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Chapter 6. System Time-response Characteristics

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A system response is determined by

6.2. System Time Response

Consider a continuous-time (open-loop) system

$$G(s) = K \frac{\prod (s - z_i)}{\prod (s - p_i)}$$

or a discrete-time (open-loop) system

$$G^{\mathsf{d}}(z) = K \frac{\prod (z - z_i^{\mathsf{d}})}{\prod (z - p_i^{\mathsf{d}})}.$$

The system response is determined by

- **Poles** p_i or p_i^d : Generate a certain "mode" of the output
- ightharpoonup Zeros z_i or z_i^{d} : Block a certain "mode" of the input
- ▶ DC gain $\lim_{s\to 0} G(s)$ or $\lim_{z\to 1} G^{\mathsf{d}}(z)$
- ▶ Relative degree (상대차수) = # of poles # of zeros

Some features of the sampled-data system

6.2. System Time Response

Consider a sampled-data system with the pulse transfer function

$$G^{\mathsf{d}}(z) = \mathcal{Z}\left[\frac{1 - \mathsf{e}^{-Ts}}{s}G(s)\right].$$

▶ (Discrete-time) poles: Each pole has the exact form

$$p_i^{\mathsf{d}} = \mathsf{e}^{p_i T}$$

▶ (Discrete-time) zeros: Some are related to continuous-time zeros as

$$z_i^{\rm d} \approx {\rm e}^{z_i T} \quad ({\rm for \ sufficiently \ small} \ T)$$

while the others do not have a continuous-time counterpart.

- ightharpoonup DC gain: $\lim_{z\to 1} G^{\mathsf{d}}(z) = \lim_{s\to 0} G(s)$
- ▶ Relative degree: For most cases, $\operatorname{rel.deg}(G^{\operatorname{d}}(z)) = 1$.

How to compute step response of closed-loop system

6.2. System Time Response

Example 6.1: We want to compute the inverse z-transform of

$$C^{\mathsf{d}}(z) = \frac{G^{\mathsf{d}}(z)}{1 + G^{\mathsf{d}}(z)} R^{\mathsf{d}}(z), \quad \text{with } R^{\mathsf{d}}(z) = \frac{1}{1 - z^{-1}} = \frac{z}{z - 1}.$$

In the example above, the pulse transfer function is computed by

$$\begin{split} G^{\mathsf{d}}(z) &= \mathcal{Z}\left[\frac{1 - \mathsf{e}^{-Ts}}{s} \frac{4}{s+2}\right] = \frac{z-1}{z} \mathcal{Z}\left[\frac{4}{s(s+2)}\right] = \frac{z-1}{z} \frac{2(1 - \mathsf{e}^{-2T})z}{(z-1)(z - \mathsf{e}^{-2T})} \\ &= \frac{0.3625}{z - 0.8187}, \quad \text{if } T = 0.1 \text{ sec} \end{split}$$

(Cont'd)

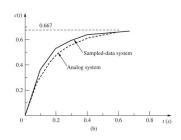
6.2. System Time Response

... We have the closed-loop transfer function

$$T^{\rm d}(z) = \frac{G^{\rm d}(z)}{1+G^{\rm d}(z)} = \frac{0.3625}{z-0.4562}.$$

⇒ the step response in discrete time has the form

$$\begin{split} c^{\mathsf{d}}(k) \\ &= \mathcal{Z}^{-1} \left[\frac{G^{\mathsf{d}}(z)}{1 + G^{\mathsf{d}}(z)} \frac{z}{z - 1} \right] \\ &= \mathcal{Z}^{-1} \left[\frac{0.667z}{z - 1} + \frac{-0.667z}{z - 0.4562} \right] \\ &= 0.667 - 0.667(0.4562)^k \end{split}$$



Q. How can we derive the step response in continuous time?

(Cont'd)

6.2. System Time Response

The continuous-time closed-loop transfer function is given by

$$T_a(s) = \frac{G_p(s)}{1 + G_p(s)} = \frac{4}{s+6}, \quad G_p(s) = \frac{4}{s+2}$$

 \Rightarrow We have the unit-step response in continuous time as:

$$C_a(s) = T_a(s) \frac{1}{s} = \frac{4}{s(s+6)} = \frac{0.667}{s} - \frac{0.667}{s+6},$$

$$\Rightarrow c_a(t) = 0.667(1 - e^{-6t})$$

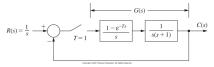
One important lesson: $c^{\rm d}(k)$ above is NOT the same as $c_a(kT)$: i.e.,

$$c^{\mathsf{d}}(k) = 0.667 - 0.667(0.4562)^{k} \neq c_{a}(kT) = 0.667 - 0.667\mathrm{e}^{-0.6k}$$
$$\approx 0.667 - 0.667(0.5488)^{k}$$

Example 6.4

6.2. System Time Response

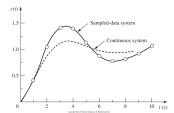
Consider the following sampled-data system with T=1:



The pulse transfer function can be computed as

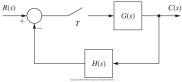
$$G^{\mathsf{d}}(z) = \frac{z-1}{z} \mathcal{Z} \left[\frac{1}{s^2(s+1)} \right]_{T-1} = \dots = \frac{0.368z + 0.264}{z^2 - 1.368z + 0.368}$$

In this case, we have $c^{d}(k)$ (solid) and $c_{a}(t)$ (dashed) as



System response of a closed-loop system

6.3. System Characteristic Equation



Then one can express the z transform of $c^{\rm d}(k)=c(kT)$ as

$$C^{d}(z) = \frac{G^{d}(z)}{1 + \overline{GH}^{d}(z)} R^{d}(z) = \frac{K \prod^{m} (z - z_{i}^{d})}{\prod^{n} (z - p_{i}^{d})} R^{d}(z)$$
$$= \frac{k_{1}z}{z - p_{1}^{d}} + \dots + \frac{k_{n}z}{z - p_{n}^{d}} + C_{R}(z)$$

where p_i^d is the root of $1 + \overline{GH}^d(z) = 0$ (assumed to be simple in this case).

Note:

- $ightharpoonup rac{k_1z}{z-p_a^d}+\cdots+rac{k_nz}{z-p_a^d}$ is called the natural response of $C^{\sf d}(z)$.
- ▶ $1 + \overline{GH}^{d}(z) = 0$ is termed the system characteristic equation.

Mapping from s-plane to z-plane: The case for $e(t) = e^{-at}$

6.4. Mapping the s-plane into the z-plane

Main interest is to reveal the relation btw. continuous- and discrete-time responses.

For the simplest case $e(t) = e^{-at}$, we have

$$E(s) = \frac{1}{s+a} \rightarrow E^*(s) = \frac{e^{Ts}}{e^{Ts} - e^{-aT}} \rightarrow E^{d}(z) = \frac{z}{z - e^{-aT}}$$

which means that,

(the pole
$$s=-a$$
 in s -plane) \leftrightarrow (the pole $z=\mathrm{e}^{-aT}$ in z -plane)

Note:

- ▶ This equivalence is satisfied for both $a \in \mathbb{R}$ and $a \in \mathbb{C}$.
- No such relation can be found in the zeros.

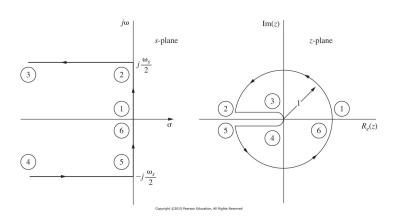
In the following slides, we examine the contour of $\mathrm{e}^{(\sigma+j\omega)T}$ in the complex plane.

Contour 1: Unit circle in z-plane

6.4. Mapping the s-plane into the z-plane

Case 1: For $z = e^{j\omega T}$, we have

$$z = \mathrm{e}^{sT} = \mathrm{e}^{\sigma T} \mathrm{e}^{j\omega T} = \mathrm{e}^{j\omega T} = \cos(\omega T) + j\sin(\omega T) = 1 \angle \omega T.$$

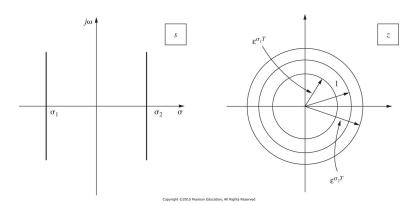


Contour 2: Circles with various radius

6.4. Mapping the s-plane into the z-plane

Case 2: For
$$s_1 = \sigma_1 + j\omega$$
,

$$z = e^{\sigma_1 T} e^{j\omega T} = e^{\sigma_1 T} \angle \omega T$$
 whose radius is $e^{\sigma_1 T}$

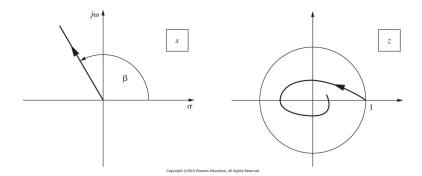


Contour 3: Constant damping ratio

6.4. Mapping the s-plane into the z-plane

Case 3: For $s_1 = \sigma + j\omega$ where $\omega/\sigma = \tan(\beta)$ with fixed β ,

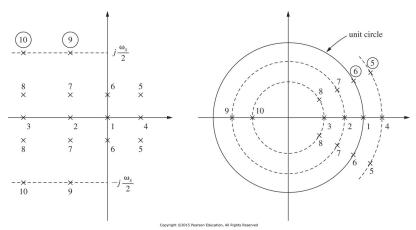
$$z = e^{\sigma T} e^{j\omega T} = e^{\sigma T} \angle \omega T = e^{\sigma T} \angle (\sigma T \tan \beta)$$



Summarizing so far,

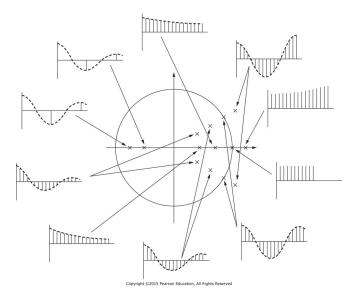
6.4. Mapping the s-plane into the z-plane

Each points in the z-plane are associated with points in the s-plane, with the equivalence $z=\mathrm{e}^{sT}.$



Transient response characteristics w.r.t. pole locations

6.4. Mapping the s-plane into the z-plane



Extension to 2nd-order system

6.4. Mapping the s-plane into the z-plane

Consider a continuous-time 2nd-order system

$$G(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

where the (continuous-time) system response of G(s) is governed by

- $\triangleright \zeta$: damping ratio
- $\triangleright \omega_n$: natural frequency
- ⇒ The continuous-time poles of the system is computed by

$$p_{1,2} = -\zeta \omega_n \pm j\omega_n \sqrt{1 - \zeta^2}$$

Equivalent discrete-time poles in z-plane?

6.4. Mapping the s-plane into the z-plane

Following the same procedure, we have the equivalent discrete-time poles in z-plane

$$p_{1,2}^{\mathsf{d}} = \mathsf{e}^{sT}|_{s=p_{1,2}} = \mathsf{e}^{-\zeta\omega_n T} \angle (\pm \omega_n T \sqrt{1-\zeta^2}) = \angle r(\pm \theta).$$

where the radius and the phase are computed by

$$r=\mathrm{e}^{-\zeta\omega_nT}=\mathrm{e}^{-T/\tau},\quad \text{(with }\tau:=1/(\zeta\omega_n)\text{)}$$

$$\theta=\omega_nT\sqrt{1-\zeta^2}$$

On the other hand, one can also derive ζ and ω_n as

$$\zeta = \frac{-\ln r}{\sqrt{(\ln r)^2 + \theta^2}}, \quad \omega_n = \frac{1}{T}\sqrt{(\ln r)^2 + \theta^2}.$$

Understanding the role of time constant au

6.4. Mapping the s-plane into the z-plane

In summary, a part of the system characteristic equation in s-plane will be changed into the following in the z-plane:

▶ 1st-order equation:

$$s + 1/\tau \rightarrow z - e^{-T/\tau}$$

2nd-order equation:

$$\begin{split} (s+1/\tau)^2 + \omega^2 & \to & z^2 - 2z \mathrm{e}^{-T/\tau} \cos(\omega T) + \mathrm{e}^{-2T/\tau} \\ & = (z-z_1)(z-\overline{z}_1) \text{ (where } z_1 = \mathrm{e}^{-T/\tau} \angle \omega T \text{)} \end{split}$$

Note: In both cases, τ and the radius r are related with each other as

$$\frac{\tau}{T} = -\frac{1}{\ln r}$$

which means that, the ratio btw. τ and T is important.

System type of a discrete-time system

6.5. Steady-state Accuracy

Consider a discrete-time closed-loop system

$$\frac{C^{\mathsf{d}}(z)}{R^{\mathsf{d}}(z)} = \frac{G^{\mathsf{d}}(z)}{1 + G^{\mathsf{d}}(z)}$$

where the open-loop transfer function is expressed by

$$G^{\mathsf{d}}(z) = \frac{K \prod^{m} (z - z_i^{\mathsf{d}})}{(z - 1)^{\mathsf{N}} \prod^{n - \mathsf{N}} (z - p_j^{\mathsf{d}})} \quad \text{with } p_j^{\mathsf{d}} \neq 1.$$

 \Rightarrow *N* is called the type of the system (in discrete time).

Note: For a system with $N \ge 1$, its DC gain $= \infty$.

Q. How about continuous-time case?

Steady-state step response and system type

6.5. Steady-state Accuracy

For the unit-step input whose z-transform is computed by

$$R^{\mathsf{d}}(z) = \frac{z}{z - 1}$$

one has the steady-state error of the closed-loop system as

$$\begin{split} e_{ss}^{\mathsf{d}} &= \lim_{k \to \infty} e^{\mathsf{d}}(k) = \lim_{z \to 1} (z - 1) E^{\mathsf{d}}(z) = \lim_{z \to 1} \frac{(z - 1) R^{\mathsf{d}}(z)}{1 + G^{\mathsf{d}}(z)} = \frac{1}{1 + \lim_{z \to 1} G^{\mathsf{d}}(z)} \\ &= \begin{cases} \mathsf{constant} \neq 0, & \text{if } N = 0 \\ 0, & \text{if } N \geq 1 \end{cases} \end{split}$$

provided that the closed-loop system is stable (as we will discuss later).

Lesson 1: A system with type $N \ge 1$ has zero-steady-state error in step response.

Steady-state ramp response and system type

6.5. Steady-state Accuracy

For the unit-ramp input whose z-transform is computed by

$$R^{\mathsf{d}}(z) = \frac{Tz}{(z-1)^2}$$

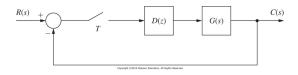
one has the steady-state error of the closed-loop system as

$$\begin{split} e_{ss}^{\mathsf{d}} &= \lim_{k \to \infty} e^{\mathsf{d}}(k) = \lim_{z \to 1} (z-1) E^{\mathsf{d}}(z) = \lim_{z \to 1} \frac{(z-1) R^{\mathsf{d}}(z)}{1 + G^{\mathsf{d}}(z)} = \frac{T}{\lim_{z \to 1} (z-1) G^{\mathsf{d}}(z)} \\ &= \begin{cases} \infty, & \text{if } N = 0 \\ \text{constant } \neq 0, & \text{if } N = 1 \\ 0, & \text{if } N \geq 2 \end{cases} \end{split}$$

Lesson 2: A system with type $N \geq 2$ has zero-steady-state error in ramp response.

Example 6.8: Role of integrator in control

6.5. Steady-state Accuracy



$$G(s) = \frac{1}{s+1} \frac{1 - e^{-Ts}}{s}, \quad D^{\mathsf{d}}(z) = \frac{K_I z}{z-1} + K_P$$

Questions:

- ► Steady-state error for the step input?
- Steady-state error for the ramp input?

Note: A generalization of this concept = Internal model principle (내부모델이론) That is, to regulate the reference or compensate the disturbance exactly, $D^{\rm d}(z)$ should have the generating model (such as z/z-1 for step reference)