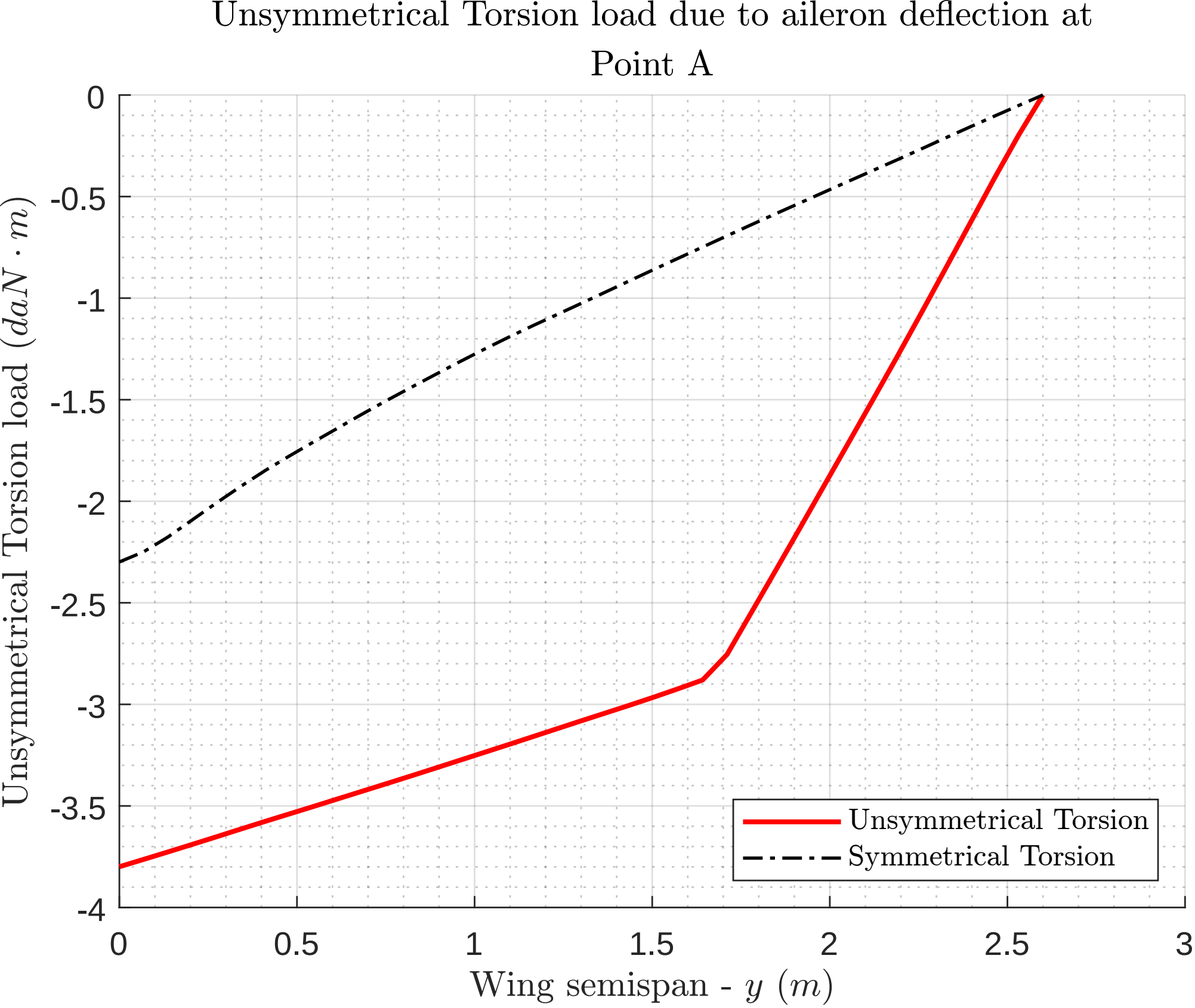
Unsymmetrical Torsion load due to aileron deflection at Point A
Unsymmetrical Torsion load $(daN \cdot m)$
Wing semispan - $y$ $(m)$
0
-0.5
-1
-1.5
-2
-2.5
-3
-3.5
-4
0
0.5
1
1.5
2
2.5
3
Unsymmetrical Torsion
Symmetrical Torsion