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Vector Problems on Finite Elements

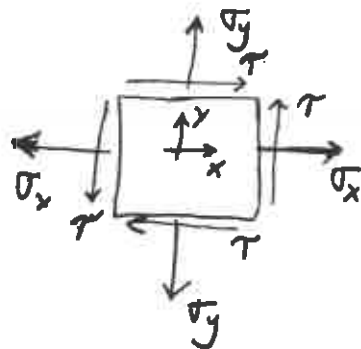
- multiple "unknowns" ... "degrees-of-freedom" exist at each node
- must enforce multiple Galerkin Equations at each node to balance ... generally coupled sets of eqns
- Key... breakdown vector equations into scalar components, apply FE techniques to each scalar equations

e.g. Stress Balance

$$\nabla \cdot \underline{\underline{\sigma}} = \underline{\underline{f}}$$

$$\underline{\underline{\sigma}} \equiv \begin{bmatrix} \sigma_x & \tau \\ \tau & \sigma_y \end{bmatrix}$$

"Stress Tensor"



Breakdown into Cartesian Components:

$$\hat{x} : \frac{\partial \sigma_x}{\partial x} + \frac{\partial \tau}{\partial y} = f_x$$

$$\hat{y} : \frac{\partial \tau}{\partial x} + \frac{\partial \sigma_y}{\partial y} = f_y$$

Apply weighted residual approach to equations

Alternately... keep in vector form; manipulate w/ vector identities (i.e. "integration by parts"); then breakdown at the final stage

$$\text{i.e. } \langle \nabla \cdot \underline{\underline{\sigma}} \phi_i \rangle = \langle \underline{\underline{f}} \phi_i \rangle$$

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$$- \langle \underline{\sigma} \cdot \nabla \phi_i \rangle + \underbrace{\oint \underline{\sigma} \cdot \hat{n} \phi_i ds}_{\text{stress on boundary}} = \langle \underline{B} \phi_i \rangle$$

X-component:

$$\left\langle \sigma_x \frac{\partial \phi_i}{\partial x} \right\rangle + \left\langle \tau \frac{\partial \phi_i}{\partial y} \right\rangle = \hat{x} \cdot \underbrace{\left[\oint \underline{\sigma} \cdot \hat{n} \phi_i ds - \langle \underline{B} \phi_i \rangle \right]}_{R_{x_i}}$$

y-component

$$\left\langle \tau \frac{\partial \phi_i}{\partial x} \right\rangle + \left\langle \sigma_y \frac{\partial \phi_i}{\partial y} \right\rangle = \hat{y} \cdot \underbrace{\left[\oint \underline{\sigma} \cdot \hat{n} \phi_i ds - \langle \underline{B} \phi_i \rangle \right]}_{R_{y_i}}$$

Constitutive Relations: $\underline{\sigma}$ related to $u, v \dots$ displacements in x, y directions

- Plane Stress (linear elasticity)

$$\begin{Bmatrix} \sigma_x \\ \sigma_y \\ \tau \end{Bmatrix} = \frac{E}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix} \begin{Bmatrix} \frac{\partial u}{\partial x} \\ \frac{\partial v}{\partial y} \\ \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \end{Bmatrix}$$

- Plane Strain

$$\begin{Bmatrix} \sigma_x \\ \sigma_y \\ \tau \end{Bmatrix} = \frac{E(1-\nu)}{(1+\nu)(1-2\nu)} \begin{bmatrix} 1 & \frac{\nu}{1-\nu} & 0 \\ \frac{\nu}{1-\nu} & 1 & 0 \\ 0 & 0 & \frac{1-2\nu}{2(1-\nu)} \end{bmatrix} \begin{Bmatrix} \frac{\partial u}{\partial x} \\ \frac{\partial v}{\partial y} \\ \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \end{Bmatrix}$$

E, ν material properties

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Work Through Plane Stress case:

$$\text{We have ... } \langle \sigma_x \frac{\partial \phi_i}{\partial x} \rangle + \langle \tau \frac{\partial \phi_i}{\partial y} \rangle = R_{x_i}$$

$$\langle \tau \frac{\partial \phi_i}{\partial x} \rangle + \langle \sigma_y \frac{\partial \phi_i}{\partial y} \rangle = R_{y_i}$$

$$\Rightarrow \left\langle \frac{E}{1-\nu^2} \left(\frac{\partial u}{\partial x} + \nu \frac{\partial v}{\partial y} \right) \frac{\partial \phi_i}{\partial x} \right\rangle + \left\langle \frac{E}{2(1+\nu)} \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) \frac{\partial \phi_i}{\partial y} \right\rangle = \bar{R}_{x_i}$$

$$\left\langle \frac{E}{1-\nu^2} \left(\nu \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right) \frac{\partial \phi_i}{\partial y} \right\rangle + \left\langle \frac{E}{2(1+\nu)} \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) \frac{\partial \phi_i}{\partial x} \right\rangle = \bar{R}_{y_i}$$

For convenience... assume E, ν constant... multiply through

$$\left\langle \frac{\partial u}{\partial x} \frac{\partial \phi_i}{\partial x} + \frac{1-\nu}{2} \frac{\partial u}{\partial y} \frac{\partial \phi_i}{\partial y} \right\rangle + \left\langle \nu \frac{\partial v}{\partial y} \frac{\partial \phi_i}{\partial x} + \frac{1-\nu}{2} \frac{\partial v}{\partial x} \frac{\partial \phi_i}{\partial y} \right\rangle = \bar{R}_{x_i}$$

$$\left\langle \nu \frac{\partial u}{\partial x} \frac{\partial \phi_i}{\partial y} + \frac{1-\nu}{2} \frac{\partial u}{\partial y} \frac{\partial \phi_i}{\partial x} \right\rangle + \left\langle \frac{\partial v}{\partial y} \frac{\partial \phi_i}{\partial y} + \frac{1-\nu}{2} \frac{\partial v}{\partial x} \frac{\partial \phi_i}{\partial x} \right\rangle = \bar{R}_{y_i}$$

Can write in matrix form: $[K]\{\bar{u}\} = \{R\}$ where

$$K_{ij} = \begin{bmatrix} \left\langle \frac{\partial \phi_j}{\partial x} \frac{\partial \phi_i}{\partial x} + \frac{1-\nu}{2} \frac{\partial \phi_j}{\partial y} \frac{\partial \phi_i}{\partial y} \right\rangle & \left\langle \nu \frac{\partial \phi_j}{\partial x} \frac{\partial \phi_i}{\partial y} + \frac{1-\nu}{2} \frac{\partial \phi_j}{\partial y} \frac{\partial \phi_i}{\partial x} \right\rangle \\ \left\langle \nu \frac{\partial \phi_j}{\partial y} \frac{\partial \phi_i}{\partial x} + \frac{1-\nu}{2} \frac{\partial \phi_j}{\partial y} \frac{\partial \phi_i}{\partial y} \right\rangle & \left\langle \frac{\partial \phi_j}{\partial y} \frac{\partial \phi_i}{\partial y} + \frac{1-\nu}{2} \frac{\partial \phi_j}{\partial x} \frac{\partial \phi_i}{\partial x} \right\rangle \end{bmatrix}$$

$$\bar{u} = \begin{Bmatrix} u_j \\ v_j \end{Bmatrix}; \quad R_i = \begin{Bmatrix} \bar{R}_{x_i} \\ \bar{R}_{y_i} \end{Bmatrix}$$

Assembly of $[K]$:

- Composed of collection of submatrices K_{ij} , ... is 2×2 for each i, j combination
- At element level ... scalar element matrix is $\# \text{ nodes/element} \times \# \text{ nodes/element}$... i.e. Linear triangle is 3×3
- Now each entry consists of a 2×2

$$[K]^e = \begin{bmatrix} [::] & [::] & [::] \\ [::] & [::] & [::] \\ [::] & [::] & [::] \end{bmatrix}$$

\swarrow
 K_{32} locally

- In full storage mode; $[K]$ globally is $2N \times 2N$ for N nodes

Previously $i = \text{row index goes } 1 \text{ to } N$
 $j = \text{column index goes } 1 \text{ to } N$

Now for each i from 1 to N ; have 2 entries $\Rightarrow 2i-1, 2i$
 j from 1 to N ; have 2 entries $\Rightarrow 2j-1, 2j$

- In Band Storage Mode

IF HB = Half BW of Grid (i.e. max node difference in an element)

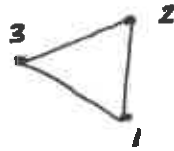
then Half BW of $[K] = 2(HB) + 1$

$$\begin{bmatrix} \square & \square & \square & \square & \square \\ \square & \square & \square & \square & \square \\ \square & \square & \square & \square & \square \\ \square & \square & \square & \square & \square \\ \square & \square & \square & \square & \square \end{bmatrix}$$

Total Bandwidth
 $2[2(HB) + 1] + 1$

Assembly of $[K]$ can't

e.g. Linear triangle



$$K_{ij} = \begin{bmatrix} \frac{\Delta y_j \Delta y_i}{4A} + \frac{1-\nu}{2} \left(\frac{\Delta x_j \Delta x_i}{4A} \right) & -\nu \frac{\Delta x_j \Delta y_i}{4A} - \frac{1-\nu}{2} \frac{\Delta y_j \Delta x_i}{4A} \\ -\nu \frac{\Delta y_j \Delta x_i}{4A} - \frac{1-\nu}{2} \frac{\Delta x_j \Delta y_i}{4A} & \frac{\Delta x_j \Delta x_i}{4A} + \frac{1-\nu}{2} \frac{\Delta y_j \Delta y_i}{4A} \end{bmatrix}$$

Assembly Loop:

Do $i = 1$ TO 3 Loop over local rows

$IIX = 2IN(L, i) - 1$
 $IY = IIX + 1$ } Get Global row #'s for vector system

Do $j = 1$ TO 3 Loop over local columns

$JIX = 2IN(L, j)$
 $JY = JIX + 1$ } Get Global column #'s for vector system

$$k_{11} = \frac{\Delta y(j) \Delta y(i)}{4A} + \frac{1-\nu}{2} \left(\frac{\Delta x(j) \Delta x(i)}{4A} \right)$$

$$k_{12} = -\nu \frac{\Delta x(j) \Delta y(i)}{4A} - \frac{1-\nu}{2} \frac{\Delta y(j) \Delta x(i)}{4A}$$

$$k_{21} = -\nu \frac{\Delta y(j) \Delta x(i)}{4A} - \frac{1-\nu}{2} \frac{\Delta x(j) \Delta y(i)}{4A}$$

$$k_{22} = \frac{\Delta x(j) \Delta x(i)}{4A} + \frac{1-\nu}{2} \frac{\Delta y(j) \Delta y(i)}{4A}$$

Compute coefficients for each (i, j) pair

$$A(IIX, NDIA6 + JIX - IIX) = A(IIX, NDIA6 + JIX - IIX) + k_{11}$$

$$A(IIX, NDIA6 + JY - IIX) = A(IIX, NDIA6 + JY - IIX) + k_{12}$$

$$A(IY, NDIA6 + JIX - IY) = A(IY, NDIA6 + JIX - IY) + k_{21}$$

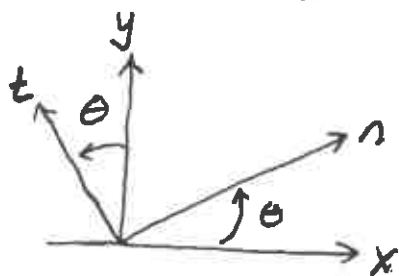
$$A(IY, NDIA6 + JY - IY) = A(IY, NDIA6 + JY - IY) + k_{22}$$

END j , END i

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Boundary conditions: usually specified in terms of Normal/Tangential (to the boundary) stress (Type II) or displacement (Type I)

Common Strategy: Rotate system of equations + variables into a local (n, t) system



For any vector \underline{F}

$$\begin{Bmatrix} F_n \\ F_t \end{Bmatrix} = \underbrace{\begin{bmatrix} \cos\theta & \sin\theta \\ -\sin\theta & \cos\theta \end{bmatrix}}_R \begin{Bmatrix} F_x \\ F_y \end{Bmatrix}$$

$$\begin{Bmatrix} F_x \\ F_y \end{Bmatrix} = \underbrace{\begin{bmatrix} \cos\theta & -\sin\theta \\ \sin\theta & \cos\theta \end{bmatrix}}_{R^{-1} = R^T} \begin{Bmatrix} F_n \\ F_t \end{Bmatrix}$$

$$R^{-1} = R^T$$

So if we want to rotate the force balance equations at node i into local (n, t) system:

$$[R_i] \begin{Bmatrix} FB_{x_i} \\ FB_{y_i} \end{Bmatrix} = \begin{Bmatrix} FB_{n_i} \\ FB_{t_i} \end{Bmatrix}$$

Then in terms of our overall system of equations

$$[K] \{z\} = \{b\}; \text{ rotating } i\text{th equation pair}$$

$$\begin{bmatrix} I & I & I & \dots & 0 \\ 0 & & & & R_{iI} \end{bmatrix} [K] \{z\} = \begin{bmatrix} I & I & I & \dots & 0 \\ & & & & R_{iI} \end{bmatrix} \{b\}$$

Premultiply system matrix by rotation matrix; also right-hand side vector b

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Now rotate all equations (pairs) each by their own R_i

$$\begin{bmatrix} R_1 & R_2 & \dots & R_i & \dots & R_N \end{bmatrix} \begin{bmatrix} K \end{bmatrix} \begin{Bmatrix} z \end{Bmatrix} = \begin{bmatrix} R_1 & R_2 & \dots & R_i & \dots & R_N \end{bmatrix} \begin{Bmatrix} b \end{Bmatrix}$$

$\begin{matrix} \nearrow \\ \text{Unknown's still in} \\ (x,y) \text{ system} \end{matrix}$
 $\begin{matrix} \searrow \\ \text{b} \\ (n,t) \end{matrix}$

We have for node i : row $2i-1$ = Normal Force balance eqn
row $2i$ = tangential force balance eqn

For Type I BC's (displacement given): normal displacement specified in favor of normal force balance. replace with direct specification of normal displacement

$$U_i \cos \theta_i + V_i \sin \theta_i = \text{Known value}$$

2 problems arise: (1) This equation goes in the "normal" row, i.e. row $2i-1$, $\therefore \cos \theta_i$ appears on the diagonal; what if $\cos \theta_i = 0$??

(2) Symmetry destroyed

Cure: "rotate" the z vector also, so that variables are in local (θ_i) system

$$z_i = \begin{Bmatrix} U_i \\ V_i \end{Bmatrix} = \begin{bmatrix} R_i^{-1} \end{bmatrix} \begin{Bmatrix} U_{\text{normal}} \\ U_{\text{tangential}} \end{Bmatrix}$$

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$$\begin{Bmatrix} z \end{Bmatrix}_{(x,y)} = \begin{bmatrix} R_1^{-1} & R_2^{-1} & \dots & R_n^{-1} \end{bmatrix} \begin{Bmatrix} z \end{Bmatrix}_{(n,t)}$$

So our system:

$$[R][K]\begin{Bmatrix} z \end{Bmatrix}_{(x,y)} = [R]\begin{Bmatrix} b \end{Bmatrix}_{(x,y)} \equiv \begin{Bmatrix} b \end{Bmatrix}_{(n,t)}$$

becomes

$$[R][K][R^{-1}]\begin{Bmatrix} z \end{Bmatrix}_{(n,t)} = \begin{Bmatrix} b \end{Bmatrix}_{(n,t)}$$

But $R^{-1} = R^T \Rightarrow RKR^{-1}$ Similarity Transformation (Orthogonal); preserves Symmetry if $[K]$ is symmetric

Type I BC: $\begin{bmatrix} \diagdown & & & & \\ 0 & 0 & 0 & 0 & 0 \end{bmatrix} \begin{Bmatrix} \vdots \\ u_{n,i} \\ u_{t,i} \\ \vdots \end{Bmatrix} = \begin{Bmatrix} \vdots \\ \text{Known} \\ \vdots \end{Bmatrix}$

Procedure: Assemble K, R
 Rotate RKR^{-1}, Rb
 Solve $z_{n,t}$
 rotate back $z_{x,y} = Rz_{n,t}$

Common to build R, R^{-1} at the element level and perform the transformations there; also select out the boundary nodes and only rotate those equations and variables; leave others in (x,y) system

Need a way to compute; $\cos \theta$ $\sin \theta$ at local level
 Boils down to computing a "nodal" normal direction

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Note: In terms of \hat{n}, \hat{t} vectors:

$$\text{If } \underline{F} = F_x \hat{x} + F_y \hat{y} ; \quad \underline{F}_n = F_x (\hat{x} \cdot \hat{n}) + F_y (\hat{y} \cdot \hat{n})$$

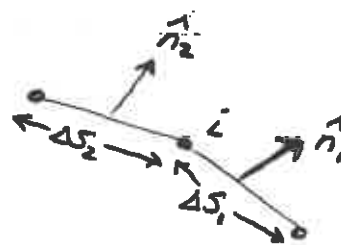
$$\underline{F}_t = F_x (\hat{x} \cdot \hat{t}) + F_y (\hat{y} \cdot \hat{t})$$

$$\begin{Bmatrix} F_n \\ F_t \end{Bmatrix} = \underbrace{\begin{bmatrix} \hat{x} \cdot \hat{n} & \hat{y} \cdot \hat{n} \\ \hat{x} \cdot \hat{t} & \hat{y} \cdot \hat{t} \end{bmatrix}}_R \begin{Bmatrix} F_x \\ F_y \end{Bmatrix}$$

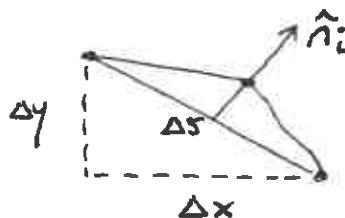
Take the "Nodal Normal"

$$\hat{n}_i \equiv \frac{\oint \hat{n} \phi_i ds}{|\oint \hat{n} \phi_i ds|}$$

$$\oint \hat{n} \phi_i ds = \frac{1}{2} [\hat{n}_1 \Delta s_1 + \hat{n}_2 \Delta s_2]$$



Works out to be that $\hat{n}_i = \frac{\Delta y \hat{x} - \Delta x \hat{y}}{\Delta s}$



Also note $\oint \hat{n} \phi_i ds \equiv \underbrace{\langle \nabla \phi_i \rangle}_{\text{Compute via element loop. Valid for all kinds of } \phi_i}$

Calculation of derived quantities

e.g. Plane Stress $\sigma_x = \frac{E}{1-\nu^2} \left(\frac{\partial u}{\partial x} + \nu \frac{\partial v}{\partial y} \right)$

Strategies: a.) $\frac{\partial u}{\partial x}$ compute on an element; constant
take as existing at element center
1 order lower in accuracy due to differentiation

b.) Galerkin treatment:

$$\sigma_x = \sum_j \sigma_j \phi_j$$

$$\langle \sigma_x \phi_i \rangle = \left\langle \frac{E}{1-\nu^2} \left(\frac{\partial u}{\partial x} + \nu \frac{\partial v}{\partial y} \right) \phi_i \right\rangle$$

$$\sum_j \sigma_j \langle \phi_j \phi_i \rangle = \left\langle \frac{E}{1-\nu^2} \left(\frac{\partial u}{\partial x} + \nu \frac{\partial v}{\partial y} \right) \phi_i \right\rangle$$

Matrix equation: $[M] \{ \sigma_x \} = \{ \tau_x \}$

\uparrow
 $m_{ij} = \langle \phi_j \phi_i \rangle$

\uparrow Computed from known
values of u, v

Likewise for σ_y, τ :

$$[M] \{ \sigma_y \} = \{ \tau_y \}$$

$$[M] \{ \tau \} = \{ \tau_\tau \}$$

Can build-up $[M]$, decompose and solve; alternatively
use "Integral lumping" to diagonalize $[M]$;

$$\sigma_{x_i} = \frac{\tau_{x_i}}{\langle \phi_i \phi_i \rangle}$$

$$\sigma_{y_i} = \frac{\tau_{y_i}}{\langle \phi_i \phi_i \rangle}$$

Inexpensive;
works well

$$\tau_i = \frac{\tau_{\tau_i}}{\langle \phi_i \phi_i \rangle}$$

Do this on interior nodes, but at boundary procedure
not very good... essentially have "one-sided" differencing

(11)

- Better to recover boundary stresses via "unused" Galerkin equations on Type I boundary

$$\text{i.e. } - \underbrace{\langle \underline{\sigma} \cdot \nabla \phi_i \rangle}_{\text{Computable through}} = \underbrace{\langle \underline{B} \phi_i \rangle}_{\text{displacements once}} - \underbrace{\int \underline{\sigma} \cdot \hat{n} \phi_i \, ds}_{\text{Sol'n is known}}$$

Computable through
displacements once
Sol'n is known

given

Recipe for
getting $\underline{\sigma} \cdot \hat{n}$

