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ME 493: Intermediate Dynamics

Dr. Fields - Spring 2019

Exam II

9 May, 2019

Problem 1a: Horizontal Disk With Bar and Spring - Derive Kane's Equations

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(1) % ME 493: Intermediate Dynamics
(2) % Final Exam Problem 1
(3) NewtonianFrame N
(4) RigidBody D, A
(5) Constant L, rD, g, k, Id, Jx, Jy
(6) Point R(A), P(A), Q()
(7) Variable Q1', Q2', U1', U2'
(8) SetGeneralizedSpeed(U1, U2)
(9) D.SetMass(Md)
(10) D.SetInertia(Dcm, Id*Nx>*Nx> + 0*Ny>*Ny> + 0*Nz>*Nz>)
(11) % D.SetInertia(Dcm, Md*rD^2/4*(Dx>*Dx> + Dy>*Dy> + 2*Dz>*Dz>))
(12) A.SetMass(Ma)
(13) A.SetInertia(Acm, (Jx*Ax>*Ax> + Jy*Ay>*Ay> + Jy*Az>*Az>))
(14) rD = 0.7*L
-> (15) rD = 0.7*L

(16) g = 9.81
-> (17) g = 9.81

(18) %%% Kinematical Equations %%%
(19) Q1' = U1
-> (20) Q1' = U1

(21) Q2' = U2
-> (22) Q2' = U2

(23) %%% Rotations %%%
(24) D.RotateX(N, -Q1)
-> (25) D_N = [1, 0, 0; 0, cos(Q1), -sin(Q1); 0, sin(Q1), cos(Q1)]
-> (26) w_D_N> = -U1*Nx>
-> (27) alf_D_N> = -U1*Nx>

(28) A.RotateY(D, Q2)
-> (29) A_D = [cos(Q2), 0, -sin(Q2); 0, 1, 0; sin(Q2), 0, cos(Q2)]
-> (30) w_A_D> = U2*Ay>
-> (31) w_A_N> = -cos(Q2)*U1*Ax> + U2*Ay> - sin(Q2)*U1*Az>
-> (32) alf_A_D> = U2'*Ay>
-> (33) alf_A_N> = (sin(Q2)*U1*U2-cos(Q2)*U1')*Ax> + U2'*Ay> + (-cos(Q2)*U1*U2-
    sin(Q2)*U1')*Az>

(34) %%% Translations %%%
(35) % R.SetPosition(Dcm, rD*Dy>)
(36) Ao.SetPosition(Dcm, -rD*Dy>)

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-> (37) $p_{Dcm_Ao} = -r_D \cdot Dy$

(38) $Acm.SetPosition(Ao, 0.5 \cdot L \cdot Ax)$

-> (39) $p_{Ao_Acm} = 0.5 \cdot L \cdot Ax$

(40) $P.SetPosition(Ao, L \cdot Ax)$

-> (41) $p_{Ao_P} = L \cdot Ax$

(42) $Q.SetPosition(Dcm, 3 \cdot L \cdot Dx)$

-> (43) $p_{Dcm_Q} = 3 \cdot L \cdot Dx$

(44) %%% Velocities %%%

(45) $Acm.SetVelocityAcceleration(N, 0.7 \cdot U1 \cdot Ax + U1 \cdot Ay + U2 \cdot Az)$

-> (46) $v_{Acm_N} = 0.7 \cdot U1 \cdot Ax + U1 \cdot Ay + U2 \cdot Az$

-> (47) $a_{Acm_N} = (U2^2 + \sin(Q2) \cdot U1^2 + 0.7 \cdot U1') \cdot Ax + (\cos(Q2) \cdot U1 \cdot U2 + U1' - 0.7 \cdot \sin(Q2) \cdot U1^2) \cdot Ay + (U2' - 0.7 \cdot U1 \cdot U2 - \cos(Q2) \cdot U1^2) \cdot Az$

(48) $Dcm.SetVelocityAcceleration(N, 0)$

-> (49) $v_{Dcm_N} = 0$

-> (50) $a_{Dcm_N} = 0$

(51) $Q.SetVelocityAcceleration(N, 0)$

-> (52) $v_{Q_N} = 0$

-> (53) $a_{Q_N} = 0$

(54) %%% Partial Velocities %%%

(55) $VD_Partials = Dcm.GetPartialVelocity(N)$

-> (56) $VD_Partials = [0; 0]$

(57) $VDcm_Partial1 = VD_Partials[1]$

-> (58) $VDcm_Partial1 = 0$

(59) $VDcm_Partial2 = VD_Partials[2]$

-> (60) $VDcm_Partial2 = 0$

(61) $VA_Partials = Acm.GetPartialVelocity(N)$

-> (62) $VA_Partials = [0.7 \cdot Ax + Ay; Az]$

(63) $VAcmm_Partial1 = VA_Partials[1]$

-> (64) $VAcmm_Partial1 = 0.7 \cdot Ax + Ay$

(65) $VAcmm_Partial2 = VA_Partials[2]$

-> (66) $VAcmm_Partial2 = Az$

(67) $WD_Partial1 = D.GetPartialAngularVelocity(N, U1)$

-> (68) $WD_Partial1> = -Nx>$

(69) $WD_Partial2> = D.GetPartialAngularVelocity(N, U2)$

-> (70) $WD_Partial2> = 0>$

(71) $WA_Partial1> = A.GetPartialAngularVelocity(N, U1)$

-> (72) $WA_Partial1> = -\cos(Q2)*Ax> - \sin(Q2)*Az>$

(73) $WA_Partial2> = A.GetPartialAngularVelocity(N, U2)$

-> (74) $WA_Partial2> = Ay>$

(75) %%% Forces %%%

(76) $System.AddForceGravity(g*Nx>)$

-> (77) $Force_Acm> = Ma*g*Nx>$

-> (78) $Force_Dcm> = Md*g*Nx>$

(79) $P.AddForceSpring(Q, k, L)$

-> (80) $Force_P_Q> = -k*L*(1-L/\sqrt{rD^2+10*L^2-6*L^2*\cos(Q2)})*Ax> + 3*k*L*(1-L/\sqrt{rD^2+10*L^2-6*L^2*\cos(Q2)})*Dx> + k*rD*(1-L/\sqrt{rD^2+10*L^2-6*L^2*\cos(Q2)})*Dy>$

(81) %%% Effective Moments %%%

(82) $NMD_Dcm> = \text{dot}(I_D_Dcm>, \text{alf_D_N>}) + \text{cross}(w_D_N>, \text{dot}(I_D_Dcm>,>, w_D_N>))$

-> (83) $NMD_Dcm> = -Id*U1'*Nx>$

(84) $NMA_Acm> = \text{dot}(I_A_Acm>,>, \text{alf_A_N>}) + \text{cross}(w_A_N>, \text{dot}(I_A_Acm>,>, w_A_N>))$

-> (85) $NMA_Acm> = Jx*(\sin(Q2)*U1*U2-\cos(Q2)*U1')*Ax> +$

$(Jx*\sin(Q2)*\cos(Q2)*U1^2 + Jy*U2'-Jy*\sin(Q2)*\cos(Q2)*U1^2)*Ay> + (Jx*\cos(Q2)*U1*U2-2*Jy*\cos(Q2)*U1*U2-Jy*\sin(Q2)*U1')*Az>$

(86) %%% Kane's Equations %%%

(87) $F1 = \text{dot}(A.GetResultantForce(), VAcM_Partial1>) + \text{dot}(D.GetResultantForce(), VDcm_Partial1>) + \text{dot}(A.GetResultantMoment(Acm), WA_Partial1>) + \text{dot}(D.GetResultantMoment(Dcm), WD_Partial1>)$

-> (88) $F1 = k*rD*(1-L/\sqrt{rD^2+10*L^2-6*L^2*\cos(Q2)}) + 0.7*\cos(Q2)*(Ma*g+3*k*L*(1-L/\sqrt{rD^2+10*L^2-6*L^2*\cos(Q2)})) - 0.7*k*L*(1-L/\sqrt{rD^2+10*L^2-6*L^2*\cos(Q2)}) - 0.5*k*L*rD*\sin(Q2)*(1-L/\sqrt{rD^2+10*L^2-6*L^2*\cos(Q2)})$

(89) $F1_N = \text{dot}(Ma*a_Acm_N>, VAcM_Partial1>) + \text{dot}(Md*a_Dcm_N>, VDcm_Partial1>) + \text{dot}(NMA_Acm>, WA_Partial1>) + \text{dot}(NMD_Dcm>, WD_Partial1>)$

-> (90) $F1_N = Id*U1' + 0.7*Ma*(U2^2+1.428571*\cos(Q2)*U1*U2+2.128571*U1')$

$$+ J_y \sin(Q2) * (2 * \cos(Q2) * U1 * U2 + \sin(Q2) * U1') - J_x \cos(Q2) * (2 * \sin(Q2) * U1 * U2 - \cos(Q2) * U1')$$

$$(91) F2 = \text{dot}(A.\text{GetResultantForce}(), V_{Acm_Partial2}) + \text{dot}(D.\text{GetResultantForce}(), V_{Dcm_Partial2}) + \text{dot}(A.\text{GetResultantMoment}(A_{cm}), W_{A_Partial2}) + \text{dot}(D.\text{GetResultantMoment}(D_{cm}), W_{D_Partial2})$$

$$\rightarrow (92) F2 = \sin(Q2) * (M_a * g + 3 * k * L * (1 - L / \sqrt{r_D^2 + 10 * L^2 - 6 * L^2 * \cos(Q2)})) - 1.5 * k * L^2 * (1 - L / \sqrt{r_D^2 + 10 * L^2 - 6 * L^2 * \cos(Q2)})$$

$$(93) F2_N = \text{dot}(M_a * a_{Acm_N}, V_{Acm_Partial2}) + \text{dot}(M_d * a_{Dcm_N}, V_{Dcm_Partial2}) + \text{dot}(NMA_{Acm}, W_{A_Partial2}) + \text{dot}(NMD_{Dcm}, W_{D_Partial2})$$

$$\rightarrow (94) F2_N = J_x \sin(Q2) * \cos(Q2) * U1^2 + J_y * U2' - J_y \sin(Q2) * \cos(Q2) * U1^2 - 0.7 * M_a * (U1 * U2 + 1.428571 * \cos(Q2) * U1^2 - 1.428571 * U2')$$

$$(95) \text{zero}[1] = \text{explicit}(F1 - F1_N)$$

$$\rightarrow (96) \text{zero}[1] = 0.1428869 * k * L^2 * \sin(Q2) * (-2.44949 + L / \sqrt{L^2 * (-1.748333 + \cos(Q2))}) + 0.8573214 * \cos(Q2) * (8.009831 * M_a - k * L * (-2.44949 + L / \sqrt{L^2 * (-1.748333 + \cos(Q2))})) + J_x \cos(Q2) * (2 * \sin(Q2) * U1 * U2 - \cos(Q2) * U1') - I_d * U1' - 0.7 * M_a * (U2^2 + 1.428571 * \cos(Q2) * U1 * U2 + 2.128571 * U1') - J_y \sin(Q2) * (2 * \cos(Q2) * U1 * U2 + \sin(Q2) * U1')$$

$$(97) \text{zero}[2] = \text{explicit}(F2 - F2_N)$$

$$\rightarrow (98) \text{zero}[2] = J_y \sin(Q2) * \cos(Q2) * U1^2 + 0.7 * M_a * (U1 * U2 + 1.428571 * \cos(Q2) * U1^2 - 1.428571 * U2') - 0.6123724 * \sin(Q2) * (2 * k * L * (-2.44949 + L / \sqrt{L^2 * (-1.748333 + \cos(Q2))})) - 16.01966 * M_a - k * L^2 * (-2.44949 + L / \sqrt{L^2 * (-1.748333 + \cos(Q2))})) - J_x \sin(Q2) * \cos(Q2) * U1^2 - J_y * U2'$$

$$(99) \text{zero_auto} = \text{System.GetDynamicsKane}()$$

$$\rightarrow (100) \text{zero_auto}[1] = (1.49 * M_a + I_d + J_x \cos(Q2)^2 + J_y \sin(Q2)^2) * U1' - 0.7 * M_a * g * \cos(Q2) - 0.5 * k * (4.2 * L * \cos(Q2) - 1.4 * L - r_D * (-2 + L * \sin(Q2))) * (1 - L / \sqrt{r_D^2 + 10 * L^2 - 6 * L^2 * \cos(Q2)}) - 0.7 * U2 * (2.857143 * J_x \sin(Q2) * \cos(Q2) * U1 - 2.857143 * J_y \sin(Q2) * \cos(Q2) * U1 - M_a * (U2 + 1.428571 * \cos(Q2) * U1))$$

$$\rightarrow (101) \text{zero_auto}[2] = (M_a + J_y) * U2' - \sin(Q2) * (M_a * g - 1.5 * k * L * (-2 + L) * (1 - L / \sqrt{r_D^2 + 10 * L^2 - 6 * L^2 * \cos(Q2)})) - 0.7 * U1 * (M_a * (U2 + 1.428571 * \cos(Q2) * U1) - 1.428571 * (J_x - J_y) * \sin(Q2) * \cos(Q2) * U1)$$

$$(102) \text{check} = \text{explicit}(\text{zero} + \text{zero_auto})$$

$$\rightarrow (103) \text{check} = [0; 0]$$

$$(104) \% \text{ODE}(\text{zero}, U1', U2') \text{Exam1_matlab.m}$$

Problem 1b: Would it have been easier to use generalized speeds defined by

$${}^N\overline{V}^{Acm} = U_1 \hat{a}_x + U_2 \hat{a}_y + U_3 \hat{a}_z ?$$

No. All of the degrees of freedom are accounted for using Q_1 and Q_2 . Adding a third variable and associated generalized speed would add a constraint between variables. This is supported by the geometry governing where \hat{a}_z is able to move since it is attached to the disk at point R.

Problem 1c: Does your answer for the velocity of ${}^N\overline{V}^{Acm}$ make sense?

Yes, the position of the center of mass for bar A in the x and y direction is heavily dependent on the rotation angle of disk D due to the attachment point at R. If we think generally about how this object would behave in the real world, we could spin the disk and the spring would apply significant velocity on the bar in the z-direction but the remaining velocities would be set by the disk above.

Problem 2a: Double Arm Welded to Disk on Slope - Derive Kane's Equations

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(1) % ME 493: Intermediate Dynamics
(2) % Final Exam Problem 2
(3) NewtonianFrame N
(4) RigidBody A, B, C
(5) RigidFrame D
(6) Point P(), Q()
(7) Constant L, rB, g, tB
(8) Variable Q{1:3}', U{1:3}'
(9) A.SetMass(Ma)
(10) A.SetInertia(Acm, 0, 0, IA)
(11) B.SetMass(Mb)
(12) B.SetInertia(Bcm, 0, 0, jB)
(13) % B.SetInertia(Bcm, (0*Bx>*Bx> + 0*By>*By> + jB*Bz>*Bz>))
(14) % B.SetInertia(Bcm, (0*Nx>*Nx> + 0*Ny>*Ny> + jB*Nz>*Nz>))
(15) % B.SetInertia(Bcm, Mb*rB^2/4*(Bx>*Bx> + By>*By> + 2*Bz>*Bz>))
(16) C.SetMass(Mc)
(17) C.SetInertia(Ccm, 0, IC, 0)
(18) SetGeneralizedSpeed(U1, U2, U3)
(19) %%% Kinematical Equations %%%
(20) Q1' = U1
-> (21) Q1' = U1

(22) Q2' = U2
-> (23) Q2' = U2

(24) Q3' = U3
-> (25) Q3' = U3

(26) %%% Rotations %%%
(27) A.RotateZ(N, tB+Q3)
-> (28) A_N = [cos(tB+Q3), sin(tB+Q3), 0; -sin(tB+Q3), cos(tB+Q3), 0; 0, 0, 1]
-> (29) w_A_N> = U3*Az>
-> (30) alf_A_N> = U3'*Az>

(31) B.RotateZ(N, tB+Q3)
-> (32) B_N = [cos(tB+Q3), sin(tB+Q3), 0; -sin(tB+Q3), cos(tB+Q3), 0; 0, 0, 1]
-> (33) w_B_N> = U3*Bz>
-> (34) alf_B_N> = U3'*Bz>

(35) C.Rotate(A, BodyY, -Q1)
-> (36) C_A = [cos(Q1), 0, sin(Q1); 0, 1, 0; -sin(Q1), 0, cos(Q1)]
-> (37) w_C_A> = -U1*Cy>
-> (38) w_C_N> = sin(Q1)*U3*Cx> - U1*Cy> + cos(Q1)*U3*Cz>

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$$\rightarrow (39) \text{alf_C_A} > = -U1 * Cy >$$

$$\rightarrow (40) \text{alf_C_N} > = (\cos(Q1) * U1 * U3 + \sin(Q1) * U3') * Cx > - U1 * Cy > + (\cos(Q1) * U3' - \sin(Q1) * U1 * U3) * Cz >$$

$$(41) \text{D.RotateZ(N, tB)}$$

$$\rightarrow (42) \text{D_N} = [\cos(tB), \sin(tB), 0; -\sin(tB), \cos(tB), 0; 0, 0, 1]$$

$$\rightarrow (43) \text{w_D_N} > = 0 >$$

$$\rightarrow (44) \text{alf_D_N} > = 0 >$$

$$(45) \text{%%\% Translations \% \% \%}$$

$$(46) \text{Ao.Translate(No, Q2 * By >)}$$

$$\rightarrow (47) \text{p_No_Ao} > = Q2 * By >$$

$$\rightarrow (48) \text{v_Ao_N} > = -Q2 * U3 * Bx > + U2 * By >$$

$$\rightarrow (49) \text{a_Ao_N} > = (-2 * U2 * U3 - Q2 * U3') * Bx > + (U2' - Q2 * U3^2) * By >$$

$$(50) \text{Do.Translate(Ao, 0 >)}$$

$$\rightarrow (51) \text{p_Ao_Do} > = 0 >$$

$$\rightarrow (52) \text{v_Do_N} > = -Q2 * U3 * Bx > + U2 * By >$$

$$\rightarrow (53) \text{a_Do_N} > = (-2 * U2 * U3 - Q2 * U3') * Bx > + (U2' - Q2 * U3^2) * By >$$

$$(54) \text{Bcm.Translate(Ao, 0 >)}$$

$$\rightarrow (55) \text{p_Ao_Bcm} > = 0 >$$

$$\rightarrow (56) \text{v_Bcm_N} > = -Q2 * U3 * Bx > + U2 * By >$$

$$\rightarrow (57) \text{a_Bcm_N} > = (-2 * U2 * U3 - Q2 * U3') * Bx > + (U2' - Q2 * U3^2) * By >$$

$$(58) \text{Acm.Translate(Ao, L * Ax >)}$$

$$\rightarrow (59) \text{p_Ao_Acm} > = L * Ax >$$

$$\rightarrow (60) \text{v_Acm_N} > = L * U3 * Ay > - Q2 * U3 * Bx > + U2 * By >$$

$$\rightarrow (61) \text{a_Acm_N} > = -L * U3^2 * Ax > + L * U3' * Ay > + (-2 * U2 * U3 - Q2 * U3') * Bx > + (U2' - Q2 * U3^2) * By >$$

$$(62) \text{Q.Translate(Ao, 2 * L * Ax >)}$$

$$\rightarrow (63) \text{p_Ao_Q} > = 2 * L * Ax >$$

$$\rightarrow (64) \text{v_Q_N} > = 2 * L * U3 * Ay > - Q2 * U3 * Bx > + U2 * By >$$

$$\rightarrow (65) \text{a_Q_N} > = -2 * L * U3^2 * Ax > + 2 * L * U3' * Ay > + (-2 * U2 * U3 - Q2 * U3') * Bx > + (U2' - Q2 * U3^2) * By >$$

$$(66) \text{Co.Translate(Q, -1.5 * L * Cz >)}$$

$$\rightarrow (67) \text{p_Q_Co} > = -1.5 * L * Cz >$$

$$\rightarrow (68) \text{v_Co_N} > = 1.5 * L * (1.333333 + \sin(Q1)) * U3 * Ay > - Q2 * U3 * Bx > + U2 * By > + 1.5 * L * U1 * Cx >$$

$$\rightarrow (69) \text{a_Co_N} > = -2 * L * U3^2 * Ax > + 1.5 * L * (2 * \cos(Q1) * U1 * U3 + 1.333333 * U3' + \sin(Q1) * U3') * Ay > + (-2 * U2 * U3 - Q2 * U3') * Bx > + (U2' - Q2 * U3^2) * By > - 1.5 * L * (\sin(Q1) * \cos(Q1) * U3^2 - U1') * Cx > + 1.5 * L * (U1^2 + \sin(Q1)^2 * U3^2) * Cz >$$


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(70) Ccm.Translate(Q, -0.75*L*Cz>)
-> (71) p_Q_Ccm> = -0.75*L*Cz>
-> (72) v_Ccm_N> = 0.75*L*(2.666667+sin(Q1))*U3*Ay> - Q2*U3*Bx> + U2*By> +
0.75*L*U1*Cx>
-> (73) a_Ccm_N> = -2*L*U3^2*Ax> +
0.75*L*(2*cos(Q1)*U1*U3+2.666667*U3'+sin(Q1)
*U3')*Ay> + (-2*U2*U3-Q2*U3')*Bx> + (U2'-Q2*U3^2)*By> - 0.75*L*(sin(Q1)
*cos(Q1)*U3^2-U1')*Cx> + 0.75*L*(U1^2+sin(Q1)^2*U3^2)*Cz>

(74) P.Translate(Do, -rB*Bx>)
-> (75) p_Do_P> = -rB*Bx>
-> (76) v_P_N> = -Q2*U3*Bx> + (U2-rB*U3)*By>
-> (77) a_P_N> = (rB*U3^2-2*U2*U3-Q2*U3')*Bx> + (U2'-Q2*U3^2-rB*U3')*By>

(78) %%% Constraints %%%
(79) Dependent[1] = dot(V_P_N>, By>)
-> (80) Dependent[1] = U2 - rB*U3

(81) Constrain(Dependent[U2])
-> (82) U2 = rB*U3
-> (83) U2' = rB*U3'

(84) %%% Forces %%%
(85) System.AddForceGravity(g*Nx>)
-> (86) Force_Acm> = Ma*g*Nx>
-> (87) Force_Bcm> = Mb*g*Nx>
-> (88) Force_Ccm> = Mc*g*Nx>

(89) zero = System.GetDynamicsKane()
-> (90) zero[1] = 0.5625*(1.777778*IC+Mc*L^2)*U1' - 0.5625*Mc*L*cos(Q1)*(1.333333
*g*cos(tB+Q3)+U3*(2.666667*U2+2.666667*L*U3+L*sin(Q1)*U3)) - 0.75*Mc*L*
Q2*cos(Q1)*U3'

-> (91) zero[2] = 0.75*g*(1.333333*Mb*(rB*sin(tB+Q3)+Q2*cos(tB+Q3))+1.333333*
Ma*(L*sin(tB+Q3)+rB*sin(tB+Q3)+Q2*cos(tB+Q3))+Mc*(1.333333*rB*sin(tB+
Q3)+1.333333*Q2*cos(tB+Q3)+L*sin(tB+Q3)*(2.666667+sin(Q1)))) + 2*Ma*Q2*
U2*U3 + 2*Mb*Q2*U2*U3 +
0.75*Mc*(2.666667*Q2*U2*U3+L*Q2*sin(Q1)*U1^2+1.5
*L^2*cos(Q1)*(2.666667+sin(Q1))*U1*U3) + 0.5625*(1.777778*IA+1.777778*
jB+1.777778*Mb*Q2^2+1.777778*Ma*(L^2+2*L*rB+Q2^2)+Mc*(1.777778*Q2^2+L^2
*(2.666667+sin(Q1))^2+1.777778*rB*(Ma*rB+Mb*rB+Mc*(rB+1.5*L*(2.666667+
sin(Q1)))))*U3' - rB*U3*(Ma*Q2*U3+Mb*Q2*U3+Mc*(Q2*U3-1.5*L*cos(Q1)*U1))
- 0.75*Mc*L*Q2*cos(Q1)*U1'

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- (92) Input $t_{\text{Final}} = 40.0$, $t_{\text{Step}} = 0.1$, $\text{absError} = 1.0\text{E-}07$
- (93) Input $t_B = 0.1745$ rad, $g = 9.81$ m/s², $r_B = 0.3$ m
- (94) Input $M_b = 1.0$ kg, $j_B = 0.06$ kg*m², $L = 0.8$ m
- (95) Input $M_a = 0.5$ kg, $I_A = 0.12$ kg*m², $M_c = 0.3$ kg
- (96) Input $I_C = 0.04$ kg*m², $Q_1 = 0.2618$ rad, $Q_2 = 0.02$ m
- (97) Input $Q_3 = 0.2618$ rad, $U_1 = 0.0873$ rad/s
- (98) Input $U_3 = 0.0$ rad/s
- (99) OutputPlot t sec, Q_1 deg, Q_2 m, Q_3 deg, U_1 rad/s, U_2 m/s, U_3 rad/s
- (100) ODE(zero, U_1' , U_3') Exam2_matlab.m

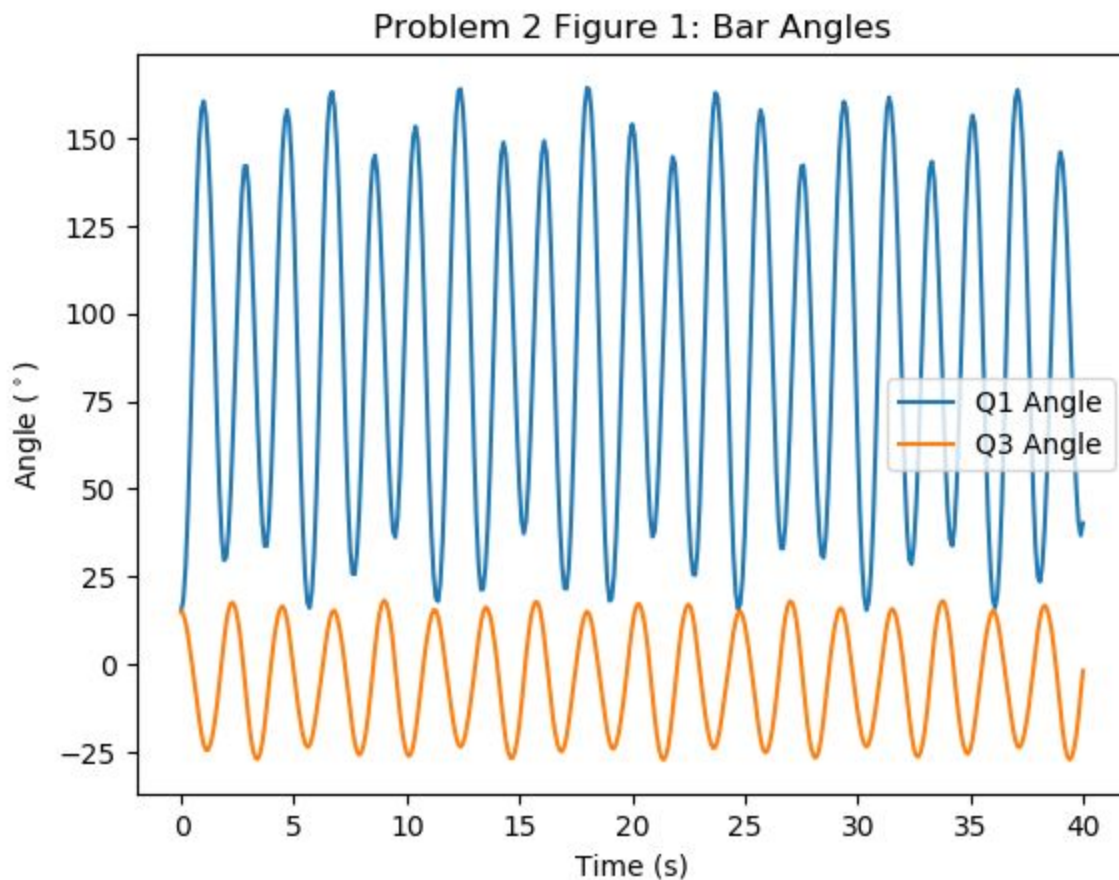
Problem 2b: How many degrees of freedom does the system have?

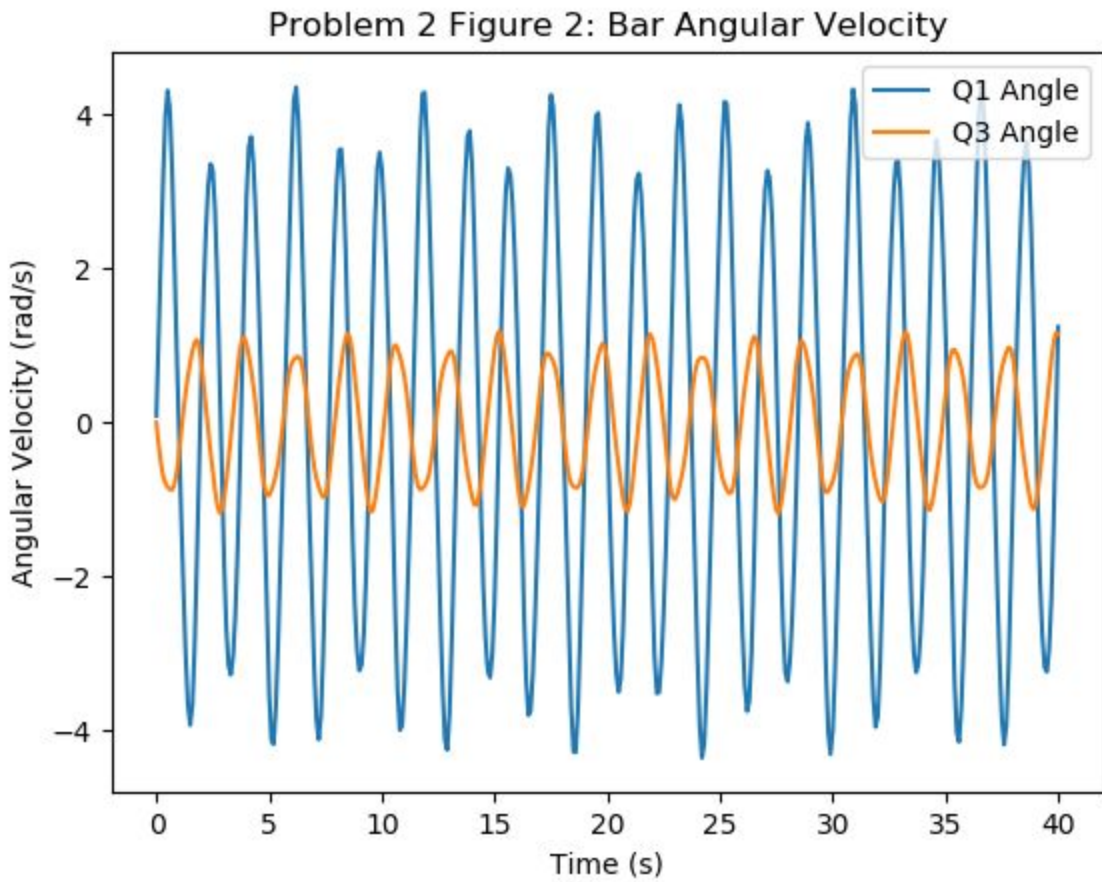
2. Since the bar A is welded to disk B, Q_2 and Q_3 are redundant. It is possible to derive the position or the angle of the top part of the arm using only one of these variables. One of the challenges in this problem was figuring out that I needed to set up an additional RigidFrame, D to describe the constrained point P at the interface between the sloped surface and the disk. The frame needed to follow the slope of the surface without turning along with the rigid bodies.

Problem 2c: If the angle θ was not constant, would you still be able to solve the problem using Kane's method?

Yes, the additional variable would change the rotation dynamics of the rigid frame and the overall behavior of the arms, but Kane's method could still handle this problem. We would need to be careful about deciding how to define the rotation of the other rigid bodies A and B.

Problem 2d: Simulation Plot





Problem 3: Comeback Can

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(1) % ME 493: Intermediate Dynamics
(2) % Final Exam Problem 3
(3) NewtonianFrame N
(4) RigidBody A, B
(5) Constant rA, rB, g, k
(6) Point P()
(7) Variable Q{1:3}', U{1:3}'
(8) A.SetMass(mA)
(9) B.SetMass(mB)
(10) A.SetInertia(Acm, IA, 0, 0)
(11) B.SetInertia(Bcm, IB, 0, 0)
(12) % A.SetInertia(Acm, mA*rA^2*(Ax>*Ax>))
(13) % B.SetInertia(Bcm, mB*rB^2/4*(2*Bx>*Bx> + By>*By> + Bz>*Bz>))
(14) SetGeneralizedSpeed(U1, U2, U3)
(15) %%% Kinematical Equations %%%
(16) Q1' = U1
-> (17) Q1' = U1

(18) Q2' = U2
-> (19) Q2' = U2

(20) Q3' = U3
-> (21) Q3' = U3

(22) %%% Rotations %%%
(23) A.RotateX(N, Q1)
-> (24) A_N = [1, 0, 0; 0, cos(Q1), sin(Q1); 0, -sin(Q1), cos(Q1)]
-> (25) w_A_N> = U1*Ax>
-> (26) alf_A_N> = U1'*Ax>

(27) B.RotateX(A, -Q3)
-> (28) B_A = [1, 0, 0; 0, cos(Q3), -sin(Q3); 0, sin(Q3), cos(Q3)]
-> (29) w_B_A> = -U3*Bx>
-> (30) w_B_N> = (U1-U3)*Bx>
-> (31) alf_B_A> = -U3'*Bx>
-> (32) alf_B_N> = (U1'-U3')*Bx>

(33) %%% Translations %%%
(34) P.SetPosition(Acm, -rA*Nz>)
-> (35) p_Acm_P> = -rA*Nz>

(36) Bcm.SetPosition(Acm, -rB*Az>)
-> (37) p_Acm_Bcm> = -rB*Az>

```

```
(38) %%% Velocities %%%
(39) P.SetVelocity(N, U2*Ny>)
-> (40) v_P_N> = U2*Ny>

(41) AcM.SetVelocityAcceleration(N, 0>)
-> (42) v_AcM_N> = 0>
-> (43) a_AcM_N> = 0>

(44) BcM.SetVelocityAcceleration(N, 0>)
-> (45) v_BcM_N> = 0>
-> (46) a_BcM_N> = 0>

(47) %%% Constraints %%%
(48) Dependent[1] = dot(v_P_N>, Ny>)
-> (49) Dependent[1] = U2

(50) Constrain(Dependent[U2])
-> (51) U2 = 0
-> (52) U2' = 0

(53) %%% Forces %%%
(54) System.AddForceGravity( -g*Nz> )
-> (55) Force_AcM> = -mA*g*Nz>
-> (56) Force_BcM> = -mB*g*Nz>

(57) B.AddTorque(A, -2*k*Q3*Ax>)
-> (58) Torque_B_A> = -2*k*Q3*Ax>

(59) zero = System.GetDynamicsKane()
-> (60) zero = [(IA+IB)*U1' - IB*U3'; IB*U3' - 2*k*Q3 - IB*U1']

(61) % ODE(zero, U1', U3') Exam3_matlab.m
```

Problem 4a: Restrained Double Pendulum

```

(1) % ME 493: Intermediate Dynamics
(2) % Final Exam Problem 4
(3) NewtonianFrame N
(4) RigidFrame A, B
(5) Constant IA, IB, g, k, lko, bA, bB
(6) Particle P, Q
(7) Variable Q{1:2}', U{1:2}'
(8) P.SetMass(mP)
(9) Q.SetMass(mQ)
(10) SetGeneralizedSpeed(U1, U2)
(11) %%% Kinematical Equations %%%
(12) Q1' = U1
-> (13) Q1' = U1

(14) Q2' = U2
-> (15) Q2' = U2

(16) %%% Rotations %%%
(17) A.RotateZ(N, Q1)
-> (18) A_N = [cos(Q1), sin(Q1), 0; -sin(Q1), cos(Q1), 0; 0, 0, 1]
-> (19) w_A_N> = U1*Az>
-> (20) alf_A_N> = U1'*Az>

(21) B.RotateZ(A, Q2)
-> (22) B_A = [cos(Q2), sin(Q2), 0; -sin(Q2), cos(Q2), 0; 0, 0, 1]
-> (23) w_B_A> = U2*Bz>
-> (24) w_B_N> = (U1+U2)*Az>
-> (25) alf_B_A> = U2'*Bz>
-> (26) alf_B_N> = (U1'+U2')*Az>

(27) %%% Velocities %%%
(28) Ao.SetVelocityAcceleration(N, 0>)
-> (29) v_Ao_N> = 0>
-> (30) a_Ao_N> = 0>

(31) %%% Translations %%%
(32) P.Translate(Ao, IA*Ax>)
-> (33) p_Ao_P> = IA*Ax>
-> (34) v_P_N> = IA*U1'*Ay>
-> (35) a_P_N> = -IA*U1'^2*Ax> + IA*U1'*Ay>

(36) Q.Translate(P, IB*Bx>)
-> (37) p_P_Q> = IB*Bx>

```

$$\rightarrow (38) v_{Q_N} = I_A U_1' A_y + I_B (U_1 + U_2)' B_y$$

$$\rightarrow (39) a_{Q_N} = -I_A U_1'^2 A_x + I_A U_1' A_y - I_B (U_1 + U_2)^2 B_x + I_B (U_1' + U_2') B_y$$

(40) %%% Forces %%%

(41) System.AddForceGravity(g*Nx)

$$\rightarrow (42) \text{Force}_P = m_P g N_x$$

$$\rightarrow (43) \text{Force}_Q = m_Q g N_x$$

(44) Q.AddForceSpring(Ao, k, lko)

$$\rightarrow (45) \text{Force}_{Q_{Ao}} = -k I_A (1 - l_{ko} / \sqrt{I_A^2 + I_B^2 + 2 I_A I_B \cos(Q_2)}) A_x - k I_B (1 - l_{ko} / \sqrt{I_A^2 + I_B^2 + 2 I_A I_B \cos(Q_2)}) B_x$$

(46) A.AddTorque(-bA*U1*Nz)

$$\rightarrow (47) \text{Torque}_A = -b_A U_1 N_z$$

(48) B.AddTorque(-bB*U2*Nz)

$$\rightarrow (49) \text{Torque}_B = -b_B U_2 N_z$$

(50) zero = System.GetDynamicsKane()

$$\rightarrow (51) \text{zero}[1] = m_P g I_A \sin(Q_1) + m_Q g (I_A \sin(Q_1) + I_B \sin(Q_1 + Q_2)) + b_A U_1 + b_B U_2 + m_Q I_A I_B \sin(Q_2) (U_1'^2 - (U_1 + U_2)^2) + m_Q I_B (I_B + I_A \cos(Q_2)) U_2' + (m_P I_A^2 + m_Q (I_A^2 + I_B^2 + 2 I_A I_B \cos(Q_2))) U_1'$$

$$\rightarrow (52) \text{zero}[2] = m_Q g I_B \sin(Q_1 + Q_2) + b_B U_2 + m_Q I_A I_B \sin(Q_2) U_1'^2 + m_Q I_B^2 U_2' + m_Q I_B (I_B + I_A \cos(Q_2)) U_1' - k I_A I_B \sin(Q_2) (1 - l_{ko} / \sqrt{I_A^2 + I_B^2 + 2 I_A I_B \cos(Q_2)})$$

(53) Input tFinal = 30, tStep = 0.1, absError = 1.0E-07

(54) Input bA = 100 N*s/m, bB = 100 N*s/m, g = 9.81 m/s^2

(55) Input k = 250 N/m, I_A = 1.5 m, I_B = 2.0 m

(56) Input lko = 1.0 m, mP = 10.0 kg, mQ = 15.0 kg

(57) Input Q1 = 1.5708 rad, Q2 = 1.5708 rad

(58) Input U1 = 0.0 rad, U2 = 0.0 rad

(59) OutputPlot t sec, Q1 rad, Q2 rad, U1 rad/s, U2 rad/s

Problem 4b: Simulation Plot