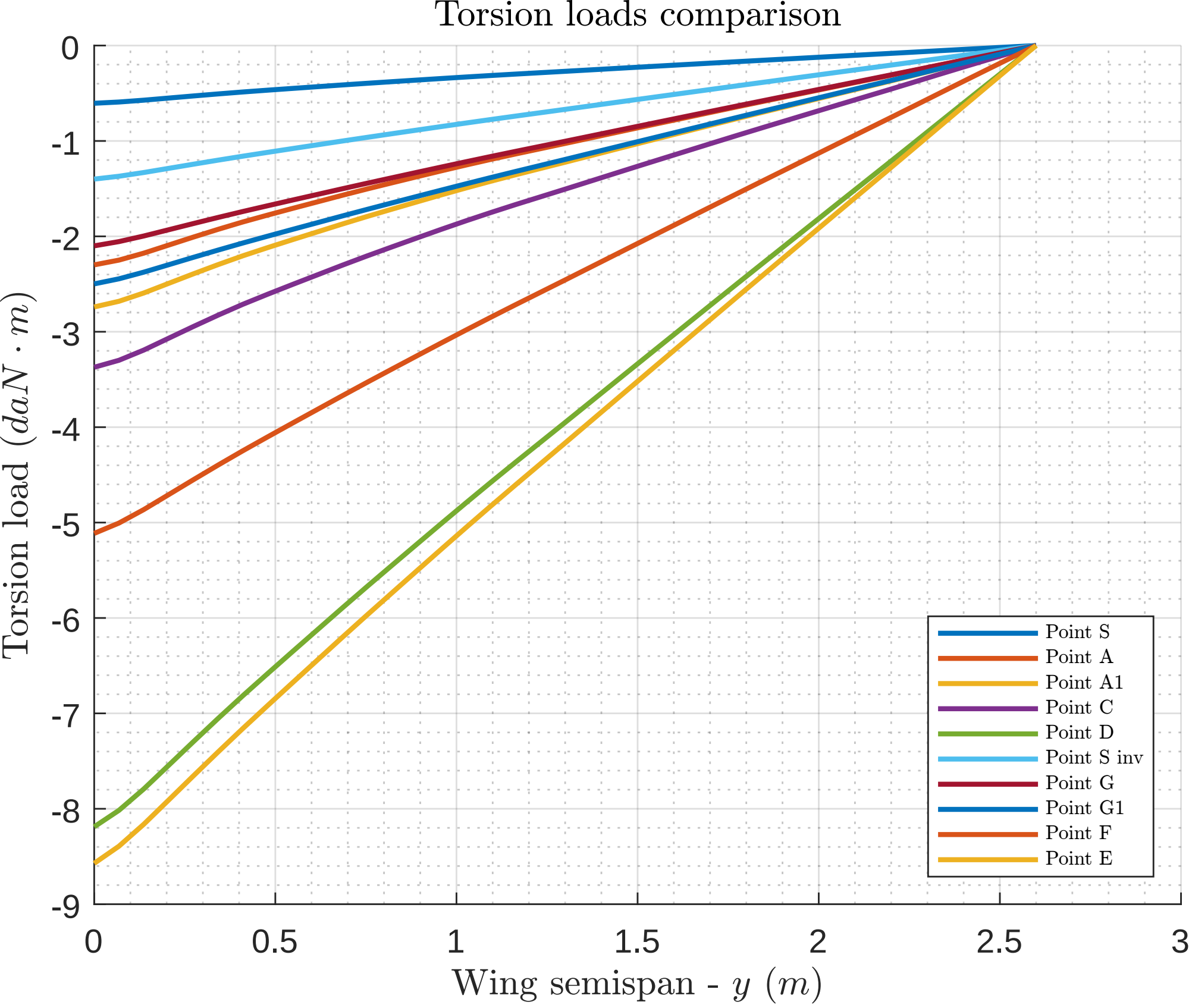
Torsion loads comparison
Torsion load $(daN \cdot m)$
Wing semispan - $y$ $(m)$
0
-1
-2
-3
-4
-5
-6
-7
-8
-9
0
0.5
1
1.5
2
2.5
3
Point S
Point A
Point A1
Point C
Point D
Point S inv
Point G
Point G1
Point F
Point E