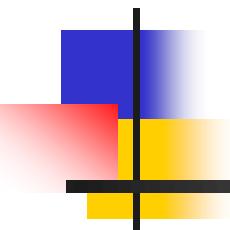


Introduction to Turbulence Modelling



Chao-An Lin

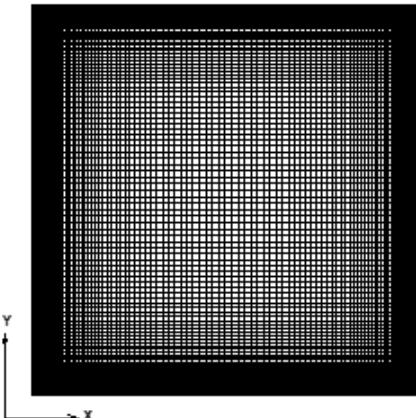
Department of Power Mechanical Engineering
National Tsing Hua University

Navier-Stokes equation

- Incompressible Navier-Stokes equations

$$\frac{\partial u_i}{\partial t} + \frac{\partial}{\partial x_j} (u_i u_j) = - \frac{1}{\rho} \frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_j} \left[\nu \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \right]$$

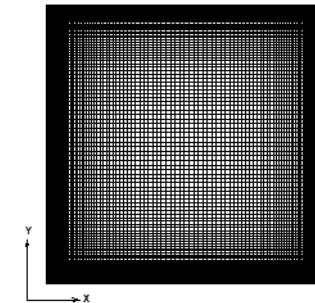
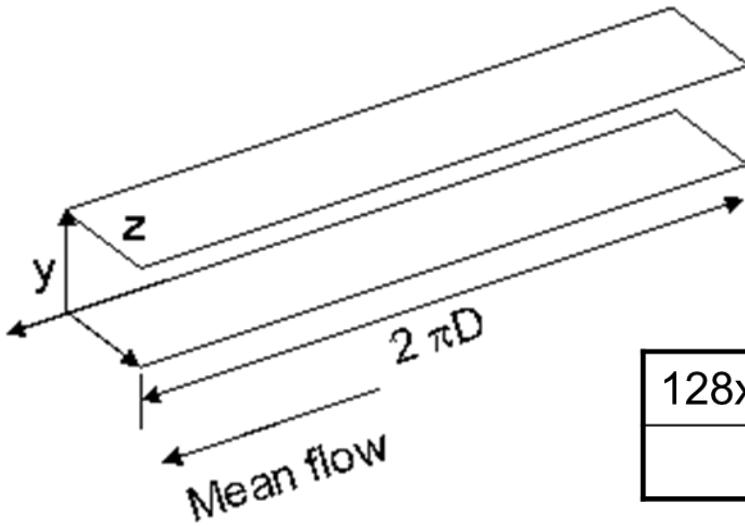
- To resolve the near wall structure
 - Grid is clustered in wall normal direction



$$(x_i = \xi [1 - \frac{\tanh \gamma (\xi - \hat{x}_i)}{\tanh \gamma \xi}], \xi = \frac{D}{2}, i = 1, 2)$$

Direct Numerical simulation

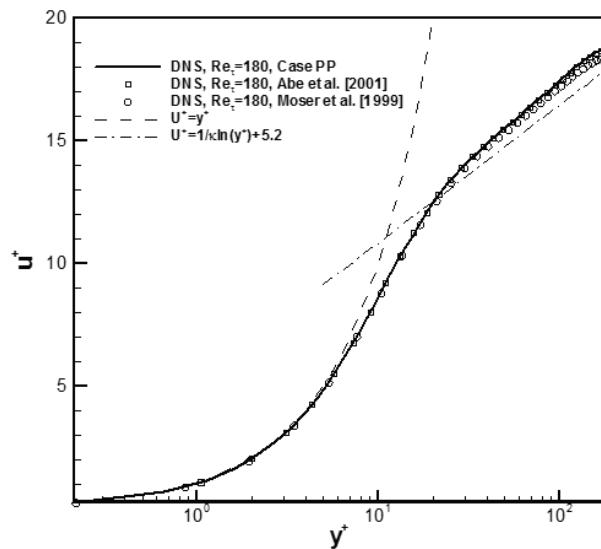
- Direct numerical simulations applied to the turbulent Poiseuille flow in a plane channel.
- The frictional Reynolds numbers is 360 based on frictional velocity and channel height.
- Simulation results are compared with DNS results of Abe et al. (2001) and Moser et al. (1999)at the same Ret.



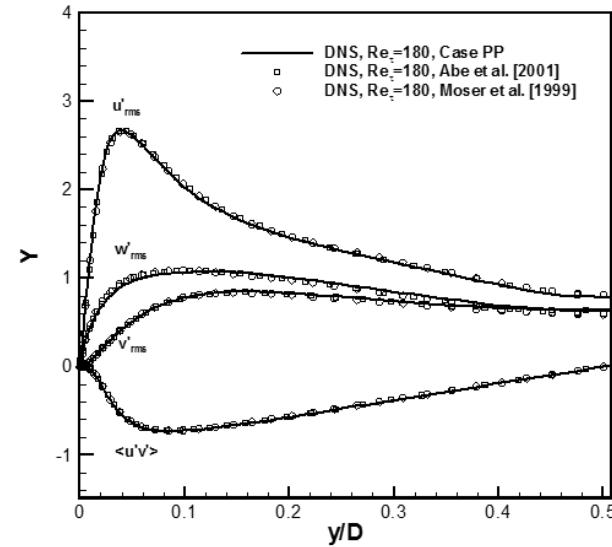
128x128x128	Δx^+	Δy^+	Δz^+
DNS	8.8357	0.2126~5.927	4.4879

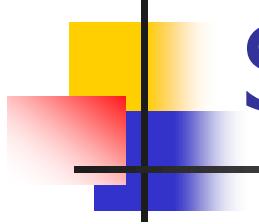
Direct Numerical simulation

□ Mean streamwise velocity profile



□ Turbulence intensities and Reynolds stress





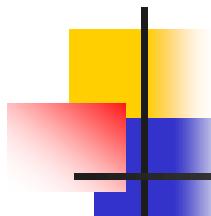
Reynolds averaged Navier-Stokes equations

- Reynolds averaged NS equation

$$\frac{\partial}{\partial x_j}(\rho U_i U_j) = -\frac{\partial P}{\partial x_i} + \frac{\partial}{\partial x_j} \left[\mu \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) - \rho \bar{u}_i \bar{u}_j \right]$$

- How to model the Reynolds stress

??



Eddy viscosity models

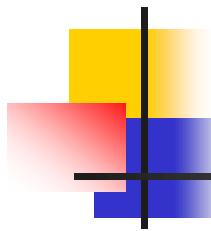
- Navier Stokes equation

$$\frac{\partial \rho U_j U_i}{\partial x_j} = - \frac{\partial P}{\partial x_i} + \frac{\partial}{\partial x_j} \left[\mu \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) - \rho \overline{u_i u_j} \right]$$

- Boussinesq Approximation

$$-\rho \overline{u_i u_j} = \mu_t \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) - \frac{2}{3} \delta_{ij} \rho k$$

$$\frac{\partial \rho U_j U_i}{\partial x_j} = - \frac{\partial (P + \frac{2}{3} \rho k)}{\partial x_i} + \frac{\partial}{\partial x_j} \left[(\mu + \mu_t) \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) \right]$$



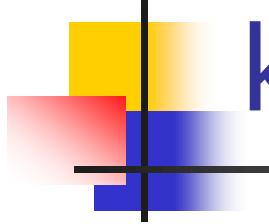
Eddy viscosity models

- k- ε model (Launder and Jones 1972)

$$-\rho \overline{u_i u_j} = \mu_t \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) - \frac{2}{3} \delta_{ij} \rho k,$$
$$\mu_t = C_\mu \rho k^a \varepsilon^b = C_\mu \rho k^2 / \varepsilon$$

- K equation

$$\frac{\partial \rho U_j k}{\partial x_j} = \frac{\partial}{\partial x_j} \left[-\frac{1}{2} \rho \overline{u_j u_i u_i} - \rho \overline{u_j p} \right] - \rho \overline{u_i u_j} \frac{\partial U_i}{\partial x_j} - \underbrace{\mu \frac{\partial u_i}{\partial x_j} \frac{\partial u_i}{\partial x_j}}_{\rho \varepsilon}$$



k- εmodels-k equation

$$\underbrace{-\rho \bar{u}_i \bar{u}_j \frac{\partial U_i}{\partial x_j}}_{P_k} = \left[\mu_t \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) - \frac{2}{3} \delta_{ij} \rho k \right] \frac{\partial U_i}{\partial x_j}$$
$$= \mu_t \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) \frac{\partial U_i}{\partial x_j}$$
$$-\frac{1}{2} \rho \bar{u}_j \bar{u}_i \bar{u}_i + \rho \bar{u}_j \bar{p} = \frac{\mu_t}{\sigma_k} \frac{\partial k}{\partial x_j}$$
$$\frac{\partial \rho U_j k}{\partial x_j} = \frac{\partial}{\partial x_j} \left[(\mu + \frac{\mu_t}{\sigma_k}) \frac{\partial k}{\partial x_j} \right] + \mu_t \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) \frac{\partial U_i}{\partial x_j} - \rho \varepsilon$$

k- εmodels-ε equation

$$\frac{\partial \rho U_j k}{\partial x_j} = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_k} \right) \frac{\partial k}{\partial x_j} \right] + P_k - \rho \varepsilon$$

$$\frac{\partial \rho U_j \varepsilon}{\partial x_j} = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_\varepsilon} \right) \frac{\partial \varepsilon}{\partial x_j} \right] + \frac{\varepsilon}{k} (C_{\varepsilon 1} P_k - C_{\varepsilon 2} \rho \varepsilon)$$

-If $k \sim x^{-n}$ decaying grid turbulence $P_k = 0$

$$\frac{dk}{dx} = -\varepsilon \Rightarrow -nx^{-n-1} = -\varepsilon \Rightarrow \varepsilon = nx^{-n-1}$$

$$\frac{d\varepsilon}{dx} = -C_{\varepsilon 2} \frac{\varepsilon^2}{k} \Rightarrow -(n+1)nx^{-n-2} = -C_{\varepsilon 2} \frac{n^2 x^{-2n-2}}{x^{-n}}$$

$$C_{\varepsilon 2} = \frac{n+1}{n} \approx 1.92 \leftarrow n \approx 1.1$$

$$U \frac{\partial k}{\partial x} + V \frac{\partial k}{\partial y} = \underbrace{-\bar{uv} \frac{\partial U}{\partial y}}_{A-convection} - \underbrace{\frac{\partial}{\partial y} (\bar{v}k' + \frac{\bar{vp}}{\rho})}_{B-production} - \underbrace{\nu \frac{\partial^2 u_i}{\partial x_j^2}}_{C-diffusion} - \underbrace{\nu \frac{\partial^2 u_i}{\partial x_j^2}}_{D-dissipation}$$

Kinetic energy of turbulence

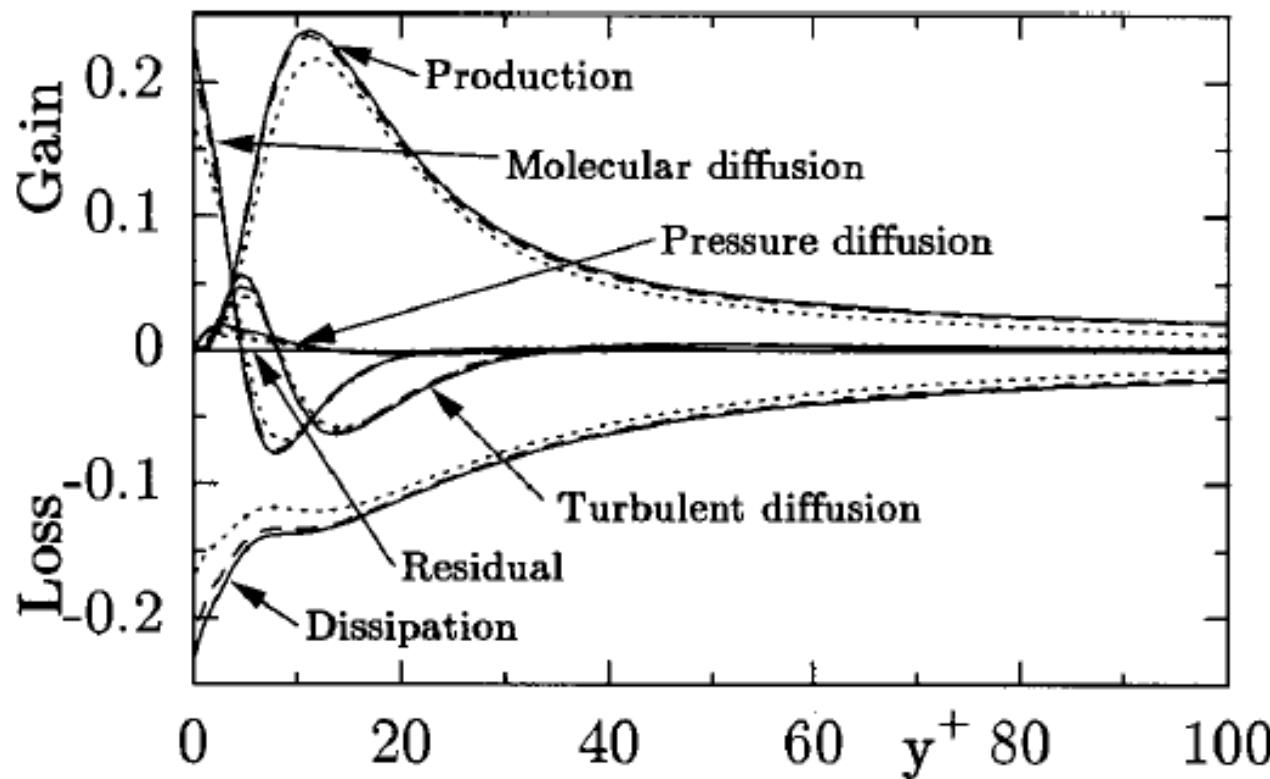


Fig. 13 Budget of turbulent kinetic energy: —, $Re_\tau = 640$; - - -, $Re_\tau = 395$; - · -, $Re_\tau = 180$

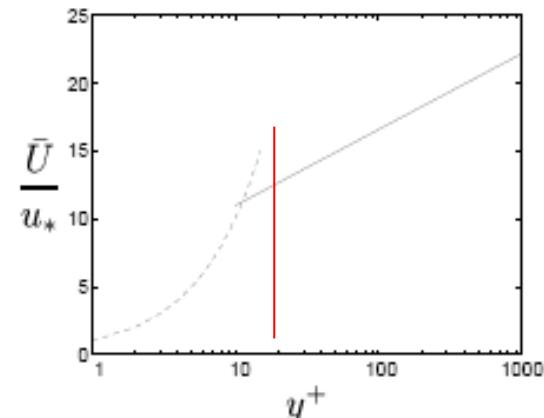
k- εmodels-local equilibrium

– For homogeneous shear flow $S = \frac{\partial U}{\partial y} \neq 0$

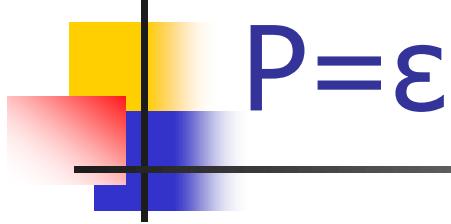
$$P_k = \mu_t \frac{\partial U}{\partial y} \frac{\partial U}{\partial y} = \rho \varepsilon \Leftrightarrow \tau = \mu_t \frac{\partial U}{\partial y} = \rho |\bar{uv}|$$

$$\rho^2 \bar{uv}^2 = \tau^2 = \mu_t^2 \left(\frac{\partial U}{\partial y} \right)^2 = \mu_t \rho \varepsilon = \rho C_\mu \frac{k^2}{\varepsilon} \rho \varepsilon$$

$$C_\mu = \frac{\bar{uv}^2}{k^2} \approx 0.3^2 = 0.09 \Leftrightarrow \left| \frac{\bar{uv}}{k} \right| \approx 0.3 \leftarrow \text{inner wall}$$



k- εmodels-local equilibrium


$$P = \varepsilon$$

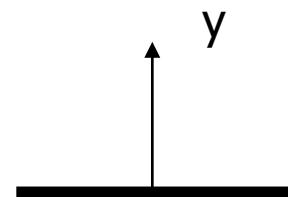
Near wall turbulence kinetic energy level

– The log layer – constant stress layer

$$|\overline{uv}| = u_*^2, \text{ and log law } \frac{\partial U}{\partial y} = \frac{u_*}{\kappa y}$$

$$u_*^2 = -\overline{uv} = \nu_t \frac{\partial U}{\partial y} = C_\mu \frac{k^2 u_*}{\varepsilon \kappa y} \Rightarrow \varepsilon = C_\mu \frac{k^2}{u_* \kappa y}$$

$$P = -\overline{uv} \frac{\partial U}{\partial y} = u_*^2 \frac{u_*}{\kappa y} = \varepsilon = C_\mu \frac{k^2}{u_* \kappa y} \Rightarrow k = \frac{u_*^2}{\sqrt{C_\mu}}$$



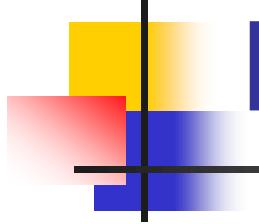
k- ε models-local equilibrium

$$0 = \frac{\partial}{\partial y} \left[\frac{\mu_t}{\sigma_\varepsilon} \frac{\partial \varepsilon}{\partial y} \right] + \frac{\varepsilon}{k} (C_{\varepsilon 1} P_k - C_{\varepsilon 2} \rho \varepsilon), \mu_t \simeq \rho u \ell \\ = \rho u_* \kappa y$$

$$\frac{\partial}{\partial y} \left[\frac{\mu_t}{\sigma_\varepsilon} \frac{\partial \varepsilon}{\partial y} \right] = \frac{\partial}{\partial y} \left[\frac{\rho u_* \kappa y}{\sigma_\varepsilon} \frac{\partial}{\partial y} \rho \frac{u_*^3}{\kappa y} \right] = \rho^2 \frac{u_*^4}{\sigma_\varepsilon y^2}$$

$$0 = (C_{\varepsilon 1} - C_{\varepsilon 2}) \frac{u_*^3}{\kappa y} \frac{1}{C_\mu^{-1/2} u_*^2} \frac{u_*^3}{\kappa y} + \frac{u_*^4}{\sigma_\varepsilon y^2}$$

$$C_{\varepsilon 2} - C_{\varepsilon 1} = \frac{\kappa^2}{C_\mu^{1/2} \sigma_\varepsilon} = \frac{0.41^2}{0.3 \times 1.3} = 0.43$$



k- εmodels

- Jet spreading rate
- Plane jet in the self-similar region

$$\frac{d\delta^{1/2}}{dx} \approx 0.11 \Leftarrow C_{\varepsilon 1} = 1.44$$

$$C_{\varepsilon 2} - C_{\varepsilon 1} = 1.92 - 1.44 = 0.48$$

The difference controls the spreading rate of jets.

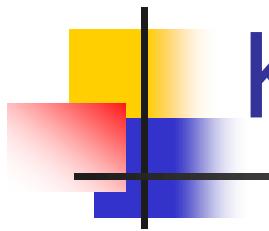
Final form of k- ε models

$$\frac{\partial \rho U_j k}{\partial x_j} = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_k} \right) \frac{\partial k}{\partial x_j} \right] + \underbrace{\mu_t \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) \frac{\partial U_i}{\partial x_j}}_{P_k} - \rho \varepsilon$$

$$\frac{\partial \rho U_j \varepsilon}{\partial x_j} = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_\varepsilon} \right) \frac{\partial \varepsilon}{\partial x_j} \right] + \frac{\varepsilon}{k} (C_{\varepsilon 1} P_k - C_{\varepsilon 2} \rho \varepsilon)$$

$$-\rho \overline{u_i u_j} = \mu_t \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) - \frac{2}{3} \delta_{ij} \rho k, \quad \mu_t = C_\mu \rho k^2 / \varepsilon$$

C_μ	$C_{\varepsilon 1}$	$C_{\varepsilon 2}$	σ_k	σ_ε
0.09	1.44	1.92	1.	1.3



K- ω model- Wilcox 1993

$$\frac{\partial \rho U_j k}{\partial x_j} = \frac{\partial}{\partial x_j} \left[(\mu + \frac{\mu_t}{\sigma_k}) \frac{\partial k}{\partial x_j} \right] + \underbrace{\mu_t \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) \frac{\partial U_i}{\partial x_j}}_{P_k} - \rho k \omega$$

$$\frac{\partial \rho U_j \omega}{\partial x_j} = \frac{\partial}{\partial x_j} \left[(\mu + \frac{\mu_t}{\sigma_\omega}) \frac{\partial \omega}{\partial x_j} \right] + C_\mu C_{\omega 1} \rho \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) \frac{\partial U_i}{\partial x_j} - C_{\omega 2} \rho \omega^2$$

$$\mu_t = C_\mu \rho \frac{k}{\omega}, \quad \varepsilon = k \omega$$

C_μ	$C_{\omega 1}$	$C_{\omega 2}$	σ_k	σ_ω
0.09	5/9	5/6	2	2

Low Reynolds number k- ϵ modeling

In the near wall region, the model has to be modified to account for the diminishing length scale

$$\frac{\partial U_j k}{\partial x_j} = \frac{\partial}{\partial x_j} \left[\left(\nu + \frac{\nu_t}{\sigma_k} \right) \frac{\partial k}{\partial x_j} \right] - \frac{1}{2} \frac{\partial}{\partial x_j} \left(\nu \frac{k}{\varepsilon} \frac{\partial \hat{\varepsilon}}{\partial x_j} \right)$$

$$- \overline{u_i u_j} \frac{\partial U_i}{\partial x_j} - (\tilde{\varepsilon} + \hat{\varepsilon})$$

$$\frac{\partial U_j \tilde{\varepsilon}}{\partial x_j} = \frac{\partial}{\partial x_j} \left[\left(\nu + \frac{\nu_t}{\sigma_\varepsilon} \right) \frac{\partial \tilde{\varepsilon}}{\partial x_j} \right] - \frac{\partial}{\partial x_j} \left(\nu \frac{\tilde{\varepsilon}}{k} \frac{\partial k}{\partial x_j} \right)$$

$$- C_{\varepsilon 1} f_1 \frac{\tilde{\varepsilon}}{k} \overline{u_i u_j} \frac{\partial U_i}{\partial x_j} - C_{\varepsilon 2} f_2 \frac{\tilde{\varepsilon}^2}{k}$$

Low Reynolds number modeling

$$\nu_t = (k^2/\tilde{\varepsilon}) C_\mu f_\mu(y_\lambda)$$

The coefficients and the damping functions are

$$C_\mu = 0.09, \quad C_{\varepsilon 1} = 1.44, \quad C_{\varepsilon 2} = 1.92$$

$$f_\mu = 1 - \exp(-0.01y_\lambda - 0.008y_\lambda^3)$$

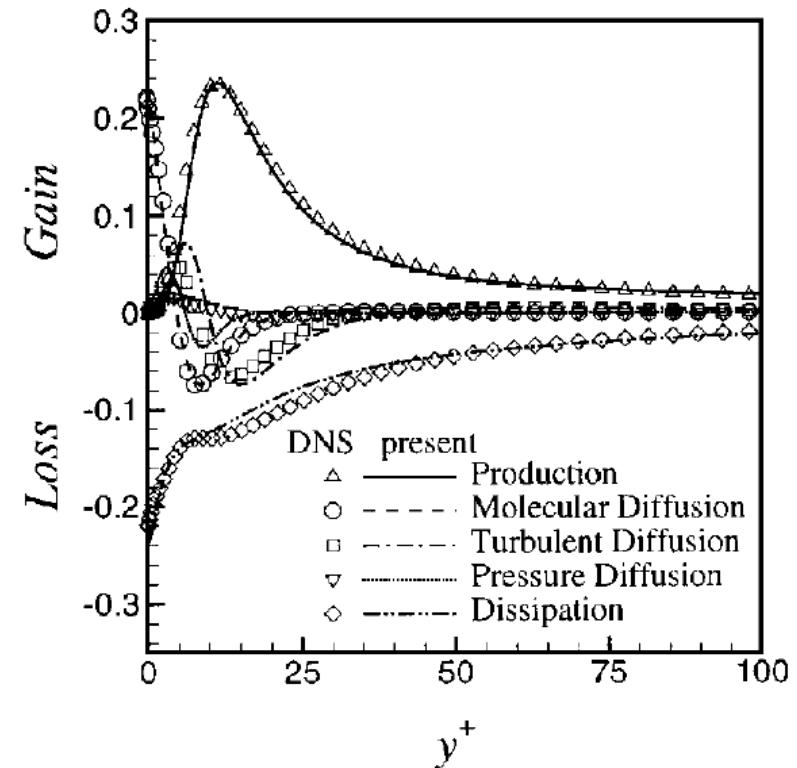
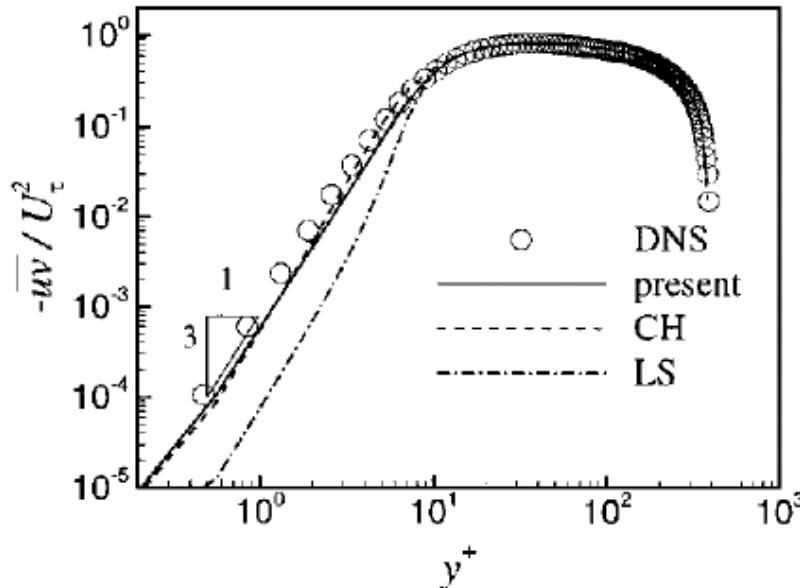
$$\sigma_k = 1.4 - 1.1 \exp[-(y_\lambda/10)]$$

$$\sigma_\varepsilon = 1.3 - 1.0 \exp[-(y_\lambda/10)]$$

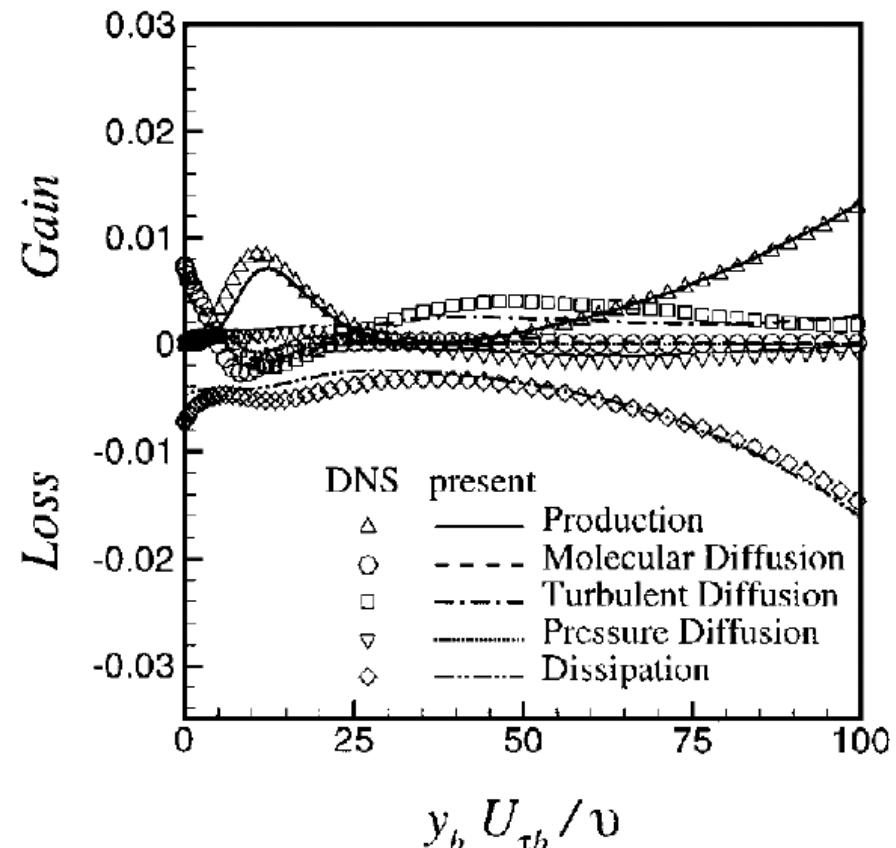
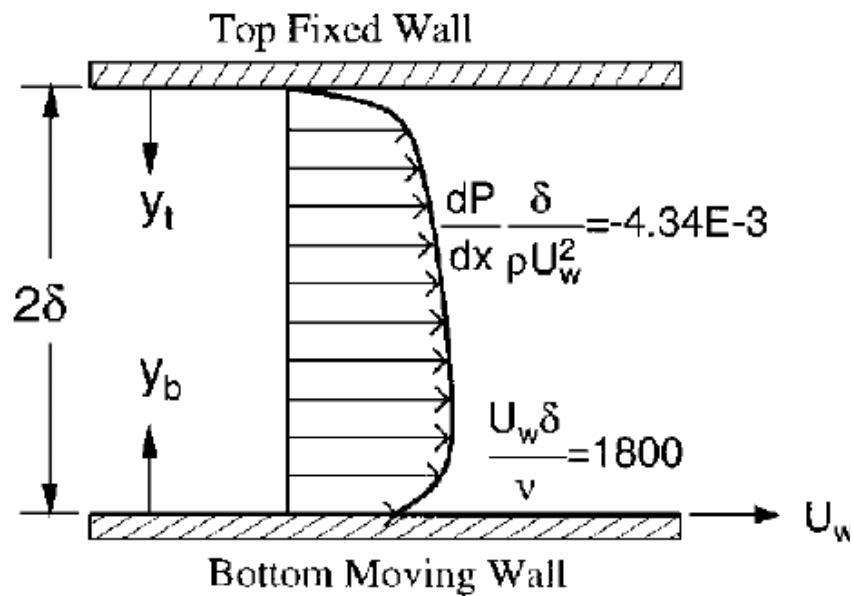
$$f_1 = 1, \quad f_2 = 1$$

Hwang, C. B. and Lin, C. A., 1998, "An improved Low-Reynolds-Number k-e model based on Direct Numerical Simulation Data", AIAA J., Vol. 36, No. 1, pp. 38-43.

Low Reynolds number modeling-Channel flow



Low Reynolds number modeling- Couette Poiseuille flow

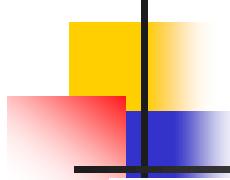


Reynolds stress model

$$\frac{\partial}{\partial x_j}(\rho U_i U_j) = -\frac{\partial P}{\partial x_i} + \frac{\partial}{\partial x_j}[\mu(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i}) - \rho \bar{u}_i \bar{u}_j]$$

■ Transport equation for Reynolds stress

$$\underbrace{\frac{\partial \rho U_k \bar{u}_i \bar{u}_j}{\partial x_k}}_{C_{ij}} = \underbrace{\frac{\partial}{\partial x_k} \left[-\rho \bar{u}_k \bar{u}_i \bar{u}_j - \rho \delta_{jk} \bar{u}_i p - \rho \delta_{ik} \bar{u}_j p + \mu \frac{\partial \bar{u}_i \bar{u}_j}{\partial x_k} \right]}_{d_{ij}} \\ \underbrace{- \rho \bar{u}_i \bar{u}_k \frac{\partial U_j}{\partial x_k} - \rho \bar{u}_j \bar{u}_k \frac{\partial U_i}{\partial x_k}}_{P_{ij}} + \underbrace{p \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)}_{\Phi_{ij}} - \underbrace{\mu \frac{\partial u_i}{\partial x_k} \frac{\partial u_j}{\partial x_k}}_{\varepsilon_{ij}}$$



Reynolds stress model

- Normal stresses clearly not describable by eddy-viscosity scheme.

$$-\rho\bar{uu} = 2\mu_t \frac{\partial U}{\partial x} - \frac{2}{3}\rho k$$

$$-\rho\bar{vv} = 2\mu_t \frac{\partial V}{\partial y} - \frac{2}{3}\rho k$$

$$-\rho\bar{ww} = 2\mu_t \frac{\partial W}{\partial z} - \frac{2}{3}\rho k$$

- Shear with streamline curvature

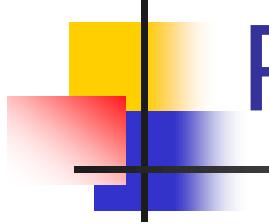
Wake curvature $\partial V / \partial x \sim 0.01 \partial U / \partial y$

Yet such small extra strains produce a significant effect on turbulence. Now

$$P_{\bar{uv}} = -\rho \left[\bar{uu} \frac{\partial V}{\partial x} + \bar{vv} \frac{\partial U}{\partial y} \right]$$

If eddy viscosity model is adopted, then

$$-\rho\bar{uv} = \mu_t \left[\frac{\partial V}{\partial x} + \frac{\partial U}{\partial y} \right]$$



Reynolds stress model

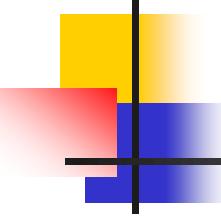
$$d_{ij} = \frac{\partial}{\partial x_k} \left(C_s \rho \frac{k}{\varepsilon} \overline{u_k u_\ell} \frac{\partial \overline{u_i u_j}}{\partial x_\ell} \right) \text{ or } \frac{\partial}{\partial x_k} \left(\frac{\mu_t}{\sigma} \frac{\partial \overline{u_i u_j}}{\partial x_k} \right)$$

– slow part

– rapid part

$$\Phi_{ij1} = -C_1 \frac{\varepsilon}{k} \left(\overline{u_i u_j} - \frac{2}{3} \delta_{ij} k \right) \quad \Phi_{ij2} = -C_2 \left(P_{ij} - \frac{2}{3} \delta_{ij} P_k \right)$$

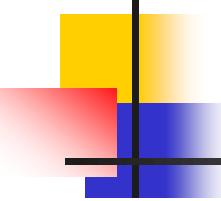
$$\varepsilon_{ij} = \frac{2}{3} \delta_{ij} \rho \varepsilon$$



Reynolds stress model

- The WET model of turbulence – Launder
 - Latter lectures focus on term-by-term approximations to the unknown fluctuating products in the second-moment equations.
 - The WET hypothesis avoids these complexities,
- Translated to turbulence. This means,
Value of 2nd moment \propto generation of 2nd moment \times turbulent time scale
- Direct application of WET hypothesis to Reynolds stress gives

$$\overline{u_i u_j} \propto P_{ij} T$$



Reynolds stress model

- Reinterpretation of WET model
 - Now in isotropic turbulence
 - It seems reasonable to suppose that WET hypothesis should apply to departure of the 2nd moment from their isotropic state.

$$(\overline{u_i u_j} - \frac{1}{3} \delta_{ij} \overline{u_k u_k}) = C_s T (P_{ij} - \frac{1}{3} \delta_{ij} P_{kk})$$

$$T = \frac{k}{\epsilon}$$

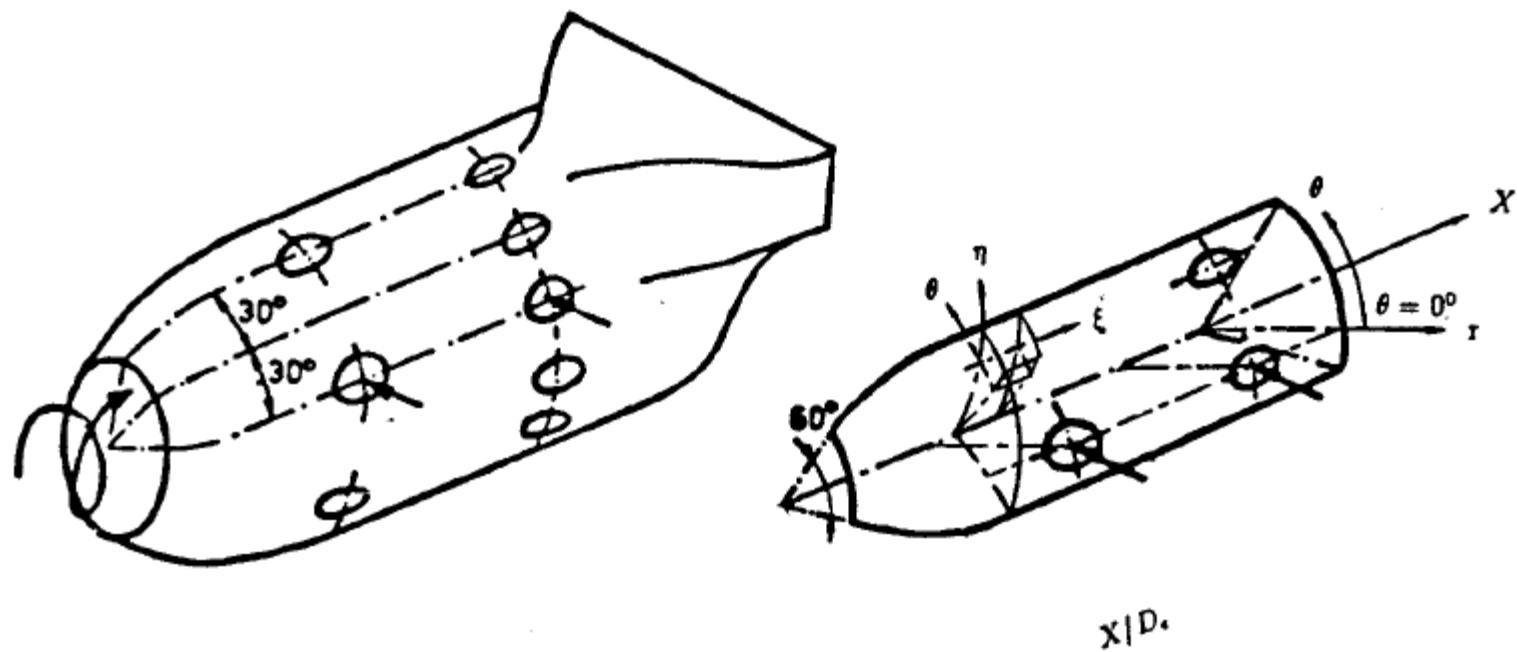
$$\phi_{ij} = \overline{p \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)}$$

$$a_{ij} = \frac{\overline{u_i u_j} - \frac{1}{3} \delta_{ij} \overline{u_k u_k}}{k} \propto \frac{P_{ij} - \frac{1}{3} \delta_{ij} P_{kk}}{\epsilon}$$

$$\phi_{ij1} = -1.8 \rho \frac{\epsilon}{k} (\overline{u_i u_j} - \frac{2}{3} \delta_{ij} k)$$

$$\phi_{ij2} = -\frac{C_2 + 8}{11} (P_{ij} - \frac{1}{3} \delta_{ij} P_{kk}) - \frac{30C_2 - 2}{55} \rho k \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) - \frac{8C_2 - 2}{11} (D_{ij} - \frac{1}{3} \delta_{ij} D_{kk})$$

Combustion Chamber



Combustion Chamber

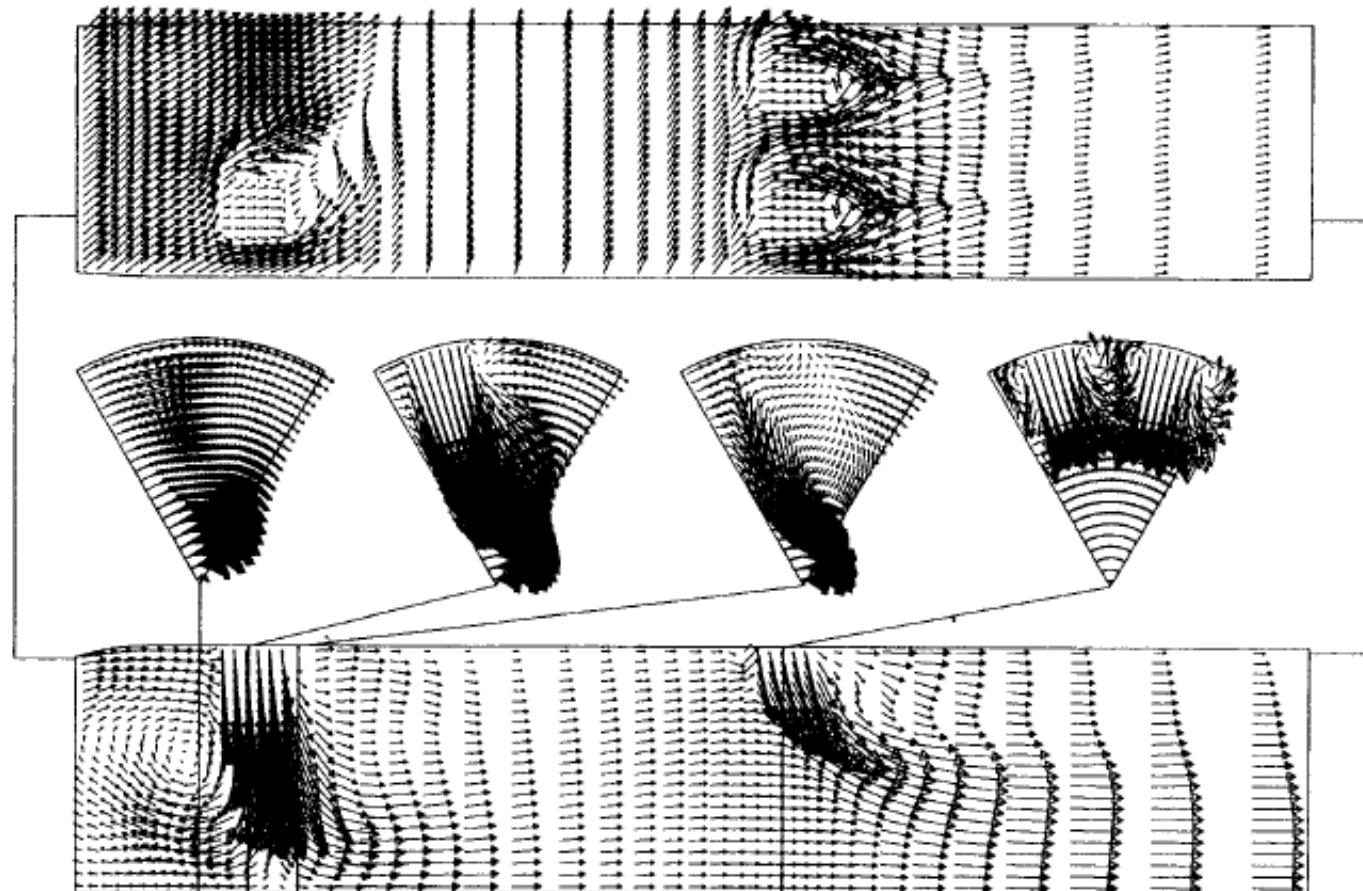
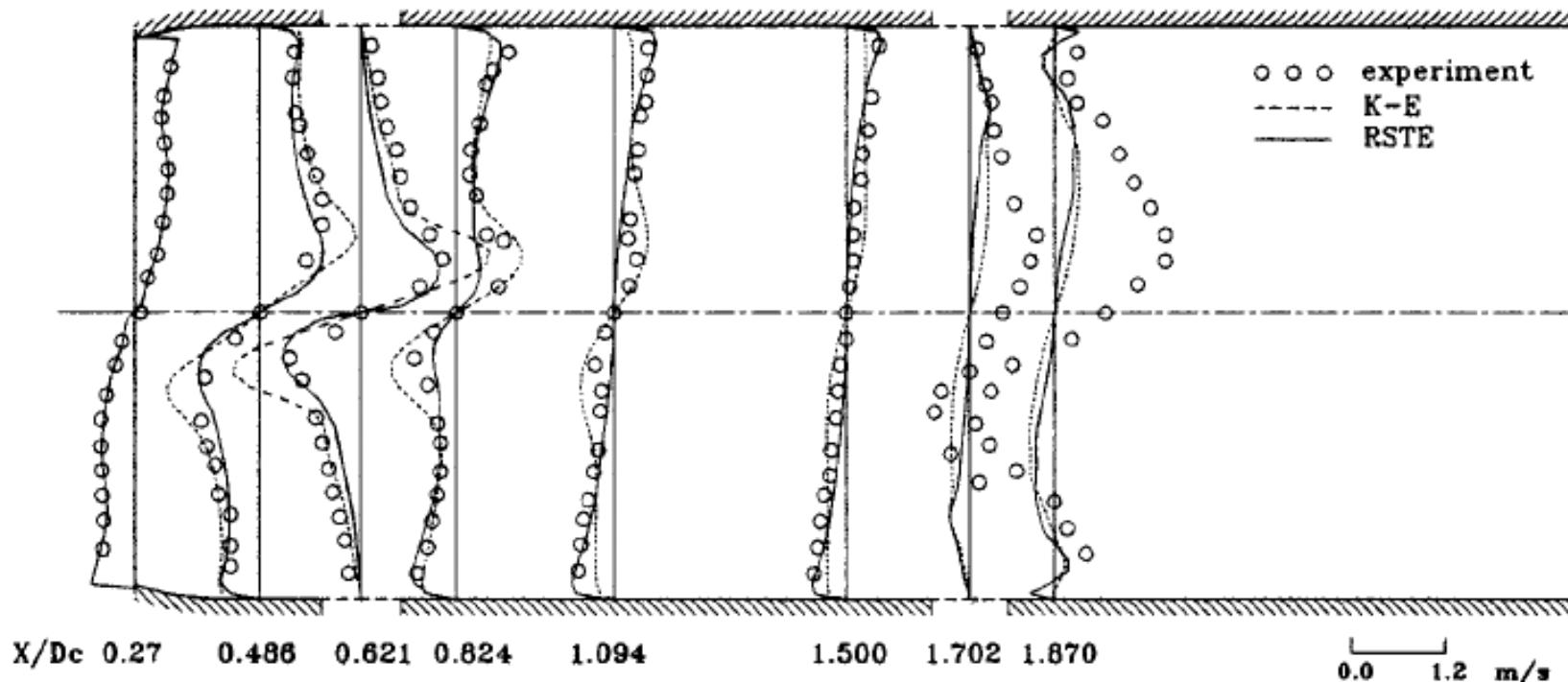
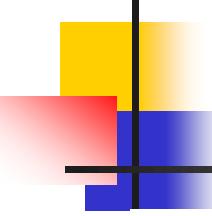


Figure 3. Overall view of the combustor flow—swirler 1.

Combustion Chamber



Lin, C. A. and Lu, C. M., 1994, "Modelling Three-Dimensional Gas-Turbine-Combustor-Model Flow using Second-Moment Closure", AIAA Journal, Vol. 32 no 7, pp. 1416-1422.



Swirling flow

INTERNATIONAL JOURNAL FOR NUMERICAL METHODS IN FLUIDS

Int. J. Numer. Meth. Fluids 30: 493–508 (1999)

COMPUTATIONS OF STRONGLY SWIRLING FLOWS WITH SECOND-MOMENT CLOSURES

J.C. CHEN AND C.A. LIN*

Department of Power Mechanical Engineering, National Tsing Hua University, Hsinchu, 30043, Taiwan

$$\frac{\partial(\rho U_i)}{\partial x_i} = 0,$$

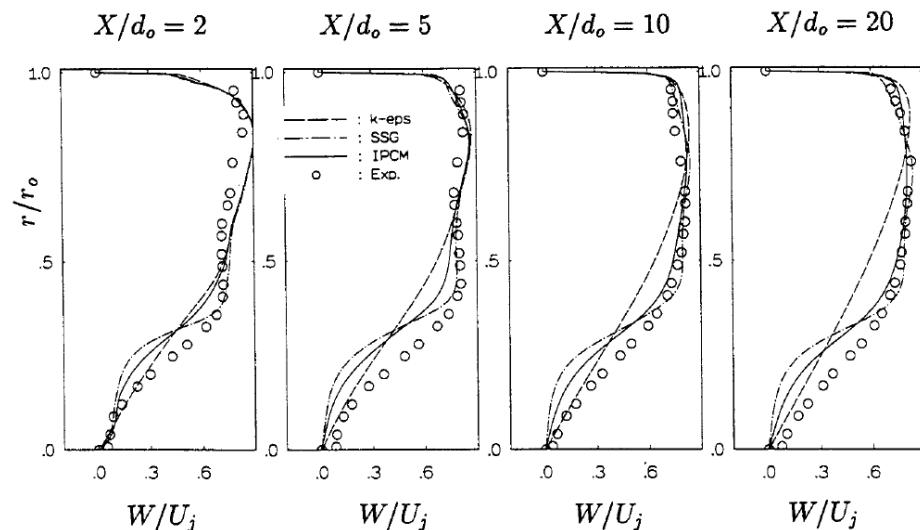
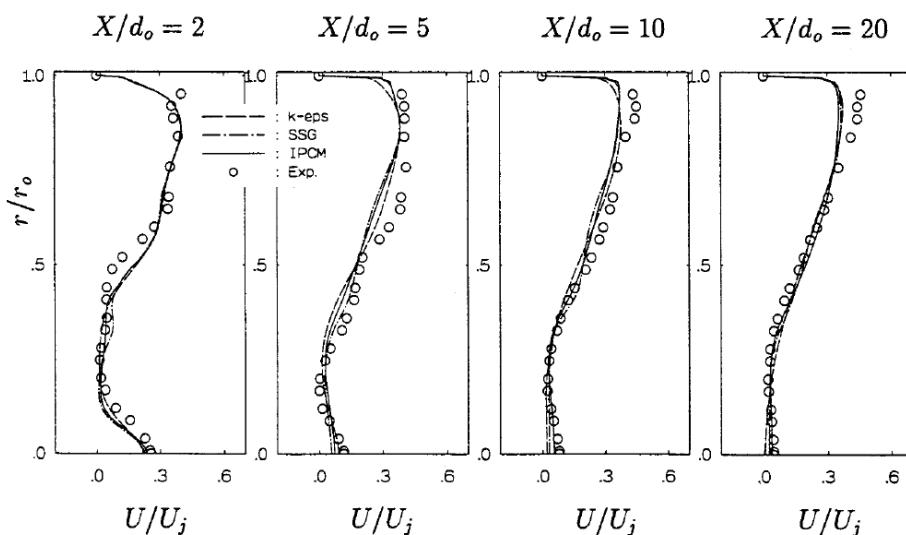
$$\frac{\partial(\rho U_i U_j)}{\partial x_j} = -\frac{\partial P}{\partial x_i} + \frac{\partial}{\partial x_j} \left[\mu_l \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) - \rho \bar{u}_i \bar{u}_j \right], \quad \frac{\partial}{\partial x_k} (\rho U_k \bar{u}_i \bar{u}_j) = d_{ij} + P_{ij} + \epsilon_{ij} + \phi_{ij},$$

Chen, J. C. and Lin, C. A., 1999, "Computations of Strongly Swirling Flows with Second-Moment Closure", *I. J. of Numerical Methods in Fluids*, Vol. 30, pp. 493-508.

$$S = \frac{\int_0^{r_o} UWr^2 dr}{r_o \int_0^{r_o} U^2 r dr}$$

Swirling flow

So et al.'s strongly swirling flow $S = 2.25$



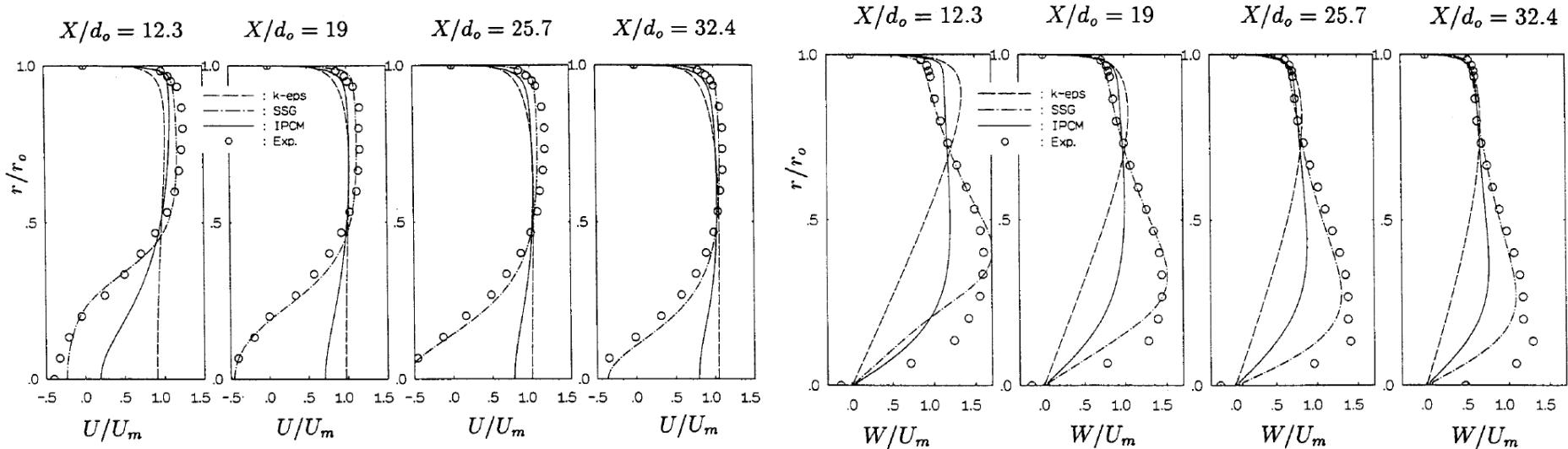
Chen, J. C. and Lin, C. A., 1999, "Computations of Strongly Swirling Flows with Second-Moment Closure", I. J. of Numerical Methods in Fluids, Vol. 30, pp. 493-508.

$$\phi_{ij1} = -(3.4\epsilon + 1.8P_k)b_{ij} + 4.2\epsilon \left(b_{ik}b_{kj} - \frac{1}{3}b_{kl}b_{kl}\delta_{ij} \right), \quad (12)$$

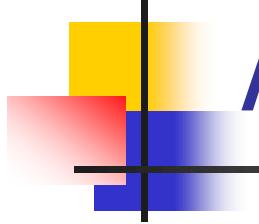
$$\phi_{ij2} = (0.8 - 1.3\sqrt{b_{kl}b_{kl}})kS_{ij} + 1.25k \left(b_{ik}S_{jk} + b_{jk}S_{ik} - \frac{2}{3}b_{kl}S_{kl}\delta_{ij} \right) + 0.4k(b_{ik}W_{jk} + b_{jk}W_{ik}), \quad (13)$$

Swirling flow

Kitoh's strongly swirling flow $S = 0.85$



Chen, J. C. and Lin, C. A., 1999, "Computations of Strongly Swirling Flows with Second-Moment Closure", I. J. of Numerical Methods in Fluids, Vol. 30, pp. 493-508.



Algebraic stress modeling

$$\underbrace{\frac{\partial}{\partial x_k} (U_k \overline{u_i u_j})}_{C_{ij}} - \underbrace{\frac{\partial}{\partial x_k} \left[\nu \frac{\partial \overline{u_i u_j}}{\partial x_k} - \overline{u_i u_j u_k} - \frac{1}{\rho} (\overline{p u_i} \delta_{jk} + \overline{p u_j} \delta_{ik}) \right]}_{d_{ij}}$$
$$= \underbrace{-\overline{u_i u_k} \frac{\partial U_j}{\partial x_k} - \overline{u_j u_k} \frac{\partial U_i}{\partial x_k}}_{P_{ij}} + \underbrace{\frac{p}{\rho} \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)}_{\phi_{ij}} - \underbrace{\left(\frac{2}{3} \delta_{ij} \varepsilon + \varepsilon_{ij,D} \right)}_{\varepsilon_{ij}}$$

ASM-linearization

$$\underbrace{\frac{\partial}{\partial x_j}(U_j k)}_{C_k} - \underbrace{\frac{\partial}{\partial x_j} \left(v \frac{\partial k}{\partial x_j} - \frac{1}{2} \overline{u_i u_i u_j} - \frac{1}{\rho} \overline{p u_j} \right)}_{d_k} = \underbrace{-\overline{u_i u_j} \frac{\partial U_i}{\partial x_j}}_P - \varepsilon$$

$$C_{ij} - d_{ij} = \frac{\overline{u_i u_j}}{k} (C_k - d_k) = \frac{\overline{u_i u_j}}{k} (P - \varepsilon)$$

assumption

$$\frac{\overline{u_i u_j}}{k} (P - \varepsilon) = P_{ij} - \frac{2}{3} \delta_{ij} \varepsilon + \underbrace{\phi_{ij} - \varepsilon_{ij,D}}_{\pi_{ij}}$$

Rodi W. The prediction of free turbulent boundary layers by use of a two-equation model of turbulence.
Ph.D. Thesis, University of London, 1972.

$$b_{ij} = \left[\frac{\overline{u_i u_j}}{2k} - \frac{1}{3} \delta_{ij} \right]$$

ASM-linearization

$$\frac{\overline{u_i u_j}}{k} (P - \varepsilon) = P_{ij} - \frac{2}{3} \delta_{ij} \varepsilon + \underbrace{\phi_{ij} - \varepsilon_{ij,D}}_{\pi_{ij}}$$

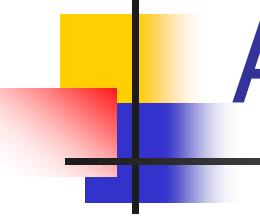
$$\pi_{ij} = - \left(C_1^0 + C_1^1 \frac{P}{\varepsilon} \right) \varepsilon b_{ij} + C_2 \varepsilon S_{ij} + C_3 \varepsilon \left(b_{ik} S_{jk} + b_{jk} S_{ik} - \frac{2}{3} b_{mn} S_{mn} \delta_{ij} \right)$$

$$+ C_4 \varepsilon (b_{ik} W_{jk} + b_{jk} W_{ik})$$

$$S_{ij} = \frac{1}{2} \frac{k}{\varepsilon} \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right), \quad W_{ij} = \frac{1}{2} \frac{k}{\varepsilon} \left(\frac{\partial U_i}{\partial x_j} - \frac{\partial U_j}{\partial x_i} \right)$$

$$S = \sqrt{2 S_{ij} S_{ij}}, \quad \Omega = \sqrt{2 W_{ij} W_{ij}}$$

$$b_{ij} \left[(C_1^0 - 2) + (C_1^1 + 2) \frac{P}{\varepsilon} \right] = \left(C_2 - \frac{4}{3} \right) S_{ij} + (C_3 - 2) \left(b_{ik} S_{jk} + b_{jk} S_{ik} - \frac{2}{3} b_{mn} S_{mn} \delta_{ij} \right) \\ + (C_4 - 2) (b_{ik} W_{jk} + b_{jk} W_{ik})$$



ASM-linearization

$$b_{ij} \left[(C_1^0 - 2) + (C_1^1 + 2) \frac{P}{\varepsilon} \right] = \left(C_2 - \frac{4}{3} \right) S_{ij} + (C_3 - 2) \left(b_{ik} S_{jk} + b_{jk} S_{ik} - \frac{2}{3} b_{mn} S_{mn} \delta_{ij} \right) \\ + (C_4 - 2) (b_{ik} W_{jk} + b_{jk} W_{ik})$$

Explicit solution: Gatski and Speziale (1993)

$$b_{ij} = - \frac{3}{3 - 2\eta_1 - 6\eta_2} \left[\beta_1 S_{ij} + \beta_2 (S_{ik} W_{kj} + S_{jk} W_{ki}) - 2\beta_3 \left(S_{ik} S_{kj} - \frac{1}{3} S_{kl} S_{kl} \delta_{ij} \right) \right]$$

where $\beta_1 = (4/3 - C_2)/\beta_4$, $\beta_2 = \beta_1(2 - C_4)/\beta_4$, $\beta_3 = \beta_1(2 - C_3)/\beta_4$, $\beta_4 = (C_1^0 - 2) + (C_1^1 + 2)P/\varepsilon$, $\eta_1 = \xi_1(2 - C_3)^2/\beta_4^2$ and $\eta_2 = \xi_2(2 - C_4)^2/\beta_4^2$. $\xi_1 = S_{ij} S_{ij}$ and $\xi_2 = W_{ij} W_{ij}$.

Non-linear eddy viscosity model

$$b_{ij} = \left[\frac{\overline{u_i u_j}}{2k} - \frac{1}{3} \delta_{ij} \right]$$

Craft TJ, Launder BE, Suga K. Prediction of turbulent transitional phenomena with a nonlinear eddy-viscosity model. International Journal of Heat and Fluid Flow 1997; 18:15–28.

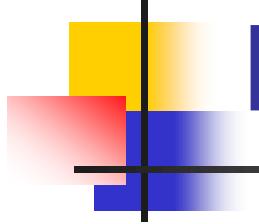
$$= -C_\mu S_{ij} + C_1 C_\mu \left[S_{ik} S_{kj} - \frac{1}{3} \delta_{ij} S_{kl} S_{kl} \right] + C_2 C_\mu [W_{ik} S_{kj} + W_{jk} S_{ki}]$$

$$+ C_3 C_\mu \left[W_{ik} W_{kj} - \frac{1}{3} \delta_{ij} W_{kl} W_{kl} \right] + C_4 C_\mu^2 \left(S_{ki} W_{lj} + S_{kj} W_{li} - \frac{2}{3} S_{km} W_{lm} \delta_{ij} \right) S_{kl}$$

$$+ C_6 C_\mu^2 S_{ij} S_{kl} S_{kl} + C_7 C_\mu^2 S_{ij} W_{kl} W_{kl}$$

$$S_{ij} = \frac{1}{2} \frac{k}{\varepsilon} \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right), \quad W_{ij} = \frac{1}{2} \frac{k}{\varepsilon} \left(\frac{\partial U_i}{\partial x_j} - \frac{\partial U_j}{\partial x_i} \right)$$

$$S = \sqrt{2S_{ij}S_{ij}}, \quad \Omega = \sqrt{2W_{ij}W_{ij}}$$



Large eddy simulation

– Filtering operation

$$\widetilde{f(\vec{x})} = \int f(\vec{x}) G(\vec{x}, \vec{x}') d\vec{x}'$$

$$\frac{\partial \rho \widetilde{U}_i}{\partial x_i} = 0$$

$$\frac{\partial \rho \widetilde{U}_i}{\partial t} + \frac{\partial \rho \widetilde{U}_j \widetilde{U}_i}{\partial x_j} = - \frac{\partial \tilde{P}}{\partial x_i} + \frac{\partial}{\partial x_j} \left[\mu \left(\frac{\partial \widetilde{U}_i}{\partial x_j} + \frac{\partial \widetilde{U}_j}{\partial x_i} \right) - \tau_{ij} \right]$$

$$\tau_{ij} = \rho \widetilde{U}_i \widetilde{U}_j - \rho \widetilde{U}_j \widetilde{U}_i$$

Table 13.2. *Filter functions and transfer functions for one-dimensional filters: the box filter function has the same second moment as the Gaussian ($\frac{1}{12}\Delta^2$); the other filters have the same value of the transfer function at the characteristic wavenumber $\kappa_c \equiv \pi/\Delta$, i.e., $\widehat{G}(\kappa_c) = \exp(-\pi^2/24)$*

Name	Filter function	Transfer function
General	$G(r)$	$\widehat{G}(\kappa) \equiv \int_{-\infty}^{\infty} e^{ikr} G(r) dr$
Box	$\frac{1}{\Delta} H(\frac{1}{2}\Delta - r)$	$\frac{\sin(\frac{1}{2}\kappa\Delta)}{\frac{1}{2}\kappa\Delta}$
Gaussian	$\left(\frac{6}{\pi\Delta^2}\right)^{1/2} \exp\left(-\frac{6r^2}{\Delta^2}\right)$	$\exp\left(-\frac{\kappa^2\Delta^2}{24}\right)$
Sharp spectral	$\frac{\sin(\pi r/\Delta)}{\pi r}$	$H(\kappa_c - \kappa),$ $\kappa_c \equiv \pi/\Delta$
Cauchy	$\frac{a}{\pi\Delta[(r/\Delta)^2 + a^2]}, \quad a = \frac{\pi}{24}$	$\exp(-a\Delta \kappa)$
Pao		$\exp\left(-\frac{\pi^{2/3}}{24}(\Delta \kappa)^{4/3}\right)$

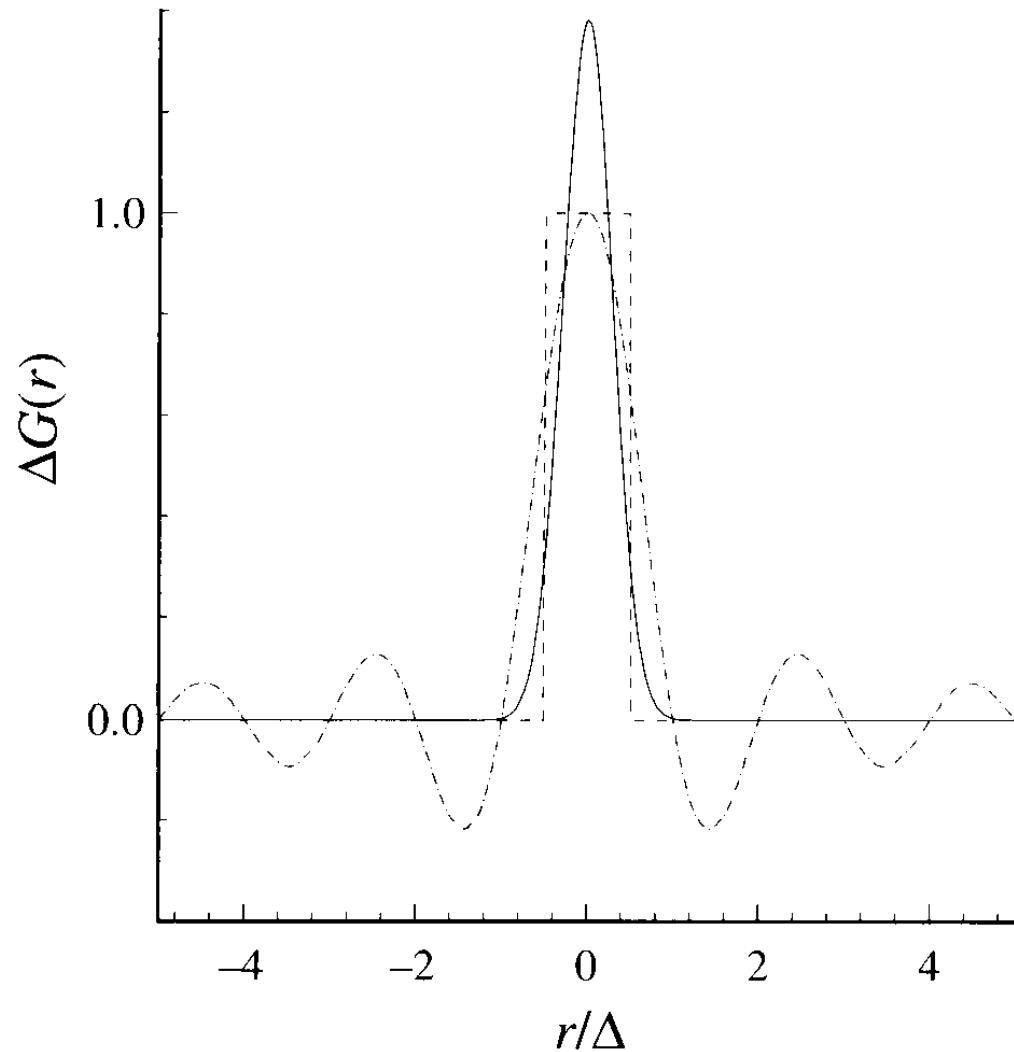


Fig. 13.1. Filters $G(r)$: box filter, dashed line; Gaussian filter, solid line; sharp spectral filter, dot-dashed line.

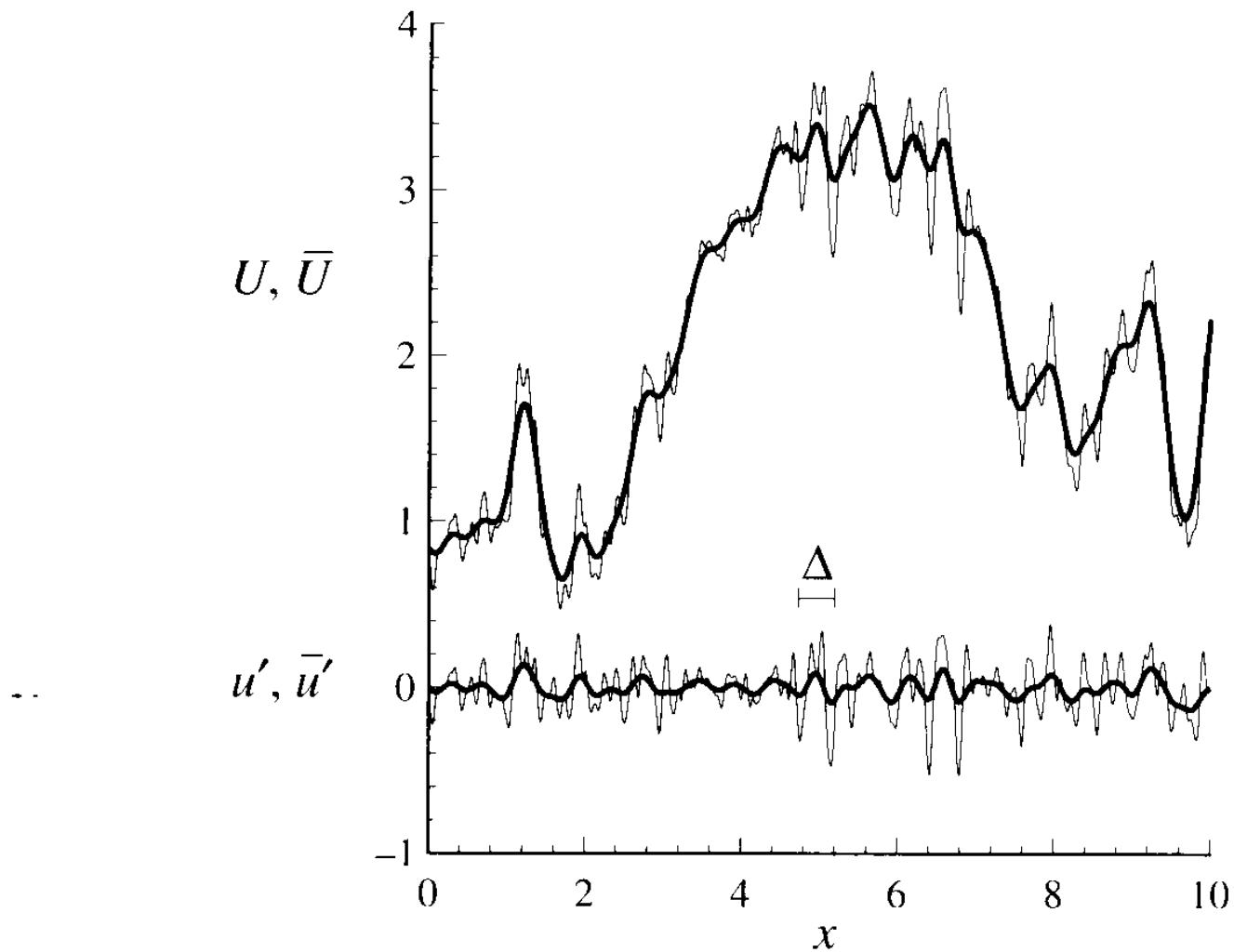
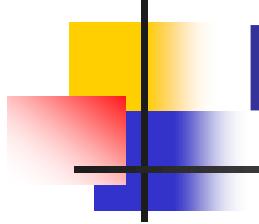


Fig. 13.2. Upper curves: a sample of the velocity field $U(x)$ and the corresponding filtered field $\bar{U}(x)$ (bold line), using the Gaussian filter with $\Delta \approx 0.35$. Lower curves: the residual field $u'(x)$ and the filtered residual field $\bar{u}'(x)$ (bold line).



Large eddy simulation

$$\frac{\partial \rho \tilde{U}_i}{\partial t} + \frac{\partial \rho \tilde{U}_j \tilde{U}_i}{\partial x_j} = -\frac{\partial \tilde{P}}{\partial x_i} + \frac{\partial}{\partial x_j} \left[\mu \left(\frac{\partial \tilde{U}_i}{\partial x_j} + \frac{\partial \tilde{U}_j}{\partial x_i} \right) - \tau_{ij} \right]$$

– Smagorinsky model (eddy viscosity model)

$$\tau_{ij} - \frac{\delta_{ij}}{3} \tau_{kk} = -\mu_t \widetilde{S}_{ij} = -\mu_t \left(\frac{\partial \tilde{U}_i}{\partial x_j} + \frac{\partial \tilde{U}_j}{\partial x_i} \right)$$

$$\mu_t = (C_S \Delta)^2 |\tilde{S}|, \quad \tilde{S} = (\widetilde{S}_{ij}/2)^2$$
$$C_S \approx 0.1, \quad \Delta$$

$$= (\Delta_x \Delta_y \Delta_z)^{1/3} \text{ or } \sqrt{\Delta_x \Delta_y + \Delta_x \Delta_z + \Delta_y \Delta_z}$$

LES-dynamic model

$$\hat{\Delta} = 2\tilde{\Delta}$$

$$\tau_{ij} - \frac{\delta_{ij}}{3}\tau_{kk} = -2C\tilde{\Delta}^2|\tilde{S}|\tilde{S}_{ij} = -2C\beta_{ij}$$

$$\tau_{ij} = \widetilde{u_i u_j} - \tilde{u}_i \tilde{u}_j$$

$$T_{ij} - \frac{\delta_{ij}}{3}T_{kk} = -2C\hat{\Delta}^2|\hat{S}|\hat{S}_{ij} = -2C\alpha_{ij}$$

$$T_{ij} = \widehat{\widetilde{u_i u_j}} - \widehat{\tilde{u}_i \tilde{u}_j}$$

Six equations, one unknown

$$\begin{aligned} L_{ij} &= T_{ij} - \widehat{\tau_{ij}} \\ &= \widehat{\widetilde{u_i u_j}} - \widehat{\tilde{u}_i \tilde{u}_j} - \widehat{\widetilde{u_i u_j}} + \widehat{\tilde{u}_i \tilde{u}_j} \\ &= \widehat{\tilde{u}_i \tilde{u}_j} - \widehat{\tilde{u}_i \tilde{u}_j} \end{aligned}$$

$$\begin{aligned} L_{kk} &= T_{kk} - \widehat{\tau_{kk}} \\ &= \widehat{\widetilde{u_k u_k}} - \widehat{\tilde{u}_k \tilde{u}_k} \end{aligned}$$

$$\begin{aligned} L_{ij}^a &= L_{ij} - \frac{\delta_{ij}}{3}L_{kk} \\ &= -2C\alpha_{ij} + 2C\widehat{\beta_{ij}} \\ &= -2C\hat{\Delta}^2|\hat{S}|\hat{S}_{ij} + 2C\tilde{\Delta}^2|\tilde{S}|\tilde{S}_{ij} \\ &= 2CM_{ij} \end{aligned}$$

Lilly, DK, 1992, A proposed modification of the Germano subgrid-scale closure method, Phys. Fluids 4, 633–635 (Citation: 5797)

LES-dynamic model

Lilly (1992)-least square error to determine C

$$\epsilon = (L_{ij}^a - 2CM_{ij})^2$$

$$\frac{1}{2} \frac{\partial \epsilon}{\partial C} = -M_{ij}(L_{ij}^a - 2CM_{ij}) = 0$$

$$C = \frac{L_{ij}^a M_{ij}}{M_{kl} M_{kl}} = \frac{L_{ij} M_{ij}}{M_{kl} M_{kl}}$$

Average in the periodic direction to prevent instability

$$C(x, t) = \frac{\langle L_{ij} M_{ij} \rangle}{\langle M_{kl} M_{kl} \rangle}$$

$$\begin{aligned} L_{ij}^a &= L_{ij} - \frac{\delta_{ij}}{3} L_{kk} \\ &= -2C\alpha_{ij} + 2\widehat{C\beta_{ij}} \\ &= -2C\widehat{\Delta}^2 |\widehat{S}| \widehat{S}_{ij} + 2C\widetilde{\Delta}^2 |\widetilde{S}| \widetilde{S}_{ij} \\ &= 2CM_{ij} \end{aligned}$$

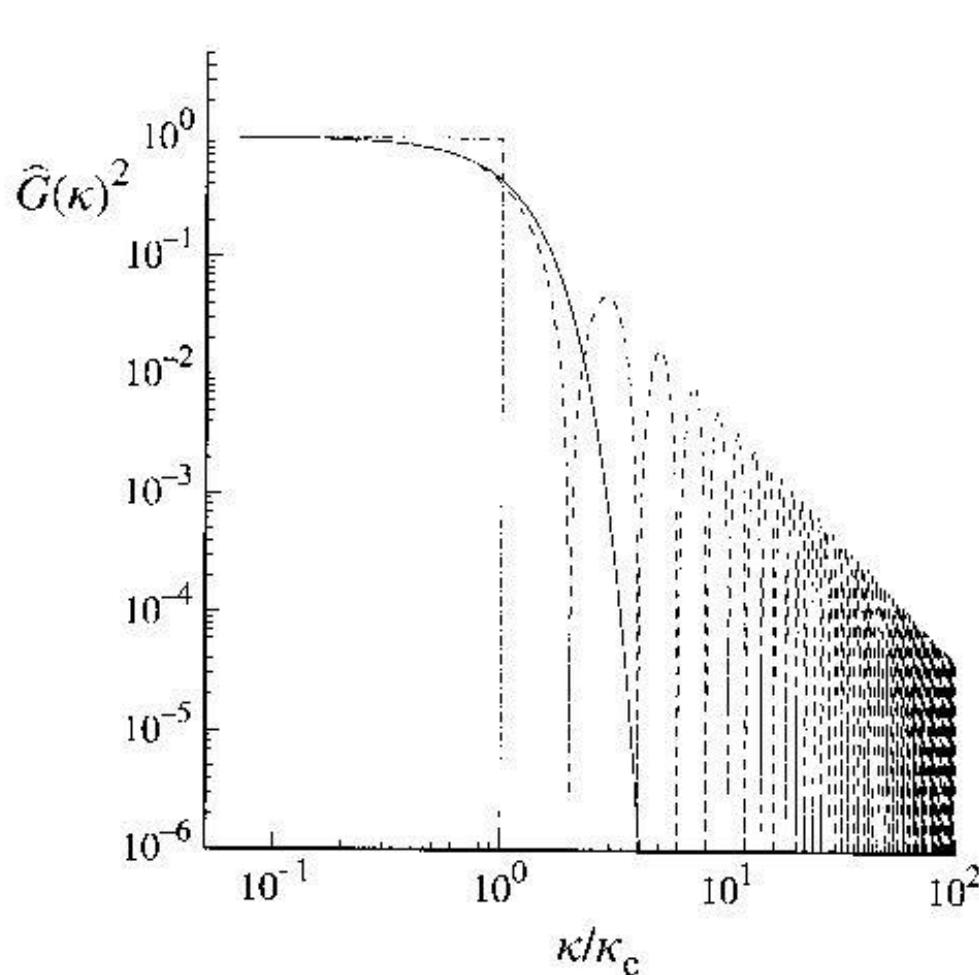


Fig. 13.4. Attenuation factors $\hat{G}(\kappa)^2$: box filter, dashed line; Gaussian filter, solid line; arp spectral filter, dot-dashed line.