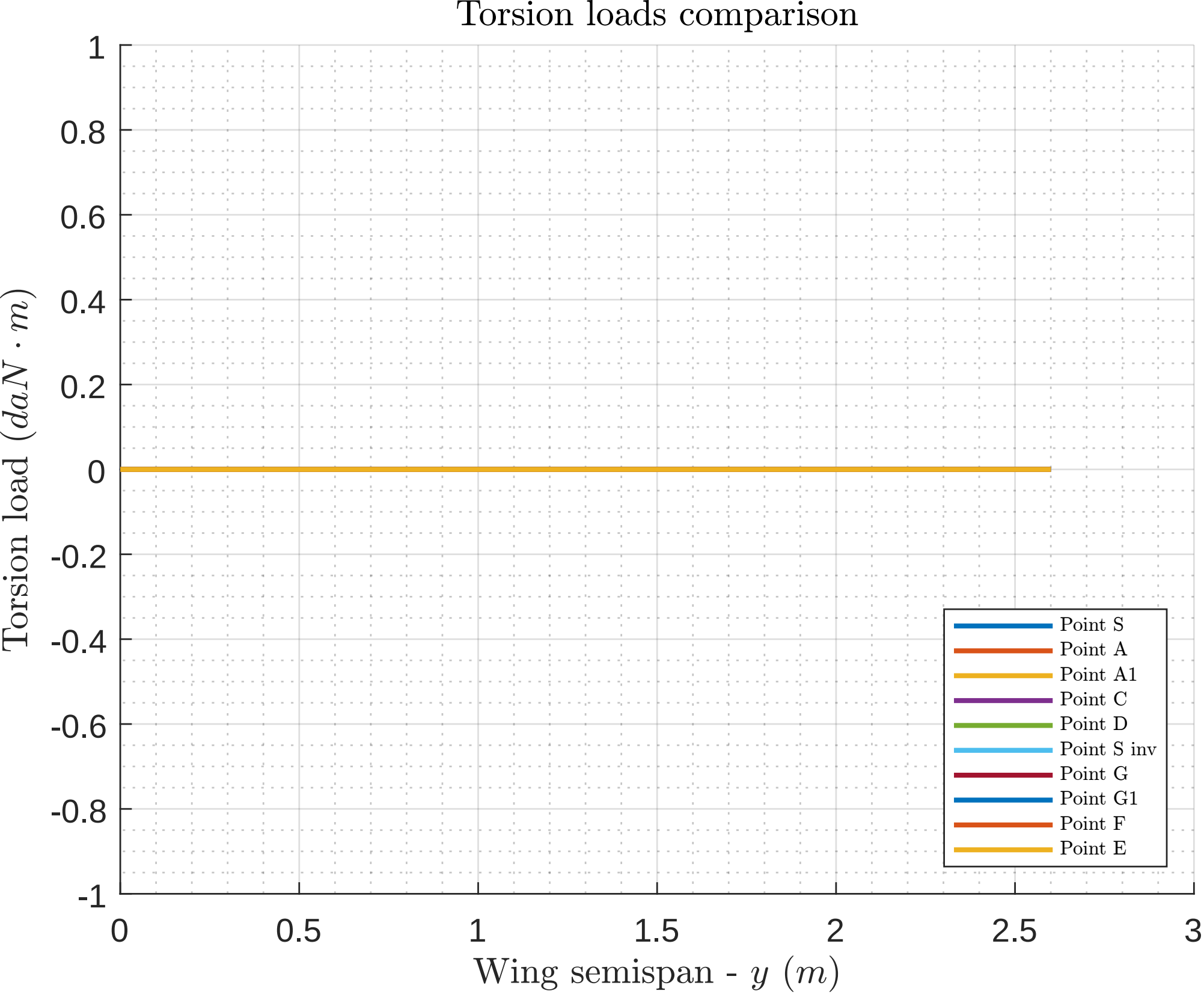
Torsion loads comparison
Torsion load $(daN \cdot m)$
Wing semispan - y (m)
1
0.8
0.6
0.4
0.2
0
-0.2
-0.4
-0.6
-0.8
-1
0
0.5
1
1.5
2
2.5
3
Point S
Point A
Point A1
Point C
Point D
Point S inv
Point G
Point G1
Point F
Point E