

# PICO: Probe of Inflation and Cosmic Origins

Thematic Area: Space Based Projects

Lead Author: Shaul Hanany (hanany@umn.edu)<sup>1</sup>

## Authors

|                                   |                                      |                                    |                                     |
|-----------------------------------|--------------------------------------|------------------------------------|-------------------------------------|
| Marcelo Alvarez <sup>9,16</sup>   | Brendan P. Crill <sup>6</sup>        | Eric Hivon <sup>30</sup>           | Levon Pogosian <sup>47</sup>        |
| Emmanuel Artis <sup>12</sup>      | Gianfranco De Zotti <sup>35</sup>    | Renée Hložek <sup>40</sup>         | Clem Pryke <sup>1</sup>             |
| Peter Ashton <sup>24,16,9</sup>   | Jacques Delabrouille <sup>5,12</sup> | Johannes Hubmayr <sup>19</sup>     | Giuseppe Puglisi <sup>48,49</sup>   |
| Jonathan Aumont <sup>25</sup>     | Eleonora Di Valentino <sup>36</sup>  | Bradley R. Johnson <sup>41</sup>   | Mathieu Remazeilles <sup>33</sup>   |
| Ragnhild Auriel <sup>26</sup>     | Joy Didier <sup>37</sup>             | William Jones <sup>4</sup>         | Graca Rocha <sup>6,8</sup>          |
| Ranajoy Banerji <sup>26</sup>     | Olivier Doré <sup>6,8</sup>          | Terry Jones <sup>1</sup>           | Marcel Schmittfull <sup>18</sup>    |
| R. Belen Barreiro <sup>27</sup>   | Hans K. Eriksen <sup>26</sup>        | Lloyd Knox <sup>20</sup>           | Douglas Scott <sup>50</sup>         |
| James G. Bartlett <sup>5,6</sup>  | Josquin Errard <sup>5</sup>          | Al Kogut <sup>21</sup>             | Peter Shirron <sup>21</sup>         |
| Soumen Basak <sup>28</sup>        | Tom Essinger-Hileman <sup>21</sup>   | Marcos López-Caniego <sup>42</sup> | Ian Stephens <sup>51</sup>          |
| Nick Battaglia <sup>7</sup>       | Stephen Feeney <sup>17</sup>         | Charles Lawrence <sup>6</sup>      | Brian Sutin <sup>6</sup>            |
| Jamie Bock <sup>8,6</sup>         | Jeffrey Filippini <sup>38</sup>      | Alex Lazarian <sup>43</sup>        | Maurizio Tomasi <sup>52</sup>       |
| Kimberly K. Boddy <sup>2</sup>    | Laura Fissel <sup>14</sup>           | Zack Li <sup>4</sup>               | Amy Trangsrud <sup>6</sup>          |
| Matteo Bonato <sup>29</sup>       | Raphael Flauger <sup>15</sup>        | Mathew Madhavacheril <sup>4</sup>  | Alexander van Engelen <sup>23</sup> |
| Julian Borrill <sup>9,10</sup>    | Unni Fuskeland <sup>26</sup>         | Jean-Baptiste Melin <sup>12</sup>  | Flavien Vansyngel <sup>53</sup>     |
| François Bouchet <sup>30</sup>    | Vera Gluscevic <sup>39</sup>         | Joel Meyers <sup>44</sup>          | Ingunn K. Wehus <sup>26</sup>       |
| François Boulanger <sup>31</sup>  | Krzysztof M. Gorski <sup>6</sup>     | Calum Murray <sup>5</sup>          | Qi Wen <sup>1</sup>                 |
| Blakesley Burkhardt <sup>32</sup> | Dan Green <sup>16</sup>              | Mattia Negrello <sup>45</sup>      | Siyao Xu <sup>43</sup>              |
| Jens Chluba <sup>33</sup>         | Shaul Hanany <sup>1</sup>            | Giles Novak <sup>46</sup>          | Karl Young <sup>1</sup>             |
| David Chuss <sup>11</sup>         | Brandon Hensley <sup>4</sup>         | Roger O'Brient <sup>6,8</sup>      | Andrea Zonca <sup>54</sup>          |
| Susan E. Clark <sup>18,34</sup>   | Diego Herranz <sup>27</sup>          | Christopher Paine <sup>6</sup>     |                                     |
| Joelle Cooperrider <sup>6</sup>   | J. Colin Hill <sup>17,18</sup>       | Tim Pearson <sup>8</sup>           |                                     |

## Endorsers

|                     |                        |                        |                       |
|---------------------|------------------------|------------------------|-----------------------|
| Maximilian Abitbol  | Carlo Burigana         | Kyle Helson            | Tomotake Matsumura    |
| Zeehan Ahmed        | Giovanni Cabass        | Sophie Henrot-Versillé | Darragh McCarthy      |
| David Alonso        | Robert Caldwell        | Thiem Hoang            | P. Daniel Meerburg    |
| Mustafa A. Amin     | John Carlstrom         | Kevin M. Huffenberger  | Alessandro Melchiorri |
| Adam Anderson       | Xingang Chen           | Kent Irwin             | Marius Millea         |
| James Annis         | Francis-Yan Cyr-Racine | Reijo Keskitalo        | Amber Miller          |
| Jason Austermann    | Paolo de Bernardis     | Rishi Khatri           | Joseph Mohr           |
| Carlo Baccigalupi   | Tijmen de Haan         | Chang-Goo Kim          | Lorenzo Moncelsi      |
| Darcy Barron        | C. Darren Dowell       | Theodore Kisner        | Pavel Motloch         |
| Ritoban Basu Thakur | Cora Dvorkin           | Arthur Kosowsky        | Tony Mroczkowski      |
| Elia Battistelli    | Chang Feng             | Ely Kovetz             | Suvodip Mukherjee     |
| Daniel Baumann      | Ivan Soares Ferreira   | Kerstin Kunze          | Johanna Nagy          |
| Karim Benabed       | Aurelien Fraisse       | Guilaine Lagache       | Pavel Naselsky        |
| Bradford Benson     | Andrei V. Frolov       | Daniel Lenz            | Federico Nati         |
| Paolo de Bernardis  | Nicholas Galitzki      | François Levrier       | Paolo Natoli          |
| Marco Bersanelli    | Silvia Galli           | Marilena Loverde       | Michael Niemack       |
| Federico Bianchini  | Ken Ganga              | Philip Lubin           | Elena Orlando         |
| Daniel Bilbao-Ahedo | Tuhin Ghosh            | Juan Macias-Perez      | Bruce Partridge       |
| Colin Bischoff      | Sunil Golwala          | Nazzareno Mandolesi    | Marco Peloso          |
| Sebastian Bocquet   | Riccardo Gualtieri     | Enrique Martínez       | Francesco Piacentini  |
| J. Richard Bond     | Jon E. Gudmundsson     | González               | Michel Piat           |
| Jeff Booth          | Nikhel Gupta           | Carlos Martins         | Elena Pierpaoli       |
| Sean Bryan          | Nils Halverson         | Silvia Masi            | Giampaolo Pisano      |

|                            |                      |                      |                   |
|----------------------------|----------------------|----------------------|-------------------|
| Nicolas Ponthieu           | Benjamin Saliwanchik | Eric Switzer         | Jan Vrtilek       |
| Giuseppe Puglisi           | Neelima Sehgal       | Andrea Tartari       | Benjamin Wallisch |
| Benjamin Racine            | Sarah Shandera       | Grant Teply          | Benjamin Wandelt  |
| Christian Reichardt        | Erik Shirokoff       | Peter Timbie         | Gensheng Wang     |
| Christophe Ringeval        | Anže Slosar          | Matthieu Tristram    | Scott Watson      |
| Karwan Rostem              | Tarun Souradeep      | Caterina Umiltà      | Edward J. Wollack |
| Anirban Roy                | Suzanne Staggs       | Rien van de Weygaert | Zhilei Xu         |
| Jose Alberto Rubino-Martin | George Stein         | Vincent Vennin       | Siavash Yasini    |
| Matarrese Sabino           | Radek Stompor        | Licia Verde          |                   |
| Maria Salatino             | Rashid Sunyaev       | Patricio Vielva      |                   |
|                            | Aritoki Suzuki       | Abigail Vieregg      |                   |

## Affiliations

1. University of Minnesota - Twin Cities.
2. Johns Hopkins University.
3. Carnegie Mellon University.
4. Princeton University.
5. APC, Univ Paris Diderot, CNRS/IN2P3, CEA/Irfu, Obs de Paris, Sorbonne Paris Cité, France.
6. Jet Propulsion Laboratory, California Institute of Technology.
7. Cornell University.
8. California Institute of Technology.
9. Lawrence Berkeley National Laboratory.
10. Space Sciences Laboratory, University of California, Berkeley.
11. Villanova University.
12. IRFU, CEA, Université Paris-Saclay, France.
13. University of Pennsylvania.
14. National Radio Astronomy Observatory.
15. University of California, San Diego.
16. University of California, Berkeley.
17. Center for Computational Astrophysics, Flatiron Institute.
18. Institute for Advanced Study, Princeton.
19. National Institute of Standards and Technology.
20. University of California, Davis.
21. NASA Goddard Space Flight Center.
22. University of Michigan.
23. Canadian Institute for Theoretical Astrophysics, University of Toronto, Canada.
24. Kavli Institute for the Physics and Mathematics of the Universe (WPI).
25. IRAP, Université de Toulouse, France.
26. University of Oslo, Norway.
27. Instituto de Física de Cantabria (CSIC-Universidad de Cantabria), Spain.
28. School of Physics, Indian Institute of Science Education and Research Thiruvananthapuram, India.
29. INAF-Istituto di Radioastronomia and Italian ALMA Regional Centre, Italy.
30. Institut d'Astrophysique de Paris, CNRS and Sorbonne Université, France.
31. Ecole Normale Supérieure, Paris, France.
32. Rutgers University.
33. JBCA, University of Manchester.
34. Hubble Fellow
35. INAF-Osservatorio Astronomico di Padova, Italy.
36. University of Manchester.
37. University of Southern California.
38. University of Illinois, Urbana-Champaign.
39. University of Florida.
40. Department of Astronomy & Astrophysics and Dunlap Institute, University of Toronto, Canada.
41. Columbia University.
42. European Space Astronomy Centre.
43. University of Wisconsin - Madison.
44. Southern Methodist University.
45. Cardiff University School of Physics and Astronomy.
46. Northwestern University.
47. Simon Fraser University.
48. Stanford University.
49. Kavli Institute for Particle Astrophysics and Cosmology.
50. University of British Columbia, Canada.
51. Harvard-Smithsonian Center for Astrophysics.
52. Università degli studi di Milano.
53. Institut d'Astrophysique Spatiale, CNRS, Univ. Paris-Sud, Université Paris-Saclay, France.
54. San Diego Supercomputer Center, University of California, San Diego.

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The information presented about the PICO mission concept is pre-decisional and is provided for planning and discussion purposes only.

The cost information contained in this document is of a budgetary and planning nature and is intended for informational purposes only. It does not constitute a commitment on the part of JPL and/or Caltech.

# 1 Executive Summary

The Probe of Inflation and Cosmic Origins (PICO) is an imaging polarimeter that will scan the sky for 5 years in 21 frequency bands from 21 to 799 GHz. It will produce full-sky surveys of intensity and polarization with a final combined-map noise level equivalent to 3300 *Planck* missions for the baseline required specifications, and according to our current best-estimate would perform as 6400 *Planck* missions. With these capabilities, unmatched by any other existing or proposed platform:

- PICO could determine the energy scale of inflation and give a first, direct probe of quantum gravity by searching for the signal that arises from gravitational waves sourced by inflation and parameterized by the tensor-to-scalar ratio  $r$ . The PICO requirement is to reach  $r = 5 \times 10^{-4}$  ( $5\sigma$ ), a level that is 100 times lower than current upper limits, and 5 times lower than limits forecast by any planned experiment. If the signal is not detected, PICO is the only instrument that can exclude at  $5\sigma$  models for which the characteristic scale in the potential is the Planck scale, a key threshold in inflation physics.
- The mission will measure the minimum expected sum of the neutrino masses with  $4\sigma$  confidence, rising to  $7\sigma$  if the sum is near 0.1 eV.
- The measurements will either detect or strongly constrain deviations from the standard model of particle physics by counting the number of light particle species  $N_{\text{eff}}$  in the early universe with  $\Delta N_{\text{eff}} < 0.06$  ( $2\sigma$ ).
- PICO will elucidate the processes affecting the evolution of cosmic structures by measuring the optical depth to reionization  $\tau$  with an error  $\sigma(\tau) = 0.002$ , limited only by the number of spatial modes available in the largest angular scale cosmic microwave background (CMB) polarization.
- The data will give a full sky map of the projected gravitational potential due to all structures in the Universe with the highest signal-to-noise ratio (SNR) relative to any foreseeable experiment, and it will give a catalog of 150,000 clusters extending to their earliest formation redshift. Each of these datasets will be used in combination with other data to constrain the evolution of the amplitude of linear fluctuations  $\sigma_8(z)$  with sub-percent accuracy and thus constrain dark energy and modified gravity models.
- PICO will determine the cosmological paradigm of the 2030s by reducing the allowed volume of uncertainty in an 11-dimensional  $\Lambda$ CDM parameter space by a factor of nearly a billion relative to current *Planck* constraints on only six parameters. Such exquisite scrutiny will either give strong validation of the model or require yet-to-be discovered revisions.
- With 86,000,000 independent polarization measurements across the Milky Way, 3,000 times more than *Planck* had, PICO’s data will be used to resolve long-standing questions about our Galaxy including the composition, temperature, and emissivities of Galactic dust, and the relative roles of gas turbulence and magnetic fields in the dynamics of the Galaxy and in the observed low star-formation efficiency.
- The data will constrain generic models of dark matter; enable a search for primordial magnetic fields with sufficient sensitivity to rule them out as the sole source for the largest observed galactic magnetic fields; constrain string-theory-motivated axions; and will give precise tracing of the evolution with  $z$  of thermal pressure in the universe.
- PICO’s deep, full-sky legacy maps will constrain the early phases of galaxy and cluster evolution; perform a census of cold dust in thousands of low  $z$  galaxies; make cosmic infrared background maps due to dusty star-forming galaxies; and map magnetic fields in 70 nearby galaxies.

With its broad frequency coverage, PICO is better equipped than any other current or planned instrument to separate the detected signals into their original sources of emission. This capability is important for many of the science goals, and is critical for unveiling the faintest of signals, the telltale signature of inflation, which is already known to be dominated by Galactic foregrounds. PICO’s large multiplicity of independent maps and sky surveys, and its stable thermal environment will give control of systematic uncertainties unmatched by any other platform. Mission operations are simple: PICO has a single instrument that surveys the sky with a continuously repetitive pattern. The required technologies have either already been proven by past missions, or are simple extensions of technologies now being used by sub-orbital experiments.

The science PICO will deliver addresses some of the most fundamental quests of human knowledge. Its science advances will enrich many areas of astrophysics, and will form the basis for the cosmological paradigm of the 2030s and beyond. Progress in CMB science requires a scale-up of investment. PICO is the most cost-effective way to achieve this scale-up. It has no competitor in terms of raw sensitivity, and it is the only single-platform instrument with the combination of angular resolution, frequency bands, and control of systematic effects that can deliver the compelling, timely, and broad science.

## 2 Key Science Goals and Objectives

### 2.1 Gravitational Waves and Inflation

Measurements of the CMB  $BB$  angular power spectrum are the only foreseeable way to detect inflationary gravitational waves. The strength of the signal, quantified by the tensor-to-scalar ratio  $r$ , is a direct measure of the expansion rate of the Universe during inflation, and together with the Friedmann equation, it reveals the energy scale of inflation. A detection of  $r$  “would be a watershed discovery”, a quote from the 2010 decadal panel report [1].

PICO will detect primordial gravitational waves if inflation occurred at an energy scale of at least  $5 \times 10^{15}$  GeV, or equivalently  $r = 5 \times 10^{-4}$  ( $5\sigma$ ). In a community white paper setting targets for measurements of inflationary gravitational waves in the next decade, Shandera et al. [2] quote two theoretically motivated targets: (1) rejecting  $r = 0.01$ , and (2) rejecting  $r = 0.001$ . The second threshold is motivated by the goal of rejecting all inflationary models that naturally explain the observed value of the spectral index of primordial fluctuations  $n_s$  and having a characteristic scale in the potential that is larger than the Planck scale. They write “If these thresholds are passed without a detection, most textbook models of inflation will be ruled out; and, while the possibility of an early inflationary phase would still remain viable, the data would then force a significant change in our understanding of the primordial Universe.” PICO is the only next decade experiment with the raw sensitivity to reject both targets at high confidence; see Figure 2.1. It is the only next decade experiment that can detect inflationary models that have  $r \geq 5 \times 10^{-4}$  at high confidence.

Uncertainty in the characterization of Galactic foregrounds already limits our ability to constrain  $r$ . These foregrounds are anticipated to be nearly 1000 times stronger than next-decade-

Table 1.1: Mission Parameters

|  |   |
|--|---|
| Full sky CMB polarization map depth <sup>a</sup> : |   |
| Baseline . . . . .                                 | $0.87 \mu\text{K}_{\text{CMB}}$ arcmin<br>equivalent to 3300 <i>Planck</i> missions |
| CBE <sup>b</sup> . . . . .                         | $0.61 \mu\text{K}_{\text{CMB}}$ arcmin<br>equivalent to 6400 <i>Planck</i> missions |
| Survey duration / start . . . . .                  |   |
| Orbit type . . . . .                               | Sun-Earth L2  |
| Launch mass . . . . .                              | 2147 kg   |
| Total power . . . . .                              | 1320 W  |
| Data rate . . . . .                                | 6.1 Tbits/day   |
| Cost . . . . .                                     | \$958M  |

<sup>a</sup> rms noise in  $1 \times 1$  arcmin $^2$  pixel.

<sup>b</sup> CBE = Current best estimate.

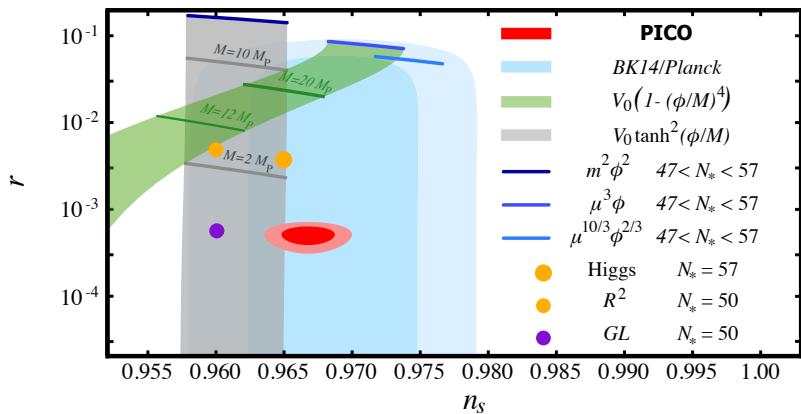


Figure 2.1: Current  $1\sigma$  and  $2\sigma$  limits on  $r$  and  $n_s$  (cyan) and forecasted constraints for a fiducial model with  $r = 0.0005$  for PICO, together with predictions for selected models of inflation. Characteristic super-Planckian scales in the potentials are marked with darker lines. GL is the Goncharev-Linde model (see text).

targeted inflationary  $B$ -mode signals at low  $\ell$  multipoles. ‘Lensing’  $B$ -modes, created by gravitational lensing of  $E$ -modes, are an additional effective foreground for the higher multipoles. With sufficiently high resolution to remove at least 73% of the lensing effects, and 21 frequency bands to account for foregrounds, no other next-decade experiment is better equipped than PICO to overcome the challenges in robustly finding the faint inflationary signal, or in rejection confusion due to foregrounds.

## 2.2 Fundamental Particles and Fields

- **Light Relics** The ‘effective number of light relic particle species’  $N_{\text{eff}}$  gives information about particle species that are predicted to have existed in the early Universe in extensions of the Standard Model. The canonical value with three neutrino families is  $N_{\text{eff}} = 3.046$ . Additional light particles contribute a change  $\Delta N_{\text{eff}}$  that is a function only of the decoupling temperature of the additional species and the spin of the particle  $g$ . PICO will provide a constraint  $\Delta N_{\text{eff}} < 0.06$  (95%) and will either detect new particle species, or constrain the lowest temperature  $T_F$  at which any vector particle (spin 1) could have fallen out of equilibrium to a factor of 400 higher than today’s constraint. No other next decade experiment will provide a tighter constraint. [add Neff white paper, and check the statement.](#)

- **Neutrino Mass** The origin, structure, and values of the neutrino masses are among the great outstanding questions about the nature of the Standard Model particles. All cosmological measurements of  $\sum m_\nu$  relate the amplitudes of the matter power spectrum and the primordial fluctuation power spectrum  $A_s$ . Both are limited by degeneracies; the former is limited by our knowledge of  $\omega_m$  and the latter by the optical depth to reionization  $\tau$ . PICO is the only instrument that will self consistently provide three of these four ingredients:  $\tau$ ,  $A_s$  from the primary CMB, and the matter power spectrum via CMB lensing. In combination with  $\omega_m$  coming from DESI and EUCLID data, PICO will give  $\sigma(\sum m_\nu) = 14$  meV, giving a  $4\sigma$  detection of the minimum sum of 58 meV. PICO will measure  $\sum m_\nu$  in two additional ways, which will give equivalent constraints.

- **Dark Matter** CMB experiments are effective in constraining dark matter candidates in the lower mass range, which is not available for terrestrial direct detection experiments [3–8]. PICO’s constraining power comes from making high SNR maps of the lensing-induced deflections of polarized photons, and cosmic-variance limited determinations of the  $TT$ ,  $TE$  and  $EE$  spectra up to  $\ell \simeq 2500$ . For a spin-independent velocity-independent contact-interaction between dark matter and protons, chosen as our fiducial model, PICO will improve upon *Planck*’s dark matter cross-section constraints by a factor of 25 over a broad range of candidate dark matter masses. If 2%

of the total dark content is made of axions, PICO’s measurement of the  $TT$ ,  $TE$  and  $EE$  spectra with additional constraints from the lensing reconstruction will detect this species at between 7 and  $13\sigma$ , depending on the mass range. **need to put these constraints in ‘next decade’ perspective**

- **Primordial Magnetic Fields** One of the long-standing puzzles in astrophysics is the origin of observed 1–10  $\mu\text{G}$  galactic magnetic fields [9]. Producing such fields through a dynamo mechanism requires a primordial seed field [10]. Moreover,  $\mu\text{G}$ -strength fields have been observed in proto-galaxies that are too young to have gone through the number of revolutions necessary for the dynamo to work [11]. A 0.1 nG primordial magnetic field (PMF), present at the time of galaxy formation, could provide the seed or even eliminate the need for the dynamo altogether [12]. A detection of PMFs with the CMB would be a major discovery because it would signal new physics beyond the Standard Model, and discriminate among different theories of the early Universe [17–19]. PICO will probe PMFs as weak as 0.1 nG ( $1\sigma$ ), a precision not attainable by any next decade experiment, and can thus conclusively rule out the purely primordial (i.e., no-dynamo driven) origin of the largest galactic magnetic fields.

- **Cosmic Birefringence** A number of well-motivated extensions of the Standard Model involve fields with parity-violating coupling [23–25, 25–28]. Their presence may cause cosmic birefringence – a rotation of the polarization of an electromagnetic wave as it propagates across cosmological distances [25, 29, 30]. PICO’s constraints on cosmic birefringence are more stringent than any other next decade experiment [? ].

## 2.3 Cosmic Structure Formation and Evolution

- **The Formation of the First Luminous Sources** A few hundred million years after the Big Bang, the neutral hydrogen gas permeating the Universe was reionized by photons emitted by the first luminous sources to have formed. The nature of these sources and the exact history of this epoch are key missing links in our understanding of structure formation. With full sky coverage, multiple frequency bands, and ample sensitivity to remove foregrounds, PICO is uniquely suited to make the low- $\ell$   $EE$ -spectrum measurements and reach cosmic-variance-limited precision with  $\sigma(\tau) = 0.002$ , settling some of these questions and significantly constraining the others. Data from PICO’s frequency bands above 400 GHz – which have better than 2 arcmin resolution – will be used to provide clean maps for higher resolution ground-based instruments that can reconstruct the patchy  $\tau$  field. No other experiment can provide these data.

- **Probing the Evolution of Structures via Gravitational Lensing and Cluster Counts** The amplitude of linear fluctuations as a function of redshift, parameterized by  $\sigma_8(z)$ , is a sensitive probe of physical processes affecting growth of structures in the Universe. CMB photons are affected by, and thus probe,  $\sigma_8(z)$  as they traverse the entire Universe. The PICO sub-percent constraints on  $\sigma_8(z)$ , obtained through measurements of gravitational lensing and independently through using cluster counts, will translate to constraints on dark energy, models of modified gravity, baryonic feedback process, and limits on the particle content of the Universe.

- **Gravitational Lensing** Matter between us and the last-scattering surface deflects the path of photons through gravitational lensing, imprinting the three-dimensional matter distribution across the volume of the Universe onto the CMB maps. The specific quantity being mapped by these data is the projected gravitational potential  $\phi$  that is lensing the photons. With SNR of more than 560, the PICO  $C_L^{\phi\phi}$  angular power spectrum is the highest of any foreseeable CMB experiment in the range  $2 \leq L \lesssim 1500$ . PICO’s  $\phi$  map will be a key ingredient in the delensing process that improves constraints on  $r$ , in extracting neutrino mass constraints, in constraining shear biases for LSST and

WFIRST [? ], and in measuring  $\sigma_8(z)$  in multiple redshift bins with sub-percent accuracy [58].

**– Cluster Counts** The distribution of galaxy clusters over redshift is one consequence of the evolution of structures and is thus a sensitive measure of  $\sigma_8(z)$ . We forecast that PICO will find  $\sim 150,000$  galaxy clusters, assuming the cosmological parameters from *Planck* and using the 70% of sky not obscured by the Milky Way. Information provided by the high frequency bands will mitigate the potential reduction in detection efficiency due to dust emission by cluster members [42]. This catalog will provide  $\sigma_8$  with sub-percent precision for  $0.5 < z < 2$ , and a neutrino mass constraint  $\sigma(\Sigma m_V) = 14$  meV that is independent from the one coming from the CMB lensing measurements. A significant fraction of the PICO-detected clusters will also be detected by eROSITA, giving an exceptional catalog of multi-wavelength observations for detailed studies of cluster astrophysics.

**• Constraining Feedback Processes through the Sunyaev–Zeldovich Effect** About 6% of CMB photons are Thomson-scattered by free electrons in the intergalactic medium (IGM) and intercluster medium (ICM), and a fraction of these are responsible for the thermal and kinetic Sunyaev–Zeldovich effects (tSZ and kSZ) [43, 44]. The amplitude of the tSZ is proportional to the integrated electron pressure along the line of sight, and it thus contains information about the thermodynamic properties of the IGM and ICM, which are highly sensitive to astrophysical feedback. With its low noise and broad frequency coverage, which is essential for separating out other signals, PICO will yield a definitive tSZ map over the full sky with a total SNR of 1270 for the CBE and 10% lower for the baseline configurations (Fig. ??). **what is unique? Full sky? frequencies? resolution?** The 150,000 clusters forecast to be detected by PICO will be found in this map. Considering the LSST gold weak-lensing sample, with a source density of 26 galaxies/arcmin<sup>2</sup> covering 40% of the sky, we forecast a detection of the tSZ–weak-lensing cross-correlation with SNR = 3000. Cross-correlations with the galaxies themselves will be measured at even higher SNR. At this immense significance, the signal will be broken down into dozens of tomographic redshift bins, precisely tracing the evolution of thermal pressure over cosmic time.

## 2.4 Testing $\Lambda$ CDM

PICO will set the cosmological paradigm for the 2030’s and beyond by measuring the six parameter  $\Lambda$ CDM with 100,000 more constraining power compared to *Planck*; see Figure 2.3 (the improvement between WMAP and *Planck* was by 100). For an 11-parameter set that include  $r$ ,  $N_{\text{eff}}$ , and  $\Sigma(m_V)$ , the improvement is by a factor of  $0.5 \times 10^9$ . These improvements will test  $\Lambda$ CDM so stringently that it is hard to imagine it surviving such a scrutiny if it is not fundamentally correct. If tensions deepen to become discrepancies, it would be even more exciting if a new cosmological model emerged.

## 2.5 Galactic Structure and Star Formation

PICO will produce 21 polarization maps of Galactic emission, all much deeper than *Planck*’s seven maps. At 799 GHz PICO will have five times finer resolution than *Planck*; see Fig. ??). Such a data set can only be obtained by PICO. These data will complement a rich array of other polarization observations forthcoming in the next decade, including stellar polarization surveys to be combined with Gaia astrometry, and Faraday rotation measurements from observations at radio wavelengths with the Square Kilometer Array (SKA) and its precursors.

- Test models of the composition of interstellar dust** Less than 1  $\mu\text{m}$  in size, dust grains

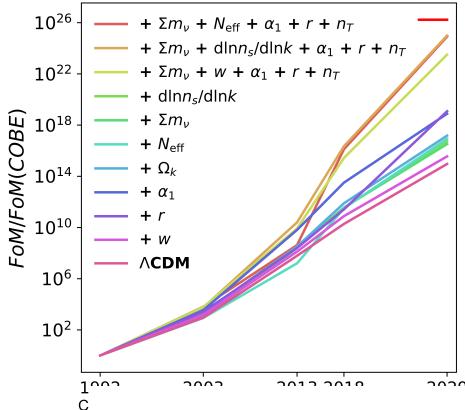


Figure 2.2: The increase in cosmological parameter constraining power using only CMB data since *COBE*. The FoM is the inverse of the uncertainty volume in parameter space. For an 11-parameter set that includes  $N_{\text{eff}}$  (red increasing line) PICO will improve the FoM by a factor of  $0.5 \times 10^9$  relative to *Planck*. It will extract nearly the same information as that attainable by a mission with twice higher resolution and nine times lower noise (top right red horizontal bar), that is, PICO’s performance on cosmological parameters is equivalent to that of a ‘CMB flagship-scale mission’. The constituents of the 11-parameter set are given in Hanany et al. [58].

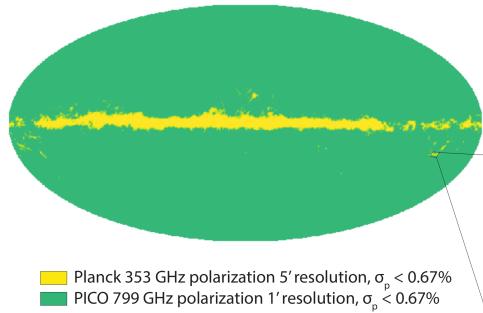


Figure 2.3: Caption will be here

are intermediate in the evolution from atoms and molecules to large solid bodies such as comets, asteroids, and planets. Through vastly improved spectral characterization of Galactic polarization, the PICO data will validate or reject state-of-the-art dust models [e.g. 54, 55], test for the presence of additional dust grain species with distinct polarization signatures, such as magnetic nanoparticles [56], and will be used as an input for the foreground separation necessary to extract cosmological  $E$ - and  $B$ -mode science.

• **Determine how magnetic fields affect molecular cloud and star formation** Stars are formed through interactions between gravitational and magnetic fields, turbulence, and gas over more than four orders of magnitude of spatial scales, which span the diffuse ISM (kpc scale), molecular clouds (10 pc), and molecular cloud cores (0.1 pc). However, the role magnetic fields play in the large-scale structure of the diffuse interstellar medium (ISM) and in the observed low star-formation efficiency has been elusive, owing to the dearth of data. With  $1.1'$  resolution PICO will expand the number of independent magnetic field measurements across the sky by a factor of 2900, from *Planck*’s 30,000 to 86,000,000 (Fig. ??). The data will robustly characterize turbulent properties like the Alfvén Mach number across a previously unexplored regime of parameter space.

## 2.6 Legacy Surveys

PICO will generate a rich and unique catalog of hundreds of thousands of new sources serving astrophysicists across a broad range of interests including in galaxy and cluster evolution, correlations of cold galactic dust with galaxy properties, the physics of jets in active galactic nuclei, and the properties of the cosmic infrared background. This information will be embedded in catalogs including 50,000 proto-clusters extending to  $z \simeq 4.5$ , 4,500 strongly lensed galaxies extending to  $z \simeq 5$ , 30,000 galaxies with  $z \leq 0.1$ , polarization data for few thousand radio sources and dust galaxies, and the deepest maps of the CIB with as high a resolution as  $1'$ .

## 2.7 Signal Separation

### 2.7.1 The Signal Separation Challenge

Galactic emission dominates the sky's polarized intensity on large angular scales ( $\ell \lesssim 10$ ), it dominates the cosmological  $B$ -modes signals for  $\ell \lesssim 150$  for all allowed levels of  $r$ , and it is expected to be significant even at  $\ell \simeq 1000$ , posing challenge for reconstructing the  $B$ -mode signal from lensing. This is illustrated in Figs. ?? and 2.4, which show Galactic emission power spectra calculated for the cleanest – that is, the least Galactic-emission-contaminated – 60% of the sky. But even in small patches of the sky, far from the Galactic plane and with the least foreground contamination, Galactic emission levels are substantial relative to an inflationary signal of  $r \sim 0.01$  [57]. Separating the cosmological and Galactic emission signals is one of two primary challenges facing any next-decade experiment attempting to reach these levels of constraints on  $r$  (the second is control of systematic uncertainties).

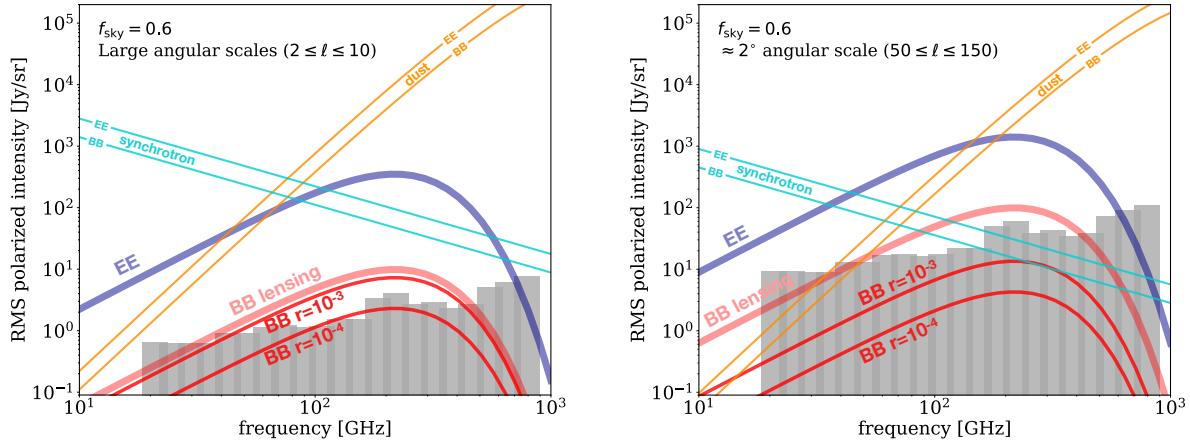


Figure 2.4: Polarization  $BB$  spectra of Galactic synchrotron and dust, compared to CMB polarization  $EE$  and  $BB$  spectra of different origins for two values of  $r$  and for two ranges of angular scales: large-scale,  $\ell \leq 10$ , corresponding to the reionization peak (left panel); and intermediate scales  $50 \leq \ell \leq 150$ , corresponding to the recombination peak (right panel). Data from *Planck* indicate that for Galactic emission the level of the  $E$ -mode is approximately twice that of  $B$  [57]. The PICO baseline noise (grey bands) is low compared to the Galactic emission components, and thus they will be measured with high SNR in many frequency bands.

To investigate the efficacy of PICO in addressing the foreground-separation challenge, we used both an analytic forecast and map-domain simulations; a complete description is given in Hanany et al. [58]. Here we present results only from the more conservative map-domain analysis. In this analysis we simulate sky maps that are constrained by available data, but otherwise have a mixture of foreground properties; we ‘observe’ these maps just like a realistic experiment would do, and then apply foreground separation techniques to separate the Galactic and CMB emissions. Our results indicate that:

- the combination of PICO’s sensitivity and broad frequency coverage are effective in foreground removal and that PICO will reach the requirement of  $r = 5 \times 10^{-4}$  ( $5\sigma$ ); see Figure 2.5;
- the high frequency bands, above 400 GHz, may be essential for proper subtraction of foregrounds; see Figure 2.6.

Among all next decade experiments, PICO is best suited to handle and model the foregrounds because it has more frequency bands than any other experiment, because it has the lowest noise, and because it has full sky coverage, giving access to several independent small regions that are

very low in dust emission. This is a distinct advantage relative to CMB-S4, which targets a single patch of the sky, and relative to liteBIRD, which does not have frequency bands above 400 GHz.

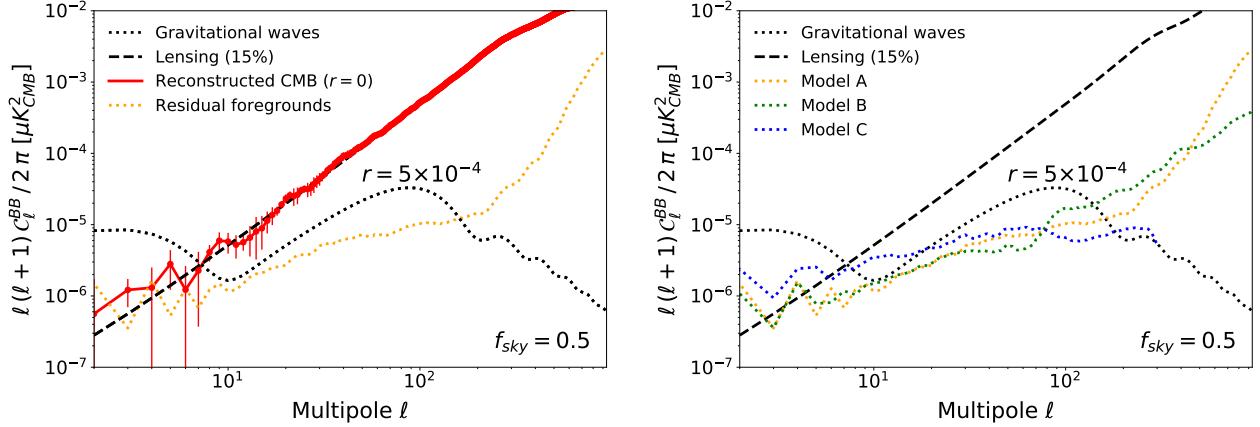


Figure 2.5: Angular power spectra of  $BB$  due to the CMB and of residual foregrounds after an end-to-end map-based foreground-separation exercise. The PICO low noise levels and breadth in frequency coverage enable separation of model A foregrounds such that the residual foreground spectrum (left, yellow dotted) is a factor of ten (four) below a  $BB$  inflationary signal with  $r = 5 \times 10^{-4}$  (black dotted) at  $\ell = 5(80)$ . Within errors, the recovered CMB (red) matches the input CMB, which consists of only lensing  $BB$  (dashed black), over all angular scales  $\ell \gtrsim 6$ . The results for model B are similar (right, green dots), while model C has somewhat higher residuals at low  $\ell$ . In this exercise we used 50% of the sky. Lower foreground residual levels are obtainable with smaller, cleaner patches of  $\sim 5\%$  of sky, which would reduce the residual foregrounds at  $\ell \simeq 80$ .

## 2.8 Systematic Uncertainties

Properly modeling, engineering for, and controlling systematic effects are key for the success of any experimental endeavor striving to achieve  $\sigma(r) \lesssim 1 \times 10^{-3}$ . Based on extensive community experience with both hardware and analysis of data we make the following points.

- Relative to other platforms, a space-based mission provides the most thermally stable platform,

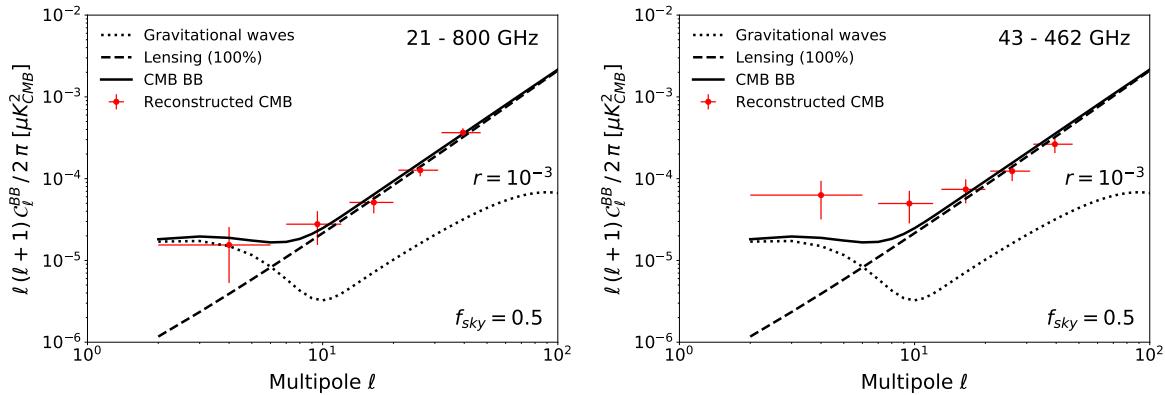


Figure 2.6: **Left:** Foreground separation with all of PICO’s 21 frequency bands recovers the input CMB  $BB$  power spectrum (solid black) without bias (red). The input CMB spectrum has a contribution from lensing (dashed) and an inflationary signal with  $r = 0.001$  (dotted). This exercise uses a parametric approach [59] with foregrounds varying on 4° pixels, and using 50% sky fraction. **Right:** Running the same foreground separation algorithm on the same sky but using only PICO’s bands between 43 and 462 GHz produces an output spectrum (red) that is biased at low multipoles relative to the input. With real data, such a bias would be erroneously interpreted as a higher value of  $r$ .

and thus the prerequisite for improved control of systematic effects. PICO’s orbit at L2 is among the most thermally stable of possible orbits.

- PICO’s sky scan pattern gives strong data redundancy, which enables numerous cross-checks. Each of the 12,996 detectors makes independent maps of the  $I$ ,  $Q$ , and  $U$  Stokes parameters enabling many comparisons within and across frequency bands, within and across sections of the focal plane, and within and across bolometers that have either the same or different polarization sensitivities. Half the sky is scanned every two weeks, and the entire sky is scanned in 6 months. Thus combinations of maps constructed at different times during of the mission will be differenced to search for residual time-dependent systematic effects.
- The scan pattern gives almost continuous scans of planets and large amplitude ( $\geq 4$  mK) CMB dipole signals [60]. These features result in continuous, high SNR calibration and antenna-pattern characterization. In comparison, *Planck* observed each of the planets with only a 6 month cadence and had nearly 100 days/year during which the dipole calibration signals were below 4 mK, at times dipping below 1 mK.
- We showed that two of the highest priority systematic effects can be controlled to levels that are small compared to requirements. More analysis and planning is required to address systematic uncertainties arising from the far-sidelobe response of the telescope.

We direct the reader to the mission study report for more details on our work on systematic effects for PICO [58].

### 3 Technical Overview

PICO meets all of its science-derived instrument requirements (§ ??) with a single instrument: an imaging polarimeter with 21 logarithmically spaced frequency bands centered between 21 and 799 GHz (Table ??). The instrument has a two-reflector Dragone-style telescope (§ 3.1 and Fig. 3.1). The focal plane is populated by 12,996 transition-edge-sensor (TES) bolometers (§ 3.2) and read out using a time-domain multiplexing scheme (§ 3.3). PICO employs a single science observing mode: fixed rate imaging while scanning the sky (§ 4.1.2).

The instrument is configured inside the shadow of a V-groove assembly that thermally and optically shields it from the Sun (Fig. 3.1 and § 3.4.3). The Sun shadow cone depicted in Fig. 3.1 is  $29^\circ$ . The angle to the Sun during the survey,  $\alpha = 26^\circ$  (§ 4.1.2 and Fig. 4.2), is supplemented with a margin of  $3^\circ$  to account for the radius of the Sun ( $0^\circ 25$ ), pointing control error, design margin, and alignment tolerances.

The V-groove assembly is attached to the bipod struts that support the instrument structural ring. The ring supports the primary reflector and telescope box. The telescope box contains the actively cooled components (§ 3.4.1, § 3.4.2), including the secondary reflector, the focal plane and sub-kelvin refrigerator structures. Just inside the box, a thermal liner serves as a cold optical baffle and aperture stop. Instrument integration and test are described in § 3.5.

During the survey, the instrument is spun at 1 rpm and the spin axis is made to precess about the anti-Sun direction (§ 4.1.2). Spacecraft control is simplified by mounting the instrument on a spinning spacecraft module, while a larger non-spinning module houses most spacecraft subsystems (§ 4.3). Instrument elements that act as heat sources are accommodated on the spinning module of the spacecraft. Only power and digital data lines cross between the spinning and non-spinning modules. A functional block diagram of the instrument is shown in Figure 3.2.

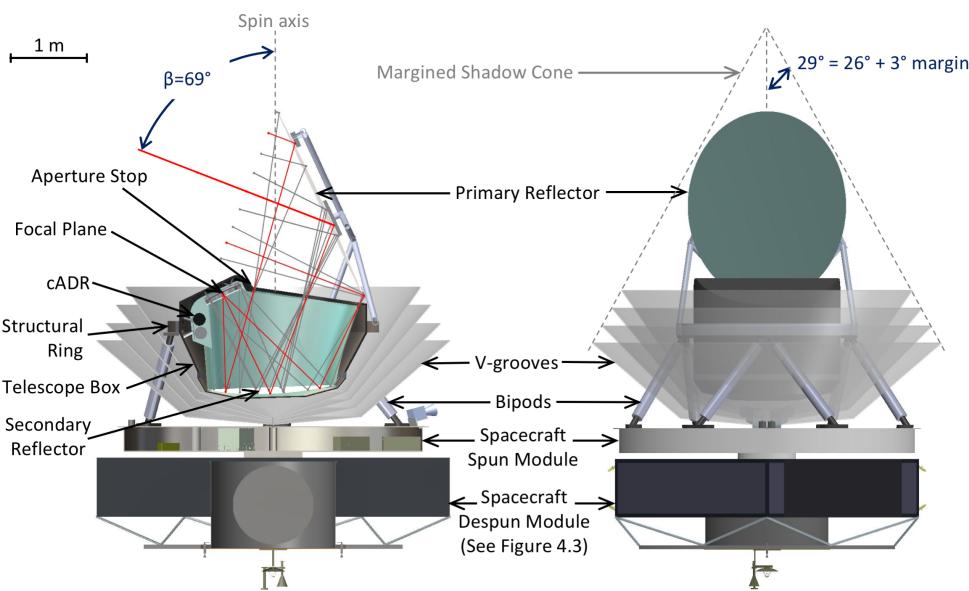


Figure 3.1: PICO overall configuration in side view and cross section (left), and front view with V-Groove assembly shown semi-transparent (right). The mission consists of a single science instrument mounted on a structural ring. The ring is supported by bipods on a stage spinning at constant speed relative to a despun module. Figure 3.2 shows the functions hosted by each of the modules.

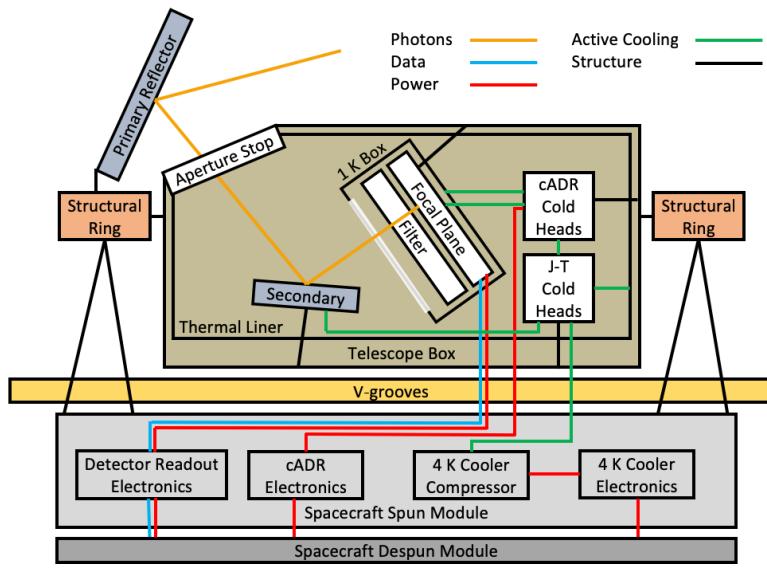


Figure 3.2: PICO instrument block diagram. Active coolers provide cooling to the 100 mK focal plane, the surrounding 1 K box, the 4.5 K secondary reflector, and the 4.5 K thermal liner that acts as a cold aperture stop. V-grooves provide passive cooling. The instrument, V-grooves, and spacecraft spun module spin together at a rate of 1 RPM. The spacecraft spun module hosts the 4 K cooler compressor and drive electronics, the sub-K cooler drive electronics, and the detector warm readout electronics. Only power and digital data lines cross to the spacecraft despun module, which hosts the spacecraft power, telemetry, attitude control, and communication systems (§ 4.3).

### 3.1 Telescope

The PICO telescope design is driven by a combination of science requirements and physical volume limits. The science requirements are: a large diffraction-limited field of view (DLFOV) sufficient to support approximately  $10^4$  detectors; arcminute resolution at 800 GHz; low spurious polarization; and low sidelobe response. All requirements are met with PICO's 1.4 m aperture modified open-Dragone design. There are no moving parts in the PICO optical system.

The PICO optical design was selected following a trade study examining cross-Dragone, Gre-

gorian Dragone, and open-Dragone designs [63]. The open-Dragone and crossed-Dragone systems offer more diffraction-limited focal-plane area than the Gregorian Dragone one [64] and are able to support enough detectors to provide the required sensitivity. The open-Dragone design does not require the more massive and voluminous baffles that the cross-Dragone does, and hence can satisfy the aperture size requirement within the shadow cone.

PICO’s initial open-Dragone design [65, 66] has been modified with the addition of an aperture stop and adding corrections to the primary and secondary reflectors to enlarge the DLFOV. The detailed geometric parameterization of the PICO optical design is described by Young et al. [63]. The primary reflector ( $270\text{ cm} \times 205\text{ cm}$ ) is passively cooled and the secondary reflector ( $160\text{ cm} \times 158\text{ cm}$ ) is actively cooled. The highest frequency (900 GHz) sets the surface accuracy requirement of the reflectors at  $\lambda/14 = 24\mu\text{m}$ . The focal ratio is 1.42. The slightly concave focal surface, which has a radius of curvature of 4.55 m, is telecentric to within  $0.12^\circ$  across the entire FOV.

An actively cooled circular aperture stop between the primary and secondary reflectors reduces detector noise and shields the focal plane from stray radiation. Stray-light analysis of the PICO open-Dragone design using GRASP confirms that the focal plane is protected from direct view of the sky, and that spillover past the primary is suppressed by 80 dB relative to the main lobe for both co-pol and cross-pol beams. Detailed baffle design will be performed during mission formulation.

## 3.2 Focal Plane

PICO’s focal plane is populated by an array of TES bolometers operating in 21 frequency bands, each with 25% fractional bandwidth, and band centers ranging from 21 to 799 GHz. The layout of the PICO focal plane is shown in Fig. 3.3 and detailed in Table 3.1.

Bolometers operating in the mm/sub-mm wave band are photon-noise limited. Therefore, increase in sensitivity is achieved through an increase in detector count. The PICO focal plane has 12,996 detectors, 175 times the number flown aboard *Planck*, thereby providing a breakthrough increase in sensitivity

with a comparably sized telescope. This breakthrough is enabled by development and demonstration in suborbital projects, which now commonly operate arrays of  $10^3$ – $10^4$  detectors (§ 5). Further technology maturation required for PICO is described in Section § 5.

### 3.2.1 21–462 GHz Bands

Several optical-coupling technologies have matured over the past ten years to efficiently use focal-plane area: horns with ortho-mode transducers (OMTs) [67]; lithographed antenna arrays [68]; and sinuous antennas under lenslets [69]. Horn-coupling and sinuous antenna/lenslet-coupling deliver quantum efficiency  $> 70\%$  over more than an octave of bandwidth, which have been partitioned into two or three colors per pixel. Only single-color pixels have been demonstrated to date with antenna-arrays, but this coupling enables smaller pixels and therefore they can be more densely packed.

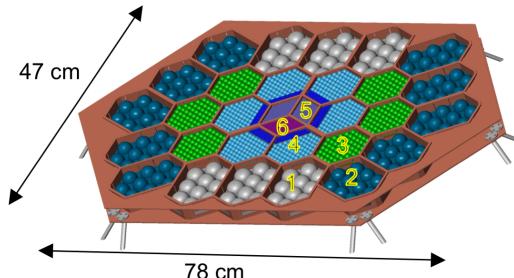


Figure 3.3: PICO focal plane. Detectors are fabricated on six types of tiles (shown numbered and colored as in Table 3.1). The wafers are located on the focal plane such that higher frequency bands, which require better optical performance, are placed nearer to the center. All detectors are within the diffraction-limited performance for their respective frequency bands.

The PICO baseline focal plane employs three-color sinuous antenna/lenslet pixels [70] for the 21–462 GHz bands. Niobium microstrips mediate the signals between the antenna and detectors, and partition the wide continuous bandwidth into three narrow channels using integrated, on-wafer, micro-machined filter circuits [71]. Six transition edge sensor bolometers per pixel detect the radiation in two orthogonal polarization states.

### 3.2.2 555–799 GHz Bands

PICO’s highest three frequency channels are beyond the niobium superconducting band-gap, rendering on-wafer, microstrip filters a poor solution for defining the optical passband. For these bands we use feedhorns to couple the radiation to two single-color polarization-sensitive TES bolometers. The waveguide cut-off defines the lower edge of the band, and quasi-optical metal-mesh filters define the upper edge. Numerous experiments have successfully used similar approaches [72–74].

### 3.2.3 Polarimetry

Polarimetry is achieved by differencing the signals from pairs of two co-pointed bolometers within a pixel that are sensitive to two orthogonal polarization states. Half the pixels in the focal plane are sensitive to the  $Q$  and half to the  $U$  Stokes parameters of the incident radiation. Two layouts for the distribution of the  $Q$  and  $U$  pixels on the focal plane have been investigated [75]; both would satisfy mission requirements. Stokes  $I$  is obtained from the sum of the signals of orthogonal detectors.

### 3.2.4 Sensitivity

PICO’s Current Best Estimate (CBE) sensitivity meets the requirements of the baseline mission with  $> 40\%$  margin (Table 3.2).

We developed an end-to-end noise model of the PICO instrument to predict mission sensitivity and provide a metric by which to evaluate mission design trades. The model includes four noise sources per bolometer: photon, phonon, Johnson, and readout (from both cold and warm readout electronics). To validate our calculations, we compared two independent software packages that have been validated with several operating CMB instruments. The calculations agreed within 1% both for individual noise terms and for overall mission noise. A detailed description of the PICO noise model and its inputs is available in Young et al. [63]; small differences between that publication and Table 3.2 are due to refinements of the primary mirror and stop temperatures.

Laboratory experiments have demonstrated that TES bolometers can be made background-limited in the low loading environment they would experience at L2 [76]. For PICO, the primary contributor to noise is the optical load. The sources of optical load are the CMB, reflectors, aperture stop, and low-pass filters. The CMB and stop account for at least 50% of the optical load at all frequencies up to and including 555 GHz. At higher bands emission from the primary mirror

Table 3.1: PICO makes efficient use of the focal area with multichroic pixels (three bands per pixel, § 3.2.1). The sampling rate is based on the smallest beam (Table 3.2), with 3 samples per FWHM at a scan speed ( $360^\circ/\text{min}$ )  $\sin(\beta = 69^\circ) = 336^\circ/\text{min}$ . Scaling from suborbital experience, we anticipate that TES bolometers can support these sampling rates with  $\sim 4\times$  margin.

| Tile type | $N_{\text{tile}}$ | Pixels/tile | Pixel type | Band centers [GHz] | Sampling rate [Hz] |
|-----------|-------------------|-------------|------------|--------------------|--------------------|
| 1         | 6                 | 10          | A          | 21, 30, 43         | 45                 |
| 2         | 10                | 10          | B          | 25, 36, 52         | 55                 |
| 3         | 6                 | 61          | C          | 62, 90, 129        | 136                |
| 4         | 6                 | 85          | D          | 75, 108, 155       | 163                |
|           |                   | 80          | E          | 186, 268, 385      | 403                |
| 5         | 2                 | 450         | F          | 223, 321, 462      | 480                |
| 6         | 1                 | 220         | G          | 555                | 917                |
|           |                   | 200         | H          | 666                |                    |
|           |                   | 180         | I          | 799                |                    |

Table 3.2: PICO has 21 partially overlapping frequency bands with band centers ( $v_c$ ) from 21 GHz to 799 GHz and each with bandwidth  $\Delta v/v_c = 25\%$ . The beams are single mode, with FWHM sizes of  $6.2 \times (155 \text{ GHz}/v_c)$ . The CBE per-bolometer sensitivity is photon-noise limited (§ 3.2.4). The total number of bolometers for each band is equal to (number of tiles)  $\times$  (pixels per tile)  $\times$  (2 polarizations per pixel), from Table 3.1. Array sensitivity assumes 90% detector operability. The map depth assumes 5 yr of full sky survey at 95% survey efficiency, except the 25 and 30 GHz frequency bands, which are conservatively excluded during 4 hr/day Ka-band (26 GHz) telecom periods (§ 4.2).

| Band center | Beam FWHM | CBE bolo NET                                | $N_{\text{bolo}}$ | CBE array NET                               | Baseline array NET                          | Baseline polarization map depth            |                       |
|-------------|-----------|---|-------------------|---|---|--|-----------------------|
| [GHz]       | [arcmin]  | $[\mu\text{K}_{\text{CMB}} \text{s}^{1/2}]$ |                   | $[\mu\text{K}_{\text{CMB}} \text{s}^{1/2}]$ | $[\mu\text{K}_{\text{CMB}} \text{s}^{1/2}]$ | $[\mu\text{K}_{\text{CMB}} \text{arcmin}]$ | $[\text{Jy sr}^{-1}]$ |
| 21.....     | 38.4      | 112   | 120               | 12.0  | 17.0  | 23.9                                       | 8.3                   |
| 25.....     | 32.0      | 103   | 200               | 8.4   | 11.9  | 18.4                                       | 10.9                  |
| 30.....     | 28.3      | 59.4  | 120               | 5.7   | 8.0   | 12.4                                       | 11.8                  |
| 36.....     | 23.6      | 54.4  | 200               | 4.0   | 5.7   | 7.9  | 12.9                  |
| 43.....     | 22.2      | 41.7  | 120               | 4.0   | 5.6   | 7.9  | 19.5                  |
| 52.....     | 18.4      | 38.4  | 200               | 2.8   | 4.0   | 5.7  | 23.8                  |
| 62.....     | 12.8      | 69.2  | 732               | 2.7   | 3.8   | 5.4  | 45.4                  |
| 75.....     | 10.7      | 65.4  | 1020              | 2.1   | 3.0   | 4.2  | 58.3                  |
| 90.....     | 9.5       | 37.7  | 732               | 1.4   | 2.0   | 2.8  | 59.3                  |
| 108.....    | 7.9       | 36.2  | 1020              | 1.1   | 1.6   | 2.3  | 77.3                  |
| 129.....    | 7.4       | 27.8  | 732               | 1.1   | 1.5   | 2.1  | 96.0                  |
| 155.....    | 6.2       | 27.5  | 1020              | 0.9   | 1.3   | 1.8  | 119                   |
| 186.....    | 4.3       | 70.8  | 960               | 2.0   | 2.8   | 4.0  | 433                   |
| 223.....    | 3.6       | 84.2  | 900               | 2.3   | 3.3   | 4.5  | 604                   |
| 268.....    | 3.2       | 54.8  | 960               | 1.5   | 2.2   | 3.1  | 433                   |
| 321.....    | 2.6       | 77.6  | 900               | 2.1   | 3.0   | 4.2  | 578                   |
| 385.....    | 2.5       | 69.1  | 960               | 2.3   | 3.2   | 4.5  | 429                   |
| 462.....    | 2.1       | 133   | 900               | 4.5   | 6.4   | 9.1  | 551                   |
| 555.....    | 1.5       | 658   | 440               | 23.0  | 32.5  | 45.8                                       | 1580                  |
| 666.....    | 1.3       | 2210  | 400               | 89.0  | 126   | 177  | 2080                  |
| 799.....    | 1.1       | 10400                                       | 360               | 526   | 744   | 1050                                       | 2880                  |
| Total ..... |           | 12996                                       |                   | 0.43  | 0.61  | 0.87                                       |                       |

dominates.

The sensitivity model assumes white noise at all frequencies. Sub-orbital submillimeter experiments have demonstrated TES detectors that are stable to at least as low as 20 mHz [77], meeting the requirements for PICO’s scan strategy (§ 4.1.2).

### 3.3 Detector Readout

Suborbital experiment teams over the past ten years have chosen to use voltage-biased TESs because their current readout scheme lends itself to superconducting quantum interface device (SQUID)-based multiplexing. Multiplexing reduces the number of wires to the cryogenic stages and thus the total thermal load that the cryocoolers must dissipate. This approach also simplifies the instrument design.

In the multiplexing circuitry, SQUIDs function as low-noise amplifiers and cryogenic switches. The current baseline for PICO is to use a time-domain multiplexer (TDM), which assigns each detector’s address in a square matrix of simultaneously read columns, and sequentially cycles through each row of the array [78]. The PICO baseline architecture uses a matrix of 128 rows and

102 columns. The thermal loading on the cold stages from the wire harnesses is subdominant to conductive loading through the mechanical support structures.

Because SQUIDs are sensitive magnetometers, suborbital experiments have developed techniques to shield them from Earth’s magnetic field using highly permeable or superconducting materials [79]. Total suppression factors better than  $10^7$  have been demonstrated for dynamic magnetic fields [80]. PICO will use these demonstrated techniques to shield SQUID readout chips from the ambient magnetic environment, which is 20,000 times smaller than near Earth, as well as from fields generated by on-board components, including the 0.1 K cooler (§ 3.4.1). This cooler is delivered with its own magnetic shielding, which reduces the field at the distance of the SQUIDs to less than 0.1 G, which is less than Earth’s field experienced by SQUIDs aboard suborbital experiments. SQUIDs are also sensitive to radio-frequency interference (RFI). Several suborbital experiments have demonstrated RFI shielding using aluminized mylar wrapped at cryogenic stages to form a Faraday cage around the SQUIDs [81–83]. Cable shielding extends the Faraday cage to the detector warm readout electronics.

Redundant warm electronics boxes perform detector readout and instrument housekeeping using commercially available radiation-hardened analog-to-digital converters, requiring 75 W total. The readout electronics compress the data before delivering them to the spacecraft, requiring an additional 15 W. PICO detectors produce a total of 6.1 Tbits/day assuming 16 bits/sample, sampling rates from Table 3.1, and bolometer counts from Table 3.2. *Planck* HFI had a typical  $4.7 \times$  compression in flight, with information loss increasing noise by only about 10% [84, 85]. Suborbital work has demonstrated  $6.2 \times$  lossless compression [86]. PICO assumes  $4 \times$  lossless compression.

## 3.4 Thermal

Like the *Planck*-HFI instrument, PICO’s focal plane is maintained at 0.1 K to ensure low detector noise while implementing readily available technology (§ 3.4.1). To minimize detector noise due to instrument thermal radiation, the aperture stop and reflectors are cooled using both active and radiative cooling (§ 3.4.2, § 3.4.3, Fig. 3.2). All thermal requirements are met with robust margins (Table 3.3).

### 3.4.1 cADR Sub-Kelvin Cooling

A multi-stage continuous adiabatic demagnetization refrigerator (cADR) maintains the PICO focal plane at 0.1 K and the surrounding enclosure, filter, and readout components at 1 K. The cADR employs three refrigerant assemblies operating sequentially to absorb heat from the focal plane at 0.1 K and reject it to 1 K. Two additional assemblies, also operating sequentially, absorb this rejected heat at 1 K, cool other components to 1 K, and reject heat at 4.5 K. This configuration provides continuous cooling with small temperature variations at both the 0.1 K and 1 K. Heat straps connect the two cADR cold sinks to multiple points on the focal-plane assembly, which has high thermal conductance paths built in, to provide spatial temperature uniformity and stability during operation. The detector arrays are thermally sunk to the mounting frame. Heat loads in the range of  $30 \mu\text{W}$  at 0.1 K and  $1 \text{ mW}$  at 1 K (time-average) are within the capabilities of current cADRs developed by GSFC (§ 7.1) [90, 91]. The PICO sub-kelvin heat loads are estimated at less than half of this capability (Table 3.3).

### 3.4.2 The 4.5 K Cooler

Table 3.3: Projected cooler heat lift capabilities offer more than 100 % heat lift margin, complying with cooler technology best practices [87].

| Component                                  | Temperature [K] |     | Active heat lift [mW]           |                            |                             |
|--|-----------------|-----|---------------------------------|----------------------------|-----------------------------|
|  | Required        | CBE | Required per model <sup>a</sup> | Capability today           | Projected capability        |
| Primary reflector . . . . .                | < 40            | 17  | N/A (radiatively cooled)        |                            |                             |
| Secondary reflector . . . . .              | < 8             | 4.5 |                                 |                            |                             |
| Aperture stop . . . . .                    | 4.5             | 4.5 | 42 at 4.5 K                     | > 55 at 6.2 K <sup>b</sup> | > 100 at 4.5 K <sup>c</sup> |
| cADR heat rejection <sup>d</sup> . . . . . | 4.5             | 4.5 |                                 |                            |                             |
| Focal plane enclosure and filter . . . . . | 1.0             | 1.0 | 0.36                            | 1.0                        | N/A <sup>e</sup>            |
| Focal plane . . . . .                      | 0.1             | 0.1 | $5.7 \times 10^{-3}$            | $32 \times 10^{-3}$        | N/A <sup>e</sup>            |

<sup>a</sup> The required loads were calculated using Thermal Desktop. Reference [88] was used to estimate the thermal conductive loads through mechanical supports. In addition to the listed components, the total 4.5 K heat load includes the intercept on the focal plane mechanical supports. <sup>b</sup> Reference [89]. <sup>c</sup> Both NGAS and Ball project > 100 mW

lift capability at 4.5 K using higher compression-ratio compressors currently in development (§ 3.4.2 and Fig. 3.4).

<sup>d</sup> The cADR lift capability at 1 K and 0.1 K is from a GSFC quote.

<sup>e</sup> Capability today already exceeds requirement.

A cryocooler system similar to that used on JWST to cool the MIRI detectors [92, 93] removes the heat rejected from the cADR and cools the aperture stop and secondary reflector to 4.5 K. Both NGAS (which provided the MIRI coolers) and Ball Aerospace have developed such coolers under the NASA-sponsored Advanced Cryocooler Technology Development Program [94]. NGAS and Ball use slightly different but functionally-equivalent hardware approaches. A 3-stage precooler provides 16 K precooling to a separate circulated-gas loop. The circulated-gas loop utilizes Joule–Thomson (J-T) expansion, further cooling the gas to 4.5 K. The J-T expansion point is located close to the cADR heat rejection point and provides to it the lowest temperature. Subsequently, the gas flow intercepts heat conducted to the focal-plane enclosure, then cools the aperture stop and the secondary reflector before returning to the circulation compressor.

NGAS and Ball are actively working on increasing the flow rate and compression ratio of the J-T compressor, which should result in higher system efficiency and greater heat-lift relative to the current MIRI cooler. NGAS uses  $^4\text{He}$  as the circulating gas, as was used for MIRI. Ball uses a somewhat larger compressor and  $^3\text{He}$  as the circulating gas. Both employ re-optimized heat exchangers. The NGAS project has completed PDR-level development, and is expected to reach CDR well before PICO begins Phase-A. The projected performance of this cooler is shown in Fig. 3.4; it gives 100 mW at 250 W input power, which is more than 100 % heat lift margin relative to PICO’s requirements (Table 3.3). For PICO we have assumed an input power of

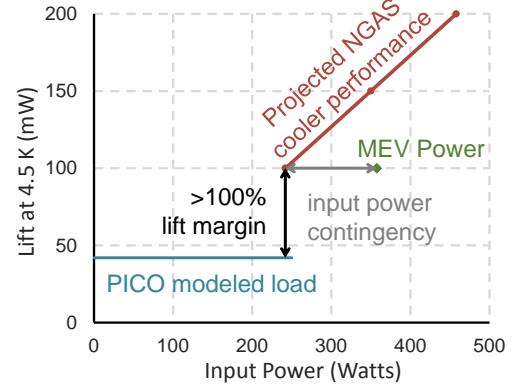


Figure 3.4: Projected performance of the NGAS cooler using a multi-stage compressor and  $^4\text{He}$  circulating gas [92] meets PICO’s requirements with > 100 % margin. PICO requires heat lift of 42 mW at 4.5 K (Table 3.3). With 250 W of input power the NGAS cooler is projected to provide 100 mW of heat lift. We conservatively specify a maximum expected value (MEV) of 350 W as the compressor’s input power, giving 100 W of additional input power contingency.

350 W.

The entire precooler assembly and the J-T circulator compressor are located on the warm space-craft spun module (Fig. 3.2). All waste heat rejected by the cooler compressors and drive electronics is transferred to the spacecraft heat-rejection system. Unlike JWST, the PICO cooler does not require deployment of the remote cold head.

### 3.4.3 Radiative Cooling

An assembly of four nested V-groove radiators, acting as radiation shields, provides passive cooling (Fig. 3.1). This is standard, 30-years old technology (§ 7.1). The outermost shield shadows the interior ones from the Sun. The V-grooves radiate to space, each reaching successively cooler temperatures. The assembly provides a cold radiative environment to the primary reflector, structural ring, and telescope box. As a consequence radiative loads on those elements are smaller than the conductive loads through the mechanical support structures.

## 3.5 Instrument Integration and Test

The PICO instrument integration and testing plan benefits from heritage and experience with the *Planck* HFI instrument [95].

We screen detector wafers prior to selection of flight wafers and focal-plane integration. The cADR and 4 K cryocooler vendors will qualify them prior to delivery. We will determine the relative alignment of the two reflectors under in-flight thermal conditions using a thermal vacuum (TVAC) chamber and photogrammetry. We integrate the flight focal-plane assembly and flight cADR in a dedicated sub-kelvin cryogenic testbed. We characterize noise, responsivity, and focal-plane temperature stability using a representative optical load for each frequency band (temperature-controlled blackbody), and we perform polarimetric and spectroscopic calibration.

The focal plane is integrated with the reflectors and structures, and alignment verified with photogrammetry at cold temperatures in a TVAC chamber. The completely integrated observatory (instrument and spacecraft bus) is tested in TVAC to measure parasitic optical loading from the instrument, noise, microphonics, and RFI. The observatory is 4.5 m in diameter and 6.1 m tall. There are no deployables.

## 4 Design Reference Mission

The PICO design reference mission is summarized in Table 4.1.

Table 4.1: PICO carries margin on key mission parameters. Maximum Expected Value (MEV) includes contingency.

### 4.1 Concept of Operations

The PICO concept of operations is similar to that of the successful *WMAP* [96] and *Planck* [97] missions. After launch, PICO cruises to a quasi-halo orbit around the Earth–Sun L2 Lagrange point (§ 4.1.1). A two-week decontamination period is followed by instrument cooldown, lasting about two months. After in-orbit checkout is complete, PICO begins its science survey.

PICO has a single science observing mode, surveying the sky continuously for 5 years using stabilized

|   |   |
|---|---|
| Orbit type .....  | Sun-Earth L2 Quasi-Halo                                     |
| Mission class .....   | Class B   |
| Mission duration .....  | 5 years   |
| Propellant (hydrazine).....   | 213 kg (77 % tank fill)                                     |
| Launch mass (MEV) .....   | 2147 kg (3195 kg capability)                                |
| Max power (MEV) .....   | 1320 W (with 125 % margin on available solar array area)    |
| Onboard data storage .....  | 4.6 Tb (3 days of compressed data, enabling retransmission) |
| Survey implementation .....   | Instrument on spin table                                    |
| Attaining the surveying the sky continuously for 5 years using stabilized |   |

a pre-planned repetitive survey pattern (§ 4.1.2). Instrument data are compressed and stored on-board, then returned to Earth in daily 4-hr Ka-band science downlink passes (concurrent with science observations). Because PICO is observing relatively static Galactic, extragalactic, and cosmological targets, there are no requirements for time-critical observations or data latency. Presently, there are no plans for targets of opportunity or guest observer programs during the prime mission. The PICO instrument does not require cryogenic consumables (as the *Planck* mission did), permitting consideration of significant mission extension beyond the prime mission.

#### 4.1.1 Mission Design and Launch

The science survey is conducted from a quasi-halo orbit around the Earth–Sun L2 Lagrange point. *Planck* and *WMAP* also operated in L2 orbits. L2 orbits provide favorable survey geometry relative to Earth orbits by mitigating viewing restrictions imposed by terrestrial and lunar stray light. The PICO orbit around L2 is small enough to ensure that the Sun–Probe–Earth (SPE) angle is less than  $15^\circ$ . This maintains the telescope boresight  $> 70^\circ$  away from the Earth (Fig. 4.2,  $70^\circ = 180^\circ - \alpha - \beta - \text{SPE}$ ).

High data-rate downlink to the Deep Space Network (DSN) is available from L2 using near-Earth Ka bands. L2 provides a stable thermal environment, simplifying thermal control. The PICO orbit exhibits no post-launch eclipses.

NASA requires that Probes be compatible with an Evolved Expendable Launch Vehicle (EELV). For the purpose of this study, the Falcon 9 [98] is used as the reference vehicle. Figure 4.1 shows PICO configured for launch in a Falcon 9 fairing. The Falcon 9 launch capability for ocean recovery exceeds PICO’s 2147 kg total launch mass (including contingency) by a 50 % margin.

Insertion to the halo manifold and associated trajectory correction maneuvers require  $150 \text{ m s}^{-1}$  of total  $\Delta V$  by the spacecraft. Orbit maintenance requires minimal propellant (statistical  $\Delta V \sim 2 \text{ m s}^{-1} \text{ year}^{-1}$ ). The orbital period is  $\sim 6$  months. There are no disposal requirements for L2 orbits, but spacecraft are customarily decommissioned to heliocentric orbit.

#### 4.1.2 Survey Design

PICO employs a highly repetitive scan strategy to map the full sky. During the survey, PICO spins with a period  $T_{\text{spin}} = 1 \text{ min}$  about a spin axis oriented  $\alpha = 26^\circ$  from the anti-solar direction

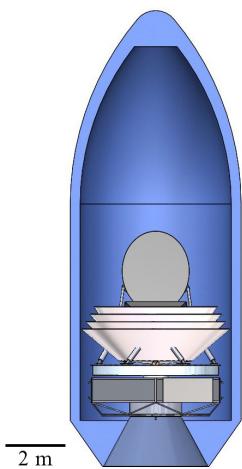


Figure 4.1: PICO is compatible with the Falcon 9.

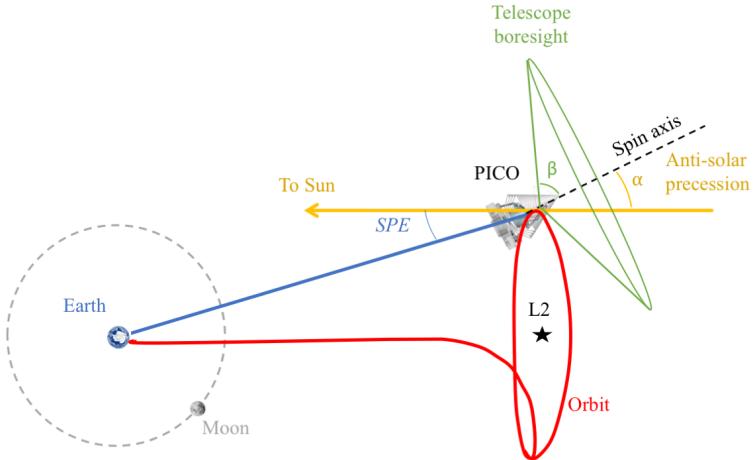


Figure 4.2: PICO surveys by continuously spinning the instrument about a precessing axis.

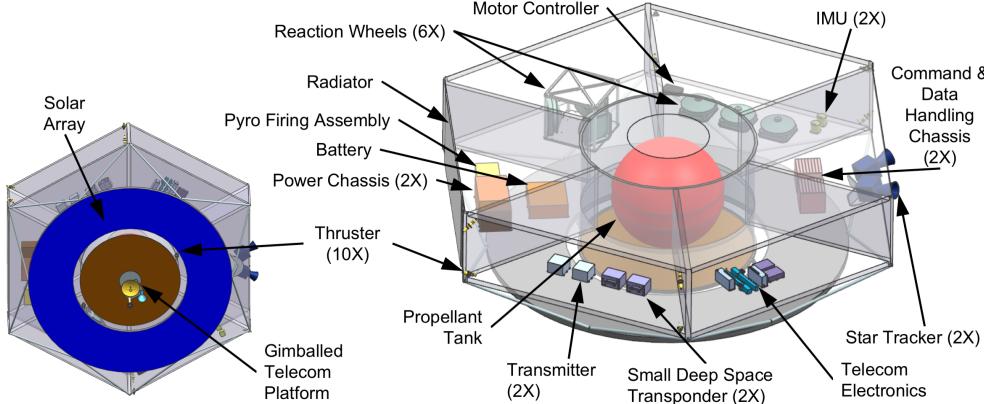


Figure 4.3: Modular equipment bays provide easy access to all components in the spacecraft de-spun module and enable parallel integration of spacecraft subsystems.

(Fig. 4.2). This spin axis is forced to precess about the anti-solar direction with a period  $T_{\text{prec}} = 10 \text{ hr}$ . The telescope boresight is oriented at an angle  $\beta = 69^\circ$  away from the spin axis (Fig. 3.1). This  $\beta$  angle is chosen such that  $\alpha + \beta > 90^\circ$ , enabling mapping of all ecliptic latitudes. The precession axis tracks along with the Earth in its yearly orbit around the Sun, so this scan strategy maps the full sky (all ecliptic longitudes) within 6 months.

PICO’s  $\alpha = 26^\circ$  value is chosen to be substantially larger than the *Planck* mission’s  $\alpha$  angle ( $7.5^\circ$ ) to mitigate systematic effects by scanning across each sky pixel with a greater diversity of orientations [99]. Increasing  $\alpha$  further would decrease the Sun-shadowed volume available for the optics and consequently reduce the telescope aperture size. A deployable Sunshade was considered, but found not to be required, and was thus excluded in favor of a more conservative and less costly approach.

The instrument spin rate, selected through a trade study, matches that of the *Planck* mission. The study balanced low-frequency ( $1/f$ ) noise subtraction (improves with spin rate) against implementation cost and heritage, pointing reconstruction ability (anti-correlated with spin rate), and data volume (linearly correlated with spin rate). The CMB dipole appears in the PICO data timestream at the spin frequency (1 rpm = 16.7 mHz). Higher multipole signals appear at harmonics of the spin frequency, starting at 33 mHz, above the knee in the detector low-frequency noise (§ 3.2.4). A destriping mapmaker applied in data post-processing effectively operates as a high-pass filter, as demonstrated by *Planck* [100]. PICO’s spin-axis precession frequency is more than 400 times faster than that of *Planck*, greatly reducing the effects of any residual  $1/f$  noise by spreading the effects more isotropically across pixels.

## 4.2 Ground Segment

The PICO Mission Operations System (MOS) and Ground Data System (GDS) can be built with extensive reuse of standard tools. The PICO concept of operations is described in § 4.1. All space-ground communications, ranging, and tracking are performed by the DSN 34 m Beam Wave Guide (BWG). X-band is used to transmit spacecraft commanding, return engineering data, and provide navigation information (S-band is a viable alternative, and could be considered in a future trade). Ka-band is used for high-rate return of science data. The baseline 150 Mb/s transfer rate (130 Mb/s information rate after CCSDS encoding) is an existing DSN catalog service [101]. The instrument produces 6.1 Tb/day, which is compressed to 1.5 Tb/day (§ 3.3). Daily 4 hr DSN passes return PICO data in 3.1 hr, with the remaining 0.9 hr available as needed for retransmission or missed-pass recovery.

## 4.3 Spacecraft

The PICO spacecraft bus is Class B and designed for a minimum lifetime of 5 years in the L2 environment. Mission-critical elements are redundant. Flight spares, engineering models, and prototypes appropriate to Class B are budgeted.

The aft end of the spacecraft (the “de-spun module”) is comprised of six equipment bays that house standard components (Fig. 4.3). The instrument and V-grooves are mounted on bipods from the spacecraft “spun module,” which contains hosted instrument elements (Fig. 3.1). A motor drives the spun module at 1 rpm to support the science survey requirements (§ 4.1.2). Reaction wheels on the despun module cancel the angular momentum of the spun module and provide three-axis control (§ 4.3.1).

The bipods that mechanically support the instrument are thermally insulating. The passively radiating V-groove assembly thermally isolates the instrument from solar radiation and from the bus (§ 3.4.3). Like *Planck* [97], the V-grooves are manufactured using honeycomb material. Additional radiators on the spun and despun spacecraft modules ( $\sim 1 \text{ m}^2$  each) reject heat dissipated by spacecraft subsystems and hosted instrument elements.

PICO’s avionics are dual-string with standard interfaces. Solid-state recorders provide three days of science data storage (4.6 Tbit, § 7.1), enabling retransmission of missed data (§ 4.2).

PICO employs a fully redundant Ka- and X-band telecommunications architecture. The Ka-band system uses a 0.3 m high-gain antenna to support a science data downlink information rate of 130 Mb/s to a 34 m BWG DSN ground station with a link margin of 4.8 dB. The X-band system provides command and engineering telemetry communication through all mission phases using medium- and low-gain antennas. Amplifiers, switches, and all three antennas are on a gimballed platform, enabling Ka and X-band downlink concurrent with science observations.

The heritage power electronics are dual-string. A 74 A-hr Li-ion battery is sized for a 3 hr launch phase with 44 % depth of discharge. After the launch phase, the driving mode is telecom concurrent with science survey (1320 W including 43 % contingency). Solar cells on the aft side of the bus ( $5.8 \text{ m}^2$  array,  $\alpha = 26^\circ$  off-Sun) support this mode with positive power, and unused area in the solar array plane ( $7.4 \text{ m}^2$  more area by growing to 4.5 m diameter) affords 125 % margin (Fig. 4.3).

The propulsion design is a simple mono-propellant blow-down hydrazine system with standard redundancy. Two aft-pointed 22 N thrusters provide  $\Delta V$  and attitude control for orbit insertion and maintenance (§ 4.1.1), requiring 140 kg of propellant. Eight 4 N thrusters provide reaction-wheel momentum management and backup attitude-control authority (60 kg of propellant). Accounting for ullage (14 kg), the baseline propellant tank fill fraction is 77 %.

### 4.3.1 Attitude Determination and Control

PICO uses a zero net angular momentum control architecture with heritage from the SMAP mission (§ 7.1). PICO’s instrument spin rate (1 rpm) matches that of the *Planck* mission, but the precession of the spin axis is faster (10 hr vs 6 months), and the precession angle larger ( $26^\circ$  vs  $7.5^\circ$ ). These differences make the spin-stabilized *Planck* control architecture impractical.

The PICO instrument spin rate is achieved and maintained using a spin motor. The spin motor drive electronics provide the coarse spin rate knowledge used for controlling the spin rate to meet the  $\pm 0.1$  rpm requirement. Data and power are passed across the interface using slip rings.

PICO requires 220 N m s to cancel the angular momentum of the instrument and spacecraft spun module at 1 RPM. This value includes mass contingency and is based on the CAD model. Three

Honeywell HR-16 reaction wheel assemblies (RWAs), each capable of  $150 \text{ N m s}$ , are mounted on the despun module parallel to the instrument spin axis, and spin opposite to the instrument to achieve zero net angular momentum. The despun module is three-axis stabilized. The spin axis is precessed using three RWAs mounted normal to the spin axis in a triangle configuration. Each set of three RWAs is sized such that two could perform the required function with margin, providing single fault tolerance.

Spin-axis pointing and spin-rate knowledge are achieved and maintained using star tracker and inertial measurement unit (IMU) data. The attitude determination system is single-fault tolerant, with two IMUs each on the spun and despun modules, and two star trackers each on the spun and despun modules. Two Sun sensors on the despun module are used for safe-mode contingencies and instrument Sun avoidance. All attitude control and reconstruction requirements are met, including spin axis control  $< 60 \text{ arcmin}$  with  $< 1 \text{ arcmin/min}$  stability, and reconstructed pointing knowledge  $< 10 \text{ arcsec}$  (each axis,  $3\sigma$ ).

Additional pointing reconstruction is performed in post-processing using the science data. The PICO instrument will observe planets (compact, bright sources) nearly every day. By fitting the telescope pointing to the known planetary ephemerides, the knowledge of the telescope boresight pointing and the relative pointing of each detector will improve to better than  $1 \text{ arcsec}$  (each axis,  $3\sigma$ ). *Planck*, with fewer detectors, making lower SNR measurements of the planets, and observing with a scan strategy that acquired measurements of each planet only once every 6 months, demonstrated  $0.8 \text{ arcsec}$  ( $1\sigma$ ) pointing reconstruction uncertainty in-scan and  $1.9 \text{ arcsec}$  ( $1\sigma$ ) cross-scan [102].

## 5 Technology Drivers

PICO builds off of the heritage of *Planck-HFI* and *Herschel*. Since the time of *Planck* and *Herschel*, suborbital experiments have used monolithically fabricated TES bolometers and multiplexing schemes to field instruments with thousands of TES bolometers per camera (Fig. 5.1). By the time PICO enters Phase A, S3 experiments plan to be operating nearly 100,000

TES bolometers in several independent cameras [104–106].

The remaining technology developments required to enable the PICO baseline design are:

1. extension of three-color antenna-coupled bolometers down to 21 GHz and up to 462 GHz (§ 5.1);
2. construction of high-frequency direct absorbing arrays and laboratory testing (§ 5.2);
3. beam line and 100 mK testing to simulate the cosmic ray environment at L2 (§ 5.3);
4. expansion of time-division multiplexing to support 128 switched rows per readout column (§ 5.4).

All of these developments are straightforward extensions of technologies already available today. We recommend APRA and SAT support to complete development of these technologies through the milestones described in Table 5.1.

### 5.1 21–462 GHz Bands

Suborbital teams have successfully demonstrated a variety of optical-coupling schemes, including horns with ortho-mode transducers (OMTs), lithographed antenna arrays, and sinuous antennas under lenslets (Table 5.2). All have achieved background-limited performance with sufficient margin on design parameters to achieve this performance in the lower background environment at L2. All have been packaged into modules and focal-plane units in working cameras representative of the PICO integration. Experiments have already used a number of PICO’s observing bands between 27 GHz and 270 GHz (Table 5.2). To date, statistical map depths of  $3 \mu\text{K}_{\text{CMB}}$  arcmin have been achieved over small sky areas, which is within a factor of five of PICO’s CBE over the entire sky (Table 3.2).

The baseline PICO instrument requires three-color dual-polarized antenna-coupled bolometers covering bands from 21 to 462 GHz (§ 3.2.1). The sinuous antenna has the bandwidth to service three bands per pixel, whereas horns and antenna arrays have only been used for two. Our baseline is to use a three-band sinuous antenna, although we have designs that use two- or one-band per pixel and have the same or similar baseline noise as PICO (§ 5.5). SPT-3G has used the PICO-baselined three-color pixel design to deploy 16,000 detectors covering 90/150/220 GHz [103].

The extension to lower frequencies requires larger antennas and therefore control of film properties and lithography over larger areas. Scaling to higher frequencies requires tighter fabrication tolerances and electromagnetic wave transmission losses tend to increase due to material properties. Current anti-reflection technologies for the lenslets need to be extended with thicker and thinner layers to cover the lowest and highest frequency channels. These developments will require



Figure 5.1: SPT-3G operates a focal plane with sinuous antenna-coupled, three-band pixels with 16,000 bolometers [103]. Each pixel couples radiation to bands at 95, 150, and 220 GHz.

Table 5.1: PICO technologies can be developed to TRL 5 prior to a 2023 Phase A start using the APRA and SAT programs, requiring a total of about \$ 13M. Per NASA guidance, these costs are outside the mission cost (§ ??).

| Task   | Current status                             | Milestone A                               | Milestone B                  | Milestone C                | Current funding | Required funding                   | Date TRL5 achieved |
|--|--|---|------------------------------|----------------------------|-----------------|------------------------------------|--------------------|
| 1a. Three-color arrays<br>$v < 90 \text{ GHz}$     | 2-color lab demos<br>$v > 30 \text{ GHz}$  | Field demo of 30–40 GHz (2020)            | Lab demos 20–90 GHz (2022)   | —                          | APRA & SAT      | \$2.5M over 4 yr (1 APRA + 1 SAT)  | 2022               |
| 1b. Three-color arrays<br>$v > 220 \text{ GHz}$    | 2-color lab demos<br>$v < 300 \text{ GHz}$ | Field demo of 150–270 GHz (2021)          | Lab demos 150–460 GHz (2022) | —                          | APRA & SAT      | \$3.5M over 4 yr (2 SATs)          | 2022               |
| 2. Direct absorbing arrays<br>$v > 50 \text{ GHz}$ | 0.1–5 THz unpolarized                      | Design & prototype of arrays (2021)       | Lab demo of 555 GHz (2022)   | Lab demo of 799 GHz (2023) | None            | \$2M over 5 yr (1 SAT)             | 2023               |
| 3. Cosmic ray studies                              | 250 mK w/ sources                          | 100 mK tests with sources (2021)          | Beamline tests (2023)        | —                          | APRA & SAT      | \$0.5–1M over 5 yr (part of 1 SAT) | —                  |
| 4a. Fast readout electronics                       | MUX66 demo                                 | Engineering and Fab of electronics (2020) | Lab demo (2021)              | Field demo (2023)          | No direct funds | \$4M over 5 yr (1 SAT)             | 2023               |
| 4b. System engineering;<br>128× MUX demo           | MUX66 demo                                 | Design of cables (2020)                   | Lab demo (2021)              | Field demo (2023)          | No direct funds | —                                  | —                  |

Table 5.2: Multiple active suborbital efforts are advancing technologies relevant to PICO.

| Project                 | Type    | Optical Coupling | $v_c$<br>[GHz] | Colors per pixel | $N_{\text{bolo}}$ | Significance     | Reference      |
|-------------------------|---------|------------------|----------------|------------------|-------------------|------------------|----------------|
| PICO baseline .....     | Flight  |                  | 21 – 462       | Three            | 11,796            |                  | § 3.2.1        |
| SPT-3G .....            | Ground  | Sinuous          | 90 – 220       | Three            | 16,260            | Trichroic        | [103]          |
| Advanced ACT-pol .      | Ground  | Horns            | 27 – 230       | Two              | 3,072             | Dichroic         | [107]          |
| BICEP/Keck .....        | Ground  | Antenna arrays   | 90 – 270       | One              | 5,120             | 50 nK-deg        | [108]          |
| Berkeley, Caltech, NIST | Lab     | Various          | 30 – 270       | Various          | –                 | Band coverage    | [79, 109, 110] |
| SPIDER .....            | Balloon | Antenna arrays   | 90 – 150       | One              | 2,400             | Stable to 10 mHz | [77]           |

control of cleanliness and understanding of process parameters. Changes to elements in the light path will require characterization of beam properties.

The direction of polarization sensitivity of the sinuous antenna varies with frequency, thus presenting a potential source of systematic error. Over 25% bandwidth, the variation is approximately  $\pm 5 \text{ deg}$  [111]. There are solutions to this in the focal-plane design, measurements, data analysis, and free parameters of the sinuous antenna geometry. A recent study found that pre-flight characterization of the effect through measurements can readily mitigate it as a source of systematic uncertainty [112]. Studies with current field demonstrations, such as with the data of SPT-3G, will be particularly important. The PICO concept is robust to any challenges in developing three-color

Table 5.3: PICO high-frequency detectors leverage development and demonstration by *Planck*, *Herschel*, and SPT.

| Project                 | Type   | Polarized   | Mono-lithic | $v_c$<br>[GHz] | Colors<br>per pixel | $N_{\text{bolo}}$ | Significance            | Reference |
|-------------------------|--------|-------------|-------------|----------------|---------------------|-------------------|-------------------------|-----------|
| PICO baseline .....     | Flight | Yes         | Yes         | 555 – 799      | One                 | 1,200             |                         | § 3.2.2   |
| <i>Planck</i> HFI ..... | Flight | 143–343 GHz | No          | 143 – 857      | One                 | 48                | TRL 9 polarized         | [74]      |
| <i>Herschel</i> .....   | Flight | No          | Yes         | 570 – 1200     | One                 | 270               | TRL 9 monolithic        | [114]     |
| SPT-SZ .....            | Ground | No          | Yes         | 90 – 220       | One                 | 840               | Monolithic array TESs   | [72]      |
| SPT-pol-90 .....        | Ground | Yes         | No          | 90             | One                 | 180               | Dual pol absorbing TESs | [115]     |

pixels; § 5.5 describes options to descope to two- and one-color pixels, technologies for which the polarization sensitivity is constant as a function of frequency.

## 5.2 555–799 GHz bands

The baseline PICO instrument requires single-color, horn-coupled, dual-polarization, direct-absorbing bolometers from 555 to 799 GHz (§ 3.2.2). *Planck* and *Herschel* demonstrated the architecture of horns coupled to direct absorbing bolometers. Ground experiments with similar designs have deployed focal planes with hundreds of horn-coupled spiderweb bolometers, replacing the *Planck* and *Herschel* NTD-Ge thermistors with TESs, and adjusting time constants as necessary (Table 5.3). *Planck*-HFI, SPT-pol, and BICEP demonstrated dual-polarized detectors. *Herschel* and SPT-SZ demonstrated monolithic unpolarized detectors. PICO will require detectors that merge these two designs in monolithic dual-polarized arrays. Since all the components of the technology already exist, the remaining necessary development is the packaging. Filled arrays of detectors such as Backshort Under Ground (BUG) bolometers are also an option [113].

## 5.3 Environmental Testing

Laboratory tests and in-flight data from balloons suggest that TES bolometer arrays may be more naturally robust against cosmic rays than the individual NTD-Ge bolometers used in *Planck*. PICO will leverage lessons learned from *Planck* and ensure robust thermal sinking of detector array substrates. Cosmic-ray glitches have fast recovery times and low coincidence rates [116, 117]. Residual risk can be retired with 100 mK testing where the array heat sinking may be weaker, and beam-line tests to simulate the expected flight environment.

## 5.4 Multiplexing

More than ten experiments have used time-domain multiplexer (TDM) readout. SCUBA2 on JCMT has 10,000 pixels, nearly as many detectors as planned for PICO [118]. Most of these experiments have used 32-row multiplexing. Recently ACT has expanded this to 64-row multiplexing [78].

PICO’s sensitivity requirements dictate the use of 13,000 transition-edge-sensor bolometers and a multiplexed system. Our baseline design is to use TDM readout with 128 switched rows per readout column (TDM-128×). The leap to TDM-128× requires:

- development of fast-switched room temperature electronics; and
- system engineering of room temperature to cryogenic row-select cabling to ensure sufficiently fast row-switch settling times.

The historical row revisit rate for bolometric instruments using 32× TDM has been 25 kHz [e.g.,

[68](#). However, X-ray instruments using TDM routinely switch between rows at 6.25 MHz [\[119\]](#). The PICO baseline assumes a 6.25 MHz switch rate and TDM- $128\times$ , which dictates a row-revisit rate of 48.8 kHz. To limit aliased noise, PICO implements low-pass filters in each readout channel with a bandwidth of 6 kHz, dictated by detector stability considerations and the required  $\sim 1$  kHz signal bandwidth. With these parameters and using the same TDM multiplexer SQUID design, the increased total noise due to aliasing is less than 15 % and is included in our detector noise budget. The system engineering study will culminate in a demonstration of TDM- $128\times$  SQUID aliased noise below PICO detector sensitivity requirements.

## 5.5 Technology Descores

A descope from three-color sinuous antenna/lenslet-coupled pixels to two-color horn-coupled, or to single color antenna-array pixels remains a viable alternative should the three-color technology not mature as planned. In both alternative options, bands above 555 GHz are the same as the baseline. For the lower frequencies, the two-color horn-coupled pixel option contains 8,840 detectors and has 19 colors. Because horns have a 2.3 : 1 bandwidth, each of the two bands in a pixel has 35 % bandwidth (compared to the baseline 25 %), which compensates for pixel count, resulting in  $0.61 \mu\text{K}_{\text{CMB}}$  arcmin aggregate CBE map depth. This is the same as the three-color CBE map depth, and affords the same 40 % margin relative to the  $0.87 \mu\text{K}_{\text{CMB}}$  arcmin baseline requirement (Table [3.2](#)). Detailed analysis would be performed to assess the impact of the coarser spectral resolution on signal component separation. Single color antenna-array pixels can have higher packing density than the other two architectures. This option has 6,540 detectors, 21 colors, each with 30 % bandwidth, and a noise level of  $0.74 \mu\text{K}_{\text{CMB}}$  arcmin, leaving only 17% noise margin relative to the requirement.

## 5.6 Enhancing Technologies

The following technologies are neither required nor assumed by the PICO baseline concept. However, they represent opportunities to extend scientific capabilities or simplify engineering.

PICO baselines TDM readout because of its relative maturity and demonstrated sensitivity and stability in relevant science missions. Lab tests of frequency-domain multiplexing (FDM) give comparable performance with higher multiplexing factors and lower thermal loads on cryogenic stages relative to TDM, but with higher ambient temperature power consumption. Suborbital experiments such as SPT-3G are using FDM to read out focal planes comparable in size to PICO.

Microwave frequency SQUID multiplexing can increase the multiplexing density and reduce the number of wires between the 4 K and ambient temperature stages [\[120, 121\]](#). Kinetic inductance detectors and Thermal KIDs can further reduce the wire count, obviate the need for SQUID-based amplifiers, and simplify integration by integrating the multiplexing function on the same substrate as the detectors [\[122–124\]](#). The cost to develop these technologies is \$3–4M/year, with a high chance of reaching TRL-5 before Phase A.

# 6 Organization, Partnerships, and Current Status

PICO is the result of an 18-months mission study funded by NASA (total grant = \$150,000). The study was open to the entire mm/sub-mm community. Seven working groups were led by members of PICO’s Executive Committee, which had a telephone conference weekly, led by the PI. A three-member steering committee, composed of two experimentalists experienced with CMB space missions, and a senior theorist gave occasional advice to the PI. More than 60 scientists, international-

and US-based, participated in-person in each of two community workshops (November 2017 and May 2018). The study report has been submitted by NASA to the decadal panel, and it is available on the arXiv and on the PICO website [58, 125]. It has contributions from 82 authors, and has been endorsed by additional 131 members of the community.

The PICO team designed an entirely US-based mission, so that the full cost of the mission can be assessed. We excluded contributions by other space agencies, despite expression of interest by international scientists. The PICO concept has wide support in the international community. If the mission is selected to proceed, a path that would be scientifically and financially optimal relative to other options, it is reasonable to expect that international partners would participate and thus reduce the US cost of the mission.

## 7 Schedule and Cost

|                | FY24           | FY25              | FY26              | FY27         | FY28             | FY29                        | FY30  | FY31  | FY32  | FY33  | FY34  | FY35         |
|----------------|----------------|-------------------|-------------------|--------------|------------------|-----------------------------|-------|-------|-------|-------|-------|--------------|
| CY 2023        | CY 2024        | CY 2025           | CY 2026           | CY 2027      | CY 2028          | CY 2029                     | CY 30 | CY 31 | CY 32 | CY 33 | CY 34 |              |
| PH A (12 mths) | PH B (12 mths) | PHASE C (22 mths) | PHASE D(18 mths ) |              | PHASE E (5 yrs ) |                             |       |       |       |       |       | F 4 mths     |
| ♦ 10/23 KDP-A  | ♦ 10/24 KDP-B  | ♦10/25 KDP-C      |                   | ♦ 8/27 KDP-D |                  | ♦ 2/29 PLAR (Start of Ph E) |       |       |       |       |       | KDP-F 2/34 ♦ |
| Reviews        | 10/25 PDR ♦    |                   | ♦ 7/26 CDR        | ♦ 7/27 ARR   |                  | Launch 1/29★                |       |       |       |       |       |              |

Figure 7.1: PICO development and operations schedule.

- Schedule** NASA-funded Probe studies including PICO assume a Phase A start in October 2023. PICO development phases B-D are similar in duration to recent comparably sized NASA missions such as Juno and SMAP. PICO is a cryogenic mission similar to *Planck*, but the cryogenic design is simpler because all PICO’s bolometric detectors are maintained at 0.1 K (*Planck*’s bolometers were maintained at 0.1 K, and the radiometers at 20 K). We used experience from *Planck* and from current implementations of ground-based kilo-pixel arrays to allocate appropriate time for integration and testing (I&T).

The baseline mission lifetime is 5 years. The PICO instrument does not have cryogenic consumables (as *Planck* did), permitting mission extension beyond the prime mission duration.

- Cost** We estimate PICO’s total Phase A–E lifecycle cost between \$870M and \$960M, including the \$150M allocation for the Launch Vehicle (per NASA direction). These cost estimates include 30 % reserves for development (Phases A–D) and 13 % reserves for operations (Phase E). Table 7.1 shows the JPL Team X

Table 7.1: Detailed breakdown of Team X and PICO Team cost estimates (in FY18\$). Costs are based on the schedule in Fig. 7.1, which includes 5 years of operations.

| Work Breakdown Structure<br>(WBS) elements                                 | Team X  | PICO        |
|--|---------|-------------|
| Development Cost (Phases A–D)  | \$ 724M | \$ 634–677M |
| 1.0, 2.0, 3.0 Management,<br>Systems Engineering, and<br>Mission Assurance | \$ 54M  | \$ 47– 50M  |
| 4.0 Science  |         | \$ 19M      |
| 5.0 Payload System   |         | \$ 168M     |
| 6.0 Flight System  | \$ 248M | \$ 210–240M |
| 10.0 Assembly, Test, and<br>Launch Operations (ATLO)                       | \$ 24M  |             |
| 7.0 Mission Operations<br>Preparation                                      |         | \$ 16M      |
| 9.0 Ground Data Systems  |         | \$ 21M      |
| 12.0 Mission and Navigation<br>Design                                      |         | \$ 7M       |
| Development Reserves (30%)   | \$167M  | \$ 146–156M |
| Operations Cost (Phase E)  |         | \$ 84M      |
| 1.0 Management   |         | \$ 6M       |
| 4.0 Science  |         | \$ 20M      |
| 7.0 Mission Operations   |         | \$ 34M      |
| 9.0 Ground Data Systems  |         | \$ 14M      |
| Operations Reserves (13%)  |         | \$ 10M      |
| Launch Vehicle Cost  |         | \$ 150M     |
| Total Cost   | \$ 958M | \$ 868–911M |

and the PICO team mission cost breakdown.

Team X estimates are generally model-based, and were generated after a series of instrument and mission-level studies. The PICO team adopted the Team X estimates, but also obtained a parametrically estimated cost range for the Flight System and Assembly, Test, and Launch Operations from Lockheed Martin Corporation to represent the cost benefits that might be realized by working with an industry partner. After adding estimated JPL overhead the PICO team cost is in-family with but lower than the Team X cost.

Science team costs are assessed by Team X based on PICO science team estimates of the numbers and types of contributors and meetings required for each year of PICO mission development and operations. These workforce estimates are informed by recent experience with the *Planck* mission. PICO’s spacecraft cost reflects a robust Class B architecture. Mission-critical elements are redundant. Appropriate flight spares, engineering models and prototypes are included. Mission operations, Ground Data Systems, and Mission Navigation and Design costs reflect the relatively simple operations: PICO has a single instrument and a single, repetitive science observing mode.

The active cooling system (the 0.1 K cADR and 4 K cryocooler) comprises nearly half of the payload cost. The cADR cost for this study is an estimate from Goddard Space Flight Center. The 4 K cryocooler cost for this study is based on the NASA Instrument Cost Model (NICM) VIII CER Cryocooler model [126], assuming a commercial build. Based on JPL experience, 18 % of the instrument cost is allocated for integration and testing (I&T). More details on the cost of PICO are available in the full PICO report [58].

## 7.1 Heritage

PICO’s reflectors are similar to *Planck*’s, but somewhat larger (270 cm × 205 cm primary versus 189 cm × 155 cm) [127]. *Herschel* observed at shorter wavelengths that required higher surface accuracy and had a larger reflector (350 cm diameter primary) [128]. PICO’s detectors are cooled by a cADR with requirements that are within the capabilities of current ADRs developed by Goddard Space Flight Center. These systems have been applied to several JAXA missions, including *Hitomi* [91]. PICO’s 4 K cryocooler (§ 3.4.2) is a direct extension of the JWST MIRI design [92, 93]. PICO benefits from a simpler and more reliable implementation of the J-T system than was required for MIRI, in that no deployment of cooling lines is required, and all flow valving is performed on the warm spacecraft. Structures similar to PICO’s V-groove radiator assembly are a standard approach for passive cooling, and were first described more than thirty years ago [129]. PICO’s spin system is less demanding than the successful SMAP spin system. The PICO spin rate is 1 rpm, and the mission requires ∼ 220 N m s of spin angular momentum cancellation. The PICO’s data volume and downlink rates are already surpassed by missions in development.

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