

# Wing-Body Lift

Wing-Body Lift -  $L_{wb}$  (daN)

2000  
1500  
1000  
500  
0  
-500  
-1000

15

20

25

30

35

40

45

50

55

60

65

70

Airspeed -  $V$  (m/s)

Point S

Point S inv

Point G

Point F

Point C

Point D

Point E

