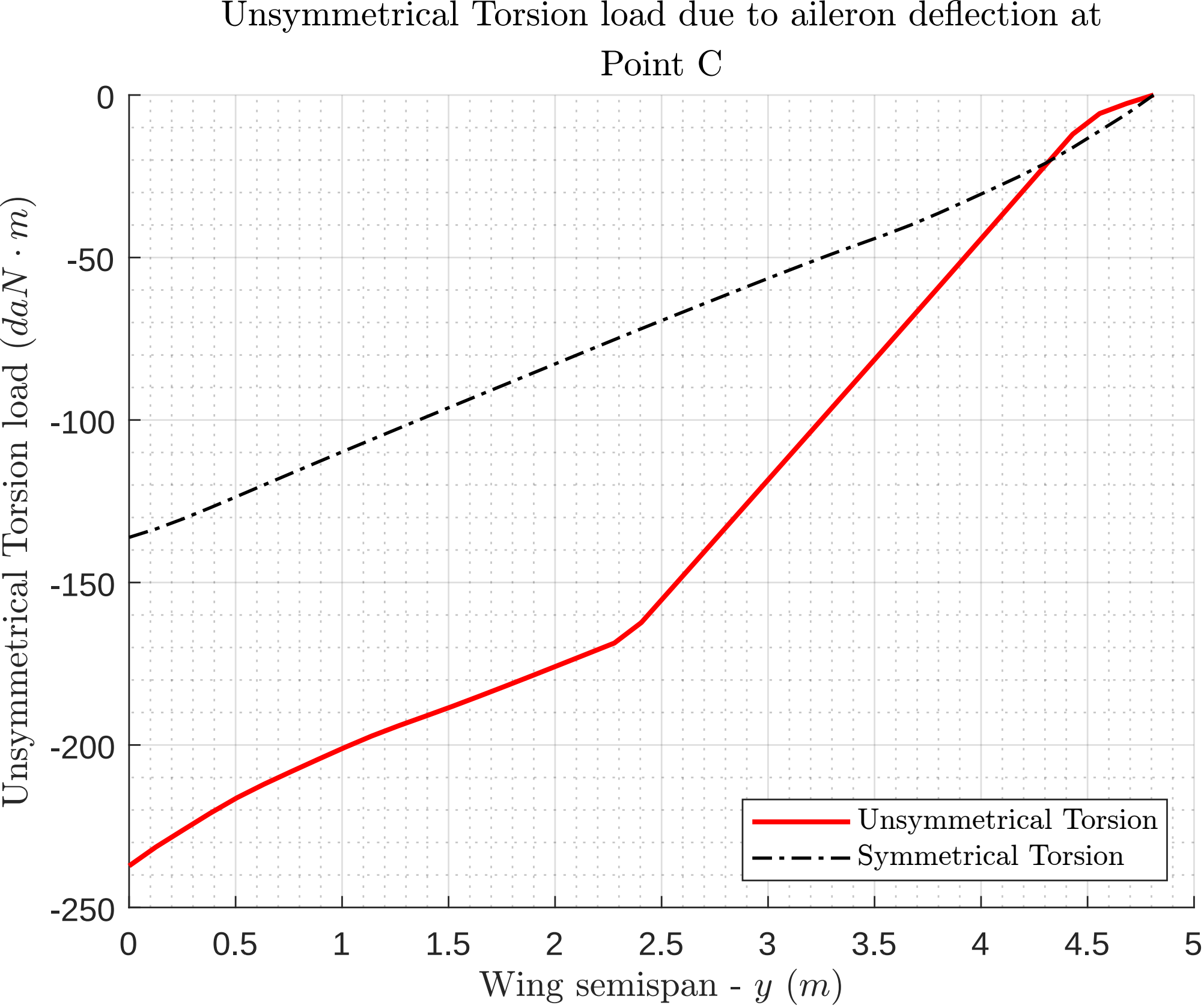
Unsymmetrical Torsion load due to aileron deflection at Point C
Unsymmetrical Torsion load $(daN \cdot m)$
Wing semispan - y (m)
0
-50
-100
-150
-200
-250
0
0.5
1
1.5
2
2.5
3
3.5
4
4.5
5
Unsymmetrical Torsion
Symmetrical Torsion