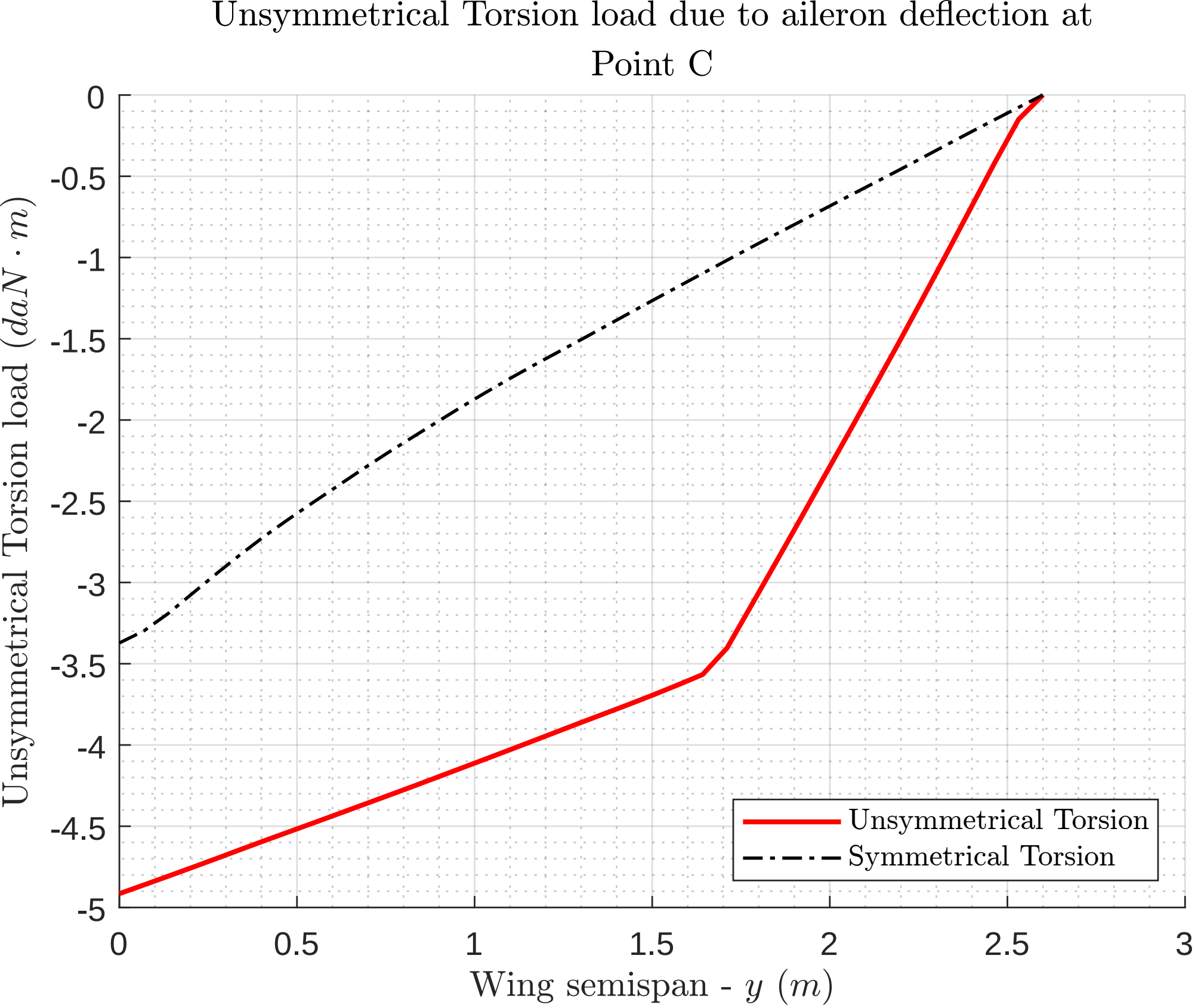
Unsymmetrical Torsion load due to aileron deflection at Point C
Unsymmetrical Torsion load $(daN \cdot m)$
Wing semispan - y (m)
0
-0.5
-1
-1.5
-2
-2.5
-3
-3.5
-4
-4.5
-5
0
0.5
1
1.5
2
2.5
3
Unsymmetrical Torsion
Symmetrical Torsion