

## Building an Estimator

### README

#### I. Adding sensor noise

The goal here is to measure the standard deviation of the GPS measurement and the Accelerometer measurement which are Gaussian. Samples of data are collected from each of these distributions.

For the GPS, it is recorded in Graph1.txt. To measure the standard deviation, I followed the steps below:

- I took the first 10 samples;
- I computed the mean using the formula  $\hat{x} = \frac{1}{10} \sum_{i=1}^{10} x_i$ ;
- Having the mean, I computed the variance using 
$$\hat{\sigma}^2 = \frac{1}{10} \sum_{i=1}^{10} (x_i - \hat{x})^2;$$
- The standard deviation  $std = \sqrt{\hat{\sigma}^2} = 0.6521$

I followed the same process for the Accelerometer ( $std = \sqrt{\hat{\sigma}^2} = 0.576$ ).

Having the two standard deviations, I tuned them slightly to meet the ~68% requirement.

#### II. Implemented Estimator

##### ➤ Attitude Estimation

It is responsible for improving the complementary filter using the measurements from the accelerometer and the gyro.

```

92 /////////////////////////////////////////////////////////////////// BEGIN STUDENT CODE ///////////////////////////////////////////////////////////////////
93 // SMALL ANGLE GYRO INTEGRATION:
94 // (replace the code below)
95 // make sure you comment it out when you add your own code -- otherwise e.g. you might integrate yaw twice
96
97 float r = rollEst, p = pitchEst;
98 float v[9] = { 1, sin(r)*tan(p), cos(r)*tan(p), 0, cos(r), -sin(r), 0, sin(r)/cos(p), cos(r)/cos(p)};
99 Mat3x3F R(v); // Matrix that turns instantaneous turn rate in body frame to inertial frame
100
101 //Transform the angular velocity from body frame to inertial frame
102 V3F i_rate = R* gyro;
103
104 //Create a predicted attitude in the global frame
105 float predictedPitch = pitchEst + dtIMU * i_rate.y ;
106 float predictedRoll = rollEst + dtIMU * i_rate.x;
107 ekfState(6) = ekfState(6) + dtIMU * i_rate.z; // yaw
108
109 // normalize yaw to -pi .. pi
110 if (ekfState(6) > F_PI) ekfState(6) -= 2.*F_PI;
111 if (ekfState(6) < -F_PI) ekfState(6) += 2.*F_PI;
112
113 /////////////////////////////////////////////////////////////////// END STUDENT CODE ///////////////////////////////////////////////////////////////////

```

### ➤ The Prediction step

It is responsible for predicting the next state of the vehicle given the current acceleration in world frame and the angular velocity in the z-axis.

It uses 3 functions:

- The predictState function which predicts the state forward excluding the yaw angle.

```

173 /////////////////////////////////////////////////////////////////// BEGIN STUDENT CODE ///////////////////////////////////////////////////////////////////
174 V3F accel_gf = attitude.Rotate_BtoI(accel * dt);
175
176 predictedState(0) = predictedState(0) + predictedState(3)*dt;
177 predictedState(1) = predictedState(1) + predictedState(4)*dt;
178 predictedState(2) = predictedState(2) + predictedState(5)*dt;
179 predictedState(3) = predictedState(3) + accel_gf.x;
180 predictedState(4) = predictedState(4) + accel_gf.y;
181 predictedState(5) = predictedState(5) - 9.81f * dt + accel_gf.z;
182
183 /////////////////////////////////////////////////////////////////// END STUDENT CODE ///////////////////////////////////////////////////////////////////

```

- The GetRbgPrime which computes the Jacobian at the current state.

```

206
207 ////////////////////////////////////////////////// BEGIN STUDENT CODE ///////////////////////////////////
208
209 RbgPrime(0, 0) = -cos(pitch) * sin(yaw);
210 RbgPrime(0, 1) = -sin(roll) * sin(pitch) * sin(yaw) - cos(roll)*cos(yaw);
211 RbgPrime(0, 2) = -cos(roll) * sin(pitch) * sin(yaw) + sin(roll) * cos(yaw);
212 RbgPrime(1, 0) = cos(pitch) * cos(yaw);
213 RbgPrime(1, 1) = sin(roll) * sin(pitch) * cos(yaw) - cos(roll) * sin(yaw);
214 RbgPrime(1, 2) = cos(roll) * sin(pitch) * cos(yaw) + sin(roll) * sin(yaw);
215 RbgPrime(2, 0) = 0;
216 RbgPrime(2, 1) = 0;
217 RbgPrime(2, 2) = 0;
218 ////////////////////////////////////////////////// END STUDENT CODE ///////////////////////////////////

```

- The predict function which predicts the current covariance forward.

```

260
261 ////////////////////////////////////////////////// BEGIN STUDENT CODE ///////////////////////////////////
262
263 //Generating gPrime
264 VectorXf ac_prime(3), ac(3);
265 ac(0) = accel.x;
266 ac(1) = accel.y;
267 ac(2) = accel.z;
268 ac_prime = RbgPrime * ac;
269 gPrime(0, 3) = dt;
270 gPrime(1, 4) = dt;
271 gPrime(2, 5) = dt;
272 gPrime(3, 6) = ac_prime(0) * dt;
273 gPrime(4, 6) = ac_prime(1) * dt;
274 gPrime(5, 6) = ac_prime(2) * dt;
275
276 //Calculating the predicted covariance
277 ekfCov = gPrime * ekfCov;
278 gPrime.transposeInPlace();
279 ekfCov = ekfCov * gPrime + Q;
280
281 ////////////////////////////////////////////////// END STUDENT CODE ///////////////////////////////////

```

### ➤ The Magnetometer update

It updates the value of the yaw angle given measurement from the magnetometer.

```

334 ////////////////////////////////////////////////// BEGIN STUDENT CODE ///////////////////////////////////
335 zFromX(0) = ekfState(6);
336 hPrime(0, 6) = 1;
337
338 ////////////////////////////////////////////////// END STUDENT CODE ///////////////////////////////////

```

```

360 zt = z - zFromX;
361 if (z.size() == 1) //If the update function is call while uptading yaw, normalize the difference btw the measured and the esti
362 {
363     if (zt(0) > F_PI) zt(0) -= 2.*F_PI;
364     if (zt(0) < -F_PI) zt(0) += 2.*F_PI;
365 }

```

### ➤ The GPS update

It updates the vehicle position and velocity given measurement from the GPS.

```
303 ////////////////////////////////////////////////// BEGIN STUDENT CODE //////////////////////////////////////
304 zFromX(0) = ekfState(0);
305 zFromX(1) = ekfState(1);
306 zFromX(2) = ekfState(2);
307 zFromX(3) = ekfState(3);
308 zFromX(4) = ekfState(4);
309 zFromX(5) = ekfState(5);
310 for (int i = 0; i < 6; i++)
311 {
312     hPrime(i, i) = 1;
313 }
314
315 ////////////////////////////////////////////////// END STUDENT CODE //////////////////////////////////////
```