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```
clear all;
close all;
figure(1);
figure(2);
figure(3);
% Parameters that are used for the Whipple bicycle model. The model is
% based on the linearized 4th order model and analysis of eigenvalues
% from IEEE CSM (25:4) August 2005 pp 26-47

% Basic data is given by 26 parameters
g = 9.81; % Acceleration of gravity [m/s^2]
b = 1.00; % Wheel base [m]
c = 0.08; % Trail [m]
Rrw = 0.35; Rfw = 0.35; % Wheel radii
lambda = pi*70/180; % Head angle [radians]

% Rear frame mass [kg], center of mass [m], and inertia tensor [kgm^2]
mrf=12;xrf=0.439;zrf=0.579;
Jxxrf=0.475656;Jxzrf=0.273996;Jyyrf=1.033092;Jzzrf=0.527436;
mrf=87;xrf=0.491586;zrf=1.028138;
Jxxrf=3.283666;Jxzrf=0.602765;Jyyrf=3.8795952;Jzzrf=0.565929;

% Front frame mass [kg], center of mass [m], and inertia tensor [kgm^2]
mff=2;xff=0.866;zff=0.676;
Jxxff=0.08;Jxzff=-0.02;Jyyff=0.07;Jzzff=0.02;

% Rear wheel mass [kg], center of mass [m], and inertia tensor [kgm^2]
mrw=1.5;Jxxrw=0.07;Jyyrw=0.14;

% Front wheel mass [kg], center of mass [m], and inertia tensor [kgm^2]
mfw=1.5;Jxxfw=0.07;Jyyfw=0.14;

% Auxiliary variables
xrw=0;zrw=Rrw;xfw=b;zfw=Rfw;
Jzzrw=Jxxrw;Jzzfw=Jxxfw;
mt=mrf+mrw+mff+mfw;
xt=(mrf*xrf+mrw*xrw+mff*xff+mfw*xfw)/mt;
zt=(mrf*zrf+mrw*zrw+mff*zff+mfw*zfw)/mt;
Jxxt=Jxxrf+mrf*zrf^2+Jxxrw+mrw*zrw^2+Jxxff+mff*zff^2+Jxxfw+mfw*zfw^2;
Jxzt=Jxzrf+mrf*xrf*zrf+mrw*xrw*zrw+Jxzff+mff*xff*zff+mfw*xfw*zfw;
Jzzt=Jzzrf+mrf*xrf^2+Jzzrw+mrw*xrw^2+Jzzff+mff*xff^2+Jzzfw+mfw*xfw^2;
mf=mff+mfw;
xf=(mff*xff+mfw*xfw)/mf;zf=(mff*zff+mfw*zfw)/mf;
Jxxf=Jxxff+mff*(zff-zf)^2+Jxxfw+mfw*(zfw-zf)^2;
Jxzf=Jxzff+mff*(xff-xf)*(zff-zf)+mfw*(xfw-xf)*(zfw-zf);
Jzzf=Jzzff+mff*(xff-xf)^2+Jzzfw+mfw*(xfw-xf)^2;
d=(xf-b-c)*sin(lambda)+zf*cos(lambda);
F11=mf*d^2+Jxxf*cos(lambda)^2+2*Jxzf*sin(lambda)*cos(lambda)+Jzzf*sin(lambda)^2;
Flx=mf*d*zf+Jxxf*cos(lambda)+Jxzf*sin(lambda);
Flz=mf*d*xf+Jxzf*cos(lambda)+Jzzf*sin(lambda);
gamma=c*sin(lambda)/b;
Sr=Jyyrw/Rrw;Sf=Jyyfw/Rfw;St=Sr+Sf;Su=mf*d+gamma*mt*xt;

% Matrices for linearized fourth order model
M=[Jxxt -Flx-gamma*Jxzt;-Flx-gamma*Jxzt F11+2*gamma*Flz+gamma^2*Jzzt];
```

```
K0=[-mt*g*zt g*Su;g*Su -g*Su*cos(lambda)];  
K2=[0 -(St+mt*zt)*sin(lambda)/b;0 (Su+Sf*cos(lambda))*sin(lambda)/b];  
c12=gamma*St+Sf*sin(lambda)+Jxzt*sin(lambda)/b+gamma*mt*zt;  
c22=Flz*sin(lambda)/b+gamma*(Su+Jzzt*sin(lambda)/b);  
C0=[0 -c12;(gamma*St+Sf*sin(lambda)) c22];  
one=diag([1 1]);null=zeros(2,2);  
  
% Nominal velocity  
v0=5;
```

Part A

$M \ddot{\theta} + C \dot{\theta} + (K_0 + K_2 \dot{\theta}) \theta = [0; T]$; where $\theta = [\gamma; \delta]$ -> angles of bicycle dynamics let $x_1 = \gamma$, $x_2 = \dot{\gamma}$, $x_3 = \delta$, $x_4 = \dot{\delta}$ thus, $\dot{x}_1 = x_2$, $\dot{x}_2 = \ddot{\gamma}$, $\dot{x}_3 = x_4$, $\dot{x}_4 = \ddot{\delta}$ expand matrix equations for 2 eom in terms of state-variables

```
syms x1 x2 x3 x4 x1_d x2_d x3_d x4_d T real
% get coefficients from matrices
m1 = M(1,1); m2 = M(1,2); m3=M(2,1); m4=M(2,2);
c1 = C0(1,1); c2=C0(1,2); c3=C0(2,1); c4=C0(2,2);
K = [K0 + K2*v0^2];
k1 = K(1,1); k2=K(1,2); k3=K(2,1); k4=K(2,2);

% form equations of motion using state-variables

% x2_d = (-m2/m1)*x4_d - (c1*v0/m1)*x2 - (c2*v0/m1)*x4 - (k1/m1)*x1 - (k2/m1)*x3;
% x4_d = (-m3/m4)*x2_d - (c3*v0/m4)*x2 - (c4*v0/m4)*x4 - (k3/m4)*x1 - (k4/m4)*x3 + T/m4;

% rearrange and substitute expression for x2_d into eom of x4_d; eom2
x2_d_eqn = (-m2/m1)*x4_d - (c1*v0/m1)*x2 - (c2*v0/m1)*x4 - (k1/m1)*x1 - (k2/m1)*x3;
eom2 = (-m3/m4)*(x2_d_eqn) - (c3*v0/m4)*x2 - (c4*v0/m4)*x4 - (k3/m4)*x1 - (k4/m4)*x3 + (T/m4) - x4_d;

% solve for x4_d and x2_d, solution denoted with _s
x4_d_s = collect(solve(eom2 == 0, x4_d), [x1; x2; x3; x4; T]);
x2_d_s = collect(subs(x2_d_eqn, x4_d, x4_d_s), [x1; x2; x3; x4; T]);
x1_d_s = x2;
x3_d_s = x4;

% construct state-space presentation
A = [0 1 0 0; ...
     0 0 0 0; ...
     0 0 0 1; ...
     0 0 0 0];

B = [0; 0; 0; 0];

x = [x1; x2; x3; x4];

% put the coefficients from the solution to x2_d, into A,
% and input term into B
```

```

[C_, T_] = coeffs(vpa(x2_d_s));

for i = 1:size(T_,2)
    if T_(i) == x1
        A(2, 1) = C_(i);
    elseif T_(i) == x2
        A(2, 2) = C_(i);
    elseif T_(i) == x3
        A(2, 3) = C_(i);
    elseif T_(i) == x4
        A(2, 4) = C_(i);
    elseif T_(i) == T
        B(2) = C_(i);
    end
end

% put the coefficients from x4_d into A, and input term into B
[C_, T_] = coeffs(vpa(x4_d_s));

for i = 1:size(T_,2)
    if T_(i) == x1
        A(4, 1) = C_(i);
    elseif T_(i) == x2
        A(4, 2) = C_(i);
    elseif T_(i) == x3
        A(4, 3) = C_(i);
    elseif T_(i) == x4
        A(4, 4) = C_(i);
    elseif T_(i) == T
        B(4) = C_(i);
    end
end

% view poles of the state-space model, in Hz
P_A = eig(A)/2/pi;

% findings: v0=5 yields two unstable poles, whilst v0=10 is stable
%
%
% x1_d =      0      1.0000      0      0 * x1 +      0 * T
% x2_d      8.7611 -0.6370 23.2100 2.1486 x2      0.2922
% x3_d      0      0      0      1.0000 x3      0
% x4_d     -14.9477 -17.2465 29.1529 -12.9089 x4      7.9109

```

Part B

$u = -kx$, $\dot{x}_d = Ax + Bu$, $y = Cx \rightarrow \dot{x}_d = (A-Bk)x$ poles of closed loop system = eig(A-Bk) desired poles:

```

Pol = [-2 -10 -1+1i -1-1i];
% required feedback gain, k:
k = acker(A, B, Pol);

%k =
%   -1.2565    5.2073   -32.5252   -0.4145
clp = eig(A-B*k);

```

Part C

```

%
%      u      +      x /-----\ x
%      --> B_f -->() ----->| integral |-----> y=Cx ----> y
%      +|      \-----/ |
%      |      |
%      |<---(A_f = A-Bk)<---|
%
%

```

```

% with state-feedback and input change in delta
% x_d = (A - Bk) [x1 ; x2 ; (x3 + u) ; x4]
% x_d = (A - Bk) x + (A-Bk)[0 ; 0 ; 1 ; 0] u
% x_d = A_f * x + B_f * u

sr = [0;0;1;0]; % selected state for step input
A_f = A-B*k; % state matrix with state-feedback
B_f = A_f*sr; % construct B matrix
C = [0;0;1;0]'; % view only delta on the output
D = 0;
sys_delta = ss(A_f,B_f,C,D);

% apply a step input with size of 0.002, plot response of each state
% using the step() function
opt = stepDataOptions;
opt.StepAmplitude = 0.002; % step amplitude set
t = (0:0.005:10); % time set
figure(1); hold on;
step(sys_delta, t, opt); % response on delta
sys_gamma = ss(A_f,B_f,[1,0,0,0],D); % response on gamma
sys_gamma_d = ss(A_f,B_f,[0,1,0,0],D); % response on gamma_d
step(sys_gamma, t, opt);
sys_delta_d = ss(A_f,B_f,[0,0,0,1],D); % response on delta_d
step(sys_delta_d, t, opt);
hold off;

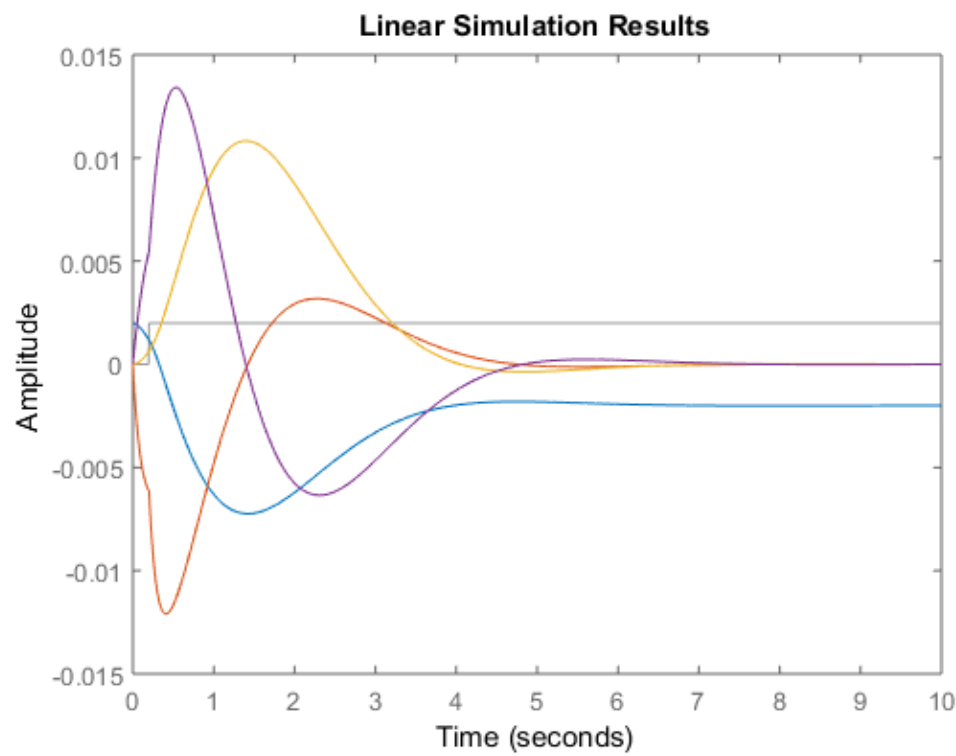
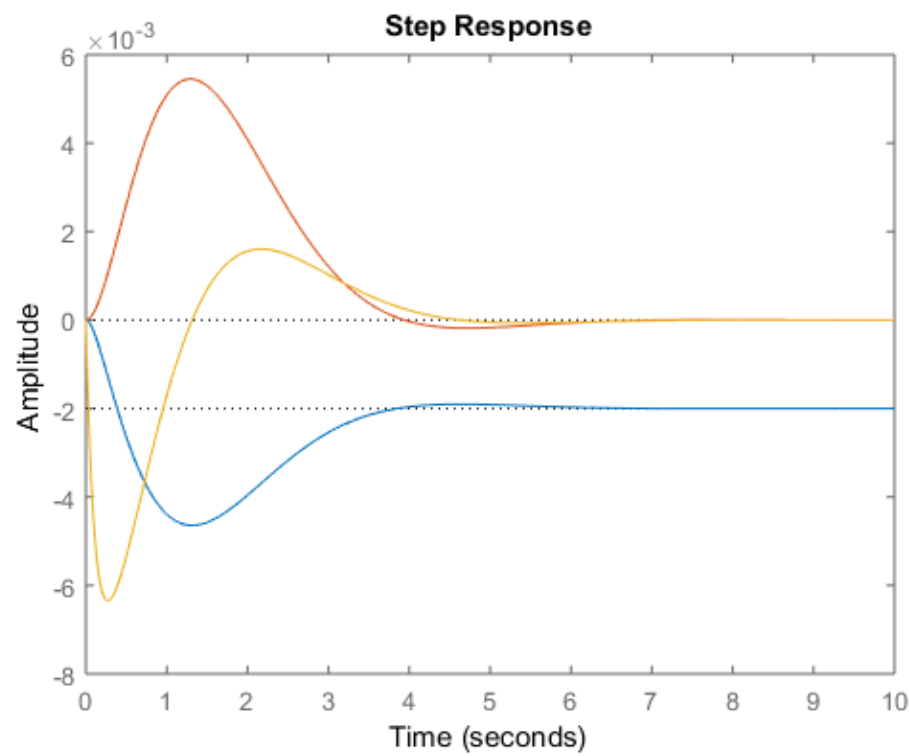
% model using lsim, providing an initial condition for delta = 0.002

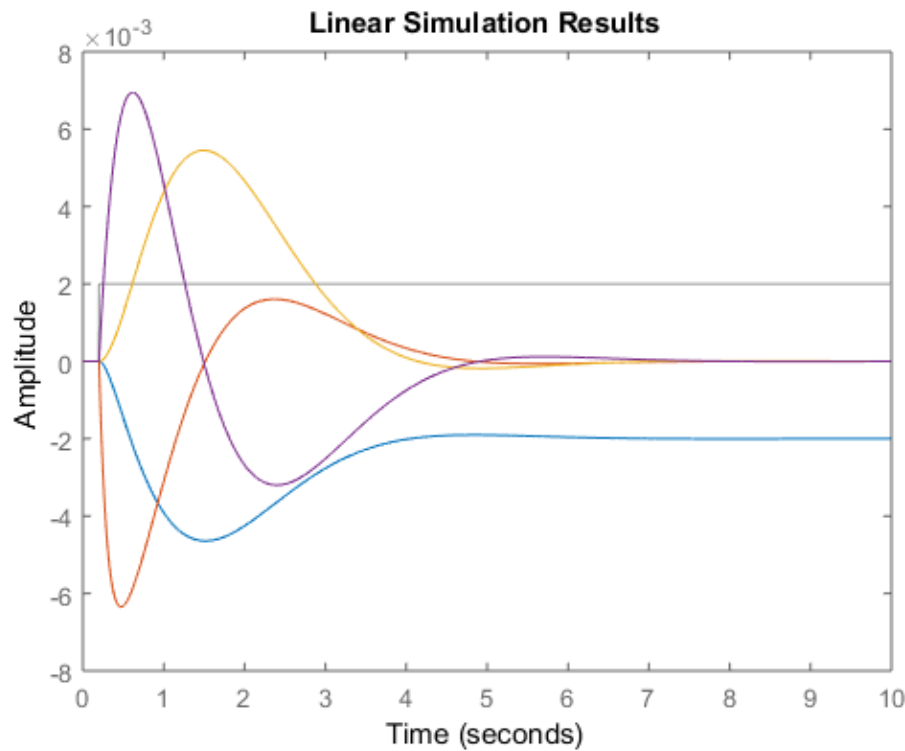
% construct a step input, u
u = [];
for n = 1:size(t,2)
    if t(n) < 0.2
        u(n) = 0;
    else
        u(n) = 0.002;
    end
end

figure(2); hold on;
lsim(sys_delta, u, t, [0;0;0.002;0]);
lsim(sys_delta_d, u, t, [0;0;0.002;0]);
lsim(sys_gamma, u, t, [0;0;0.002;0]);
lsim(sys_gamma_d, u, t, [0;0;0.002;0]);
hold off;

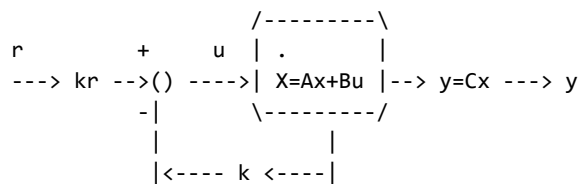
% without the initial condition, can be seen that the same as
% using the step function
figure(3); hold on;
lsim(sys_delta, u, t, [0;0;0;0]);
lsim(sys_delta_d, u, t, [0;0;0;0]);
lsim(sys_gamma, u, t, [0;0;0;0]);
lsim(sys_gamma_d, u, t, [0;0;0;0]);
% lsim(syslqrcl, u, t);
hold off;

```





Part C - 2.0



insert a reference for state (delta)

```
% x_d = (A-Bk)*x + (kr*B)*r
% x_d = A_star*x + B_star*r
% where r is a reference step input for delta

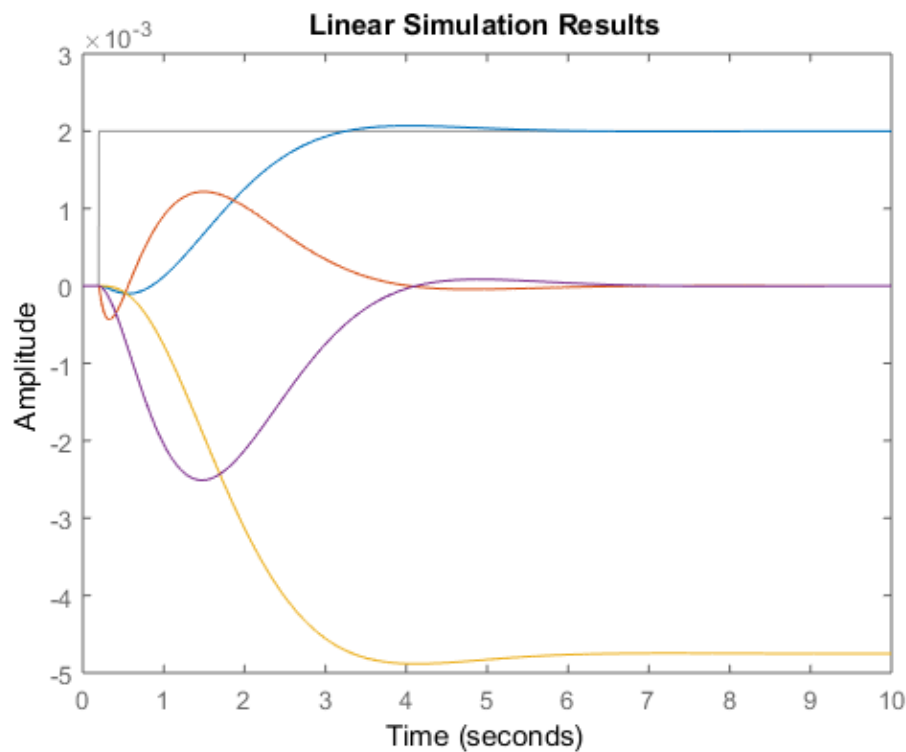
A_star = A-B*k;
C = [0,0,1,0]; %design kr for only delta on output
D = 0;
kr = -(C*(A-B*k)^-1 *B)^-1;%-(rscale(ss(A-B*k,B,C,D),k) - 1.68); %equal to -0.541 (rscale calculates Kr)
B_star = kr*B;

% apply a step input with size of 0.002, plot response of each state
% using the step() function
t = (0:0.005:10); % time set
figure(6); hold on;
sys_delta = ss(A_star,B_star,C,D);
sys_gamma = ss(A_star,B_star,[1,0,0,0],D); % response on gamma
sys_gamma_d = ss(A_star,B_star,[0,1,0,0],D); % response on gamma_d
sys_delta_d = ss(A_star,B_star,[0,0,0,1],D); % response on delta_d

lsim(sys_delta, u, t);
lsim(sys_delta_d, u, t);
lsim(sys_gamma, u, t);
lsim(sys_gamma_d, u, t);
% lsim(syslqrcl, u, t);

hold off;
%Nbar = rscale(ss(A-B*k,B,C,D),k)
```

```
%aa = ss(A-B*k, -(Nbar-1.68)*B, C, D)
%figure(7); lsim(aa, u, t)
```



Part D

```
P = [-8 -16 -4+1i -4-1i];           % pole locations for observer
L = place(A', C', P)';               % poles at eig(A-LC);

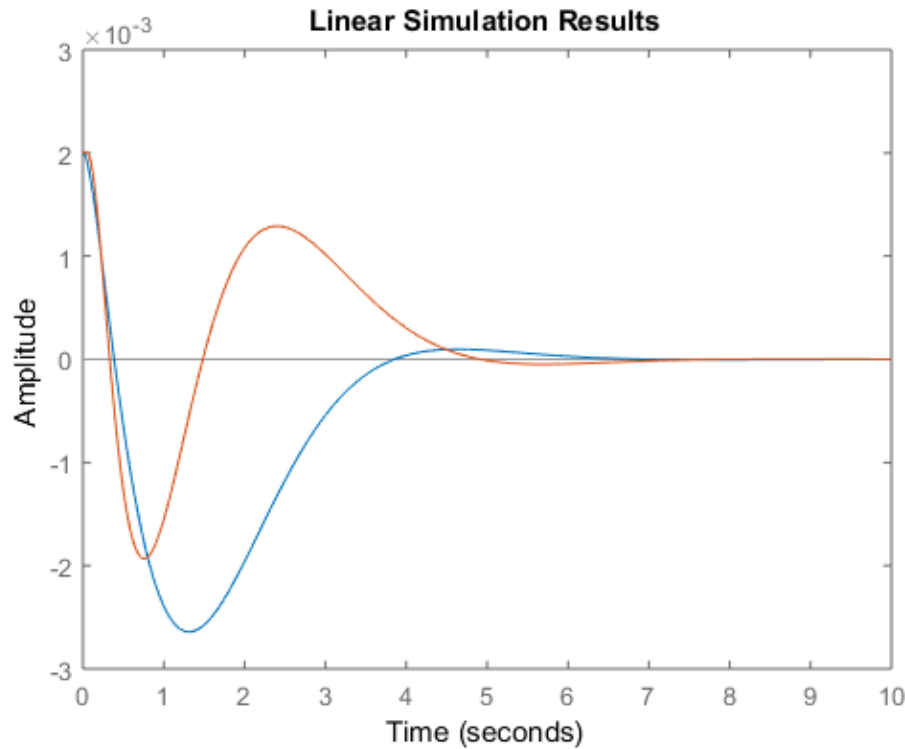
% state space model for complete state-feedback and observer

A_p = [A -B*k; L*C A-B*k-L*C];
B_p = [B*k_r; B*k_r];
C_p = [C zeros(size(C))];
C_p_hat = [zeros(size(C)) C];

sys_p = ss(A_star, B_star, C, D);    % system which outputs plant delta
sys_p_hat = ss(A_p, B_p, C_p, 0);    % system which outputs observer delta

figure(8); hold on;
no_input = zeros(size(t));           % no step input

%show response for initial disturbance on plant delta
lsim(sys_p, no_input, t, [0, 0, 0.002, 0]);
lsim(sys_p_hat, no_input, t, [0, 0, 0.002, 0, zeros(1,4)]);
hold off;
```

Part E

to show the effect of observer poles on replicating the plant Pol is a matrix containing the poles of the state-feedback component of the plant

```
% make the observer poles a multiple of the plant observes, my multiplying
% Pol by an element in co before evaluating L.
co = [2, 5, 10, 20];

% For each multiple of Pol, plot the response against the plant for an
% initial state offset -> delta by 0.002

for i = 1:4
    % outputting delta
    C = [0,0,1,0];
    % place poles of observer at a multiple of Pol
    L = place(A', C', co(i)*Pol)'; % poles at eig(A-LC);
    A_p = [A -B*k; L*C A-B*k-L*C]; % observer A
    B_p = [B*k_r; B*k_r]; % observer B
    C_p = [C zeros(size(C))]; % observer C

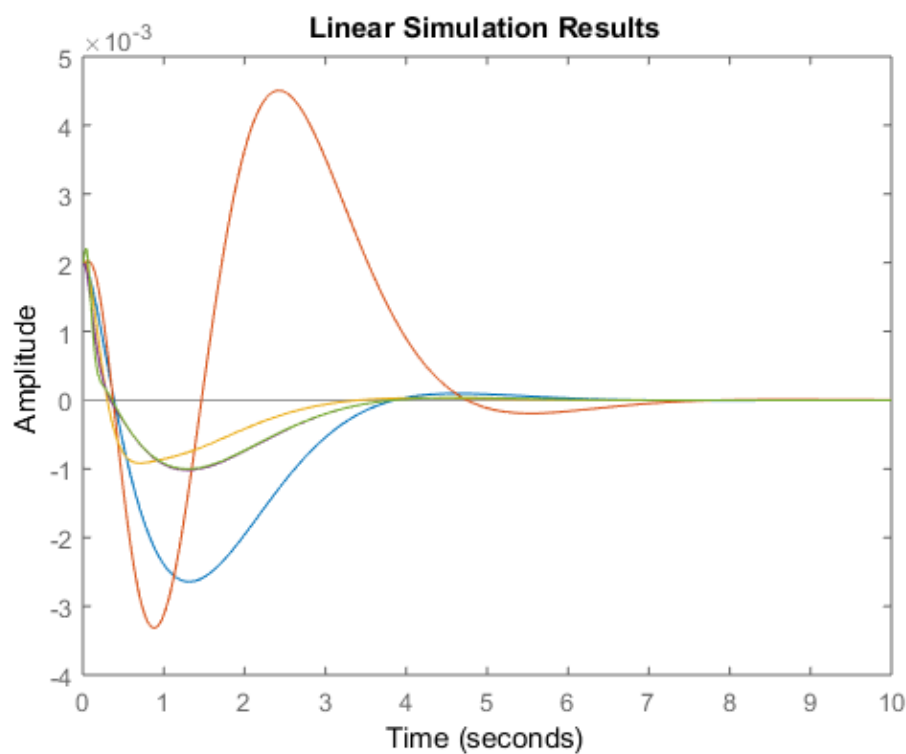
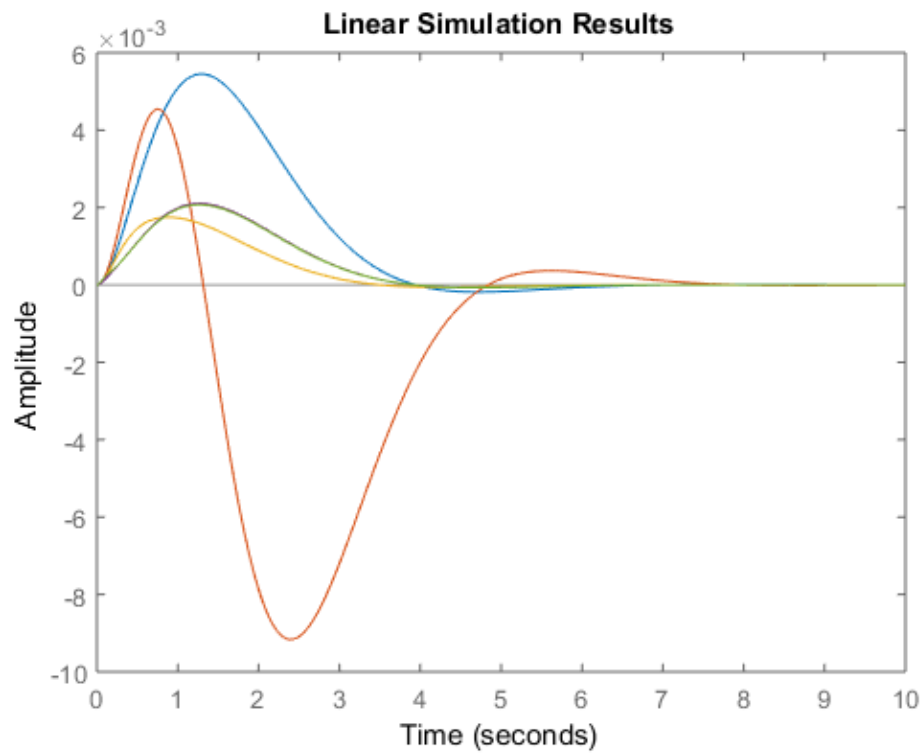
    % retrieve the state-space model for an output of delta only
    if i == 1
        sys_delta = ss(A_star, B_star, C, D); % system which outputs plant delta
    end
    sys_delta_hat = ss(A_p, B_p, C_p, 0); % system which outputs observer delta
    % retrieve the state-space model for an output of phi
    C = [1,0,0,0];
    C_p_hat = [C, zeros(size(C))];
    if i == 1
        sys_phi = ss(A_star, B_star, C, D); % system which outputs plant delta
    end
    sys_phi_hat = ss(A_p, B_p, C_p_hat, 0); % system which outputs observer delta
    figure(9); hold on;
    % plot response on phi
    if i == 1
        % plot the plant response, only once
        lsim(sys_phi, no_input, t, [0, 0, 0.002, 0]);
    end
    lsim(sys_phi_hat, no_input, t, [0, 0, 0.002, 0, zeros(1,4)]);
end
```

```

hold off;
figure(10); hold on;
% plot reponse on delta
if i == 1
    % plot the plant response, only once
    lsim(sys_delta, no_input, t, [0, 0, 0.002, 0]);
end
lsim(sys_delta_hat, no_input, t, [0, 0, 0.002, 0, zeros(1,4)]);
hold off;

```

```
end
```

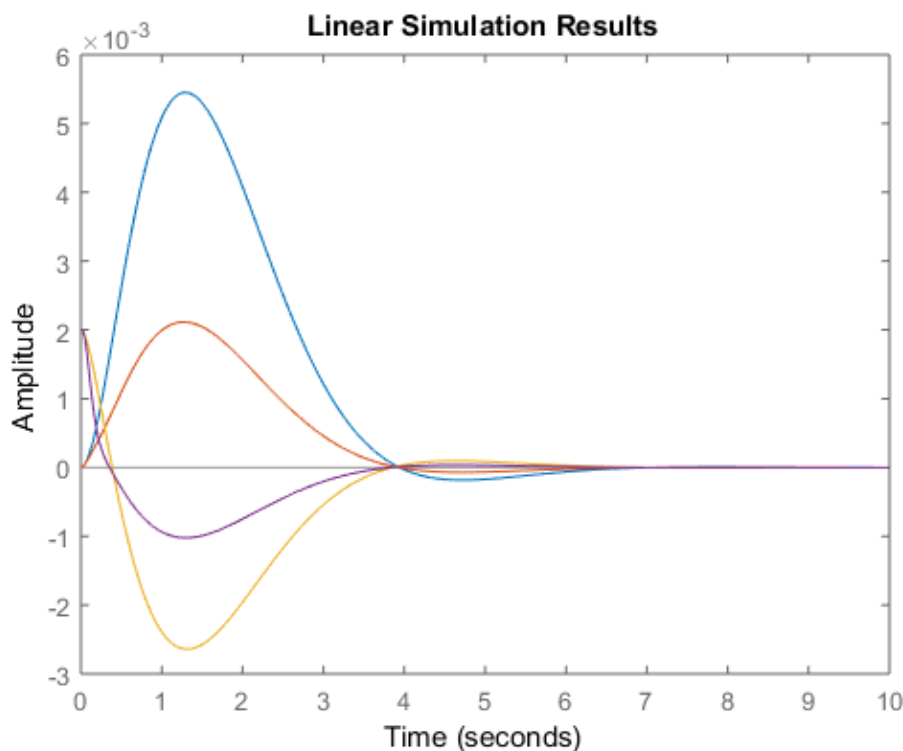


```

i = 1;
% outputing delta
C = [0,0,1,0];
% place poles of observer at a multiple of Pol
L = place(A', C', [-15+7.5i -15-7.5i -20+10i -20-10i]); % poles at eig(A-LC);
A_p = [A -B*k; L*C A-B*k-L*C]; % observer A
B_p = [B*k; B*k]; % observer B
C_p = [C zeros(size(C))]; % observer C

% retrieve the state-space model for an output of delta only
if i == 1
    sys_delta = ss(A_star, B_star, C, D); % system which outputs plant delta
end
sys_delta_hat = ss(A_p, B_p, C_p, 0); % system which outputs observer delta
% retrieve the state-space model for an output of phi
C = [1,0,0,0];
C_p_hat = [C, zeros(size(C))];
if i == 1
    sys_phi = ss(A_star, B_star, C, D); % system which outputs plant delta
end
sys_phi_hat = ss(A_p, B_p, C_p_hat, 0); % system which outputs observer delta
figure(11); hold on;
% plot response on phi
if i == 1
    % plot the plant response, only once
    lsim(sys_phi, no_input, t, [0, 0, 0.002, 0]);
end
lsim(sys_phi_hat, no_input, t, [0, 0, 0.002, 0, zeros(1,4)]);
hold off;
figure(11); hold on;
% plot reponse on delta
if i == 1
    % plot the plant response, only once
    lsim(sys_delta, no_input, t, [0, 0, 0.002, 0]);
end
lsim(sys_delta_hat, no_input, t, [0, 0, 0.002, 0, zeros(1,4)]);
hold off;

```



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