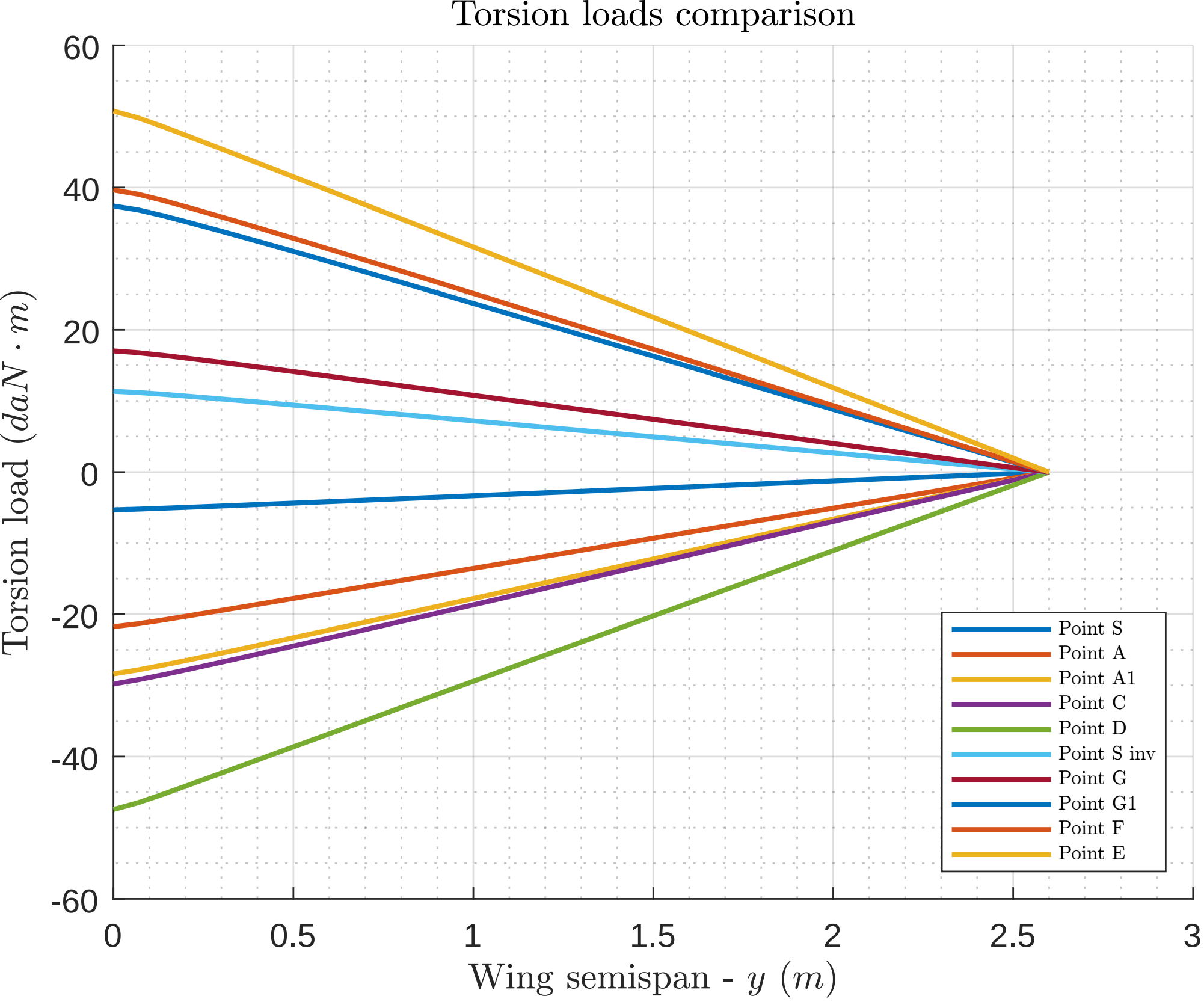
Torsion loads comparison
Torsion load $(daN \cdot m)$
Wing semispan - y (m)
60
40
20
0
-20
-40
-60
0
0.5
1
1.5
2
2.5
3
Point S
Point A
Point A1
Point C
Point D
Point S inv
Point G
Point G1
Point F
Point E