

Part-FCL Question Bank

PPL(A)

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(Excerpt)

80 – Aircraft General Knowledge

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1		thickness of the wing is defined as the distance between the lower and the upper of the wing at the (1,00 P.)			
	□ ☑ □	most inner part of the wing. thinnest part of the wing. thickest part of the wing. most outer part of the wing.			
2	How P.)	is referred to a tubular steel construction with a non self-supporting skin? (1,00			
		Semi-monocoque construction.			
		Grid construction Honeycomb structure			
		Monocoque construction			
3		ary fuselage structures of wood or metal planes are usually made up by what ponents? (1,00 P.) Girders, rips and stringers Frames and stringer Rips, frames and covers			
		Covers, stringers and forming parts			
4	Whic	ch wing configuration is shown in the attachment?			
	See figure (AGK-002) (1,00 P.)				
	Sieh	e Anlage 1			
		Strut-braced high wing High wing Mid wing Low wing			



5	A construction made of frames and stringer with a supporting skin is called (1,00 P.)		
		Honeycomb structure. Grid construction. Wood- or mixed construction. Semi-monocoque construction.	

6 Which tail assembly is shown in the attachment?

See figure (AGK-003) (1,00 P.)

Siehe Anlage 2

V-tail
T-tail

☐ Fuselage-mounted tail

☐ Cruciform tail



7	Wha	t are the major components of an aircraft's tail? (1,00 P.)
		Ailerons and elevator
		Horizontal tail and vertical tail Rudder and ailerons Steering wheel and pedals
8	The	sandwich structure consists of two (1,00 P.)
		thin layers and a light core material. thin layers and a heavy core material. thick layers and a heavy core material. thick layers and a light core material.
9	Whic	ch constructional elements give the wing its profile shape? (1,00 P.)
		Rips Planking Spar Tip
10	The	load factor "n" describes the relationship between (1,00 P.)
		thrust and drag. drag and lift. lift and weight.
		weight and thrust.

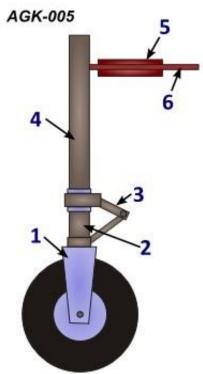
11	Whi	ch are the advantages of sandwich structures? (1,00 P.)
		High temperature durability and low weight Low weight, high stiffness, high stability, and high strength Good formability and high temperature durability High strength and good formability
12	Whi	ch of the stated materials shows the highest strength? (1,00 P.)
		Carbon fiber re-inforced plastic Aluminium Wood Magnesium
13	Wha	at needs to be considered if permissible limitations were exceeded? (1,00 P.)
		The helicopter must be inspected by the pilot in command and if no defect is found, no appropriate entry in the aircraft technical log is necessary The helicopter must be inspected by at least two pilots licensed on the type, one of whom must be the pilot in command The helicopter must be inspected by a qualified engineer before the next flight The helicopter must be subjected to a duplicate inspected by two engineers
14	Rea	sons for dents in the helicopter structure are (1,00 P.)
		material defects or old colour.
		intense erosion or high wear.
	\square	hard landing or excessive stress.
		excessive engine rpm and cylinder defects.
15	Wha	at kind of hydraulic oil is used in aeroplane systems today? (1,00 P.)
		Mineral oil Vegetable oil Bio-oil Synthetic oil

16 Number 1 on the gear designates the...

See figure (AGK-005) (1,00 P.)

Siehe Anlage 3

- □ strut.□ torque link.
- □ inner shock absorber.



17 The fuselage structure may be damaged by... (1,00 P.)

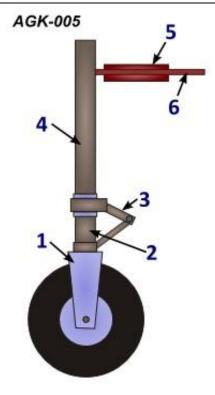
- ☑ exceeding the manoeuvering speed in heavy gusts.
- □ neutralizing stick forces according to actual flight state.
- □ stall after exceeding the maximum angle of attack.
- ☐ airspeed decreasing below a certain value.

18 Number 2 on the gear designates the...

See figure (AGK-005) (1,00 P.)

Siehe Anlage 3

- ☐ fork.
- □ torque link.
- ☐ fixed outer cylinder.



19	How is the nose or tail wheel usually controlled on small aircraft and motor gliders?
	(1,00 P.)

- ☐ By the steering wheel
- ☐ By the control column
- ☐ By weight movement
- ☑ By the pedals

20 Where is the brake system installed to slow the aircraft on ground? (1,00 P.)

- ☐ On the tail wheel
- ☐ On the nose and main gear
- ☐ Only on the nose gear
- ☑ Only on the main gear

21 What kind of control surface is connected with the nose wheel? (1,00 P.)

- □ Elevator
- ☐ Trim rudder
- ☐ Aileron
- ☑ Rudder

22 What is checked by the marking in the attachment?

See figure (AGK-006) (1,00 P.)

Siehe Anlage 4

- ☐ The tread wear of the tire
- ☐ The operating pressure of the carcass
- oxdot The correct postion of the tire relative to the rim



23 Abc	out how many a	axes does an	aircraft move	and how are	these axes	called? (1.00) P.	١
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- 4; optical axis, imaginary axis, sagged axis, axis of evil
- 4; vertical axis, lateral axis, longitudinal axis, axis of speed
- ☐ 3; x-axis, y-axis, z-axis
- ☑ 3; vertical axis, lateral axis, longitudinal axis

24 A movement around the longitudinal axis is primarily initiated by the... (1,00 P.)

- □ rudder.
- ☑ ailerons.
- □ elevator.
- □ trim tab.

25		v are the flight controls on a small single-engine piston aircraft normally controlled actuated? (1,00 P.)
		Power-assisted through hydraulic pumps or electric motors Hydraulically through hydraulic pumps and actuators Manually through rods and control cables Electrically through fly-by-wire
26	Wha	at are the primary and the secondary effects of a rudder input to the left? (1,00 P.)
		Primary: yaw to the left
	\square	Secondary: roll to the right Primary: yaw to the left
		Secondary: roll to the left Primary: yaw to the right
		Secondary: roll to the left Primary: yaw to the right Secondary: roll to the right
27		at is the effect of pulling the control yoke or stick backwards? (1,00 P.)
		The aircraft's tail will produce an increased downward force, causing the aircraft's nose to drop The aircraft's tail will produce an decreased upward force, causing the aircraft's nose to drop The aircraft's tail will produce an increased downward force, causing the aircraft's nose to rise The aircraft's tail will produce an increased upward force, causing the aircraft's nose to rise
28		at happens to a helicopter during cruise when the stick is moved forward without er corrections? (1,00 P.)
		The speed increases and the sink rate increases
		The speed decreases and the sink rate increases
		The speed increases and the sink rate decreases The speed decreases and the sink rate decreases
29	Whi	ch of the following options states all primary flight controls of an aircraft? (1,00 P.)
		All movable parts on the aircraft which aid in controlling the aircraft Flaps, slats, speedbrakes Elevator, rudder, aileron
		Elevator, rudder, aileron, trim tabs, high-lift wing devices, power controls

30	t is the purpose of the secondary flight controls? (1,00 P.)				
		To constitute a backup system for the primary flight controls To improve the performance characteristics of an aircraft and relieve the pilot of excessive control forces			
		To improve the turn characteristics of an aircraft in the low speed regime during approach and landing			
		To enable the pilot to control the aircraft's movements about its three axes			
31	Wha	t is the purpose of a ground adjustable trim tab? (1,00 P.)			
		It is set on the ground to correct an out-of-trim condition arising from the mass and balance situation of an aircraft			
		It is preset on the ground and further adjusted in flight to reduce the need to constantly re-trim the aircraft			
		It is used to optimize the handling characteristics of an aircraft during ground operations It is a non-movable metal tab on a flight control which is adjusted on the ground to optimize the inflight characteristics of an aircraft			
32	The trim wheel or lever in the cockpit is moved aft by the pilot.				
	What effect does this action have on the trim tab and on the elevator? (1,00 P.)				
		The trim tab moves up, the elevator moves up The trim tab moves up, the elevator moves down			
		The trim tab moves down, the elevator moves up The trim tab moves down, the elevator moves down			
	_				
33	Whe P.)	n trimming an aircraft nose up, in which direction does the trim tab move? (1,00			
		Depends on CG position It moves up			
		In direction of rudder deflection It moves down			
34		does a balance tab move in relation to the flight control surface that it is coupled ? (1,00 P.)			
	☑	In the opposite direction At an angle of 90°			
		At an angle of 45° In the same direction			

35	The trim is used to (1,00 P.)		
		increase adverse yaw.	
		move the centre of gravity. adapt the control force. lock control elements.	
36	Whi	ch of the following are an aircraft's secondary flight controls? (1,00 P.)	
		Elevator, rudder, aileron All movable parts on the aircraft which aid in controlling the aircraft Elevator, rudder, aileron, trim tabs, high-lift wing devices, power controls Wing flaps, leading edge devices, spoilers or speedbrakes, trim systems	
37	Wha	at has to be considered during refueling? (1,00 P.)	
		Check the fuel content of the tank with a torch and remove fire protection No open fires, obey smoking ban and apply ground cables Refuel through a soaked rag and keep a fire extinguisher available Apply ground wires, turn on the main switch and magneto ignition	
38	The	primer is (1,00 P.)	
	$\overline{\checkmark}$	an auxiliary pump in the fuel system to facilitate engine starting.	
		a mechanical switch in the cockpit to engage the turbocharger. a valve in the fuel control system for automatic mixture regulation. a nozzle in the Venturi tube of a carburettor for atomising the fuel.	
39	Wha	at is the purpose of the tank ventilation? (1,00 P.)	
		To prevent underpressure caused by fuel consumption To prevent water disposal during parking To distribute the fuel from one tank segment to the other during flight To prevent fuel spillage during refueling on the filler plug	
40	1. Ba 2. Ga 3. Ra	ircuit breaker (1,00 P.)	
		1 and 4 3 and 4 2 and 3 1 and 2	

41	Wha	at is the unit for voltage? (1,00 P.)
		Ampere Watt Ohm
	☑	Volt
42	Wha	at is the unit for electrical power? (1,00 P.)
		Ampere Ohm Watt Volt
43	Wha	at is the reason for static dischargers on aircraft? (1,00 P.)
		To ensure grounding during refueling To discharge static charging during flight To eliminate electrical interferences during intensive radio traffic To improve the quality of radio transmission in high altitudes
44	Wha	at must be considered if the alternator fails in a helicopter? (1,00 P.)
		All instruments and warning systems will fail No change as long as the battery provides enough power The engine runs roughly and is prone to knocking Only high power consumers will failing
45		en using direct current, electrically driven flight instruments are marked with 0 P.)
		"EL". "DC". "AL". "CO".
46	Wha	at devices can be affected by a failure of the electrical system in a helicopter? (1,00
		Fuel quantity indication, radio equipment and altimeter Radio equipment, navigation equipment and magnetic compass Airspeed indicator, altimeter and artificial horizon Radio equipment, navigation equipment and gyros

47	Which cylinder arragement is commonly used on small aircrafts and motor gliders?
	(1,00 P.)

- ☑ Horizontally opposed engine
- ☐ Radial engine
- ☐ In-line engine
- □ V-type engine

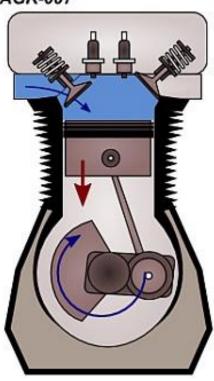
48 Which part of the cycle of a four-stroke Otto engine can be seen in the attachment?

See figure (AGK-007) (1,00 P.)

Siehe Anlage 5

- ☐ Second stroke compression
- ☐ Fourth stroke exhaust
- ☑ First stroke induction
- ☐ Third stroke power

AGK-007



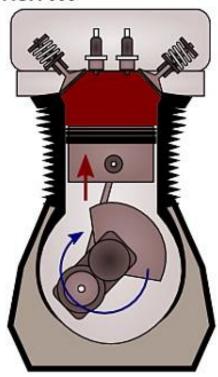
49 Which part of the cycle of a four-stroke Otto engine can be seen in the attachment?

See figure (AGK-008) (1,00 P.)

Siehe Anlage 6

- ☑ Second stroke compression
- ☐ Third stroke power
- ☐ Fourth stroke exhaust
- ☐ First stroke induction

AGK-008



50	What is likely to be the cause if the engine runs unusually rough while checking the
	magnetos? (1,00 P.)

	Starter	ic	four	ltv,
ш	Starter	ıs	iau	IΙV

- ☑ Sparking plug is defective
- ☐ Short circuit on the ground cable
- ☐ Ignition switch is faulty

51 The highest absorbtion of humidity in fuel can be observed in which situation? (1,00 P.)

- ☐ During parking on wet gras areas
- ☐ During parking on cold aprons
- ☑ Almost empty tanks
- ☐ Almost full tanks

52 Where does the condensation water converge in the tank? (1,00 P.)

- □ It floats on the fuel
- ☐ Near the cap of the tank
- ☑ At the lowest position
- ☐ It is mixed with the fuel

53	Wha	at does the octane rating or fuel grade describe? (1,00 P.)
		Anti-knock rating Flame front speed Ignition timing Combustion temperature
54	Whi	ich colour does Avgas 100 LL have? (1,00 P.)
		Red Green Yellow Blue
55		at is the direct influence by switching on the carburettor heating on fixed propeller ines during engine run-up? (1,00 P.)
		RPM decreases Angle of attack decreases
		Angle of attack increases RPM increaes
56	Wha	at is the major task of a carburettor? (1,00 P.)
		To pump fuel from the tanks into the cylinder To control the aircraft's speed through the throttle valve To provide additional fuel to cool the engine To produce an ignitable air/fuel mixture
57		which phase of flight must the carburettor heating be switched off although burettor icing might be expected? (1,00 P.)
		During climb During cruise During taxi During take-off
58	ln w	hich outside air temperature is icing most likely? (1,00 P.)
		Between -5° C and +20° C Between -10° C and +10° C
		Between -20° C and +5° C
		Between -15° C and 0° C

59	What is the task of cooling fins on air-cooled engine cylinders? (1,00 P.)			
		Quick heat transfer to the surrounding air flow by expanding the surface Cooling of the cylinder surrounding airflow and forwarding to hotter engine parts Leading the airflow to parts designated to be cooled Increasing the airflow thus improving cooling of cylinder parts		
60	Cyli	nder head temperature indication relates to (1,00 P.)		
	□ □	all Cylinders. a random cylinder. the critical cylinder. the average of all cylinders.		
61	Wha	at happens during oil filter clocking? (1,00 P.)		
		A bypass valve opens thus enabling the circulation to continue, debris will be filtered by an		
		alternate filter The oil circulation will end after 15 minutes so that a proper engine run will not be guaranteed A bypass valve opens thus enabling the circulation to continue, debris will not be filtered The oil circulation will end after 30 minutes so that a proper engine run will not be guaranteed		
62	How	v should an Otto engine be shut down? (1,00 P.)		
		By completely leaning the mixture By moving the propeller to the feathered position By closing the throttle lever completely By shutting down the generator		
63	Pist	on engines of helicopters have (1,00 P.)		
		one magneto ignition system. one electrical ignition system. two coupled ignition systems. two independent ignition systems.		
64	How	v do you call fuel mixtures with a high amount of fuel? (1,00 P.)		
		Lean Full Empty Rich		

The angle indicated by arrow number 1 shows the propeller's... 65

See figure (AGK-011)

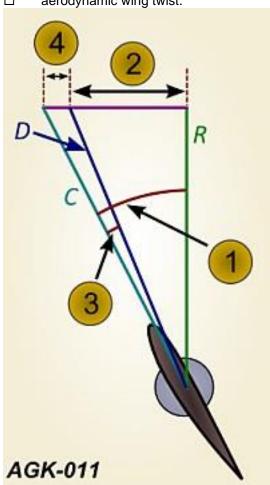
D: Direction of air flow

C: Chord line

R: Direction of rotation (1,00 P.)

Siehe Anlage 7

- $\overline{\mathbf{V}}$ angle of incidence.
- angle of attack.
- geometric wing twist.
- aerodynamic wing twist.



66 How should a power increase be executed on a constant-speed propeller, provided that no other procedure is described in the flight manual? (1,00 P.)

- 1) Decrease manifold pressure
 - 2) Increase RPM
- 1) Decrease RPM
 - 2) Increase manifold pressure
- 1) Increase manifold pressure
 - 2) Increase RPM
- $\overline{\mathbf{V}}$ 1) Increase RPM
 - 2) Increase manifold pressure

67	that	How should a power decrease be executed on a constant-speed propeller, provided that no other procedure is described in the flight manual? (1,00 P.)				
		1) Decrease RPM				
		Decrease manifold pressure Decrease manifold pressure				
		2) Increase RPM 1) Decrease RPM				
		2) Increase manifold pressure				
	Ø	Decrease manifold pressure Decrease RPM				
68	Witl	h increasing altitude and unchanged mixture setting, the air/fuel mixture (1,00 P.)				
		becomes more lean.				
		stays constant. becomes liquid.				
	<u>a</u>	becomes richer.				
69	Witl	h increasing altitude, the power of a carburettor engine (1,00 P.)				
		remains constant.				
		first decreases, from 5000 ft on increases. increases. decreases.				
70	D					
70		ing ignition check the ignition is switched to OFF for a short moment and sequently back to BOTH.				
	Wha	at happens to the RPM if the ignition system is properly grounded? (1,00 P.)				
		The RPM increases once switched to OFF and resumes a value below the previous one when				
	$\overline{\checkmark}$	switched back to BOTH The RPM decreases once switched to OFF and resumes the previous value when switched				
		back to BOTH The RPM increases once switched to OFF and resumes the previous value when switched back				
		to BOTH The RPM decreases once switched to OFF and resumes a value below the previous one when				
	Ш	switched back to BOTH				
71	Whi	ich gauges involve a measurement of temperature? (1,00 P.)				
		Engine lubricant, exhaust gas, outside air, cabin air, directional gyro				
		Engine lubricant, cylinder head, exhaust gas, suction, outside air, cabin air Engine lubricant, engine cooling fluid, cylinder head, exhaust gas, outside air, cabin air				
		Engine oil, cylinder head, exhaust gas, altimeter, airspeed indicator, vertical speed indicator				

72 Fuel quantity in small aircraft is most commonly measured through a level sensor.

	The	European Certification Specifications CS 23 require that (1,00 P.)
		at least one fuel quantity indicator must be available to indicate the total amount of fuel abord an aircraft. This Indicator must be calibrated adequately to indicate the correct amount of fuel during all phases of flight. if the fuel indicator fails, the pilot must land at the next available airport. if no fuel quantity indicator is available to the pilot in flight, the pilot must check the fuel quantity before commencing the flight and recalculate the remaining fuel on board in regular intervals during the flight. a fuel quantity indicator must be available for each tank and the indicator must be calibrated to read "zero" during level flight when the quantity of fuel remaining in the tank is equal to the unusable fuel supply.
73		ich of the instruments listed below obtain their readings through pressure asurement? (1,00 P.)
		Oil pressure gauge, fuel pressure gauge, manifold pressure gauge, altimeter, vertical speed indicator, airspeed indicator, suction gauge Oil pressure gauge, fuel pressure gauge, fuel quantity gauge, manifold pressure gauge, differential pressure gauge, altimeter Airspeed indicator, vertical speed indicator, altimeter, directional gyro, turn and bank coordinator, oil pressure gauge, fuel pressure gauge Airspeed indicator, vertical speed indicator, altimeter, magnetic compass, oil pressure gauge, fuel pressure gauge
74	The	Pitot / static system is required to (1,00 P.)
		prevent potential static buildup on the aircraft. correct the reading of the airspeed indicator to zero when the aircraft is static on the ground. measure total and static air pressure. prevent icing of the Pitot tube.
75	Whi	ch pressure is sensed by the Pitot tube? (1,00 P.)
		Cabin air pressure Total air pressure Dynamic air pressure Static air pressure
76	QFE	E is the (1,00 P.)
		barometric pressure at a reference datum, typically the runway threshold of an airfield. barometric pressure adjusted to sea level, using the international standard atmosphere (ISA). magnetic bearing to a station. altitude above the reference pressure level 1013.25 hPa.

77	QNE	E is the (1,00 P.)
		barometric pressure at a reference datum, typically the runway threshold of an airfield. magnetic bearing to a station. barometric pressure adjusted to sea level, using the international standard atmosphere (ISA). altitude above the reference pressure level 1013.25 hPa.
78	Whi	ch is the purpose of the altimeter subscale? (1,00 P.)
		To set the reference level for the altitude decoder of the transponder To adjust the altimeter reading for non-standard temperature To reference the altimeter reading to a predetermined level such as mean sea level, aerodrome level or pressure level 1013.25 hPa
		To correct the altimeter reading for system errors
79		which way may an altimeter subscale which is set to an incorrect QNH lead to an orrect altimeter reading? (1,00 P.)
		If the subscale is set to a lower than actual pressure, the indication is too high. This may lead to much closer proximity to the ground than intended
	Ø	If the subscale is set to a higher than actual pressure, the indication is too high. This may lead to much closer proximity to the ground than intended
		If the subscale is set to a lower than actual pressure, the indication is too low. This may lead to much closer proximity to the ground than intended If the subscale is set to a higher than actual pressure, the indication is too low. This may lead to much greater heights above the ground than intended
80	Low	ver-than-standard temperature may lead to (1,00 P.)
		a blockage of the Pitot tube by ice, freezing the altimeter indication to its present value. an altitude indication which is too low. an altitude indication which is too high. a correct altitude indication as long as the altimeter subscale is set to correct for non-standard temperature.
81	A fli	ght level is a (1,00 P.)
		pressure altitude. density altitude. altitude above ground. true altitude.
82	A tr	ue altitude is (1,00 P.)
		a height above ground level corrected for non-standard pressure. a height above ground level corrected for non-standard temperature. a pressure altitude corrected for non-standard temperature. an altitude above mean sea level corrected for non-standard temperature.

83	During a flight in colder-than-ISA air the indicated altitude is (1,00 P.)			
		higher than the true altitude. lower than the true altitude. eqal to the true altitude. equal to the standard altitude.		
84	corre	ng a flight in an air mass with a temperature equal to ISA and the QNH set ectly, ndicated altitude is (1,00 P.)		
		lower than the true altitude. equal to the true altitude. equal to the standard atmosphere. higher than the true altitude.		
85	Whic	ch instrument can be affected by the hysteresis error? (1,00 P.)		
		Vertical speed indicator Altimeter Direct reading compass Tachometer		
86	The	measurement of altitude is based on the change of the (1,00 P.)		
		dynamic pressure. differential pressure. static pressure. total pressure.		
87		ch of the following options states the working principle of a vertical speed cator? (1,00 P.)		
	$\overline{\mathbf{V}}$	Measuring the present static air pressure and comparing it to the static air pressure inside a reservoir		
		Total air pressure is measured and compared to static pressure Measuring the vertical acceleration through the displacement of a gimbal-mounted mass Static air pressure is measured and compared against a vacuum		
88	The	vertical speed indicator measures the difference of pressure between (1,00 P.)		
		the present total pressure and the total pressure of a previous moment.		
		the present dynamic pressure and the dynamic pressure of a previous moment. the present dynamic pressure and the static pressure of a previous moment. the present static pressure and the static pressure of a previous moment.		

89	Calib	orated airspeed (CAS) equals (1,00 P.)
		equivalent airspeed (EAS) corrected for altitude. indicated airspeed (IAS) corrected for instrument and position error. ground speed (GS) corrected for instrument and position error. true airspeed (TAS) corrected for wind.
	_	
90	At hi	igher altitudes, true airspeed (TAS) tends to be higher than calibrated airspeed S).
	A ro	ugh estimate of the TAS can be obtained by (1,00 P.)
		subtracting 2 % of the CAS for every 1000 ft altitude. adding 10 % of the CAS for every 1000 ft altitude. subtracting 10 % of the CAS for every 1000 m altitude. adding 2 % of the CAS for every 1000 ft altitude.
91		ircraft cruises on a heading of 180° with a true airspeed of 100 kt. The wind comes 180° with 30 kt.
		ecting instrument and position errors, which will be the approximate reading of airspeed indicator? (1,00 P.)
	☑ □ □	100 kt 130 kt 30 kt
	П	70 kt
92	Whic	ch of the following factors could cause an erroneous airspeed indication? (1,00 P.)
		A circuit breaker was pulled The earth wire is still attached
		The boost pressure line is defective The aircraft is covered with adhesive foils
93	Whic	ch of the following states the working principle of an airspeed indicator? (1,00 P.)
33	∀	Total air pressure is measured and compared against static air pressure.
		Total air pressure is measured by the static ports and converted into a speed indication by the airspeed indicator
		Dynamic air pressure is measured by the Pitot tube and converted into a speed indication by the airspeed indicator
	ш	Static air pressure is measured and compared against a vacuum.

94	Wha	t values are usually marked with a red line on instrument displays? (1,00 P.)
		Operational limits Caution areas Recommended areas Operational areas
95	Wha (1,00	t is necessary for the determination of speed (IAS) by the airspeed indicator?
		The difference between the total pressure and the dynamic pressure
		The difference between the standard pressure and the total pressure The difference betweeen the total pressure and the static presssure
		The difference between the dynamic pressure and the static pressure
96		t is the meaning of the white arc on the airspeed indicator? (1,00 P.)
		Speed range in bumpy air Speed range in smooth air Speed range for extended flaps Speed range not to exceed
97	Wha	t is the meaning of the red range on the airspeed indicator? (1,00 P.)
		Speed which must not be exceeded within bumpy air
		Speed which must not be exceeded with flaps extended
		Speed which must not be exceeded regardless of circumstances Speed which must not be exceeded in turns with more than 45° bank
98	The	compass error caused by the aircraft's magnetic field is called (1,00 P.)
		variation. declination. inclination. deviation.
99		indication of a magnetic compass deviates from magnetic north direction due to t errors? (1,00 P.)
		Deviation, turning and acceleration errors Gravity and magnetism Inclination and declination of the earth's magnetic field Variation, turning and acceleration errors

100	Whic	ch of the mentioned cockpit instruments is connected to the pitot tube? (1,00 P.)
		Vertical speed indicator Direct-reading compass Altimeter Airspeed indicator
101	Whic	ch cockpit instruments are connected to the static port? (1,00 P.)
		Altimeter, vertical speed indicator, airspeed indicator Airspeed indicator, direct-reading compass, slip indicator
		Altimeter, slip indicator, navigational computer
		Airspeed indicator, altimeter, direct-reading compass
102		ircraft in the northern hemisphere intends to turn on the shortest way from a ling of 270° to a heading of 360°.
At approximately which indication of the magnetic compass should terminated? (1,00 P.)		oproximately which indication of the magnetic compass should the turn be inated? (1,00 P.)
		330° 360° 270° 030°
103		ircraft in the northern hemisphere intends to turn on the shortest way from a ling of 360° to a heading of 270°.
		pproximately which indication of the magnetic compass should the turn be inated? (1,00 P.)
		300° 360° 240° 270°
104	The	term "static pressure" is defined as pressure (1,00 P.)
		inside the airplane cabin. of undisturbed airflow. resulting from orderly flow of air particles. sensed by the pitot tube.

105 An aircraft in the northern hemisphere intends to turn on the shortest way from a

heading of 030° to a heading of 180°. At approximately which indicated magnetic heading should the turn be terminated? (1,00 P.) 360°. П 150°. $\overline{\mathsf{V}}$ 210°. 180°. 106 What does the dynamic pressure depend directly on? (1,00 P.) Air pressure and air temperature $\overline{\mathbf{V}}$ Air density and airflow speed squared Air density and lift coefficient Lift- and drag coefficient 107 What is a cause for the dip error on the direct-reading compass? (1,00 P.) Deviation in the cockpit Acceleration of the airplane $\overline{\mathbf{V}}$ Inclination of earth's magnetic field lines Temperature variations 108 What behaviour is shown by a rotating gyro in space? (1,00 P.) It moves in accordance with the body surrounding it $\overline{\mathbf{V}}$ It tends to maintain its position in space It swings from east to west like a pendulum It moves in circles with a stadily decreasing radius 109 A gyro which is rotating in space responds to forces being applied to it by an evasive movement... (1,00 P.) at an angle of 45° to the force being applied. in a northern direction. $\overline{\mathbf{V}}$ at an angle of 90° to the force being applied. at an angle of 180° to the force being applied.

110	The	bank angle of a 2-minutes circle depends on the (1,00 P.)
		CAS. IAS. Ground speed. TAS.
111	A tu P.)	rn and bank (or turn and slip) coordinator provides information regarding (1,00
		the coordination of the turn and slip angle. the rate of turn and coordination, i.e. slip or skid, of the turn. the pitch and bank angle of the aircraft. the rate of turn and bank angle of the aircraft.
112	The	airspeed indicator is unservicable.
	The	airplane may only be operated (1,00 P.)
	\square	when the airspeed indicator is fully functional again.
		when a GPS with speed indication is used during flight.
		if no maintenance organisation is around.
		if only airfield patterns are flown.
113		ch light colour advises the pilot to the condition "corrective action may be lired in the future"? (1,00 P.)
		Amber (orange) Red Blue Green
114	The	Caution Area is marked on an airspeed indicator by what color? (1,00 P.)
		Green Red Yellow White
115	An a	attitude director indicator (ADI) combines the information provided by (1,00 P.)
		the attitude indicator and the flight director. the directional gyro and the rate gyro. the attitude indicator and the relative bearing indicator. the attitude indicator and the radio magnetic indicator.

116	What does the abbreviation HSI stand for? (1,00 P.)			
		Horizontal Situation Indicator Hybernating System Indication Horizontal Slip Indicator Horrifying Steep Inclination		
	_	Thermying etecp incimation		
117	A horizontal situation indicator (HSI) combines the information provided by (1,00 P.)			
		the directional gyro and the flight director. the rate gyro and the slip indicator. the directional gyro and the VHF navigation receiver. the attitude indicator and the flight director.		
118	What difference in altitude is shown by an altimeter, if the reference pressure scale setting is changed from 1000 hPa to 1010 hPa? (1,00 P.)			
		80 m less than before		
		Values depending on QNH Zero		
	Ø	80 m more than before		
119	When is it necessary to adjust the pressure in the reference scale of an alitimeter? (1,00 P.)			
		Before every flight and during cross country flights Every day before the first flight		
		Once a month before flight operation		
		After maintance has been finished		
120	The	altimeter's reference scale is set to airfield pressure (QFE).		
	What indication is shown during the flight? (1,00 P.)			
		Pressure altitude Height above airfield		
		Airfield elevation Altitude above MSL		

121	1 A vertical speed indicator connected to a too big equalizing tank results in (
		no indication mechanical overload indication too low		
	V	indication too high		
122	A vertical speed indicator measures the difference between (1,00 P.)			
		instantaneous total pressure and previous total pressure.		
		total pressure and static pressure. instantaneous static pressure and previous static pressure. dynamic pressure and total pressure.		
123	3 The term "inclination" is defined as (1,00 P.)			
		deviation induced by electrical fields.		
		angle between airplane's longitudinal axis and true north. angle between earth's magnetic field lines and horizontal plane. angle between magnetic and true north.		
124	Wha	At does the slip indicator show? (1,00 P.) Apparent vertical Rate of turn Vertical to horizon Airplane's bank		
125	5 The slip indicator's ball (apparent vertical) has moved to the right during a By what actions may the ball led back to middle position? (1,00 P.)			
	 ✓	Reduce bank, increase rate of turn		
		Increase bank, increase rate of turn		
		Reduce speed, increase bank Reduce rate of turn, reduce bank		
126				

What engines are commonly used with Touring Motor Gliders (TMG)? (1,00 P.)			
	4 Cylinder; 4 stroke 4 Cylinder 2 stroke 2 Cylinder Diesel 2 plate Wankel		
What is the purpose of engine oil in piston engine? (1,00 P.)			
	Cooling, lubrication of the engine and disposal of fine abrasion. Cooling of the engine and lubrication of propeller shaft. For nose reduction and bearing of the propeller shaft. Cooling, lubrication and increasing engine performance.		
	Wha		

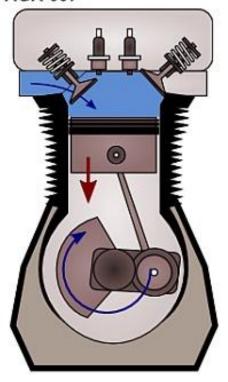




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AGK-007



AGK-008

