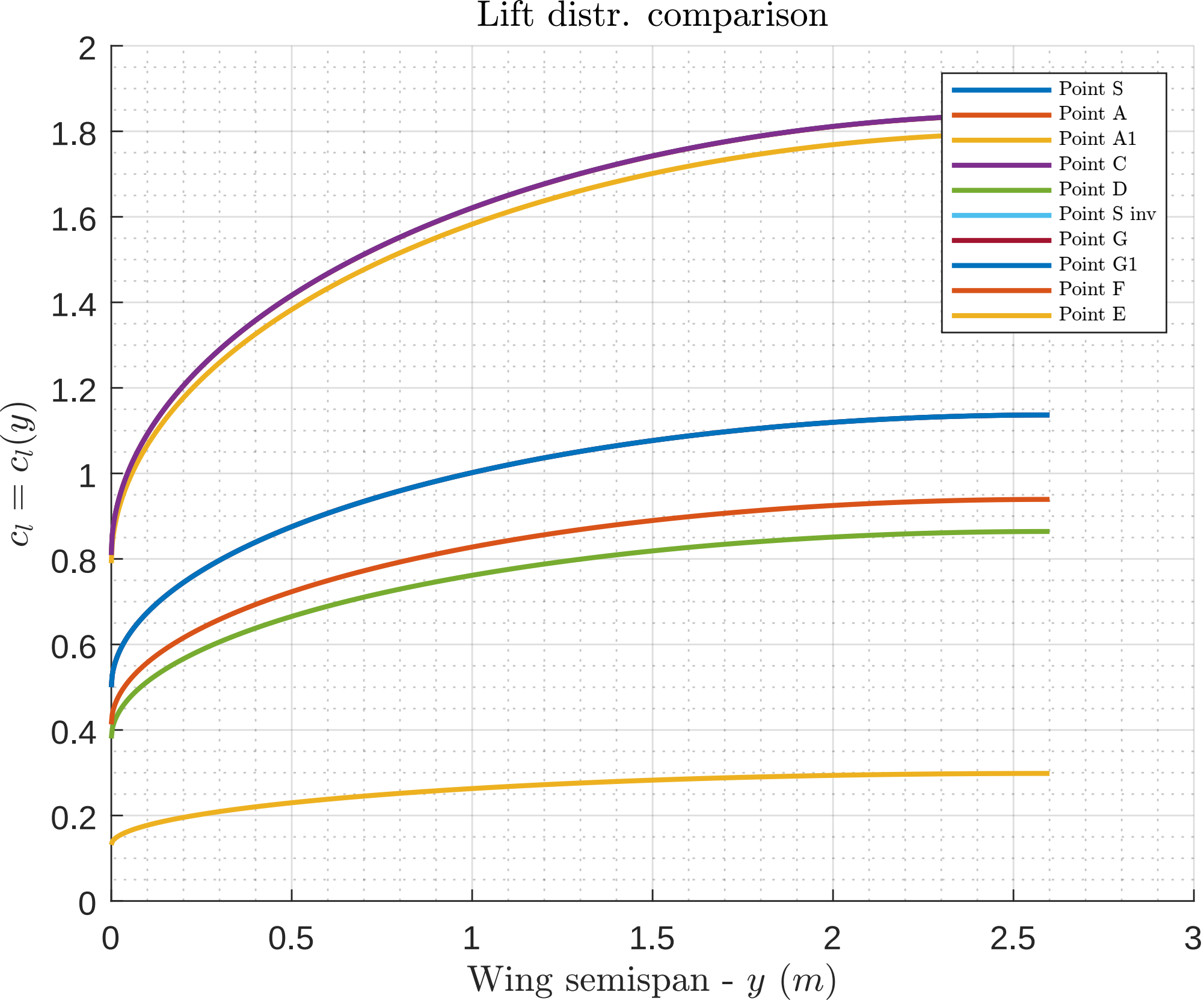
Lift distr. comparison
Point S
Point A
Point A1
Point C
Point D
Point S inv
Point G
Point G1
Point F
Point E
$c_l = c_l(y)$
Wing semispan - y (m)
0
0.2
0.4
0.6
0.8
1
1.2
1.4
1.6
1.8
2
0
0.5
1
1.5
2
2.5
3