CIEM5000: Structural Engineering Base Matrix Method – Final Details

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The Matrix Method

Main steps:

- Extract element matrices
- Impose nodal equilibrium
- Impose boundary conditions
- Solve for unknown displacements
- Postprocess results

This week:

- Element loads
- Non-zero Dirichlet boundary conditions in two different ways
- Postprocessing: support reactions and element fields
- Matrix method versus FEM parallels and differences
- Example: A fully-resolved example by hand
- Workshop: Wrap up the code and solve a frame structure

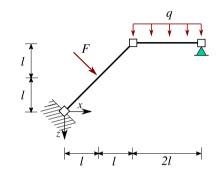
Flement loads

The matrix method is a discrete approach

- Nodal loads treated easily
- What if we have loads applied inside elements?

A number of approaches to handle this:

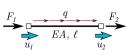
- Further discretization (seldom helps)
- ODE approach
- Work-based approach



Element loads — ODE approach

We can follow the same steps as before:
General solution for the ODE:

$$EA\frac{\mathrm{d}^2 u}{\mathrm{d}x^2} = -q \quad \Rightarrow \quad u(x) = -\frac{qx^2}{2EA} + C_1x + C_2$$



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Boundary conditions and final solution:

$$u(0) = u_1$$
 $u(\ell) = u_2$ \Rightarrow $C_1 = \frac{q\ell}{2EA} + \frac{u_2 - u_1}{\ell}$ $C_2 = u_1$
 $u(x) = \frac{q}{2EA} (\ell x - x^2) + u_1 (1 - \frac{x}{\ell}) + \frac{u_2 x}{\ell}$



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Relate forces at the edges with internal stresses:

$$N(x) = \frac{EA}{\ell} (u_2 - u_1) + \frac{q\ell - 2qx}{2}$$
 $F_1 = -N_1$ $F_2 = N_2$



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Relate forces and displacements, but now an extra term appears:

$$\frac{EA}{\ell} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} - \begin{bmatrix} \frac{q\ell}{2} \\ \frac{q\ell}{2} \end{bmatrix} = \begin{bmatrix} F_1 \\ F_2 \end{bmatrix}$$

Dealing with equivalent loads

The new term is an equivalent nodal load:

- Element loads ⇒ nodal loads
- Force equilibrium at the nodes therefore changes a bit:

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Remember, this is a force in the global coordinate system!

$$\mathbf{f}_{\mathrm{eq}} = \mathbf{T}^{\mathrm{T}} \overline{\mathbf{f}}_{\mathrm{eq}}$$

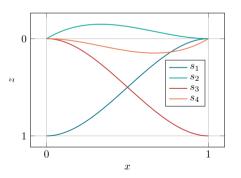


TUDelft

Euler-Bernoulli bending, point load at midspan:

• Displacement field for arbitrary DOFs (ODE with q=0):

$$w(x) = \underbrace{\left(\frac{2x^3}{\ell^3} - \frac{3x^2}{\ell^2} + 1\right)}_{s_1} w_1 + \underbrace{\left(-\frac{x^3}{\ell^2} + \frac{2x^2}{\ell} - x\right)}_{s_2} \varphi_1 + \underbrace{\left(-\frac{2x^3}{\ell^3} + \frac{3x^2}{\ell^2}\right)}_{s_3} w_2 + \underbrace{\left(-\frac{x^3}{\ell^2} + \frac{x^2}{\ell}\right)}_{s_4} \varphi_2$$



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Work performed by edge forces:

$$W_F = F_1^{\text{eq}} w_1 + T_1^{\text{eq}} \varphi_1 + F_2^{\text{eq}} w_2 + T_2^{\text{eq}} \varphi_2$$

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• Work performed by *P* under the same displacement:

$$W_q = Pw(\ell/2) = (Ps_1(\ell/2)) w_1 + (Ps_2(\ell/2)) \varphi_1 + (Ps_3(\ell/2)) w_2 + (Ps_4(\ell/2)) \varphi_2$$

ŤuDelft 6 / 13

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• Enforcing $W_F = W_q$ and isolating terms:

$$\mathbf{f_{eq}} = \begin{bmatrix} F_1^{\text{eq}} \\ T_1^{\text{eq}} \\ T_2^{\text{eq}} \\ \end{bmatrix} = \begin{bmatrix} \frac{P}{2} \\ -\frac{P\ell}{8} \\ \frac{P}{2} \\ \frac{P\ell}{8} \end{bmatrix}$$

TUDelft

Euler-Bernoulli bending, constant distributed load:

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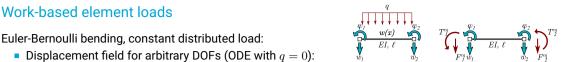
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TUDelft

 $\begin{array}{c|c} w(x) & \varphi_2 & \varphi_1 \\ \hline \vdots & EI, \ell & \vdots \\ \hline \end{array}$

Dirichlet BCs — Static condensation

Up until now displacement BCs have been simple:

- For a fixed DOF, we can simply discard the corresponding equation
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8/13

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Note: fc includes both nodal loads, equivalent loads and support reactions.

Dirichlet BCs — Size-preserving approach

The approach from before can be annoying to code:

- Reordering the system costs computation time
- Gains when inverting the stiffness matrix are very limited $(N_c \ll N_f)$

Alternatively, we can modify the relevant equations and solve the full system:

Support reactions recovered later from the unconstrained system

$$\begin{bmatrix} K_{11} & K_{12} & K_{13} \\ K_{21} & K_{22} & K_{23} \\ K_{31} & K_{32} & K_{33} \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \end{bmatrix} = \begin{bmatrix} f_1 \\ f_2 \\ f_3 \end{bmatrix}$$

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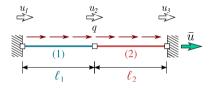
Support reactions recovered later from the unconstrained system

$$\begin{bmatrix} K_{11} & 0 & K_{13} \\ 0 & 1 & 0 \\ K_{31} & 0 & K_{33} \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \end{bmatrix} = \begin{bmatrix} f_1 - K_{12} \Delta_2 \\ \Delta_2 \\ f_3 - K_{32} \Delta_2 \end{bmatrix}$$

Example – element loads and Dirichlet BCs

Let us use what we have just learned and show a quick example:

- Same two-element bar model as before ⇒ stiffness matrix does not change!
- Added distributed load and non-zero prescribed displacement on the right



$$\mathbf{K} = \begin{bmatrix} \frac{EA_1}{\ell_1} & -\frac{EA_1}{\ell_1} & 0\\ -\frac{EA_1}{\ell_1} & \frac{EA_1}{\ell_1} + \frac{EA_2}{\ell_2} & -\frac{EA_2}{\ell_2}\\ 0 & -\frac{EA_2}{\ell_2} & \frac{EA_2}{\ell_2} \end{bmatrix}$$

Element-level postprocessing

From discrete nodal displacements to continuum element fields:

Assemble and solve the global system of equations:

 $\mathbf{K}\mathbf{u} = \mathbf{f}$

TUDelft 11/13

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From the ODE solution, recover relevant fil propose pypi as students don't neeelds as function of $\overline{\mathbf{u}}^e$:

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Finally, plot the results! Works for displacements and any internal field (e.g. moments)

Matrix method versus FEM

The two methods give the same results for bars. However:

- The matrix method solves the strong form ODEs exactly
- FEM solves the weak form problem on the shape function space
- Matrix method: strong form solved locally, elements glued together through equilibrium
- FEM: The weak form is solved globally

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Then why don't we just use the matrix method for everything?

- Gluing elements through equilibrium only works in 1D
- Exact ODE solutions in 2D generally do not exist

Example – 3D frame with torsion

Full solution by hand to demonstrate all steps:

- Definition of a new element (torsion)
- Element reduction for tractability (bending)
- Element loads, support reactions (including distributed loads), postprocessing

Values for numerical calculation:

- $EI = 1000 \text{ kNm}^2$
- $GI_t = 800 \text{ kNm}^2$
- $\ell = 2 \text{ m}$
- T = 4 kNm
- q = 6 kN/m
- = m = 2 kNm/m

