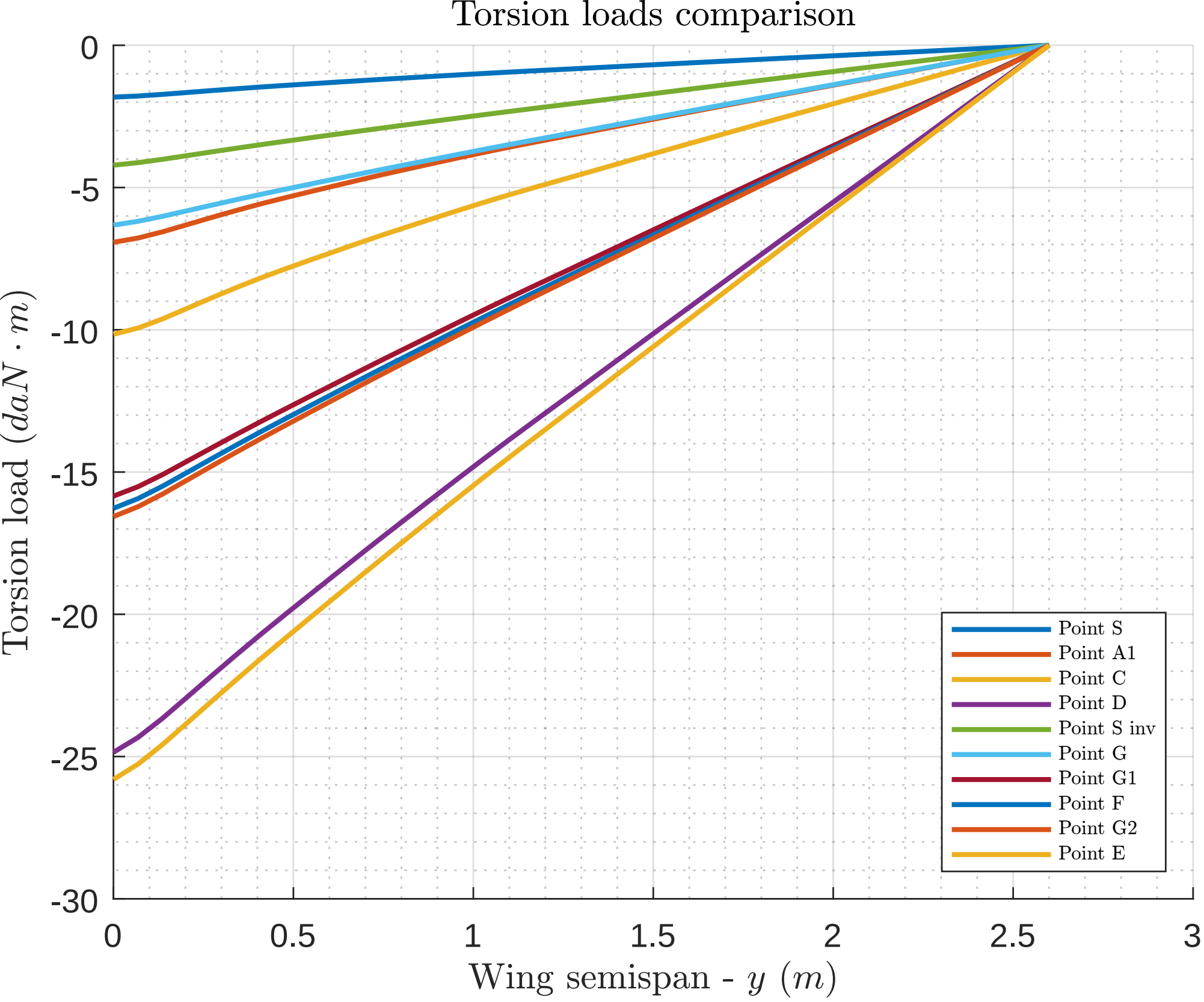
Torsion loads comparison
Torsion load $(daN \cdot m)$
Wing semispan - y (m)
0
-5
-10
-15
-20
-25
-30
0
0.5
1
1.5
2
2.5
3
Point S
Point A1
Point C
Point D
Point S inv
Point G
Point G1
Point F
Point G2
Point E