

Quadcopter Platform Linearization

Nathan Michael

The following is rough code (unpolished) to linearize the platform model. A few checks or symbolic/numeric substitutions are included.

Provided “as-is” to save time.

Define the rotation matrices about the axes.

```
In[44]:= Rx[θ_] := {{1, 0, 0}, {0, Cos[θ], -Sin[θ]}, {0, Sin[θ], Cos[θ]}};
Ry[θ_] := {{Cos[θ], 0, Sin[θ]}, {0, 1, 0}, {-Sin[θ], 0, Cos[θ]}};
Rz[θ_] := {{Cos[θ], -Sin[θ], 0}, {Sin[θ], Cos[θ], 0}, {0, 0, 1}};
```

Define the full rotation matrix (ZYX parameterization).

```
In[47]:= Rot[φ_, θ_, ψ_] := Rz[ψ].Ry[θ].Rx[φ];
```

Assume a diagonal inertia matrix.

```
In[48]:= J = DiagonalMatrix[{J11, J22, J33}];
```

Vee operation:

```
In[49]:= vee[M_] := {M[[3, 2]], M[[1, 3]], M[[2, 1]]};
```

Relationship between body frame angular velocities and Euler angle derivatives

```
In[50]:= H = Normal[CoefficientArrays[FullSimplify[vee[Transpose[Rot[φ[t], θ[t], ψ[t]]].
D[Rot[φ[t], θ[t], ψ[t]], t]]], {φ'[t], θ'[t], ψ'[t]}][[2]]];
```

```
In[51]:= omega = FullSimplify[H.{φ'[t], θ'[t], ψ'[t]}];
```

Nonlinear equations of motion.

```
In[52]:= nonlinEqns = FullSimplify[Join[ $\frac{1}{m} * (\text{Rot}[\phi[t], \theta[t], \psi[t]].\{0, 0, f[t]\} - \{0, 0, m * g\})$ ,
Inverse[J].({M1[t], M2[t], M3[t]} - Cross[omega, J.omega])]]];
```

```
In[53]:= nonlinEqns // MatrixForm
```

Out[53]//MatrixForm=

$$\begin{pmatrix} \frac{f[t] (\cos[\phi[t]] \cos[\psi[t]] \sin[\theta[t]] + \sin[\phi[t]] \sin[\psi[t]])}{m} \\ \frac{f[t] (-\cos[\psi[t]] \sin[\phi[t]] + \cos[\phi[t]] \sin[\theta[t]] \sin[\psi[t]])}{m} \\ -g + \frac{\cos[\theta[t]] \cos[\phi[t]] f[t]}{m} \\ \frac{M1[t] + (J22 - J33) (-\cos[\phi[t]] \sin[\phi[t]] \theta'[t]^2 + \cos[\theta[t]] \cos[2\phi[t]] \theta'[t] \psi'[t] + \cos[\theta[t]]^2 \cos[\phi[t]] \sin[\phi[t]] \psi'[t]^2)}{J11} \\ \frac{M2[t] + (J11 - J33) (\sin[\phi[t]] \theta'[t] - \cos[\theta[t]] \cos[\phi[t]] \psi'[t]) (\phi'[t] - \sin[\theta[t]] \psi'[t])}{J22} \\ \frac{M3[t] + (J11 - J22) (\phi'[t] - \sin[\theta[t]] \psi'[t]) (\cos[\phi[t]] \theta'[t] + \cos[\theta[t]] \sin[\phi[t]] \psi'[t])}{J33} \end{pmatrix}$$

Create the state space model, linearized about ϕ_i , θ_i , etc.

```

In[54]:= ss = StateSpaceModel[
  {Join[{x'[t], y'[t], z'[t],  $\phi'$ [t],  $\theta'$ [t],  $\psi'$ [t]}, nonlinEqns],
    {x[t], y[t], z[t],  $\phi$ [t],
       $\theta$ [t],  $\psi$ [t], x'[t], y'[t], z'[t],  $\phi'$ [t],  $\theta'$ [t],  $\psi'$ [t]}},
  {{x[t], 0},
    {y[t], 0},
    {z[t], 0},
    { $\phi$ [t],  $\phi_i$ },
    { $\theta$ [t],  $\theta_i$ },
    { $\psi$ [t],  $\psi_i$ },
    {x'[t], 0},
    {y'[t], 0},
    {z'[t], 0},
    { $\phi'$ [t],  $\phi_{di}$ },
    { $\theta'$ [t],  $\theta_{di}$ },
    { $\psi'$ [t],  $\psi_{di}$ }},
  {{f[t], fm},
    {M1[t], m1},
    {M2[t], m2},
    {M3[t], m3}}
];

```

Extract A and B for visual inspection.

```

In[55]:= A = ss[[1, 1]];
B = ss[[1, 2]];

```

Process model at hover.

```
In[59]:= subs = {phi -> 0, theta -> 0, phi d -> 0, theta d -> 0, fm -> m * g};
```

```
In[60]:= A //. subs // MatrixForm
```

```
B //. subs // MatrixForm
```

```
Out[60]//MatrixForm=
```

$$\begin{pmatrix} 0 & 0 & 0 & 0 & 0 & 0 & 1 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & g \sin[\psi_i] & g \cos[\psi_i] & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & -g \cos[\psi_i] & g \sin[\psi_i] & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & \frac{(J_{22}-J_{33}) \psi_{di}^2}{J_{11}} & 0 & 0 & 0 & 0 & 0 & 0 & \frac{(J_{22}-J_{33}) \psi_{di}}{J_{11}} & 0 & 0 \\ 0 & 0 & 0 & 0 & \frac{(J_{11}-J_{33}) \psi_{di}^2}{J_{22}} & 0 & 0 & 0 & 0 & -\frac{(J_{11}-J_{33}) \psi_{di}}{J_{22}} & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \end{pmatrix}$$

```
Out[61]//MatrixForm=
```

$$\begin{pmatrix} 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ \frac{1}{m} & 0 & 0 & 0 \\ 0 & \frac{1}{J_{11}} & 0 & 0 \\ 0 & 0 & \frac{1}{J_{22}} & 0 \\ 0 & 0 & 0 & \frac{1}{J_{33}} \end{pmatrix}$$

Attitude dynamics (lower block)

```
In[62]:= FullSimplify[ss[[1, 1]][[7 ;; 12, 7 ;; 12]]] // MatrixForm
```

```
A22 = FullSimplify[ss[[1, 1]][[7 ;; 12, 7 ;; 12]]];
```

```
Out[62]//MatrixForm=
```

$$\begin{pmatrix} 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & \frac{(J_{22}-J_{33}) (\psi_{di} \cos[\theta_i] \cos[2\phi_i] - 2\theta_{di} \cos[\phi_i] \sin[\phi_i])}{J_{11}} \\ 0 & 0 & 0 & \frac{(J_{11}-J_{33}) (-\psi_{di} \cos[\theta_i] \cos[\phi_i] + \theta_{di} \sin[\phi_i])}{J_{22}} & \frac{(J_{11}-J_{33}) (\phi_{di} - \psi_{di} \sin[\theta_i]) \sin[\phi_i]}{J_{22}} - \frac{(J_{11}-J_{33}) (\phi_{di} \cos[\theta_i] - \psi_{di} \sin[\theta_i])}{J_{33}} \\ 0 & 0 & 0 & \frac{(J_{11}-J_{22}) (\theta_{di} \cos[\phi_i] + \psi_{di} \cos[\theta_i] \sin[\phi_i])}{J_{33}} & \frac{(J_{11}-J_{22}) \cos[\phi_i] (\phi_{di} - \psi_{di} \sin[\theta_i])}{J_{33}} - \frac{(J_{11}-J_{22}) (\phi_{di} \cos[\theta_i] - \psi_{di} \sin[\theta_i])}{J_{33}} \end{pmatrix}$$

Should be controllable, verify:

```
In[64]:= ControllableModelQ[ss]
```

```
Out[64]= True
```

Select full-state feedback gains via LQR:

```
In[65]:= Q = 0.1 * IdentityMatrix[12];
```

```
R = 2 * IdentityMatrix[4];
```

```
In[67]:= Chop[LQRegulatorGains[ss //. subs, {Q, R}], 10-8] // MatrixForm
```

... **RiccatiSolve**: RiccatiSolve has received a matrix with non-numerical elements.

⋮ **RiccatiSolve**: RiccatiSolve has received a matrix with non-numerical elements.

Out[67]//MatrixForm=

[illegible]

$$\begin{aligned} & \{ \{ \{ 0.1, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0 \}, \{ 0, 0.1, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0 \}, \\ & \{ 0, 0, 0.1, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0 \}, \{ 0, 0, 0, 0.1, 0, 0, 0, 0, 0, 0, 0, 0, 0 \}, \\ & \{ 0, 0, 0, 0, 0.1, 0, 0, 0, 0, 0, 0, 0, 0 \}, \{ 0, 0, 0, 0, 0, 0.1, 0, 0, 0, 0, 0, 0 \}, \\ & \{ 0, 0, 0, 0, 0, 0, 0.1, 0, 0, 0, 0, 0, 0 \}, \{ 0, 0, 0, 0, 0, 0, 0, 0.1, 0, 0, 0, 0 \}, \\ & \{ 0, 0, 0, 0, 0, 0, 0, 0, 0.1, 0, 0, 0, 0 \}, \{ 0, 0, 0, 0, 0, 0, 0, 0, 0, 0.1, 0, 0 \}, \\ & \{ 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0.1, 0 \}, \{ 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0.1 \} \}, \\ & \{ \{ 2, 0, 0, 0 \}, \{ 0, 2, 0, 0 \}, \{ 0, 0, 2, 0 \}, \{ 0, 0, 0, 2 \} \} \} \end{aligned}$$

```
In[68]:= A22 // MatrixForm
```

Out[68]//MatrixForm=

$$\begin{pmatrix} 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & (J22-J33) (\psi_{di} \cos[\theta_i] \cos[2\phi_i] - 2\theta_{di} \cos[\phi_i] \sin[\phi_i]) \\ 0 & 0 & 0 & (J11-J33) (-\psi_{di} \cos[\theta_i] \cos[\phi_i] + \theta_{di} \sin[\phi_i]) & J11 \\ 0 & 0 & 0 & J22 & (J11-J33) (\phi_{di} - \psi_{di} \sin[\theta_i]) \sin[\phi_i] \\ 0 & 0 & 0 & (J11-J22) (\theta_{di} \cos[\phi_i] + \psi_{di} \cos[\theta_i] \sin[\phi_i]) & J22 \\ 0 & 0 & 0 & J33 & (J11-J22) \cos[\phi_i] (\phi_{di} - \psi_{di} \sin[\theta_i]) \end{pmatrix} \begin{matrix} \\ \\ \\ \\ \\ - (J11- \\ - (J11- \\ - (J11- \end{matrix}$$