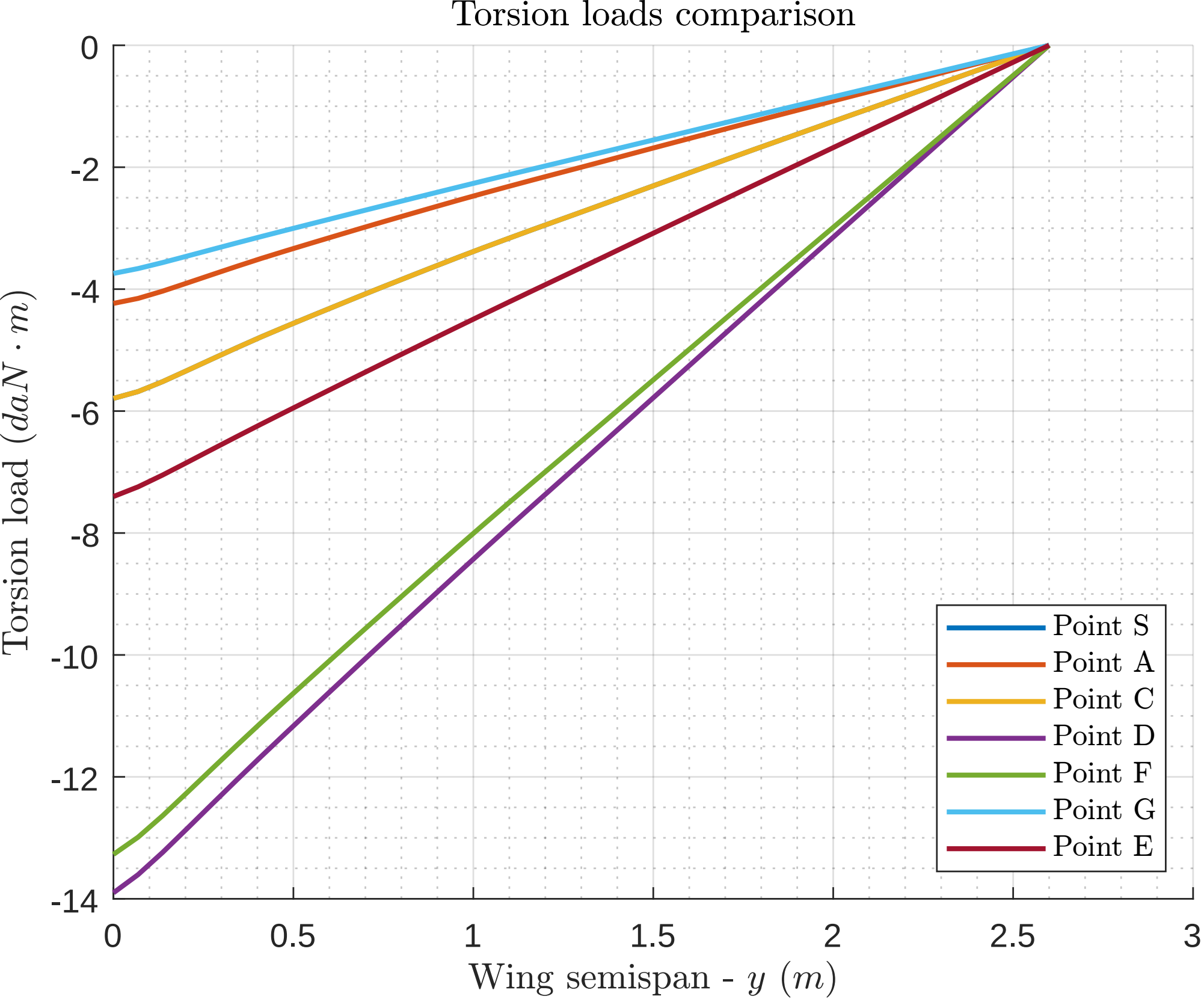
Torsion loads comparison
Torsion load $(daN \cdot m)$
Wing semispan - y (m)
0
-2
-4
-6
-8
-10
-12
-14
0
0.5
1
1.5
2
2.5
3
Point S
Point A
Point C
Point D
Point F
Point G
Point E