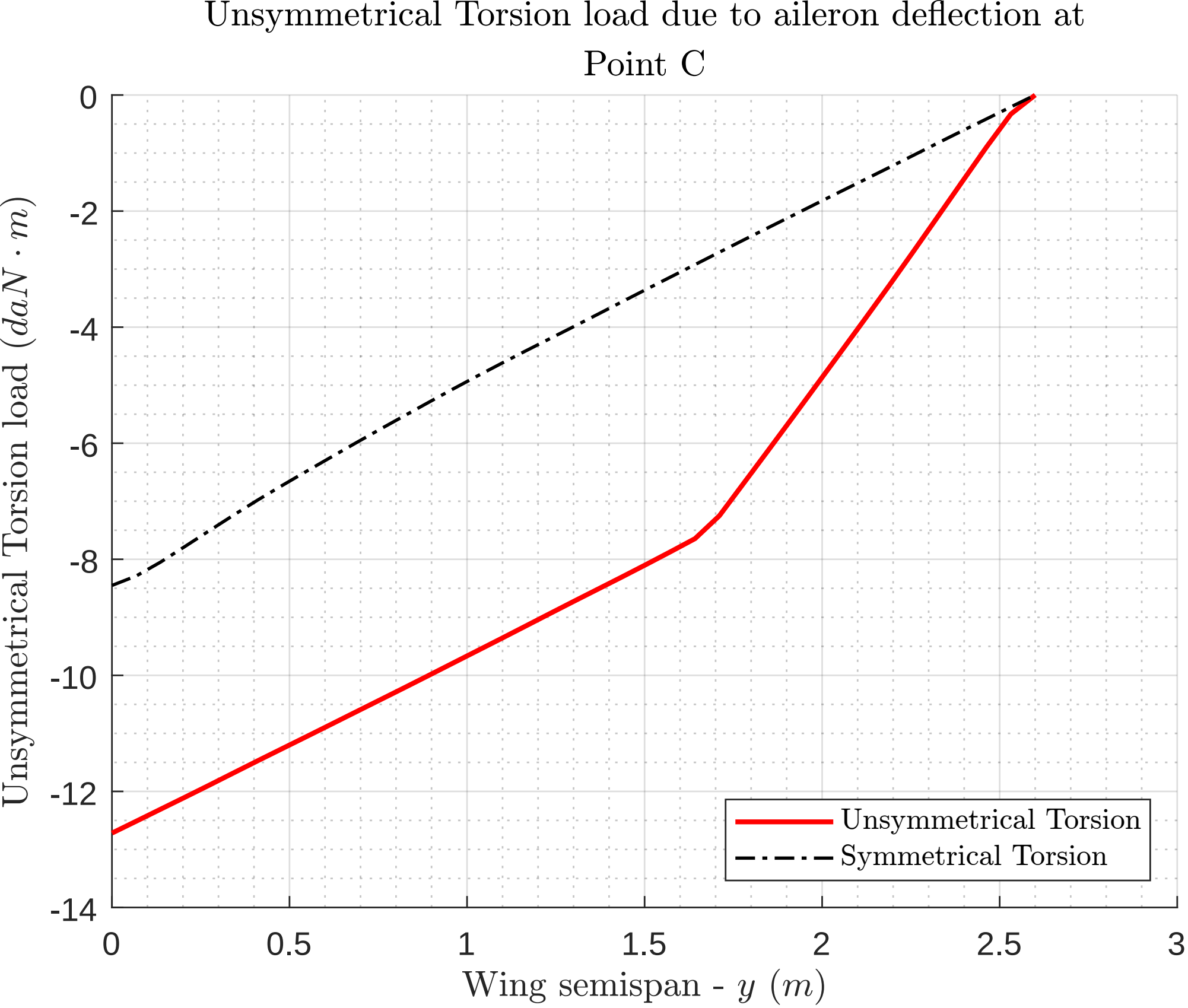
Unsymmetrical Torsion load due to aileron deflection at Point C
Unsymmetrical Torsion load $(daN \cdot m)$
Wing semispan - $y$ $(m)$
0
-2
-4
-6
-8
-10
-12
-14
0
0.5
1
1.5
2
2.5
3
Unsymmetrical Torsion
Symmetrical Torsion