# Evaluacion1

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# 1 Realizacion de codigo

En el codigo se configuro como lo pide la actividad de la evaluación, usar el método de Euler para ilustrar el comportamiento de un proyectil en una grafica realizada por Gnuplot, para el cual ya se nos daban valores especificos para ese comportamiento, solo cambiando el anguro en el que este era lanzado, al igual qu realicar graficas donde al proyectil le influia, la fricción y donde no era afectada por esta, con angulos de 15, 30, 45, 60 y 75 grados. Para el programa dado por la actividad se crearon diferentes archivos con los respectivos datos con fricción, sin friccion y con los diferentes angulos para que arrojara las siguentes graficas:

## 1.1 codigo de programa

#### Program projectile2

```
!-----
! Realistic projectile motion with air resistance
! method: program may call various ODU solvers
   key = 0 modified Euler
  key = 1 Runge-Kutta 4th order
  key = 2 code Rkf45 (Runge-Kutta 4th-5th order)
! written by: Alex Godunov
! last revision: October 2006
1-----
! input from a file (self explanatory)
   see file cannon.dat
! output ...
   to a file named by a user
1-----
    implicit none
    Real*8 d1x, d2x, d1y, d2y, ti, tf
    Real*8 xi(2), xf(2), yi(2), yf(2)
    character output*20, tabla*20
    real*8 g, v0, angle, dt, C, rho, Rp, Mp, yrho, u
    real*8 rad, CdO, energy, energyO, xc, yc, vxc, vyc
```

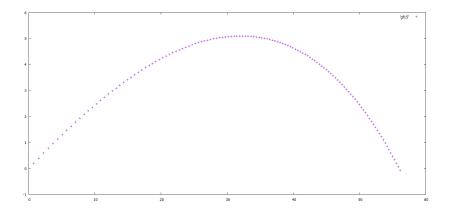


Figure 1: con fricción a 15 grados

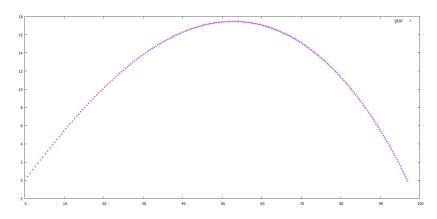


Figure 2: con fricción a 30 grados

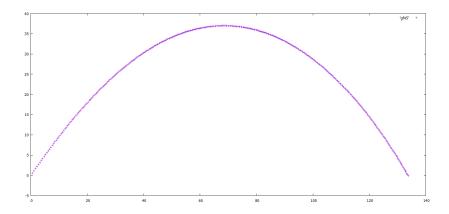


Figure 3: con fricción a 45 grados

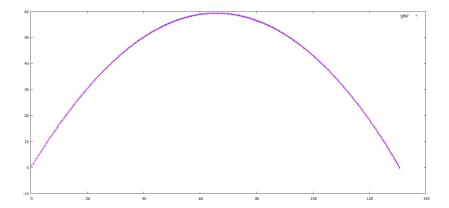


Figure 4: con fricción a 60 grados

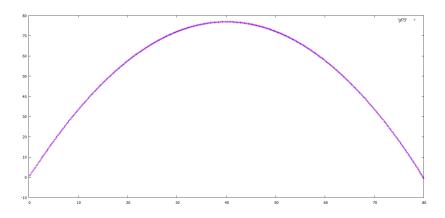


Figure 5: con fricción a 75 grados

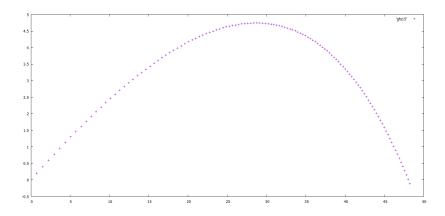


Figure 6: sin fricción a 15 grados

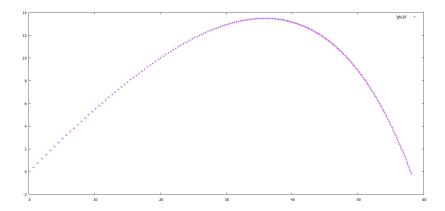


Figure 7: sin fricción a 30 grados

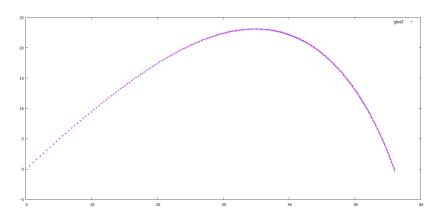


Figure 8: sin fricción a 45 grados

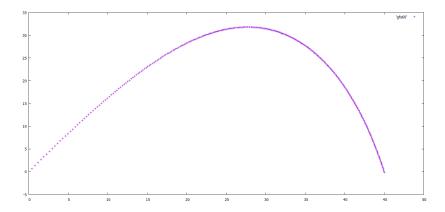


Figure 9: sin fricción a 60 grados

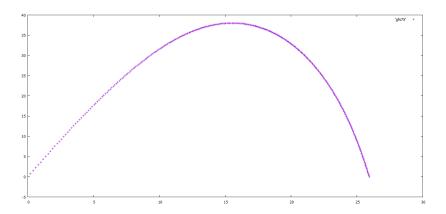


Figure 10: sin fricción a 75 grados

```
real*8 xfly(5000), yfly(5000), xrange
      integer*4 i, j, key, jmax
      integer iflag, iwork(5), ne
     real*8 y(4), relerr, abserr, work(27)
     parameter (rad=3.1415926/180.0, jmax=5000)
     parameter (relerr=1.0e-9, abserr=0.0)
      common/const/ Cd0, g, yrho
      !external d1x, d2x, d1y, d2y, cannon
      !*** read initial data from a file
      print*, " dame el nombre del archivo"
     read 201, output
      read 201, tabla
      open (unit=7,file=output)
     read(7, 202) key
     read(7,203) g
     read(7, 203) xi(1)
     read(7, 203) yi(1)
     read(7, 203) v0
     read(7, 203) angle
     read(7, 203) dt
     read(7,203) C
     read(7,203) rho
     read(7,203) Rp
     read(7, 203) Mp
     read(7,204) yrho
     read(7, 203) u
```

```
!*** end reading and set initial time to 0.0
     ti = 0.0
!*** end initial data
     xi(2) = v0*cos(angle*rad)
     yi(2) = v0*sin(angle*rad)
! CdO is the air resistance coefficient /Mp projectile
     Cd0 = C*rho*3.141592*Rp**2/Mp
! energyO is the initial energy of the projectile
! later energy is calculated that is printed as a fraction of energy0
! if there is no frictional forces the energy must be conserved
     energy0= Mp*g*yi(1) + 0.5*Mp*(xi(2)**2+yi(2)**2)
      open(unit=8,file=tabla,status='unknown')
!write(7,210)
     write(7,211) xi(1), yi(1)
!*** loop over time till the projectile hits the ground
! {\tt rkf45} initial data and conditions for {\tt rkf45} and first call
       it is very important to call rkf45 for the first time with
       iflag = 1 (otherwise the code does not run)
     if(key.eq.2) then
  ne = 4
  iflag = 1
  y(1) = xi(1)
  y(2) = yi(1)
  y(3) = xi(2)
  y(4) = yi(2)
     end if
!*** loop till the projectile hits the ground i.e. yf=y1
     do while (yf(1).gt.-0.01)
       j = j+1
       tf = ti + dt
       if(key.eq.0) call euler22m(ti,tf,xi,xf,yi,yf)
        if(key.eq.1) call rk4_d22(d1x,d2x,d1y,d2y,ti,tf,xi,xf,yi,yf)
       if(key.eq.2) then
     call rkf45(cannon,ne,y,ti,tf,relerr,abserr,iflag,work,iwork)
          ! xf(1)=y(1)
```

```
! yf(1)=y(2)
    !xf(2)=y(3)
    !yf(2)=y(4)
    if(iflag.eq.7) iflag = 2
  end if
        energy = Mp*g*yf(1) + 0.5*Mp*(xf(2)**2+yf(2)**2)
        energy = energy/energy0
        xfly(j) = xf(1)/u
  yfly(j) = yf(1)/u
        write(8, 211) xf(1)/u, yf(1)/u
!* TEST section
! good test for the code: no air resistance
! then one may compare with analytic solution
       xc = 0.0 + v0*cos(angle*rad)*tf
        yc = 0.0 + v0*sin(angle*rad)*tf-0.5*g*(tf)**2
       vxc= v0*cos(angle*rad)
       vyc= v0*sin(angle*rad)-g*(tf)
! remove comment from the next line to print
      write(7, 211) tf,xf(1)/xc,yf(1)/yc,xf(2)/vxc,yf(2)/vyc,energy
! preparation for the next step
        ti = tf
        do i=1,2
            xi(i) = xf(i)
            yi(i) = yf(i)
         end do
!*** max number of time steps is 2000
if(j.ge.jmax) exit
      end do
!*** calculate max range (using linear interpolation on the last two points)
      xrange = xfly(j-1)
      xrange = xrange+(xfly(j)-xfly(j-1))*yfly(j-1)/(yfly(j-1)-yfly(j))
      !write (7, 213) xrange
201
     format (a12)
202
     format (i5)
203 format (f10.4)
204 format (e10.2)
210 format(7x,'X',11x,'Y')
211
     format (f8.2, 4f12.3,1pe12.3)
212
     format (' Iflag from Rkf45 = ',i2,' -> increase time step')
     format (/, 'Range is =',f12.3)
      contains
```

### end program projectile2

```
Function d1x(t,x,y)
!-----
! function dx/dt
    implicit none
    Real*8 d1x, t, x(2), y(2)
    d1x = x(2)
    return
  end Function d1x
    Function d1y(t,x,y)
!-----
! function dy/dt
ļ-----
    implicit none
    Real *8 d1y, t, x(2), y(2)
    d1y = y(2)
    return
  end Function D1y
    Function d2x(t,x,y)
!-----
! function d2x/dt2
!----
    implicit none
    Real*8 d2x, t, x(2), y(2), Cd0, g, v, yrho
    common/const/ Cd0, g, yrho
    v = sqrt(x(2)**2+y(2)**2)
    d2x = (-1.0)*(Cd0*exp(-y(1)/yrho))*v*x(2)
    return
  end Function d2x
    Function d2y(t,x,y)
1-----
! function d2y/dt2
!----
    implicit none
    Real*8 d2y, t, x(2), y(2), Cd0, g, v, yrho
    common/const/ Cd0, g, yrho
    v = sqrt(x(2)**2+y(2)**2)
```

```
return
   end Function d2v
     subroutine cannon(t, y, yp)
! first and second derivatives for rkf45
! definition of the differential equations
! y(1) = x yp(1) = vx = y(3)
! y(2) = y yp(2)=vy=y(4)
! y(3) = vx 	 yp(3)=d2x/dt2 = - Cd*v*vx
! y(4) = vy 	 yp(4) = d2y/dt2 = -g - Cd*v*vy
1-----
     implicit none
     Real*8 t, y(4), yp(4), Cd0, g, v, yrho
     common/const/ Cd0, g, yrho
     yp(1) = y(3)
     yp(2) = y(4)
! equation of motion
     v = sqrt(y(3)**2+y(4)**2)
 yp(3) = (-1.0)*(Cd0*exp(-y(2)/yrho))*v*y(3)
yp(4) = (-1.0)*(g + (Cd0*exp(-y(2)/yrho))*v*y(4))
     return
   end subroutine cannon
     Subroutine euler22m(ti,tf,xi,xf,yi,yf)
|-----
! euler22m.f: Solution of the second-order 2D ODE
!method:
         modified Euler (predictor-corrector)
! written by: Alex Godunov
! last revision: 21 October 2006
!-----
! input ...
! d1x(t,x,y)- function dx/dt (supplied by a user)
! d2x(t,x,y)- function d2x/dt2 (supplied by a user)
! d1y(t,x,y) - function dy/dt (supplied by a user)
! d2y(t,x,y)- function d2y/dt2 (supplied by a user)
    where x(2) and y(2) (x(1)-position, x(2)-speed, etc.)
! ti - initial time
! tf - time for a solution
! xi(2) - initial position and speed for x component
! yi(2) - initial position and speed for y component
! output ...
! xf(2) - solutions (x position and speed) at point tf
```

d2y = (-1.0)\*(g + (Cd0\*exp(-y(1)/yrho))\*v\*y(2))

```
! yf(2) - solutions (y position and speed) at point tf
implicit none
     Real*8 d1x, d2x, d1y, d2y, ti, tf
     Real*8 xi(2), xf(2), yi(2), yf(2)
     Real*8 h,t, x1, x2, y1, y2
     Real*8 k1x(2), k2x(2), k3x(2), k4x(2), k1y(2), k2y(2), k3y(2), k4y(2)
     h = tf-ti
     t = ti
!*** Euler
     xf(1) = xi(1) + h*d1x(t,xi,yi)
     xf(2) = xi(2) + h*d2x(t,xi,yi)
     yf(1) = yi(1) + h*d1y(t,xi,yi)
     yf(2) = yi(2) + h*d2y(t,xi,yi)
!*** modified Euler
     xf(1) = xi(1) + (d1x(t,xi,yi)+d1x(t,xf,yf))*0.5*h
     xf(2) = xi(2) + (d2x(t,xi,yi)+d2x(t,xf,yf))*0.5*h
     yf(1) = yi(1) + (d1y(t,xi,yi)+d1y(t,xf,yf))*0.5*h
     yf(2) = yi(2) + (d2y(t,xi,yi)+d2y(t,xf,yf))*0.5*h
     Return
   End Subroutine Euler22m
```