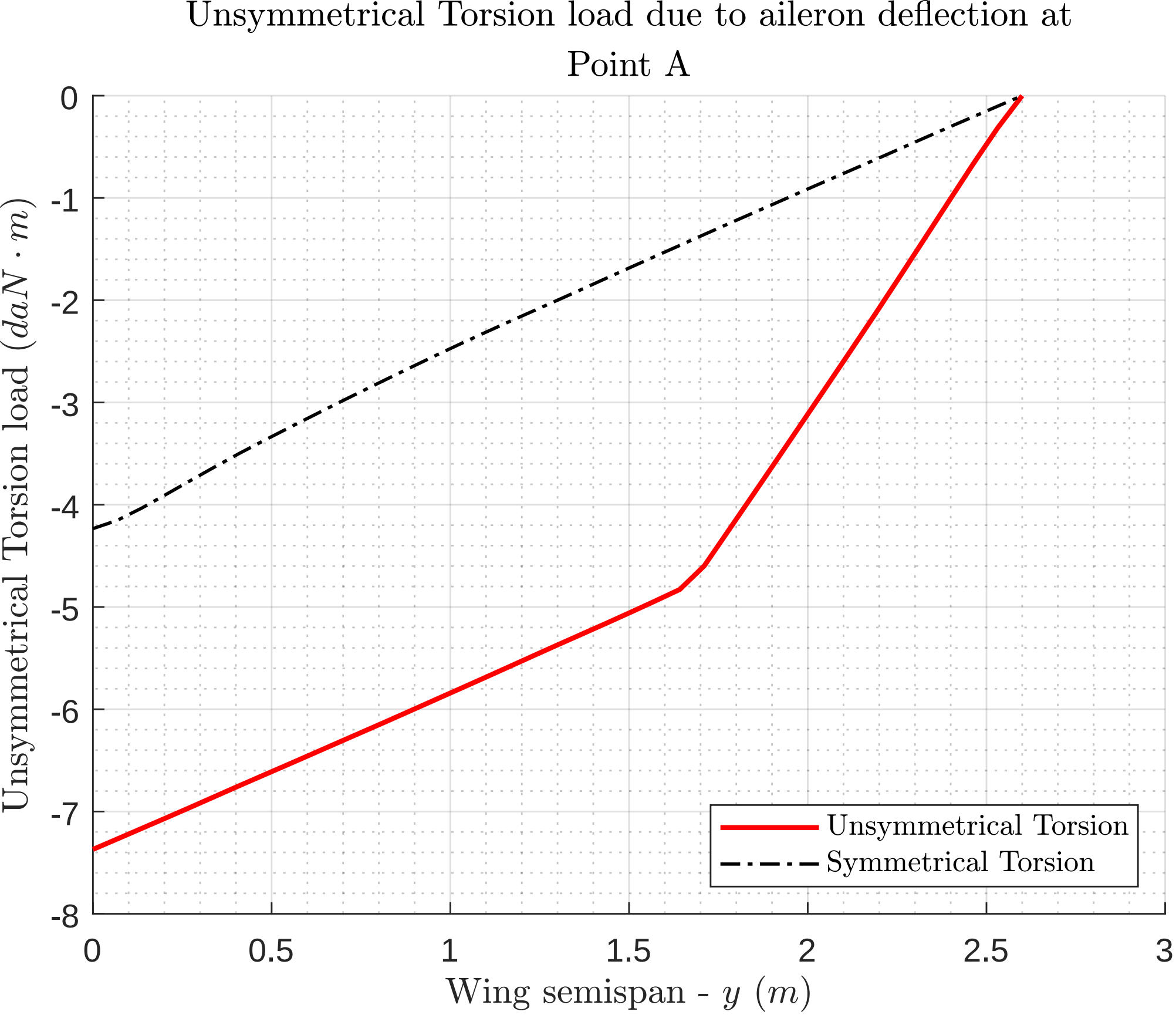
Unsymmetrical Torsion load due to aileron deflection at Point A
Unsymmetrical Torsion load $(daN \cdot m)$
Wing semispan - y (m)
0
-1
-2
-3
-4
-5
-6
-7
-8
0
0.5
1
1.5
2
2.5
3
Unsymmetrical Torsion
Symmetrical Torsion