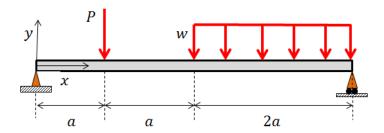
AE323 – Homework Assignment #2 – Spring 2019 Wednesday, Jan. 30, 2019 Due on Friday, Feb. 8 at class time

Topics: Internal Forces and Moments in Statically Determinate Structures

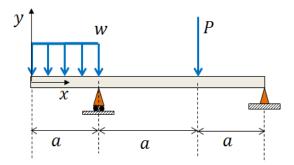
For the three structural problems shown below, answer the following three questions:

- a) Is the problem statically determinate or indeterminate, and why?
- b) Compute the reactions at the supports
- c) Find the expression of the resultant shear force $V_{\nu}(x)$ and bending moment $M_{z}(x)$ in the beam.

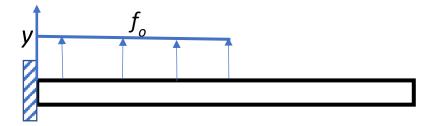
Problem 1. (Note: P is in N, w is in N/m)



Problem 2.



<u>Problem 3.</u> (Taken from the Fall 2017 final exam): The cantilever beam of length L is subjected to a transverse load f_o (given in N/m) applied over the first half of the beam.

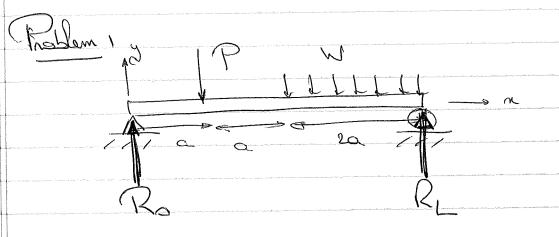


Problem 4. Assuming that the lift distribution over a wing of length L is quadratic, as in

$$l(x) = l_0 \left(1 - \left(\frac{x}{L} \right)^2 \right),$$

with l_0 given in N/m, and assuming that the wing is cantilever to the fuselage,

- a) Compute the reactions at the fuselage
- b) Compute the distribution of the resultant shear force $(V_z(x))$ and resultant bending moment $(M_y(x))$. Put your solution in a non-dimensional form (i.e., as a function of (x/L)), and plot the two non-dimensional solutions for $0 \le x/L \le 1$.
- c) Check that your solution found in b) matches the reactions found in a)
- d) Check also that your solution found in b) corresponds to your expectation at the end of the wing (i.e., at x = L)



(a) 2 reactions: Ro, RL 2 agrs: EE-0 \(\Sigma \sqrt{2}\) = 0

as statically determinate

(b) ZF=0 == Ro+Re=P+2aw
4a
ZMzl-0 == 4aRe=Pa+ lwxdx

= Pa + w(2a)(3a)

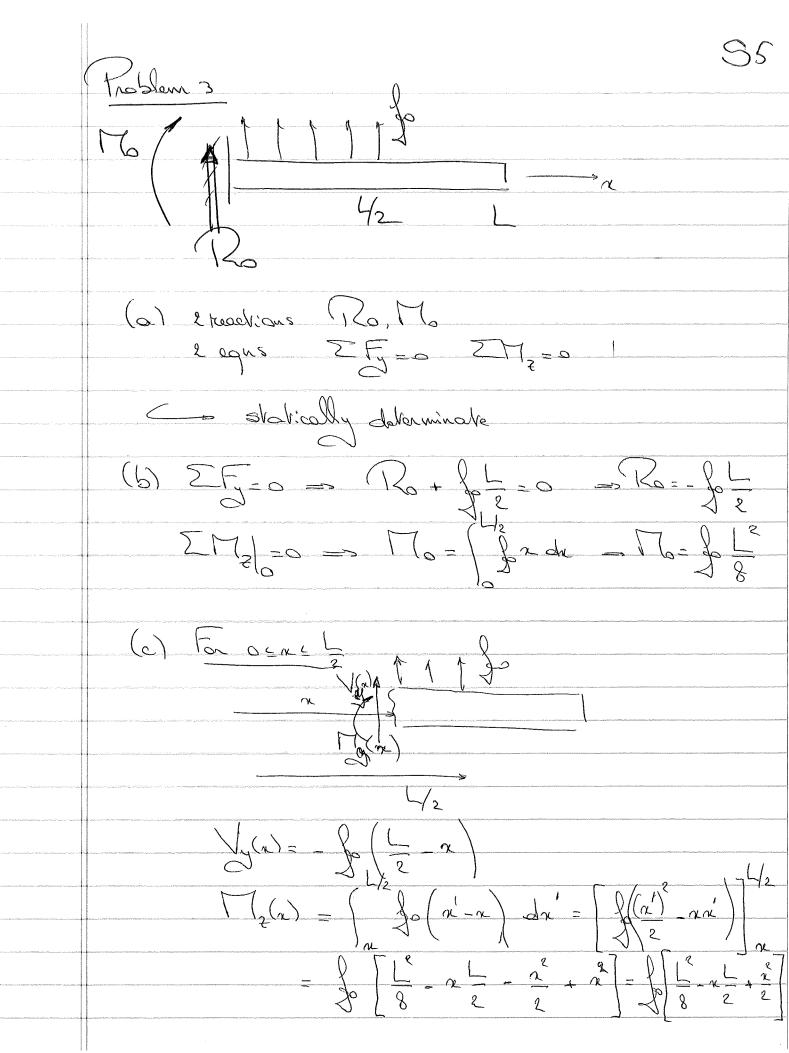
R=P + 3aw R-P + 2aw-R=3P aw 4 2

(c) We need to "at" the beam at 3 books ans

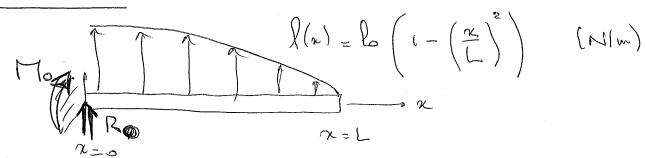
2 igns: ZF=0 Statically determinate (b) ZF=0 => R,+R2=P+aw ZP1/21=0 => Pa+wa 5a=2aR, = P + Saw R=P+au-R=P-au le again meed to cut the beam at 3 localions in = - War 12

، نو،

$$\frac{\alpha \leq \alpha \leq 2\alpha}{M_2(n)} = \frac{1}{N_2(n)} = \frac{1}{N_2(n$$



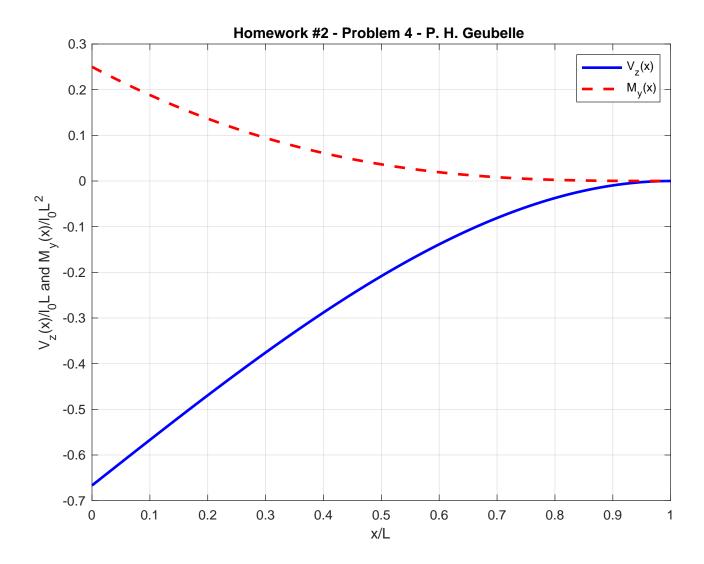
Problem 4.



Shear and Moment Liagrams

My(x) / 1x

$$\frac{1}{2}(x) = -\int_{\alpha}^{1} \left[\frac{1}{x^{2}} \right] dx = -\int_{\alpha}^{1} \left[\frac{1}{x^{2}$$



e At x=0, we find from the solution in b) $V_{2}^{(0)} = -\frac{2}{3} l_{0} l_{0}$ $V_{3}^{(0)} = -\frac{2}{3} l_{0} l_{0}$ Assume as part a) $V_{3}^{(0)} = l_{0} l_{0}^{2}$

· At n=L, we expect to find $V_2(L)=T_3(L)=0$ Indeed, that's what the solution b) evaluated out N=L is.

.

yer e e e

.

and good towards a