

Isogeometric Methods: Homework #5

Problem 1: Structural Vibrations

In this problem, you will solve the problem of structural vibrations of a solid aluminum NACA 0012 airfoil. Assume that the length of the corresponding aircraft wing is infinitely long, resulting in a plane strain stress state. The resulting problem definition is as depicted in Figure 1. For the problem in consideration, the physical and geometric parameters are as follows:

$$E = 69 \text{ GPa}$$

$$\nu = 0.35$$

$$\rho = 2700 \text{ kg/m}^3$$

$$c = 1 \text{ m}$$

Compute the first five non-rigid body modes and corresponding natural frequencies, and plot both components of the resulting displacement field and the σ_{xx} , σ_{yy} , and σ_{xy} stress components over the *deformed* physical geometry for each of the computed modes.

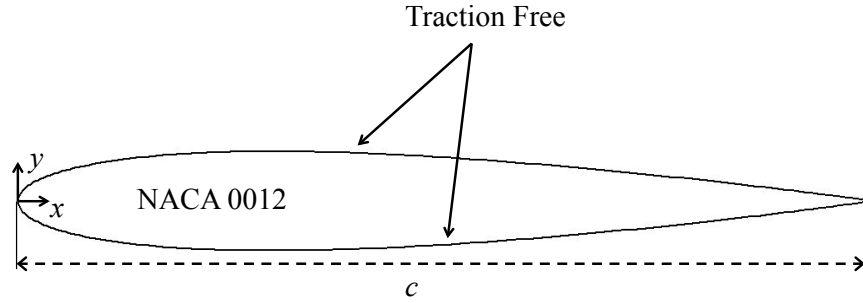


Figure 1: Setup for Problem 1.

Note: The formula for the shape of a NACA 00xx airfoil, with “xx” being replaced by the percentage of thickness to chord, is:

$$y_t = \frac{t}{0.2}c \left[0.2969\sqrt{\frac{x}{c}} - 0.1260\left(\frac{x}{c}\right) - 0.3516\left(\frac{x}{c}\right)^2 + 0.2843\left(\frac{x}{c}\right)^3 - 0.1036\left(\frac{x}{c}\right)^4 \right]$$

where c is the chord length, x is the position along the chord from 0 to c , y_t is the half thickness at a given value of x (centerline to surface), and t is the maximum thickness as a fraction of the chord (i.e., $t = \text{xx}/100$).

Problem 2: Wave Propagation

In this problem, you will solve the problem of elastic wave propagation in a thick walled steel cylindrical pressure vessel due to an ultrasonic internal pressure loading of the form:

$$p(t) = \delta p_i \exp i\omega t$$

where ω , the frequency of the loading, is significantly larger than the upper limit of human hearing (greater than 20 MHz). Assume that the ends of the cylinder are capped, resulting in a plane strain stress state. The resulting problem definition is as depicted in Figure 2. For the problem in consideration, the physical and geometric parameters are as follows:

$$E = 200 \text{ GPa}$$

$$\nu = 0.3$$

$$\rho = 7850 \text{ kg/m}^3$$

$$\delta p_i = 5 \text{ MPa}$$

$$r_i = 75 \text{ mm}$$

$$t = 15 \text{ mm}$$

Compute the resulting amplitude field $\mathbf{u}(\mathbf{x})$ for the two frequency values $\omega = 5 \text{ MHz}$ and $\omega = 30 \text{ MHz}$. Plot both components of the amplitude field and the two in-plane principal stress amplitudes over the *undeformed* physical geometry for each considered frequency value. Are the resulting elastic waves primary or secondary waves? Estimate the wave-speeds using your computations and compare your results with the theoretical wave-speed values. Discuss any discrepancies you observe.

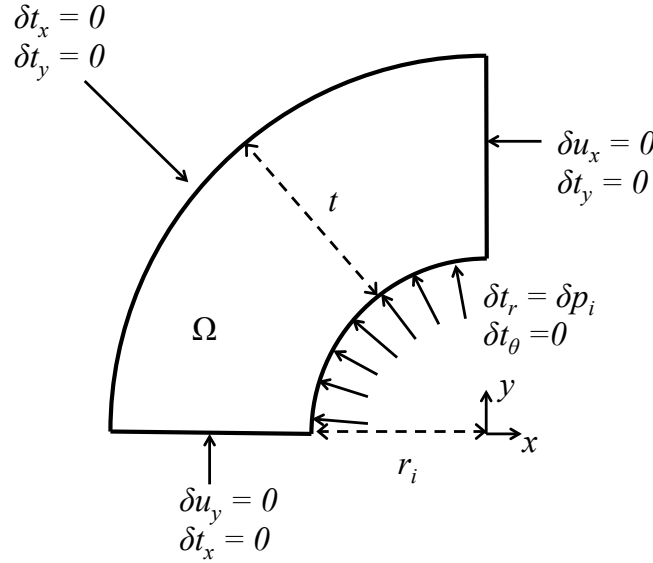


Figure 2: Setup for Problem 2.

Problem 3: Elastodynamics

In this problem, you will solve the problem of time-dependent elastic deformation of a thick walled steel cylindrical pressure vessel due to a dynamic internal pressure loading of the form:

$$p(t) = p_i \left(\frac{t}{t_{peak}} \right) F \left(\frac{t_{peak} - t}{t_{decay}} \right)$$

where $F(\eta) = 0.5(1 + \tanh(\eta))$, t_{peak} corresponds to the time of peak loading, and t_{decay} is the period of decay. The resulting non-dimensional pressure field $\tilde{p}(t) := p(t)/p_i$ is illustrated in Figure 3 for $t_{peak} = 0.05$ ms and $t_{decay} = 0.15$ ms. This problem is motivated by deformation due to detonation of a gas inside of a steel pipe. Note that such a problem is three-dimensional in nature, however, with resulting pressure waves traversing in both radial and axial directions. Assume that the ends of the cylinder are capped, resulting in a plane strain stress state. The resulting problem definition is as depicted in Figure 4. For the problem in consideration, the physical and geometric parameters are as follows:

$$E = 200 \text{ GPa}$$

$$\nu = 0.3$$

$$\rho = 7850 \text{ kg/m}^3$$

$$p_i = 30 \text{ MPa}$$

$$t_{peak} = 0.05 \text{ ms}$$

$$t_{decay} = 0.15 \text{ ms}$$

$$r_i = 75 \text{ mm}$$

$$t_{pipe} = 15 \text{ mm}$$

Assuming that there is no viscous damping and that the vessel is initially at rest, compute the resulting time-dependent displacement field $\mathbf{u}(\mathbf{x}, t)$ up to a final time of five milliseconds. Plot the two in-plane principal stress amplitudes over the *undeformed* physical geometry at several representative time instances, and plot the inner radius (i.e., $r = r_i$) hoop stress $\sigma_{\theta\theta}$ versus time. Assess the accuracy of your results, and explain your methodology for determining your results are sufficiently accurate.

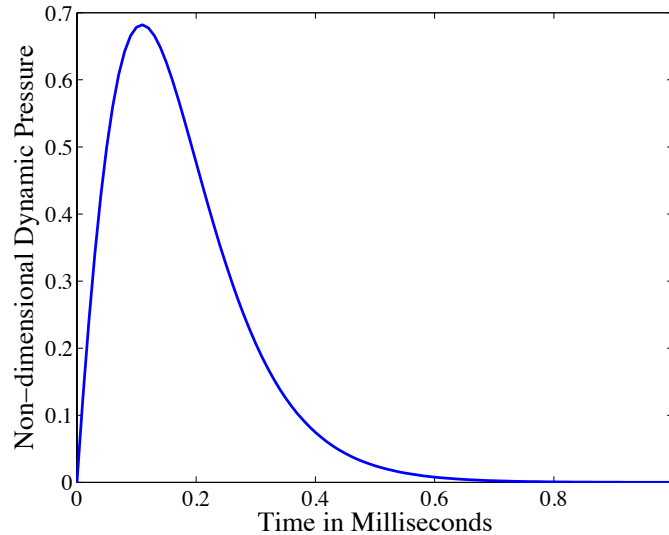


Figure 3: Non-dimensional dynamic pressure field for Problem 3.

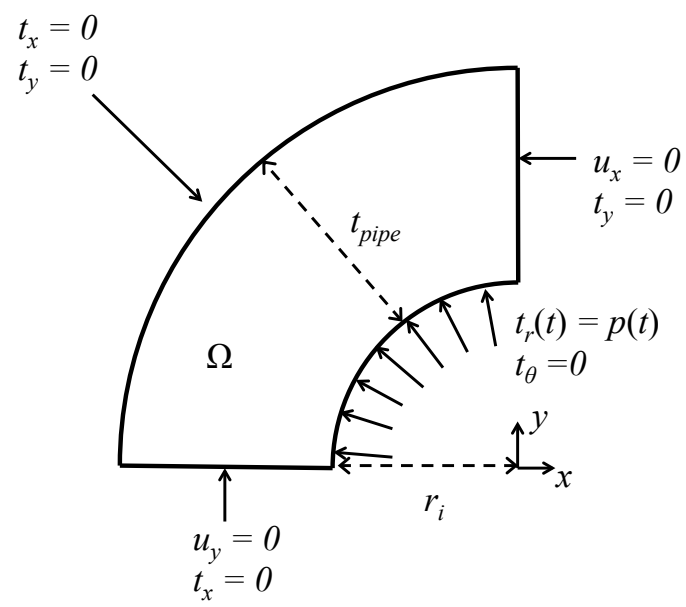


Figure 4: Setup for Problem 3.