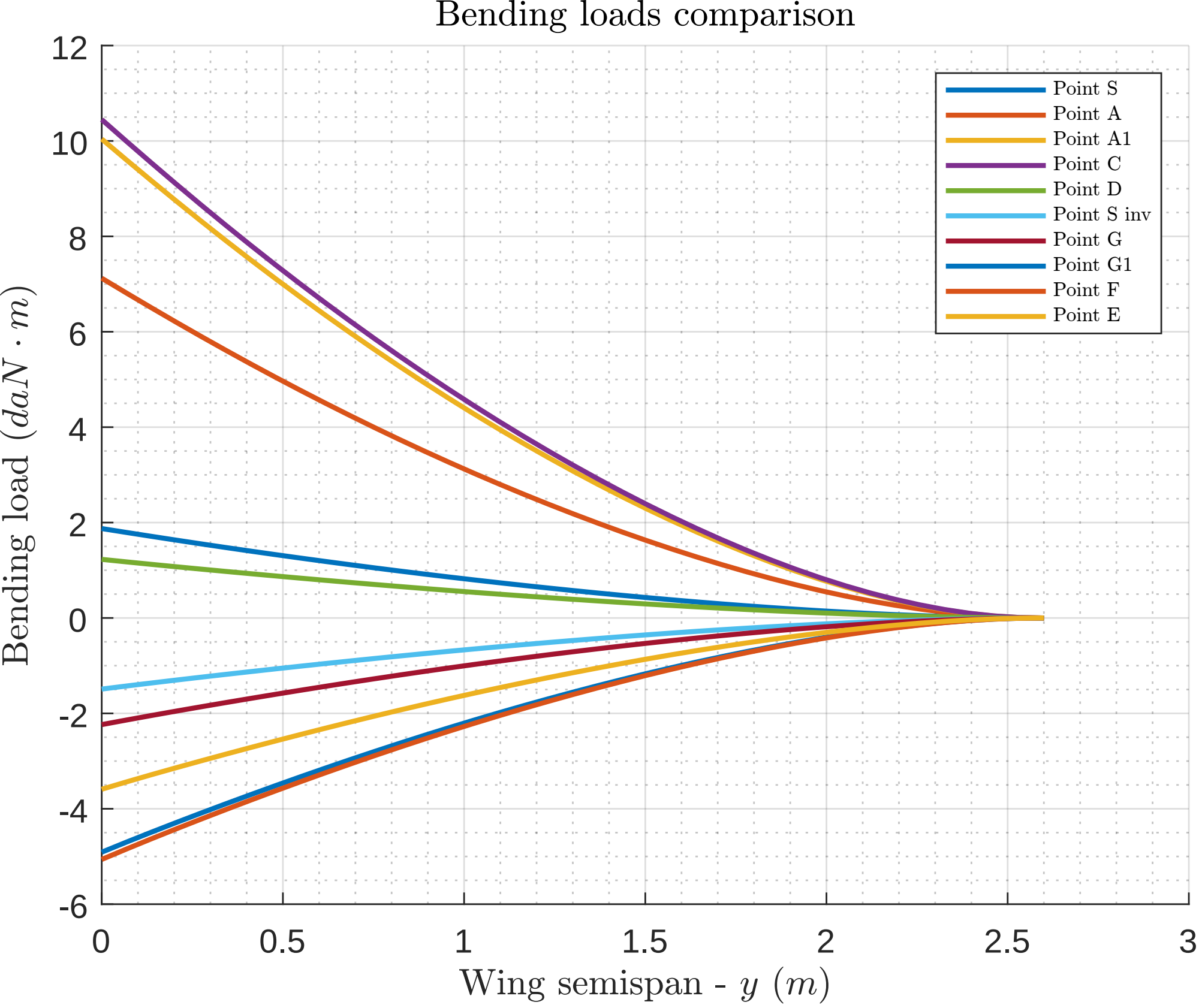
Bending loads comparison
Bending load $(daN \cdot m)$
Wing semispan - y (m)
12
10
8
6
4
2
0
-2
-4
-6
0
0.5
1
1.5
2
2.5
3
Point S
Point A
Point A1
Point C
Point D
Point S inv
Point G
Point G1
Point F
Point E