Spin Orbit Coupling Applied to Earth's rotation

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1 Some preliminaries

We apply spin orbit coupling to the problem. Let $\mathbf{R_{cm}}$ be the radial vector from the center of the sun to the center of the earth. Let \mathbf{r}' be the vector from the center of mass of the earth to element of mass under consideration. Let α' be the angle between those 2 vectors. Let v denote the orientation of the earth about a certain axis relative to the earth/sun plane. The potential energy can be written as:

$$V(R_{cm}, v) = \int dV' \rho(\mathbf{r}') \frac{-GM}{\sqrt{R_{cm}^2 + r'^2 - 2r'R_{cm}\cos(\alpha')}}$$
(1)

This form of the integral makes it easy to taylor expand the integrand, according to the famous multipole expansion.

$$V(R_{cm}, v) = -\int dV' \rho(\mathbf{r}') \frac{GM}{R_{cm}} \sum_{l} \left(\frac{r'}{R_{cm}}\right)^{l} P_{l}(\cos(\alpha'))$$
 (2)

Where P_l denotes orthonormalized legendre polynomials. We note the integral vanishes for all l > 0, for a spherical distribution. Hence, this expansion allows us to put increasing accuracy to corrections from a spherical matter distribution.

2 Extracting information

Let's extract information by examining successive values of 1.

 $\mathbf{l} = \mathbf{1}$: For $\mathbf{l} = 1$, we have an integral over $\rho(\mathbf{r}') \frac{r'}{R} \cos(\alpha')$ which vanishes if we choose the center of mass to be our origin (we can make dipole moment of any charge distribution of single polarity to vanish).

$$1 = 2$$
:

$$P_2(x) = \frac{1}{2}(3x^2 - 1) = \frac{1}{2}(2 - 3\sin(\alpha')^2)$$
(3)

$$int\rho(\mathbf{r}')dV'r'^2\frac{1}{2}(2-3\sin(\alpha')^2) = \frac{1}{2}(I_1+I_2+I_3-3I)$$
 (4)

Where I_i represent moments about orthogonal axis and I represent the moment about the axis going through R_{cm} . We can arbitrarily pick the 3 moments to be along principle axis.

For our purposes, expanding along l = 2 is enough.

3 Application to Earth's rotation

Let us now see why the spin of the earth changes. We can model the earth as an oblate ellipsoid of constant density ρ_0 that rotates initially rotates along the principle axis defining rotational symmetry (Call the moment about that one I_2). This sets our initial conditions. The 2 other principle axis are equal and called I_1 . Let the angle between the plane of rotation and the ellipsoid be approximately $\theta = 23$ degrees. The precession time for the earth's axis is 23000 years (see Kleppner), hence we will ignore precession as its time scale is much larger than the time scale of the evolution we are interested in.

Let us break in terms of the free parameters the system can have. The free parameters are: $R_{cm} = R$ because the distance to the sun varies, ϕ because the angular variable about the sun of the earth's CM varies, γ because the axis of spin of the earth varies, and γ because the spin rotation angle also varies. Note we have reduced the number of free parameters of the rigid body of 6 to 4.

4 Lagrangian

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$$L = T - V = \frac{1}{2} \left(MR^2 \dot{\phi}^2 + M\dot{R}^2 + I_2 \dot{\theta}^2 + I_1 \dot{\gamma}^2 \right) + \frac{GM_sM}{R} + GM_s \frac{1}{R^3} \left(2I_1 + I_2 - 3I \right)$$
 (5)

For an ellipsoid of constant density, $I_1 = M \frac{R_1^2 + R_2^2}{5}$, $I_2 = M \frac{2R_1^2}{5}$ where R_1 denotes the radial distance to the equator, R_2 denotes the radial distance to the north pole.

5 Level 1

For level 1 design, we do not need to use such an elaborate scheme. We will instead consider conservation of angular momentum in the direction perpendicular to the motion, call it z.

Call L_s the spin angular momentum magnitude, equal to $I_2\dot{\theta}$. The orbital angular momentum, on the other hand, can be $MR^2\dot{\phi}$. We apply conservation of energy and momentum, ignoring the sun's contribution (E, L are constants of motion):

$$frac12(M\dot{R}^2 + MR^2\dot{\phi}^2) + \frac{(I_2\dot{\theta})^2}{2MR^2} - GM_sM\frac{1}{R} = E$$
 (6)

$$I_2 \dot{\theta} \cos(\gamma) + MR^2 \dot{\phi} = L \tag{7}$$

Level 1.1

Assume that the repetition of ω_s will repeat annually at the epogee of the elliptical orbit. Then \dot{R} will be 0 at those points.

If we plug the second equation into the first for $\dot{\phi}$ we have:

$$\left(\dot{R}^2 + R^2 \left(\frac{L - I_2 \dot{\theta} \cos \gamma}{MR^2}\right)^2 + \frac{(I_2 \dot{\theta})^2}{M^2 R^2} - \frac{2GM_S}{R}\right) = \frac{2E}{M}$$
 (8)

Because E and L does not tell anything about the physics of spin orbit coupling, we do not expect to be able to extract ω_s maximum and minimum from just assuming ideal elliptical orbital parameters and solving for ω at R_{max} and R_{min} .

However, we can approach the problem as follows.

$$L_1 + S_1|_{apoapsis} = L_2 + S_2|_{periapsis} \tag{9}$$

$$fracL_1^2 2mr_1^2 + \frac{S_1^2}{2I} - \frac{\alpha}{r_1} = \frac{L_2^2}{2mr_2^2} + \frac{S_2^2}{2I} - \frac{\alpha}{r_2}$$
 (10)

We have 6 variables: $L_1, L_2, S_1, S_2, r_1, r_2$, where S is the spin angular momentum, L is the orbital angular momentum. Note α is a constant corresponding to the gravitational energy. We can make the following assumptions:

- The orbit-coupling is annually periodic
- The times of maximum and minimum spin occur at the apoapsis and periapsis, where $\dot{R}=0$
- The spin is larger when it is at the apoapsis (experimental data), i. e. at 1
- $S_1 = S_0 + \Delta S$, $S_2 = S_0 \Delta S$ and trivially: $L_1 = L_0 + \Delta S$, $L_2 = L_0 \Delta S$, where S_0 and L_0 denote the average spin and orbital angular momenta respectively

Because the original orbital motion satisfies the following integral of motion:

$$\frac{L_0^2}{2mr^2} - \frac{\alpha}{r} = Constant \tag{11}$$

This implies that:

$$\frac{L_0^2}{2mr_1^2} - \frac{\alpha}{r_1} = \frac{L_0^2}{2mr_2^2} - \frac{\alpha}{r_2} \tag{12}$$

because L_0 corresponds to the orbital angular momentum before any spin-coupling was applied. We plug in our ansatz about the spin coupling term, and obtain the following equations:

$$= \frac{L_0}{m} \left(\frac{1}{r_1^2} + \frac{1}{r_2^2} \right) + \frac{\Delta S}{2m} \left(\frac{1}{r_2^2} - \frac{1}{r_1^2} \right) - 2\frac{S_0}{I}$$
 (13)

This method does not lead to a sensible answer!! Our approach does not endogenize ΔI , the seasonal change in moment of inertia.

PHYSICS FAIL !!!!!!

Level 1.2

We have an unknown variable, R. If we approximate that the radial coordinate is not very different even when coupling is applied, we can substitute the normal values of R obtained from solving point particle central force motion into the equations.