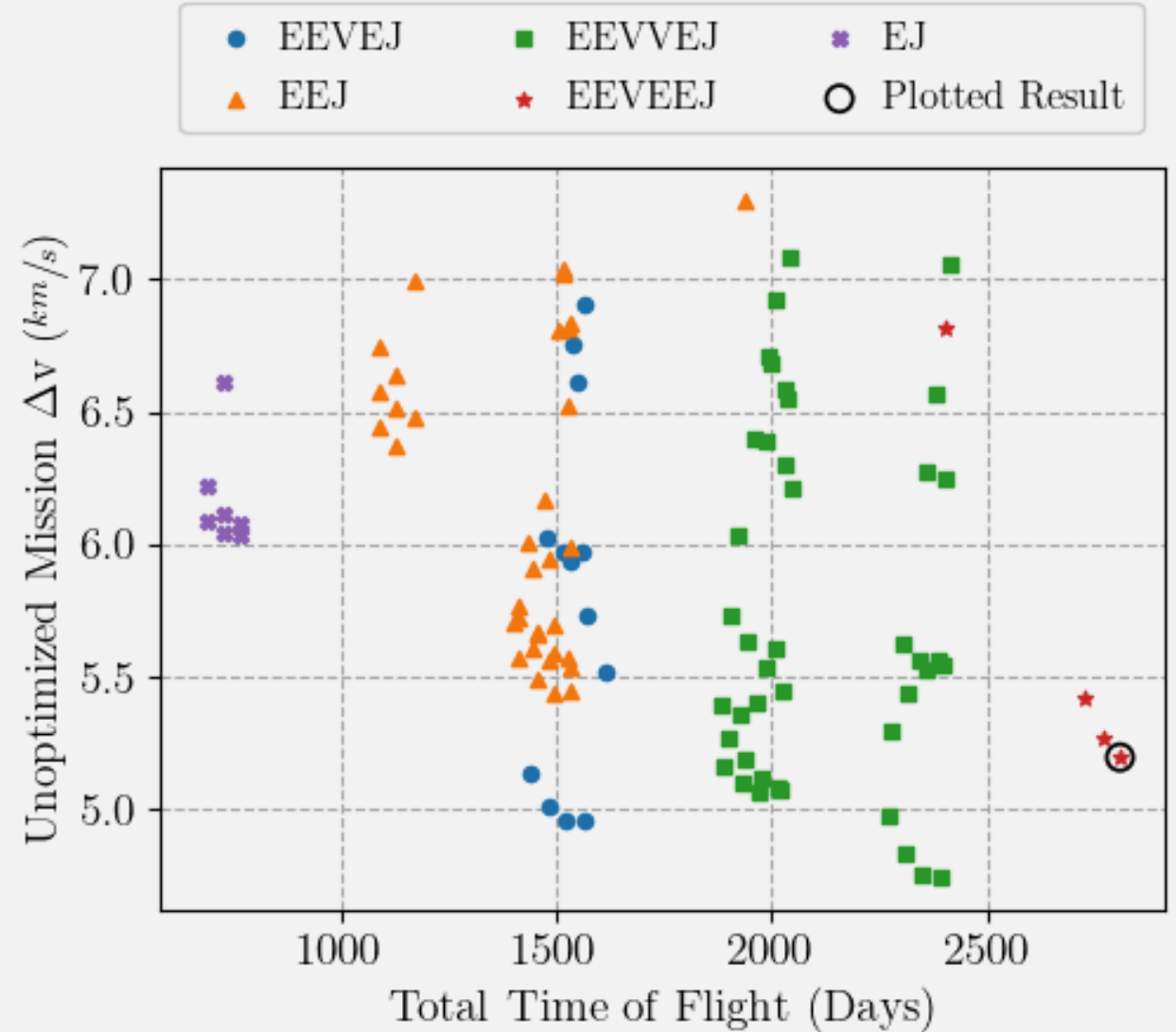
# Europa Clipper (1/2)

- Simulation Goal

- Find EELV Europa Clipper trajectory (EEVEEJ) as described by Buffington<sup>[4]</sup>
- Assess algorithm's ability to find long flyby sequences

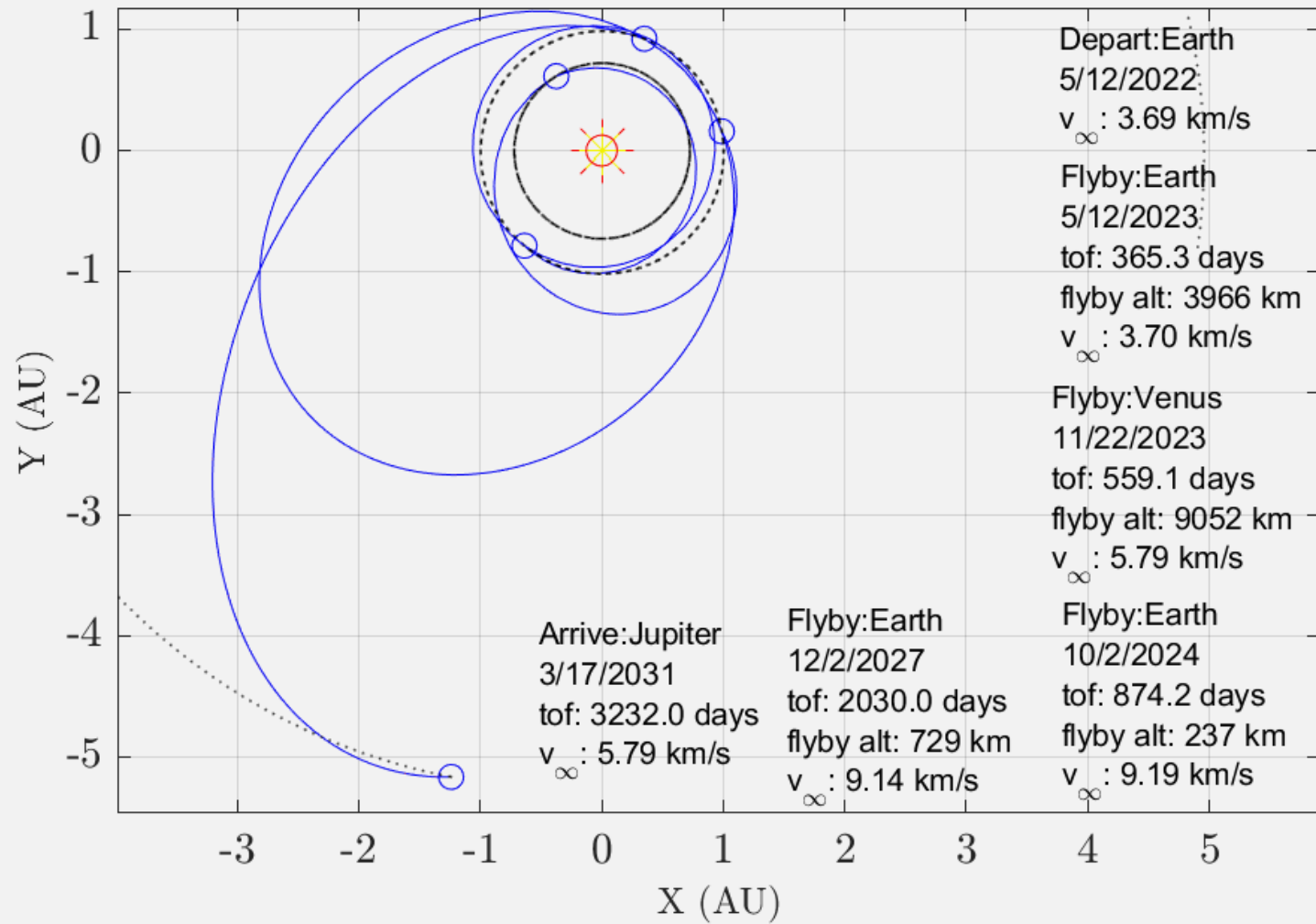
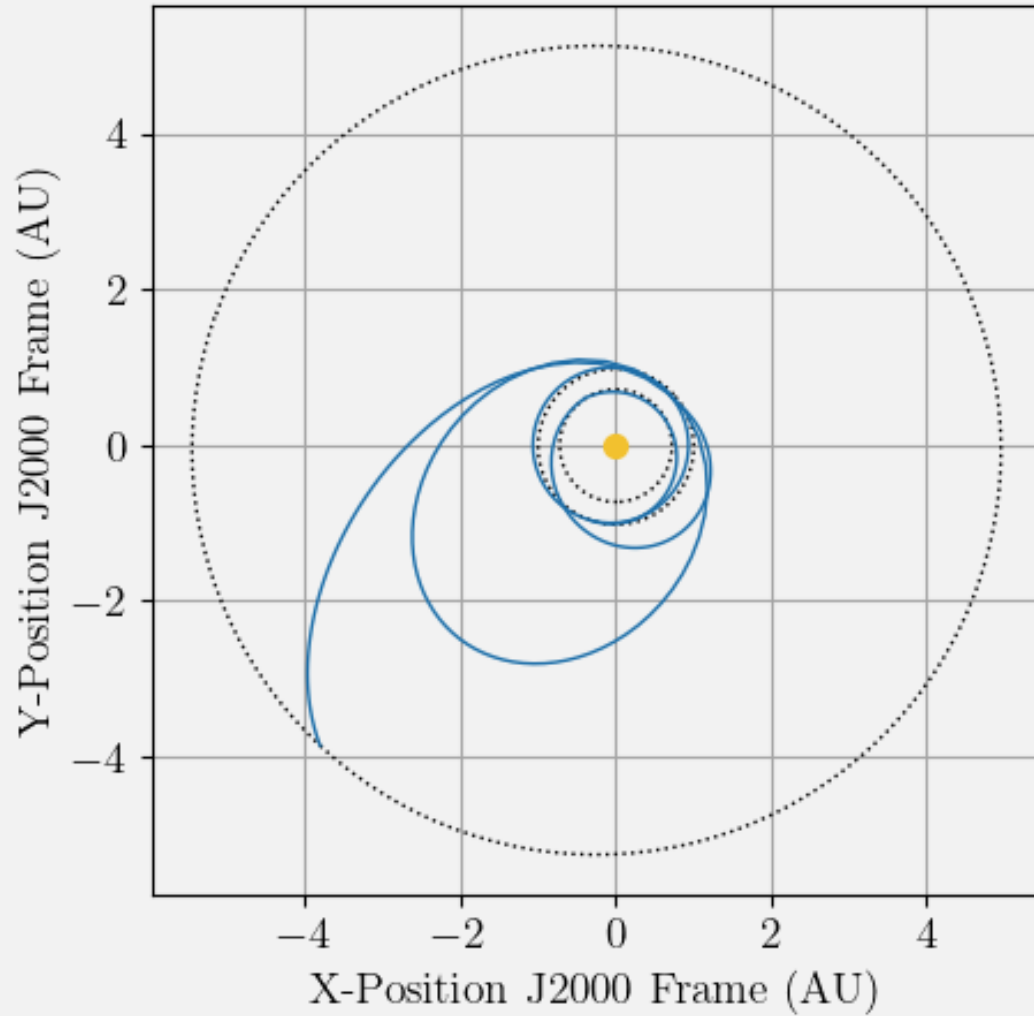
- Tree Search Inputs

Input	Value
Arrival Planet	Jupiter
Launch Window	March 01 — September 01, 2022
Iterations	75,000
$\Delta V$ Budget	10 $km/s$
Max C3	10 $km^2/s^2$
Detail ( $d$ )	24



After the run completed, the tree search found 275 trajectories from 1.9B possibilities

# Europa Clipper (2/2)



The tree search results (left) are within 30 days of their optimized counterparts (right) excluding Jupiter encounter. This confirms the algorithm's ability to find multi-flyby trajectories



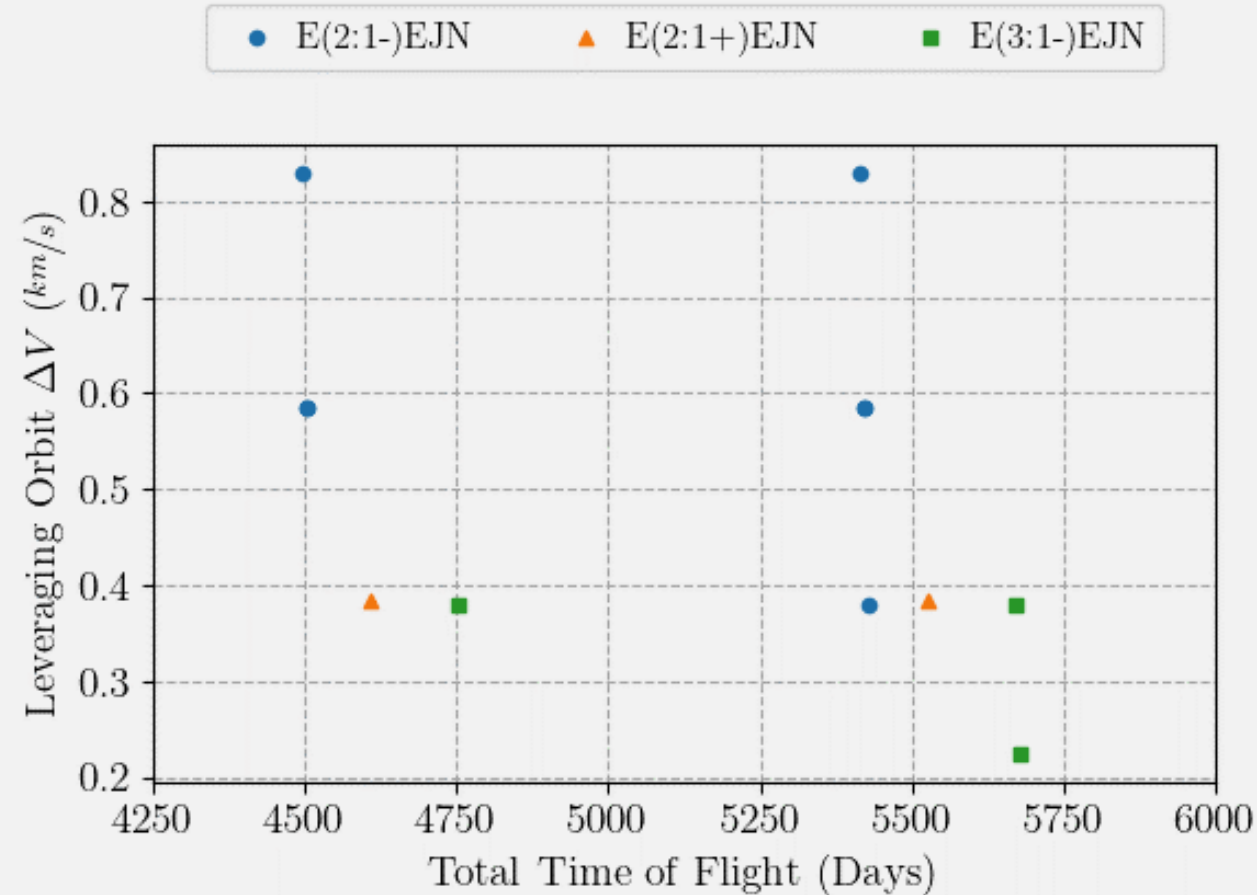
# Trajectories to Neptune (1/2)

- Simulation Goal

- Find trajectories to Neptune via a JGA and Earth orbit leveraging
- Test  $\Delta$ VEGA trajectories and the predicted required DSM  $\Delta V$

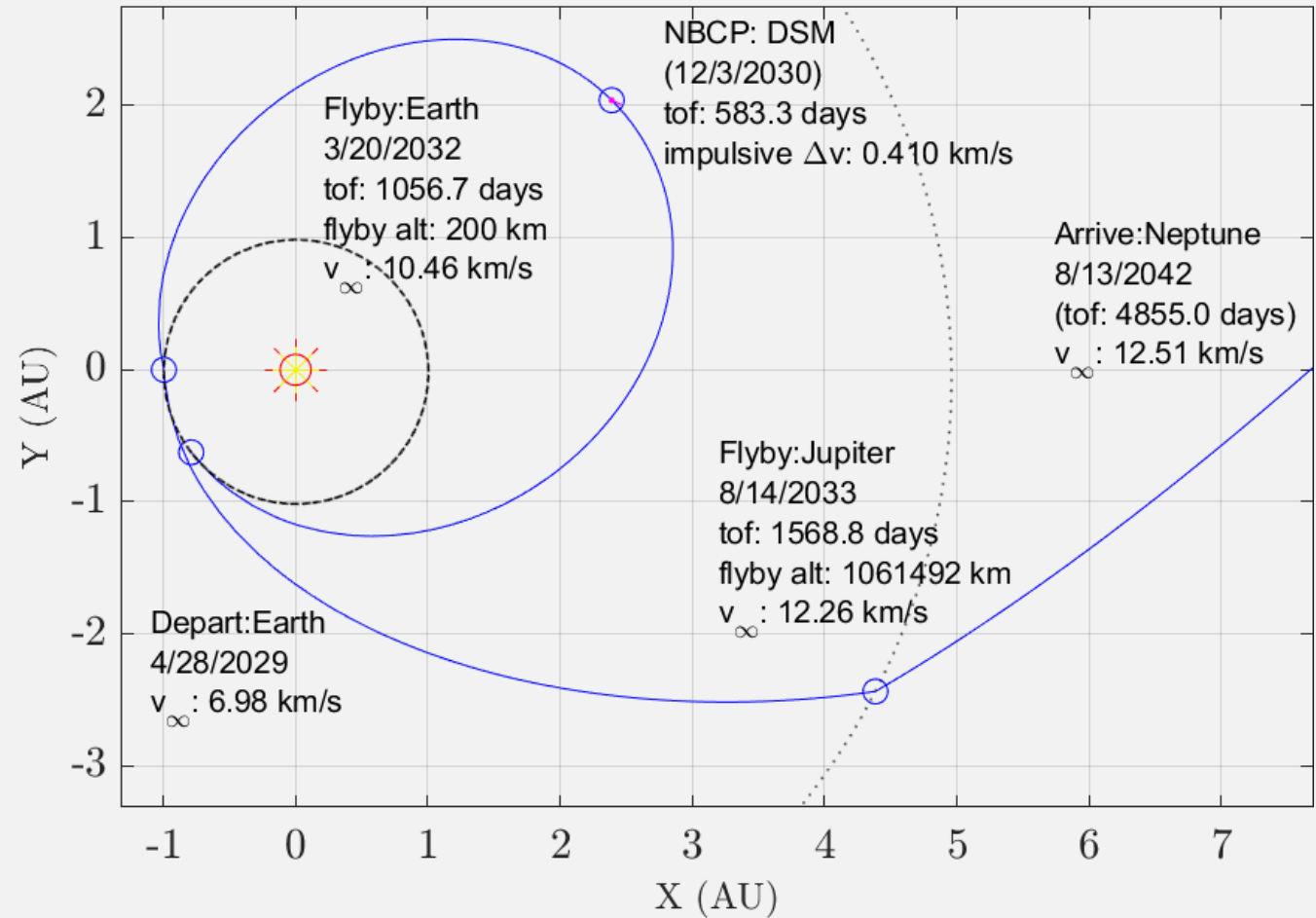
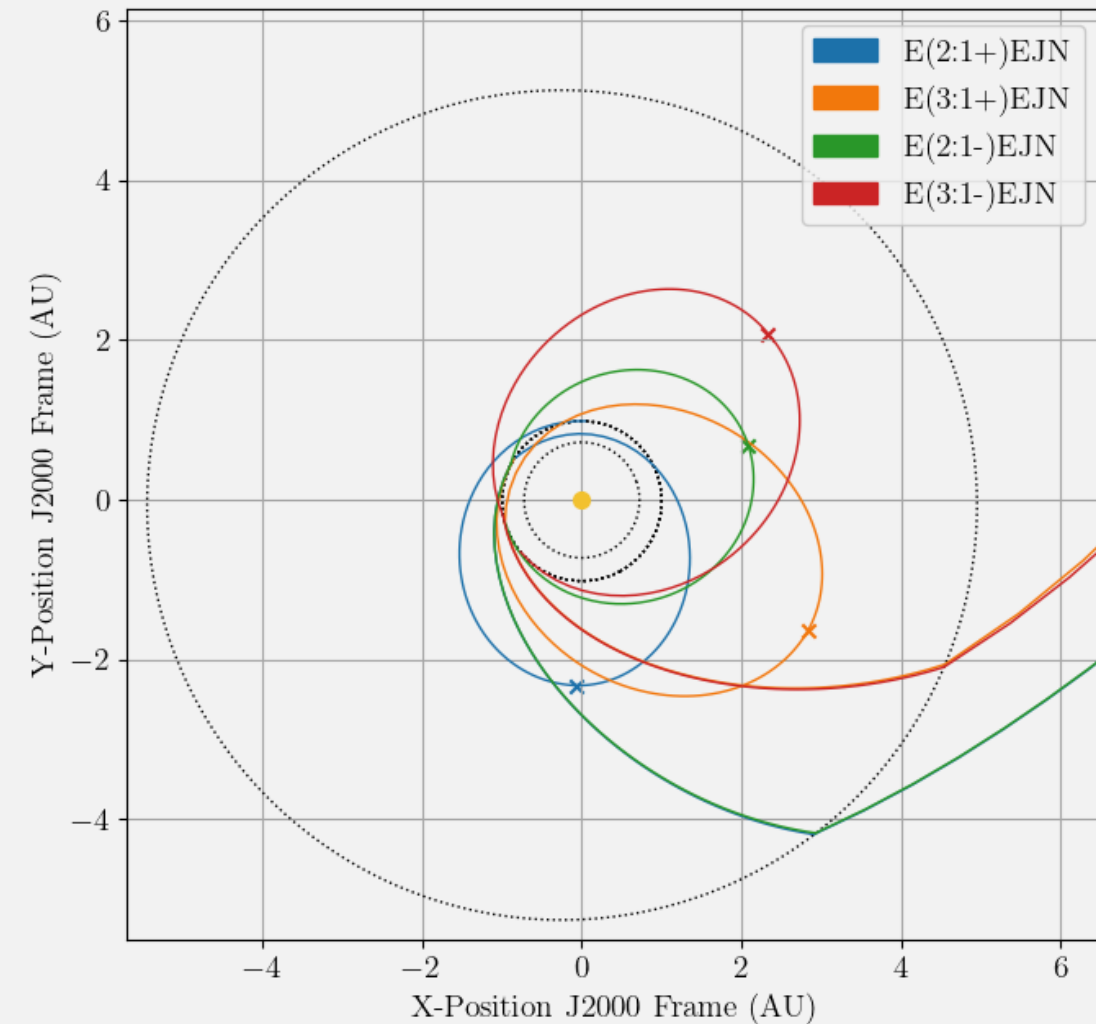
- Tree Search Inputs

Input Name	Input Value
Arrival Planet	Neptune
Launch Window	Jan 01, 2029 — Jan 01, 2030
Iterations	15,000
$\Delta V$ Budget	$3 \text{ km/s}$
Max C3	$60 \text{ km}^2/\text{s}^2$
Detail ( $d$ )	16



Various  $\Delta$ VEGA trajectory options exist for the EEJN search space.

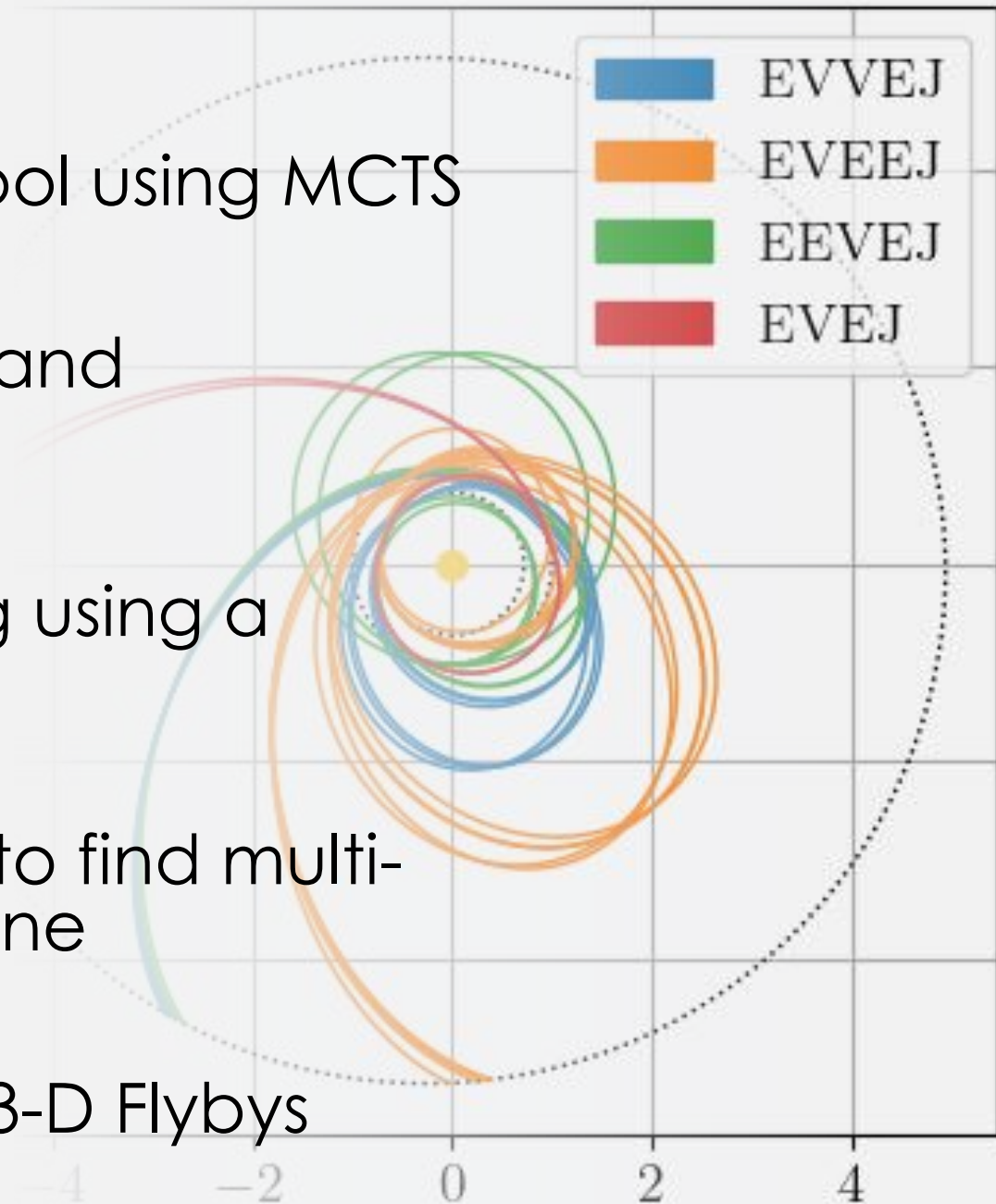
# Trajectories to Neptune (2/2)



The MCTS results with Earth leveraging make for useful initial guesses for the optimizer, and the DSM  $\Delta V$  estimate is  $<0.3$  km/s of the predicted for all tested cases.

# Conclusion

- Created a multi-flyby broad search tool using MCTS
- Good balance between exploration and exploitation of the search space
- Implemented  $\Delta$ VEGA orbit leveraging using a lookup table solution
- Demonstrated algorithm's capability to find multi-flyby sequences to Jupiter and Neptune
- Future Work:  $V_\infty$  leveraging of Venus, 3-D Flybys



For any questions regarding the paper

Please join Virtual Room [Trajectory Design and Optimization IV](#) on [August 12, 2020](#) at [11:20 AM EST](#).

Thank you

# References

- [1] D. Hennes and D. Izzo, "Interplanetary trajectory planning with Monte Carlo tree search," *IJCAI International Joint Conference on Artificial Intelligence*, Vol. 2015-Janua, No. Ijcai, 2015, pp. 769–775.
- [2] S. Wagner and B. Wie, "Hybrid algorithm for multiple gravity-assist and impulsive Delta-V maneuvers," *Journal of Guidance, Control, and Dynamics*, Vol. 38, No. 11, 2015, pp. 2096–2107, 10.2514/1.G000874.
- [3] A. Sims, Jon; Longuski, James; Staugler, "V(infinity) Leveraging for Interplanetary Missions Multiple-Revolutuion Orbit Techniques," *Journal of Guidance, Control, and Dynamics*, Vol. 20, No. 3, 1997, pp. 409–415.
- [4] B. Buffington, "Trajectory design for the europa clipper mission concept," *AIAA/AAS Astrodynamics Specialist Conference 2014*, 2014.