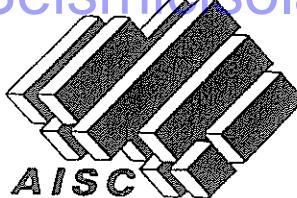
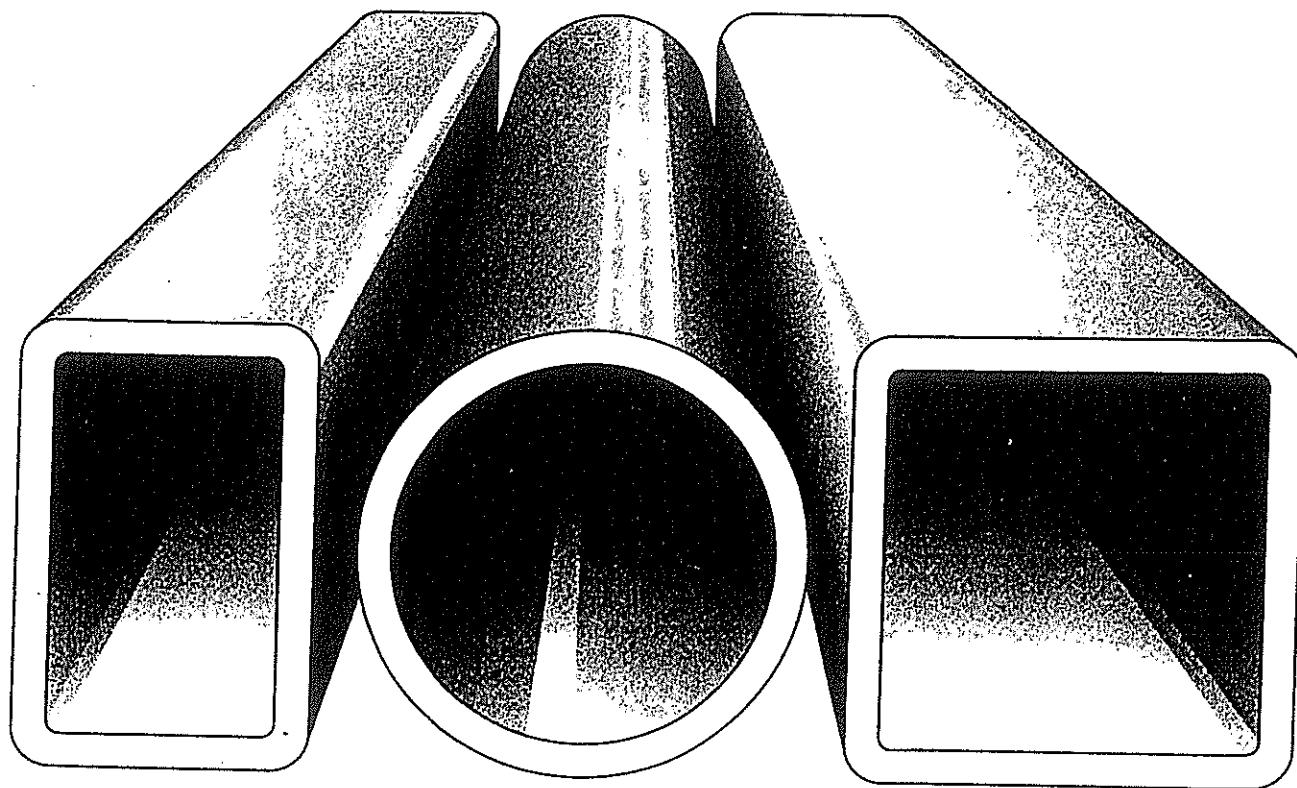


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# design capacity tables for structural steel hollow sections

1st edition



AUSTRALIAN INSTITUTE OF STEEL CONSTRUCTION  
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LIMIT STATES  
EDITION TO  
AS 4100-1990  
 $S_{v} \leq \phi R_u$

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# DESIGN CAPACITY TABLES FOR STRUCTURAL STEEL HOLLOW SECTIONS

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## Foreword

The Australian Institute of Steel Construction (AISC) was formed in 1962 to serve the interests of all concerned in the design, specification, fabrication, and erection of steel structures.

Today, the AISC is a national body whose purpose is to promote and optimise the use of fabricated structural steel through engineering, research, and the dissemination of knowledge to specifiers, fabricators, and suppliers.

Its activities are directed towards fostering the economic and efficient use of steel in buildings, bridges, civil engineering projects, pressure vessels, pipelines, marine and offshore structures, and other fabricated forms.

The Institute is a non-profit organisation which is primarily financed by membership dues.

Engineers, architects, and others interested in steel construction are invited to make use of the Institute's services.

The Design Capacity Tables for Structural Steel Hollow Sections were prepared for publication by the Tubular Structures Development Group to aid designers in the selection and economical use of hollow sections in steel structures.

The Tubular Structures Development Group consists of the AISC, Tubemakers of Australia Limited, Structural Products Division, and Palmer Tube Mills (Aust.) Pty Ltd. The aim of the group is to encourage the efficient use of steel hollow sections in construction.

## Acknowledgments

In the compilation of the *Design Capacity Tables for Structural Steel Hollow Sections - First Edition*, AISC acknowledges with grateful thanks the contribution and assistance of the following organisations and individuals:

- AISC – Technical Services
- Tubemakers of Australia Limited, Structural Products Division
- Palmer Tube Mills (Aust.) Pty Ltd
- and all those who gave constructive comment and advice on the technical and editorial content of the document.

## Preface

The first edition of the *Design Capacity Tables for Structural Steel Hollow Sections* (DCTHS) is to be used as a design aid when complying with the limit states code AS 4100-1990: *Steel Structures*, published by Standards Australia.

This publication supplements the *AISC Design Capacity Tables for Structural Steel* (DCT). The DCTHS presents similar data as that included in the DCT for hollow sections complying with the material standard AS 1163-1991: *Structural Steel Hollow Sections*. Significant additions in this publication include:

- a method to convert the tabulated limit states design capacities to safe working loads
- section properties for hollow sections with a wall thickness less than 3 mm and design capacities for designing to the member design provisions of AS 4100. (Hollow sections less than 3 mm thick are not considered in the DCT.)
- tables for telescoping hollow sections
- a set of tables for all design section capacities
- maximum design load tables for strength and serviceability limit states including simply supported, continuous, fixed end, and cantilever beams

Improved table presentation coupled with informative text on the use of this publication will enhance the use of structural steel hollow sections when designing to AS 4100.

AISC 1992.

## INTRODUCTION

### Steel Structures Standard

The tables in this publication have been calculated in conformity with the requirements of the Australian Standard AS 4100–1990: Steel Structures (published by Standards Australia), and as far as possible, the nomenclature and symbols used throughout are the same as those adopted in this Standard. In these notes and in the tables:

- "the Steel Structures Code" (or simply "the Code") and AS 4100 refer to AS 4100–1990
- "the Tables" or "Tables" refer to this edition of the Design Capacity Tables for Structural Steel Hollow Sections.

Cold-formed hollow sections manufactured in accordance with Australian Standard AS 1163–1991: Structural Steel Hollow Sections are included within the scope of AS 4100. Extensive research<sup>[1], [2], [3]</sup> undertaken over a number of years has confirmed that cold-formed hollow sections meet the inherent requirements of AS 4100.

Clause 1.1 of AS 4100 excludes material other than packers less than 3 mm thick. For sections less than 3 mm thick designers are required to refer to AS 1538–1988: Cold-Formed Steel Structures Code, which at this stage is still based on a permissible stress method of design.

A range of tests<sup>[1], [2], [3]</sup> conducted on hollow sections less than 3 mm thick have shown that the member design rules in AS 4100 can be used to determine the behaviour of these thin structural hollow sections manufactured to AS 1163.

In order to provide a consistent approach to design in the tables, hollow sections less than 3 mm thick, are presented in these tables as complying with the design rules appearing in AS 4100 although this code does not cover sections less than 3 mm thick. Therefore designers should where necessary, satisfy themselves as to the adequacy of members less than 3 mm thick, by verification as outlined in AS 1538.

Further research<sup>[4], [5], [6]</sup> has shown that the mechanical properties of cold-formed hollow sections are not reduced by a wide range of welding operations. The grade designations of cold-formed hollow sections based on yield strength are also not affected by hot dip galvanising.

## Reference Standards

"AS 4100" refers to AS 4100–1990: Steel Structures

"AS 1163" refers to AS 1163–1991: Structural Steel Hollow Sections

"AS 1538" refers to AS 1538–1988: Cold-Formed Steel Structures Code

"AS 1554.1" refers to AS 1554.1–1980: Structural Steel Welding

"AS 1170" refers to AS 1170–1989: SAA Loading Code

"AS 1650" refers to AS 1650–1989: Hot-Dipped Galvanized Coatings on Ferrous Articles

## LIMIT STATES DESIGN USING THESE TABLES

**Definition of limit states** - When a structure or part of a structure is rendered unfit for use it reaches a 'limit state'. In this state it ceases to perform the functions or to satisfy the conditions for which it was designed. Relevant limit states for structural steel include strength, serviceability, stability, fatigue, brittle fracture, fire, and earthquake. Only two limit states for structural steel are considered in these tables - strength limit state, and where applicable, serviceability limit state.

*Australian Standard AS 4100–1990: Steel Structures* introduces a limit states approach to structural steel design within Australia. The code follows a semi-probabilistic limit states basis presented in a deterministic format.

Limit states design requires structural members and connections to be proportioned such that the **design capacity** ( $\phi R_u$ ) is less than the **design action effect** ( $S^*$ ) resulting from the **design action** ( $W^*$ ), i.e.

$$S^* \leq \phi R_u$$

**Design action or design load** ( $W^*$ ) is the combination of the nominal actions or loads (e.g. transverse loads on a beam) imposed upon the structure, multiplied by the appropriate load factors as specified in AS 1170. These design actions/loads are identified by a superscript (\*) after the appropriate action/load (e.g.  $W_L^*$  in the tables describes the design transverse load on a beam).

**Design action effects** ( $S^*$ ) are the actions (e.g. design bending moments, shear forces, axial loads) computed from the design actions or design loads using an acceptable method of analysis. These effects are identified by a superscript (\*) after the appropriate action effect (e.g.  $M^*$  describes the design bending moment).

**Design capacity** ( $\phi R_u$ ) is the product of the nominal capacity ( $R_u$ ) and the appropriate capacity factor ( $\phi$ ) found in Table 3.4 of AS 4100.  $R_u$  is determined from Sections 5 to 8, as appropriate, in AS 4100.

For example, consider the strength limit state design of a simply supported beam subject to a total transverse design load ( $W_L^*$ ) distributed uniformly along the beam with full lateral restraint.

The corresponding design action effect ( $S^*$ ) is the design bending moment ( $M^*$ ) which is determined by:

$$M^* = \frac{W_L^* L}{8}$$

where  $L$  = span of the beam

In this case the design capacity ( $\phi R_u$ ) is equal to the design section moment capacity ( $\phi M_s$ ), given by:

$$\phi M_s = \phi f_y Z_e$$

where  $\phi$  = the capacity factor

$f_y$  = yield stress used in design

$Z_e$  = effective section modulus

To satisfy the requirement for limit state design the following relationship must be satisfied,

$$M^* \leq \phi M_s$$

The maximum design bending moment is therefore equal to the design section moment capacity ( $M^* = \phi M_s$ ), and the **maximum design load** is that design load ( $W_L^*$ ) which corresponds to the maximum design bending moment. It should be noted that in this instance the bending capacity of the beam may not be the only criteria in the strength limit state which needs to be considered. (e.g. shear capacity, bearing capacity).

The DCTHS gives values of **design capacity** ( $\phi R_u$ ) and **maximum design load** ( $W^*$ ), where applicable, determined in accordance with AS 4100. When using these tables, the designer must determine the relevant **strength limit state design action** ( $W^*$ ) and/or the corresponding **design action effects** ( $S^*$ ) to ensure that the **strength limit state** requirements of AS 4100 are satisfied. Other limit states (e.g. serviceability, fatigue) must also be considered by the designer. Section 8 of the tables contains design aids for checking the serviceability limit state for some specific beam load and support configurations.

## GENERAL NOTES ON THE TABLES

### Table Contents and Usage

For the commonly available Australian structural steel hollow sections, tables are provided for:

- (i) **section dimensions and section properties**, i.e:
  - Dimensions and Properties (PART 1)
  - Surface Areas (PART 1)
  - Properties for Assessing Section Capacity to AS 4100 (PART 1)
  - Properties for Fire Design (PART 1)
  - Telescoping Sections (PART 1)
- (ii) **design capacity** ( $\phi R_u$ ) for:
  - Section Capacities (PART 3)
  - Members Subject to Bending (PART 4)
  - Members Subject to Axial Compression (PART 5)
  - Members Subject to Axial Tension (PART 6)
- (iii) **elastic buckling load** ( $N_{om}$ ) (PART 7)
- (iv) **maximum design load** ( $W^*$ ) for:
  - Strength Limit State ( $W_L^*$ ) for Beams (PART 8)
  - Serviceability Limit State ( $W_s^*$ ) for Beams (PART 8)
    - (simply supported, continuous, fixed end, and cantilever beams)

Acceptable methods of analysis for determining the **design action effects** are described in Section 4 of AS 4100 and PART 2 of this publication. Information relevant to such methods of analysis is presented briefly in PART 7 of this publication.

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## Range of Sections

The tables cover the full range of hollow sections, manufactured in accordance with AS 1163, which are generally available and commonly used in Australia. The following grades of structural steel hollow sections are detailed in AS 1163.

Australian Standard	Steel Grade	Yield Stress $f_y$ MPa	Tensile Strength $f_u$ MPa
AS 1163	C250 and C250L0	250	320
	C350 and C350L0	350	430
	C450 and C450L0	450	500

The grade designation (e.g. C250) is based on the nominal minimum yield strength of the steel (in MPa). The prefix 'C' is used before the value of the nominal yield strength of the steel to indicate that the section is cold-formed. AS 1163 only considers cold-formed structural steel hollow sections. The suffix 'L0' denotes impact properties at 0° C as specified in AS 1163.

In these notes and in the tables:

- Grade "C250" refers to **both** C250 and C250L0
- Grade "C350" refers to **both** C350 and C350L0
- Grade "C450" refers to **both** C450 and C450L0

## Specifications

Hollow sections produced in Australia are manufactured by cold-forming and high frequency Electric Resistance Welding (ERW). The ERW process allows cold-formed hollow sections to be welded at ambient temperatures without subsequent stress relieving.

The structural hollow sections produced in Australia are generally manufactured and tested to meet the requirements of one or more of the following specifications:

- AS 1163-1991: Structural Steel Hollow Sections
- AS 1450-1983: Steel Tubes for Mechanical Purposes
- AS 1074-1989: Steel Tubes and Tubulars for Ordinary Service
- API5L-1990: Steel Linepipe

However the Tables only apply to those hollow sections manufactured in accordance with AS 1163.

## Availability

The sections listed in the tables are normally readily available from steel distributors ex-stock in standard lengths. However, the availability should be checked for the larger sizes, for large tonnages of individual sections, or for non-standard lengths.

The standard lengths given in the table below are produced by Palmer Tube Mills (Aust.) Pty Ltd and Tubemakers of Australia Limited, Structural Products Division, although each manufacturer may not produce all of the sizes and stock lengths shown. Sections may be ordered in other lengths ex mill rollings subject to the manufacturers' length limitations and minimum order requirements.

Section Type	Sizes	Standard Lengths (m)
Circular Hollow Sections Grade C250	13.5 O.D. 17.2 O.D. to 165.1 O.D. 508.0 O.D. to 610.0 O.D.	5.8 6.5 6.0, 9.0, and 12.0
Circular Hollow Sections Grade C350	21.3 O.D. to 88.9 O.D. 114.3 O.D. to 457.0 O.D.	6.5 6.5 and 12.0
Rectangular Hollow Sections Grade C350	50 x 20 to 150 x 50 150 x 100 to 250 x 150	8.0 8.0 and 12.0
Rectangular Hollow Sections Grade C450	50 x 20 to 125 x 75 x 2.8 125 x 75 x 3.3 and 125 x 75 x 3.8	8.0 8.0 and 12.0
Square Hollow Sections Grade C350	13 x 13 to 25 x 25 30 x 30 to 89 x 89 100 x 100 to 250 x 250	6.5 8.0 8.0 and 12.0
Square Hollow Sections Grade C450	25 x 25 30 x 30 to 100 x 100	6.5 8.0

Hollow sections in grades C250L0, C350L0, and C450L0 have equivalent or superior performance to grades C250, C350 and C450 respectively. The "L0" grades are the most readily available grades from the manufacturers.

Hollow sections are readily available in black, painted, and galvanized finishes. The galvanized finishes available are:

- Continuously hot-dipped galvanized.  
(min. 300 g/m<sup>2</sup> on each surface to the requirements of AS 1650)
- Z275 for hollow sections fabricated from continuously hot-dipped galvanized steel sheet.  
(275 g/m<sup>2</sup> total on both surfaces to AS 1650)
- Continuously hot-dipped galvanized – external surface only.  
(min. 180 g/m<sup>2</sup> to AS 1650)
- Continuously hot-dipped galvanized – external surface only.  
(min. 100 g/m<sup>2</sup>)

## Welding

The following consumables are recommended by AS 1554.1–1980 when welding hollow sections in steel grade C250 and C350. AS 1554.1 is currently being revised to include C450 grade steel.

		C250	C350	C450*
Manual metal-arc	(MMAW)	E41XX E48XX	E41XX E48XX	E48XX
Gas metal-arc (known as MIG)	(GMAW)	W50X	W50X	W50X
Submerged arc	(SAW)	W40X W50X	W40X W50X	—
Flux-cored arc	(FCAW)	W40X W50X	W40X W50X	—

\* Note: C450 grade steel has not yet been included in AS 1554.1–1980, but tests<sup>[5], [6]</sup> have shown that E48XX and W50X consumables are satisfactory for providing the following requirements so that strength and ductility are met:

- 1) The yield and tensile strengths of the welding consumable are greater than the nominal values of the C450 steel
- 2) Welding consumables are grade 2 (or higher) depositing welds giving Charpy notch impact toughness of 47 J at 0° C.

Check with local suppliers for consumables satisfying these requirements.

The above recommendations apply to predominantly statically loaded structures but also in broad principle to dynamically loaded structures subject to moderate cyclic loads, as specified in AS 4100: Section 11 – Fatigue.

## Properties of Steel

The properties of steel adopted in the Tables are shown below.

Property	Symbol	Value
Elastic Modulus	$E$	$200 \times 10^3$ MPa
Shear Modulus	$G$	$80 \times 10^3$ MPa
Density	$\rho$	7850 kg/m <sup>3</sup>
Poisson's Ratio	$\nu$	0.25
Coefficient of Thermal Expansion	$\alpha_T$	$11.7 \times 10^{-6}$ per °C

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- [1] Hasan, S.W., Hancock, G.J., "Plastic Bending Tests of Cold-Formed Rectangular Hollow Sections", *Steel Construction*, AISC, Vol. 23, No. 4, 1989.
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- [3] Zhao, X.L., Hancock, G.J., "Tests to Determine Plate Slenderness Limits for Cold-Formed Rectangular Hollow Sections of Grade C450", *Steel Construction*, AISC, Vol. 25, No. 4, 1990.
- [4] HERA, "Investigation of the Brittle Fracture Resistance of Cold-Formed Rectangular Hollow Section. (Part 2)", *HERA Report R4-39*, Auckland Industrial Development Division Department of Scientific and Industrial Research, 1987.
- [5] Corderoy, D.J.H., Landy, R.J., "Weldability Trials with Grade C450 RHS Structural Tubing", *Second National Structural Engineering Conference*, The Institution of Engineers Australia, 1990.
- [6] Corderoy, D.J.H., Dempsey, R.I., Landy, R.J., "Welding Aspects of Grade C450L0 Structural Hollow Sections", *Welding Towards 2000 Conference Proceedings*, Welding Technology Institute of Australia, 1991.

## CONVERSION TO SAFE WORKING LOADS

The design capacities given in these tables are limit states design capacities calculated in accordance with AS 4100, and must be equal to or greater than the design action effect (e.g. bending moment, shear force, axial force) resulting from the design loads. These *design loads* are not working loads, but are obtained by factoring the nominal (working) loads applied to the structure in accordance with the loading code AS 1170.

For working stress design, working load capacities (safe working loads) based on a load factor of 1.5 as used for the live load component of dead and live load in AS 1170.1, may be obtained using:

$$\text{safe working load} = \frac{\text{maximum design load from tables}}{1.5}$$

or       $\text{safe working load} = \frac{\text{design capacity from tables}}{1.5}$

## LIST OF PRINCIPAL SYMBOLS USED IN THE TABLES

$A_e$	effective area of a cross-section
$A_g$	gross area of a cross-section
$A_n$	net area of a cross-section
$b$	width of a section
$b_e$	effective width of a plate element
$b_b, b_{bf},$	
$b_{bw}, b_s$	bearing widths defined in Section 4.3.2
$C$	torsional section modulus
$c_m$	factor for unequal moments
$d$	depth of a section
$d_e$	effective outside diameter of a circular hollow section
$d_o$	outside diameter of a circular hollow section
$d_w$	depth of web
$d_l$	clear depth between flanges
$E$	Young's modulus of elasticity
$f_u$	tensile strength used in design
$f_y$	yield stress used in design
$f_{va}^*$	average design shear stress in the web
$f_{vm}^*$	maximum design shear stress in the web
$G$	shear modulus of elasticity; or nominal dead load
$I$	second moment of area of a cross-section
$I_w$	warping section constant
$I_x$	$I$ about the cross-sectional major principal x-axis
$I_y$	$I$ about the cross-sectional minor principal y-axis
$J$	torsional section constant
$k_e$	member effective length factor
$k_f$	form factor for members subject to axial compression
$k_h$	load height effective length factor
$k_r$	effective length factor for restraint against lateral rotation
$k_{sm}$	exposed surface area to mass ratio
$k_t$	correction factor for distribution of forces in a tension member; or twist restraint effective length factor
$L$	span or member length; or segment or sub-segment length
$L_e$	effective length of a compression member; or effective length of a laterally unsupported flexural member
$M_b$	nominal member moment capacity
$M_{bx}$	$M_b$ about major principal x-axis
$M_{ix}$	nominal in-plane member moment capacity about major principal x-axis
$M_{iy}$	nominal in-plane member moment capacity about minor principal y-axis
$M_o$	reference elastic buckling moment for a member subject to bending
$M_{oa}$	amended elastic buckling moment for a member subject to bending
$M_{ox}$	nominal out-of-plane member moment capacity about major principal x-axis

$M_{rx}$	$M_s$ about major principal x-axis reduced by axial force
$M_{ry}$	$M_s$ about minor principal y-axis reduced by axial force
$M_s$	nominal section moment capacity
$M_{sx}$	$M_s$ about major principal x-axis
$M_{sy}$	$M_s$ about minor principal y-axis
$M_z$	nominal torsional moment section capacity
$M^*$	design bending moment
$M_m^*$	maximum calculated design bending moment along the length of a member or in a segment
$M_x^*$	design bending moment about major principal x-axis
$M_y^*$	design bending moment about minor principal y-axis
$M_z^*$	design torsional moment
$N_c$	nominal member capacity in compression
$N_{cx}$	$N_c$ for member buckling about major principal x-axis
$N_{cy}$	$N_c$ for member buckling about minor principal y-axis
$N_{om}$	elastic flexural buckling load of a member
$N_{omb}$	$N_{om}$ for a braced member
$N_{omx}$	$\frac{\pi^2 EI_x}{(k_e L)^2} N_{om}$ about major principal x-axis
$N_{omy}$	$\frac{\pi^2 EI_x}{(k_e L)^2} N_{om}$ about minor principal y-axis
$N_s$	nominal section capacity of a concentrically loaded compression member
$N_t$	nominal section capacity in tension
$N^*$	design axial force, tensile or compressive
$P$	applied load
$R_b$	nominal bearing capacity of a web
$R_{bb}$	nominal bearing buckling capacity
$R_{by}$	nominal bearing yield capacity
$R_u$	nominal capacity
$r$	radius of gyration
$r_{ext}$	external corner radius
$r_x$	radius of gyration about major principal x-axis
$r_y$	radius of gyration about minor principal y-axis
$R^*$	design bearing force
$S$	plastic section modulus
$S_x$	$S$ about major principal x-axis
$S_y$	$S$ about minor principal y-axis
$S^*$	design action effect
$t$	thickness of a section
$t_f$	thickness of a flange
$t_w$	thickness of a web
$V_u$	nominal shear capacity of a web with a uniform shear stress distribution

$V_v$	nominal shear capacity of a web
$V_{vx}$	$V_v$ of a member in the major principal x-axis direction
$V_{vy}$	$V_v$ of a member in the minor principal y-axis direction
$V^*$	design shear force
$W$	applied load
$W^*$	design action
$W_t^*$	strength limit state maximum design load
$W_s^*$	serviceability limit state maximum design load
$Z$	elastic section modulus
$Z_e$	effective section modulus
$Z_{ex}$	$Z_e$ for bending about major principal x-axis
$Z_{ey}$	$Z_e$ for bending about minor principal y-axis
$Z_n$	$Z$ for bending about n-axis
$Z_x$	$Z$ for bending about major principal x-axis
$Z_y$	$Z$ for bending about minor principal y-axis
$\alpha_a$	compression member factor (as defined in Clause 6.3.3 of AS 4100)
$\alpha_b$	compression member section constant (as defined in Clause 6.3.3 of AS 4100)
$\alpha_c$	compression member slenderness reduction factor
$\alpha_m$	moment modification factor for bending
$\alpha_s$	slenderness reduction factor
$\alpha_T$	coefficient of thermal expansion for steel
$\beta_m$	ratio of smaller to larger bending moments at the ends of a member
$\Delta$	deflection of a member
$\delta_b$	moment amplification factor for a braced member
$\delta_m$	moment amplification factor, taken as the greater of $\delta_b$ and $\delta_s$
$\delta_s$	moment amplification factor for a sway member
$\eta$	compression member imperfection factor (as defined in Clause 6.3.3 of AS 4100)
$\theta$	angle of twist per unit length slenderness ratio
$\lambda_c$	elastic buckling load factor
$\lambda_e$	plate element slenderness
$\lambda_{ed}$	plate element deformation slenderness limit
$\lambda_{ep}$	plate element plasticity slenderness limit
$\lambda_{ey}$	plate element yield slenderness limit
$\lambda_n$	modified compression member slenderness
$\nu$	Poisson's ratio
$\xi$	compression member factor (as defined in Clause 6.3.3 of AS 4100)
$\pi$	pi ( $\approx 3.14159$ )
$\rho$	density of a material
$\phi$	capacity factor

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NOTE: SEE PAGE (IX) FOR THE SPECIFIC MATERIAL  
 STANDARD REFERRED TO BY THE SECTION TYPE  
 AND STEEL GRADE IN THESE TABLES

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**PART 1 SECTION PROPERTIES****1.1 Introduction**

The section property tables include all relevant section dimensions and properties necessary for assessing steel structures in accordance with AS 4100. The structural hollow sections included in these tables are:

- Circular Hollow Sections                          Grade C250 and C350
- Rectangular Hollow Sections                          Grade C350 and C450
- Square Hollow Sections                                  Grade C350 and C450

*The size ranges considered for the above hollow sections are based on the normative ranges listed in AS 1163. For an explanation of this standard in terms of these Tables reference must be made to "General Notes on the Tables" in particular to "Range of Sections" [see page (ix)]. The grades include the L0 grades as described in "Range of Sections".*

**1.2 Section Property Tables**

For each group of structural hollow sections the tables include:

- Dimensions and Properties
- Properties for Assessing Section Capacity to AS 4100

**1.2.1 Dimensions and Properties**

The tables give standard dimensions and properties for the structural steel hollow sections listed in Section 1.1. The second moments of area are required for serviceability calculations and the radii of gyration are required for assessing member stability. The elastic and plastic section moduli for bending about the various axes are also tabulated. These are utilised in an intermediate step to determine the effective section modulus for flexural design to AS 4100. The elastic section moduli are also used in the determination of elastic stresses where design for fatigue must be considered, or where the stress state at serviceability loads may need to be checked. The torsion constants are used in determining the torsional moment and angle of twist per unit length.

The general format of these tables follows that of the equivalent tables contained in the current "Design Capacity Tables for Structural Steel"-1st Edition.

### 1.2.1.1 Torsion Constants

The torsional inertia constant ( $J$ ) and the torsional modulus constant ( $C$ ) for square and rectangular hollow sections are defined as follows:

$$J = \left( t^3 \frac{h}{3} + 2kA_h \right)$$

$$C = \left( \frac{t^3 \frac{h}{3} + 2kA_h}{t + \frac{k}{t}} \right)$$

where  $R_c = \frac{R_o + R_i}{2}$

$$h = 2[(b - t) + (d - t)] - 2R_c(4 - \pi)$$

$$A_h = (b - t)(d - t) - R_c^2(4 - \pi)$$

$$k = \frac{2A_ht}{h}$$

- and
- $t$  = specified thickness of section
  - $b$  = width of section
  - $d$  = depth of section
  - $R_o$  = outer corner radius
  - $R_i$  = inner corner radius
  - $R_c$  = mean corner radius
  - $h$  = length of the mid-contour
  - $A_h$  = area enclosed by  $h$
  - $k$  = integration constant

as shown in Figure 1.2.1.1

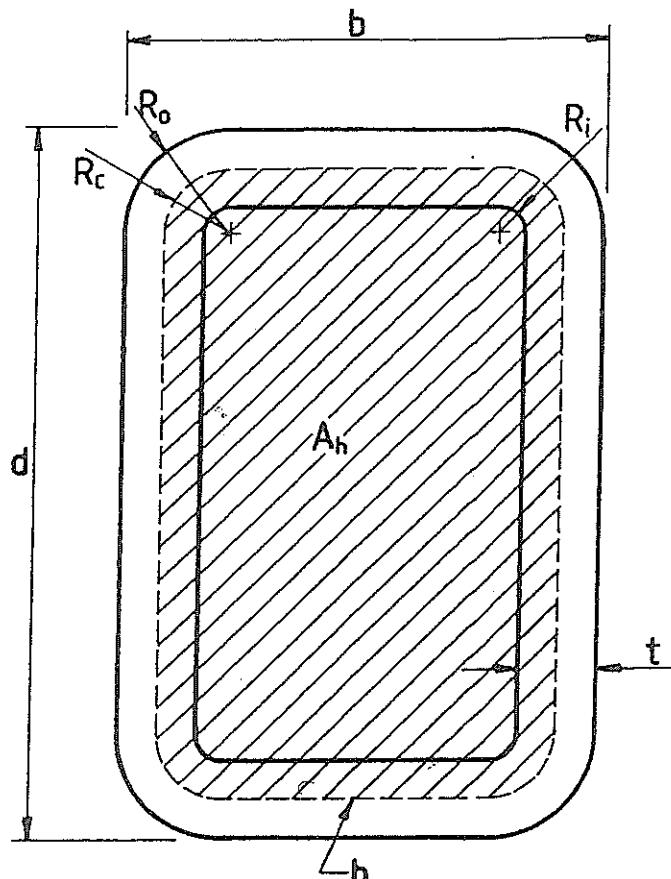


Figure 1.2.1.1: Parameters for Calculation of Torsion Constants

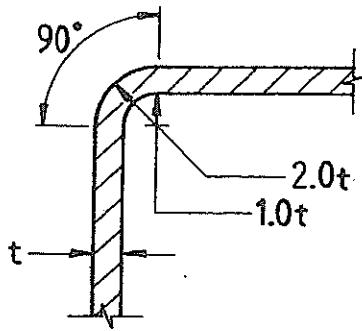
The information contained in Section 1.2.1.1 was extracted from:

- International Standard ISO 657/XIV, "Hot-rolled steel sections – Part XIV : Hot-finished structural hollow sections – Dimensions and sectional properties", 1977.

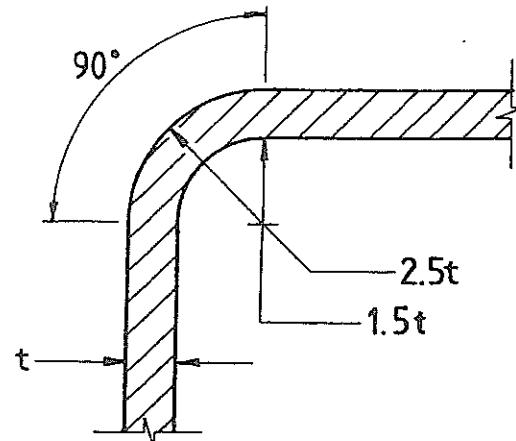
## 1.2.1.2 Corner Radii

The section properties presented in this publication are calculated in accordance with AS 1163.

Figure 1.2.1.2 shows the corner radii detail used in determining section properties. However it should be noted that the actual corner geometry may vary from that shown.



a) thickness less than 3.0 mm



b) thickness 3.0 mm and greater

Figure 1.2.1.2: Corner Geometry for Determining Section Properties

## 1.2.2 Surface Areas

Surface area data may be used in estimating quantities of protective coatings. Tables 1.2–1 to 1.2–6 include values of external surface area per metre and external surface area per tonne.

## 1.2.3 Properties for Assessing Section Capacities

These properties are necessary for calculating the section capacities of the structural hollow sections in accordance with AS 4100. The effective section moduli, "compactness" of section, and the form factor are tabulated according to steel grade.

### 1.2.3.1 Compactness

In Clauses 5.2.3, 5.2.4, and 5.2.5 of AS 4100, sections are described as **compact**, **non-compact** or **slender**. This type of categorisation provides a measure of the relative importance of yielding and local buckling on the effective section modulus.

The tables include a column headed "compactness" where the compactness or otherwise of the sections is indicated for a given axis of bending as follows:

- C compact
- N non-compact
- S slender

These terms are important with respect to selecting the methods of analysis that may be used to determine the design action effects (see Clause 4.5 of AS 4100) or in using the provisions of Section 8 of AS 4100 for designing members subject to combined actions. Clause 4.5 of AS 4100 does not currently permit plastic analysis when designing with structural hollow sections and further research is required to determine the suitability of plastic analysis in the design of hollow sections.

### 1.2.3.2 Effective Section Modulus

Subsequent to the evaluation of "compactness" the effective section modulus ( $Z_e$ ) is also tabulated.  $Z_e$  is determined by the requirements of Clauses 5.2.2 to 5.2.5 inclusive, of AS 4100 and is used in the calculation of the nominal section moment capacity ( $M_s$ ) as defined in Clause 5.2.1 of AS 4100.

Table 1.2.3.2 gives values of plate element slenderness limits for structural hollow sections used in the determination of  $Z_e$  in Tables 1.2–1 to 1.2–6. It should be noted that the deformation limit ( $\lambda_{ed}$ ) is never exceeded for hollow sections manufactured in accordance with AS 1163 and therefore noticeable deformations will not occur under service loadings.

Section	Element	Residual Stresses	Plasticity Limit $\lambda_{ep}$	Yield Limit $\lambda_{ey}$	Deformation Limit $\lambda_{ed}$
RHS, SHS	Compression Flange	CF	30	40	90
	Web	CF	82	115	—
CHS	—	CF	50	120	—

Table 1.2.3.2: Plate Element Slenderness Limits for Members Subject to Bending

The CF residual stress classification is used for all sections as they are manufactured in Australia by the cold forming process.

### 1.2.3.3 Form Factor

The form factor ( $k_f$ ) determined in accordance with Clause 6.2.2 of AS 4100 is given by:

$$k_f = \frac{A_e}{A_g}$$

where  $A_g$  = gross cross-sectional area  
 $A_e$  = effective area

$A_e$  was calculated by summing the effective areas of the individual elements whose effective widths are specified:

for RHS and SHS by

$$b_e = b \left( \frac{\lambda_{ey}}{\lambda_e} \right) \leq b$$

and for CHS as the lesser of

$$d_e = d_o \sqrt{\left( \frac{\lambda_{ey}}{\lambda_e} \right)} \leq d_o$$

and

$$d_e = d_o \left( \frac{3\lambda_{ey}}{\lambda_e} \right)^2$$

where  $d_e$  = effective outside diameter of CHS (Clause 6.2.4 of AS 4100)

$b_e$  = effective width of section (Clause 6.2.4 of AS 4100)

$b$  = full width of section

$d_o$  = outside diameter of CHS

$\lambda_{ey}$  = yield slenderness limit (see Table 1.2.3.3)

$\lambda_e$  = plate element slenderness (see Table 1.2.3.3)

Table 1.2.3.3: Plate Element Slenderness Limits for Members Subject to Axial Compression

Section	Residual Stresses	Yield Slenderness Limit $\lambda_{ey}$	Plate Element Slenderness $\lambda_e$
RHS, SHS	CF	40	$\frac{b}{t} \sqrt{\left( \frac{f_y}{250} \right)}$
CHS	CF	82	$\frac{d_o}{t} \left( \frac{f_y}{250} \right)$

$k_f$  must be known in order to determine the nominal section capacity of a concentrically loaded compression member ( $N_s$ ) as defined in Clause 6.2.1 of AS 4100. The calculation of  $k_f$  indicates the degree to which the column section will buckle locally before squashing (i.e.  $k_f = 1.0$  signifies a column section which will yield rather than buckle locally in a short or stub column test). A knowledge of  $k_f$  is also important when using the provisions of Section 8 of AS 4100 for designing members subject to combined actions.

## 1.3 Properties for Fire Design

To assist in the design of the structural steel hollow sections for fire resistance in accordance with Section 12 of AS 4100, values of the exposed surface area to mass ratio ( $k_{sm}$ ) are tabulated for the various cases shown in Figure 1.3.

For **unprotected structural hollow sections** the values of  $k_{sm}$  corresponding to four- and three-sided exposure should be taken as those corresponding to Cases 1 and 4 respectively. In these instances fire protection is necessary where a fire rating is required.

For members requiring the addition of fire protection materials, the "Handbook of Fire Protection Materials for Structural Steel" published by AISC<sup>[1]</sup> may be consulted to determine the thickness of proprietary materials required for a given value of  $k_{sm}$  and Fire-Resistance Level. In the AISC Handbook, the exposed surface area to mass ratio ( $E$ ) may be taken as equivalent to  $k_{sm}$ .

					
Case 1	Case 2	Case 3	Case 4	Case 5	Case 6
4-SIDED EXPOSURE TO FIRE			3-SIDED EXPOSURE TO FIRE		

Cases of fire exposure considered:

- 1 = Profile-protected
- 2 = Total Perimeter, Box-protected, No Gap
- 3 = Total Perimeter, Box-protected, 25 mm Gap
- 4 = Top Flange Excluded, Profile-protected
- 5 = Top Flange Excluded, Box-protected, No Gap
- 6 = Top Flange Excluded, Box-protected, 25 mm Gap

Figure 1.3: Cases for Calculation of Exposed Surface Area to Mass Ratio

Suggested references for Fire Design:

- [1] Proe, D.J., Bennetts, I.D., Thomas, I.R., Szeto, W.T., "Handbook of Fire Protection Materials for Structural Steel", *Australian Institute of Steel Construction*, 1990.
- [2] Bennetts, I.D., Proe, D.J., Thomas, I.R., "Guidelines for Assessment of Fire Resistance of Structural Steel Members", *Australian Institute of Steel Construction*, 1987.

## 1.4 Telescoping Sections

### 1.4.1 Scope

The tables of telescoping sections provided can be used to determine hollow sections which are suitable for telescoping.

### 1.4.2 Method

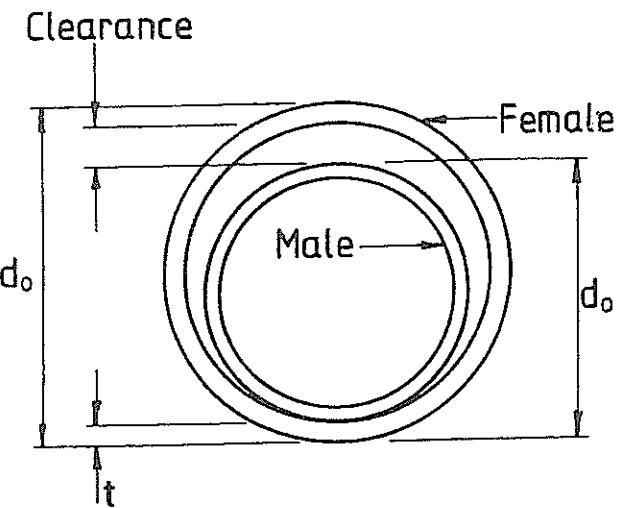
Total available clearance is tabulated to allow designers to select sections with suitable clearance for the type of fit required. Sections with clearances less than 2.0 mm are shown **bold** in the tables. Figure 1.4.2 shows typical telescoping data required to select appropriate sections.

All calculations used in preparation of the tables are based on the nominal dimensions of hollow sections and manufacturing tolerances specified in AS 1163. Owing to dimensional tolerances permitted within that standard actual clearances of sections manufactured to this specification will vary marginally from the values tabulated.

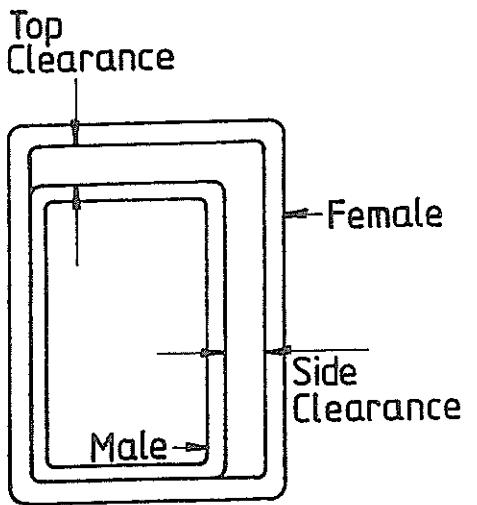
For tight fits, varying corner radii and internal weld heights can affect telescoping of sections and it is recommended that some form of testing is carried out prior to committing material. Where telescoping over some length is required, additional clearance may be needed to allow for straightness of the section.

Telescoping of SHS and RHS where the female (outer) has a larger wall thickness requires careful consideration of corner clearances due to the larger corner radii of the thicker section.

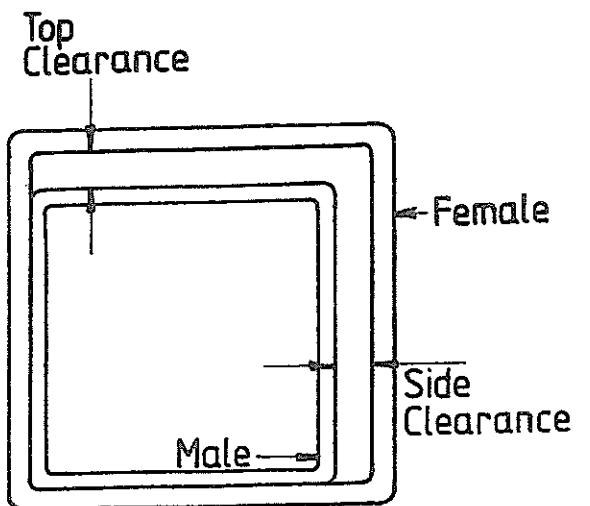
Typical corner geometry may differ from that used for calculation of section properties and reference should be made to individual manufacturers.



a) CHS



b) RHS



c) SHS

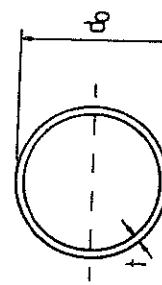


TABLE 1.2 – 1(1)

## DIMENSIONS AND PROPERTIES

## CIRCULAR HOLLOW SECTIONS: GRADE C250

DIMENSIONS AND RATIOS				PROPERTIES								Properties for Design to AS 4100	
Designation	Mass per m	External Surface Area per m	$\frac{d_o}{t}$	Gross Section Area				About any axis				Torsion Constant	Torsion Modulus
				$A_g$	$10^3 \text{ mm}^2$	$10^3 \text{ mm}^4$	$10^3 \text{ mm}^3$	$I$	$Z$	$S$	$r$		
610.0 x 12.7 CHS	187	1.92	10.2	48.0	23800	1060	3490	4520	211	2130	10 <sup>6</sup> mm <sup>4</sup>	$10^3 \text{ mm}^3$	(C,N,S) $10^3 \text{ mm}^3$
9.5 CHS	141	1.92	13.6	64.2	17900	809	2650	3450	212	1620	6970	1.00	C 4530
6.4 CHS	95.3	1.92	20.1	95.3	12100	553	1810	2330	213	1110	5300	1.00	N 3270
508.0 x 12.7 CHS	155	1.60	10.3	40.0	19800	606	2380	3120	175	1210	4770	0.927	N 2000
9.5 CHS	117	1.60	13.7	53.5	14900	462	1820	2380	176	925	3640	1.00	C 3120
6.4 CHS	79.2	1.60	20.2	79.4	10100	317	1250	1610	177	634	2500	1.00	N 2330
165.1 x 5.4 CHS	21.3	0.519	24.4	30.6	2710	8.65	105	138	56.5	17.3	209	1.00	C 1460
5.0 CHS	19.7	0.519	26.3	33.0	2510	8.07	97.7	128	56.6	16.1	195	1.00	N 128
139.7 x 5.4 CHS	17.9	0.439	24.5	25.9	2280	5.14	73.7	97.4	47.5	10.3	147	1.00	C 138
5.0 CHS	16.6	0.439	26.4	27.9	2120	4.81	68.8	90.8	47.7	9.61	138	1.00	C 97.4
114.3 x 5.4 CHS	14.5	0.359	24.8	21.2	1850	2.75	48.0	64.1	38.5	5.49	96.1	1.00	C 90.8
4.5 CHS	12.2	0.359	29.5	25.4	1550	2.34	41.0	54.3	36.9	4.69	82.0	1.00	C 64.1
101.6 x 5.0 CHS	11.9	0.319	26.8	20.3	1520-	1.77	34.9	46.7	34.2	3.55	69.9	1.00	C 54.3
4.0 CHS	9.63	0.319	33.2	25.4	1230	1.46	28.8	38.1	34.5	2.93	57.6	1.00	C 46.7
8819 x 5.9 CHS	12.1	0.279	23.1	15.1	1540	1.33	30.0	40.7	28.4	2.65	59.9	1.00	C 38.1
5.0 CHS	10.3	0.279	27.0	17.8	1320	1.16	26.2	35.2	29.7	2.33	52.4	1.00	C 40.7
4.0 CHS	8.38	0.279	33.3	22.2	1070	0.953	21.7	28.9	30.0	1.93	43.3	1.00	C 35.2
76.1 x 5.9 CHS	10.2	0.239	23.4	12.9	1300	0.807	21.2	29.1	24.9	1.61	42.4	1.00	C 28.9
4.5 CHS	7.95	0.239	30.1	16.9	1010	0.651	17.1	23.1	25.4	1.30	34.2	1.00	C 29.1
3.6 CHS	6.44	0.239	37.1	21.1	820	0.540	14.2	18.9	25.7	1.08	28.4	1.00	C 23.1
60.3 x 5.4 CHS	7.31	0.189	25.9	11.2	931	0.354	11.8	16.3	19.5	0.709	23.5	1.00	C 18.9
4.5 CHS	6.19	0.189	30.6	13.4	789	0.309	10.2	14.0	19.8	0.618	20.5	1.00	C 16.3
3.6 CHS	5.03	0.189	37.6	16.8	641	0.259	8.58	11.6	20.1	0.517	17.2	1.00	C 14.0
												11.6	

Notes: 1. For Grade C250  $f_y = 250 \text{ MPa}$  and  $\epsilon_u = 220 \text{ MPa}$ .

2. Grade C250 to AS 1163 is cold formed and therefore is allocated the CF residual stresses classification in AS 4100.

3. C = Compact Section; N = Non-compact Section; S = Slender Section, as defined in AS 4100.

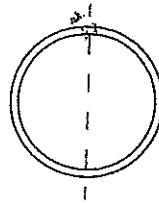




TABLE 1.2 – 1(2)  
DIMENSIONS AND PROPERTIES  
CIRCULAR HOLLOW SECTIONS: GRADE C250

Designation	DIMENSIONS AND RATIOS				PROPERTIES						
	Mass per m	External Surface Area per m	$\frac{d_o}{t}$	Gross Section Area $A_g$	About any axis			Torsion Constant $J$	Torsion Modulus $C$	Form Factor $k_f$	
					1	2	S				
mm	kg/m	$m^2/m$		$mm^3$	$10^3 mm^4$	$10^3 mm^3$	$mm$	$10^4 mm^4$	$10^3 mm^3$		
48.3 x 5.4 CHS	5.71	0.152	26.6	8.9	728	0.170	7.04	9.99	15.3	0.340	14.1
4.0 CHS	4.37	0.152	34.7	12.1	557	0.138	5.70	7.87	15.7	0.275	11.4
3.2 CHS	3.56	0.152	42.6	15.1	453	0.116	4.80	6.52	16.0	0.232	9.59
42.4 x 4.9 CHS	4.53	0.133	29.4	8.7	577	0.103	4.87	6.93	13.4	0.206	9.74
4.0 CHS	3.79	0.133	35.2	10.6	483	0.099	4.24	5.92	13.6	0.180	8.48
3.2 CHS	3.09	0.133	43.1	13.3	394	0.0762	3.59	4.93	13.9	0.152	7.19
33.7 x 4.5 CHS	3.24	0.106	32.7	7.5	413	0.0450	2.67	3.87	10.4	0.0901	5.35
4.0 CHS	2.93	0.106	36.1	8.4	373	0.0419	2.49	3.65	10.6	0.0838	4.97
3.2 CHS	2.41	0.106	44.0	10.5	307	0.0360	2.14	2.99	10.8	0.0721	4.28
26.9 x 4.0 CHS	2.26	0.0845	37.4	6.7	288	0.0194	1.45	2.12	8.22	0.0389	2.89
3.2 CHS	1.87	0.0845	45.2	8.4	238	0.0170	1.27	1.81	8.46	0.0341	2.53
2.6 CHS	1.56	0.0845	54.2	10.3	198	0.0148	1.10	1.54	8.64	0.0296	2.20
21.3 x 3.6 CHS	1.57	0.0669	42.6	5.9	200	0.0016	0.767	1.14	6.39	0.0163	1.53
3.2 CHS	1.43	0.0669	46.8	6.7	182	0.00768	0.722	1.06	6.50	0.0154	1.44
2.6 CHS	1.20	0.0669	55.8	8.2	153	0.00681	0.639	0.915	6.68	0.0136	1.28
17.2 x 2.9 CHS	1.02	0.0540	52.8	5.9	130	0.00347	0.403	0.601	5.16	0.00693	0.806
2.3 CHS	0.845	0.0540	63.9	7.5	108	0.00306	0.356	0.515	5.33	0.00612	0.711
13.5 x 2.9 CHS	0.758	0.0424	55.9	4.7	98.6	0.00146	0.216	0.334	3.89	0.00292	0.432
2.3 CHS	0.635	0.0424	66.8	5.9	80.9	0.00132	0.196	0.293	4.04	0.00264	0.392

Notes:

- For Grade C250  $f_y = 250$  MPa and  $f_u = 320$  MPa.
- $f_y$  = yield stress used in design;  $f_u$  = tensile strength used in design as defined in AS 4100.
- Grade C250 to AS 183 is cold formed and therefore is allocated the CF residual stresses classification in AS 4100.
- $N$  = Compact Section;  $N$  = Non-compact Section;  $S$  = Slender Section; as defined in AS 4100.

TABLE 1.2 - 2(1)

## DIMENSIONS AND PROPERTIES

## CIRCULAR HOLLOW SECTIONS: GRADE C350

DIMENSIONS AND RATIOS				PROPERTIES						PROPERTIES FOR DESIGN TO AS 4100				
Designation	Mass per m	External Surface Area per t	$\frac{d_o}{t}$	Gross Section Area			About any axis			Torsion Constant	Torsion Modulus	Form Factor	About any axis	
				$A_g$	$10^6 \text{ mm}^4$	$10^3 \text{ mm}^3$	$I$	$Z$	$S$					
mm	kg/m	$\text{m}^2/\text{m}$		mm <sup>2</sup>			mm <sup>4</sup>		mm <sup>3</sup>	$10^6 \text{ mm}^4$				
457.0 x 12.7 CHS	139	1.44	10.3	36.0	17700	438	1920	2510	157	876	3830	1.00	N	
9.5 CHS	105	1.44	13.7	48.1	13400	334	1460	1900	158	669	2930	1.00	2500	
6.4 CHS	71.1	1.44	20.2	71.4	9060	230	1010	1300	159	460	2010	0.904	1790	
406.4 x 12.7 CHS	123	1.28	10.4	32.0	15700	305	1500	1970	139	609	3000	1.00	N	
9.5 CHS	93.0	1.28	13.7	42.8	11800	233	150	1500	140	467	2300	1.00	1090	
6.4 CHS	63.1	1.28	20.2	63.5	8040	161	792	1020	141	322	1580	0.980	N	
365.6 x 12.7 CHS	107	1.12	10.4	28.0	13700	201	10300	165	871	1140	122	403	1.00	C
9.5 CHS	81.1	1.12	13.8	37.4	10300	165	871	1140	121	310	1740	1.00	1450	
6.4 CHS	55.1	1.12	20.3	55.6	7020	107	602	781	123	214	1200	1.00	895	
323.9 x 12.7 CHS	97.5	1.02	10.4	25.5	12400	151	930	1230	110	301	1860	1.00	C	
9.5 CHS	73.7	1.02	13.8	34.1	9380	116	717	939	111	232	1480	1.00	1490	
6.4 CHS	50.1	1.02	20.3	50.6	6380	80.5	497	645	112	161	984	1.00	N	
273.1 x 9.3 CHS	60.5	0.858	14.2	29.4	7710	67.1	492	647	93.3	134	983	1.00	N	
6.4 CHS	42.1	0.858	20.4	42.7	5360	47.7	349	455	94.3	95.4	699	1.00	710	
4.8 CHS	31.8	0.858	27.0	56.9	4050	36.4	267	346	94.9	72.8	533	1.00	1230	
219.1 x 8.2 CHS	42.6	0.688	16.1	26.7	5430	30.3	276	365	75.2	60.5	552	1.00	C	
6.4 CHS	33.6	0.688	20.5	34.2	4280	24.2	221	290	44.4	48.4	442	1.00	939	
.4.8 CHS	25.4	0.688	27.1	45.6	3230	18.6	169	220	75.8	37.1	339	1.00	601	
166.3 x 7.1 CHS	28.2	0.529	18.7	23.7	3600	11.7	139	185	57.0	23.4	278	1.00	C	
6.4 CHS	25.6	0.529	20.7	26.3	3260	10.7	127	168	57.3	21.4	254	1.00	441	
4.8 CHS	19.4	0.529	27.3	35.1	2470	8.25	98.0	128	57.8	16.5	196	1.00	312	
165.1 x 3.5 CHS <sup>1</sup>	13.9	0.519	37.2	47.2	1780	5.80	70.3	91.4	57.1	11.6	141	1.00	C	
3.0 CHS	12.0	0.519	43.2	55.0	1550	5.02	60.8	78.8	57.3	10.0	122	1.00	365	
139.7 x 3.5 CHS	11.8	0.499	37.3	39.9	1500	3.47	49.7	64.9	48.2	6.95	99.5	1.00	N	
3.0 CHS	10.1	0.499	43.4	46.6	1250	3.01	43.1	56.1	48.3	6.02	86.2	1.00	63.7	
114.3 x 6.0 CHS	16.0	0.389	22.4	19.1	2040	3.00	52.5	70.4	38.3	6.00	105	1.00	C	
4.8 CHS	13.0	0.389	27.7	23.8	1650	2.48	43.4	57.6	38.8	4.96	86.8	1.00	53.3	
3.8 CHS	9.83	0.389	36.5	31.8	1250	1.92	33.6	44.1	39.2	3.84	67.2	1.00	57.6	
3.2 CHS	8.77	0.389	41.0	35.7	1120	1.72	30.2	39.5	39.3	3.45	60.4	1.00	44.1	

Notes: 1. For Grade C350 fy = 350 MPa and  $f_y = 430$  MPa.

2. fy = Yield stress used in design; fu = tensile strength used in design; as defined in AS 4100.

3. C = Compact Section; N = Non-compact Section; S = Slender Section; as defined in AS 4100.

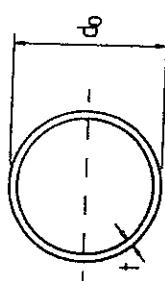
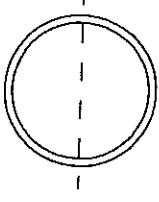


TABLE 1.2 – 2(2)  
DIMENSIONS AND PROPERTIES  
CIRCULAR HOLLOW SECTIONS: GRADE C350



DIMENSIONS AND RATIOS				PROPERTIES						PROPERTIES FOR DESIGN TO AS 4100		
Designation	Mass per m	External Surface Area per m part t	$\frac{d_o}{t}$	Gross Section Area			About any axis			Torsion Constant $J$	Torsion Modulus $10^6 \text{ mm}^4$	Form Factor $C$
				$A_g$ $\text{mm}^2$	$10^6 \text{ mm}^4$	$10^3 \text{ mm}^3$	$I$	$Z$	$S$			
101.6 x 3.2 CHS	7.77	0.319	41.1	31.8	969	1.20	23.6	31.0	34.8	2.40	47.2	1.00
2.6 CHS	6.35	0.319	50.3	39.1	809	0.991	19.5	25.5	35.0	1.98	39.0	1.00
88.9 x 5.5 CHS	11.3	0.279	24.7	16.2	1440	1.26	26.3	38.3	29.6	2.52	56.6	1.00
4.8 CHS	9.96	0.279	28.1	18.5	1270	1.12	25.3	34.0	29.8	2.25	50.6	1.00
3.2 CHS	6.76	0.279	41.3	27.8	862	0.792	17.8	23.5	30.3	1.58	35.6	1.00
2.6 CHS	5.53	0.279	50.6	34.2	705	0.657	14.8	19.4	30.5	1.31	29.6	1.00
76.1 x 3.2 CHS	5.75	0.239	41.6	23.8	733	0.486	12.8	17.0	25.8	0.976	25.6	1.00
2.3 CHS	4.19	0.239	57.1	33.1	533	0.363	9.35	12.5	26.1	0.727	19.1	1.00
60.3 x 2.9 CHS	4.11	0.189	46.1	20.8	523	0.216	7.16	9.56	20.3	0.432	14.3	1.00
2.3 CHS	3.29	0.189	57.6	26.2	419	0.177	5.85	7.74	20.5	0.353	11.7	1.00
48.3 x 2.9 CHS	3.25	0.152	46.7	16.7	414	0.107	4.43	5.99	16.1	0.214	8.86	1.00
2.3 CHS	2.61	0.152	58.2	21.0	332	0.0881	3.65	4.87	16.3	0.176	7.30	1.00
42.4 x 2.6 CHS	2.55	0.133	52.2	16.3	325	0.0846	3.05	4.12	14.1	0.129	6.10	1.00
2.0 CHS	1.89	0.133	66.8	21.2	254	0.0519	2.45	3.27	14.3	0.104	4.90	1.00
33.7 x 2.6 CHS	1.99	0.106	53.1	13.0	254	0.0309	1.84	2.52	11.0	0.0619	3.67	1.00
2.0 CHS	1.56	0.106	67.7	16.9	199	0.0251	1.49	2.01	11.2	0.0502	2.98	1.00
26.9 x 2.3 CHS	1.40	0.0845	60.6	11.7	178	0.0136	1.01	1.40	8.74	0.0271	2.02	1.00
2.0 CHS	1.23	0.0845	68.8	13.5	156	0.0122	0.907	1.24	8.83	0.0244	1.81	1.00
21.3 x 2.0 CHS	0.952	0.0689	70.3	10.7	121	0.00571	0.536	0.748	6.86	0.0114	1.07	1.00
												C 0.748

Notes: 1. For Grade C350  $f_y = 350 \text{ MPa}$ , and  $f_u = 430 \text{ MPa}$ .

$f_y$  = yield stress used in design;  $f_u$  = tensile strength used in design; as defined in AS 4100.

2. Grade C350 to AS 1163 is cold formed and therefore is allocated the CF residual stresses classification in AS 4100.

3. C = Compact Section; N = Non-compact Section; S = Shaped Section; as defined in AS 4100.

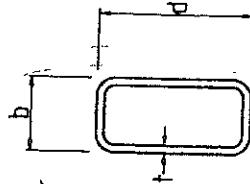
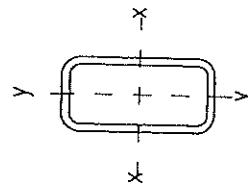


TABLE 1.2-3(1)

## DIMENSIONS AND PROPERTIES

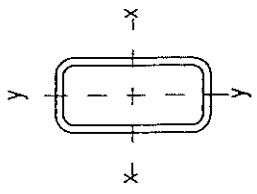
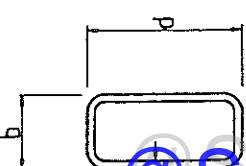
## RECTANGULAR HOLLOW SECTIONS: GRADE C350

DIMENSIONS AND RATIOS							PROPERTIES										PROPERTIES											
Designation	Mass per m	External Surface Area per m	$\frac{b-2t}{t}$	Gross Section			About x-axis						About y-axis						About x-axis						About y-axis			
				$A_g$	$\frac{10^3 \text{ mm}^2}{\text{m}}$	$\frac{10^3 \text{ mm}^3}{\text{m}}$	$I_x$	$S_x$	$t_x$	$b$	$Z_x$	$S_y$	$t_y$	$Z_y$	$I_y$	$J$	$C$	$k_t$	$f_y$	$Z_{xy}$	$I_{xy}$	Torsion Constant	Torsion Modulus	Form Factor	About x-axis	Compactness	About y-axis	
250 x 150 x 9.0 RHS	51.8	0.761	14.7	14.7	25.8	6600	53.7	430	533	90.2	-24.3	324	375	60.7	56.0	554	1.00	C	533	N	373	(CNS) $10^3 \text{ mm}^3$	(CNS) $10^3 \text{ mm}^3$	(CNS) $10^3 \text{ mm}^3$	Compactness	Compactness	Compactness	
6.0 RHS	35.6	0.774	21.8	23.0	39.7	4830	38.4	307	374	92.0	-17.5	233	264	62.2	39.0	395	0.907	N	374	S	208							
5.0 RHS	29.9	0.779	26.0	28.0	48.0	3810	32.7	262	317	92.6	15.0	199	224	62.6	33.0	337	0.814	N	300	S	156							
200 x 100 x 9.0 RHS	37.2	0.561	14.9	9.11	20.2	4800	22.8	228	293	68.9	7.64	153	180	39.9	19.9	272	1.00	C	293	C	180	(CNS) $10^3 \text{ mm}^3$	(CNS) $10^3 \text{ mm}^3$	(CNS) $10^3 \text{ mm}^3$	Compactness	Compactness	Compactness	
6.0 RHS	22.1	0.574	22.0	14.7	31.3	3330	16.7	167	210	70.8	5.69	114	130	41.3	14.2	200	1.00	C	210	N	119							
5.0 RHS	17.9	0.583	32.5	23.0	48.0	2810	11.9	119	144	179	71.5	4.92	98.3	111	41.8	12.1	172	0.925	C	179	S	90.1						
150 x 100 x 6.0 RHS	21.4	0.474	22.1	14.7	23.0	2730	8.17	109	134	54.7	4.36	81.5	91.0	42.3	9.89	142	0.891	C	147	S	63.1							
5.0 RHS	18.2	0.479	26.3	18.0	28.0	2310	7.07	94.3	115	55.3	3.79	87.3	102	40.0	9.51	147	1.00	C	134	C	102							
4.0 RHS	14.8	0.483	32.7	23.0	35.5	1880	5.87	78.2	94.6	55.9	3.15	63.0	71.8	40.4	8.12	127	1.00	C	115	C	94.6							
150 x 50 x 5.0 RHS	14.2	0.379	26.6	8.00	28.0	1810	4.44	59.2	78.9	49.5	0.765	30.6	36.7	20.5	6.64	105	0.971	C	94.6	S	60.9							
4.0 RHS	11.6	0.383	32.9	10.5	35.5	1480	3.74	49.8	65.4	50.2	0.653	26.1	29.8	21.0	4.92	66.8	1.00	C	78.9	N	34.1							
3.0 RHS	8.96	0.390	43.5	14.7	48.0	1140	2.99	39.8	51.4	51.2	0.526	21.1	23.5	21.5	1.50	38.3	0.776	C	65.4	S	25.1							
125 x 75 x 6.0 RHS	16.7	0.374	22.4	10.5	18.8	2130	4.16	66.6	84.2	44.2	1.87	50.0	59.1	29.6	4.44	98.2	1.00	C	72.7	C	59.1							
5.0 RHS	14.2	0.379	26.6	13.0	23.0	1810	3.64	58.3	72.7	44.8	1.39	43.9	51.1	30.1	4.44	88.2	1.00	C	63.0	C	51.1							
4.0 RHS	11.6	0.383	32.9	16.8	29.3	1480	3.05	48.9	60.3	45.4	1.11	37.0	42.4	30.6	3.16	63.0	1.00	C	60.3	C	39.9							
3.0 RHS	8.96	0.390	43.5	23.0	39.7	1140	2.43	38.9	47.3	46.1	1.11	29.5	33.3	31.1	2.43	49.5	0.908	C	47.3	S	26.3							
100 x 50 x 6.0 RHS	12.0	0.274	22.8	6.33	14.7	1530	1.71	34.2	45.3	33.4	0.567	22.7	27.7	19.2	1.53	40.9	1.00	C	45.3	C	27.7							
4.0 RHS	10.3	0.279	27.0	8.00	18.0	1310	1.53	30.6	39.8	34.1	0.511	20.4	24.4	19.7	1.35	36.5	1.00	C	39.8	C	24.4							
3.5 RHS	7.53	0.283	33.3	10.5	23.0	1080	1.31	26.1	33.4	34.8	0.441	17.6	20.2	1.13	31.2	1.00	C	33.4	C	20.6								
3.0 RHS	6.60	0.290	43.5	37.9	12.3	959	1.18	23.6	35.1	40.0	0.400	16.0	18.5	20.4	1.01	28.2	1.00	C	29.9	N	18.1							
2.5 RHS	5.56	0.291	52.4	14.7	31.3	841	1.06	21.3	35.6	36.1	0.361	14.4	16.4	20.7	0.886	25.0	1.00	C	28.7	S	15.0							
2.0 RHS	4.50	0.293	65.1	23.0	48.0	574	0.750	15.0	18.5	36.2	0.257	10.3	14.0	20.9	0.754	21.5	0.926	C	22.7	S	11.4							
																0.616	17.7	0.802	C	18.5	S	7.98						

Notes: 1. For Grade C350  $f_y = 350$  MPa and end  $t_0 = 40$  MPa.

2. Grade C350 to AS 1163 is cold formed and therefore is allocated the CF (fluted) stress class; classification in AS 4100.

3. C = Compact Section; N = Non-compact Section; S = Slender Section; as defined in AS 4100.



**TABLE 1.2-3(2)**  
**DIMENSIONS AND PROPERTIES**  
**RECTANGULAR HOLLOW SECTIONS: GRADE C350**

DIMENSIONS AND RATIOS						PROPERTIES										PROPERTIES FOR DESIGN TO AS 4100			
Designation	Mass per m	External Surface Area per t	$\frac{b-2t}{t}$	$\frac{d-2t}{t}$	Gross Section Area $A_g$	About x-axis				About y-axis				Torsion Constant $J$	Torsion Modulus $C$	Form Factor $k_f$	About x-axis Compactness (C.N.S) $10^3 \text{mm}^3$	About y-axis Compactness (C.N.S) $10^3 \text{mm}^3$	
						$I_x$	$Z_x$	$S_x$	$I_y$	$Z_y$	$S_y$	$I_y$	$Z_y$						
55 x 50 x 6.0 RHS	9.67	0.224	23.2	6.33	10.5	1230	0.800	21.3	28.1	25.5	0.421	16.9	21.1	29.3	1.00	C	28.1	21.1	
55 x 50 x 5.0 RHS	8.35	0.229	27.4	8.00	13.0	1060	0.726	19.4	24.9	26.5	0.384	15.4	18.9	0.891	1.00	C	24.9	18.8	
55 x 50 x 4.0 RHS	6.92	0.233	33.7	10.5	16.8	881	0.630	16.8	21.1	26.7	0.335	13.4	16.0	0.754	22.7	C	21.1	16.0	
55 x 50 x 3.0 RHS	5.42	0.240	44.2	14.7	23.0	691	0.522	13.9	17.1	27.5	0.278	11.1	12.9	0.593	18.4	C	17.1	12.9	
55 x 50 x 2.5 RHS	4.58	0.241	52.7	18.0	28.0	584	0.450	12.0	14.6	24.0	0.240	9.60	11.0	0.505	15.9	C	14.6	10.6	
55 x 50 x 2.0 RHS	3.72	0.243	66.4	23.0	35.5	474	0.372	9.91	12.0	28.0	0.199	7.96	9.06	0.205	0.414	13.1	0.971	12.0	
55 x 25 x 25 RHS	3.60	0.191	53.1	8.00	26.0	459	0.285	7.60	10.1	24.9	0.0487	3.89	4.53	10.3	0.144	7.14	1.00	N	
55 x 25 x 2.0 RHS	2.89	0.195	65.8	10.5	35.5	374	0.238	6.36	8.31	25.3	0.0414	3.31	3.77	10.5	0.120	6.04	0.964	C	
55 x 25 x 1.6 RHS	2.38	0.195	81.7	13.6	44.9	303	0.197	5.26	6.81	25.5	0.0347	2.78	3.11	10.7	0.0993	5.05	0.813	S	
65 x 35 x 30 RHS	4.25	0.190	44.7	9.67	19.7	541	0.281	8.65	11.0	22.8	0.106	6.04	7.11	14.0	0.289	10.4	1.00	C	
65 x 35 x 25 RHS	3.60	0.191	53.1	12.0	30.5	459	0.244	7.52	9.45	23.1	0.0926	5.29	6.13	14.2	0.223	9.10	1.00	C	
65 x 35 x 20 RHS	2.93	0.193	65.8	15.5	30.5	374	0.204	6.28	7.80	23.4	0.0778	4.44	5.07	14.4	0.184	7.62	1.00	C	
50 x 25 x 30 RHS	3.07	0.140	45.5	6.33	14.7	391	0.112	4.47	5.86	16.9	0.0367	2.93	3.56	9.69	0.0964	5.18	1.00	C	
50 x 25 x 2.5 RHS	2.62	0.141	54.0	8.00	18.0	334	0.0989	3.95	5.11	17.2	0.0328	2.62	3.12	9.91	0.0443	4.60	1.00	C	
50 x 25 x 2.0 RHS	2.15	0.143	66.6	10.5	23.0	274	0.0838	3.35	4.26	17.5	0.0281	2.25	2.62	10.1	0.0706	3.92	1.00	C	
50 x 25 x 1.6 RHS	1.75	0.145	82.5	13.6	29.3	223	0.0702	2.81	3.53	17.7	0.0237	1.90	2.17	10.3	0.0385	3.29	1.00	C	
50 x 20 x 30 RHS	2.83	0.130	45.8	4.67	14.7	361	0.0951	3.81	5.16	16.2	0.0212	2.12	2.63	7.67	0.0520	3.88	1.00	C	
50 x 20 x 2.5 RHS	2.42	0.131	54.2	6.00	18.0	309	0.0848	3.39	4.51	16.6	0.0192	1.92	2.32	7.89	0.0550	3.49	1.00	C	
50 x 20 x 2.0 RHS	1.99	0.133	66.8	8.00	23.0	254	0.0723	2.89	3.78	16.9	0.0167	1.67	1.95	8.11	0.0466	3.00	1.00	C	
50 x 20 x 1.6 RHS	1.63	0.135	82.7	10.5	29.3	207	0.0668	2.43	3.14	17.1	0.0142	1.42	1.63	8.29	0.0389	2.55	1.00	C	

Notes: 1. For Grade C350  $y = 350 \text{ MPa}$  and  $f_u = 430 \text{ MPa}$ .

2.  $y$  = yield stress used in design;  $f_u$  = tensile strength used in design as defined in AS 4100.

3. C = Compact Section; N = Non-compact Section; S = Slender Section; as defined in AS 4100.

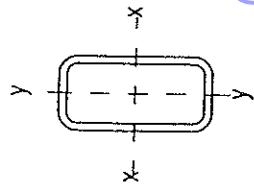
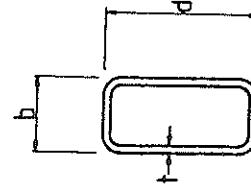


TABLE 1.2-4  
DIMENSIONS AND PROPERTIES  
RECTANGULAR HOLLOW SECTIONS: GRADE C450

DIMENSIONS AND RATIOS							PROPERTIES										PROPERTIES FOR DESIGN TO AS 4100					
Designation	Mass per m	External Surface Area per m	$\frac{b-2t}{t}$	$\frac{d-2t}{t}$	Gross Section		About x-axis				About y-axis				Torsion Constant	Torsion Modulus	Form Factor	About x-axis Compactness	About y-axis Compactness	About x-axis Slenderness	About y-axis Slenderness	
					Area $A_g$	$10^3 \text{ mm}^2$	$I_x$ $10^8 \text{ mm}^4$	$Z_x$ $10^3 \text{ mm}^3$	$S_x$ $10^6 \text{ mm}^3$	$I_y$ $10^8 \text{ mm}^4$	$Z_y$ $10^3 \text{ mm}^3$	$S_y$ $10^6 \text{ mm}^3$	$J$	$J_y$	$C$	$10^3 \text{ mm}^3$						
125 x 75 x 3 RHS	11.1	0.384	34.6	17.7	30.9	1410	2.93	46.8	57.7	45.5	1.33	35.5	40.6	30.7	3.02	60.4	0.978	C	57.7	S	34.6	
3 x 3 RHS	9.73	0.386	39.7	20.7	35.9	1240	2.60	41.6	50.9	45.8	1.19	31.6	35.9	30.9	2.66	53.6	0.893	C	50.9	S	27.7	
2 x 3 RHS	8.39	0.390	46.5	24.8	42.6	1070	2.29	36.6	44.4	46.2	1.04	27.8	31.3	31.2	2.28	46.6	0.812	N	41.8	S	21.7	
100 x 50 x 3 RHS	7.14	0.286	40.1	13.2	28.3	909	1.13	22.6	28.5	35.2	0.875	23.3	26.1	31.4	1.90	39.1	0.721	C	30.2	S	16.1	
—	6.19	0.290	46.9	15.9	33.7	655	1.00	20.1	25.1	35.7	0.841	13.6	15.3	17.6	20.5	0.964	26.9	1.00	C	28.5	N	15.8
2 x 3 RHS	5.14	0.292	56.8	19.7	41.5	655	0.848	17.0	21.0	36.0	0.290	11.6	13.6	15.5	20.8	0.824	23.6	0.922	C	26.1	S	12.5
75 x 50 x 2.8 RHS	5.09	0.240	47.2	15.9	24.8	648	0.493	13.2	16.1	27.6	0.263	10.5	12.2	20.1	0.558	17.4	1.00	C	21.0	S	9.12	
2 x 3 RHS	4.24	0.242	57.1	19.7	30.8	540	0.419	11.2	13.6	27.9	0.224	8.96	10.3	20.4	0.469	14.8	0.984	13.6	N	11.6	S	8.80
65 x 35 x 2.8 RHS	3.99	0.190	47.7	10.5	21.2	508	0.267	8.21	10.4	22.9	0.101	5.75	6.73	14.1	0.245	9.92	1.00	C	10.4	C	6.73	
2 x 3 RHS	3.34	0.192	57.6	13.2	26.3	425	0.229	7.04	8.81	23.2	0.069	4.96	5.72	14.3	0.208	8.53	1.00	C	8.81	N	5.32	
50 x 25 x 2.8 RHS	2.89	0.140	48.5	6.83	16.9	368	0.107	4.27	6.57	17.0	0.0552	2.82	3.39	9.78	0.0917	4.96	1.00	C	5.57	C	3.39	
2 x 3 RHS	2.44	0.142	58.4	8.87	19.7	310	0.0931	3.72	4.78	17.3	0.0310	2.48	2.92	10.0	0.0790	4.34	1.00	C	4.78	C	2.92	
50 x 20 x 2.8 RHS	2.67	0.130	48.8	5.14	15.9	340	0.0912	3.65	4.91	16.4	0.0205	2.05	2.51	7.76	0.594	3.73	1.00	C	4.91	C	2.51	
2 x 3 RHS	2.25	0.132	58.6	6.70	19.7	287	0.0800	3.20	4.23	16.7	0.0183	1.83	2.18	7.98	0.0518	3.31	1.00	C	4.23	C	2.18	

Notes: 1. For Grade C450  $f_y = 450 \text{ MPa}$  and  $f_u = 500 \text{ MPa}$ .

2. Yield stress used in design;  $f_u$  = tensile strength used in design; as defined in AS 4100.

3. C = Compact Section; N = Non-compact Section; S = Slender Section; as defined in AS 4100.

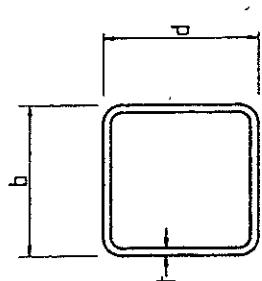
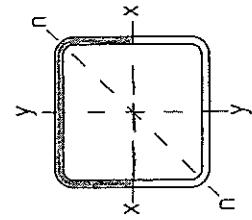


TABLE 1.2 – 5(1)

**DIMENSIONS AND PROPERTIES  
SQUARE HOLLOW SECTIONS: GRADE C350**



DIMENSIONS AND RATIOS				PROPERTIES								PROPERTIES FOR DESIGN TO AS 4100			
Designation	Mass per m	External Surface Area per m	$\frac{b - 2t}{4}$	Cross Section Area				About x-x, y-y and n-n axis				Torsion Constant	Torsion Modulus	Form Factor	
				$A_g$	$I_x$	$Z_x$	$Z_n$	$S_x$	$r_x$	$J$	$C$				
				$\text{mm}^2$	$10^6 \text{ mm}^4$	$10^3 \text{ mm}^3$	$10^3 \text{ mm}^3$	$\text{mm}$	$10^6 \text{ mm}^4$			$10^3 \text{ mm}^3$			
250 x 250 x 9.0 SHS	65.9	0.981	14.8	25.8	8400	78.8	639	477	750	97.5	129	972	1.00	N	
250 x 250 x 9.0 SHS	45.0	0.974	21.7	39.7	5730	56.2	450	330	521	99.0	88.7	681	0.953	S	
200 x 200 x 9.0 SHS	51.8	0.781	14.7	20.2	6600	39.2	392	297	465	77.1	64.5	599	1.00	C	
200 x 200 x 9.0 SHS	35.6	0.774	21.8	31.3	4520	28.0	280	207	327	78.6	44.8	425	1.00	N	
200 x 200 x 9.0 SHS	5.0 SHS	29.9	26.0	38.0	3810	23.9	239	175	277	79.1	37.8	362	0.980	S	
150 x 150 x 9.0 SHS	37.7	0.561	14.9	14.7	4800	15.4	205	158	248	56.6	26.1	216	1.00	C	
150 x 150 x 9.0 SHS	6.0 SHS	26.2	22.0	23.0	3330	11.3	150	113	178	58.2	18.4	229	1.00	C	
150 x 150 x 9.0 SHS	5.0 SHS	22.1	0.574	28.2	2810	9.70	129	96.1	151	58.7	15.6	197	1.00	N	
125 x 125 x 9.0 SHS	30.6	0.461	15.1	11.9	3900	8.38	134	106	165	46.4	14.5	208	1.00	C	
125 x 125 x 9.0 SHS	6.0 SHS	21.4	0.474	22.1	18.8	2730	6.29	101	76.5	120	48.0	154	1.00	C	
125 x 125 x 9.0 SHS	5.0 SHS	18.2	0.479	26.3	23.0	2310	5.44	87.1	65.4	103	48.5	88.7	1.00	C	
125 x 125 x 9.0 SHS	4.0 SHS	14.8	0.483	32.7	29.3	1680	4.52	72.3	53.6	84.5	49.0	7.25	110	1.00	N
100 x 100 x 9.0 SHS	23.5	0.361	15.4	9.11	3000	3.91	78.1	63.6	98.6	36.1	7.00	123	1.00	C	
100 x 100 x 9.0 SHS	6.0 SHS	16.7	0.374	22.4	14.7	2130	3.04	60.7	47.1	73.5	37.7	5.15	93.6	1.00	C
100 x 100 x 9.0 SHS	5.0 SHS	14.2	0.379	26.6	18.0	1810	2.66	53.1	40.5	63.5	38.3	4.42	81.4	1.00	C
100 x 100 x 9.0 SHS	4.0 SHS	11.6	0.383	32.9	23.0	1480	2.23	44.6	33.5	52.6	38.8	3.63	68.0	1.00	C
100 x 100 x 9.0 SHS	3.0 SHS	8.96	0.380	43.5	31.3	1410	1.77	35.4	26.0	41.2	39.4	2.79	53.2	1.00	N
89 x 89 x 6.0 SHS	14.6	0.330	22.5	12.8	1870	2.06	46.2	36.3	56.6	33.2	3.54	71.6	1.00	C	
89 x 89 x 6.0 SHS	5.0 SHS	12.5	0.334	26.7	15.8	1590	1.81	40.7	31.4	49.1	33.7	3.05	62.7	1.00	C
89 x 89 x 6.0 SHS	3.5 SHS	9.06	0.341	37.6	23.4	1150	1.37	30.9	23.2	36.5	34.5	2.24	47.1	1.00	C
75 x 75 x 6.0 SHS	12.0	0.274	22.6	10.5	1530	1.16	30.9	24.7	38.4	27.5	2.04	48.2	1.00	C	
75 x 75 x 6.0 SHS	5.0 SHS	10.3	0.279	27.0	13.0	1310	1.03	27.5	21.6	33.6	28.0	1.77	42.6	1.00	C
75 x 75 x 6.0 SHS	4.0 SHS	8.49	0.283	33.3	16.8	1080	0.882	23.5	18.0	28.2	28.6	1.48	36.1	1.00	C
75 x 75 x 6.0 SHS	3.5 SHS	7.53	0.285	37.9	19.4	959	0.797	21.3	16.1	25.3	26.8	1.32	32.5	1.00	C
75 x 75 x 6.0 SHS	3.0 SHS	6.60	0.290	43.9	23.0	841	0.716	19.1	14.2	22.5	29.2	1.15	28.7	1.00	C
75 x 75 x 6.0 SHS	2.5 SHS	5.56	0.291	52.4	28.0	709	0.614	16.4	12.0	19.1	29.4	0.971	24.6	1.00	N

Notes: 1. For Grade C350  $f_y = 350 \text{ MPa}$  and  $f_u = 420 \text{ MPa}$ .

$f_y$  = yield stress used in design;  $f_u$  = tensile strength used in design; as defined in AS 4100.

2. Grade C350 to AS 1163 is cold formed and therefore is allocated the CF residual stresses classification in AS 4100.

3. C = Compact Section; N = Non-compact Section; S = Slender Section; as defined in AS 4100.

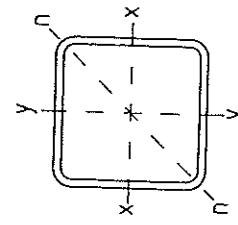


TABLE 1.2 – 5(2)

**DIMENSIONS AND PROPERTIES  
SQUARE HOLLOW SECTIONS: GRADE C350**

DIMENSIONS AND RATIOS				PROPERTIES								PROPERTIES FOR DESIGN - TO AS 4100			
Designation	Mass per m	External Surface Area per m	$\frac{b-2t}{t}$	Gross Section Areas				About x-, y-, and n-axis				Torsion Constant	Torsion Modulus	$K_t$	Compactness $Z_c$
				$A_g$	$I_x$	$Z_x$	$Z_n$	$S_x$	$I_x$	$J$	$C$				
d mm	b mm	t mm	$\frac{b-2t}{t}$	$\text{mm}^2/\text{m}$	$\text{mm}^3/\text{m}$	$10^8 \text{ mm}^4$	$10^3 \text{ mm}^3$	$10^6 \text{ mm}^3$	$10^6 \text{ mm}^4$	$10^6 \text{ mm}^4$	$10^3 \text{ mm}^3$	$10^3 \text{ mm}^3$	$(\text{C}, \text{N}, \text{S})$	$10^3 \text{ mm}^3$	
65 x 65 x 6.0 SHS	10.1	0.234	23.1	8.83	1290	0.706	21.7	17.8	27.5	23.4	1.27	34.2	1.00	C	27.5
50 x 50 x 7.2 SHS	8.75	0.239	27.3	11.0	1110	0.638	19.6	15.6	24.3	22.9	1.12	30.6	1.00	C	24.3
40 x 40 x 8.0 SHS	7.23	0.243	33.6	14.3	921	0.552	17.0	13.2	20.6	24.5	0.939	26.2	1.00	C	20.6
30 x 30 x 5.6 SHS	5.66	0.250	44.1	19.7	721	0.454	14.0	10.4	16.6	25.1	0.733	21.0	1.00	C	16.6
25 x 25 x 4.78 SHS	4.78	0.251	52.6	24.0	609	0.391	12.0	8.91	14.1	25.3	0.624	18.1	1.00	C	14.1
20 x 20 x 3.88 SHS	3.88	0.263	65.3	30.5	494	0.323	9.94	7.29	11.6	25.6	0.509	14.9	1.00	N	10.6
60 x 60 x 5.0 SHS	6.39	0.179	27.9	8.00	814	0.257	10.3	8.61	13.2	17.8	0.469	16.3	1.00	C	13.2
50 x 50 x 5.35 SHS	5.35	0.183	34.2	10.5	681	0.229	9.15	7.33	11.4	18.3	0.403	14.3	1.00	C	11.4
40 x 40 x 4.25 SHS	4.25	0.190	44.7	14.7	541	0.195	7.79	5.92	9.39	19.0	0.321	11.8	1.00	C	9.39
25 x 25 x 3.60 SHS	3.60	0.191	53.1	18.0	459	0.169	6.78	5.09	8.07	19.2	0.275	10.2	1.00	C	8.07
20 x 20 x 2.93 SHS	2.93	0.193	65.8	23.0	374	0.141	5.66	4.20	6.66	19.5	0.225	8.51	1.00	C	6.66
16 x 16 x 2.38 SHS	2.38	0.195	61.7	29.3	303	0.117	4.68	3.44	5.46	19.6	0.185	7.03	1.00	N	5.10
40 x 40 x 4.0 SHS	4.09	0.143	34.9	8.00	521	0.105	5.26	4.36	6.74	14.2	0.192	8.33	1.00	C	6.74
30 x 30 x 3.30 SHS	3.30	0.150	45.3	11.3	421	0.0932	4.66	3.61	5.72	14.9	0.168	7.07	1.00	C	5.72
25 x 25 x 2.82 SHS	2.82	0.161	53.7	14.0	359	0.0822	4.11	3.13	4.97	15.1	0.136	6.21	1.00	C	4.97
20 x 20 x 2.31 SHS	2.31	0.163	66.4	18.0	294	0.0694	3.47	2.61	4.13	15.4	0.113	5.23	1.00	C	4.13
16 x 16 x 1.88 SHS	1.88	0.155	82.3	23.0	239	0.0579	2.90	2.15	3.41	15.6	0.0927	4.36	1.00	C	3.41
35 x 35 x 3.0 SHS	2.83	0.130	46.8	9.67	361	0.0595	3.40	2.67	4.23	12.8	0.102	5.18	1.00	C	4.23
25 x 25 x 2.42 SHS	2.42	0.131	54.2	12.0	309	0.0529	3.02	2.33	3.69	13.1	0.0889	4.58	1.00	C	3.69
20 x 20 x 1.99 SHS	1.99	0.133	66.8	15.5	254	0.0451	2.38	1.95	3.09	13.3	0.0741	3.89	1.00	C	3.09
16 x 16 x 1.63 SHS	1.63	0.135	82.7	19.9	207	0.0379	2.16	1.62	2.57	13.5	0.0511	3.26	1.00	C	2.57
30 x 30 x 3.0 SHS	2.36	0.110	46.5	8.0	301	0.0350	2.34	1.87	2.96	10.8	0.0615	3.58	1.00	C	2.96
25 x 25 x 2.03 SHS	2.03	0.111	54.8	10.0	259	0.0316	2.10	1.65	2.61	11.0	0.0540	3.20	1.00	C	2.61
20 x 20 x 1.88 SHS	1.88	0.113	67.4	13.0	214	0.0272	1.81	1.39	2.21	11.3	0.0454	2.75	1.00	C	2.21
16 x 16 x 1.38 SHS	1.38	0.115	83.3	16.8	175	0.0231	1.54	1.16	1.84	11.5	0.0377	2.32	1.00	C	1.84
25 x 25 x 3.0 SHS	1.89	0.0897	47.4	6.33	241	0.0184	1.47	1.21	1.91	8.74	0.0333	2.27	1.00	C	1.91
25 x 25 x 2.5 SHS	1.64	0.0914	55.7	8.00	209	0.0169	1.35	1.08	1.71	8.99	0.0297	2.07	1.00	C	1.71
20 x 20 x 1.86 SHS	1.36	0.0931	68.3	10.5	174	0.0148	1.19	0.926	1.47	9.24	0.0253	1.80	1.00	C	1.47
16 x 16 x 1.12 SHS	1.12	0.0945	84.1	13.6	143	0.0128	1.02	0.780	1.24	9.44	0.0212	1.54	1.00	C	1.24
20 x 20 x 1.6 SHS	0.873	0.0745	85.4	10.5	111	0.00808	0.608	0.474	0.751	7.39	0.0103	0.924	1.00	C	0.751

Notes: 1. For Grade C350  $f_y = 350$  MPa and  $f_u = 420$  MPa.  
 $f_y$  = yield stress used in design;  $f_u$  = tensile strength used in design; as defined in AS 4100.  
2. Grade C350 to AS 1169 is cold formed and therefore is allocated the CF restraint stresses classification in AS 4100.  
3. C = Compact Section; N = Non-compact Section; S = Slender Section; as defined in AS 4100.

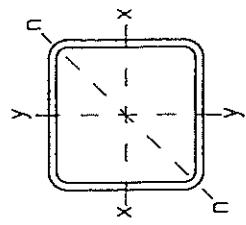
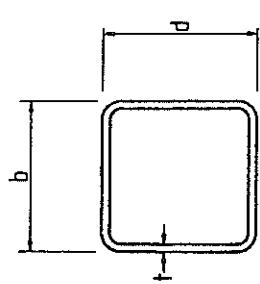


TABLE 1.2 – 6  
DIMENSIONS AND PROPERTIES  
SQUARE HOLLOW SECTIONS: GRADE C450

DIMENSIONS AND RATIOS				PROPERTIES								PROPERTIES FOR DESIGN – TO AS 4100			
Designation	Mass per m	External Surface Area per m	$\frac{b - 2t}{t}$	About x, y, and n-axis				Torsion Constant	Torsion Modulus	Form Factor	About x and y-axis				
				$A_g$	$I_x$	$Z_x$	$Z_n$				$J$	$C$	$k_t$	$Z_u$	
mm mm mm	kg/m	m <sup>2</sup> /m		mm <sup>2</sup>	10 <sup>6</sup> mm <sup>4</sup>	10 <sup>3</sup> mm <sup>3</sup>	10 <sup>3</sup> mm <sup>3</sup>	mm	10 <sup>6</sup> mm <sup>4</sup>	10 <sup>3</sup> mm <sup>3</sup>	(C,N,S)	10 <sup>3</sup> mm <sup>3</sup>			10 <sup>3</sup> mm <sup>3</sup>
100 x 100 x 3.8 SHS	11.1	0.384	34.6	24.3	1410	2.14	42.7	32.0	50.3	38.9	3.47	65.1	1.00	N	48.3
3.3 SHS	9.73	0.386	39.7	28.3	1240	1.90	38.0	28.2	44.5	39.2	3.05	57.7	1.00	N	39.3
2.8 SHS	8.39	0.390	46.5	33.7	1070	1.67	33.3	24.4	38.7	39.5	2.61	50.0	0.886	S	31.0
2.3 SHS	6.95	0.392	56.4	41.5	885	1.40	27.9	20.3	32.3	39.7	2.17	41.9	0.721	S	23.1
75 x 75 SHS	7.14	0.266	40.1	20.7	909	0.761	20.3	15.3	24.1	28.9	1.25	31.0	1.00	C	24.1
2.8 SHS	6.19	0.290	46.9	24.8	788	0.676	18.0	13.3	21.2	29.3	1.08	27.1	1.00	N	20.1
2.3 SHS	5.14	0.292	56.8	30.6	655	0.571	15.2	11.2	17.7	29.5	0.900	22.9	0.974	S	15.0
65 x 65 SHS	5.31	0.260	47.2	21.2	676	0.429	13.2	9.64	15.6	25.2	0.690	19.9	1.00	C	15.6
2.3 SHS	4.42	0.262	57.0	26.3	563	0.364	11.2	8.27	13.1	25.4	0.579	16.9	1.00	N	12.1
50 x 50 x 2.8 SHS	3.99	0.190	47.7	15.9	508	0.185	7.40	5.60	8.87	19.1	0.303	11.2	1.00	C	8.87
2.3 SHS	3.34	0.192	57.6	19.7	425	0.159	6.34	4.74	7.52	19.3	0.256	9.55	1.00	C	7.52
40 x 40 x 2.8 SHS	3.11	0.160	48.3	12.3	396	0.0890	4.45	3.43	5.43	15.0	0.149	6.74	1.00	C	5.43
2.3 SHS	2.62	0.162	58.1	15.4	333	0.0773	3.86	2.93	4.64	15.2	0.127	5.83	1.00	C	4.84
35 x 35 x 2.8 SHS	2.67	0.130	48.8	10.5	340	0.0570	3.26	2.54	4.02	12.9	0.097	4.95	1.00	C	4.02
2.3 SHS	2.25	0.132	58.6	13.2	287	0.0499	2.85	2.19	3.46	13.2	0.0891	4.32	1.00	C	3.46
30 x 30 x 2.8 SHS	2.23	0.110	49.4	8.71	284	0.0357	2.25	1.79	2.83	10.9	0.0586	3.44	1.00	C	2.83
2.3 SHS	1.89	0.112	58.2	11.0	241	0.0299	2.00	1.55	2.45	11.1	0.0507	3.03	1.00	C	2.45
25 x 25 x 2.3 SHS	1.53	0.0921	60.1	8.87	195	0.0161	1.29	1.02	1.62	9.09	0.0280	1.97	1.00	C	1.62

Notes:

1. For Grade C450  $f_y = 450$  MPa and  $f_{u\alpha} = 500$  MPa.

$f_y$  = yield stress used in design;  $f_{u\alpha}$  = tensile strength used in design; as defined in AS 4100.

2. Grade C450 to AS 1163 is cold formed and therefore is allocated the CF residual stresses classification in AS 4100.

3. C = Compact Section; N = Non-compact Section; S = Slender Section; as defined in AS 4100.

FIRE ENGINEERING DESIGN  
EXPOSED SURFACE AREA TO MASS RATIO ( $m^2/t$ )  
CIRCULAR HOLLOW SECTIONS: GRADE C250

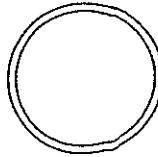
- 1 = TOTAL PERIMETER, PROFILE-PROTECTED  
 2 = TOTAL PERIMETER, BOX-PROTECTED, NO GAP  
 3 = TOTAL PERIMETER, BOX-PROTECTED, 25mm GAP  
 4 = TOP FLANGE EXCLUDED, PROFILE-PROTECTED  
 5 = TOP FLANGE EXCLUDED, BOX-PROTECTED, NO GAP  
 6 = TOP FLANGE EXCLUDED, BOX-PROTECTED, 25mm GAP

Designation $d_o$ mm	Mass perm kg/m	1	2	3	4	5	6
610.0 x 12.7 CHS	187	10.2	—	11.1	—	—	—
9.5 CHS	141	13.6	—	14.7	—	—	—
6.4 CHS	95.3	20.1	—	21.8	—	—	—
508.0 x 12.7 CHS	155	10.3	—	11.3	—	—	—
9.5 CHS	117	13.7	—	15.0	—	—	—
6.4 CHS	79.2	20.2	—	22.1	—	—	—
165.1 x 5.4 CHS	21.3	24.4	—	31.8	—	—	—
5.0 CHS	19.7	26.3	—	34.2	—	—	—
139.7 x 5.4 CHS	17.9	24.5	—	33.3	—	—	—
5.0 CHS	16.6	26.4	—	35.9	—	—	—
114.3 x 5.4 CHS	14.5	24.8	—	35.6	—	—	—
4.5 CHS	12.2	29.5	—	42.4	—	—	—
101.6 x 5.0 CHS	11.9	26.8	—	40.0	—	—	—
4.0 CHS	9.63	33.2	—	49.5	—	—	—
88.9 x 5.9 CHS	12.1	23.1	—	36.1	—	—	—
5.0 CHS	10.3	27.0	—	42.2	—	—	—
4.0 CHS	8.38	33.3	—	52.1	—	—	—
76.1 x 5.9 CHS	10.2	23.4	—	38.8	—	—	—
4.5 CHS	7.95	30.1	—	49.9	—	—	—
3.6 CHS	6.44	37.1	—	61.5	—	—	—
60.3 x 5.4 CHS	7.31	25.9	—	47.4	—	—	—
4.5 CHS	6.19	30.6	—	56.0	—	—	—
3.6 CHS	5.03	37.6	—	68.8	—	—	—
48.3 x 5.4 CHS	5.71	26.6	—	54.1	—	—	—
4.0 CHS	4.37	34.7	—	70.7	—	—	—
3.2 CHS	3.56	42.6	—	86.8	—	—	—
42.4 x 4.9 CHS	4.53	29.4	—	64.1	—	—	—
4.0 CHS	3.79	35.2	—	76.6	—	—	—
3.2 CHS	3.09	43.1	—	93.8	—	—	—
33.7 x 4.5 CHS	3.24	32.7	—	81.1	—	—	—
4.0 CHS	2.93	36.1	—	89.8	—	—	—
3.2 CHS	2.41	44.0	—	109	—	—	—
26.9 x 4.0 CHS	2.26	37.4	—	107	—	—	—
3.2 CHS	1.87	45.2	—	129	—	—	—
2.6 CHS	1.56	54.2	—	155	—	—	—
21.3 x 3.6 CHS	1.57	42.6	—	143	—	—	—
3.2 CHS	1.43	46.8	—	157	—	—	—
2.6 CHS	1.20	55.8	—	187	—	—	—
17.2 x 2.9 CHS	1.02	52.8	—	206	—	—	—
2.3 CHS	0.845	63.9	—	250	—	—	—
13.5 x 2.9 CHS	0.758	55.9	—	263	—	—	—
2.3 CHS	0.635	66.8	—	314	—	—	—

See page 1-7 for details on cases of fire exposure considered.

TABLE 1.3-2

**FIRE ENGINEERING DESIGN  
EXPOSED SURFACE AREA TO MASS RATIO ( $m^2/t$ )  
CIRCULAR HOLLOW SECTIONS: GRADE C350**



- 1 = TOTAL PERIMETER, PROFILE-PROTECTED
- 2 = TOTAL PERIMETER, BOX-PROTECTED, NO GAP
- 3 = TOTAL PERIMETER, BOX-PROTECTED, 25mm GAP
- 4 = TOP FLANGE EXCLUDED, PROFILE-PROTECTED
- 5 = TOP FLANGE EXCLUDED, BOX-PROTECTED, NO GAP
- 6 = TOP FLANGE EXCLUDED, BOX-PROTECTED, 25mm GAP

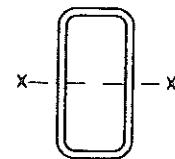
Designation $d_o$ mm	t mm	Mass per m kg/m	1	2	3	4	5	6
457.0 x 12.7 CHS	139	10.3	—	11.4	—	—	—	—
9.5 CHS	105	13.7	—	15.2	—	—	—	—
6.4 CHS	71.1	20.2	—	22.4	—	—	—	—
406.4 x 12.7 CHS	123	10.4	—	11.6	—	—	—	—
9.5 CHS	93.0	13.7	—	15.4	—	—	—	—
6.4 CHS	63.1	20.2	—	22.7	—	—	—	—
355.6 x 12.7 CHS	107	10.4	—	11.9	—	—	—	—
9.5 CHS	81.1	13.8	—	15.7	—	—	—	—
6.4 CHS	55.1	20.3	—	23.1	—	—	—	—
323.9 x 12.7 CHS	97.5	10.4	—	12.1	—	—	—	—
9.5 CHS	73.7	13.8	—	15.9	—	—	—	—
6.4 CHS	50.1	20.3	—	23.4	—	—	—	—
273.1 x 9.3 CHS	60.5	14.2	—	16.8	—	—	—	—
6.4 CHS	42.1	20.4	—	24.1	—	—	—	—
4.8 CHS	31.8	27.0	—	32.0	—	—	—	—
219.1 x 8.2 CHS	42.6	16.1	—	19.8	—	—	—	—
6.4 CHS	33.6	20.5	—	25.2	—	—	—	—
4.8 CHS	25.4	27.1	—	33.3	—	—	—	—
168.3 x 7.1 CHS	28.2	18.7	—	24.3	—	—	—	—
6.4 CHS	25.6	20.7	—	26.8	—	—	—	—
4.8 CHS	19.4	27.3	—	35.4	—	—	—	—
165.1 x 3.5 CHS	13.9	37.2	—	48.4	—	—	—	—
3.0 CHS	12.0	43.2	—	56.3	—	—	—	—
139.7 x 3.5 CHS	11.8	37.3	—	50.7	—	—	—	—
3.0 CHS	10.1	43.4	—	58.9	—	—	—	—
114.3 x 6.0 CHS	16.0	22.4	—	32.2	—	—	—	—
4.8 CHS	13.0	27.7	—	39.8	—	—	—	—
3.6 CHS	9.83	36.5	—	52.5	—	—	—	—
3.2 CHS	8.77	41.0	—	58.9	—	—	—	—
101.6 x 3.2 CHS	7.77	41.1	—	61.3	—	—	—	—
2.6 CHS	6.35	50.3	—	75.0	—	—	—	—
88.9 x 5.5 CHS	11.3	24.7	—	38.6	—	—	—	—
4.8 CHS	9.96	28.1	—	43.8	—	—	—	—
3.2 CHS	6.76	41.3	—	64.5	—	—	—	—
2.6 CHS	5.53	50.5	—	78.9	—	—	—	—
76.1 x 3.2 CHS	5.75	41.6	—	68.9	—	—	—	—
2.3 CHS	4.19	57.1	—	94.6	—	—	—	—
60.3 x 2.9 CHS	4.11	46.1	—	84.4	—	—	—	—
2.3 CHS	3.29	57.6	—	105	—	—	—	—
48.3 x 2.9 CHS	3.25	46.7	—	95.1	—	—	—	—
2.3 CHS	2.61	58.2	—	118	—	—	—	—
42.4 x 2.6 CHS	2.55	52.2	—	114	—	—	—	—
2.0 CHS	1.99	66.8	—	146	—	—	—	—
33.7 x 2.6 CHS	1.99	53.1	—	132	—	—	—	—
2.0 CHS	1.56	67.7	—	168	—	—	—	—
26.9 x 2.3 CHS	1.40	60.6	—	173	—	—	—	—
2.0 CHS	1.23	68.8	—	197	—	—	—	—
21.3 x 2.0 CHS	0.952	70.3	—	235	—	—	—	—

@Seismicisolation  
See page 1-7 for details on cases of fire exposure considered.

TABLE 1.3-3(A)

**FIRE ENGINEERING DESIGN**  
**EXPOSED SURFACE AREA TO MASS RATIO ( $m^2/t$ )**  
**RECTANGULAR HOLLOW SECTIONS: GRADE C350**  
**about x-axis**

- 1 = TOTAL PERIMETER, PROFILE-PROTECTED  
 2 = TOTAL PERIMETER, BOX-PROTECTED, NO GAP  
 3 = TOTAL PERIMETER, BOX-PROTECTED, 25mm GAP  
 4 = TOP FLANGE EXCLUDED, PROFILE-PROTECTED  
 5 = TOP FLANGE EXCLUDED, BOX-PROTECTED, NO GAP  
 6 = TOP FLANGE EXCLUDED, BOX-PROTECTED, 25mm GAP



Designation d b t	Mass perm kg/m	1	2	3	4	5	6
mm mm mm							
250 x 150 x 9.0 RHS	51.8	14.7	15.4	19.3	12.7	12.5	14.5
6.0 RHS	35.6	21.8	22.5	28.1	18.4	18.3	21.1
5.0 RHS	29.9	26.0	26.7	33.4	21.8	21.7	25.0
200 x 100 x 9.0 RHS	37.7	14.9	15.9	21.2	13.4	13.3	15.9
6.0 RHS	26.2	22.0	22.9	30.6	19.3	19.1	22.9
5.0 RHS	22.1	26.2	27.2	36.2	22.8	22.6	27.2
4.0 RHS	17.9	32.5	33.5	44.7	28.1	27.9	33.5
150 x 100 x 6.0 RHS	21.4	22.1	23.3	32.6	18.8	18.6	23.3
5.0 RHS	18.2	26.3	27.5	38.5	22.2	22.0	27.5
4.0 RHS	14.8	32.7	33.9	47.4	27.3	27.1	33.9
150 x 50 x 5.0 RHS	14.2	26.6	28.1	42.1	24.8	24.6	31.6
4.0 RHS	11.6	32.9	34.4	51.6	30.3	30.1	38.7
3.0 RHS	8.96	43.5	44.7	67.0	39.3	39.1	50.2
125 x 75 x 6.0 RHS	16.7	22.4	23.9	35.8	19.7	19.4	25.4
5.0 RHS	14.2	26.6	28.1	42.1	23.1	22.8	29.8
4.0 RHS	11.6	32.9	34.4	51.6	28.2	28.0	36.6
3.0 RHS	8.96	43.5	44.7	67.0	36.5	36.3	47.5
100 x 50 x 6.0 RHS	12.0	22.8	24.9	41.6	21.1	20.8	29.1
5.0 RHS	10.3	27.0	29.1	48.5	24.6	24.2	33.9
4.0 RHS	8.49	33.3	35.4	58.9	29.8	29.5	41.2
3.5 RHS	7.53	37.9	39.9	66.4	33.5	33.2	46.5
3.0 RHS	6.60	43.9	45.5	75.8	38.1	37.9	53.0
2.5 RHS	5.56	52.4	53.9	89.8	45.2	44.9	62.9
2.0 RHS	4.50	65.1	66.6	111	55.8	55.5	77.7
75 x 50 x 6.0 RHS	9.67	23.2	25.8	46.5	21.1	20.7	31.0
5.0 RHS	8.35	27.4	29.9	53.9	24.4	23.9	35.9
4.0 RHS	6.92	33.7	36.1	65.1	29.3	28.9	43.4
3.0 RHS	5.42	44.2	46.1	83.0	37.2	36.9	55.3
2.5 RHS	4.58	52.7	54.5	98.2	43.9	43.6	65.4
2.0 RHS	3.72	65.4	67.2	121	54.1	53.8	80.7
75 x 25 x 2.5 RHS	3.60	53.1	55.5	111	49.0	48.6	76.3
2.0 RHS	2.93	65.8	68.2	136	60.0	59.7	93.7
1.6 RHS	2.38	81.7	84.0	168	73.9	73.5	116
65 x 35 x 3.0 RHS	4.25	44.7	47.1	94.2	39.3	38.9	62.4
2.5 RHS	3.60	53.1	55.5	111	46.2	45.8	73.6
2.0 RHS	2.93	65.8	68.2	136	56.6	56.2	90.3
50 x 25 x 3.0 RHS	3.07	45.5	48.9	114	41.3	40.7	73.3
2.5 RHS	2.62	54.0	57.2	134	48.2	47.7	85.8
2.0 RHS	2.15	66.6	69.8	163	58.7	58.2	105
1.6 RHS	1.75	82.5	85.6	200	71.9	71.4	128
50 x 20 x 3.0 RHS	2.83	45.8	49.4	120	43.0	42.4	77.7
2.5 RHS	2.42	54.2	57.7	140	50.1	49.5	90.7
2.0 RHS	1.99	66.8	70.3	171	60.8	60.3	110
1.6 RHS	1.63	82.7	86.1	209	74.3	73.8	135

See page 1-7 for details on cases of fire exposure considered.

TABLE 1.3-3(B)

**FIRE ENGINEERING DESIGN**  
**EXPOSED SURFACE AREA TO MASS RATIO ( $m^2/t$ )**  
**RECTANGULAR HOLLOW SECTIONS: GRADE C350**  
**about y-axis**

- 1 = TOTAL PERIMETER, PROFILE-PROTECTED  
 2 = TOTAL PERIMETER, BOX-PROTECTED, NO GAP  
 3 = TOTAL PERIMETER, BOX-PROTECTED, 25mm GAP  
 4 = TOP FLANGE EXCLUDED, PROFILE-PROTECTED  
 5 = TOP FLANGE EXCLUDED, BOX-PROTECTED, NO GAP  
 6 = TOP FLANGE EXCLUDED, BOX-PROTECTED, 25mm GAP



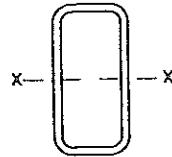
Designation d    b    t	Mass per m	1	2	3	4	5	6
mm   mm   mm	kg/m						
250 x 150 x 9.0 RHS	51.8	14.7	15.4	19.3	10.7	10.6	13.0
6.0 RHS	35.6	21.8	22.5	28.1	15.6	15.5	19.0
5.0 RHS	29.9	26.0	26.7	33.4	18.5	18.4	22.5
200 x 100 x 9.0 RHS	37.7	14.9	15.9	21.2	10.8	10.6	13.9
6.0 RHS	26.2	22.0	22.9	30.6	15.5	15.3	20.1
5.0 RHS	22.1	26.2	27.2	36.2	18.3	18.1	23.8
4.0 RHS	17.9	32.5	33.5	44.7	22.5	22.3	29.3
150 x 100 x 6.0 RHS	21.4	22.1	23.3	32.6	16.5	16.3	22.1
5.0 RHS	18.2	26.3	27.5	38.5	19.5	19.3	26.1
4.0 RHS	14.8	32.7	33.9	47.4	23.9	23.7	32.2
150 x 50 x 6.0 RHS	16.7	22.4	23.9	35.8	16.7	16.4	23.9
5.0 RHS	14.2	26.6	28.1	42.1	17.8	17.6	26.3
4.0 RHS	11.6	32.9	34.4	51.6	21.7	21.5	32.3
3.0 RHS	8.96	43.5	44.7	67.0	28.1	27.9	41.9
125 x 75 x 5.0 RHS	14.2	26.6	28.1	42.1	19.6	19.3	28.1
4.0 RHS	11.6	32.9	34.4	51.6	23.9	23.7	34.4
3.0 RHS	8.96	43.5	44.7	67.0	30.9	30.7	44.7
100 x 50 x 6.0 RHS	12.0	22.8	24.9	41.6	17.0	16.6	27.0
5.0 RHS	10.3	27.0	29.1	48.5	19.7	19.4	31.5
4.0 RHS	8.49	33.3	35.4	58.9	23.9	23.6	38.3
3.5 RHS	7.53	37.9	39.9	66.4	26.9	26.6	43.2
3.0 RHS	6.60	43.9	45.5	75.8	30.6	30.3	49.2
2.5 RHS	5.56	52.4	53.9	89.8	36.2	35.9	58.4
2.0 RHS	4.50	65.1	66.6	111	44.7	44.4	72.2
75 x 50 x 6.0 RHS	9.67	23.2	25.8	46.5	18.5	18.1	31.0
5.0 RHS	8.35	27.4	29.9	53.9	21.4	20.9	35.9
4.0 RHS	6.92	33.7	36.1	65.1	25.7	25.3	43.4
3.0 RHS	5.42	44.2	46.1	83.0	32.6	32.3	55.3
2.5 RHS	4.58	52.7	54.5	98.2	38.5	38.2	65.4
2.0 RHS	3.72	65.4	67.2	121	47.4	47.1	80.7
75 x 25 x 2.5 RHS	3.60	53.1	55.5	111	35.1	34.7	69.4
2.0 RHS	2.93	65.8	68.2	136	43.0	42.6	85.2
1.6 RHS	2.38	81.7	84.0	168	52.9	52.5	105
65 x 35 x 3.0 RHS	4.25	44.7	47.1	94.2	32.2	31.8	61.2
2.5 RHS	3.60	53.1	55.5	111	37.9	37.5	72.2
2.0 RHS	2.93	65.8	68.2	136	46.4	46.0	88.6
50 x 25 x 3.0 RHS	3.07	45.5	48.9	114	33.1	32.6	73.3
2.5 RHS	2.62	54.0	57.2	134	38.7	38.2	85.8
2.0 RHS	2.15	66.6	69.8	163	47.1	46.5	105
1.6 RHS	1.75	82.5	85.6	200	57.6	57.1	128
50 x 20 x 3.0 RHS	2.83	45.8	49.4	120	32.4	31.8	75.9
2.5 RHS	2.42	54.2	57.7	140	37.7	37.1	88.7
2.0 RHS	1.99	66.8	70.3	171	45.8	45.2	108
1.6 RHS	1.63	82.7	86.1	209	55.9	55.3	132

See page 1-7 for details on cases of fire exposure considered.

TABLE 1.3-4(A)

**FIRE ENGINEERING DESIGN**  
**EXPOSED SURFACE AREA TO MASS RATIO ( $m^2/t$ )**  
**RECTANGULAR HOLLOW SECTIONS: GRADE C450**  
**about x-axis**

- 1 = TOTAL PERIMETER, PROFILE-PROTECTED
- 2 = TOTAL PERIMETER, BOX-PROTECTED, NO GAP
- 3 = TOTAL PERIMETER, BOX-PROTECTED, 25mm GAP
- 4 = TOP FLANGE EXCLUDED, PROFILE-PROTECTED
- 5 = TOP FLANGE EXCLUDED, BOX-PROTECTED, NO GAP
- 6 = TOP FLANGE EXCLUDED, BOX-PROTECTED, 25mm GAP



Designation d b t	Mass per m	1	2	3	4	5	6
mm mm mm	kg/m						
125 x 75 x 3.8 RHS	11.1	34.6	36.1	54.1	29.5	29.3	38.3
3.3 RHS	9.73	39.7	41.1	61.7	33.7	33.4	43.7
2.8 RHS	8.39	46.5	47.7	71.5	38.9	38.7	50.7
2.3 RHS	6.95	56.4	57.6	86.3	47.0	46.8	61.2
100 x 50 x 3.3 RHS	7.14	40.1	42.0	70.1	35.4	35.0	49.0
2.8 RHS	6.19	46.9	48.5	80.8	40.6	40.4	56.5
2.3 RHS	5.14	56.8	58.3	97.2	48.9	48.6	68.0
75 x 50 x 2.8 RHS	5.09	47.2	49.1	88.4	39.6	39.3	58.9
2.3 RHS	4.24	57.1	59.0	106	47.5	47.2	70.7
65 x 35 x 2.8 RHS	3.99	47.7	50.1	100	41.7	41.3	66.4
2.3 RHS	3.34	57.6	59.9	120	49.8	49.4	79.4
50 x 25 x 2.8 RHS	2.89	48.5	51.9	121	43.8	43.2	77.8
2.3 RHS	2.44	58.4	61.6	144	51.9	51.3	92.4
50 x 20 x 2.8 RHS	2.67	48.8	52.4	127	45.5	44.9	82.3
2.3 RHS	2.25	58.6	62.1	151	53.8	53.2	97.6

See page 1-7 for details on cases of fire exposure considered.

TABLE 1.3-4(B)

**FIRE ENGINEERING DESIGN**  
**EXPOSED SURFACE AREA TO MASS RATIO ( $m^2/t$ )**  
**RECTANGULAR HOLLOW SECTIONS: GRADE C450**  
**about y-axis**

- 1 = TOTAL PERIMETER, PROFILE-PROTECTED
- 2 = TOTAL PERIMETER, BOX-PROTECTED, NO GAP
- 3 = TOTAL PERIMETER, BOX-PROTECTED, 25mm GAP
- 4 = TOP FLANGE EXCLUDED, PROFILE-PROTECTED
- 5 = TOP FLANGE EXCLUDED, BOX-PROTECTED, NO GAP
- 6 = TOP FLANGE EXCLUDED, BOX-PROTECTED, 25mm GAP



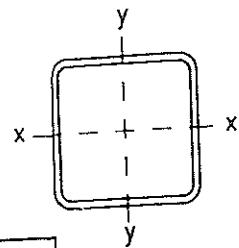
Designation d b t	Mass per m	1	2	3	4	5	6
mm mm mm	kg/m						
125 x 75 x 3.8 RHS	11.1	34.6	36.1	54.1	25.0	24.8	33.8
3.3 RHS	9.73	39.7	41.1	61.7	28.5	28.3	38.6
2.8 RHS	8.39	46.5	47.7	71.5	33.0	32.8	44.7
2.3 RHS	6.95	56.4	57.6	86.3	39.8	39.6	54.0
100 x 50 x 3.3 RHS	7.14	40.1	42.0	70.1	28.4	28.0	42.0
2.8 RHS	6.19	46.9	48.5	80.8	32.6	32.3	48.5
2.3 RHS	5.14	56.8	58.3	97.2	39.1	38.9	58.3
75 x 50 x 2.8 RHS	5.09	47.2	49.1	88.4	34.7	34.4	54.0
2.3 RHS	4.24	57.1	59.0	106	41.6	41.3	64.8
65 x 35 x 2.8 RHS	3.99	47.7	50.1	100	34.2	33.8	58.9
2.3 RHS	3.34	57.6	59.9	120	40.8	40.4	70.4
50 x 25 x 2.8 RHS	2.89	48.5	51.9	121	35.1	34.6	69.1
2.3 RHS	2.44	58.4	61.6	144	41.6	41.1	82.1
50 x 20 x 2.8 RHS	2.67	48.8	52.4	127	34.3	33.7	71.1
2.3 RHS	2.25	58.6	62.1	151	40.5	39.9	84.3

See page 1-7 for details on cases of fire exposure considered.

TABLE 1.3-5

**FIRE ENGINEERING DESIGN**  
**EXPOSED SURFACE AREA TO MASS RATIO ( $m^2/t$ )**  
**SQUARE HOLLOW SECTIONS: GRADE C350**  
**about x- and y-axis**

- 1 = TOTAL PERIMETER, PROFILE-PROTECTED
- 2 = TOTAL PERIMETER, BOX-PROTECTED, NO GAP
- 3 = TOTAL PERIMETER, BOX-PROTECTED, 25mm GAP
- 4 = TOP FLANGE EXCLUDED, PROFILE-PROTECTED
- 5 = TOP FLANGE EXCLUDED, BOX-PROTECTED, NO GAP
- 6 = TOP FLANGE EXCLUDED, BOX-PROTECTED, 25mm GAP

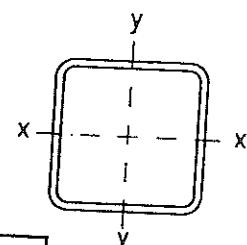


Designation d b t	Mass per m	1	2	3	4	5	6
mm mm mm	kg/m						
250 x 250 x 9.0 SHS	65.9	14.6	15.2	18.2	11.5	11.4	12.9
6.0 SHS	45.0	21.7	22.2	26.7	16.8	16.7	18.9
200 x 200 x 9.0 SHS	51.8	14.7	15.4	19.3	11.7	11.6	13.5
6.0 SHS	35.6	21.8	22.5	28.1	17.0	16.9	19.7
5.0 SHS	29.9	26.0	26.7	33.4	20.2	20.0	23.4
150 x 150 x 9.0 SHS	37.7	14.9	15.9	21.2	12.1	11.9	14.6
6.0 SHS	26.2	22.0	22.9	30.6	17.4	17.2	21.0
5.0 SHS	22.1	26.2	27.2	36.2	20.5	20.4	24.9
125 x 125 x 9.0 SHS	30.6	15.1	16.3	22.9	12.5	12.3	15.5
6.0 SHS	21.4	22.1	23.3	32.6	17.7	17.5	22.1
5.0 SHS	18.2	26.3	27.5	38.5	20.8	20.6	26.1
4.0 SHS	14.8	32.7	33.9	47.4	25.6	25.4	32.2
100 x 100 x 9.0 SHS	23.5	15.4	17.0	25.5	13.0	12.7	17.0
6.0 SHS	16.7	22.4	23.9	35.8	18.2	17.9	23.9
5.0 SHS	14.2	26.6	28.1	42.1	21.3	21.1	28.1
4.0 SHS	11.6	32.9	34.4	51.6	26.0	25.8	34.4
3.0 SHS	8.96	43.5	44.7	67.0	33.7	33.5	44.7
89 x 89 x 6.0 SHS	14.6	22.5	24.3	37.9	18.5	18.2	25.0
5.0 SHS	12.5	26.7	28.5	44.5	21.6	21.3	29.3
3.5 SHS	9.06	37.6	39.3	61.4	29.7	29.5	40.5
75 x 75 x 6.0 SHS	12.0	22.8	24.9	41.6	19.1	18.7	27.0
5.0 SHS	10.3	27.0	29.1	48.5	22.2	21.8	31.5
4.0 SHS	8.49	33.3	35.4	58.9	26.8	26.5	38.3
3.5 SHS	7.53	37.9	39.9	66.4	30.2	29.9	43.2
3.0 SHS	6.60	43.9	45.5	75.8	34.3	34.1	49.2
2.5 SHS	5.56	52.4	53.9	89.8	40.7	40.4	58.4
65 x 65 x 6.0 SHS	10.1	23.1	25.6	45.3	19.6	19.2	29.1
5.0 SHS	8.75	27.3	29.7	52.6	22.7	22.3	33.7
4.0 SHS	7.23	33.6	36.0	63.6	27.4	27.0	40.8
3.0 SHS	5.66	44.1	45.9	81.3	34.8	34.5	52.1
2.5 SHS	4.78	52.6	54.4	96.2	41.1	40.8	61.7
2.0 SHS	3.88	65.3	67.1	119	50.6	50.3	76.1
50 x 50 x 5.0 SHS	6.39	27.9	31.3	62.6	24.0	23.5	39.1
4.0 SHS	5.35	34.2	37.4	74.8	28.6	28.1	46.8
3.0 SHS	4.25	44.7	47.1	94.2	35.7	35.3	58.9
2.5 SHS	3.60	53.1	55.5	111	42.0	41.6	69.4
2.0 SHS	2.93	65.8	68.2	136	51.5	51.1	85.2
1.6 SHS	2.38	81.7	84.0	168	63.4	63.0	105
40 x 40 x 4.0 SHS	4.09	34.9	39.1	88.0	30.0	29.3	53.8
3.0 SHS	3.30	45.3	48.4	109	36.8	36.3	66.6
2.5 SHS	2.82	53.7	56.8	128	43.1	42.6	78.1
2.0 SHS	2.31	66.4	69.4	156	52.5	52.0	95.4
1.6 SHS	1.88	82.3	85.2	192	64.4	63.9	117
35 x 35 x 3.0 SHS	2.83	45.8	49.4	120	37.7	37.1	72.4
2.5 SHS	2.42	54.2	57.7	140	43.9	43.3	84.5
2.0 SHS	1.99	66.8	70.3	171	53.3	52.7	103
1.6 SHS	1.63	82.7	86.1	209	65.1	64.6	126
30 x 30 x 3.0 SHS	2.36	46.5	50.8	136	38.8	38.1	80.5
2.5 SHS	2.03	54.8	59.0	157	45.0	44.3	93.5
2.0 SHS	1.68	67.4	71.5	191	54.3	53.7	113
1.6 SHS	1.38	83.3	87.3	233	66.1	65.5	138
25 x 25 x 3.0 SHS	1.89	47.4	52.9	159	40.6	39.7	92.6
2.5 SHS	1.64	55.7	61.0	183	46.6	45.7	107
2.0 SHS	1.36	68.3	73.3	220	55.8	55.0	128
1.6 SHS	1.12	84.1	89.0	267	67.5	66.7	156
20 x 20 x 1.6 SHS	0.873	85.4	91.7	321	69.8	68.8	183

See page 1-7 for details on cases of fire exposure considered.

TABLE 1.3-6  
**FIRE ENGINEERING DESIGN**  
**EXPOSED SURFACE AREA TO MASS RATIO ( $m^2/t$ )**  
**SQUARE HOLLOW SECTIONS: GRADE C450**  
**about x- and y-axis**

- 1 = TOTAL PERIMETER, PROFILE-PROTECTED
- 2 = TOTAL PERIMETER, BOX-PROTECTED, NO GAP
- 3 = TOTAL PERIMETER, BOX-PROTECTED, 25mm GAP
- 4 = TOP FLANGE EXCLUDED, PROFILE-PROTECTED
- 5 = TOP FLANGE EXCLUDED, BOX-PROTECTED, NO GAP
- 6 = TOP FLANGE EXCLUDED, BOX-PROTECTED, 25mm GAP



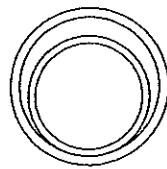
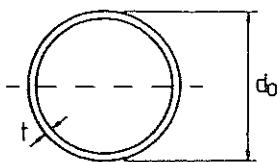
Designation d b t mm mm mm	Mass per m kg/m	1	2	3	4	5	6
100 x 100 x 3.8 SHS	11.1	34.6	36.1	54.1	27.3	27.1	36.1
3.3 SHS	9.73	39.7	41.1	61.7	31.1	30.8	41.1
2.8 SHS	8.39	46.5	47.7	71.5	36.0	35.8	47.7
2.3 SHS	6.95	56.4	57.6	86.3	43.4	43.2	57.6
75 x 75 x 3.3 SHS	7.14	40.1	42.0	70.1	31.9	31.5	45.5
2.8 SHS	6.19	46.9	48.5	80.8	36.6	36.4	52.5
2.3 SHS	5.14	56.8	58.3	97.2	44.0	43.7	63.2
65 x 65 x 2.8 SHS	5.31	47.2	49.0	86.6	37.0	36.7	55.6
2.3 SHS	4.42	57.0	58.8	104	44.4	44.1	66.7
50 x 50 x 2.8 SHS	3.99	47.7	50.1	100	38.0	37.6	62.6
2.3 SHS	3.34	57.6	59.9	120	45.3	44.9	74.9
40 x 40 x 2.8 SHS	3.11	48.3	51.4	116	39.1	38.6	70.7
2.3 SHS	2.62	58.1	61.2	138	46.4	45.9	84.1
35 x 35 x 2.8 SHS	2.67	48.8	52.4	127	39.9	39.3	76.7
2.3 SHS	2.25	58.6	62.1	151	47.1	46.6	90.9
30 x 30 x 2.8 SHS	2.23	49.4	53.7	143	41.0	40.3	85.1
2.3 SHS	1.89	59.2	63.4	169	48.2	47.5	100
25 x 25 x 2.3 SHS	1.53	60.1	65.3	196	49.8	48.9	114

See page 1-7 for details on cases of fire exposure considered.

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## TELESCOPING INFORMATION CIRCULAR HOLLOW SECTIONS

Female (Outer) <i>d<sub>o</sub></i> mm	Male (Inner) <i>d<sub>o</sub></i> mm	Clearance mm
165.1 x 5.4 CHS	139.7	11.6
5.0 CHS	139.7	12.4
3.5 CHS	139.7	15.4
3.0 CHS	139.7	16.4
139.7 x 5.4 CHS	114.3	12.1
5.0 CHS	114.3	12.9
3.5 CHS	114.3	15.9
3.0 CHS	114.3	16.9
114.3 x 6.0 CHS	88.9	11.4
5.4 CHS	88.9	12.6
4.8 CHS	101.6	0.9
4.5 CHS	101.6	1.5
3.6 CHS	101.6	3.3
3.2 CHS	101.6	4.1
101.6 x 5.0 CHS	88.9	0.8
4.0 CHS	88.9	2.8
3.2 CHS	88.9	4.4
2.6 CHS	88.9	5.6
88.9 x 5.9 CHS	60.3	15.3
5.5 CHS	76.1	0.2
5.0 CHS	76.1	1.2
4.8 CHS	76.1	1.6
4.0 CHS	76.1	3.2
3.2 CHS	76.1	4.8
76.1 x 5.9 CHS	60.3	2.6
4.5 CHS	60.3	5.4
3.6 CHS	60.3	7.2
3.2 CHS	60.3	8.0
2.6 CHS	60.3	9.2
2.3 CHS	60.3	9.8
60.3 x 5.4 CHS	48.3	0.2
4.5 CHS	48.3	2.0
3.6 CHS	48.3	3.8
2.9 CHS	48.3	5.2
2.3 CHS	48.3	6.4

Female (Outer) <i>d<sub>o</sub></i> mm	Male (Inner) <i>d<sub>o</sub></i> mm	Clearance mm
48.3 x 5.4 CHS	33.7	2.6
4.0 CHS	33.7	5.4
3.2 CHS	33.7	7.0
2.9 CHS	33.7	7.6
2.3 CHS	42.4	0.1
42.4 x 4.9 CHS	26.9	4.5
4.0 CHS	26.9	6.3
3.2 CHS	33.7	1.1
2.6 CHS	33.7	2.3
2.0 CHS	33.7	3.5
33.7 x 4.5 CHS	21.3	2.2
4.0 CHS	21.3	3.2
3.2 CHS	21.3	4.8
2.6 CHS	26.9	0.4
2.0 CHS	26.9	1.6
26.9 x 4.0 CHS	17.2	0.5
3.2 CHS	17.2	2.1
2.6 CHS	17.2	3.3
2.3 CHS	17.2	3.9
2.0 CHS	21.3	0.4
21.3 x 3.6 CHS	—	—
3.2 CHS	13.5	0.2
2.6 CHS	13.5	1.4
2.0 CHS	13.5	2.6
17.2 x 2.9 CHS	NO SECTION AVAILABLE	
2.3 CHS	NO SECTION AVAILABLE	
13.5 x 2.9 CHS	NO SECTION AVAILABLE	
2.3 CHS	NO SECTION AVAILABLE	

Note: Clearance = (AS 1163 Min *d<sub>o</sub>* - 2*t*) - (AS 1163 Max *d<sub>o</sub>*)

CHS is not a precision tube and all dimensions shown in this chart, although in accordance with the specifications, may vary marginally. Internal weld bead may need to be considered when a close fit is required.

#### SIZES WITH A CLEARANCE LESS THAN 2.0 mm ARE SHOWN BOLD IN THE CHARTS.

For tight fits it is recommended that some form of testing is carried out prior to committing material. Where telescoping over some length is required, additional allowance may be needed for straightness.

#### HOW TO USE THIS CHART

1. Select the size of Female (or Outer) member closest to your requirements from the left hand column.
2. Depending on the application select the clearance required between the two members. Members may need to slide freely inside each other, or be locked with a pin, spot welded or fixed with wedges. This means, in some cases, a 'slippery' fit may be suitable, while for others the tightest fit possible may be more appropriate.
3. Having selected the most suitable clearance for your application, take the appropriate size of the Male (Inner) section from the centre column, e.g.

Female Section (Outer)	Male Section (Inner)	Clearance
		mm
76.1 x 5.9	60.3	2.6

Note that the clearance is the total available difference between member dimensions, not the gap on both sides.

4. Where two telescoping sections are being used, thickness should be similar and will be determined by normal structural requirements. If a third section is to be used consideration of both clearance and thickness within the size list available may be required.
5. Pipe may need to be fixed against twisting by welding or bolting.
6. Press Fit: for short pieces with no need for separation or sliding an interference fit can be achieved using the available ductility of the steel.

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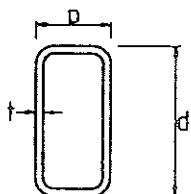
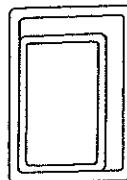


TABLE 1.4-2

## TELESCOPING INFORMATION RECTANGULAR HOLLOW SECTIONS



Female (Outer) d    b    t	Nominal Clearance		Male (Inner) d    b	
	Top	Side		
mm   mm   mm	mm	mm	mm	mm
250 x 150 x 9.0 RHS	40.0	40.0	200	100
6.0 RHS	38.0	38.0	200	100
5.0 RHS	32.0	32.0	200	100
200 x 100 x 9.0 RHS	42.0	42.0	150	50
6.0 RHS	40.0	40.0	150	50
5.0 RHS	38.0	38.0	150	50
4.0 RHS	32.0	32.0	150	50
150 x 100 x 6.0 RHS	17.0	17.0	125	75
5.0 RHS	15.0	15.0	125	75
4.0 RHS	13.0	13.0	125	75
125 x 75 x 6.0 RHS	13.0	13.0	100	50
5.0 RHS	15.0	15.0	100	50
4.0 RHS	17.0	17.0	100	50
3.8 RHS	17.4	17.4	100	50
3.3 RHS	18.4	18.4	100	50
3.0 RHS	19.0	19.0	100	50
2.8 RHS	19.4	19.4	100	50
2.3 RHS	20.4	20.4	100	50
100 x 50 x 6.0 RHS	23.0	3.0	65	35
5.0 RHS	25.0	5.0	65	35
4.0 RHS	27.0	7.0	65	35
3.5 RHS	28.0	8.0	65	35
3.3 RHS	28.4	8.4	65	35
3.0 RHS	29.0	9.0	65	35
2.8 RHS	29.4	9.4	65	35
2.5 RHS	30.0	10.0	65	35
2.3 RHS	30.4	10.4	65	35
2.0 RHS	31.0	11.0	65	35
100 x 50 x 6.0 RHS	13.0	13.0	75	25
5.0 RHS	15.0	15.0	75	25
4.0 RHS	17.0	17.0	75	25
3.5 RHS	18.0	18.0	75	25
3.3 RHS	18.4	18.4	75	25
3.0 RHS	19.0	19.0	75	25
2.8 RHS	19.4	19.4	75	25
2.5 RHS	20.0	20.0	75	25
2.3 RHS	20.4	20.4	75	25
2.0 RHS	21.0	21.0	75	25

Female (Outer) d    b    t	Nominal Clearance		Male (Inner) d    b	
	Top	Side		
mm   mm   mm	mm	mm	mm	mm
75 x 50 x 6.0 RHS	13.0	13.0	50	25
5.0 RHS	0.0	5.0	65	35
4.0 RHS	2.0	7.0	65	35
3.0 RHS	4.0	9.0	65	35
2.8 RHS	4.4	9.4	65	35
2.5 RHS	5.0	10.0	65	35
2.3 RHS	5.4	10.4	65	35
2.0 RHS	6.0	11.0	65	35
75 x 25 x 2.5 RHS	20.0	0.0	50	20
2.0 RHS	21.0	1.0	50	20
1.6 RHS	21.8	1.8	50	20
65 x 35 x 3.0 RHS	9.0	4.0	50	25
2.8 RHS	9.4	4.4	50	25
2.5 RHS	10.0	5.0	50	25
2.3 RHS	10.4	5.4	50	25
2.0 RHS	11.0	6.0	50	25
50 x 25 x 3.0 RHS	NO SECTION AVAILABLE			
2.8 RHS				
2.5 RHS				
2.3 RHS				
2.0 RHS				
1.6 RHS				
50 x 20 x 3.0 RHS	NO SECTION AVAILABLE			
2.8 RHS				
2.5 RHS				
2.3 RHS				
2.0 RHS				
1.6 RHS				

Note: RHS is not a precision tube and all dimensions shown in this chart, although in accordance with the specifications, may vary marginally. Varying corner radii and the internal weld bead may need to be considered when a closer fit is required.  
**SIZES WITH A CLEARANCE LESS THAN 2.0 mm ARE SHOWN BOLDER IN THE CHARTS.**

For tight fits it is recommended that some form of testing is carried out prior to committing material.

Where telescoping over some length is required, additional allowance may be needed for straightness.

### HOW TO USE THIS CHART

1. Select the size of Female (or Outer) member closest to your requirements from the left hand column.
2. Depending on the application select the clearance required between the two members.  
Members may need to slide freely inside each other, or be locked with a pin, spot welded or fixed with wedges.  
This means, in some cases, a 'sloppy' fit may be suitable, while for others the tightest fit possible may be more appropriate.
3. Having selected the most suitable clearance for your application, take the appropriate size of the Male (Inner) section from the right hand column, eg.

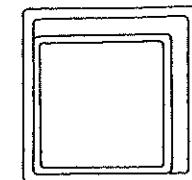
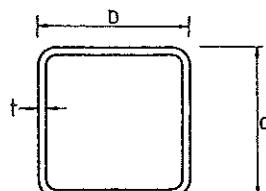
Female Section (Outer)	Clearance	Male Section (Inner)
75 x 50 x 3.0	mm 4.0 x 9.0	65 x 35

Note that the clearance is the total available difference between member dimensions, not the gap on both sides.

4. Where two telescoping sections are being used, thickness should be similar and will be determined by normal structural requirements.  
If a third section is to be used consideration of both clearance and thickness within the size list available may be required.
5. RHS has the obvious advantage that its shape prevents rotation of the section.
6. Press Fit: for short pieces with no need for separation or sliding an interference fit can be achieved using the available ductility of the steel. Sizes where clearance is shown as 0.0 may occasionally require press fit.

TABLE 1.4-3

## TELESCOPING INFORMATION SQUARE HOLLOW SECTIONS



Female (Outer) d    b    t			Nominal Clearance Top      Side		Male (Inner) d      b	
mm	mm	mm	mm	mm	mm	mm
250	250	9.0 SHS	32.0	32.0	200	200
		6.0 SHS	38.0	38.0	200	200
200	200	9.0 SHS	32.0	32.0	150	150
		6.0 SHS	38.0	38.0	150	150
		5.0 SHS	40.0	40.0	150	150
150	150	9.0 SHS	7.0	7.0	125	125
		6.0 SHS	13.0	13.0	125	125
		5.0 SHS	15.0	15.0	125	125
125	125	9.0 SHS	7.0	7.0	100	100
		6.0 SHS	13.0	13.0	100	100
		5.0 SHS	15.0	15.0	100	100
		4.0 SHS	17.0	17.0	100	100
100	100	9.0 SHS	7.0	7.0	75	75
		6.0 SHS	13.0	13.0	75	75
		5.0 SHS	1.1	1.1	89	89
		4.0 SHS	3.1	3.1	89	89
		3.8 SHS	3.5	3.5	89	89
		3.3 SHS	4.5	4.5	89	89
		3.0 SHS	5.1	5.1	89	89
		2.8 SHS	5.5	5.5	89	89
		2.3 SHS	6.5	6.5	89	89
89	89	6.0 SHS	1.9	1.9	75	75
		5.0 SHS	3.9	3.9	75	75
		3.5 SHS	6.9	6.9	75	75
75	75	6.0 SHS	13.0	13.0	50	50
		5.0 SHS	0.0	0.0	65	65
		4.0 SHS	2.0	2.0	65	65
		3.3 SHS	3.4	3.4	65	65
		3.0 SHS	4.0	4.0	65	65
		2.8 SHS	4.4	4.4	65	65
		2.5 SHS	5.0	5.0	65	65
		2.3 SHS	5.4	5.4	65	65
		3.5 SHS	3.0	3.0	65	65
65	65	6.0 SHS	3.0	3.0	50	50
		5.0 SHS	5.0	5.0	50	50
		4.0 SHS	7.0	7.0	50	50
		3.0 SHS	9.0	9.0	50	50
		2.8 SHS	9.4	9.4	50	50
		2.5 SHS	10.0	10.0	50	50
		2.3 SHS	10.4	10.4	50	50
		2.0 SHS	11.0	11.0	50	50

Female (Outer) d    b    t			Nominal Clearance Top      Side		Male (inner) d      b	
mm	mm	mm	mm	mm	mm	mm
50	50	5.0 SHS	0.0	0.0	40	40
		4.0 SHS	2.0	2.0	40	40
		3.0 SHS	4.0	4.0	40	40
		2.8 SHS	4.4	4.4	40	40
		2.5 SHS	5.0	5.0	40	40
		2.3 SHS	5.4	5.4	40	40
		2.0 SHS	6.0	6.0	40	40
		1.6 SHS	6.8	6.8	40	40
40	40	4.0 SHS	2.0	2.0	30	30
		3.0 SHS	4.0	4.0	30	30
		2.8 SHS	4.4	4.4	30	30
		2.5 SHS	0.0	0.0	35	35
		2.3 SHS	0.4	0.4	35	35
		2.0 SHS	1.0	1.0	35	35
		1.6 SHS	1.8	1.8	35	35
35	35	3.0 SHS	4.0	4.0	25	25
		2.8 SHS	4.4	4.4	25	25
		2.5 SHS	0.0	0.0	30	30
		2.3 SHS	0.4	0.4	30	30
		2.0 SHS	1.0	1.0	30	30
		1.6 SHS	1.8	1.8	30	30
30	30	3.0 SHS	4.0	4.0	20	20
		2.8 SHS	4.4	4.4	20	20
		2.5 SHS	0.0	0.0	25	25
		2.3 SHS	0.4	0.4	25	25
		2.0 SHS	1.0	1.0	25	25
		1.6 SHS	1.8	1.8	25	25
25	25	3.0 SHS	4.0	4.0	15	15
		2.5 SHS	0.0	0.0	20	20
		2.3 SHS	0.4	0.4	20	20
		2.0 SHS	1.0	1.0	20	20
		1.6 SHS	1.8	1.8	20	20
20	20	1.6 SHS	1.8	1.8	15	15
15	15	1.8 SHS	NO SECTION AVAILABLE			
13	13	1.8 SHS	NO SECTION AVAILABLE			

Note: SHS is not a precision tube and all dimensions shown in this chart, although in accordance with the specifications, may vary marginally. Varying corner radii and the internal weld bead may need to be considered when a closer fit is required.

SIZES WITH A CLEARANCE LESS THAN 2.0 mm ARE SHOWN BOLDER IN THE CHARTS.

For tight fits it is recommended that some form of testing is carried out prior to committing material.

Where telescoping over some length is required, additional allowance may be needed for straightness.

### HOW TO USE THIS CHART

- Select the size of Female (or Outer) member closest to your requirements from the left hand column.
- Depending on the application select the clearance required between the two members. Members may need to slide freely inside each other, or be locked with a pin, spot welded or fixed with wedges. This means, in some cases, a 'sloppy' fit may be suitable, while for others the tightest fit possible may be more appropriate.
- Having selected the most suitable clearance for your application, take the appropriate size of the Male (inner) section from the right hand column, eg.

Female Section (Outer)	Clearance	Male Section (Inner)
75 x 75 x 3.0	mm 4.0 x 4.0	65 x 65

Note that the clearance is the total available difference between member dimensions, not the gap on both sides.

- Where two telescoping sections are being used, thickness should be similar and will be determined by normal structural requirements. If a third section is to be used consideration of both clearance and thickness within the size list available may be required.
- SHS has the obvious advantage that its shape prevents rotation of the section.
- Press Fit: for short pieces with no need for separation or sliding an interference fit can be achieved using the available ductility of the steel. Sizes where clearance is shown as 0 may occasionally require press fit.

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## PART 2 DETERMINATION OF DESIGN ACTION EFFECTS

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2.4.1    First-Order Elastic Analysis .....	2-3
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## PART 2 DETERMINATION OF DESIGN ACTION EFFECTS

### 2.1 Methods of Analysis

The methods of structural analysis that are recognised in AS 4100 and most likely to be used for structural hollow sections are:

- a) First-order elastic analysis
- b) First-order elastic analysis with moment amplification (Clause 4.4.2 of AS 4100)
- c) Second-order elastic analysis in accordance with Appendix E of AS 4100

Plastic analysis is currently not permitted by AS 4100 for structural steel hollow sections, although research<sup>[1], [2]</sup> has already been performed to establish the suitability of square and rectangular hollow sections.

### 2.2 Second-Order Effects

When combined bending and axial compression forces are present in members, SECOND-ORDER EFFECTS must be considered. Second-order bending moments are often classified as  $P\Delta$  which arise from the relative end displacements ( $\Delta$ ), or as  $P\delta$  which arise from the member deflecting ( $\delta$ ) from a straight line joining the member's ends (Figure 2.2). In braced frames the relative member end displacements ( $\Delta$ ) are small, and consideration is only given to the  $P\delta$  effects. In sway frames the  $P\Delta$  effects are often more significant than the  $P\delta$  effects.

### 2.3 Use of Tables

The tabulated values in PARTS 4, 5, and 6 may be used for design in those cases where second-order effects:

- can be neglected
- are accounted for using moment amplification factors in conjunction with a first-order elastic analysis
- are accounted for in a second-order elastic analysis

[1] Hasan, S.W., Hancock, G.J., "Plastic Bending Tests of Cold-Formed Rectangular Hollow Sections", *Steel Construction*, AISC, Vol. 23, No.4, 1989.

[2] Zhao, X.L., Hancock, G.J., "Tests to Determine Plate Slenderness Limits for Cold-Formed Rectangular Hollow Sections of Grade C4501", *Steel Construction*, AISC, Vol. 25, No. 4 1990.

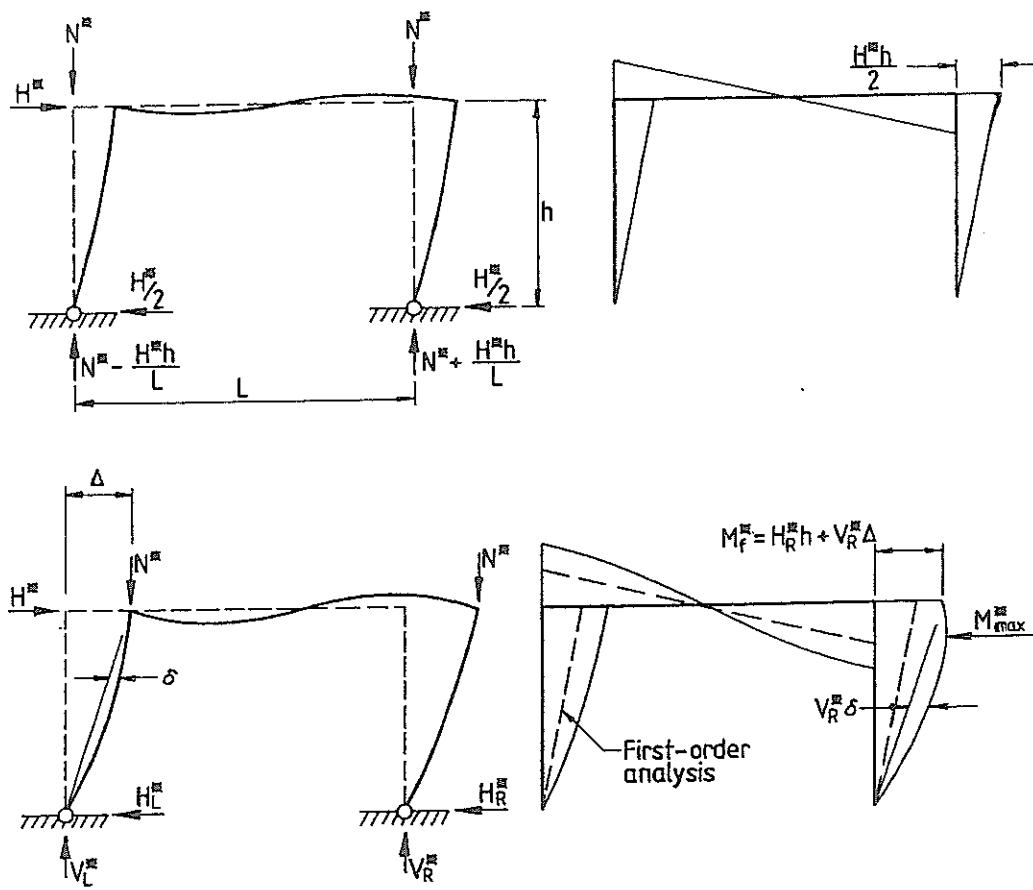


Figure 2.2: First-order Analysis and Second-order Behaviour

## 2.4 Use of Analysis Methods

### 2.4.1 First-order Elastic Analysis

This method can be used to analyse members which do not have second order effects. They are members with:

- bending moments only
- axial tension force only
- combined bending moments and tension force

## 2.4.2 First-order Elastic Analysis with Moment Amplification

This method can be used to analyse members with combined bending and axial compression and when the moment amplification factors  $\delta_b$  or  $\delta_s$  (see Clauses 4.4.2.2 and 4.4.2.3 of AS 4100) are less than 1.4 (i.e. when second-order effects are less than 40 %). The maximum moment in the member  $M_m^*$ , as determined by the first-order elastic analysis, is multiplied by the moment amplification factor  $\delta_b$  or  $\delta_s$ . See PART 7, Figures 7.3(1) and 7.3(2) for the determination of  $\delta_b$  and  $\delta_s$  respectively.

## 2.4.3 Second-order Elastic Analysis in Accordance with Appendix E of AS 4100

This method can be used to analyse members with combined bending and axial compression and must be used when the moment amplification factors  $\delta_b$  or  $\delta_s$  are greater than 1.4 (i.e. when second-order effects are greater than 40 %).

A suitable computer analysis program is normally used due to the iterative nature of this analysis.

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# SECTION CAPACITIES

## PART 3 | SECTION CAPACITIES

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## TABLES

### TABLES 3-1 TO 3-6

Design Section Capacities ( $\phi V$ ,  $\phi N$ ,  $\phi M$ ,  $\phi V_{pl}$ ,  $\phi U$ )

3-5

NOTE: SEE PAGE 10 FOR THE SPECIFIC MATERIAL  
STANDARD REFERRED TO BY THE SECTION TYPE  
AND STEEL GRADE IN THESE TABLES

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## PART 3 SECTION CAPACITIES

### 3.1 Scope

The following tables give values of design section capacities for axial tension ( $\phi N$ ), axial compression ( $\phi N_c$ ), moment ( $\phi M_c$ ), shear ( $\phi V_c$ ), and torsional moment ( $\phi M_z$ ).

### 3.2 Method

The determination of each of the design section capacities (Tables 3.1-1 to 3.1-6) is detailed in Sections 3.2.1 to 3.2.5. PART 3 of the Tables contains design section capacities whilst PART 4 to PART 6 contain design member capacities. Section capacities give the maximum capacity of a section subjected to design action effects. Member capacities are determined by reducing the section capacities by factors accounting for restraints and loading conditions.

#### 3.2.1 Design Section Capacity for Axial-Tension

The design section capacity for axial-tension ( $\phi N$ ) is determined from Clauses 7.1 and 7.2 of AS 4100 as the lesser of:

$$\phi N = 0.9 \sigma_y A_g$$

$$\text{and } \phi N = 100.85 k_A A_n$$

where  $\phi = 0.9$  (Table 3.4 of AS 4100)

$A_g =$  gross cross-sectional area

$\sigma_y =$  yield stress used in design

$k_A = 1.0$

$A_n =$  net section area

$A_n = A_g - A_t$  (gross cross-sectional area (assuming full perimeter welded connections

with no penetrations or holes))

$\sigma_y =$  tensile strength used in design

#### 3.2.2 Design Section Capacity for Axial-Compression

The design section capacity for axial-compression ( $\phi N_c$ ) is determined from Clauses 6.1 and 6.2 of AS 4100 as:

$$\phi N_c = 0.9 \sigma_y A_g$$

where  $\phi = 0.9$  (Table 3.4 of AS 4100)

$A_g = A_g / A_n$  (see Section 12.3.3 and Tables 12-1 to 12-6)

$A_n =$  net section area

$A_n = A_g$

$A_g =$  gross cross-sectional area (assuming no penetrations or holes)

$\sigma_y =$  yield stress used in design

### 3.2.3 Design Moment Section Capacity

The design moment section capacity ( $\phi M_s$ ) is determined from Clauses 5.1 and 5.2.1 of AS 4100 as:

$$\phi M_s = \phi f_y Z_e$$

where  $\phi = 0.9$  (Table 3.4 of AS 4100)

$f_y$  = yield stress used in design

$Z_e$  = effective section modulus (see Section 1.2.3.2 and Tables 1.2-1 to 1.2-6).

Values of the design section moment capacity ( $\phi M_s$ ) can be found in the tables for members bent about either principal x- or y-axis. It should be noted that the design member capacity in the minor principal y-axis is the design section capacity ( $\phi M_{sy}$ ). For members which are fully restrained against flexural buckling the design member moment capacity equals the full section moment capacity ( $\phi M_s$ ).

### 3.2.4 Design Shear Capacity of a Web

The design shear capacity of a web ( $\phi V_v$ ) is determined from Clauses 5.11.3 and 5.11.4 of AS 4100, for RHS and SHS and as the *lesser* of

$$\phi V_v = 0.6 f_y (d - 2t) 2t \quad (\text{Clause 5.11.4 of AS 4100})$$

and

$$\phi V_v = \frac{2\phi V_u}{0.9 + \left( \frac{f_{vm}^*}{f_{va}^*} \right)} \quad (\text{Clause 5.11.3 of AS 4100})$$

and for CHS

$$\phi V_v = 0.36 f_y A_g \quad (\text{Clause 5.11.4 of AS 4100})$$

where  $\phi = 0.9$  (Table 3.4 of AS 4100)

$f_y$  = yield stress used in design

$A_g$  = net section area

$A_g$  = gross cross-sectional area (assuming there are no holes larger than those required for fasteners, or that the net area is greater than 0.9 times the gross area)

$d$  = full depth of section

$t$  = thickness of section

$V_u = 0.6 f_y (d - 2t) 2t$

$f_{va}^*$  = average design shear stress in the web

$f_{vm}^*$  = maximum design shear stress in the web

The ratio of maximum to average design shear stress in the web ( $f_{vm}^* / f_{va}^*$ ) for bending about the x-axis, is calculated<sup>[1]</sup> using:

$$\frac{f_{vm}^*}{f_{va}^*} = \frac{3(2b + d)}{2(3b + d)}$$

where  $d$  = full depth of section

$b$  = full width of section

Note: for bending about the y-axis  $b$  and  $d$  are interchanged in the calculation of the maximum to average design web shear stress ratio. Non-uniform shear stress governs when  $d/b > 0.75$

For calculating the web area, the web depth has been taken as the clear depth between flanges ( $d_1$ ) for RHS and SHS, and as 0.6 times the gross cross-sectional area ( $0.6 A_g$ ) for CHS.

[1] Bridge, R.Q., Trahair, N.S., "Thin Wall Beams", *Steel Construction*, AISC, Vol. 15, No. 1, 1981.

## 2.5 Design Torsional Moment Section Capacity

The design torsional moment section capacity ( $\phi M_z$ ) is determined in accordance with Sections 3.2.5.1 and 3.2.5.2.

### 3.2.5.1 Introduction

Although AS 4100 makes no provision for the design of members subject to torsion it is nevertheless considered appropriate to supply torsional capabilities for hollow sections in the tables. Hollow sections perform particularly well in torsion and their behaviour under torsional loading is readily analysed by simple procedures. An explanation of torsional effects is provided in the references listed in Section 3.2.5.2.

The general theory of torsion established by Saint-Venant is based on uniform torsion. The theory assumes that all cross-sections rotate as a body around the centre of torsion.

When the torsional moment that is applied is non-uniform, such as when the torsional load is applied midspan between rigid supports or when the free warping of the sections is restricted, then the torsional load is shared between uniform and non-uniform torsion or warping. However in the case of hollow sections the contribution of non-uniform torsion is negligible and sections can be treated as subject to uniform torsion without any significant loss of precision.

### 3.2.5.2 Method

For hollow sections, torsional actions can be considered using the following formulae:

$$\begin{aligned} M_z^* &\leq \phi M_z \\ \phi M_z &= \phi \cdot 0.6 f_y C \end{aligned}$$

where  $M_z^*$  = design torsional moment

$\phi$  = 0.9 (Table 3.4 of AS 4100)

$\phi M_z$  = design torsional moment section capacity

$f_y$  = yield stress used in design

$C$  = torsional section modulus

Note: The angle of twist per unit length  $\theta$  (radians) can be determined using the following formulae:

$$\theta = \frac{M_z^*}{GJ}$$

where  $M_z^*$  = design torsional moment for serviceability limit state

$G$  =  $80 \times 10^3$  MPa

$J$  = torsional section constant

The method for determining the torsion section constants  $C$  and  $J$  is detailed in Section 1.2.1.1.

Suggested references for design for torsion:

- [1] "AS 4100 Supplement 1-1990: Steel Structures Commentary (Supplement to AS 4100-1990)", Standards Australia, Section C8.5.
- [2] Trahair, N.S., Bradford, M.A., "The Behaviour and Design of Steel Structures", 2nd ed., Chapman and Hall, London, 1988.

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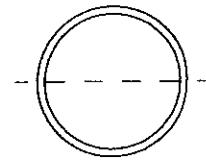
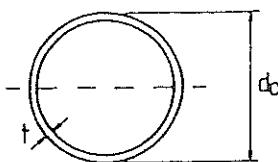


TABLE 3.1-1

**DESIGN SECTION CAPACITIES  
CIRCULAR HOLLOW SECTIONS: GRADE C250  
about any axis**

Designation		Mass per m	Axial Tension	Axial Compression	Moment	Shear	Torsion
$d_o$	$t$		$\phi N_t$	$\phi N_s$	$\phi M_s$	$\phi V_v$	$\phi M_z$
mm	mm	kg/m	kN	kN	kNm	kN	kNm
610.0 x 12.7 CHS	187	5360	5360	1020	1930	941	
9.5 CHS	141	4030	4030	735	1450	715	
6.4 CHS	95.3	2730	2530	449	983	489	
508.0 x 12.7 CHS	155	4450	4450	701	1600	645	
9.5 CHS	117	3350	3350	525	1210	491	
6.4 CHS	79.2	2270	2270	328	817	337	
165.1 x 5.4 CHS	21.3	610	610	31.0	218	28.3	
5.0 CHS	19.7	566	566	28.8	204	26.4	
139.7 x 5.4 CHS	17.9	513	513	21.9	185	19.9	
5.0 CHS	16.6	476	476	20.4	171	18.6	
114.3 x 5.4 CHS	14.5	416	416	14.4	150	13.0	
4.5 CHS	12.2	349	349	12.2	126	11.1	
101.6 x 5.0 CHS	11.9	341	341	10.5	123	9.43	
4.0 CHS	9.63	276	276	8.58	99.3	7.77	
88.9 x 5.9 CHS	12.1	346	346	9.16	125	8.09	
5.0 CHS	10.3	297	297	7.93	107	7.07	
4.0 CHS	8.38	240	240	6.49	86.4	5.85	
76.1 x 5.9 CHS	10.2	293	293	6.56	105	5.73	
4.5 CHS	7.95	228	228	5.20	82.0	4.62	
3.6 CHS	6.44	184	184	4.26	66.4	3.83	
60.3 x 5.4 CHS	7.31	210	210	3.67	75.4	3.17	
4.5 CHS	6.19	177	177	3.16	63.9	2.77	
3.6 CHS	5.03	144	144	2.61	51.9	2.32	
48.3 x 5.4 CHS	5.71	164	164	2.25	59.0	1.90	
4.0 CHS	4.37	125	125	1.77	45.1	1.54	
3.2 CHS	3.56	102	102	1.47	36.7	1.30	
42.4 x 4.9 CHS	4.53	130	130	1.56	46.8	1.31	
4.0 CHS	3.79	109	109	1.33	39.1	1.15	
3.2 CHS	3.09	88.7	88.7	1.11	31.9	0.970	
33.7 x 4.5 CHS	3.24	92.9	92.9	0.870	33.4	0.722	
4.0 CHS	2.93	84.0	84.0	0.799	30.2	0.671	
3.2 CHS	2.41	69.0	69.0	0.672	24.8	0.578	
26.9 x 4.0 CHS	2.26	64.7	64.7	0.477	23.3	0.390	
3.2 CHS	1.87	53.6	53.6	0.407	19.3	0.342	
2.6 CHS	1.56	44.7	44.7	0.347	16.1	0.297	
21.3 x 3.6 CHS	1.57	45.0	45.0	0.257	16.2	0.207	
3.2 CHS	1.43	40.9	40.9	0.238	14.7	0.195	
2.6 CHS	1.20	34.4	34.4	0.206	12.4	0.173	

Notes: 1.  $\phi = 0.9$  (Table 3.4 of AS 4100)  
 2.  $N_t = A_g f_y$  (Clause 7.2 of AS 4100)  
 3.  $N_s = k_t A_g f_y$  (Clause 6.2.1 of AS 4100)  
 4.  $M_s = f_y Z_e$  (Clause 5.2.1 of AS 4100)  
 5.  $V_v = 0.36 f_y A_g$  (Clause 5.11.4 of AS 4100)  
 6.  $M_z = 0.6 f_y C$  (Section 3.2.5 of the Tables)

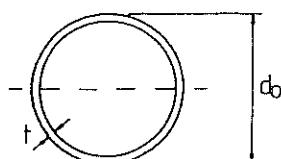
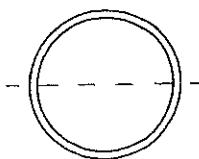


TABLE 3.1-2



**DESIGN SECTION CAPACITIES  
CIRCULAR HOLLOW SECTIONS: GRADE C350  
about any axis**

Designation <i>d<sub>o</sub></i> mm	Mass per m kg/m	Axial Tension $\phi N_t$	Axial Compression $\phi N_g$	Moment $\phi M_s$	Shear $\phi V_v$	Torsion $\phi M_z$
		kN	kN	kNm	kN	kNm
457.0 x 12.7 CHS	139	5580	5580	789	2010	724
	9.5 CHS	105	4210	4210	565	1510
	6.4 CHS	71.1	2850	2580	343	1030
406.4 x 12.7 CHS	123	4950	4950	620	1780	567
	9.5 CHS	93.0	3730	3730	456	1340
	6.4 CHS	63.1	2530	2430	282	912
355.6 x 12.7 CHS	107	4310	4310	471	1550	428
	9.5 CHS	81.1	3250	3250	356	1170
	6.4 CHS	55.1	2210	2210	224	796
323.9 x 12.7 CHS	97.5	3910	3910	388	1410	351
	9.5 CHS	73.7	2960	2960	296	1060
	6.4 CHS	50.1	2010	2010	189	724
273.1 x 9.3 CHS	60.5	2430	2430	204	874	186
	6.4 CHS	42.1	1690	1690	139	608
	4.8 CHS	31.8	1270	1270	98.3	459
219.1 x 8.2 CHS	42.6	1710	1710	115	616	104
	6.4 CHS	33.6	1350	1350	91.2	485
	4.8 CHS	25.4	1020	1020	66.3	366
168.3 x 7.1 CHS	28.2	1130	1130	58.2	408	52.6
	6.4 CHS	25.6	1030	1030	52.9	369
	4.8 CHS	19.4	777	777	40.4	280
165.1 x 3.5 CHS	13.9	560	560	27.3	201	26.6
	3.0 CHS	12.0	481	481	22.6	173
139.7 x 3.5 CHS	11.8	472	472	20.1	170	18.8
	3.0 CHS	10.1	406	406	16.8	146
114.3 x 6.0 CHS	16.0	643	643	22.2	231	19.9
	4.8 CHS	13.0	520	520	18.1	187
	3.6 CHS	9.83	394	394	13.9	142
	3.2 CHS	8.77	352	352	12.4	127
101.6 x 3.2 CHS	7.77	312	312	9.76	112	8.92
	2.6 CHS	6.35	255	255	7.90	91.7
88.9 x 5.5 CHS	11.3	454	454	12.1	163	10.7
	4.8 CHS	9.96	399	399	10.7	144
	3.2 CHS	6.76	271	271	7.41	97.7
	2.6 CHS	5.53	222	222	6.10	79.9
76.1 x 3.2 CHS	5.75	231	231	5.36	83.1	4.85
	2.3 CHS	4.19	168	168	3.95	60.5
60.3 x 2.9 CHS	4.11	165	165	3.01	59.3	2.71
	2.3 CHS	3.29	132	132	2.44	47.5
48.3 x 2.9 CHS	3.25	130	130	1.89	46.9	1.67
	2.3 CHS	2.61	105	105	1.53	37.7
42.4 x 2.6 CHS	2.55	102	102	1.30	36.9	1.15
	2.0 CHS	1.99	80.0	80.0	1.03	28.8
33.7 x 2.6 CHS	1.99	80.0	80.0	0.794	28.8	0.694
	2.0 CHS	1.56	62.7	62.7	0.634	22.6
26.9 x 2.3 CHS	1.40	56.0	56.0	0.440	20.2	0.381
	2.0 CHS	1.23	49.3	49.3	0.391	17.7
21.3 x 2.0 CHS	0.952	38.2	38.2	0.236	13.8	0.203

- Notes: 1.  $\phi = 0.9$  (Table 3.4 of AS 4100)  
 2.  $N_t = A_g f_y$  (Clause 7.2 of AS 4100)  
 3.  $N_g = k_f A_g f_y$  (Clause 6.2.1 of AS 4100)  
 4.  $M_s = f_y Z$  (Clause 5.2.1 of AS 4100)  
 5.  $V = 0.384 A$  (Clause 5.1.4 of AS 4100)  
 6.  $M_z = 0.6 M_s$  (Section 3.2.5 of the Tables)

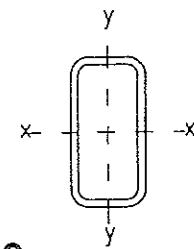
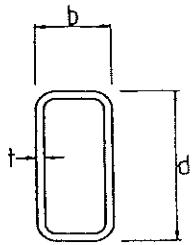


TABLE 3.1-3

**DESIGN SECTION CAPACITIES  
RECTANGULAR HOLLOW SECTIONS: GRADE C350  
about x- and y-axis**

Designation <i>d b t</i> mm mm mm	Mass per m kg/m	Axial	Axial	Moment		Shear	Torsion	
		Tension $\phi N_t$ kN	Compression $\phi N_s$ kN	x-axis $\phi M_{sx}$ kNm	y-axis $\phi M_{sy}$ kNm	x-axis $\phi V_{vx}$ kN	y-axis $\phi V_{vy}$ kN	$\phi M_z$ kNm
250 x 150 x 9.0 RHS	51.8	2080	2080	168	117	759	449	105
	6.0 RHS	35.6	1430	1290	118	65.6	519	313
	5.0 RHS	29.9	1200	978	94.4	49.3	436	265
200 x 100 x 9.0 RHS	37.7	1510	1510	92.4	56.8	590	279	51.4
	6.0 RHS	26.2	1050	1050	66.2	37.3	406	200
	5.0 RHS	22.1	886	820	56.5	28.4	342	170
	4.0 RHS	17.9	719	575	46.2	19.9	276	139
150 x 100 x 6.0 RHS	21.4	861	861	42.3	32.0	303	200	27.7
	5.0 RHS	18.2	729	729	36.3	26.3	256	170
	4.0 RHS	14.8	593	575	29.8	19.2	208	139
150 x 50 x 5.0 RHS	14.2	571	571	24.8	10.7	246	75.6	10.7
	4.0 RHS	11.6	467	449	20.6	7.9	200	63.5
	3.0 RHS	8.96	359	279	16.2	5.03	152	49.9
125 x 75 x 6.0 RHS	16.7	672	672	26.5	18.6	247	143	16.3
	5.0 RHS	14.2	571	571	22.9	16.1	209	123
	4.0 RHS	11.6	467	467	19.0	12.6	170	101
	3.0 RHS	8.96	359	326	14.9	8.30	130	78.2
100 x 50 x 6.0 RHS	12.0	483	483	14.3	8.74	190	86.2	7.73
	5.0 RHS	10.3	414	414	12.5	7.69	162	75.6
	4.0 RHS	8.49	341	341	10.5	6.48	132	63.5
	3.5 RHS	7.53	302	302	9.43	5.71	117	56.9
	3.0 RHS	6.60	265	265	8.40	4.73	102	49.9
	2.5 RHS	5.56	223	207	7.14	3.59	85.5	42.5
	2.0 RHS	4.50	181	145	5.83	2.51	69.1	34.8
75 x 50 x 6.0 RHS	9.67	388	388	8.84	6.66	138	86.2	5.32
	5.0 RHS	8.35	335	335	7.84	5.92	119	75.6
	4.0 RHS	6.92	278	278	6.66	5.03	98.0	63.5
	3.0 RHS	5.42	218	218	5.38	4.07	75.7	49.9
	2.5 RHS	4.58	184	184	4.60	3.34	64.0	42.5
	2.0 RHS	3.72	149	145	3.77	2.43	51.9	34.8
75 x 25 x 2.5 RHS	3.60	145	145	3.17	1.36	61.5	18.9	1.35
	2.0 RHS	2.93	118	113	2.62	1.00	49.9	15.9
	1.6 RHS	2.38	95.5	77.6	2.15	0.699	40.4	13.2
65 x 35 x 3.0 RHS	4.25	170	170	3.46	2.24	64.0	32.9	1.97
	2.5 RHS	3.60	145	145	2.98	1.93	54.2	28.3
	2.0 RHS	2.93	118	118	2.46	1.48	44.1	23.4
50 x 25 x 3.0 RHS	3.07	123	123	1.85	1.12	47.5	21.5	0.979
	2.5 RHS	2.62	105	105	1.61	0.982	40.5	18.9
	2.0 RHS	2.15	86.2	86.2	1.34	0.824	33.1	15.9
	1.6 RHS	1.75	70.3	70.3	1.11	0.644	27.0	13.2
50 x 20 x 3.0 RHS	2.83	114	114	1.62	0.827	46.9	15.9	0.733
	2.5 RHS	2.42	97.3	97.3	1.42	0.729	40.0	14.2
	2.0 RHS	1.99	79.9	79.9	1.19	0.616	32.7	12.1
	1.6 RHS	1.63	65.3	65.3	0.989	0.484	26.6	10.2

- Notes:
1.  $\phi = 0.9$  (Table 3.4 of AS 4100)
  2.  $N_t = A_g f_y$  (Clause 7.2 of AS 4100)
  3.  $N_s = k_f A_g f_y$  (Clause 6.2.1 of AS 4100)
  4.  $M_s = f_y Z_e$  (Clause 5.2.1 of AS 4100)
  5.  $V_v$  (See Section 3.2.4 of the Tables)
  6.  $M_z = 0.6 f_y C$  (Section 3.2.5 of the Tables)

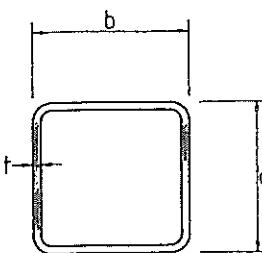
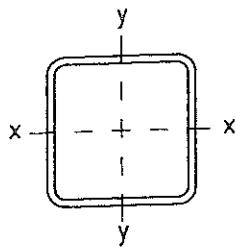


TABLE 3.1-4  
DESIGN SECTION CAPACITIES  
SQUARE HOLLOW SECTIONS: GRADE C350  
about x- and y-axis



Designation		Mass per m	Axial Tension	Axial Compression	Moment	Shear	Torsion
d	b		$\phi N_t$	$\phi N_c$	$\phi M_a$	$\phi V_v$	$\phi M_z$
mm	mm	kg/m	kN	kN	kNm	kN	kNm
250 x 250 x 9.0 SHS	6.0 SHS	65.9	2650	2650	234	780	184
	4.0 SHS	45.0	1810	1540	129	533	129
200 x 200 x 9.0 SHS	6.0 SHS	51.8	2080	2080	147	612	113
	5.0 SHS	35.6	1430	1430	92.5	421	80.3
	4.0 SHS	29.9	1200	1070	70.1	355	68.3
150 x 150 x 9.0 SHS	6.0 SHS	37.7	1510	1510	78.2	444	59.7
	5.0 SHS	26.2	1050	1050	55.9	309	43.3
	4.0 SHS	22.1	886	886	45.5	261	37.1
125 x 125 x 9.0 SHS	6.0 SHS	30.6	1230	1230	52.0	360	39.3
	5.0 SHS	21.4	861	861	37.8	253	29.1
	4.0 SHS	18.2	729	729	32.4	215	25.1
	3.0 SHS	14.8	593	593	24.8	175	20.7
100 x 100 x 9.0 SHS	6.0 SHS	23.5	944	944	31.0	276	23.2
	5.0 SHS	16.7	672	672	23.2	197	17.7
	4.0 SHS	14.2	571	571	20.0	168	15.4
	3.0 SHS	11.6	467	467	16.6	137	12.8
	2.0 SHS	8.96	359	359	11.7	105	10.1
89 x 89 x 6.0 SHS	5.0 SHS	14.6	588	588	17.8	172	13.5
	3.5 SHS	12.5	502	502	15.5	147	11.8
	2.0 SHS	9.06	363	363	11.5	107	8.91
75 x 75 x 6.0 SHS	5.0 SHS	12.0	483	483	12.1	141	9.11
	4.0 SHS	10.3	414	414	10.6	121	8.06
	3.5 SHS	8.49	341	341	8.90	100	6.83
	3.0 SHS	7.53	302	302	7.97	88.9	6.15
	2.5 SHS	6.60	265	265	7.08	77.3	5.43
	2.0 SHS	5.56	223	223	5.75	65.3	4.65
65 x 65 x 6.0 SHS	5.0 SHS	10.1	407	407	8.67	119	6.46
	4.0 SHS	8.75	351	351	7.66	103	5.78
	3.0 SHS	7.23	290	290	6.49	85.1	4.95
	2.5 SHS	5.66	227	227	5.22	66.1	3.98
	2.0 SHS	4.78	192	192	4.45	56.0	3.42
	1.6 SHS	3.88	156	156	3.33	45.5	2.82
50 x 50 x 5.0 SHS	4.0 SHS	6.39	256	256	4.14	74.7	3.07
	3.0 SHS	5.35	215	215	3.59	62.7	2.70
	2.5 SHS	4.25	170	170	2.96	49.3	2.22
	2.0 SHS	3.60	145	145	2.54	42.0	1.93
	1.6 SHS	2.93	118	118	2.10	34.3	1.61
	1.2 SHS	2.38	95.5	95.5	1.61	28.0	1.33
40 x 40 x 4.0 SHS	3.0 SHS	4.09	164	164	2.12	47.8	1.57
	2.5 SHS	3.30	133	133	1.80	38.1	1.34
	2.0 SHS	2.82	113	113	1.56	32.7	1.17
	1.6 SHS	2.31	92.5	92.5	1.30	26.9	0.989
	1.2 SHS	1.88	75.3	75.3	1.07	22.0	0.824
35 x 35 x 3.0 SHS	2.5 SHS	2.83	114	114	1.33	32.5	0.978
	2.0 SHS	2.42	97.3	97.3	1.16	28.0	0.866
	1.6 SHS	1.99	79.9	79.9	0.975	23.1	0.735
	1.2 SHS	1.63	65.3	65.3	0.808	19.0	0.616
30 x 30 x 3.0 SHS	2.5 SHS	2.36	94.8	94.8	0.932	26.9	0.676
	2.0 SHS	2.03	81.6	81.6	0.822	23.3	0.605
	1.6 SHS	1.68	67.3	67.3	0.695	19.4	0.519
	1.2 SHS	1.38	55.2	55.2	0.580	16.0	0.439
25 x 25 x 3.0 SHS	2.5 SHS	1.89	75.9	75.9	0.603	21.3	0.430
	2.0 SHS	1.64	65.8	65.8	0.539	18.7	0.391
	1.6 SHS	1.36	54.7	54.7	0.462	15.7	0.341
	1.2 SHS	1.12	45.1	45.1	0.389	13.0	0.292
20 x 20 x 1.6 SHS		0.873	35.0	35.0	0.236	10.0	0.175

Notes: 1.  $\phi = 0.9$  (Table 3.4 of AS 4100)  
 2.  $N_t = A_g f_y$  (Clause 7.2 of AS 4100)  
 3.  $N_c = k_1 A_g f_y$  (Clause 6.2.1 of AS 4100)  
 4.  $M_a = f_y Z_g$  (Clause 5.1.1 of AS 4100)  
 5.  $V_v$  (See Section 3.2.4 of the Tables)  
 6.  $M_z = 0.6 f_y C$  (Section 3.2.5 of the Tables)

TABLE 3.1-5

**DESIGN SECTION CAPACITIES  
RECTANGULAR HOLLOW SECTIONS: GRADE C450  
about x- and y-axis**

Designation  d b t	Mass per m  mm mm mm	kg/m	Axial Tension	Axial Compression	Moment		Shear		Torsion
			$\phi N_t$	$\phi N_s$	$\phi M_{sx}$	$\phi M_{sy}$	$\phi V_{vx}$	$\phi V_{vy}$	$\phi M_z$
125 x 75 x 3.8 RHS	11.1	540	559	23.3	14.0	209	124	14.7	
3.3 RHS	9.73	474	448	20.6	11.2	183	110	13.0	
2.8 RHS	8.39	409	351	16.9	8.80	156	94.4	11.3	
2.3 RHS	6.95	339	259	12.3	6.50	129	78.7	9.50	
100 x 50 x 3.3 RHS	7.14	348	368	11.5	6.39	143	69.6	6.54	
2.8 RHS	6.19	302	295	10.2	5.05	122	60.4	5.74	
2.3 RHS	5.14	251	215	8.52	3.70	102	50.7	4.86	
75 x 50 x 2.8 RHS	5.09	248	263	6.52	4.71	91.4	60.4	4.23	
2.3 RHS	4.24	207	215	5.49	3.56	76.2	50.7	3.59	
65 x 35 x 2.8 RHS	3.99	194	206	4.21	2.72	77.3	40.0	2.41	
2.3 RHS	3.34	163	172	3.57	2.16	64.6	34.0	2.07	
50 x 25 x 2.8 RHS	2.89	141	149	2.26	1.37	57.5	26.4	1.21	
2.3 RHS	2.44	119	126	1.94	1.18	48.3	22.8	1.05	
50 x 20 x 2.8 RHS	2.67	130	138	1.99	1.02	56.8	19.6	0.907	
2.3 RHS	2.25	110	116	1.71	0.882	47.7	17.2	0.803	

Notes: 1.  $\phi = 0.9$  (Table 3.4 of AS 4100)  
 2.  $N_t = A_g f_y$  (Clause 7.2 of AS 4100)  
 3.  $N_s = k_f A_g f_y$  (Clause 6.2.1 of AS 4100)  
 4.  $M_s = f_y Z_e$  (Clause 5.2.1 of AS 4100)  
 5.  $V_v$  (See Section 3.2.4 of the Tables)  
 6.  $M_z = 0.6 f_y C$  (Section 3.2.5 of the Tables)

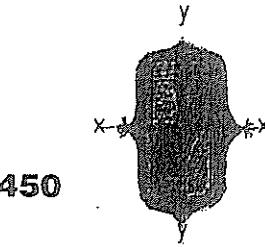
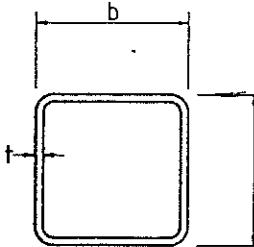


TABLE 3.1-6

**DESIGN SECTION CAPACITIES  
SQUARE HOLLOW SECTIONS: GRADE C450  
about x- and y-axis**

Designation  d b t	Mass per m  mm mm mm	kg/m	Axial Tension	Axial Compression	Moment	Shear	Torsion
			$\phi N_t$	$\phi N_s$	$\phi M_z$	$\phi V_v$	$\phi M_z$
100 x 100 x 3.8 SHS	11.1	540	572	19.6	169	15.8	
3.3 SHS	9.73	474	502	15.9	148	14.0	
2.8 SHS	8.39	409	383	12.5	127	12.2	
2.3 SHS	6.95	339	259	9.35	105	10.2	
75 x 75 x 3.3 SHS	7.14	348	368	9.75	108	7.54	
2.8 SHS	6.19	302	319	8.16	93.3	6.59	
2.3 SHS	5.14	251	259	6.07	77.7	5.56	
65 x 65 x 2.8 SHS	5.31	259	274	6.32	79.8	4.83	
2.3 SHS	4.42	215	228	4.91	66.7	4.10	
50 x 50 x 2.8 SHS	3.99	194	206	3.59	59.7	2.71	
2.3 SHS	3.34	163	172	3.05	50.1	2.32	
40 x 40 x 2.8 SHS	3.11	152	161	2.20	46.2	1.64	
2.3 SHS	2.62	127	135	1.88	39.1	1.42	
35 x 35 x 2.8 SHS	2.67	130	138	1.63	39.5	1.20	
2.3 SHS	2.25	110	116	1.40	33.6	1.05	
30 x 30 x 2.8 SHS	2.23	109	115	1.14	32.8	0.835	
2.3 SHS	1.89	92.3	97.7	0.994	28.0	0.736	
25 x 25 x 2.3 SHS	1.53	74.7	79.1	0.656	22.5	0.479	

Notes: Refer to TABLE 3.1-5

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## PART 4 | MEMBERS SUBJECT TO BENDING

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NOTE: SEE PAGE (4) FOR THE SPECIFIC MATERIAL STANDARD REFERRED TO BY THE SECTION TYPE AND STEEL GRADE IN THESE TABLES.

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## PART 4 | MEMBERS SUBJECT TO BENDING

### 4.1 Design Moment Capacity for Members Without Full Lateral Restraint

#### 4.1.1 Scope

These tables for RHS bending about the x-axis without full lateral restraint have been prepared in accordance with Section 5 of AS 4100. Values of design moment capacity ( $\phi M_u$ ) are given for various values of effective length ( $L_e$ ).

CHS and SHS are not included in these tables as they are not susceptible to lateral buckling. The design member moment capacity ( $\phi M_u$ ) always equals the design section moment capacity ( $\phi M_u$ ) as given in Tables B4-1 to B4-2 for CHS and Tables B4-5 to B4-6 for SHS, except for the extreme case when the load acts at above the shear centre (Clause C5.6.1.4 for the Commentary of AS 4100).

#### 4.1.2 Method

The values of design moment capacity ( $\phi M_u$ ) are determined in accordance with Clause 5.6.1.4 of AS 4100 as:

$$\phi M_u = \phi M_u^0 \times \psi$$

Where  $\phi M_u^0 = 0.9$  (Table 3.4 of AS 4100)

$\psi = 1.0$  (Table 5.6.4 of AS 4100 corresponding to the case of uniform moment over the effective length ( $L_e$ )))

for  $L_e \leq \text{FLR}$  ( $\text{FLR}$  = maximum segment length for full lateral restraint as determined in Section 4.1.3)  
 $\psi = 1.0$

or  $L_e > \text{FLR}$

$$\psi = 1.05 \left[ \left( \frac{L_e}{\text{FLR}} \right)^{0.5} + 1 \right] - \left( \frac{M_u}{M_u^0} \right) \quad (\text{Equation 5.6.1.1(2) of AS 4100})$$

$M_u$  = moment of resistance (moment of resistance at member subjected to bending)  $\approx M_u$  (reference buckling moment) (Clause 5.6.1.1(3)(v)(A) of AS 4100)

$$\Rightarrow \psi = \left[ \left( \frac{L_e}{\text{FLR}} \right)^{0.5} + 1 \right] - \left( \frac{M_u}{M_u^0} \right) \quad (\text{Equation 5.6.1.1(3) of AS 4100})$$

$F_c = 280 \times 10^3 \text{ N/mm}^2$

$L_e$  = selected segment of structural link between principal y-axes (Tables 1.2-1 to 1.2-6)

$\sigma_c = 210 \times 10^3 \text{ N/mm}^2$

$\sigma_{cr}$  = theoretical buckling stress (see Section 1.2-1 and Tables 1.2-1 to 1.2-6)

$L$  = effective length (see Section 4.1.3)

$M_u^0$  =  $\phi M_u^0$  (see Section 4.1.2 and Tables 4.1-1 to 4.1-6)

$M_u$  = yield moment (see Section 4.1.2)

$\psi$  = effective segment ratio (see Section 4.1.3 and Tables 4.1-1 to 4.1-6)

### 4.1.3 Segment Length for Full Restraint

For SHS and RHS a segment without full or partial restraints at both ends may be considered to have full lateral restraint provided its length, as determined in accordance with Clause 5.3.2.4 of AS 4100, satisfies:

$$L \leq (1800 + 1500\beta_m) \left( \frac{b_f}{b_w} \right) \left( \frac{250}{f_y} \right) r_y$$

where  $L$  = maximum segment length for full lateral restraint

$\beta_m = -1.0$  (Clause 5.3.2.4 of AS 4100)

$b_f = b$

= width of section

$b_w = d$

= depth of section

$r_y$  = radius of gyration about minor principal y-axis

The FLR values listed in Tables 4.1–1 to 4.1–4 are calculated using  $\beta_m = -1.0$ . However  $\beta_m = -0.8$  may be used for segments with transverse loads, or  $\beta_m$  may be taken as the ratio of the smaller to larger end moments in the length ( $L$ ) for segments without transverse loads (positive when the segment is bent in reverse curvature and negative when bent in single curvature).

### 4.1.4 Effective Length

Before using these tables it is necessary to determine the effective length ( $L_e$ ), which depends on the restraint against twisting and lateral rotation and the load height.  $L_e$  is determined in accordance with Clause 5.6.3 of AS 4100 and given by:

$$L_e = k_t k_l k_r L$$

where  $k_t$  = twist restraint factor (Table 4.1.3(1))

$k_l$  = load height factor (Table 4.1.3(2))

$k_r$  = lateral rotation restraint factor (Table 4.1.3(3))

$L$  = length of segment

Table 4.1.4(1): Twist Restraint Factors ( $k_t$ )

Restraint Arrangement	Factor ( $k_t$ )
FF, FL, LL, FU	1.0
FP, PL, PU	$1 + \left[ \left( \frac{d_w}{L} \right) \left( \frac{t_f}{2t_w} \right)^3 \right]$
PP	$1 + \left[ 2 \left( \frac{d_w}{L} \right) \left( \frac{t_f}{2t_w} \right)^3 \right]$

Table 4.1.4(2): Load Height Factors ( $k_l$ ) for Gravity Loads

Longitudinal load position	Restraint arrangement	Load height position	
		Shear	Top Flange
Within segment	FF, FP, FL, PP, PL, LL FU, PU	1.0 1.0	1.4 2.0
At segment end	FF, FP, FL, PP, PL, LL FU, PU	1.0 1.0	1.0 2.0

Table 4.1.4(3): Lateral Rotation Restraint Factors ( $k_r$ )

Restraint arrangement	Ends with lateral rotation restraints	Factor ( $k_r$ )
FU, PU	Any	1.0
FF, FP, FL, PP, PL, LL	None	1.0
FF, FP, FL, PP, PL, LL	One	0.85
FF, FP, FL, PP, PL, LL	Both	0.70

where  $d_w$  = depth of web

$t_f$  = thickness of critical flange

$t_w$  = thickness of web

F = fully restrained

L = laterally restrained

P = partially restrained

U = unrestrained

and two of the symbols F, L, P, U are used to indicate the conditions at the ends of the segment. Restraint requirements are detailed in Clause 5.4.3 of AS 4100.

## 4.1.5 Other Loading and Restraint Conditions

The design moment capacities presented in these tables can be used for other restraints and loading conditions. For these situations the effective length ( $L_e$ ) corresponding to the relevant conditions must be assessed and the relevant value of  $\alpha_m$  determined in accordance with Clause 5.6.1.1(a) of AS 4100. The design moment capacity ( $\phi M_b$ ) can then be determined as the lesser of:

and

$$\begin{aligned}\phi M_{sx} &= \phi Z_e f_y \\ \phi M_{bx} &= \phi \alpha_m \alpha_s Z_e f_y\end{aligned}$$

where  $\phi = 0.9$  (Table 3.4 of AS 4100)

$\phi M_{sx}$  = the design section moment capacity (Tables 3.1–1 to 3.1–6)

$\phi M_{bx}$  =  $\alpha_m$  times the value of ( $\phi M_b = \phi \alpha_s Z_e f_y$ ) given in Tables 4.1–1 to 4.1–4

$\alpha_m$  = moment modification factor

$$\leq \frac{1}{\alpha_s}$$

$\alpha_s$  = slenderness reduction factor (see Section 4.1.2)

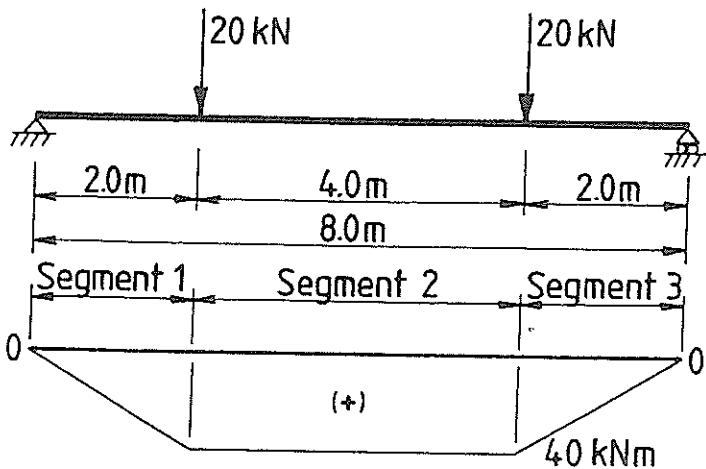
It should be noted that:

-  $\alpha_m \leq 1.0$  for CHS and SHS as they are not susceptible to lateral buckling and  $\alpha_s = 1.0$

- generally  $\alpha_m \leq 1.0$  for RHS, as these sections (with the exception of 150 x 50 and 75 x 25) are only susceptible to lateral buckling at larger spans (i.e.  $a_s > 10$ )

#### 4.1.6 Examples

- A simply supported beam shown below has two concentrated loads applied in such a way that full restraint is provided at the location of the loads. The calculated design load at each point is 20 kN. What size beam is required to support these loads?



Bending Moment Diagram

Design Data:

Design bending moment  $M^* = 40 \text{ kNm}$

Solution:

For beam segment 2:

End restraint condition = FF (fully restrained at both ends of the segment)

Twist restraint factor  $k_t = 1.0$  (Table 4.1.3(1))

Load Height Factor  $k_l = 1.0$  (Table 4.1.3(2))  
(For loading at segment end and top flange loading)

Lateral rotation restraint factor  $k_r = 0.7$  (Table 4.1.3(3))

∴ Effective length

$$\begin{aligned} L_e &= k_t k_l k_r L \\ &= 1.0 \times 1.0 \times 0.7 \times 4.0 \\ &= 2.8 \text{ m} \end{aligned}$$

## Q Seismic Isolation

A rectangular hollow section is the most efficient and most practical hollow section for this application. As a uniform bending moment is applied to segment 2,  $\alpha_m = 1.0$  (Table 5.6.1 of AS 4100). Thus alternatives can be read directly from Table 4.1–1(1) for a uniform bending moment of 40 kNm on segment 2 with an effective length ( $L_e = 2.8$  m). They are:

150 x 100 x 6.0 RHS Grade – C350 (21.4 kg/m)  $\phi M_b = 42.3 \text{ kNm} > M^*$

200 x 100 x 4.0 RHS Grade – C350 (17.9 kg/m)  $\phi M_b = 46.2 \text{ kNm} > M^*$

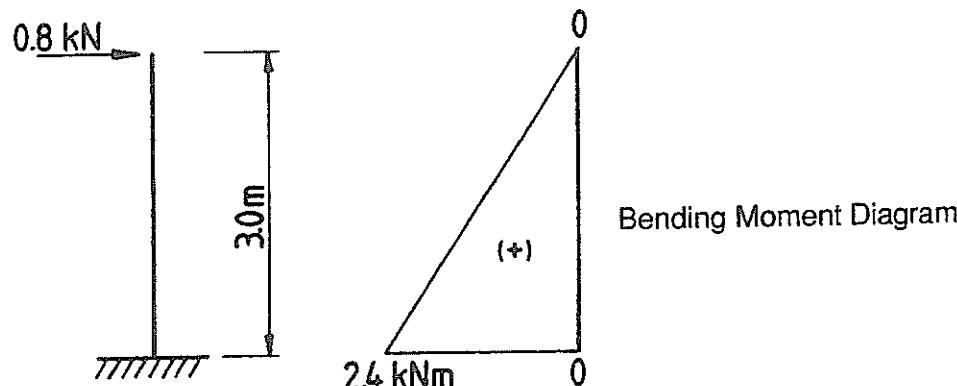
Based on mass select 200 x 100 x 4.0 RHS – Grade C350 (17.9 kg/m). The extra depth will provide increased stiffness which may be important.

Additional design checks which should be performed are:

- Additional design bending moment due to self-weight
- Shear (Section 4.2)
- Interaction of Shear and Bending (Section 4.2.3)
- Bearing (Section 4.3)
- Bearing and Bending Interaction (Section 4.4)
- Deflection (Section 4.5)

Beam segments 1 and 3 do not have to be checked because they have the same design bending moment and end restraints with a shorter effective length.

2. A free standing sign post which is securely concreted into the ground is required to resist a calculated horizontal design force of 0.8 kN at a height of 3 m. What size CHS is required?



### Design Data:

Design bending moment  $M^* = 2.4 \text{ kNm}$

### Solution:

The appropriate size of CHS may be selected from the **section capacity tables** in PART 3.

The alternatives are:

60.3 x 3.6 CHS – Grade C250 (5.03 kg/m)  $\phi M_s = 2.61 \text{ kNm} > M^*$  (Table 3.1–1)  
60.3 x 2.3 CHS – Grade C350 (3.29 kg/m)  $\phi M_s = 2.44 \text{ kNm} > M^*$  (Table 3.1–2)

Based on mass select 60.3 x 2.3 CHS (3.29 kg/m) in Grade C350 steel.

Additional design checks which should be performed are:

- Shear (Section 4.2)
- Interaction of Shear and Bending (Section 4.2.3)
- Deflection (Section 4.5) if it is critical for this type of application

## 4.2 Design Shear Capacity

### 4.2.1 Scope

The section capacity tables (Tables 3.1–1 to 3.1–6) include values of design web shear capacity for bending about the x– and y-axis.

### 4.2.2 Method

The design shear capacity of a web ( $\phi V_v$ ) is determined from the *lesser* of Clauses 5.11.3 and 5.11.4 of AS 4100 for RHS and SHS, and Clause 5.11.4 of AS 4100 for CHS, as outlined in Section 3.2.4 of this publication.

### 4.2.3 Interaction of Shear and Bending

The design web shear capacity determined in Section 4.2.2 of the Tables may be significantly reduced when the section is subject to a large bending moment at the same location. The reduced shear capacity ( $\phi V_{vm}$ ) is determined in accordance with Clause 5.12.3 of AS 4100 as:

or

$$\begin{aligned}\phi V_{vm} &= \phi V_v && \text{for } M^* \leq 0.75 \phi M_b \\ &= V_v \left[ 2.2 - \left( \frac{1.6 M^*}{\phi M_b} \right) \right] && \text{for } 0.75 \phi M_b \leq M^* \leq \phi M_b\end{aligned}$$

where  $\phi V_v$  = design web shear capacity (Section 4.2.2)

$M^*$  = design moment capacity

$\phi M_b$  = design member moment capacity (Section 4.1.2)

Note: If  $V^* < 0.6 \phi V_v$  or if  $M^* < 0.75 \phi M_b$  then no check on the interaction of shear and bending is necessary.

### 4.2.4 Examples

- Check the shear capacity of the 200 x 100 x 4.0 RHS – Grade C350 beam in Example 1 from Section 4.1.6.

#### Design Data:

Design shear force  $V^* = 20 \text{ kN}$

#### Solution:

Design shear capacity of the section

$$\begin{aligned}\phi V_{vx} &= 276 \text{ kN} \\ &> V^*\end{aligned}$$

Therefore the 200 x 100 x 4.0 RHS – Grade C350 is satisfactory.

2. Check the shear capacity of the 60.3 x 2.3 CHS – Grade C350 beam in Example 2 from Section 4.1.6.

Design Data:

Design shear force  $V^* = 0.8 \text{ kN}$

Solution:

Design shear capacity of the section

$$\begin{aligned} \phi V_v &= 47.5 \text{ kN} \\ &> V^* \end{aligned} \quad (\text{Table 3.1-2})$$

Therefore the 60.3 x 2.3 CHS – Grade C350 is satisfactory.

3. Check the 200 x 100 x 4.0 RHS – Grade C350 beam in Example 1 from Section 4.1.6 for the interaction of shear and bending.

Design Data:

Design shear force	$V^* = 20 \text{ kN}$	
Design shear capacity	$\phi V_{vx} = 276 \text{ kN}$	(Table 3.1-2)
Design bending moment	$M^* = 40 \text{ kNm}$	
Design member moment capacity	$\phi M_{bx} = \phi M_{sx}$ $= 46.2 \text{ kNm}$	(full lateral restraint, $\alpha_s = 1.0$ ) (Table 3.1-3)

Solution:

$$\begin{aligned} 0.6 \phi V_{vx} &= 0.6 \times 276 \\ &= 166 \text{ kN} \\ &> V^* \end{aligned}$$

Therefore the 200 x 100 4.0 RHS – Grade C350 is satisfactory.

## 4.3 Design Web Bearing Capacity

### 4.3.1 Scope

The tables give values of design web bearing yield capacity per unit length ( $\phi R_{by}/b_b$ ) and the design web bearing buckling capacity per unit length ( $\phi R_{bb}/b_b$ ) for SHS and RHS. The tables also include useful parameters used in web bearing design ( $5r_{ext}$ ,  $b_{bw}$ ,  $L_e/r$ ).

### 4.3.2 Method

The design web bearing capacity ( $\phi R_b$ ) has been determined from Clause 5.13 of AS 4100, with modifications derived from research<sup>[1], [2]</sup>, and taken as the *lesser* of:

$$\phi R_{bb} = \phi 2 \alpha_c b_b t f_y$$

and

$$\phi R_{by} = \phi 2 \alpha_p b_b t f_y$$

where  $\phi = 0.9$  (Table 3.4 of AS 4100)

$\phi R_{bb}$  = design web bearing buckling capacity (Clause 5.13.4 of AS 4100)

$b_b$  = bearing width (see Figure 4.3.2(2))

$t$  = thickness of section

$f_y$  = yield stress used in design

$\alpha_c$  = member slenderness reduction factor determined in accordance with Section 5.3 of these Tables using  $k_f = 1.0$ ,  $\alpha_b = 0.5$ , and a modified slenderness ratio ( $L_e/r$ ) derived from research<sup>[1], [2]</sup>.

$$\frac{L_e}{r} = \frac{\sqrt{12}(d - 2r_{ext})}{t}$$

$d$  = overall depth of section

$r_{ext}$  = external corner radii (See Section 1.2.1.2)

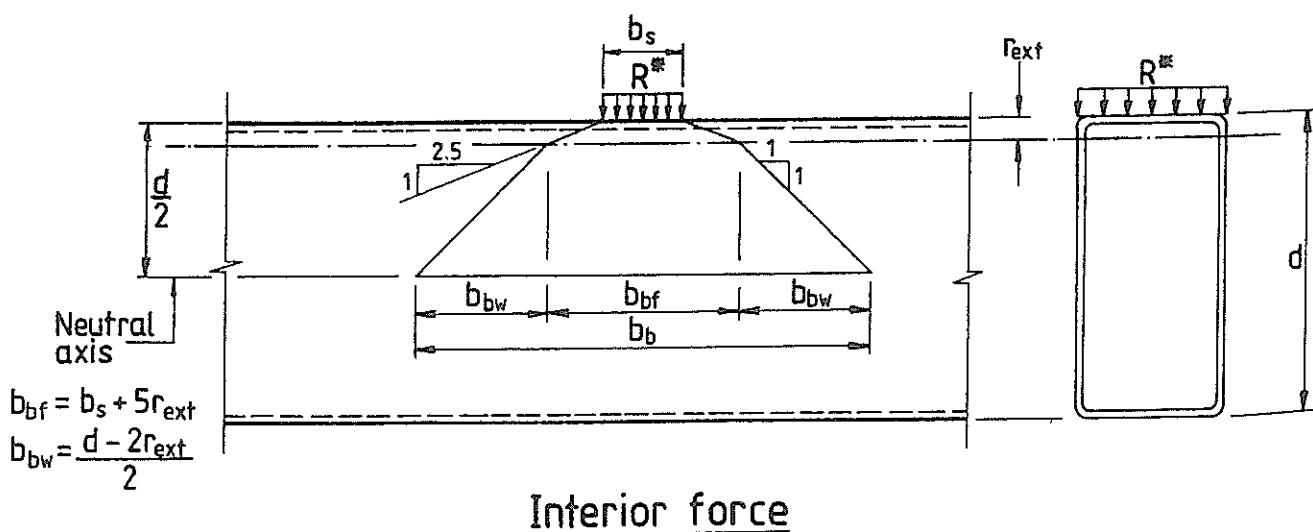


Figure 4.3.2:  $b_{bf}$ ,  $b_b$  for Dispersion of Force Through Flange and Web

and  $\phi R_{by}$  = design web bearing yield capacity (Clause 5.13.3 of AS 4100 and proposed design procedures<sup>[1], [2]</sup>)

$\alpha_p$  = load per unit length of mechanism.  $\alpha_p$  is determined using an iterative procedure to calculate the values in Tables 4.3–1 to 4.3–4.

$$= \frac{0.5}{k_r} \left[ 1 + \left( 1 - \alpha_p^2 \right) \left( 1 + \frac{k_r}{k_D} - \left( 1 - \alpha_p^2 \right) \frac{0.25}{k_D^2} \right) \right]$$

$$k_T = 2 \frac{r_{ext}}{t} - 1$$

$$k_D = \frac{d}{t} - 2 \frac{r_{ext}}{t}$$

$\alpha_p$  can also be calculated using an approximation formula, based on one step iteration, and given as:

$$\alpha_p = \frac{0.5}{k_r} \left[ 1 + \left( 1 - \alpha_{pm}^2 \right) \left( 1 + \frac{k_r}{k_D} - \left( 1 - \alpha_{pm}^2 \right) \frac{0.25}{k_D^2} \right) \right]$$

$$\text{where } \alpha_{pm} = \frac{1}{k_r} + \frac{0.5}{k_D}$$

The lesser value of  $\phi R_{by}/b_b$  and  $\phi R_{bb}/b_b$  is highlighted in bold type in the tables.

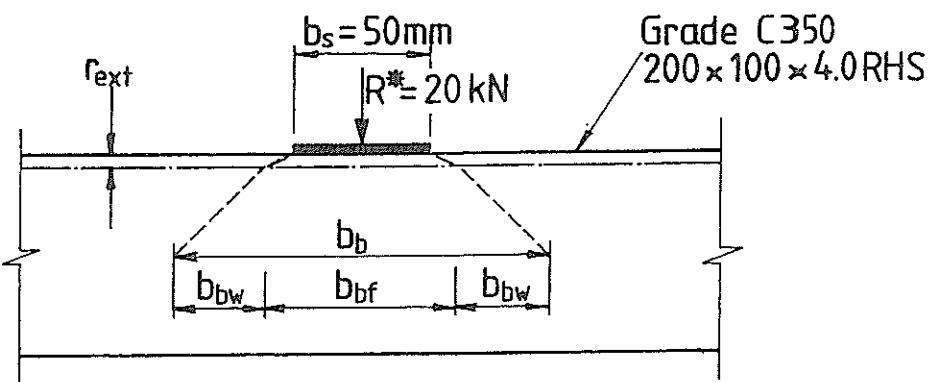
The length terms for the various situations are illustrated in Figure 4.3.2.

*Note that research only considers the development of full dispersion width along the neutral axis and does not consider end bearing conditions. This method if used for end bearing may be non-conservative and further research is required.*

- [1] Zhao, X.L., Hancock, G.J., "Square and Rectangular Hollow Sections Subject to Combined Actions", *Journal of Structural Engineering*, ASCE, Vol 118, No. 3 , pp 648–668, 1992.
- [2] Zhao, X.L., Hancock, G. J., "Design Formulae for Web Crippling of Rectangular Hollow Sections", *Proceedings*, Third Pacific Structural Steel Conference, Tokyo, Japan, 1992.

### 4.3.3 Example

- The design concentrated force of 20 kN on the 200 x 100 x 4.0 RHS – Grade C350 beam in Example 1 of Section 4.1.6 is applied over the full width of the RHS and for a length of 50 mm along the RHS. Check the bearing capacity of the beam.



#### Design Data:

Design bearing force

$$R^* = 20 \text{ kN}$$

Stiff bearing length

$$b_s = 50 \text{ mm}$$

$5r_{\text{ext}} = 50.0 \text{ mm}$  (Table 4.3-1(A))

$b_{\text{bw}} = 90.0 \text{ mm}$  (Table 4.3-1(A))

#### Solution:

Bearing length at the edge of the corner radius

$$\begin{aligned} b_{\text{bf}} &= b_s + 5r_{\text{ext}} \\ &= 50 + 50.0 \\ &= 100 \text{ mm} \end{aligned}$$

Bearing length at the centre of the web

$$\begin{aligned} b_b &= b_{\text{bf}} + 2b_{\text{bw}} \\ &= 100 + (2 \times 90) \\ &= 280 \text{ mm} \end{aligned}$$

The web bearing capacity ( $\phi R_b$ ) is the *lesser* of  $\phi R_{\text{by}}$  and  $\phi R_{\text{bb}}$ .

From Table 4.3-1(A):

Design web yield capacity  $\frac{\phi R_{\text{by}}}{b_b} = 0.636 \text{ kN/mm}$

Design web buckling capacity  $\frac{\phi R_{\text{bb}}}{b_b} = 0.488 \text{ kN/mm}$   
 $\frac{\phi R_{\text{by}}}{b_b} > \frac{\phi R_{\text{bb}}}{b_b}$

Therefore web buckling will govern.

Design web bearing capacity  $\phi R_b = \phi R_{\text{bb}}$   
 $= 0.488 \times 280$   
 $= 137 \text{ kN} \quad (> R^*)$

Therefore the 200 x 100 x 4.0 RHS – Grade C350 is satisfactory.

## 4.4 Bending and Bearing Interaction

### 4.4.1 Method

The design web bearing capacity determined in Section 4.3 of the Tables may be significantly reduced when the section is subject to a large bending moment at the same location. The effect of this interaction of bending and bearing force in hollow sections is not addressed by AS 4100, but suitable interaction equations have been developed from research<sup>[1], [2]</sup>.

The bending and bearing interaction is dependent on the ratio of bearing length to the width of bearing ( $\gamma$ ) and the web slenderness ( $d_1/t$ ). The interaction equation for  $\gamma \geq 1.0$  and  $(d_1/t) \leq 30$  is:

$$1.2 \left( \frac{R^*}{\phi R_b} \right) + \left( \frac{M^*}{\phi M_s} \right) \leq 1.5$$

When  $\gamma < 1.0$  and/or  $(d_1/t) > 30$  the interaction equation is:

$$0.8 \left( \frac{R^*}{\phi R_b} \right) + \left( \frac{M^*}{\phi M_s} \right) \leq 1.0$$

When  $\gamma < 1.0$  and  $(d_1/t) \leq 30$  or when  $\gamma \leq 1.0$  and  $(d_1/t) > 30$ , the interaction equation is:

$$\left( \frac{R^*}{\phi R_b} \right) + \left( \frac{M^*}{\phi M_s} \right) \leq 1.3$$

where  $\gamma = \frac{b_s}{b}$

$b_s$  = stiff bearing length (see Figure 4.3.2)

$b$  = width of section

$(d_1/t)$  = web slenderness

$d_1$  = clear depth between flanges

$t$  = thickness of section

$R^*$  = maximum design bearing force

$\phi$  = 0.9 (Table 3.4 of AS 4100)

$\phi R_b$  = design web bearing capacity

$M^*$  = maximum design bending moment

$\phi M_s$  = design section moment capacity

**Note: these formulae only apply to bearing across the full width of section.**

Design aids have not been produced for this interaction because of the numerous bearing lengths which may occur for each section size.

[1] Zhao, X.L., Hancock, G. J., "Square and Rectangular Hollow Sections Subject to Combined Actions", *Journal of Structural Engineering*, ASCE, Vol 118, No. 3 , pp 648–668, 1992.

[2] Zhao, X.L., Hancock, G. J., "Design Formulae for Web Crippling of Rectangular Hollow Sections", *Proceedings, Third Pacific Structural Steel Conference*, Tokyo, Japan, 1992.

#### 4.4.2 Example

- Considering further Example 1 of Section 4.1.6 and Example 1 of Section 4.3.3 the interaction of bending and bearing is checked for the 200 x 100 x 4.0 RHS – Grade C350 beam.

Design Data:

Design bearing force	$R^* = 20 \text{ kN}$
Design web bearing capacity	$\phi R_b = 137 \text{ kN}$
Design bending moment	$M^* = 40 \text{ kNm}$
Design section moment capacity	$\phi M_s = 46.2 \text{ kNm}$ (Table 3.1–3)
Stiff bearing length	$b_s = 50 \text{ mm}$
Web slenderness	$d/t = 48$

Solution:

$$\begin{aligned}\gamma &= \frac{b_s}{b} \\ &= \frac{50}{100} \\ &= 0.5 \\ &< 1.0\end{aligned}$$

$$\begin{aligned}d/t &= 48 \\ &> 30\end{aligned}$$

∴ the interaction equation is

$$\begin{aligned}0.8 \left( \frac{R^*}{\phi R_b} \right) + \left( \frac{M^*}{\phi M_s} \right) &\leq 1.0 \\ 0.8 \left( \frac{20}{137} \right) + \left( \frac{40}{46.2} \right) &= 0.98 \\ &\leq 1.0\end{aligned}$$

Therefore the 200 x 100 x 4.0 RHS – Grade C350 is satisfactory

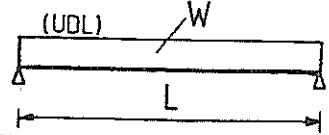
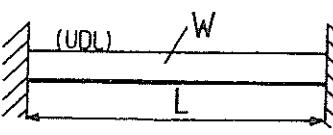
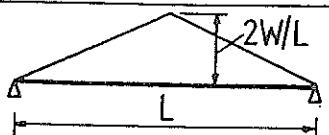
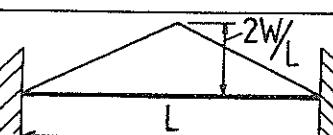
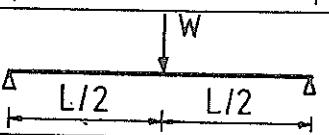
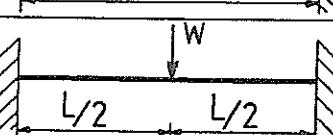
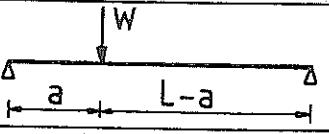
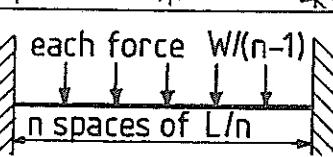
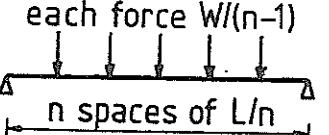
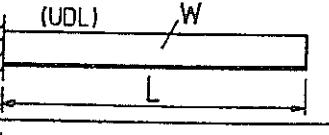
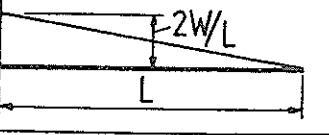
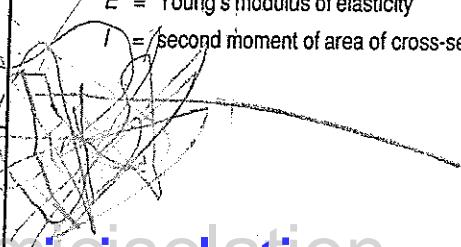
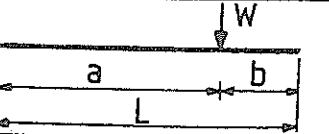
## 4.5 Calculation of Beam Deflections

Common methods for calculating the elastic deflection of a beam include:-

- integration of  $M/EI$  diagram
- moment area
- slope deflection
- published solutions for particular cases
- approximate or numerical methods (eg. finite elements)

Table 4.5 gives the more commonly used beam deflection formulae. Due to the large range of loading configurations and support conditions considered for beams in design, a more comprehensive set of beam deflection formulae are published in the AISC technical journal "Steel Construction", Volume 26 No. 1 (February 1992).

Table 4.5: Beam Deflection Formulae

SIMPLY SUPPORTED BEAMS		BUILT-IN BEAMS	
	$\Delta = \frac{5}{384} \frac{WL^3}{EI}$		$\Delta = \frac{1}{384} \frac{WL^3}{EI}$
	$\Delta = \frac{1}{60} \frac{WL^3}{EI}$		$\Delta = \frac{1.4}{384} \frac{WL^3}{EI}$
	$\Delta = \frac{1}{48} \frac{WL^3}{EI}$		$\Delta = \frac{1}{192} \frac{WL^3}{EI}$
	$\Delta = \frac{WL^3}{48EI} \left[ \frac{3a}{L} - 4 \left( \frac{a}{L} \right)^3 \right]$		$\Delta = \frac{k}{192(n-1)} \frac{WL^3}{EI}$
	$\Delta = \frac{k}{192(n-1)} \frac{WL^3}{EI}$	$n \text{ odd, } k = [n - \frac{1}{n}] \left[ 1 - \frac{1}{2} \left( 1 - \frac{1}{n^2} \right) \right]$ $n \text{ even, } k = [3 - \frac{1}{2} (1 + \frac{4}{n^2})] \times n - [2(n - \frac{1}{n})]$	
$n \text{ odd, } k = [n - \frac{1}{n}] \left[ 3 - \frac{1}{2} \left( 1 - \frac{1}{n^2} \right) \right]$ $n \text{ even, } k = n \left[ 3 - \frac{1}{2} \left( 1 + \frac{4}{n^2} \right) \right]$		$n \text{ odd, } k = [n - \frac{1}{n}] \left[ 1 - \frac{1}{2} \left( 1 - \frac{1}{n^2} \right) \right]$ $n \text{ even, } k = [3 - \frac{1}{2} (1 + \frac{4}{n^2})] \times n - [2(n - \frac{1}{n})]$	
CANTILEVERS		Where:	
	$\Delta = \frac{1}{8} \frac{WL^3}{EI}$	$\Delta$ = maximum deflection $W$ = total load on beam $L$ = span of beam $E$ = Young's modulus of elasticity $I$ = second moment of area of cross-section	
	$\Delta = \frac{1}{15} \frac{WL^3}{EI}$		
	$\Delta = \frac{Wa^3}{EI} \frac{1}{3} [1 + \frac{3b}{2a}]$		

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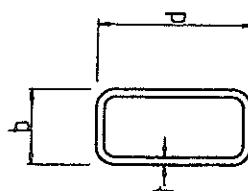
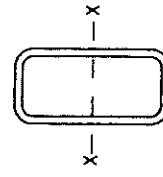


TABLE 4.1 – 1 (1)

**DESIGN MOMENT CAPACITIES FOR MEMBERS  
WITHOUT FULL LATERAL RESTRAINT  
RECTANGULAR HOLLOW SECTIONS: GRADE C350  
bending about x-axis**

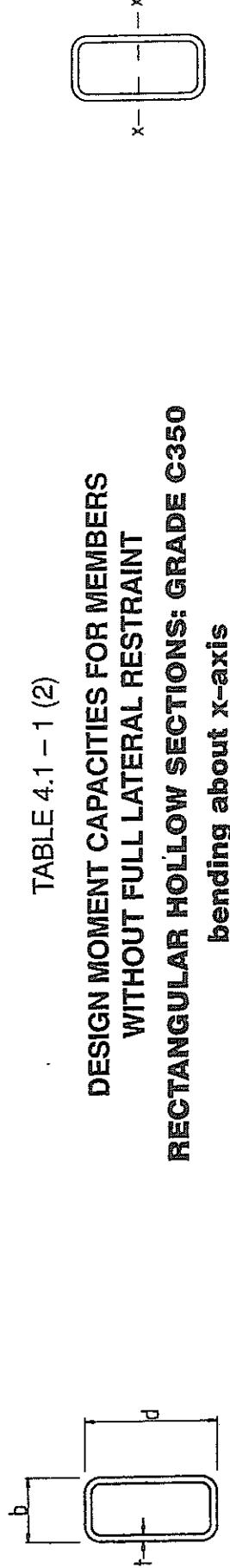


Designation <i>d b t</i>	Mass per m kg/m	Design Moment Capacities $\phi M_b$ (kNm)										FLR
		1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	9.0	10.0	
Effective Length ( $L_e$ ) in metres												
250 x 150 x 9.0 RHS	51.8	168	168	168	168	168	168	168	168	168	168	m
6.0 RHS	35.6	118	118	118	118	118	118	118	118	118	118	7.81
5.0 RHS	29.9	94.4	94.4	94.4	94.4	94.4	94.4	94.4	94.4	94.4	94.4	8.05
200 x 100 x 9.0 RHS	37.7	92.4	92.4	92.4	92.4	90.4	89.3	88.3	87.2	86.2	85.1	81.2
6.0 RHS	26.2	66.2	66.2	66.2	66.2	64.8	64.1	63.3	62.6	61.8	61.1	4.28
5.0 RHS	22.1	56.5	56.5	56.5	56.5	55.3	54.7	54.0	53.4	52.8	52.1	4.43
4.0 RHS	17.9	46.2	46.2	46.2	46.2	45.3	44.7	44.2	43.7	43.2	42.7	4.48
150 x 100 x 6.0 RHS	21.4	42.3	42.3	42.3	42.3	41.3	40.8	40.4	40.0	39.6	39.1	40.8
5.0 RHS	18.2	36.3	36.3	36.3	36.3	36.3	35.4	35.0	34.6	34.3	33.9	37.9
4.0 RHS	14.8	29.8	29.8	29.8	29.8	29.8	29.1	28.8	28.5	28.2	27.9	32.5
150 x 50 x 5.0 RHS	14.2	24.8	24.3	23.6	22.9	22.2	21.6	20.9	20.3	19.7	19.2	27.0
4.0 RHS	11.6	20.6	20.2	19.6	19.0	18.5	17.9	17.4	16.9	16.5	16.0	5.85
3.0 RHS	8.96	16.2	15.9	15.4	15.0	14.5	14.1	13.7	13.3	13.0	12.6	14.3
125 x 75 x 6.0 RHS	16.7	26.5	26.5	26.5	26.0	25.6	25.2	24.9	24.5	24.1	23.8	17.1
5.0 RHS	14.2	22.9	22.9	22.9	22.5	22.1	21.8	21.5	21.2	20.9	20.6	14.7
4.0 RHS	11.6	19.0	19.0	19.0	18.6	18.3	18.1	17.8	17.3	17.0	16.6	15.5
3.0 RHS	8.96	14.9	14.9	14.9	14.9	14.4	14.2	14.0	13.8	13.6	13.4	11.3
100 x 50 x 6.0 RHS	12.0	14.3	14.3	13.8	13.4	13.1	12.8	12.5	12.2	11.9	11.6	10.6
5.0 RHS	10.3	12.5	12.5	12.1	11.8	11.5	11.3	11.0	10.7	10.5	10.2	3.81
4.0 RHS	8.49	10.5	10.5	10.2	9.93	9.70	9.47	9.25	9.03	8.82	8.62	3.87
3.5 RHS	7.53	9.43	9.43	9.12	8.91	8.70	8.50	8.30	8.10	7.92	7.73	2.16
3.0 RHS	6.60	8.40	8.40	8.12	7.93	7.74	7.56	7.38	7.21	7.04	6.88	7.22
2.5 RHS	5.56	7.14	7.14	6.91	6.75	6.59	6.44	6.29	6.14	6.00	5.86	6.57
2.0 RHS	4.50	5.83	5.83	5.64	5.51	5.38	5.26	5.14	5.02	4.90	4.79	5.47

Notes: 1.  $\phi = 0.9$ 2. FLR - segment lengths of Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -1.0$ 

3. Values to the left of the solid line are segment lengths with full lateral restraint.

4.  $\alpha_s$  calculated in accordance with Clause 5.6.1.4 of AS 41005.  $\alpha_m = 1.0$



**TABLE 4.1 – 1 (2)**  
**DESIGN MOMENT CAPACITIES FOR MEMBERS**  
**WITHOUT FULL LATERAL RESTRAINT**  
**RECTANGULAR HOLLOW SECTIONS: GRADE C350**  
**bending about x-axis**

Designation d b t mm mm mm	Mass kg/m	Design Moment Capacities $\phi M_s$ (kNm)										FLR m	
		0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5	3.0	3.5	4.0	
100 x 50 x 6.0 RHS	12.0	14.3	14.3	14.3	14.3	14.3	14.3	14.0	13.8	13.6	13.4	13.3	12.8
5.0 RHS	10.3	12.5	12.5	12.5	12.5	12.5	12.5	12.2	12.1	12.0	11.8	11.7	11.3
4.0 RHS	8.49	10.5	10.5	10.5	10.5	10.5	10.5	10.3	10.2	10.1	9.93	9.82	9.47
3.5 RHS	7.53	9.43	9.43	9.43	9.43	9.43	9.43	9.23	9.12	9.01	8.91	8.80	8.50
3.0 RHS	6.60	8.40	8.40	8.40	8.40	8.40	8.40	8.22	8.12	8.02	7.93	7.84	7.56
2.5 RHS	5.56	7.14	7.14	7.14	7.14	7.14	7.14	6.99	6.91	6.83	6.75	6.67	6.39
2.0 RHS	4.50	5.83	5.83	5.83	5.83	5.83	5.83	5.71	5.64	5.58	5.51	5.45	5.26
75 x 50 x 6.0 RHS	9.67	8.84	8.84	8.84	8.84	8.84	8.84	8.84	8.84	8.80	8.51	8.42	8.23
5.0 RHS	8.35	7.84	7.84	7.84	7.84	7.84	7.84	7.84	7.84	7.84	7.56	7.47	7.32
4.0 RHS	6.92	6.66	6.66	6.66	6.66	6.66	6.66	6.66	6.66	6.66	6.49	6.42	6.29
3.0 RHS	5.42	5.38	5.38	5.38	5.38	5.38	5.38	5.38	5.38	5.38	5.19	5.13	5.08
2.5 RHS	4.58	4.60	4.60	4.60	4.60	4.60	4.60	4.60	4.60	4.60	4.48	4.43	4.34
2.0 RHS	3.72	3.77	3.77	3.77	3.77	3.77	3.77	3.77	3.77	3.77	3.64	3.60	3.56
75 x 25 x 2.5 RHS	3.60	3.17	3.15	3.10	3.05	3.01	2.96	2.92	2.83	2.74	2.66	2.58	2.51
2.0 RHS	2.93	2.62	2.62	2.56	2.52	2.49	2.45	2.41	2.34	2.27	2.21	2.14	2.08
1.6 RHS	2.38	2.15	2.15	2.10	2.07	2.04	2.01	1.98	1.93	1.87	1.82	1.77	1.72
65 x 35 x 3.0 RHS	4.25	3.46	3.46	3.46	3.46	3.46	3.46	3.39	3.36	3.31	3.25	3.20	3.14
2.5 RHS	3.60	2.98	2.98	2.98	2.98	2.98	2.98	2.92	2.89	2.85	2.80	2.75	2.71
2.0 RHS	2.93	2.46	2.46	2.46	2.46	2.46	2.46	2.41	2.39	2.35	2.31	2.27	2.24
50 x 25 x 3.0 RHS	3.07	1.85	1.85	1.85	1.80	1.78	1.76	1.74	1.69	1.65	1.61	1.57	1.53
2.5 RHS	2.62	1.61	1.61	1.61	1.57	1.55	1.55	1.53	1.51	1.48	1.44	1.41	1.37
2.0 RHS	2.15	1.34	1.34	1.34	1.31	1.30	1.28	1.27	1.24	1.21	1.18	1.15	1.12
1.6 RHS	1.75	1.11	1.11	1.11	1.09	1.07	1.06	1.05	1.02	0.99	0.96	0.93	0.90
50 x 20 x 3.0 RHS	2.83	1.62	1.60	1.57	1.54	1.52	1.49	1.46	1.41	1.36	1.31	1.27	1.23
2.5 RHS	2.42	1.42	1.40	1.38	1.35	1.33	1.31	1.28	1.24	1.20	1.16	1.12	1.08
2.0 RHS	1.99	1.19	1.18	1.16	1.14	1.12	1.10	1.08	1.04	1.01	0.975	0.943	0.912
1.6 RHS	1.63	0.989	0.976	0.960	0.944	0.928	0.913	0.897	0.868	0.839	0.812	0.786	0.761

Notes: 1.  $\phi = 0.9$

2. FLR – segment lengths of Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $B_m = -1.0$

3. Values to the left of the solid line are segment lengths with full lateral restraint

4.  $\alpha_m$  calculated in accordance with Clause 5.6.1.4 of AS 4100

5.  $B_m = 1.0$

**DESIGN MOMENT CAPACITIES FOR MEMBERS  
WITHOUT FULL LATERAL RESTRAINT  
RECTANGULAR HOLLOW SECTIONS: GRADE C450  
bending about x-axis**

TABLE 4.1 – 2 (1)

Designation <i>d b t</i>	Mass per m	Design Moment Capacities $\phi M_b$ (kNm)										FLR <i>m</i>				
		1.0	1.5	2.0	2.5	3.0	3.5	4.0	4.5	5.0	6.0	7.0				
125 x 76 x 3.8 RHS	11.1	23.3	23.3	23.3	23.3	22.7	22.5	22.3	22.1	21.7	21.3	20.9	20.5	20.1	3.07	
3.3 RHS	9.73	20.6	20.6	20.6	20.6	20.1	19.9	19.7	19.5	19.2	18.8	18.5	18.1	17.8	3.09	
2.8 RHS	8.39	16.9	16.9	16.9	16.9	16.5	16.4	16.3	16.1	15.8	15.6	15.3	15.0	14.8	3.12	
2.3 RHS	6.95	12.3	12.3	12.3	12.3	12.1	12.0	11.9	11.8	11.6	11.4	11.3	11.1	10.9	3.14	
100 x 50 x 3.3 RHS	7.14	11.5	11.5	11.3	11.1	10.9	10.8	10.6	10.5	10.3	9.99	9.69	9.41	9.13	8.87	1.71
2.8 RHS	6.19	10.2	10.2	9.93	9.78	9.63	9.48	9.34	9.20	9.06	8.79	8.53	8.28	8.03	7.80	1.73
2.3 RHS	5.14	8.52	8.52	8.33	8.20	8.08	7.95	7.83	7.72	7.60	7.38	7.16	6.95	6.75	6.55	1.75

Notes: Refer to TABLE 4.1 – 2 (2)

TABLE 4.1 – 2 (2)

Designation <i>d b t</i>	Mass per m	Design Moment Capacities $\phi M_b$ (kNm)										FLR <i>m</i>					
		0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5	3.0	3.5	4.0	4.5	5.0	6.0		
100 x 50 x 3.3 RHS	7.14	11.5	11.5	11.5	11.5	11.5	11.4	11.3	11.1	10.9	10.8	10.6	10.5	10.3	9.99	1.71	
2.8 RHS	6.19	10.2	10.2	10.2	10.2	10.2	10.2	10.0	9.93	9.78	9.63	9.48	9.34	9.20	9.06	8.78	1.73
2.3 RHS	5.14	8.52	8.52	8.52	8.52	8.52	8.52	8.52	8.33	8.20	8.08	7.95	7.83	7.72	7.60	7.38	1.75
75 x 50 x 2.8 RHS	5.09	6.52	6.52	6.52	6.52	6.52	6.52	6.52	6.33	6.24	6.16	6.07	5.99	5.91	5.75	2.24	
2.8 RHS	4.24	5.49	5.49	5.49	5.49	5.49	5.49	5.49	5.33	5.26	5.19	5.12	5.05	4.98	4.85	2.26	
65 x 35 x 2.8 RHS	3.99	4.21	4.21	4.21	4.21	4.10	4.05	4.01	3.93	3.84	3.76	3.68	3.60	3.53	3.38	1.26	
2.3 RHS	3.34	3.57	3.57	3.57	3.57	3.48	3.44	3.40	3.33	3.26	3.19	3.13	3.06	3.00	2.87	1.28	
50 x 25 x 2.8 RHS	2.89	2.26	2.26	2.20	2.16	2.13	2.10	2.08	2.00	1.94	1.88	1.82	1.76	1.71	1.61	0.815	
2.3 RHS	2.44	1.94	1.94	1.89	1.86	1.83	1.80	1.77	1.72	1.67	1.62	1.57	1.52	1.47	1.39	0.833	
50 x 20 x 2.8 RHS	2.67	1.99	1.93	1.88	1.84	1.80	1.76	1.72	1.64	1.57	1.51	1.44	1.38	1.32	1.22	0.517	
2.3 RHS	2.25	1.71	1.66	1.63	1.59	1.56	1.52	1.49	1.42	1.36	1.31	1.25	1.20	1.15	1.06	0.532	

Notes: 1.  $\phi = 0.9$   
 2. FLR – segment lengths of Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -1.0$   
 3. Values to the left of the solid line are segment lengths with full lateral restraint.  
 4.  $\alpha_4$  calculated in accordance with Clause 5.6.1.4 of AS 4100  
 5.  $\alpha_m = 1.0$

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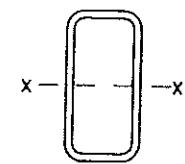
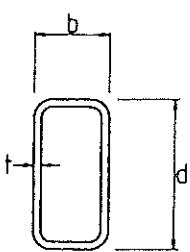


TABLE 4.3-1(A)  
DESIGN WEB CAPACITIES OF BEAMS  
RECTANGULAR HOLLOW SECTIONS: GRADE C350  
about x-axis

Designation <i>d b t</i> mm mm mm	Mass per m kg/m	$\phi V_v$	$\frac{\phi R_{by}}{b_b}$	$\frac{\phi R_{bb}}{b_b}$	Useful Parameters			
			kN	kN/mm	$s_{ext}$ mm	$b_{bw}$ mm	$L_e/r$ mm	
250 x 150 x 9.0 RHS	51.8	759	1.48	2.98	113	103	78.9	
	6.0 RHS	35.6	519	0.962	1.03	75.0	110	127
	5.0 RHS	29.9	436	0.795	0.610	62.5	113	156
200 x 100 x 9.0 RHS	37.7	590	1.52	3.83	113	77.5	59.7	
	6.0 RHS	26.2	406	0.976	1.51	75.0	85.0	98.1
	5.0 RHS	22.1	342	0.804	0.923	62.5	87.5	121
	4.0 RHS	17.9	276	0.636	0.488	50.0	90.0	156
150 x 100 x 6.0 RHS	21.4	303	1.00	2.26	75.0	60.0	69.3	
	5.0 RHS	18.2	256	0.819	1.48	62.5	62.5	86.6
	4.0 RHS	14.8	208	0.645	0.828	50.0	65.0	113
150 x 50 x 5.0 RHS	14.2	246	0.819	1.48	62.5	62.5	86.6	
	4.0 RHS	11.6	200	0.645	0.828	50.0	65.0	113
	3.0 RHS	8.96	152	0.615	0.353	30.0	69.0	159
125 x 75 x 6.0 RHS	16.7	247	1.02	2.70	75.0	47.5	54.8	
	5.0 RHS	14.2	209	0.833	1.89	62.5	50.0	69.3
	4.0 RHS	11.6	170	0.653	1.12	50.0	52.5	90.9
	3.0 RHS	8.96	130	0.619	0.493	30.0	56.5	130
100 x 50 x 6.0 RHS	12.0	190	1.06	3.10	75.0	35.0	40.4	
	5.0 RHS	10.3	162	0.855	2.32	62.5	37.5	52.0
	4.0 RHS	8.49	132	0.666	1.51	50.0	40.0	69.3
	3.5 RHS	7.53	117	0.576	1.11	43.8	41.3	81.7
	3.0 RHS	6.60	102	0.624	0.721	30.0	44.0	102
	2.5 RHS	5.56	85.5	0.516	0.441	25.0	45.0	125
	2.0 RHS	4.50	69.1	0.410	0.235	20.0	46.0	159
75 x 50 x 6.0 RHS	9.67	138	1.13	3.45	75.0	22.5	26.0	
	5.0 RHS	8.35	119	0.899	2.71	62.5	25.0	34.6
	4.0 RHS	6.92	98.0	0.691	1.94	50.0	27.5	47.6
	3.0 RHS	5.42	75.7	0.634	1.08	30.0	31.5	72.7
	2.5 RHS	4.58	64.0	0.523	0.707	25.0	32.5	90.1
	2.0 RHS	3.72	51.9	0.414	0.395	20.0	33.5	116
75 x 25 x 2.5 RHS	3.60	61.5	0.523	0.707	25.0	32.5	90.1	
	2.0 RHS	2.93	49.9	0.414	0.395	20.0	33.5	116
	1.6 RHS	2.38	40.4	0.329	0.212	16.0	34.3	149
65 x 35 x 3.0 RHS	4.25	64.0	0.641	1.25	30.0	26.5	61.2	
	2.5 RHS	3.60	54.2	0.527	0.859	25.0	27.5	76.2
	2.0 RHS	2.93	44.1	0.417	0.500	20.0	28.5	98.7
50 x 25 x 3.0 RHS	3.07	47.5	0.657	1.51	30.0	19.0	43.9	
	2.5 RHS	2.62	40.5	0.538	1.12	25.0	20.0	55.4
	2.0 RHS	2.15	33.1	0.423	0.721	20.0	21.0	72.7
	1.6 RHS	1.75	27.0	0.334	0.425	16.0	21.8	94.4
50 x 20 x 3.0 RHS	2.63	46.9	0.657	1.51	30.0	19.0	43.9	
	2.5 RHS	2.42	40.0	0.538	1.12	25.0	20.0	55.4
	2.0 RHS	1.99	32.7	0.423	0.721	20.0	21.0	72.7
	1.6 RHS	1.63	26.6	0.334	0.425	16.0	21.8	94.4

Notes: 1.  $\phi = 0.9$   
 2.  $L_e/r = \sqrt{12(d - 2r_{ex})/t}$   
 3.  $\phi R_{by} = 2\phi \alpha_p b_b t f_y$   
 4.  $\phi R_{bb} = 2\phi \alpha_b b_b t f_y$   
 5.  $\alpha_b = 0.5$   
 6.  $k_f = 1.0$

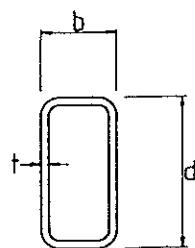
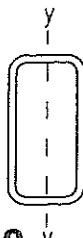


TABLE 4.3-1(B)

**DESIGN WEB CAPACITIES OF BEAMS  
RECTANGULAR HOLLOW SECTIONS: GRADE C350  
about y-axis**



Designation <i>d</i> <i>b</i> <i>t</i>	Mass per m	$\phi R_{by}$		$\phi R_{bb}$		Useful Parameters		
		$\phi V_y$	$\frac{\phi R_{by}}{b_b}$	$\frac{\phi R_{bb}}{b_b}$	$s_{ext}$	$b_{bw}$	$L_e/r$	
mm mm mm	kg/m	kN	kN/mm	kN/mm	mm	mm	mm	
250 x 150 x 9.0 RHS	51.8	449	1.59	4.65	113	52.5	40.4	
	6.0 RHS	35.6	313	1.00	2.26	75.0	60.0	69.3
	5.0 RHS	29.9	265	0.819	1.48	62.5	62.5	86.6
200 x 100 x 9.0 RHS	37.7	279	1.76	5.34	113	27.5	21.2	
	6.0 RHS	26.2	200	1.06	3.10	75.0	35.0	40.4
	5.0 RHS	22.1	170	0.855	2.32	62.5	37.5	52.0
	4.0 RHS	17.9	139	0.666	1.51	50.0	40.0	69.3
150 x 100 x 6.0 RHS	21.4	200	1.06	3.10	75.0	35.0	40.4	
	5.0 RHS	18.2	170	0.855	2.32	62.5	37.5	52.0
	4.0 RHS	14.8	139	0.666	1.51	50.0	40.0	69.3
150 x 50 x 5.0 RHS	14.2	75.6	1.02	3.04	62.5	12.5	17.3	
	4.0 RHS	11.6	63.5	0.754	2.30	50.0	15.0	26.0
	3.0 RHS	8.96	49.9	0.657	1.51	30.0	19.0	43.9
125 x 75 x 6.0 RHS	16.7	143	1.13	3.45	75.0	22.5	26.0	
	5.0 RHS	14.2	123	0.899	2.71	62.5	25.0	34.6
	4.0 RHS	11.6	101	0.691	1.94	50.0	27.5	47.6
	3.0 RHS	8.96	78.2	0.634	1.08	30.0	31.5	72.7
100 x 50 x 6.0 RHS	12.0	86.2	1.37	3.78	75.0	10.0	11.5	
	5.0 RHS	10.3	75.6	1.02	3.04	62.5	12.5	17.3
	4.0 RHS	8.49	63.5	0.754	2.30	50.0	15.0	26.0
	3.5 RHS	7.53	56.9	0.636	1.93	43.8	16.3	32.2
	3.0 RHS	6.60	49.9	0.657	1.51	30.0	19.0	43.9
	2.5 RHS	5.56	42.5	0.538	1.12	25.0	20.0	55.4
	2.0 RHS	4.50	34.8	0.423	0.721	20.0	21.0	72.7
75 x 50 x 6.0 RHS	9.67	86.2	1.37	3.78	75.0	10.0	11.5	
	5.0 RHS	8.35	75.6	1.02	3.04	62.5	12.5	17.3
	4.0 RHS	6.92	63.5	0.754	2.30	50.0	15.0	26.0
	3.5 RHS	7.53	56.9	0.636	1.93	43.8	16.3	32.2
	3.0 RHS	6.60	49.9	0.657	1.51	30.0	19.0	43.9
	2.5 RHS	5.56	42.5	0.538	1.12	25.0	20.0	55.4
75 x 25 x 2.5 RHS	9.67	34.8	0.423	0.721	20.0	21.0	72.7	
	2.0 RHS	2.93	15.9	0.456	1.12	20.0	8.50	29.4
	1.6 RHS	2.38	13.2	0.353	0.828	16.0	9.30	40.3
65 x 35 x 3.0 RHS	4.25	32.9	0.693	1.72	30.0	11.5	26.6	
	2.5 RHS	3.60	28.3	0.560	1.35	25.0	12.5	34.6
	2.0 RHS	2.93	23.4	0.436	0.977	20.0	13.5	46.8
50 x 25 x 3.0 RHS	3.07	21.5	0.759	1.85	30.0	6.50	15.0	
	2.5 RHS	2.62	18.9	0.598	1.49	25.0	7.50	20.8
	2.0 RHS	2.15	15.9	0.456	1.12	20.0	8.50	29.4
	1.6 RHS	1.75	13.2	0.353	0.828	16.0	9.30	40.3
50 x 20 x 3.0 RHS	2.83	15.9	0.844	1.89	30.0	4.00	9.24	
	2.5 RHS	2.42	14.2	0.643	1.55	25.0	5.00	13.9
	2.0 RHS	1.99	12.1	0.479	1.19	20.0	6.00	20.8
	1.6 RHS	1.63	10.2	0.365	0.899	16.0	6.80	29.4

- Notes:
1.  $\phi = 0.9$
  2.  $L_e/r = \sqrt{12(d - 2r_{ext})/t}$
  3.  $\phi R_{by} = 2\phi \alpha_p b_b t f_y$
  4.  $\phi R_{bb} = 2\phi \alpha_c b_b t f_y$
  5.  $\alpha_b = 0.5$
  6.  $\alpha_t = 1.0$

TABLE 4.3-2(A)

**DESIGN WEB CAPACITIES OF BEAMS**

**RECTANGULAR HOLLOW SECTIONS: GRADE C450**

**about x-axis**

Designation d b t	Mass per m	$\phi V_y$	$\frac{\phi R_{by}}{b_b}$	$\frac{\phi R_{bb}}{b_b}$	Useful Parameters		
			kN	kN/mm	kN/mm	$s_{ext}$	$b_{bw}$
mm mm mm	kg/m				mm	mm	mm
125 x 75 x 3.8 RHS	11.1	209	0.795	1.05	47.5	53.0	96.6
3.3 RHS	9.73	183	0.684	0.708	41.3	54.3	114
2.8 RHS	8.39	156	0.741	0.422	28.0	56.9	141
2.3 RHS	6.95	129	0.605	0.237	23.0	57.9	174
100 x 50 x 3.3 RHS	7.14	143	0.695	1.05	41.3	41.8	87.7
2.8 RHS	6.19	122	0.747	0.637	28.0	44.4	110
2.3 RHS	5.14	102	0.609	0.364	23.0	45.4	137
75 x 50 x 2.8 RHS	5.09	91.4	0.758	1.03	28.0	31.9	78.9
2.3 RHS	4.24	76.2	0.616	0.614	23.0	32.9	99.1
65 x 35 x 2.8 RHS	3.99	77.3	0.765	1.25	28.0	26.9	66.6
2.3 RHS	3.34	64.6	0.621	0.776	23.0	27.9	84.0
50 x 25 x 2.8 RHS	2.89	57.5	0.783	1.63	28.0	19.4	48.0
2.3 RHS	2.44	48.3	0.632	1.11	23.0	20.4	61.5
50 x 20 x 2.8 RHS	2.67	56.8	0.783	1.63	28.0	19.4	48.0
2.3 RHS	2.25	47.7	0.632	1.11	23.0	20.4	61.5

Notes:

- $\phi = 0.9$
- $L_e/r = \sqrt{12(d - 2r_{ext})/t}$
- $\phi R_{by} = 2\phi \alpha_p b_b t f_y$
- $\phi R_{bb} = 2\phi \alpha_c b_b t f_y$
- $\alpha_p = 0.5$
- $k_t = 1.0$

TABLE 4.3-2(B)

**DESIGN WEB CAPACITIES OF BEAMS**

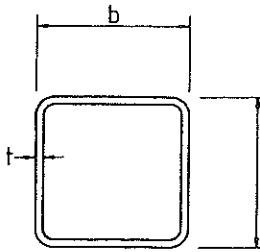
**RECTANGULAR HOLLOW SECTIONS: GRADE C450**

**about y-axis**

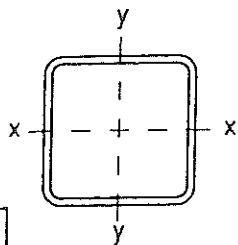
Designation d b t	Mass per m	$\phi V_y$	$\frac{\phi R_{by}}{b_b}$	$\frac{\phi R_{bb}}{b_b}$	Useful Parameters		
			kN	kN/mm	kN/mm	$s_{ext}$	$b_{bw}$
mm mm mm	kg/m				mm	mm	mm
125 x 75 x 3.8 RHS	11.1	124	0.837	2.12	47.5	28.0	51.0
3.3 RHS	9.73	110	0.714	1.59	41.3	29.3	61.4
2.8 RHS	8.39	94.4	0.758	1.03	28.0	31.9	78.9
2.3 RHS	6.95	78.7	0.616	0.614	23.0	32.9	99.1
100 x 50 x 3.3 RHS	7.14	69.6	0.761	2.20	41.3	16.8	35.2
2.8 RHS	6.19	60.4	0.783	1.63	28.0	19.4	48.0
2.3 RHS	5.14	50.7	0.632	1.11	23.0	20.4	61.5
75 x 50 x 2.8 RHS	5.09	60.4	0.783	1.63	28.0	19.4	48.0
2.3 RHS	4.24	50.7	0.632	1.11	23.0	20.4	61.5
65 x 35 x 2.8 RHS	3.99	40.0	0.821	1.97	28.0	11.9	29.4
2.3 RHS	3.34	34.0	0.655	1.48	23.0	12.9	38.9
50 x 25 x 2.8 RHS	2.89	26.4	0.890	2.16	28.0	6.90	17.1
2.3 RHS	2.44	22.8	0.694	1.69	23.0	7.90	23.8
50 x 20 x 2.8 RHS	2.67	19.6	0.974	2.26	28.0	4.40	10.9
2.3 RHS	2.25	17.2	0.737	1.79	23.0	5.40	16.3

Notes:

- $\phi = 0.9$
- $L_e/r = \sqrt{12(d - 2r_{ext})/t}$
- $\phi R_{by} = 2\phi \alpha_p b_b t f_y$
- $\phi R_{bb} = 2\phi \alpha_c b_b t f_y$
- $\alpha_p = 0.5$
- $k_t = 1.0$



**TABLE 4.3-3**  
**DESIGN WEB CAPACITIES OF BEAMS**  
**SQUARE HOLLOW SECTIONS: GRADE C350**  
**about x- and y-axis**



Designation d    b    t	Mass per m kg/m	$\phi V_y$	$\frac{\phi R_{by}}{b_b}$	$\frac{\phi R_{bb}}{b_b}$	Useful Parameters		
					$5r_{ext}$	$b_{ter}$	$L_o/r$
250 x 250 x 9.0 SHS	65.9	780	1.48	2.98	113	103	78.9
6.0 SHS	45.0	533	0.962	1.03	75.0	110	127
200 x 200 x 9.0 SHS	51.8	612	1.52	3.83	113	77.5	59.7
6.0 SHS	35.6	421	0.976	1.51	75.0	85.0	98.1
5.0 SHS	29.9	355	0.804	0.923	62.5	87.5	121
150 x 150 x 9.0 SHS	37.7	444	1.59	4.65	113	52.5	40.4
6.0 SHS	26.2	309	1.00	2.26	75.0	60.0	69.3
5.0 SHS	22.1	261	0.819	1.48	62.5	62.5	86.6
125 x 125 x 9.0 SHS	30.6	360	1.65	5.01	113	40.0	30.8
6.0 SHS	21.4	253	1.02	2.70	75.0	47.5	54.8
5.0 SHS	18.2	215	0.833	1.89	62.5	50.0	69.3
4.0 SHS	14.8	175	0.653	1.12	50.0	52.5	90.9
100 x 100 x 9.0 SHS	23.5	276	1.76	5.34	113	27.5	21.2
6.0 SHS	16.7	197	1.06	3.10	75.0	35.0	40.4
5.0 SHS	14.2	168	0.855	2.32	62.5	37.5	52.0
4.0 SHS	11.6	137	0.666	1.51	50.0	40.0	69.3
3.0 SHS	8.96	105	0.524	0.721	30.0	44.0	102
89 x 89 x 6.0 SHS	14.6	172	1.08	3.26	75.0	29.5	34.0
5.0 SHS	12.5	147	0.871	2.50	62.5	32.0	44.3
3.5 SHS	9.06	107	0.582	1.30	43.8	35.7	70.7
75 x 75 x 6.0 SHS	12.0	141	1.13	3.45	75.0	22.5	26.0
5.0 SHS	10.3	121	0.899	2.71	62.5	25.0	34.6
4.0 SHS	8.49	100	0.691	1.94	50.0	27.5	47.6
3.5 SHS	7.53	88.9	0.593	1.54	43.8	28.8	56.9
3.0 SHS	6.60	77.3	0.634	1.08	30.0	31.5	72.7
2.5 SHS	5.56	65.3	0.523	0.707	25.0	32.5	90.1
65 x 65 x 6.0 SHS	10.1	119	1.19	3.58	75.0	17.5	20.2
5.0 SHS	8.75	103	0.931	2.84	62.5	20.0	27.7
4.0 SHS	7.23	85.1	0.708	2.09	50.0	22.5	39.0
3.0 SHS	5.66	66.1	0.641	1.25	30.0	26.5	61.2
2.5 SHS	4.78	56.0	0.527	0.859	25.0	27.5	76.2
2.0 SHS	3.88	45.5	0.417	0.500	20.0	28.5	98.7
50 x 50 x 5.0 SHS	6.39	74.7	1.02	3.04	62.5	12.5	17.3
4.0 SHS	5.35	62.7	0.754	2.30	50.0	15.0	26.0
3.0 SHS	4.25	49.3	0.657	1.51	30.0	19.0	43.9
2.5 SHS	3.60	42.0	0.538	1.12	25.0	20.0	55.4
2.0 SHS	2.93	34.3	0.423	0.721	20.0	21.0	72.7
1.6 SHS	2.38	28.0	0.334	0.425	16.0	21.8	94.4
40 x 40 x 4.0 SHS	4.09	47.8	0.820	2.43	50.0	10.0	17.3
3.0 SHS	3.30	38.1	0.677	1.65	30.0	14.0	32.3
2.5 SHS	2.82	32.7	0.550	1.28	25.0	15.0	41.6
2.0 SHS	2.31	26.9	0.430	0.894	20.0	16.0	55.4
1.6 SHS	1.88	22.0	0.338	0.577	16.0	16.8	72.7
35 x 35 x 3.0 SHS	2.83	32.5	0.693	1.72	30.0	11.5	26.6
2.5 SHS	2.42	28.0	0.560	1.35	25.0	12.5	34.6
2.0 SHS	1.99	23.1	0.436	0.977	20.0	13.5	46.8
1.6 SHS	1.63	19.0	0.342	0.663	16.0	14.3	61.9
30 x 30 x 3.0 RHS	2.36	26.9	0.718	1.78	30.0	9.00	20.8
2.5 RHS	2.03	23.3	0.575	1.42	25.0	10.0	27.7
2.0 SHS	1.68	19.4	0.444	1.05	20.0	11.0	38.1
1.6 SHS	1.38	16.0	0.346	0.749	16.0	11.8	51.1
25 x 25 x 3.0 SHS	1.89	21.3	0.759	1.85	30.0	6.50	15.0
2.5 SHS	1.64	18.7	0.598	1.49	25.0	7.50	20.8
2.0 SHS	1.36	15.7	0.456	1.12	20.0	8.50	29.4
1.6 SHS	1.12	13.0	0.353	0.828	16.0	9.30	40.3
20 x 20 x 1.6 SHS	0.873	10.0	0.365	0.899	16.0	6.80	29.4

Notes: 1.  $\phi = 0.9$   
          2.  $L_o/r = \sqrt{12}(d - 2r_{ext})/l$   
          3.  $\phi R_{by} = 2\phi \alpha_p b_b t f_y$   
          4.  $\phi R_{bb} = 2\phi \alpha_p b_b t f_y$   
          5.  $\alpha_p = 0.5$   
          6.  $f_y = 1.0$

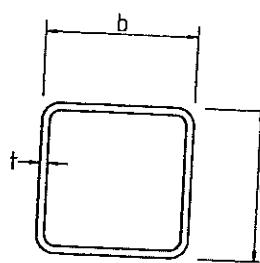
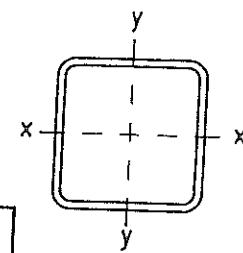


TABLE 4.3-4  
DESIGN WEB CAPACITIES OF BEAMS  
**SQUARE HOLLOW SECTIONS: GRADE C450**  
about x- and y-axis



Designation d b t mm mm mm	Mass per m kg/m	$\phi V_v$	$\frac{\phi R_{by}}{b_b}$	$\frac{\phi R_{bb}}{b_b}$	Useful Parameters		
			kN	kN/mm	$r_{ext}$ mm	$b_{bw}$ mm	$L_e/r$ mm
100 x 100 x 3.8 SHS	11.1	169	0.810	1.51	47.5	40.5	73.8
3.3 SHS	9.73	148	0.695	1.05	41.3	41.8	87.7
2.8 SHS	8.39	127	0.747	0.637	28.0	44.4	110
2.3 RHS	6.95	105	0.609	0.364	23.0	45.4	137
75 x 75 x 3.3 SHS	7.14	108	0.714	1.59	41.3	29.3	61.4
2.8 SHS	6.19	93.3	0.758	1.03	28.0	31.9	78.9
2.3 SHS	5.14	77.7	0.616	0.614	23.0	32.9	99.1
65 x 65 x 2.8 SHS	5.31	79.8	0.765	1.25	28.0	26.9	66.6
2.3 SHS	4.42	66.7	0.621	0.776	23.0	27.9	84.0
50 x 50 x 2.8 SHS	3.99	59.7	0.783	1.63	28.0	19.4	48.0
2.3 SHS	3.34	50.1	0.632	1.11	23.0	20.4	61.5
40 x 40 x 2.8 SHS	3.11	46.2	0.804	1.86	28.0	14.4	35.6
2.3 SHS	2.62	39.1	0.645	1.36	23.0	15.4	46.4
35 x 35 x 2.8 SHS	2.67	39.5	0.821	1.97	28.0	11.9	29.4
2.3 SHS	2.25	33.6	0.655	1.48	23.0	12.9	38.9
30 x 30 x 2.8 SHS	2.23	32.8	0.847	2.07	28.0	9.40	23.3
2.3 SHS	1.89	28.0	0.670	1.59	23.0	10.4	31.3
25 x 25 x 2.3 SHS	1.53	22.5	0.694	1.69	23.0	7.90	23.8

Notes:  
 1.  $\phi = 0.9$   
 2.  $L_e/r = \sqrt{12(d - 2r_{ext})}/t$   
 3.  $\phi R_{by} = 2\phi \alpha_b b_b t f_y$   
 4.  $\phi R_{bb} = 2\phi \alpha_c b_b t f_y$   
 5.  $\alpha_b = 0.5$   
 6.  $f_y = 1.0$

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**PART 5 MEMBERS SUBJECT TO AXIAL COMPRESSION**

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5.2 DESIGN CAPACITY FOR MEMBERS SUBJECT TO AXIAL COMPRESSION	5-2
5.3 METHOD	5-2
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**TABLES****TABLES 5-2-1 to 5-2-6**

Design Capacities for Members Subject to Axial Compression ( $\phi N_c$ )  
Buckling About Principal Axis

5-8

**NOTE: SEE PAGE (ix) FOR THE SPECIFIC MATERIAL STANDARD REFERRED TO BY THE SECTION TYPE AND STEEL GRADE IN THESE TABLES**

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**PART 5 MEMBERS SUBJECT TO AXIAL COMPRESSION****5.1 Scope**

The following tables and graphs give values of design axial compression capacity for various effective lengths and have been determined using Section 6 of AS 4100.

All loads are assumed to be applied through the centroid of the section and the column capacity is assumed to be associated with flexural buckling about either x- or y-axis.

**5.2 Design Capacity for Members Subject to Axial Compression**

Values of the design capacity for axial compression ( $\phi N_c$ ) for buckling about both principal axes, based on the appropriate effective length ( $L_e$ ), are given in Tables 5.2-1 to 5.2-6. The tables are supplemented by graphs of  $\phi N_c$  versus  $L_e$  placed consecutively after the tables for each corresponding grade and section.

The tables in this section have been grouped into two series for the rectangular hollow sections:

- the (A) series (e.g. Table 5.2-3(A)) for the member buckling about the x-axis; and
- the (B) series (e.g. Table 5.2-3(B)) for the member buckling about the y-axis.

For each section grade C350 and C450, the (A) series tables and graphs are immediately followed by the (B) series of tables and graphs.

**5.3 Method**

The design axial compression member capacity is obtained from Clauses 6.3 of AS 4100 and is given by:

$$\phi N_c = \phi \sigma_y A$$

where  $\phi_c = 0.9$  (Table 3.4 of AS 4100)

$\phi N_c = 0.9 K_c A_y f_y$  — see Section 3.2.2 and Tables 3.1-1 to 3.1-6

$K_c = A_c/A_b$  — see Section 1.2.3.6 and Clause 6.2.2 of AS 4100

$A_c$  = net section area

$A_b$  =

= gross cross-sectional area (assumed no penetration or holes)

$f_y$  = yield stress used in design

$\phi_c$  = member slenderness reduction factor

According to Clause 6.3.3 of AS 4100,  $\alpha_c$  depends on the modified slenderness reduction factor ( $\lambda_n$ ) and the member section constant ( $\alpha_b$ ). The member slenderness reduction factor ( $\alpha_c$ ) is determined from Clause 6.3.3 of AS 4100 and taken as:

$$\alpha_c = \xi \left\{ 1 - \sqrt{\left[ 1 - \left( \frac{90}{\xi \lambda} \right)^2 \right]} \right\}$$

$$\text{where } \xi = \frac{\left( \frac{\lambda}{90} \right)^2 + 1 + \eta}{2 \left( \frac{\lambda}{90} \right)^2}$$

$$\lambda = \lambda_n + \alpha_a \alpha_b$$

$$\eta = 0.00326(\lambda - 13.5) \geq 0$$

$$\lambda_n = \left( \frac{L_e}{r} \right) \sqrt{(k_f)} \sqrt{\left( \frac{f_y}{250} \right)}$$

$$\alpha_a = \frac{2100(\lambda_n - 13.5)}{\lambda_n^2 - 15.3\lambda_n + 2050}$$

$$\alpha_b = -0.5 \quad (\text{Table 5.3})$$

$\left( \frac{L_e}{r} \right)$  = geometrical slenderness ratio

$L_e$  = effective length of a compression member (see Section 5.4)

$r$  = radius of gyration (see tables 1.2-1 to 1.2-6)

$k_f$  = form factor (see Section 1.2.3.3 and Tables 1.2-1 to 1.2-6)

$f_y$  = yield stress used in design

Note: the member capacity equals the section capacity ( $\phi N_c = \phi N_s$ ) when the effective length ( $L_e = 0$ ). The residual stress classification used in determining  $k_f$  is shown in Table 5.3 and is described in Section 1.2.3.3 of the Tables.

Table 5.3

Section	Residual Stresses	Yield Slenderness Limit $\lambda_{ey}$	$\alpha_b$	
			$k_f = 1.0$	$k_f < 1.0$
RHS, SHS	CF	40	-0.5	-0.5
CHS	CF	82	-0.5	-0.5

## 5.4 Effective Length

Before using these tables and graphs it is necessary to determine the effective length, which depends on the rotational and translational restraints at the ends of the member and is determined using the following formula:

$$L_e = k_e L$$

The member effective length factor ( $k_e$ ) (Clause 6.3.2 of AS 4100) can be determined using Clause 4.6.3 of AS 4100 or by a rational frame buckling analysis as described in Clause 4.7 of AS 4100. Information relevant to assessing  $k_e$ , for members in frames using Clause 4.6 of AS 4100, is reproduced in Part 7 of this publication. For members with idealised end restraints  $k_e$  is given in Table 5.4.

**Table 5.4: Effective Length Factors for Members for Idealised Conditions of End Restraint**

	Braced Member			Sway Member		
	Buckled Shape					
Effective length factor ( $k_e$ )	0.7	0.85	1.0	1.2	2.2	2.2
Symbols for end restraint conditions	 = Rotation fixed, translation fixed  = Rotation free, translation fixed	 = Rotation fixed, translation free  = Rotation free, translation free				

Advice on the determination of effective lengths of members in trusses and girders is not covered in AS 4100, but suitable equations have been developed in research<sup>[1]</sup>. The following equations are applicable for:

- welded joints with gap or partial overlap
- bracing members welded along the full perimeter
- no cropping or flattening of the ends of the bracings

In all cases  $L_e / L \geq 0.5$

- (i) Chord: circular  
Bracing: circular

$$\frac{L_e}{L} = 2.20 \left( \frac{d_1^2}{Ld_o} \right)^{0.25} \leq 0.75$$

- (ii) Chord: square  
Bracing: circular

$$\frac{L_e}{L} = 2.35 \left( \frac{d_1^2}{Lb_o} \right)^{0.25} \leq 0.75$$

- (iii) Chord: square  
Bracing: square

$$\frac{L_e}{L} = 2.30 \left( \frac{b_1^2}{Lb_o} \right)^{0.25} \leq 0.75$$

where  $L_e$  = effective length of a compression member

$L$  = distance between intersection points of chord and web centre lines (see Figure 5.4)

$d_o$  = outside diameter of a circular chord member

$d_1$  = outside diameter of a circular bracing member

$b_o$  = external width of a square chord member

$b_1$  = external width of a square bracing member

The effective length of members in a pitched roof can be determined from the reference<sup>[2]</sup> below.

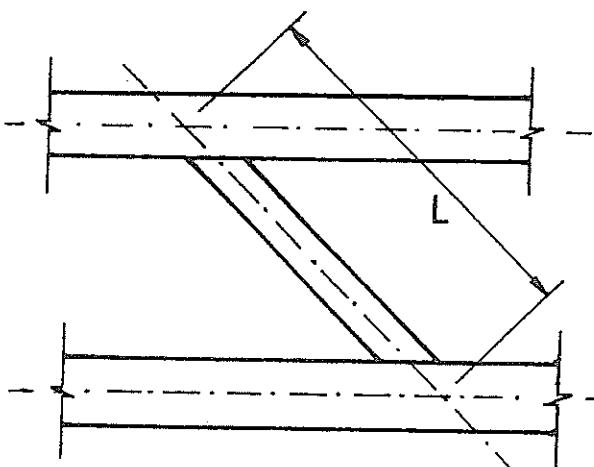


Figure 5.4:

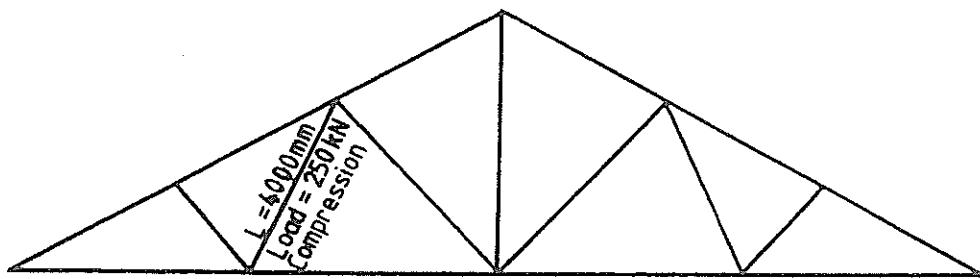
- 
- [1] Rondal, J., "Effective Lengths of Tubular Lattice Girder Members Statistical Tests", CIDECT/R 3 K - 90/3 Final Report, University of Liege, 1990.
  - [2] Fraser, D.J., "Stability of Pitched Roof Frames", Transaction of The Institution of Engineers Australia, Civil Engineering, Vol. CE28, No. 1, 1986.

## 5.5 Modes of Buckling

Although it is also possible for some doubly symmetric sections to buckle in a torsional mode, this is not the governing buckling mode for hollow sections.

## 5.6 Example

1. A compression member in a truss, shown below, is to resist a concentrically applied axial compression force of 250 kN. End connections are full perimeter welded gap joints, with chord and web members being the same size and shape. Circular hollow section web members have profiled ends.



### Design Data:

$$N^* = 250 \text{ kN}$$

### Solution:

Effective length

for  $d_1 = d_0$  ( $d_1 > 54 \text{ mm}$ )  
and  $b_1 = b_0$  ( $b_1 > 45 \text{ mm}$ )

$$\begin{aligned} L_e &= 0.75 \times 4.0 \quad (\text{Section 5.4}) \\ &= 3.0 \text{ m} \end{aligned}$$

Selecting a hollow section with the least mass from Tables 5.2–1 to 5.2–6, the alternatives are:

101.6 x 5.0 CHS – Grade C250	(11.9 kg/m)	$\phi N_c = 236 > N^*$	(Table 5.2–1(2))
114.3 x 3.2 CHS – Grade C350	(8.77 kg/m)	$\phi N_c = 237 > N^*$	(Table 5.2–2(2))
100 x 100 x 3.0 SHS – Grade C350	(8.96 kg/m)	$\phi N_c = 242 > N^*$	(Table 5.2–5(1))
100 x 100 x 2.8 SHS – Grade C450	(8.39 kg/m)	$\phi N_c = 242 > N^*$	(Table 5.2–6(1))

Based on mass select 100 x 100 x 2.8 SHS – Grade C450 (8.39 kg/m).

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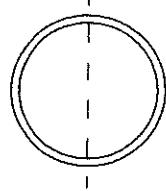
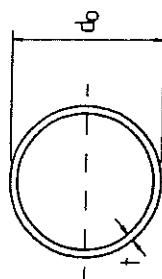


TABLE 5.2 – 1 (1)

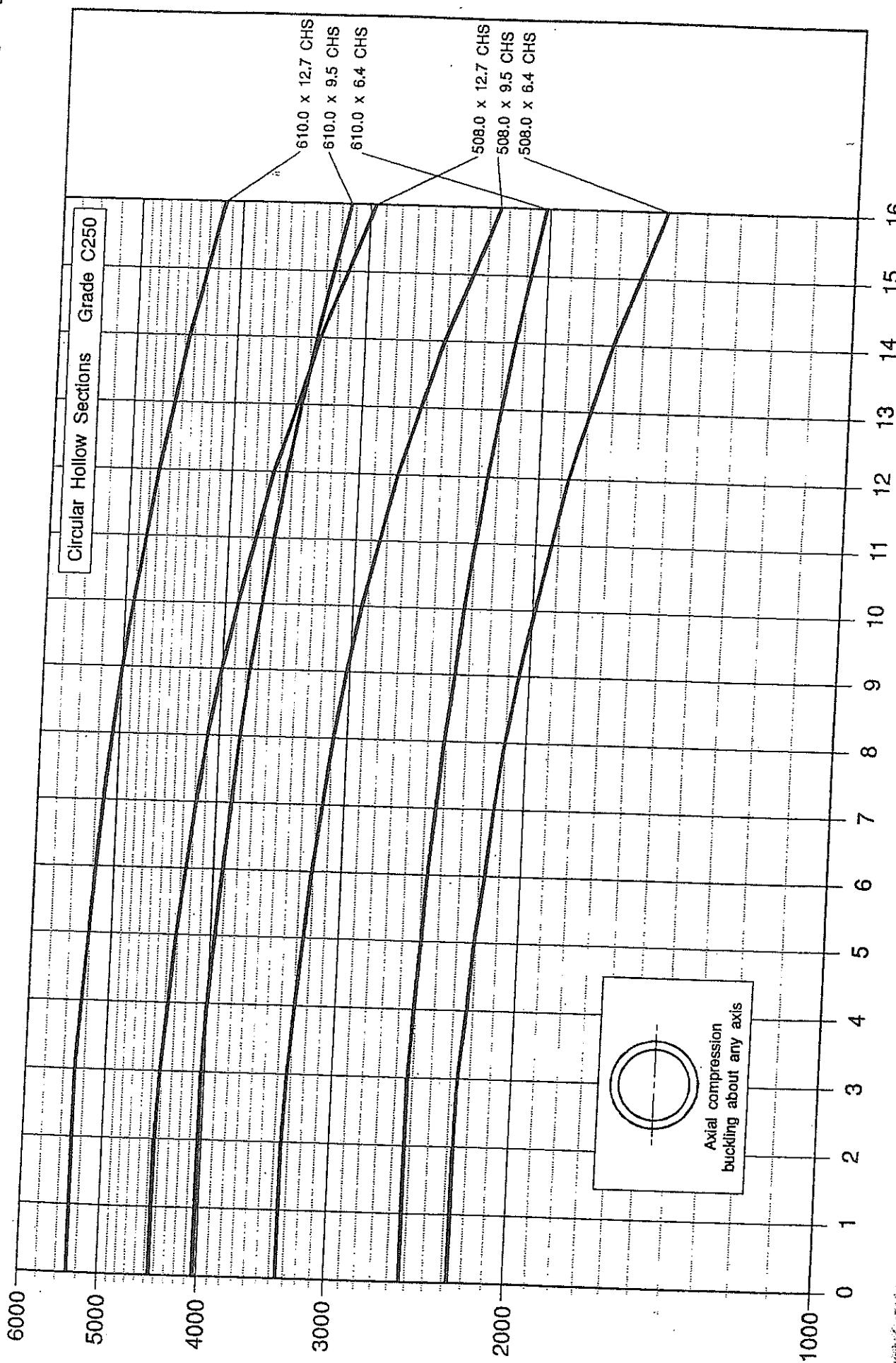
**DESIGN CAPACITIES FOR MEMBERS  
SUBJECT TO AXIAL COMPRESSION  
CIRCULAR HOLLOW SECTIONS: GRADE C250**

buckling about any axis



Designation $d_o$ mm	$t$ mm	Mass per m kg/m	Design Capacities for Axial Compression $\phi N_c$ (kN)												
			0.0	1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	9.0	10.0	12.0	14.0
610.0 x 12.7 CHS	187	5360	5360	5360	5310	5260	5210	5150	5080	5000	4910	4700	4450	4150	
	141	4030	4030	4030	4030	3960	3920	3870	3820	3760	3680	3540	3350	3130	
	9.5 CHS	2530	2530	2530	2530	2510	2490	2470	2440	2410	2370	2340	2250	2140	2020
508.0 x 12.7 CHS	155	4450	4450	4450	4420	4370	4320	4250	4180	4100	4000	3890	3640	3320	2960
	9.5 CHS	117	3350	3350	3350	3350	3290	3250	3210	3150	3090	3020	2940	2750	2510
	6.4 CHS	79.2	2270	2270	2270	2260	2230	2210	2170	2140	2100	2050	1990	1870	1710
Notes:														1530	

Notes:  
1.  $\phi = 0.9$   
2.  $\phi N_c = \phi c_c N_s$  (Clause 6.3.3 of AS4100)  
3.  $c_b = -0.5$  (Table 6.3.3 of AS4100)



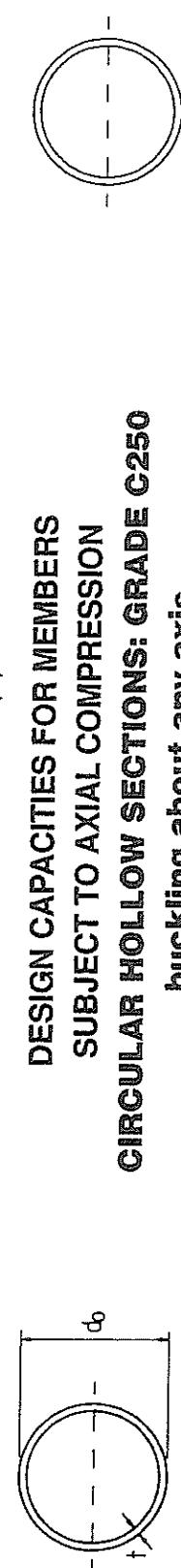
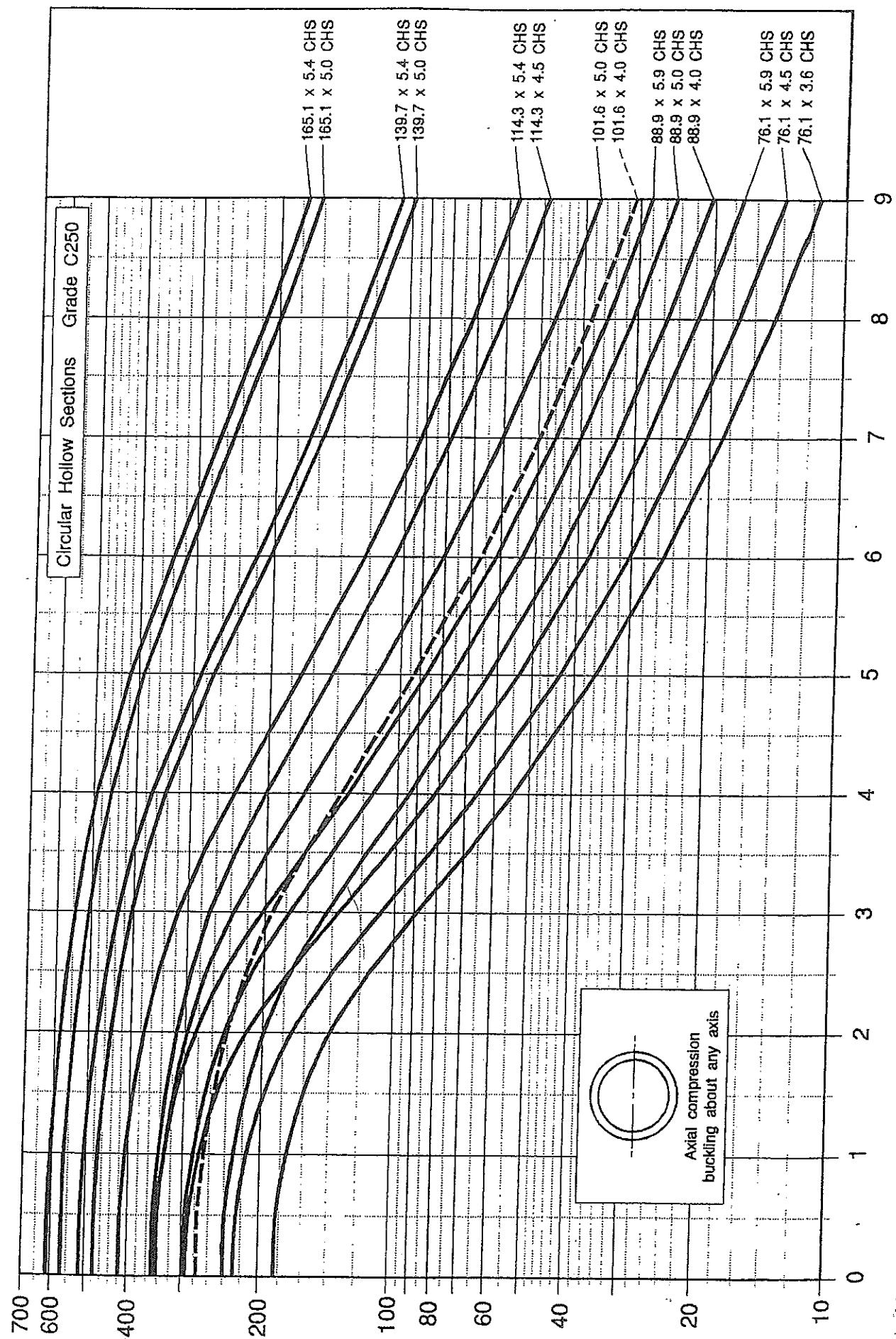


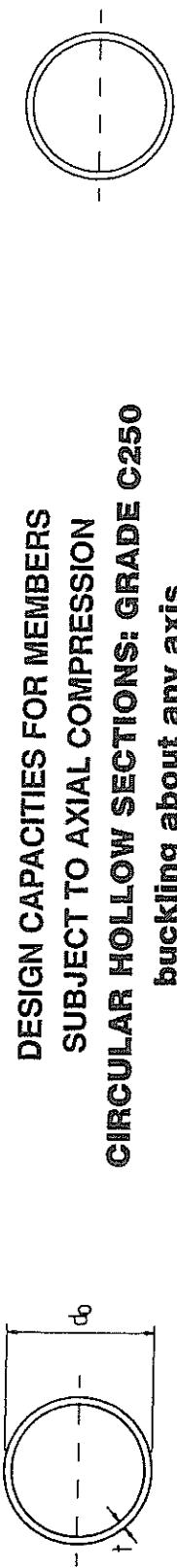
TABLE 5.2 - 1 (2)

**DESIGN CAPACITIES FOR MEMBERS  
SUBJECT TO AXIAL COMPRESSION  
CIRCULAR HOLLOW SECTIONS: GRADE C250  
buckling about any axis**

Designation $d_o$ mm	Mass per m kg/m	Design Capacities for Axial Compression $\phi N_c$ (kN)											
		0.0	0.5	1.0	2.0	2.5	3.0	3.5	4.0	5.0	6.0	7.0	8.0
165.1 x 5.4 CHS	21.3	610	610	605	595	581	565	544	520	490	418	338	214
5.0 CHS	19.7	566	566	562	552	540	524	506	483	456	389	315	199
139.7 x 5.4 CHS	17.9	513	513	506	494	479	459	434	404	375	342	288	167
5.0 CHS	16.6	476	476	470	459	445	426	403	375	342	269	204	156
114.3 x 5.4 CHS	14.5	416	416	406	392	373	348	316	278	238	170	123	131
4.5 CHS	12.2	349	349	341	330	314	293	267	235	202	144	105	105
101.6 x 5.0 CHS	11.9	341	341	331	317	297	270	236	199	165	113	80.8	46.6
4.0 CHS	9.63	276	276	268	256	241	219	193	163	135	92.9	68.5	49.6
88.9 x 5.0 CHS	12.1	346	344	331	312	284	246	203	163	130	86.9	61.5	45.7
5.0 CHS	10.3	297	295	284	268	244	213	176	141	114	75.9	53.7	40.0
4.0 CHS	8.38	240	239	230	217	199	174	144	117	93.7	62.7	44.4	33.0
76.1 x 5.9 CHS	10.2	293	289	275	252	218	175	135	104	81.7	53.6	37.8	28.0
4.5 CHS	7.95	228	225	215	197	172	139	108	83.5	65.7	43.2	30.4	22.6
3.6 CHS	6.44	184	183	174	160	140	114	89.0	69.0	54.4	35.8	25.2	18.7

Notes:  
1.  $\phi = 0.9$   
2.  $\phi N_c = \phi \alpha_{eff} N_s$   
3.  $\alpha_{eff} = 0.5$   
(Clause 6.3.3 of AS4100)  
(Table 6.3.3 of AS4100)

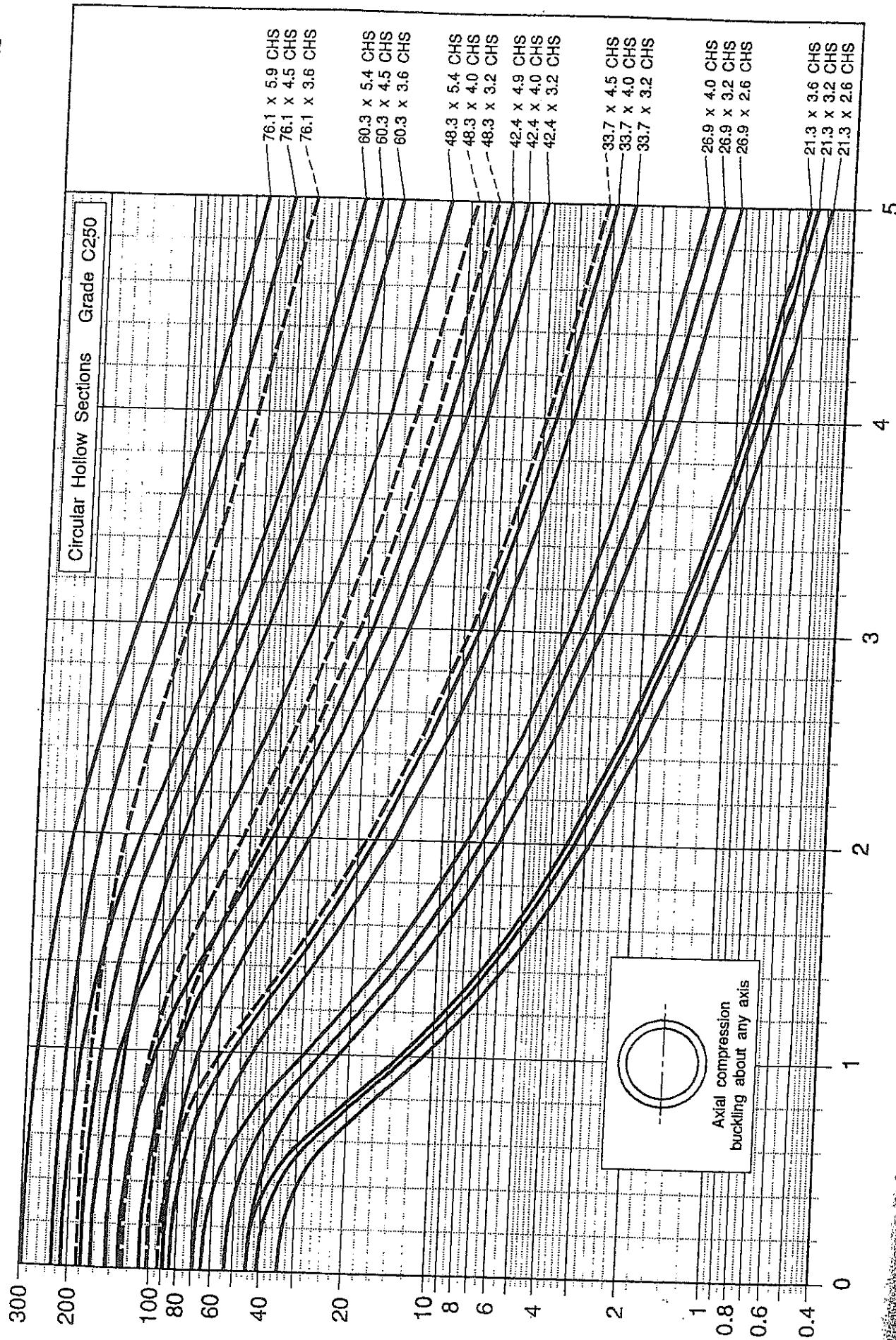
Design Member Capacity  $N_e$  (kN)@Seismicisolation  
AISC: DESIGN CAPACITY TABLES  
FOR STRUCTURAL STEEL HOLLOW SECTIONS



**TABLE 5.2 – 1 (3)**  
**DESIGN CAPACITIES FOR MEMBERS**  
**SUBJECT TO AXIAL COMPRESSION**  
**CIRCULAR HOLLOW SECTIONS: GRADE C250**  
**buckling about any axis**

Designation $d_o$ mm	Mass per m kg/m	Design Capacities for Axial Compression $\phi N_c$ (kN)												
		0.0	0.25	0.5	1.0	1.25	1.5	1.75	2.0	2.5	3.0	3.5	4.0	5.0
76.1 x 5.9 CHS	10.2	298	293	289	283	275	265	252	236	218	175	135	104	81.7
4.5 CHS	7.95	228	228	225	221	215	207	197	186	172	139	105	83.5	53.6
3.6 CHS	6.44	184	184	183	179	174	168	160	151	140	114	89.0	69.0	45.7
60.3 x 5.4 CHS	7.31	210	210	205	198	189	176	161	142	122	87.2	63.3	47.5	36.9
4.5 CHS	6.19	177	177	174	168	160	150	137	122	105	75.7	55.0	41.4	32.1
3.6 CHS	5.03	144	144	141	137	131	123	113	100	87.1	63.0	46.0	34.6	26.8
48.3 x 5.4 CHS	5.71	163	163	157	149	137	120	101	81.6	65.8	44.2	31.3	23.3	18.0
4.0 CHS	4.37	125	125	121	114	106	93.9	79.5	65.1	52.8	35.6	25.3	16.8	14.5
3.2 CHS	3.56	102	102	102	98.4	93.5	86.7	77.3	65.9	54.3	44.2	28.9	21.2	15.8
42.4 x 4.9 CHS	4.53	130	129	123	114	101	84.3	66.7	52.2	41.3	27.2	19.2	14.3	11.0
4.0 CHS	3.79	109	108	103	96.1	85.7	72.0	57.5	45.2	35.8	23.7	16.7	12.4	9.58
3.2 CHS	3.09	88.7	88.0	84.4	78.9	70.7	59.9	48.2	38.0	30.2	20.0	14.2	10.5	8.11
33.7 x 4.5 CHS	3.24	92.9	91.1	84.8	74.2	58.7	43.2	31.7	24.0	18.6	12.1	8.52	6.31	4.86
4.0 CHS	2.93	84.0	82.5	76.9	67.6	53.9	40.0	29.4	22.2	17.3	11.3	7.92	5.87	4.52
3.2 CHS	2.41	69.0	67.8	63.5	56.2	45.4	34.0	25.2	19.1	14.8	9.69	6.81	5.04	3.88
26.9 x 4.0 CHS	2.26	64.7	62.6	55.6	43.1	29.3	20.0	14.2	10.6	8.19	5.31	3.72	2.75	2.12
3.2 CHS	1.87	53.6	51.9	46.4	36.7	25.4	17.4	12.4	9.27	7.17	4.65	3.26	2.41	1.85
2.6 CHS	1.56	44.7	43.3	38.9	31.2	21.9	15.1	10.8	8.05	6.22	4.04	2.83	2.09	1.61
21.3 x 3.6 CHS	1.57	45.0	42.5	34.1	21.6	13.2	8.66	6.10	4.53	3.49	2.26	1.58	1.17	0.897
3.2 CHS	1.43	40.9	38.7	31.4	20.2	12.4	8.14	5.74	4.26	3.28	2.12	1.49	1.10	0.844
2.6 CHS	1.20	34.4	32.6	26.8	17.6	10.9	7.19	5.07	3.76	2.90	1.88	1.31	0.971	0.747

Notes:  
1.  $\phi = 0.9$   
2.  $\phi N_c = \phi c N_s$  (Clause 6.3.3 of AS4100)  
3.  $d_o = -0.5$  (Table 6.3.3 of AS4100)



@Seismicisolation

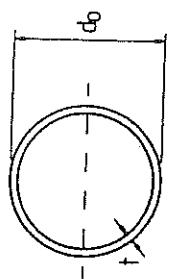
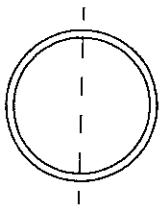


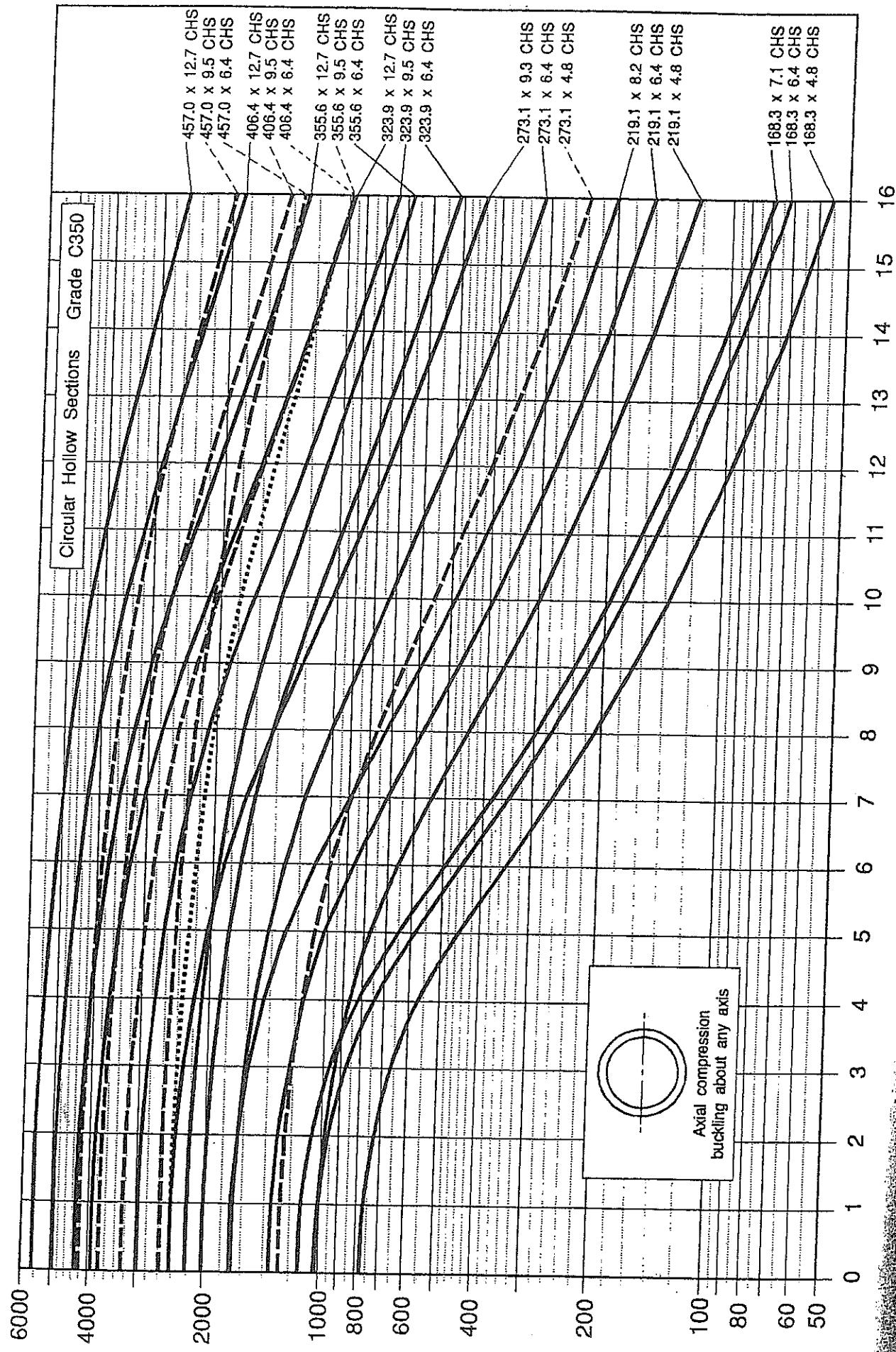
TABLE 5.2 - 2 (1)

**DESIGN CAPACITIES FOR MEMBERS  
SUBJECT TO AXIAL COMPRESSION  
CIRCULAR HOLLOW SECTIONS: GRADE C350  
buckling about any axis**



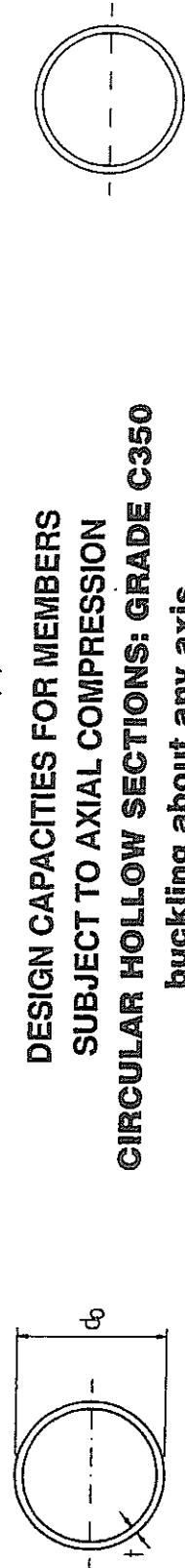
Designation $d_o$ mm	Mass per m kg/m	Design Capacities for Axial Compression $\phi N_c$ (kN)													
		0.0	1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	9.0	10.0			
457.0 x 12.7 CHS 9.5 CHS 6.4 CHS	139 105 71.1	5580 4210 2580	5570 4200 2580	5490 4140 2550	5400 4070 2510	5220 3990 2460	5150 3890 2410	4990 3770 2340	4810 3630 2270	4590 3470 2180	4340 3280 2080	3750 3130 1840	3130 2880 1570	2870 1960 1310	
	123	4950	4950	4920	4840	4740	4610	4460	4280	4060	3810	3520	2900	2320	1860
	93.0 63.1	3730 2430	3710 2420	3650 2380	3580 2340	3480 2280	3370 2210	3230 2130	3070 2030	2890 1910	2670 1780	2210 1420	2110	1420	
355.6 x 12.7 CHS 9.5 CHS 6.4 CHS	107 81.1 55.1	4310 3250 2210	4270 3220 2190	4180 3160 2150	4070 3070 2080	3920 2970 2020	3750 2840 1930	3530 2660 1830	3270 2490 1760	2980 2270 1550	2670 2030 1400	2080 1590 1090	1610 1230 852	1210 973 672	
	97.5	3910	3910	3860	3760	3640	3480	3280	3040	2750	2440	2140	1610	1230	
	73.7 50.1	2960 2010	2920 1980	2850 1940	2750 1880	2640 1800	2490 1700	2310 1580	2160 1440	1870 1280	1640 1130	1240 856	946 655	740 512	
273.1 x 9.3 CHS 6.4 CHS 4.8 CHS	60.5 42.1 31.8	2430 1680 1270	2450 1650 1250	2380 1600 1210	2300 1530 1150	2190 1430 1080	2050 1430 984	1870 1310 985	1660 1170 767	1430 1010 654	1220 1030 556	1030 728 404	748 530 304	961 740 436	
	8.2 CHS 6.4 CHS 4.8 CHS	42.6 33.6 25.4	1710 1350 1020	1700 1340 1010	1650 1300 982	1570 1230 934	1450 1140 865	1290 1020 773	1090 865 659	893 711 543	724 578 442	591 472 362	489 391 299	347 278 213	562 399 304
	7.1 CHS 6.4 CHS 4.8 CHS	28.2 25.6 19.4	1130 1030 777	1120 1010 728	1060 960 664	964 874 664	822 746 570	649 591 451	380 346 347	297 271 267	238 217 209	195 178 168	137 125 137	137 125 96.6	102 92.8 71.6

Notes:  
 1.  $\phi = 0.9$   
 2.  $\phi N_c = \phi \alpha_c N_s$  (Clause 6.3.3 of AS 4100)  
 3.  $\alpha_b = -0.5$  (Table 6.3.3 of AS 4100)



Design Member Capacity  $f_{Nc}$  (KN)

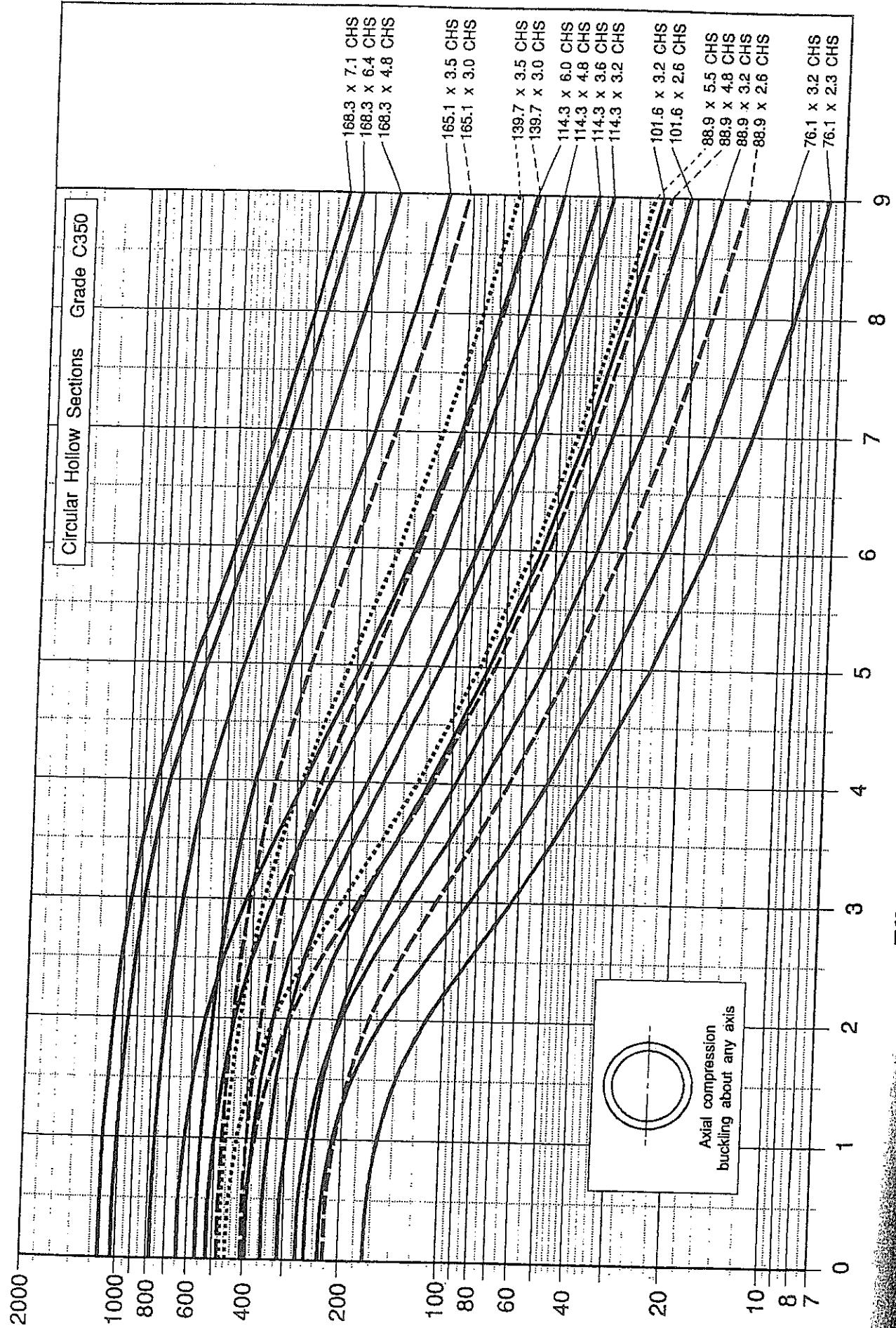
@Seismicisolation



**TABLE 5.2 – 2 (2)**  
**DESIGN CAPACITIES FOR MEMBERS**  
**SUBJECT TO AXIAL COMPRESSION**  
**CIRCULAR HOLLOW SECTIONS: GRADE C350**  
**buckling about any axis**

Designation $d_o$ mm	$t$ mm	Mass per m kg/m	Design Capacities for Axial Compression $\phi N_c$ (kN)							
			0.0	0.5	1.0	1.5	2.0	2.5	3.0	3.5
168.3 x 7.1 CHS	28.2	1130	1120	1090	1060	1020	964	899	822	649
6.4 CHS	25.6	1030	1010	990	960	922	874	816	746	591
4.8 CHS	19.4	777	757	750	728	700	664	621	570	453
165.1 x 3.5 CHS	13.9	560	553	540	524	503	477	445	407	322
3.0 CHS	12.0	481	481	475	465	451	433	410	383	351
139.7 x 3.5 CHS	11.8	472	472	462	448	429	403	371	332	289
3.0 CHS	10.1	406	406	398	386	369	348	320	287	250
114.3 x 6.0 CHS	16.0	643	641	621	591	549	492	422	349	285
4.8 CHS	13.0	520	519	503	479	446	401	345	286	234
3.6 CHS	9.83	394	393	381	364	339	306	264	220	181
3.2 CHS	8.77	352	351	340	325	303	273	237	197	162
101.6 x 3.2 CHS	7.77	312	310	298	281	256	222	183	146	117
2.6 CHS	6.35	255	253	244	230	210	182	150	121	96.8
88.9 x 5.5 CHS	11.3	454	449	427	391	338	272	210	162	127
4.8 CHS	9.96	399	395	376	345	299	242	187	145	114
3.2 CHS	6.76	271	269	256	236	206	168	131	101	74.7
2.6 CHS	5.53	222	220	210	193	169	138	108	83.9	66.1
76.1 x 3.2 CHS	5.75	231	227	213	188	153	115	85.1	64.5	50.2
2.3 CHS	4.19	168	165	155	138	113	85.0	63.2	47.9	37.4

Notes:  
1.  $\phi = 0.9$   
2.  $\phi N_c = \phi \alpha_c N_s$   
3.  $c_b = -0.5$   
(Clause 6.3.3 of AS 4100)  
(Table 6.3.3 of AS 4100)



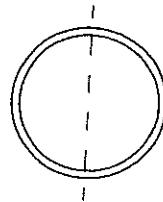
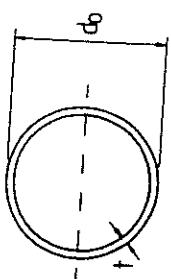
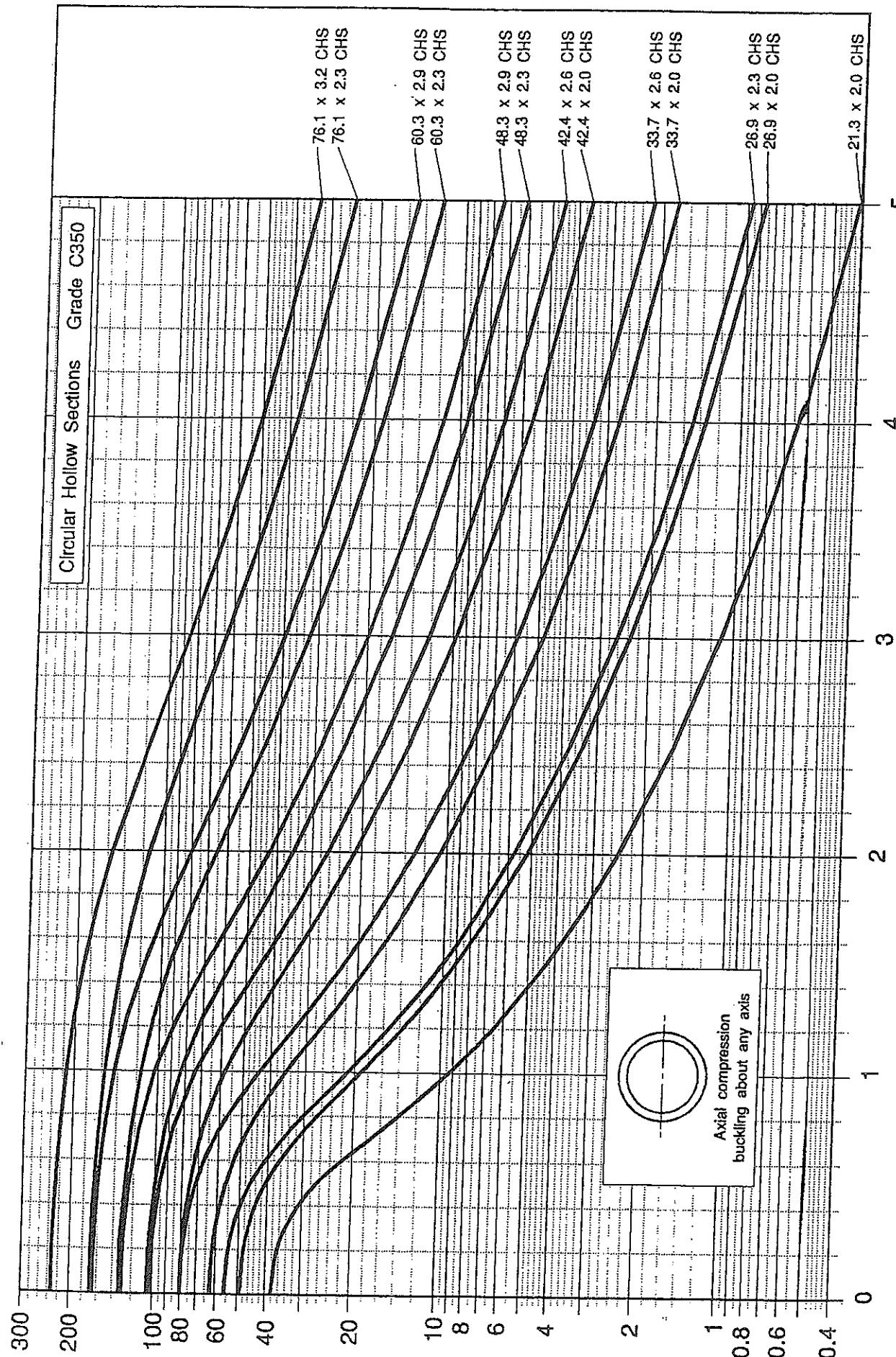


TABLE 5.2 - 2 (3)  
**DESIGN CAPACITIES FOR MEMBERS  
SUBJECT TO AXIAL COMPRESSION  
CIRCULAR HOLLOW SECTIONS: GRADE C350  
buckling about any axis**



Designation $d_o$ mm	Mass per m kg/m	Design Capacities for Axial Compression $\phi N_c$ (kN)						
		0.0	0.25	0.5	0.75	1.0	1.25	1.5
76.1x32 CHS 2.3 CHS	5.75 4.19	231 168	227 165	221 161	202 155	188 147	172 138	153 126
60.3x29 CHS 2.3 CHS	4.11 3.29	164 132	160 128	153 123	143 115	131 105	114 92.5	96.6 78.4
48.3x29 CHS 2.3 CHS	3.25 2.61	129 105	124 104	115 99.6	103 92.9	83.1 70.0	68.5 56.1	80.0 65.1
42.4x26 CHS 2.0 CHS	2.55 1.99	102 80.0	101 79.0	95.6 74.8	86.8 66.1	73.6 58.1	57.8 46.0	42.6 35.1
33.7x26 CHS 2.0 CHS	1.99 1.56	80.0 62.7	78.0 61.3	71.3 56.1	59.5 47.2	43.8 35.2	30.9 24.9	22.3 18.0
26.9x23 CHS 2.0 CHS	1.40 1.23	56.0 49.3	53.6 47.3	46.0 40.7	33.0 29.4	21.2 19.0	14.2 12.7	10.0 9.01
21.3x20 CHS	0.932	38.2	35.5	26.8	15.6	9.33	6.10	4.29

Notes:  
1.  $\phi = 0.9$   
2.  $\phi N_c = \phi \alpha_b N_a$  (Clause 6.3.3 of AS 4100)  
3.  $\alpha_b = -0.5$  (Table 6.3.3 of AS 4100)



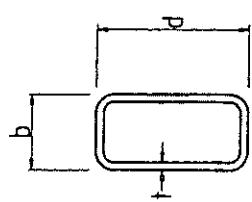
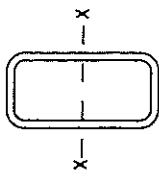


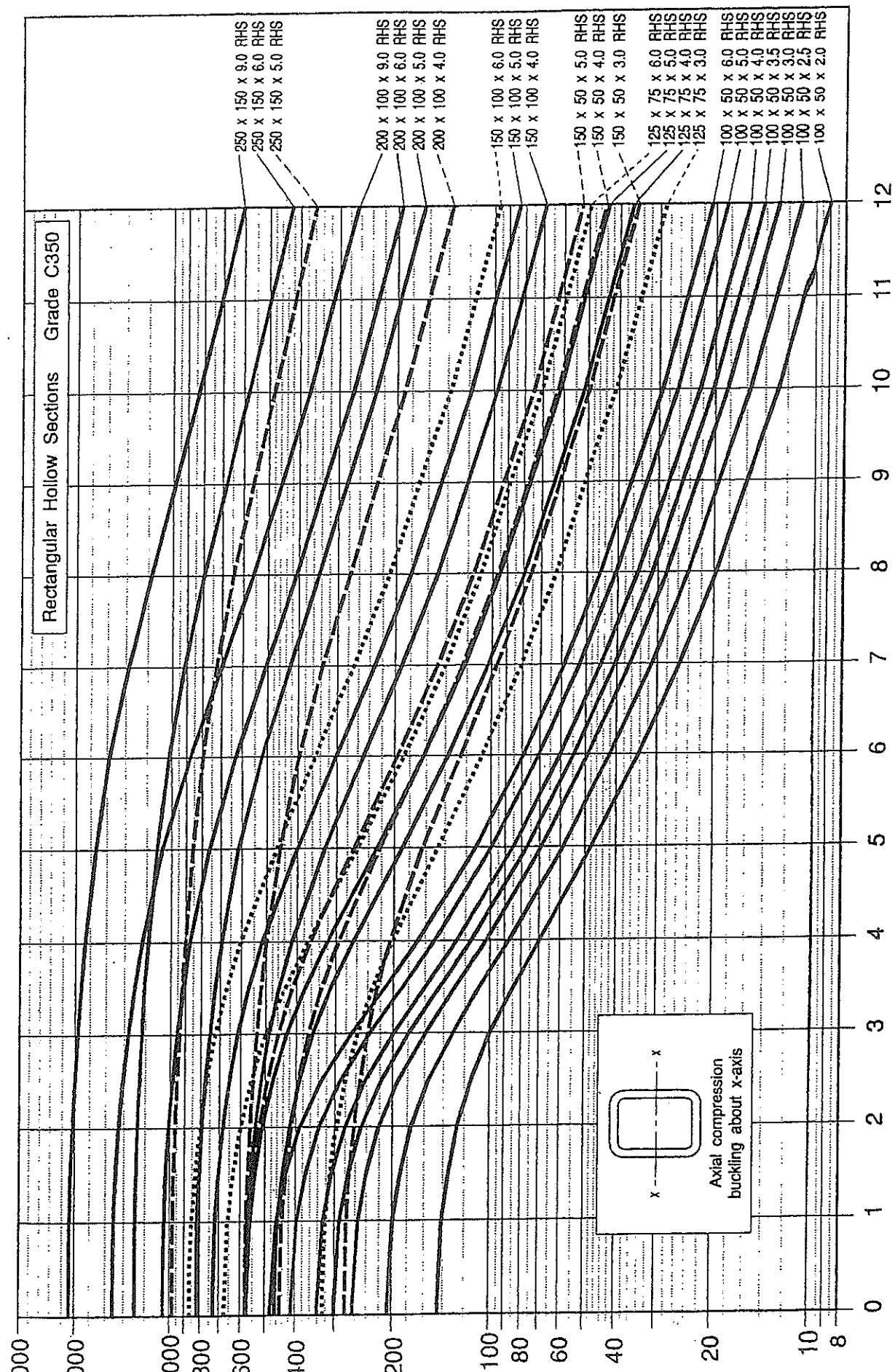
TABLE 5.2 - 3 (1)(A)

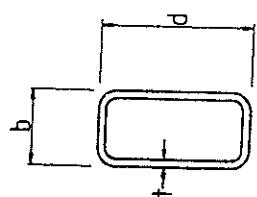
**DESIGN CAPACITIES FOR MEMBERS  
SUBJECT TO AXIAL COMPRESSION  
RECTANGULAR HOLLOW SECTIONS: GRADE C350  
buckling about x-axis**



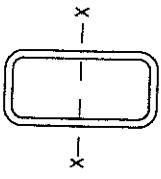
Designation d mm mm mm	Mass per m kg/m	Effective length ( $L_e$ ) in metres	Design Capacities for Axial Compression $\phi N_c$ (kN)										
			0.0	1.0	2.0	2.5	3.0	3.5	4.0	5.0	6.0	7.0	8.0
250 x 150 x 9.0 RHS	51.8	2080	2080	2030	2000	1960	1910	1860	1730	1570	1370	1170	832
6.0 RHS	35.6	1280	1280	1250	1230	1210	1180	1110	1020	914	797	581	425
5.0 RHS	29.9	978	978	972	961	949	935	918	900	854	796	725	644
200 x 100 x 9.0 RHS	37.7	1510	1500	1480	1450	1410	1360	1300	1230	1070	873	698	558
6.0 RHS	26.2	1050	1040	1030	1010	981	950	913	868	758	628	506	372
5.0 RHS	22.1	820	817	805	790	772	750	723	692	614	520	425	272
4.0 RHS	17.9	575	574	567	558	547	533	516	500	454	396	334	232
150 x 100 x 6.0 RHS	21.4	861	849	828	800	765	721	666	603	468	351	267	209
5.0 RHS	18.2	729	719	702	679	650	613	568	515	400	302	231	180
4.0 RHS	14.8	575	568	555	538	516	489	455	415	327	249	191	149
150 x 50 x 6.0 RHS	14.2	571	561	544	522	493	457	412	362	266	196	148	115
4.0 RHS	11.6	449	442	430	413	392	365	332	295	221	163	124	96.2
3.0 RHS	8.96	279	276	270	262	251	239	224	205	164	125	96.5	75.7
125 x 75 x 6.0 RHS	16.7	672	655	631	598	555	500	435	368	259	187	140	109
5.0 RHS	14.2	571	558	538	511	475	429	375	319	226	163	122	94.9
4.0 RHS	11.6	467	456	440	418	390	354	311	266	189	137	103	79.5
3.0 RHS	8.96	326	320	310	297	279	257	231	201	147	107	80.9	62.9
100 x 50 x 6.0 RHS	12.0	483	460	431	388	331	268	212	169	112	79.3	58.9	45.4
5.0 RHS	10.3	414	396	371	336	289	236	188	150	100	70.7	52.6	39.6
4.0 RHS	8.49	341	326	307	279	242	199	160	128	85.3	60.4	44.9	34.7
3.5 RHS	7.53	302	290	273	249	216	179	144	115	77.0	54.5	40.6	31.3
3.0 RHS	6.60	265	254	240	220	192	159	129	103	69.3	49.1	36.5	28.2
2.5 RHS	5.56	207	199	189	175	155	131	108	87.4	58.9	41.9	31.2	24.1
2.0 RHS	4.50	145	141	134	126	114	100	84.3	69.6	47.8	34.1	25.5	19.7

Notes:  
 1.  $\phi = 0.9$   
 2.  $\phi N_c = \phi_{\text{ax}} N_s$  (Clause 6.3.3 of AS 4100)  
 3.  $a_b = -0.5$  (Table 6.3.3 of AS 4100)



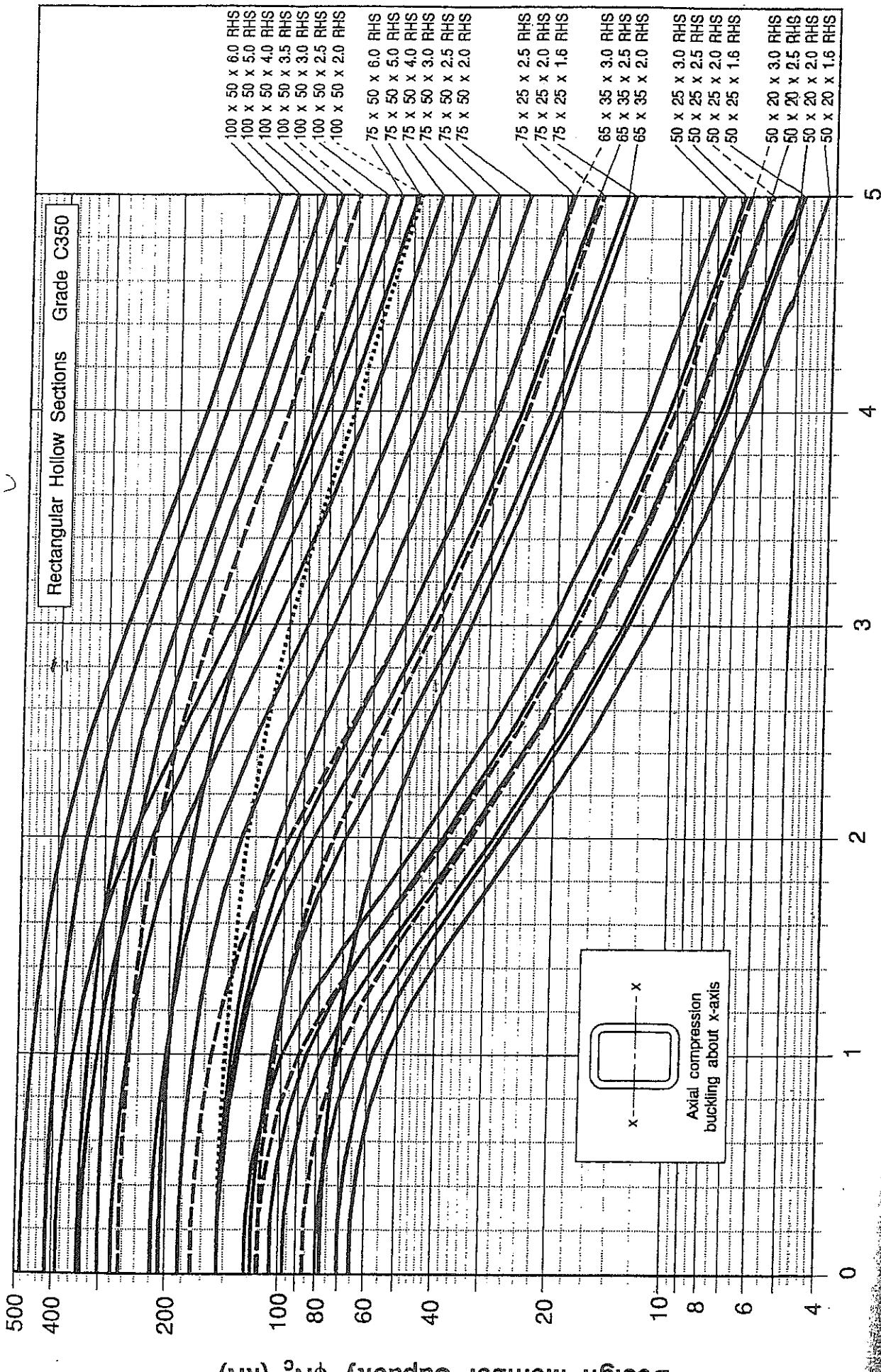


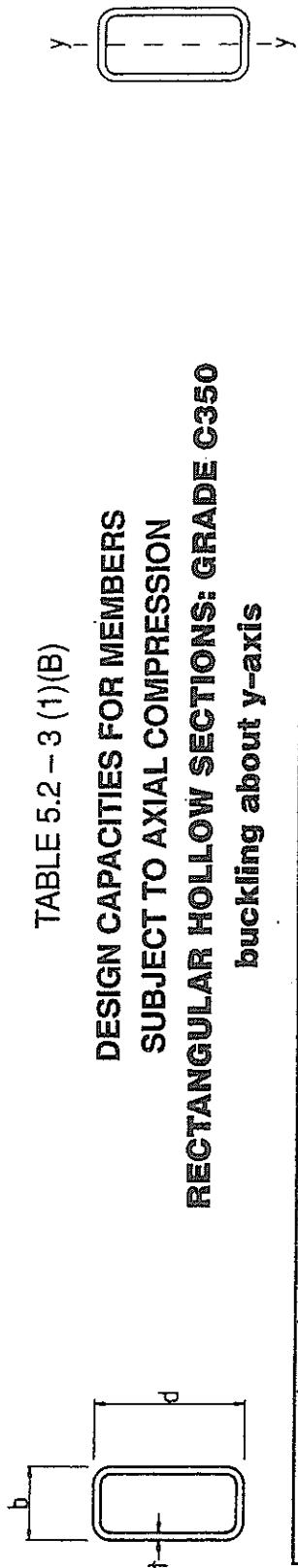
**TABLE 5.2 – 3 (2)(A)**  
**DESIGN MEMBER CAPACITIES FOR MEMBERS**  
**SUBJECT TO AXIAL COMPRESSION**  
**RECTANGULAR HOLLOW SECTIONS: GRADE C350**  
**buckling about x-axis**



Designation d   b   t mm mm mm	Mass per m kg/m	Design Member Capacities for Axial Compression $\phi N_c$ (kN)														
		0.0	0.25	0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5	3.0	3.5	4.0	5.0	
100 x 50 x 6.0 RHS	12.0	483	483	479	471	460	447	431	411	388	331	268	212	169	112	
5.0 RHS	10.3	414	414	411	404	396	385	371	355	336	289	236	188	150	100	
4.0 RHS	8.49	341	341	339	333	326	317	307	294	279	242	199	160	128	85.3	
3.5 RHS	7.53	302	302	300	296	290	282	273	262	249	216	179	144	115	77.0	
3.0 RHS	6.60	265	265	263	259	254	248	240	231	220	192	159	129	103	69.3	
2.5 RHS	5.56	207	207	206	203	199	195	189	182	175	155	131	108	87.4	58.9	
2.0 RHS	4.50	145	145	143	143	141	138	134	130	126	114	100	84.3	69.6	47.8	
75 x 50 x 6.0 RHS	9.67	388	388	382	371	357	338	315	296	254	189	140	106	82.4	53.8	
5.0 RHS	8.35	335	335	330	321	309	294	275	252	225	170	126	95.7	74.6	48.7	
4.0 RHS	6.92	278	278	273	266	257	245	230	212	190	145	109	82.8	64.6	42.2	
3.0 RHS	5.42	218	218	215	209	202	194	183	169	153	119	89.4	68.2	53.3	34.9	
2.5 RHS	4.38	184	184	181	177	171	164	155	144	130	102	76.8	56.7	45.9	30.0	
2.0 RHS	3.72	145	145	143	140	136	130	123	115	105	82.7	62.9	48.3	37.8	24.8	
75 x 25 x 2.5 RHS	3.60	145	142	138	132	125	116	105	92.2	68.2	50.2	37.9	29.5	19.2	16.0	
2.0 RHS	2.93	113	113	112	108	104	99.1	92.5	84.3	74.9	56.2	41.6	31.5	24.5	13.2	
1.6 RHS	2.38	77.6	77.6	76.7	74.9	72.6	69.6	65.9	61.4	56.0	44.0	33.5	25.6	20.1	13.8	
65 x 35 x 3.0 RHS	4.25	170	166	161	153	143	130	111	97.9	97.5	69.4	50.3	37.7	29.3	19.0	
3.60	145	141	137	130	122	111	97.9	84.1	60.2	43.7	32.8	25.4	16.5	10.5	7.68	
2.93	118	118	115	111	105	99.6	91.0	80.6	69.6	50.0	36.3	27.3	21.2	13.8	9.79	
50 x 25 x 3.0 RHS	3.07	123	122	118	110	99.6	85.4	69.4	55.2	44.0	29.3	20.7	15.4	11.9	8.89	
2.62	105	105	101	94.6	85.9	74.1	60.7	48.5	38.8	25.9	18.3	13.6	10.5	7.67	5.76	
2.0 RHS	2.15	86.2	85.7	82.6	77.8	70.9	61.6	50.9	40.8	32.7	21.9	15.5	11.5	8.89	7.44	4.82
1.6 RHS	1.75	70.3	69.9	67.4	63.6	58.2	50.8	42.2	34.0	27.3	18.3	13.0	9.63	7.44	4.82	4.82
50 x 20 x 3.0 RHS	2.83	114	113	108	101	90.0	75.8	60.6	47.7	37.8	25.0	17.7	13.1	10.1	6.56	
2.42	97.3	95.6	92.7	86.7	77.9	66.2	53.4	42.2	33.6	22.3	15.7	11.7	9.02	5.84	4.97	
2.0 RHS	1.99	79.9	79.4	76.3	71.5	64.6	55.3	45.0	35.7	28.5	19.0	13.4	9.96	7.68	4.97	4.18
1.6 RHS	1.63	65.3	64.9	62.4	58.6	53.2	45.8	37.5	29.9	23.9	15.9	11.3	8.37	6.46	4.18	4.18

Notes:  
1.  $\phi N_c = 0.9$   
2.  $\phi N_c = \phi_{\alpha} N_c$   
3.  $\phi_{\alpha} = -0.5$  (Clause 6.3.3 of AS 4100)  
(Table 6.3.3 of AS 4100)





**TABLE 5.2 – 3 (1)(B)**

**DESIGN CAPACITIES FOR MEMBERS  
SUBJECT TO AXIAL COMPRESSION**

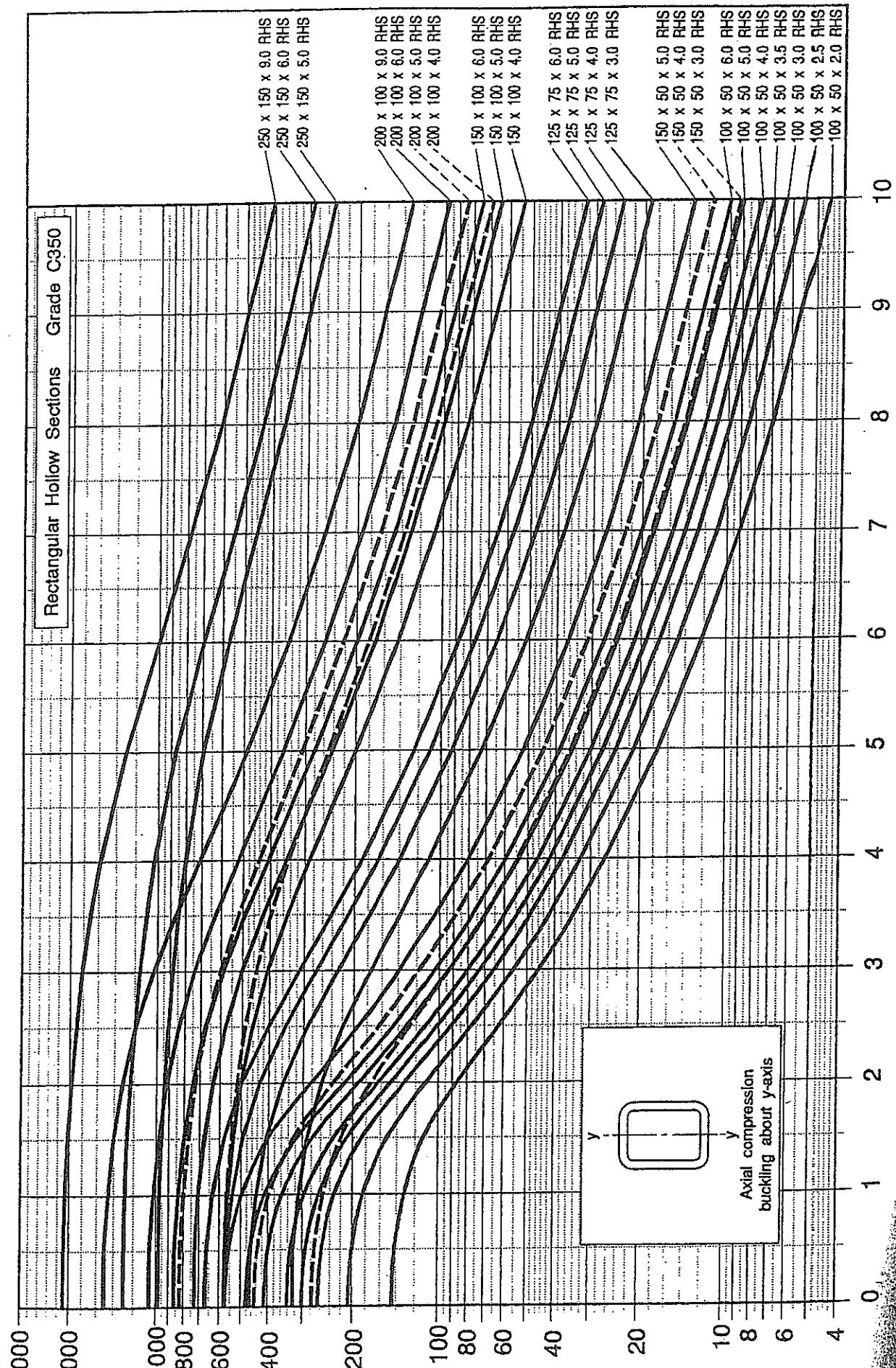
**RECTANGULAR HOLLOW SECTIONS: GRADE C350**

**buckling about y-axis**

Designation <i>d</i> <i>b</i> <i>t</i>	Mass per m kg/m	Design Capacities for Axial Compression $\phi N_c$ (kN)													
		0.0	0.5	1.0	2.0	2.5	3.0	3.5	4.0	5.0	6.0	7.0	8.0	10.0	
250 x 150 x 9.0 RHS	51.8	2080	2080	2060	2020	1960	1890	1810	1700	1580	1290	1000	778	612	403
6.0 RHS	35.6	1290	1290	1280	1260	1230	1190	1150	1050	1030	868	695	548	435	288
5.0 RHS	29.9	978	978	972	956	936	911	881	845	801	694	571	457	366	244
200 x 100 x 9.0 RHS	37.7	1510	1510	1460	1400	1310	1190	1030	866	713	488	348	260	201	130
6.0 RHS	26.2	1050	1050	977	918	840	740	629	523	361	258	193	149	96.9	
5.0 RHS	22.1	820	820	799	769	728	673	602	521	440	308	222	166	128	83.4
4.0 RHS	17.9	575	575	563	545	521	489	448	398	344	249	181	136	106	68.8
150 x 100 x 6.0 RHS	21.4	861	859	834	797	745	676	588	494	407	279	199	148.	115	74.4
5.0 RHS	18.2	729	728	707	676	634	576	504	425	351	241	172	129	99.5	64.5
4.0 RHS	14.8	575	575	559	536	504	461	407	347	289	200	143	107	82.6	53.6
150 x 50 x 5.0 RHS	14.2	571	555	499	401	282	194	139	104	80.3	52.1	36.5	27.0	20.8	13.4
4.0 RHS	11.6	449	427	397	326	236	164	118	88.3	68.3	44.4	31.1	23.0	17.7	11.4
3.0 RHS	8.96	279	273	254	221	174	127	93.0	70.1	54.5	35.5	24.9	18.5	14.2	9.18
125 x 75 x 6.0 RHS	16.7	672	684	632	579	502	404	312	241	189	124	87.7	65.0	50.1	32.4
5.0 RHS	14.2	571	566	539	496	432	351	272	211	166	109	76.9	57.1	44.0	28.5
4.0 RHS	11.6	467	462	441	407	356	292	228	177	139	91.9	64.7	48.0	37.0	24.0
3.0 RHS	8.96	326	324	310	259	219	175	138	110	72.7	51.3	38.1	29.4	19.0	
100 x 50 x 6.0 RHS	12.0	483	466	413	318	215	146	104	77.4	59.8	38.8	27.2	20.1	15.5	9.97
5.0 RHS	10.3	414	367	279	192	131	93.3	69.6	53.8	34.9	24.4	18.1	13.9	8.97	
4.0 RHS	8.49	341	330	296	235	164	112	80.3	59.9	46.3	30.1	21.1	15.6	12.0	7.73
3.5 RHS	7.53	302	283	211	148	102	72.8	54.3	42.0	27.3	19.1	14.1	10.9	7.02	
3.0 RHS	6.60	265	257	232	187	132	91.4	65.4	48.9	37.8	24.5	17.2	12.7	8.32	
2.5 RHS	5.56	207	201	184	152	111	77.8	56.0	41.9	32.5	21.1	14.8	10.9	8.42	
2.0 RHS	4.50	145	142	131	113	86.8	62.7	45.7	34.3	26.7	17.3	12.2	9.01	6.94	4.48

Notes:

1.  $\phi = 0.9$
2.  $\phi N_c = \phi \alpha_{\text{c}} N_s$   
(Clause 6.3.3 of AS4100)
3.  $\phi_b = -0.5$   
(Table 6.3.3 of AS4100)



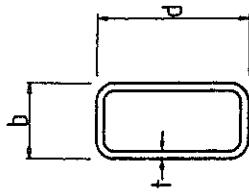


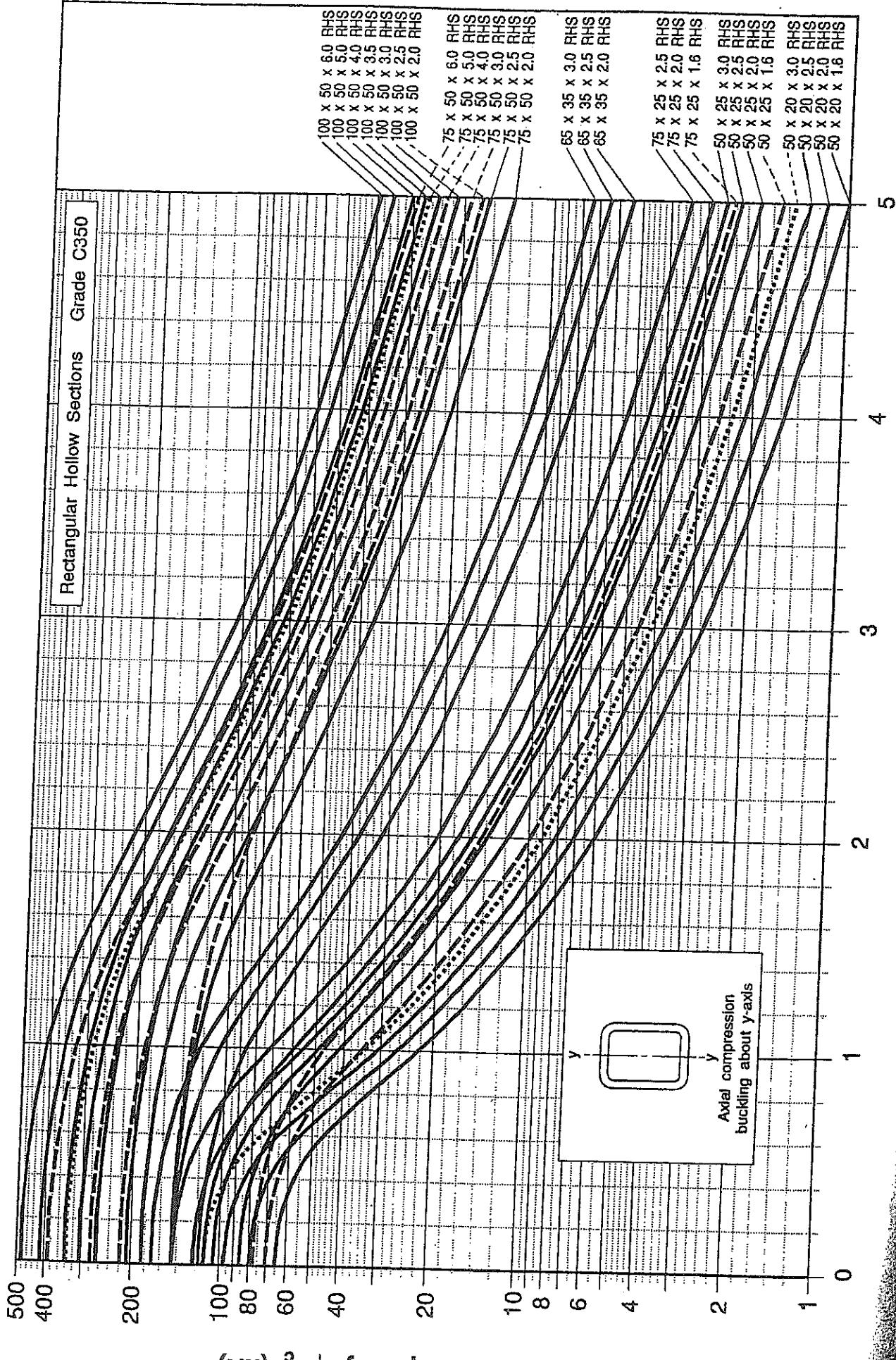
TABLE 5.2 – 3 (2)(B)

**DESIGN CAPACITIES FOR MEMBERS  
SUBJECT TO AXIAL COMPRESSION  
RECTANGULAR HOLLOW SECTIONS: GRADE C350  
buckling about y-axis**

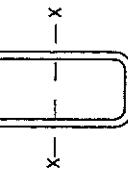
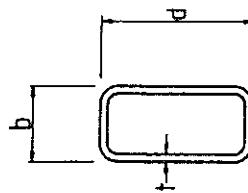


Designation d mm b mm t mm	Mass per m kg/m	Effective Length [L <sub>e</sub> ] in metres										Design Capacities for Axial Compression $\phi N_c$ (kN)
		0.0	0.25	0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5	
100 x 50 x 6.0 RHS	12.0	483	481	444	413	370	318	263	215	146	104	77.4
5.0 RHS	10.3	414	413	382	357	322	279	233	192	131	93.3	59.8
4.0 RHS	8.49	341	340	330	316	296	269	235	198	164	112	69.6
3.5 RHS	7.53	302	302	293	281	263	240	211	178	148	102	80.3
3.0 RHS	6.60	265	265	257	247	232	212	187	159	132	91.4	59.9
2.5 RHS	5.56	207	207	201	194	184	170	152	132	111	77.8	46.3
2.0 RHS	4.50	145	145	142	137	131	123	113	100	86.8	62.7	42.0
75 x 50 x 6.0 RHS	9.67	388	387	374	354	327	290	245	200	162	109	77.5
5.0 RHS	8.35	335	334	323	308	285	255	218	179	146	99.0	57.4
4.0 RHS	6.92	278	277	268	256	238	215	185	154	126	85.9	52.4
3.0 RHS	5.42	218	218	211	202	189	171	149	125	103	70.8	45.6
2.5 RHS	4.58	184	184	178	171	160	146	127	108	89.0	61.1	37.7
2.0 RHS	3.72	145	145	141	135	127	116	103	87.5	72.9	50.4	34.3
75 x 25 x 2.5 RHS	3.60	145	140	126	102	71.6	49.4	35.4	26.4	20.4	13.3	44.5
2.0 RHS	2.93	113	110	100	82.4	59.6	41.6	29.9	22.4	17.3	11.2	28.9
1.6 RHS	2.38	77.6	76.0	70.3	60.6	46.8	33.8	24.7	18.5	14.4	9.37	5.83
65 x 35 x 3.0 RHS	4.25	170	168	159	144	122	95.1	72.0	55.1	49.1	24.7	40.5
2.5 RHS	3.60	145	143	135	123	105	82.4	62.7	48.1	37.1	20.7	35.3
2.0 RHS	2.93	118	116	110	101	86.2	68.5	52.4	40.3	31.6	14.6	22.9
50 x 25 x 3.0 RHS	3.07	123	119	105	81.6	55.4	37.7	26.8	20.0	15.5	10.0	4.49
2.5 RHS	2.62	105	102	90.8	71.3	49.1	33.5	23.9	17.8	13.8	8.95	4.49
2.0 RHS	2.15	86.2	83.6	75.0	59.7	41.7	28.6	20.5	15.3	11.8	7.66	3.57
1.6 RHS	1.75	70.3	68.3	61.5	49.5	34.9	24.1	17.2	12.9	9.96	6.46	3.06
50 x 20 x 3.0 RHS	2.83	114	107	97.0	55.7	34.1	22.5	15.9	11.8	9.07	5.87	2.33
2.5 RHS	2.42	97.3	92.2	75.8	49.6	30.8	20.3	14.3	10.6	8.20	5.31	2.11
2.0 RHS	1.99	79.9	76.0	63.3	42.5	26.5	17.6	12.4	9.21	6.40	3.22	1.36
1.6 RHS	1.63	65.3	62.2	52.3	35.8	22.5	14.9	10.6	7.84	6.05	3.92	1.18

Notes:  
 1.  $\phi = 0.9$   
 2.  $\phi N_c = \phi \phi_{c,s}$  (Clause 6.3.3 of AS4100)  
 3.  $a_b = -0.5$   
 (Table 6.3.3 of AS4100)

Design Member Capacity  $\phi N_e$  (kN)

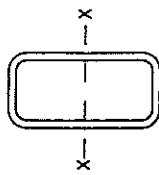
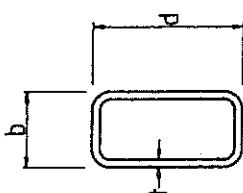
**TABLE 5.2 – 4 (1)(A)**  
**DESIGN CAPACITIES FOR MEMBERS**  
**SUBJECT TO AXIAL COMPRESSION**  
**RECTANGULAR HOLLOW SECTIONS: GRADE C450**  
**buckling about x-axis**



x—  
x

Designation d b t	Mass per m kg/m	Design Capacities for Axial Compression $\phi N_c$ (kN)							
		Effective length ( $l_e$ ) in metres							
mm mm mm	0.0	0.5	1.0	1.5	2.0	2.5	3.0	3.5	4.0
125 x 75 x 3.8 RHS	11.1	559	559	543	519	487	443	388	328
3.3 RHS	9.73	449	448	436	419	396	365	325	271
2.8 RHS	8.39	351	349	331	315	293	266	235	196
2.3 RHS	6.95	259	259	253	246	235	221	204	183
100 x 50 x 3.3 RHS	7.14	368	365	348	322	284	234	143	113
2.8 RHS	6.19	292	280	262	235	198	159	126	98.7
2.3 RHS	5.14	215	214	206	195	178	155	128	103

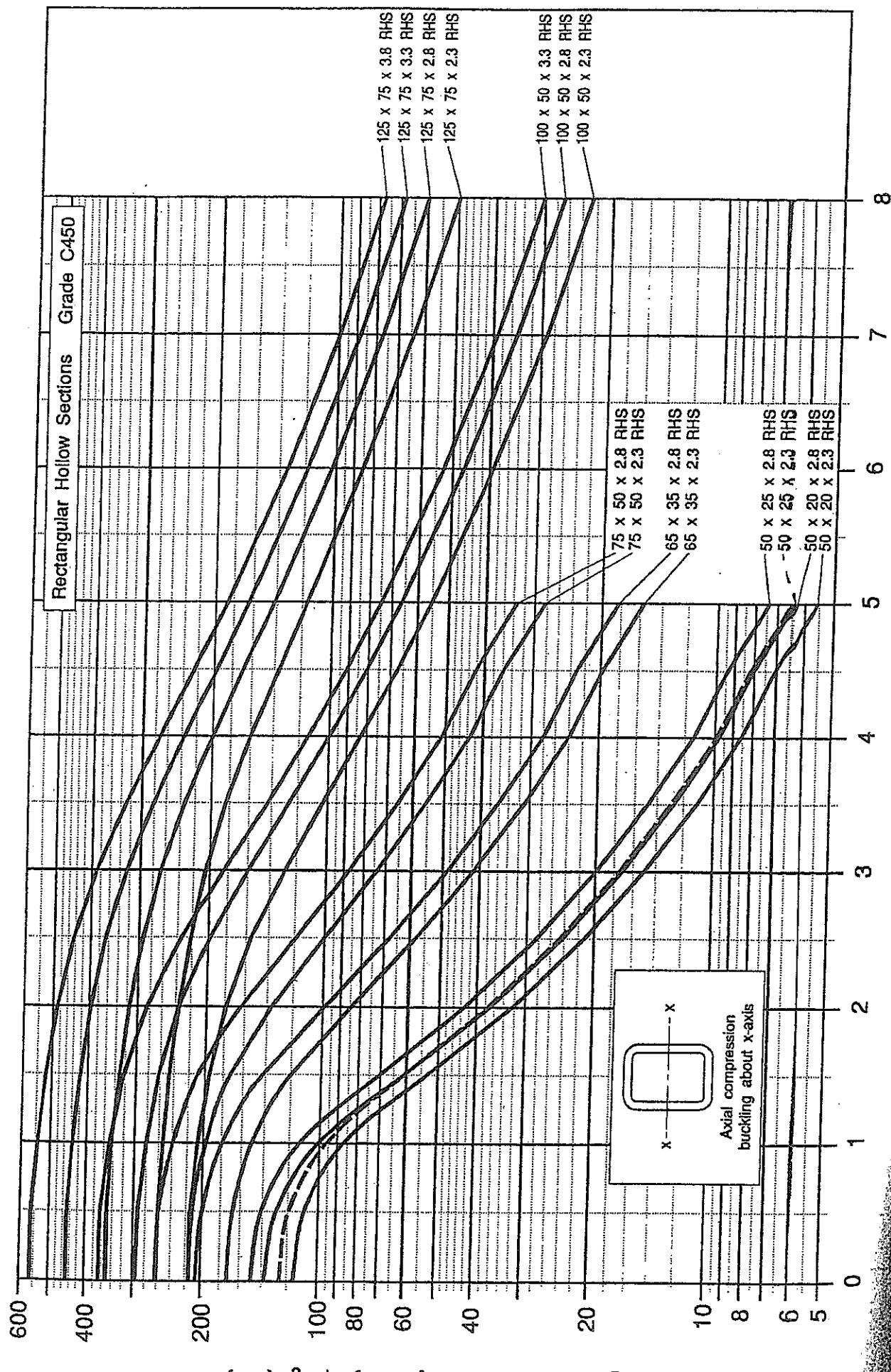
Notes: Refer to TABLE 5.2-4 (2)(A)



**TABLE 5.2 – 4 (2)(A)**  
**DESIGN CAPACITIES FOR MEMBERS**  
**SUBJECT TO AXIAL COMPRESSION**  
**RECTANGULAR HOLLOW SECTIONS: GRADE C450**  
**buckling about x-axis**

Designation d b t	Mass per m kg/m	Design Capacities for Axial Compression $\phi N_c$ (kN)							
		Effective Length ( $l_e$ ) in metres							
mm mm mm	0.0	0.25	0.5	0.75	1.0	1.25	1.5	1.75	2.0
100 x 50 x 3.8 RHS	7.14	368	368	365	358	348	337	322	305
2.8 RHS	6.19	295	292	287	280	272	262	249	234
2.3 RHS	5.14	215	214	211	206	201	195	187	178
75 x 50 x 2.8 RHS	5.09	263	256	250	239	225	208	187	163
2.3 RHS	4.24	215	211	205	197	186	172	155	136
65 x 35 x 2.8 RHS	3.99	206	206	191	179	163	142	120	119
2.3 RHS	3.34	172	172	167	160	150	137	102	84.5
50 x 50 x 2.8 RHS	2.89	149	148	141	129	113	91.3	70.8	54.8
2.3 RHS	2.44	126	124	119	109	95.9	78.4	61.2	47.5
50 x 20 x 2.8 RHS	2.67	138	136	129	118	101	80.3	61.4	47.2
2.3 RHS	2.25	116	115	109	100	86.5	69.4	55.5	41.2

Notes:  
1.  $\phi = 0.9$   
2.  $\phi N_c = \phi \alpha_c N_s$   
 $\alpha_c = \frac{1}{\sqrt{k}}$   
 $k = \frac{\pi^2 E}{l_e^2}$   
 $E = 200 GPa$   
 $\phi \alpha_c N_s = \text{Clause 6.3.3 of AS 4100}$   
 $(\text{Table 5.2-4 of AS 4100})$

Design Member Capacity ( $N_e$ ) (KN)

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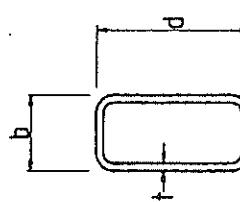


TABLE 5.2 – 4 (1)(B)

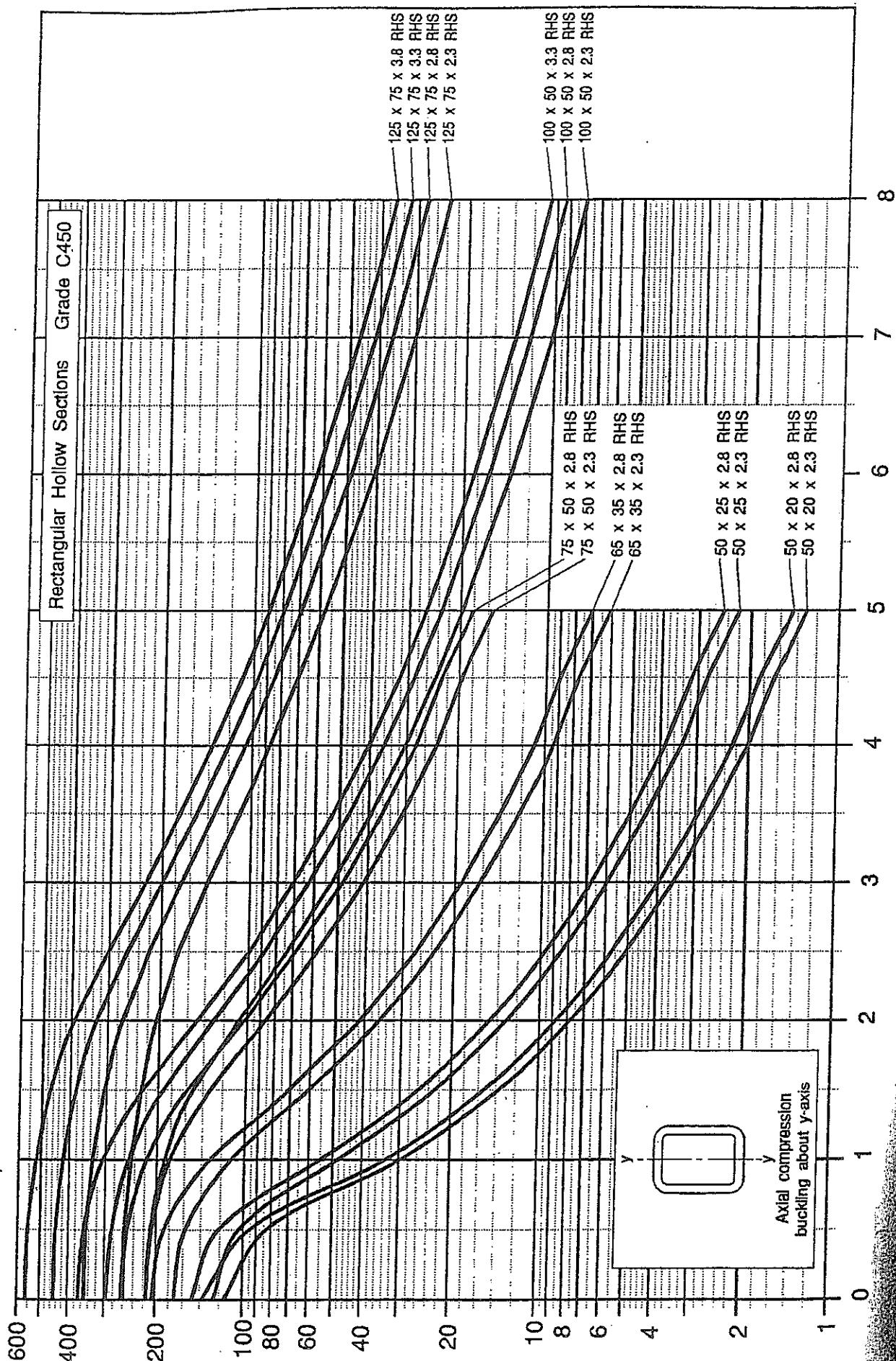
**DESIGN CAPACITIES FOR MEMBERS  
SUBJECT TO AXIAL COMPRESSION  
RECTANGULAR HOLLOW SECTIONS: GRADE C450  
buckling about y-axis**

Designation d b t mm mm mm	Mass per m kg/m	Design Capacities for Axial Compression $\phi N_c$ (kN)									
		0.0	0.5	1.0	1.5	2.0	2.5	3.0	3.5	4.0	4.5
125 x 75x38 RHS	11.1	559	552	520	469	392	303	228	174	136	109
3.3 RHS	9.73	448	443	420	383	328	261	200	153	120	99.0
2.8 RHS	8.39	351	348	332	306	268	219	171	133	105	84.2
2.3 RHS	6.95	259	257	246	229	205	173	139	109	86.9	70.1
100 x 50x33 RHS	7.14	368	354	307	226	148	99.4	70.5	52.5	40.5	32.2
2.8 RHS	6.19	296	284	251	192	130	87.8	62.5	46.6	36.0	28.6
2.3 RHS	5.14	215	209	188	151	107	73.5	52.6	39.3	30.4	24.2

Notes: Refer to TABLE 5.2-4 (2)(B)

Designation d b t mm mm mm	Mass per m kg/m	Design Capacities for Axial Compression $\phi N_c$ (kN)									
		0.0	0.25	0.5	0.75	1.0	1.25	1.5	1.75	2.0	3.0
100 x 50x33 RHS	7.14	368	366	354	335	307	270	226	184	148	99.4
2.8 RHS	6.19	296	294	284	271	251	225	192	159	130	87.8
2.3 RHS	5.14	215	215	209	200	188	172	151	128	107	73.5
75 x 50x33 RHS	5.09	263	261	252	238	217	180	158	127	102	68.5
2.8 RHS	4.24	216	214	207	196	180	158	132	107	86.7	58.2
2.3 RHS	3.34	172	169	158	139	111	98.4	70.9	53.5	41.6	35.9
65 x 50x28 RHS	3.99	206	202	188	165	131	98.4	70.9	53.5	41.6	35.9
2.8 RHS	3.34	172	169	158	139	111	82.6	60.9	46.1	35.9	23.4
50 x 50x28 RHS	2.89	149	143	122	86.4	55.2	36.8	26.1	19.4	14.9	14.9
2.3 RHS	2.44	126	120	104	74.8	48.4	32.3	22.9	17.0	13.1	8.51
50 x 50x25 RHS	2.67	138	128	96.6	56.2	33.5	21.9	15.4	11.4	8.80	5.69
2.3 RHS	2.25	116	109	83.5	49.7	29.8	19.5	13.7	10.2	7.84	5.07

Notes:  
 1.  $\phi = 0.9$   
 2.  $\phi N_c = \phi_{eff} N_c$  (Clause 6.3.3 of AS 4100)  
 $\phi_{eff} = \frac{1}{1 + \frac{t}{d}}$  (Table K.3.3 of AS 4100)



Design Member Capacity  $\phi N_c$  (kN)  
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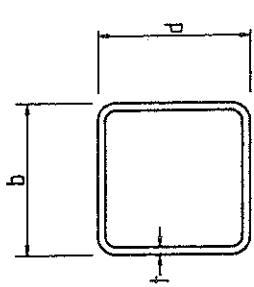
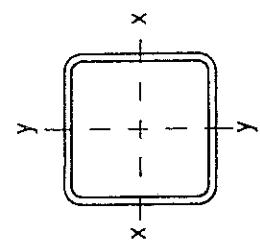


TABLE 5.2 – 5 (1)

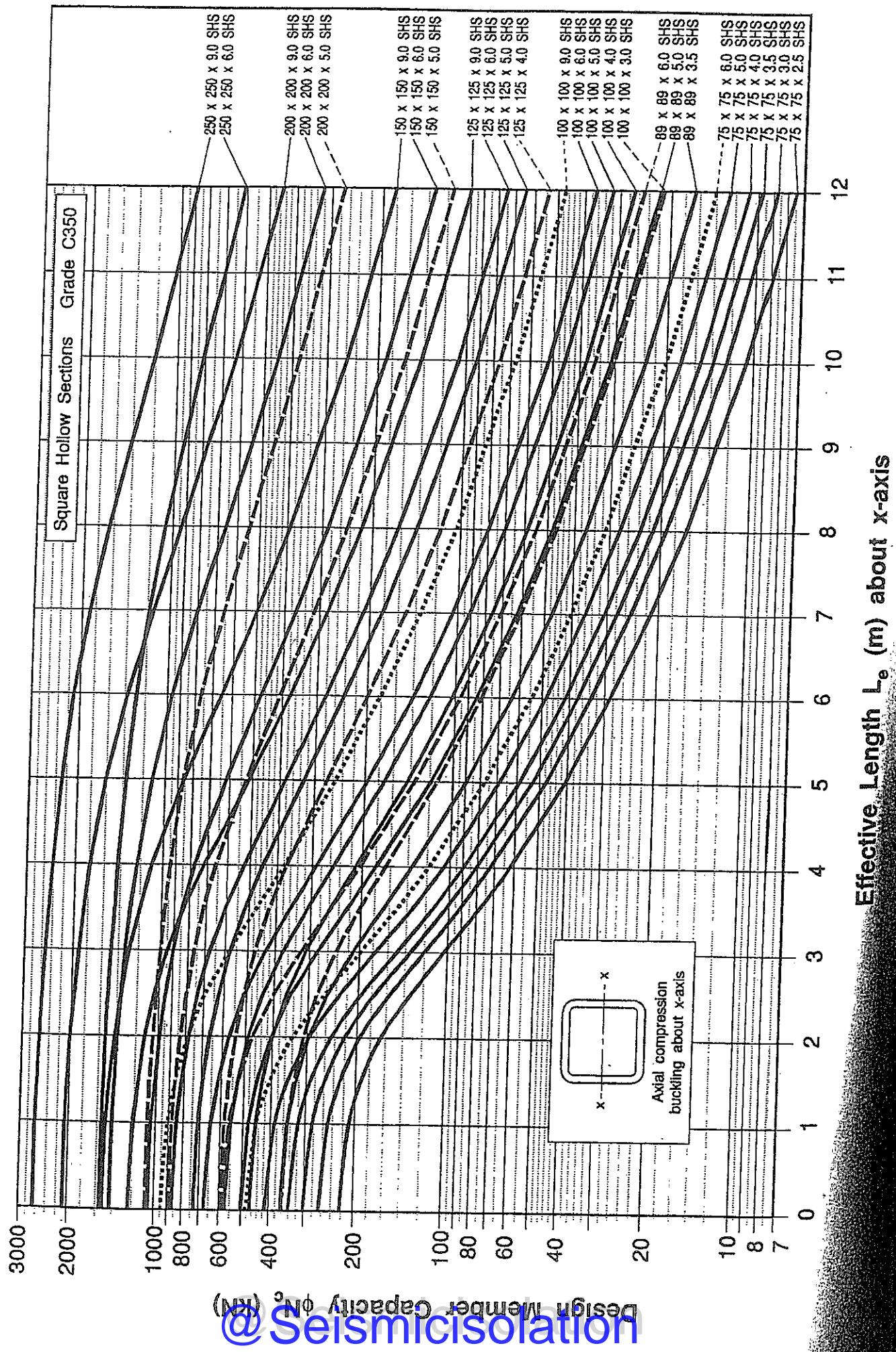
**DESIGN CAPACITIES FOR MEMBERS  
SUBJECT TO AXIAL COMPRESSION  
SQUARE HOLLOW SECTIONS: GRADE C350**

buckling about x- and y-axis



Designation d b t	Mass per m	Design Capacities for Axial Compression $\phi N_c$ (kN)													
		Effective Length ( $L_e$ ) in metres													
mm mm mm	kg/m	0.0	1.0	1.5	2.0	2.5	3.0	3.5	4.0	5.0	6.0	7.0	8.0	10.0	12.0
250 x 250 x 9.0 SHS	65.9	2650	2650	2620	2590	2560	2520	2470	2410	2270	2100	1880	1650	1200	882
6.0 SHS	45.0	1540	1540	1520	1520	1500	1480	1450	1430	1380	1280	1180	1060	810	607
200 x 200 x 9.0 SHS	51.8	2080	2070	2040	2010	1960	1910	1850	1780	1600	1370	1140	1028	630	448
6.0 SHS	35.6	1420	1420	1400	1380	1350	1320	1280	1230	1110	960	801	658	449	320
5.0 SHS	29.9	1070	1070	1050	1040	1020	998	971	940	862	763	652	544	377	271
150 x 150 x 9.0 SHS	37.7	1510	1490	1460	1440	1450	1280	1190	1090	857	652	500	391	256	180
6.0 SHS	26.2	1050	1040	1010	985	947	900	842	774	617	474	364	286	188	132
5.0 SHS	22.1	886	876	857	833	801	762	714	658	527	406	313	245	161	114
125 x 125 x 9.0 SHS	30.6	1230	1200	1160	1110	1040	944	835	718	515	374	281	218	142	99.4
6.0 SHS	21.4	861	843	817	781	735	676	604	525	382	279	210	163	106	74.5
5.0 SHS	18.2	729	715.	693	663	625	576	516	451	329	241	181	141	91.8	64.4
4.0 SHS	14.8	593	581	564	540	510	471	424	371	272	199	150	117	76.1	53.4
100 x 100 x 9.0 SHS	23.5	944	907	858	787	692	579	469	378	254	180	134	103	67.0	46.9
6.0 SHS	16.7	672	648	616	570	508	433	356	290	196	139	104	80.2	51.9	36.4
5.0 SHS	14.2	571	552	525	488	437	374	309	262	171	122	90.7	70.1	45.4	31.8
4.0 SHS	11.6	467	451	430	400	360	310	257	210	143	102	76.0	58.7	38.1	26.7
3.0 SHS	8.96	359	348	332	310	280	242	202	166	113	80.8	60.3	46.6	30.2	21.2
89 x 89 x 6.0 SHS	14.6	588	560	524	471	400	323	256	203	135	95.4	70.8	54.7	35.4	24.8
5.0 SHS	12.5	562	479	449	405	347	282	224	178	119	83.9	62.4	48.1	31.1	21.8
3.5 SHS	9.06	363	348	327	297	257	211	168	135	99.8	63.6	47.2	36.5	23.6	16.5
75 x 75 x 6.0 SHS	12.0	483	449	405	340	263	198	151	118	77.4	54.4	40.3	31.1	20.1	14.0
5.0 SHS	10.3	414	386	350	296	232	176	134	105	66.9	48.5	35.9	27.7	17.9	12.5
4.0 SHS	8.49	341	319	280	247	196	149	114	89.7	58.8	41.4	30.7	23.6	15.3	10.7
3.5 SHS	7.53	302	283	258	221	176	134	103	80.9	53.1	37.4	27.7	21.4	13.8	9.66
3.0 SHS	6.60	285	249	227	196	156	120	92.6	72.6	47.6	33.5	24.9	19.2	12.4	8.67
2.5 SHS	5.56	223	210	192	166	133	103	79.1	62.1	40.8	28.7	21.3	16.4	10.6	7.43

Notes:  
 1.  $\phi = 0.9$   
 2.  $\phi_{\text{ax}} = \phi_{\text{ax}} N_s$  (Clause 6.3.3 of AS 4100)  
 3.  $\alpha_0 = -0.5$  (Table 6.3.3 of AS 4100)



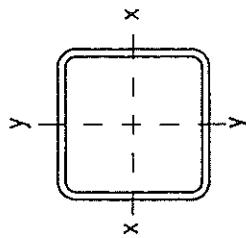
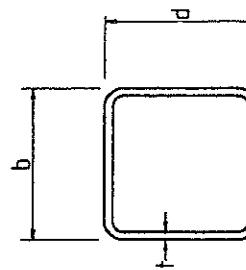


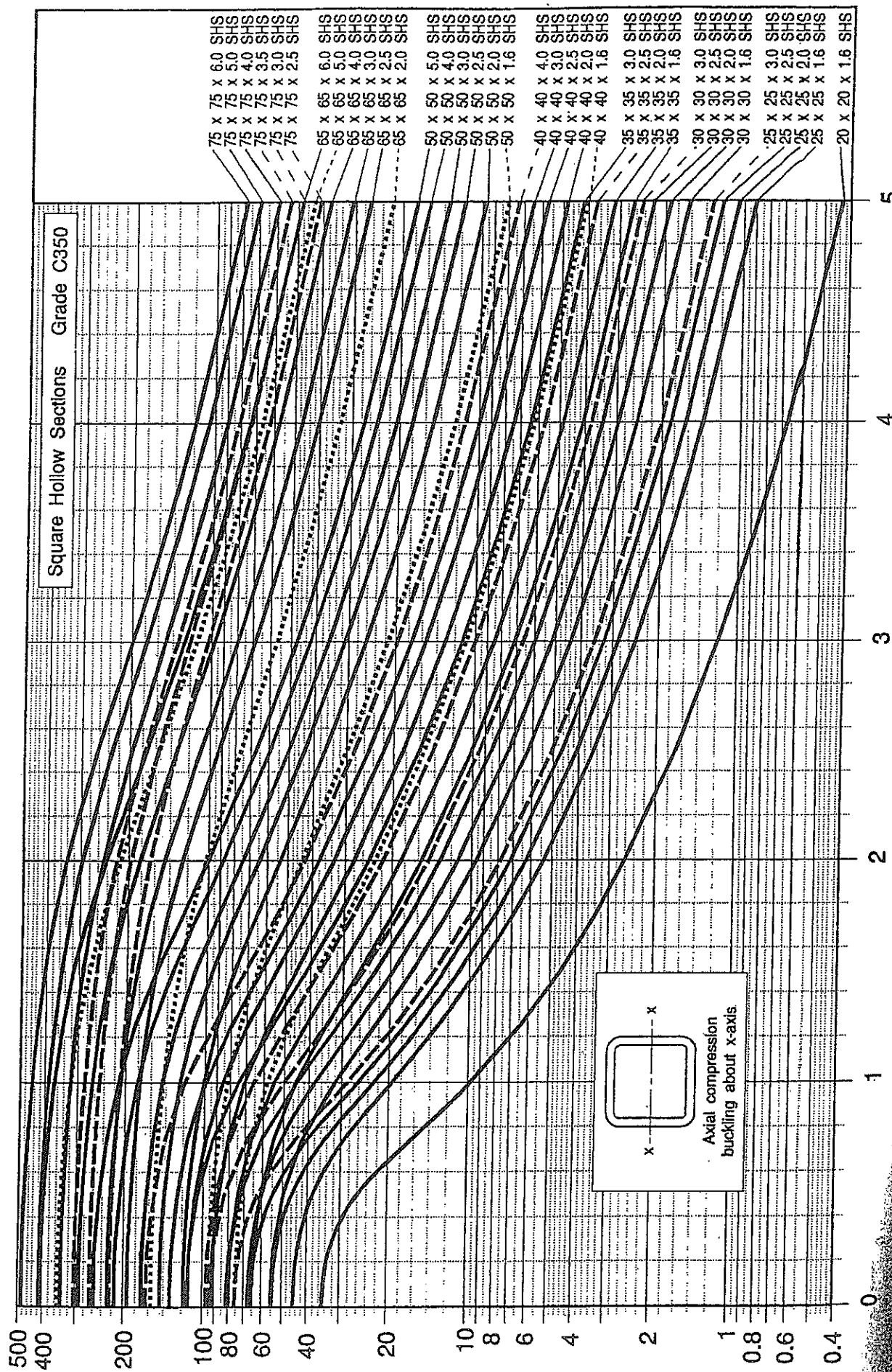
TABLE 5.2 – 5 (2)

**DESIGN CAPACITIES FOR MEMBERS  
SUBJECT TO AXIAL COMPRESSION  
SQUARE HOLLOW SECTIONS: GRADE C350  
buckling about x- and y-axis**

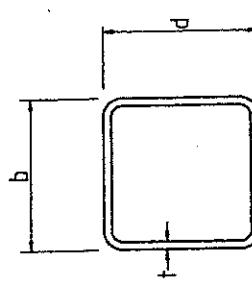


Designation <b>d b t</b>	Mass per m <b>kg/m</b>	Design Capacities for Axial Compression $\phi N_c$ (kN)													
		Effective Length ( $L_e$ ) in metres													
<b>mm mm mm</b>	<b>0.0</b>	<b>0.25</b>	<b>0.5</b>	<b>0.75</b>	<b>1.0</b>	<b>1.25</b>	<b>1.5</b>	<b>1.75</b>	<b>2.0</b>	<b>2.5</b>	<b>3.0</b>	<b>3.5</b>	<b>4.0</b>	<b>5.0</b>	
75 x 75 x 6.0 SHS	12.0	483	483	476	464	449	430	405	375	340	263	198	151	118	77.4
50 x 50 SHS	10.3	414	414	409	399	386	370	350	325	296	232	176	134	105	68.9
40 x 40 SHS	8.49	341	341	336	329	319	306	290	271	247	196	149	114	89.7	58.8
35 x 35 SHS	7.53	302	302	298	292	283	272	258	241	221	176	134	103	80.9	53.1
30 x 30 SHS	6.60	265	265	262	256	249	239	227	213	196	156	120	92.5	72.6	47.6
25 x 25 SHS	5.56	223	223	221	216	210	202	192	180	166	133	103	79.1	62.1	40.8
65 x 65 x 6.0 SHS	10.1	407	407	398	385	368	344	315	279	241	173	126	94.6	73.4	47.7
50 x 50 SHS	8.75	351	351	344	333	318	299	275	246	214	155	113	85.2	66.2	43.1
40 x 40 SHS	7.23	290	290	285	276	265	250	230	207	181	133	97.4	73.5	57.1	37.2
30 x 30 SHS	5.66	227	227	223	217	208	197	183	166	146	108	79.7	60.3	46.9	30.6
25 x 25 SHS	4.78	192	192	188	183	176	167	155	141	125	92.8	68.5	51.8	40.3	26.3
20 x 20 SHS	3.88	156	156	153	149	143	136	126	115	102	76.3	56.5	42.7	33.3	21.7
50 x 50 x 5.0 SHS	6.39	256	255	246	232	212	186	154	124	100	66.9	47.4	35.3	27.2	17.6
40 x 40 x 5.0 SHS	5.35	215	214	206	196	180	159	134	109	88.1	59.3	42.1	31.3	24.2	15.7
30 x 30 x 4.25 SHS	4.25	170	170	164	156	145	129	110	91.0	74.1	50.2	35.7	26.6	20.5	13.3
25 x 25 x 3.60 SHS	3.60	145	144	140	133	124	111	95.0	78.7	64.2	43.6	31.0	23.1	17.9	11.6
20 x 20 x 2.93 SHS	2.93	118	117	114	108	101	90.9	78.4	65.2	53.4	36.3	25.9	19.3	14.9	9.66
16 x 16 x 2.38 SHS	2.38	95.5	95.5	92.4	88.2	82.2	74.2	64.2	53.6	44.0	30.0	21.4	15.9	12.3	7.99
40 x 40 x 4.0 SHS	4.09	164	162	153	140	119	93.6	71.3	54.7	42.8	28.1	19.7	14.6	11.3	7.29
30 x 30 x 3.30 SHS	3.30	133	131	125	115	99.4	80.2	62.0	47.9	37.7	24.8	17.4	12.9	9.97	6.45
25 x 25 x 2.82 SHS	2.82	113	112	107	98.2	85.7	69.8	54.3	42.1	33.1	21.8	15.4	11.4	8.78	5.68
20 x 20 x 2.31 SHS	2.31	92.5	91.6	87.4	80.7	70.9	58.1	45.5	35.4	27.9	18.4	13.0	9.61	7.41	4.79
16 x 16 x 1.88 SHS	1.88	75.3	74.6	71.3	66.0	58.2	48.0	37.8	29.4	23.2	16.3	10.8	8.02	6.19	4.00
35 x 35 x 3.0 SHS	2.83	114	112	105	92.5	74.9	56.1	41.5	31.5	24.5	16.0	11.2	8.32	6.41	4.14
25 x 25 x 2.42 SHS	2.42	97.3	95.7	89.8	79.9	65.4	49.4	36.8	27.9	21.7	14.2	9.97	7.39	5.69	3.68
20 x 20 x 1.99 SHS	1.99	79.9	78.7	74.0	66.2	54.7	41.7	31.2	23.7	18.5	12.1	8.49	6.29	4.85	3.13
16 x 16 x 1.63 SHS	1.63	65.3	64.3	60.6	54.4	45.2	34.8	26.1	19.9	15.5	10.1	7.13	5.28	4.07	2.63
30 x 30 x 3.0 SHS	2.36	94.8	92.3	83.9	69.2	50.3	35.2	25.3	18.9	14.7	9.52	6.68	4.94	3.80	2.45
25 x 25 x 2.03 SHS	2.03	81.6	79.5	72.7	60.6	44.7	31.5	22.7	17.0	13.2	8.56	6.01	4.44	3.42	2.21
20 x 20 x 1.68 SHS	1.68	67.3	65.7	60.3	50.8	38.0	27.0	19.5	14.6	11.3	7.38	5.17	3.83	2.95	1.90
16 x 16 x 1.38 SHS	1.38	55.2	53.9	49.6	42.2	31.9	22.8	16.5	12.4	9.61	6.25	4.38	3.25	2.50	1.61
25 x 25 x 3.0 SHS	1.89	75.9	72.7	62.4	44.7	28.8	19.2	13.6	10.1	7.81	5.06	3.54	2.62	2.01	1.30
25 x 25 x 1.64 SHS	1.64	65.8	63.2	54.8	40.2	26.2	17.6	12.5	9.27	7.16	4.64	3.25	2.40	1.85	1.19
20 x 20 x 1.36 SHS	1.36	54.7	52.7	46.1	34.5	22.8	15.4	10.9	8.12	6.27	4.06	2.85	2.10	1.62	1.04
16 x 16 x 1.12 SHS	1.12	45.1	43.5	38.3	29.1	19.5	13.2	9.36	6.97	5.38	3.49	2.44	1.81	1.39	0.897
20 x 20 x 1.6 SHS	0.873	35.0	32.9	26.1	16.2	9.33	6.46	4.55	3.37	2.60	1.68	1.18	0.869	0.668	0.430

Notes:  
 1.  $\phi = 0.9$   
 2.  $\phi_{ac} N_c = \phi_{ac} N_s$  [Clause 6.3.3 of AS 4100]



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**DESIGN CAPACITIES FOR MEMBERS  
SUBJECT TO AXIAL COMPRESSION  
SQUARE HOLLOW SECTIONS: GRADE C450  
buckling about x- and y-axis**

TABLE 5.2 – 6 (1)

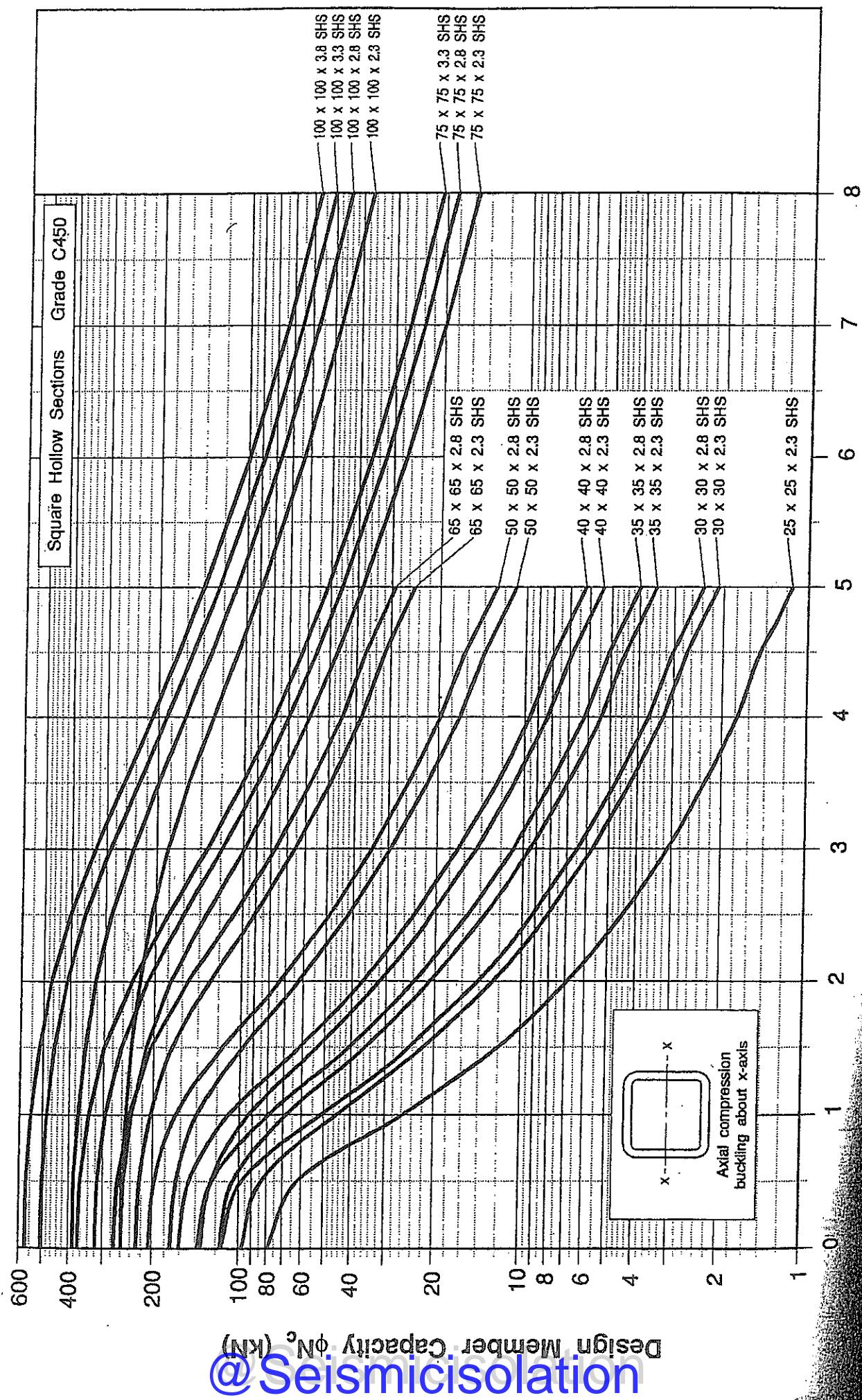
Designation d b t	Mass per m kg/m	Design Capacities for Axial Compression $\phi N_c$ (kN)											
		0.0	0.5	1.0	1.5	2.0	2.5	3.0	3.5	4.0	4.5	5.0	6.0
100 x 100 x 3.8 SHS	11.1	572	569	547	514	466	402	329	263	210	170	140	98.9
3.3 SHS	9.73	502	499	460	452	410	355	291	233	186	151	124	87.9
2.8 SHS	8.39	383	382	369	350	323	286	242	198	160	131	108	76.6
2.3 SHS	6.95	259	258	251	241	226	207	182	155	128	106	88.6	63.4
75 x 75 x 3.3 SHS	7.14	368	362	338	299	241	180	133	101	78.4	62.7	51.2	35.9
2.8 SHS	6.19	319	314	294	261	212	159	118	89.3	69.6	55.6	45.4	31.9
2.3 SHS	5.14	259	255	239	213	175	99.0	75.2	58.6	46.9	38.3	26.9	19.9

Notes: Refer to TABLE 5.2 – 6 (2)

TABLE 5.2 – 6 (2)

Designation d b t	Mass per m kg/m	Design Capacities for Axial Compression $\phi N_c$ (kN)												
		0.0	0.25	0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5	3.0	3.5	4.0
75 x 75 x 3.3 SHS	7.14	368	368	362	352	338	321	299	272	241	180	133	101	78.4
2.8 SHS	6.19	319	319	314	305	294	279	261	238	212	159	118	89.3	69.6
2.3 SHS	5.14	259	259	255	248	239	228	213	195	175	133	99.0	75.2	58.6
65 x 65 x 2.8 SHS	5.31	274	274	267	258	244	227	205	178	152	107	77.2	57.8	44.8
2.3 SHS	4.42	228	228	223	215	204	190	172	150	128	90.5	65.4	49.0	38.0
50 x 50 x 2.8 SHS	3.99	206	205	197	184	166	142	115	91.5	73.0	48.5	34.3	25.5	19.7
2.3 SHS	3.34	172	171	165	154	140	120	98.1	78.1	62.4	41.5	29.4	21.8	16.8
40 x 40 x 2.8 SHS	3.11	161	158	148	133	108	82.7	61.7	46.9	36.5	23.9	16.8	12.4	9.58
2.3 SHS	2.62	135	133	125	112	92.9	71.2	53.3	40.6	31.7	20.7	14.5	10.8	8.31
35 x 35 x 2.8 SHS	2.67	138	135	124	105	79.0	56.3	40.8	30.6	23.7	15.4	10.8	8.01	6.17
2.3 SHS	2.25	116	114	105	89.6	68.3	49.0	35.6	26.8	20.8	13.5	9.47	7.01	5.40
30 x 30 x 2.8 SHS	2.23	115	111	98.4	75.7	51.2	34.7	24.7	18.4	14.2	9.23	6.46	4.78	3.68
2.3 SHS	1.89	97.7	94.5	84.1	65.7	45.0	30.7	21.9	16.3	12.6	8.17	5.73	4.24	3.26
25 x 25 x 2.3 SHS	1.53	79.1	75.1	62.2	41.4	25.7	17.0	12.0	8.92	6.88	4.45	3.12	2.30	1.77

Notes:  
 1.  $\phi = 0.9$   
 2.  $\phi N_c = \phi c N_s$  (Clause 6.3.3 of AS 4100)  
 3.  $a_b = -0.5$  (Table 6.3.3 of AS 4100)



**PART 6 MEMBERS SUBJECT TO AXIAL TENSION**

	PAGE
6.1 SCOPE.....	6-2
6.2 METHOD.....	6-2
6.3 EXAMPLES .....	6-3

**TABLES****TABLES 6.1-1 to 6.1-6**Design Load Capacities for Members Subject to Axial Tension ( $\phi N$ )..... 6-4

**NOTE: SEE PAGE (ix) FOR THE SPECIFIC MATERIAL  
STANDARD REFERRED TO BY THE SECTION TYPE  
AND STEEL GRADE IN THESE TABLES**

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**PART 6 MEMBERS SUBJECT TO AXIAL TENSION****6.1 Scope**

Tables 6.1–1 to 6.1–6 give values of design section capacity for axial tension determined in accordance with Section 7 of AS 4100.

The tables give values of design capacity for the various grades and types of structural steel hollow sections with full perimeter welded connections.

**6.2 Method**

The design section capacity for axial tension ( $\phi N_t$ ) has been determined from Clause 7.2 of AS 4100 and taken as the *lesser* of:

$$\text{and} \quad \begin{aligned} \phi N_t &= \phi A_g f_y \\ \phi N_t &= \phi (0.85) k_t A_n f_u \end{aligned}$$

where  $\phi = 0.9$

$f_y$  = yield stress used in design

$A_n$  = net section area

=  $A_g$

= gross cross-sectional area (for full perimeter welded connections)

$f_u$  = ultimate strength used in design

$k_t = 1.0$  (Clause 7.3.1 of AS 4100)

The lesser value of  $\phi N_t = \phi A_g f_y$  and  $\phi N_t = \phi (0.85) A_g f_u$  is highlighted in bold type in the tables.

Note: for grades C250 and C350  $\phi N_t = \phi A_g f_y$  is always less than  $\phi N_t = \phi (0.85) A_g f_u$  but for grade C450  $\phi N_t = \phi (0.85) A_g f_u$  is the lesser value of  $\phi N_t$ .

For sections reduced by penetrations or holes, the value of  $\phi N_t$  can be determined from the tables as the lesser value of:

$$\text{and} \quad \begin{aligned} \phi N_t &= \phi A_g f_y \\ \phi N_t &= \phi (0.85) A_g f_u (A_n/A_g) \end{aligned}$$

where  $A_n$  = net section area

$k_t$  = tension correction factor (Clause 7.3.1 of AS 4100)

Values of  $A_g$  are tabulated in Tables 6.1–1 to 6.1–6.

Note that all the values in Tables 6.1–1 to 6.1–6 assume  $k_t = 1.0$ .

## 6.3 Examples

1. A tension member with a full perimeter welded connection is subjected to an axial tension force of 150 kN. Design a suitable RHS tension member.

Design Data:

$$N^* = 150 \text{ kN}$$

$$k_t = 1.0 \quad (\text{for a full perimeter welded connection})$$

Solution:

Select a suitable RHS member from Tables 6.1–3 and 6.1–4. The alternatives are:

$$65 \times 35 \times 3.0 \text{ RHS} - \text{Grade C350 (4.25 kg/m)} \quad \phi N_t = 170 > N^*$$

$$65 \times 35 \times 2.3 \text{ RHS} - \text{Grade C450 (3.34 kg/m)} \quad \phi N_t = 163 > N^*$$

Choose 65 x 35 x 2.3 RHS Grade C450 (3.34 kg/m) because it is more economical based on mass.

2. Select a CHS section for a tension wind brace to transmit a design force of 92 kN. A flattened end connection with a single line of M20 Grade 8.8/s bolts is to be used.

Design Data:

$$N^* = 92 \text{ kN}$$

Bolt hole diameter = 22 mm

For this type of connection the design capacity should be reduced by 20% to avoid progressive flattening of the CHS.

Solution:

Try 60.3 x 2.9 CHS – Grade C350

$$\text{Gross area of section } A_g = 523 \text{ mm}^2 \quad (\text{Table 6.1-2})$$

$$\begin{aligned} \text{Area removed by hole} &= 2 \times 22 \times 2.9 \\ &= 128 \text{ mm}^2 \end{aligned}$$

$$\begin{aligned} \text{Net area of section } A_n &= 523 - 128 \\ &= 395 \text{ mm}^2 \end{aligned}$$

$$\begin{aligned} \text{Section capacity } \phi N_t &= \phi N_t (2) \left( \frac{A_n}{A_g} \right) \quad (\text{Table 6.1-2}) \\ &= 172 \left( \frac{395}{523} \right) \\ &= 129 \text{ kN} \quad (< \phi N_t(1) = 165 \text{ kN} \text{ from Table 6.1-2}) \end{aligned}$$

$$\text{Section capacity reduced by 20\%} = 0.8 \times 129$$

$$= 103 \text{ kN} \quad (> N^*)$$

Note:  $k_t = 1.0$  is assumed for the flattened end connection.

∴ 60.3 x 2.9 CHS – Grade C250 is a suitable section.

TABLE 6.1-1  
DESIGN CAPACITIES  
FOR MEMBERS SUBJECT TO  
AXIAL TENSION  
CIRCULAR HOLLOW SECTIONS  
GRADE C250

Designation	Mass per m	Axial Tension $\phi N_i(1)$	Axial Tension $\phi N_i(2)$	Gross Section Area $A_g$
$d_o$	$t$	kg/m	kN	mm <sup>2</sup>
610.0 x 12.7 CHS	187	5360	5830	23800
9.5 CHS	141	4030	4390	17900
6.4 CHS	95.3	2730	2970	12100
508.0 x 12.7 CHS	155	4450	4840	19800
9.5 CHS	117	3350	3640	14900
6.4 CHS	79.2	2270	2470	10100
165.1 x 5.4 CHS	21.3	610	663	2710
5.0 CHS	19.7	566	616	2510
139.7 x 5.4 CHS	17.9	513	558	2280
5.0 CHS	16.6	476	518	2120
114.3 x 5.4 CHS	14.5	416	452	1850
4.5 CHS	12.2	349	380	1550
101.6 x 5.0 CHS	11.9	341	371	1520
4.0 CHS	9.63	276	300	1230
88.9 x 5.9 CHS	12.1	346	377	1540
5.0 CHS	10.3	297	323	1320
4.0 CHS	8.38	240	261	1070
76.1 x 5.9 CHS	10.2	293	319	1300
4.5 CHS	7.95	228	248	1010
3.6 CHS	6.44	184	201	820
60.3 x 5.4 CHS	7.31	210	228	931
4.5 CHS	6.19	177	193	789
3.6 CHS	5.03	144	157	641
48.3 x 5.4 CHS	5.71	164	178	728
4.0 CHS	4.37	125	136	557
3.2 CHS	3.56	102	111	453
42.4 x 4.9 CHS	4.53	130	141	577
4.0 CHS	3.79	109	118	483
3.2 CHS	3.09	88.7	96.5	394
33.7 x 4.5 CHS	3.24	92.9	101	413
4.0 CHS	2.93	84.0	91.4	373
3.2 CHS	2.41	69.0	75.1	307
26.9 x 4.0 CHS	2.26	64.7	70.4	288
3.2 CHS	1.87	53.6	58.3	238
2.6 CHS	1.56	44.7	48.6	198
21.3 x 3.6 CHS	1.57	45.0	49.0	200
3.2 CHS	1.43	40.9	44.5	182
2.6 CHS	1.20	34.4	37.4	153

Notes: 1.  $\phi = 0.9$  (Table 3.4 of AS 4100)  
 2.  $\phi N_i(1) = \phi A_g f_y$  (Clause 7.2 of AS 4100)  
 3.  $\phi N_i(2) = \phi 0.85 A_g f_u$  (Clause 7.2 of AS 4100)

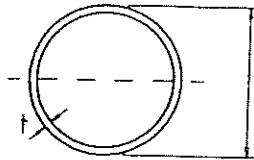


TABLE 6.1-2  
DESIGN CAPACITIES  
FOR MEMBERS SUBJECT TO  
AXIAL TENSION  
CIRCULAR HOLLOW SECTIONS  
GRADE C350

Designation:	Mass per m	Axial Tension $\phi N_i(1)$	Axial Tension $\phi N_i(2)$	Gross Section Area $A_g$
$d_o$	$t$	kg/m	kN	mm <sup>2</sup>
457.0 x 12.7 CHS	139	5580	5830	17700
9.5 CHS	105	4210	4390	13400
6.4 CHS	71.1	2850	2980	9060
406.4 x 12.7 CHS	123	4950	5170	15700
9.5 CHS	93.0	3730	3900	11800
6.4 CHS	63.1	2530	2650	8040
355.6 x 12.7 CHS	107	4310	4500	13700
9.5 CHS	81.1	3250	3400	10300
6.4 CHS	55.1	2210	2310	7020
323.9 x 12.7 CHS	97.5	3910	4080	12400
9.5 CHS	73.7	2960	3090	9380
6.4 CHS	50.1	2010	2100	6380
273.1 x 9.3 CHS	60.5	2430	2540	7710
6.4 CHS	42.1	1690	1760	5360
4.8 CHS	31.8	1270	1330	4050
219.1 x 8.2 CHS	42.6	1710	1790	5430
6.4 CHS	33.6	1350	1410	4280
4.8 CHS	25.4	1020	1060	3230
168.3 x 7.1 CHS	28.2	1130	1180	3600
6.4 CHS	25.6	1030	1070	3260
4.8 CHS	19.4	777	811	2470
165.1 x 3.5 CHS	13.9	560	585	1780
3.0 CHS	12.0	481	503	1530
139.7 x 3.5 CHS	11.8	472	493	1500
3.0 CHS	10.1	406	424	1290
114.3 x 6.0 CHS	16.0	643	672	2040
4.8 CHS	13.0	520	543	1650
3.6 CHS	9.83	394	412	1250
3.2 CHS	8.77	352	367	1120
101.6 x 3.2 CHS	7.77	312	325	989
2.6 CHS	6.35	255	266	809
88.9 x 5.5 CHS	11.3	454	474	1440
4.8 CHS	9.96	399	417	1270
3.2 CHS	7.67	271	283	862
2.6 CHS	5.53	222	232	705
76.1 x 3.2 CHS	5.75	231	241	733
2.3 CHS	4.19	168	175	533
60.3 x 2.9 CHS	4.11	165	172	523
2.3 CHS	3.29	132	138	419
48.3 x 2.9 CHS	3.25	130	136	414
2.3 CHS	2.61	105	109	332
42.4 x 2.6 CHS	2.55	102	107	325
2.0 CHS	1.99	80.0	83.5	254
33.7 x 2.6 CHS	1.99	80.0	83.6	254
2.0 CHS	1.56	62.7	65.5	199
26.9 x 2.3 CHS	1.40	56.0	58.5	178
2.0 CHS	1.23	49.3	51.5	156
21.3 x 2.0 CHS	0.952	38.2	39.9	121

Notes: 1.  $\phi = 0.9$  (Table 3.4 of AS 4100)  
 2.  $\phi N_i(1) = \phi A_g f_y$  (Clause 7.2 of AS 4100)  
 3.  $\phi N_i(2) = \phi 0.85 A_g f_u$  (Clause 7.2 of AS 4100)

TABLE 6.1-3

**DESIGN CAPACITIES  
FOR MEMBERS SUBJECT TO  
AXIAL TENSION**

**RECTANGULAR HOLLOW SECTIONS  
GRADE C350**

Designation <i>d b t</i>	Mass per m kg/m	Axial Tension $\phi N_c(1)$	Axial Tension $\phi N_c(2)$	Gross Section Area $A_g$
mm mm mm		kN	kN	mm <sup>2</sup>
250 x 150 x 9.0 RHS	51.8	2080	2170	6600
6.0 RHS	35.6	1430	1490	4350
5.0 RHS	29.9	1200	1250	3810
200 x 100 x 9.0 RHS	37.7	1510	1580	4800
6.0 RHS	26.2	1050	1100	3330
5.0 RHS	22.1	886	926	2810
4.0 RHS	17.9	719	750	2280
150 x 100 x 6.0 RHS	21.4	861	899	2730
5.0 RHS	18.2	729	761	2310
4.0 RHS	14.8	593	619	1880
150 x 50 x 5.0 RHS	14.2	571	597	1810
4.0 RHS	11.6	467	487	1480
3.0 RHS	8.96	359	375	1140
125 x 75 x 6.0 RHS	16.7	672	701	2130
5.0 RHS	14.2	571	597	1810
4.0 RHS	11.6	467	487	1480
3.0 RHS	8.96	359	375	1140
100 x 50 x 6.0 RHS	12.0	483	504	1530
5.0 RHS	10.3	414	432	1310
4.0 RHS	8.49	341	356	1080
3.5 RHS	7.53	302	315	959
3.0 RHS	6.60	265	277	841
2.5 RHS	5.56	223	233	709
2.0 RHS	4.50	181	189	574
75 x 50 x 6.0 RHS	9.67	388	405	1230
5.0 RHS	8.35	335	350	1060
4.0 RHS	6.92	278	290	881
3.0 RHS	5.42	218	227	691
2.5 RHS	4.58	184	192	584
2.0 RHS	3.72	149	156	474
75 x 25 x 2.5 RHS	3.60	145	151	459
2.0 RHS	2.93	118	123	374
1.6 RHS	2.38	95.5	99.7	303
65 x 35 x 3.0 RHS	4.25	170	178	541
2.5 RHS	3.60	145	151	459
2.0 RHS	2.93	118	123	374
50 x 25 x 3.0 RHS	3.07	123	129	391
2.5 RHS	2.62	105	110	334
2.0 RHS	2.15	86.2	90.0	274
1.6 RHS	1.75	70.3	73.4	223
50 x 20 x 3.0 RHS	2.83	114	119	361
2.5 RHS	2.42	97.3	102	309
2.0 RHS	1.99	79.9	83.5	254
1.6 RHS	1.63	65.3	68.1	207

Notes: 1.  $\phi = 0.9$  (Table 3.4 of AS 4100)  
 2.  $\phi N_c(1) = \phi A_g f_y$  (Clause 7.2 of AS 4100)  
 3.  $\phi N_c(2) = \phi 0.85 A_g f_u$  (Clause 7.2 of AS 4100)

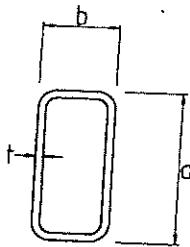


TABLE 6.1-4

**DESIGN CAPACITIES  
FOR MEMBERS SUBJECT TO  
AXIAL TENSION  
RECTANGULAR HOLLOW SECTION  
GRADE C450**

Designation <i>d b t</i>	Mass per m kg/m	Axial Tension $\phi N_c(1)$	Axial Tension $\phi N_c(2)$	Gross Section Area $A_g$
mm mm mm		kN	kN	mm <sup>2</sup>
125 x 75 x 3.8 RHS	11.1	572	540	1410
3.3 RHS	9.73	502	474	1240
2.8 RHS	8.39	433	409	1070
2.3 RHS	6.95	359	339	885
100 x 50 x 3.3 RHS	7.14	368	348	908
2.8 RHS	6.19	319	302	788
2.3 RHS	5.14	265	251	655
75 x 50 x 2.8 RHS	5.09	263	248	648
2.3 RHS	4.24	219	207	540
65 x 35 x 2.8 RHS	3.99	206	194	508
2.3 RHS	3.34	172	163	425
50 x 25 x 2.8 RHS	2.89	149	141	368
2.3 RHS	2.44	126	119	310
50 x 20 x 2.8 RHS	2.67	138	130	340

Notes: 1.  $\phi = 0.9$  (Table 3.4 of AS 4100)  
 2.  $\phi N_c(1) = \phi A_g f_y$  (Clause 7.2 of AS 4100)  
 3.  $\phi N_c(2) = \phi 0.85 A_g f_u$  (Clause 7.2 of AS 4100)

TABLE 6.1-5

**DESIGN CAPACITIES  
FOR MEMBERS SUBJECT TO  
AXIAL TENSION  
SQUARE HOLLOW SECTIONS  
GRADE C350**

Designation <i>d b t</i>	Mass per m	Axial Tension $\phi N_e(1)$		Gross Section Area $A_g$
		$\phi N_e(1)$	$\phi N_e(2)$	
<i>mm mm mm</i>	<i>kg/m</i>	<i>kN</i>	<i>kN</i>	<i>mm<sup>2</sup></i>
250 x 250 x 9.0 SHS 6.0 SHS	65.9 45.0	2650 1810	2760 1890	8400 5730
200 x 200 x 9.0 SHS 6.0 SHS 5.0 SHS	51.8 35.6 29.9	2080 1430 1200	2170 1490 1250	6600 4530 3810
150 x 150 x 9.0 SHS 6.0 SHS 5.0 SHS	37.7 26.2 22.1	1510 1050 886	1580 1100 926	4800 3330 2810
125 x 125 x 9.0 SHS 6.0 SHS 5.0 SHS 4.0 SHS	30.6 21.4 18.2 14.8	1230 861 729 593	1280 899 761 619	3900 2730 2310 1880
100 x 100 x 9.0 SHS 6.0 SHS 5.0 SHS 4.0 SHS 3.0 SHS	23.5 16.7 14.2 11.6 8.96	944 672 571 467 359	986 701 597 487 375	3000 2130 1810 1480 1140
89 x 89 x 6.0 SHS 5.0 SHS 3.5 SHS	14.6 12.5 9.06	588 502 363	614 524 379	1870 1590 1150
75 x 75 x 6.0 SHS 5.0 SHS 4.0 SHS 3.5 SHS 3.0 SHS 2.5 SHS	12.0 10.3 8.49 7.53 6.60 5.56	483 414 341 302 265 223	504 432 356 315 277 233	1530 1310 1080 959 841 709
65 x 65 x 6.0 SHS 5.0 SHS 4.0 SHS 3.0 SHS 2.5 SHS 2.0 SHS	10.1 8.75 7.23 5.66 4.78 3.88	407 351 290 227 192 156	425 367 303 237 200 162	1290 1110 921 721 609 494
50 x 50 x 5.0 SHS 4.0 SHS 3.0 SHS 2.5 SHS 2.0 SHS 1.6 SHS	6.39 5.35 4.25 3.60 2.93 2.38	256 215 170 145 118 95.5	268 224 178 151 123 99.7	814 681 541 459 374 303
40 x 40 x 4.0 SHS 3.0 SHS 2.5 SHS 2.0 SHS 1.6 SHS	4.09 3.30 2.82 2.31 1.88	164 133 113 92.5 75.3	171 138 118 96.6 78.7	521 421 359 294 239
35 x 35 x 3.0 SHS 2.5 SHS 2.0 SHS 1.6 SHS	2.83 2.42 1.99 1.63	114 97.3 79.9 65.3	119 102 83.5 68.1	361 309 254 207
30 x 30 x 3.0 SHS 2.5 SHS 2.0 SHS 1.6 SHS	2.36 2.03 1.68 1.38	94.8 81.6 67.3 55.2	99.0 85.2 70.3 57.6	301 259 214 175
25 x 25 x 3.0 SHS 2.5 SHS 2.0 SHS 1.6 SHS	1.89 1.64 1.36 1.12	75.9 65.8 54.7 45.1	79.2 68.7 57.1 47.1	241 209 174 143

Notes: 1.  $\phi = 0.9$  (Table 3.4 of AS 4100)  
2.  $\phi N_e(1) = \phi A_g f_y$  (Clause 7.2 of AS 4100)  
3.  $\phi N_e(2) = \phi 0.85 A_g f_u$  (Clause 7.2 of AS 4100)

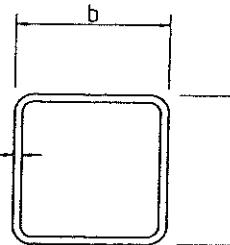


TABLE 6.1-6

**DESIGN CAPACITIES  
FOR MEMBERS SUBJECT TO  
AXIAL TENSION  
SQUARE HOLLOW SECTIONS  
GRADE C450**

Designation <i>d b t</i>	Mass per m	Axial Tension $\phi N_e(1)$		Gross Section Area $A_g$
		$\phi N_e(1)$	$\phi N_e(2)$	
<i>mm mm mm</i>	<i>kg/m</i>	<i>kN</i>	<i>kN</i>	<i>mm<sup>2</sup></i>
100 x 100 x 3.8 SHS 3.3 SHS 2.8 SHS 2.3 SHS	11.1 9.73 8.39 6.95	572 502 433 359	540 474 409 339	1410 1240 1070 885
75 x 75 x 3.3 SHS 2.8 SHS 2.3 SHS	7.14 6.19 5.14	368 319 265	348 302 251	909 788 655
65 x 65 x 2.8 SHS 2.3 SHS	5.31 4.42	274 228	259 215	676 563
50 x 50 x 2.8 SHS 2.3 SHS	3.99 3.34	206 172	194 163	508 425
40 x 40 x 2.8 SHS 2.3 SHS	3.11 2.62	161 135	152 127	396 333
35 x 35 x 2.8 SHS 2.3 SHS	2.67 2.25	138 116	130 110	340 287
30 x 30 x 2.8 SHS 2.3 SHS	2.23 1.89	115 97.7	109 92.3	284 241
25 x 25 x 2.3 SHS	1.53	79.1	74.7	195

Notes: 1.  $\phi = 0.9$  (Table 3.4 of AS 4100)  
2.  $\phi N_e(1) = \phi A_g f_y$  (Clause 7.2 of AS 4100)  
3.  $\phi N_e(2) = \phi 0.85 A_g f_u$  (Clause 7.2 of AS 4100)

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**PART 7****MEMBERS SUBJECT TO COMBINED ACTIONS****PAGE**

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**NOTE: SEE PAGE (ix) FOR THE SPECIFIC MATERIAL  
STANDARD REFERRED TO BY THE SECTION TYPE  
AND STEEL GRADE IN THESE TABLES**

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**PART 7 MEMBERS SUBJECT TO COMBINED ACTIONS****7.1 Scope**

This part of the Tables contains the interaction formulae which must be used to design members subject to combined actions in accordance with Section 8 of AS 4100.

Tables 7.3.2–1 to 7.3.2–6 give values of elastic buckling load ( $N_{om}$ ) for various effective lengths ( $L_e$ ) determined in accordance with Clause 4.6.3 of AS 4100.

**7.2 Method**

Section 7.3 describes the use and determination of moment amplification factors and the determination of the elastic buckling load for braced or sway members. The elastic buckling load is required for combined bending and axial compression when the moment is amplified by the moment amplification factors  $\delta_b$  and  $\delta_s$ .

Sections 7.4 and 7.5 give the interaction formulae for combined bending and axial compression and combined bending and axial tension respectively. Each section describes the method for uniaxial bending about the major principal x-axis, for uniaxial bending about the minor principal y-axis, and for biaxial bending. Section 7.6 gives the interaction formulae for biaxial bending without axial forces. In every case both the section capacity and the member capacity must be checked.

**7.3 Moment Amplification**

For a member subjected to combined bending and axial compression force, irrespective of whether that member is an isolated statically determinate member or part of a statically indeterminate frame, the bending moments will be amplified by the presence of axial compression force. Such amplification can be accounted for by a variety of means and these are now considered in relation to braced and sway members.

**Braced Member** – the member is braced such that its ends cannot move relative to one another.

If a first order elastic analysis is conducted then  $\delta_b$  (Clause 4.4.2.2 of AS 4100) must be used to amplify the design action effects between the ends of the member. However, when the moment amplification factor is greater than 1.4, a second order elastic analysis must be carried out (see Appendix E of AS 4100).

If an appropriate second order elastic analysis is carried out such that the design action effects at a sufficient number of locations between the ends of the member are determined then there is no need to modify the design action effects in the member using  $\delta_b$ .

The moment amplification factor ( $\delta_b$ ) must be calculated using the procedure shown in Figure 7.3(1)

**Sway Member** – the ends of the member are permitted to move relative to one another.

If a first order elastic analysis is carried out then the design action effects must be modified using the moment amplification factor ( $\delta_m$ ), which is the greater of  $\delta_b$  and  $\delta_s$  (see Clause 4.4.2.3 of AS 4100). However, when the moment amplification factor is greater than 1.4, a second order elastic analysis must be carried out (see Appendix E of AS 4100).

If an appropriate second order elastic analysis is carried out such that the design action effects at a sufficient number of locations along the length of the member are determined then there is no need to modify the design action effects using  $\delta_m$ . If this is not the situation, then the design action effects obtained from the second order elastic analysis may need to be modified using  $\delta_b$  as described in Appendix E of AS 4100.

The moment amplification factors ( $\delta_b$  and  $\delta_s$ ) must be calculated using the procedure shown in Figures 7.3(1) and 7.3(2).

### 7.3.1 Values of $c_m$

The value of  $c_m$  is specified in Clause 4.4.2.2 of AS 4100 as:

$$c_m = 0.6 - 0.4 \beta_m$$

where  $\beta_m$  is the ratio of the smaller to the larger bending moments at the ends of the member, taken as positive when the member is bent in reverse curvature. Where the member is subjected to transverse loading,  $\beta_m$  may be taken as described in Clauses 4.4.2.2(a), (b), and (c) of AS 4100. Table 7.3.1 gives values of  $c_m$  for a range of  $\beta_m$  values for single and reverse curvature bending.

### 7.3.2 Elastic Buckling Load

Values of elastic buckling load ( $N_{om}$ ) for various effective lengths ( $L_e$ ) are given in Tables 7.3.2–1 to 7.3.2–6.  $N_{om}$  values are determined in accordance with Clause 4.6.2 of AS 4100 as:

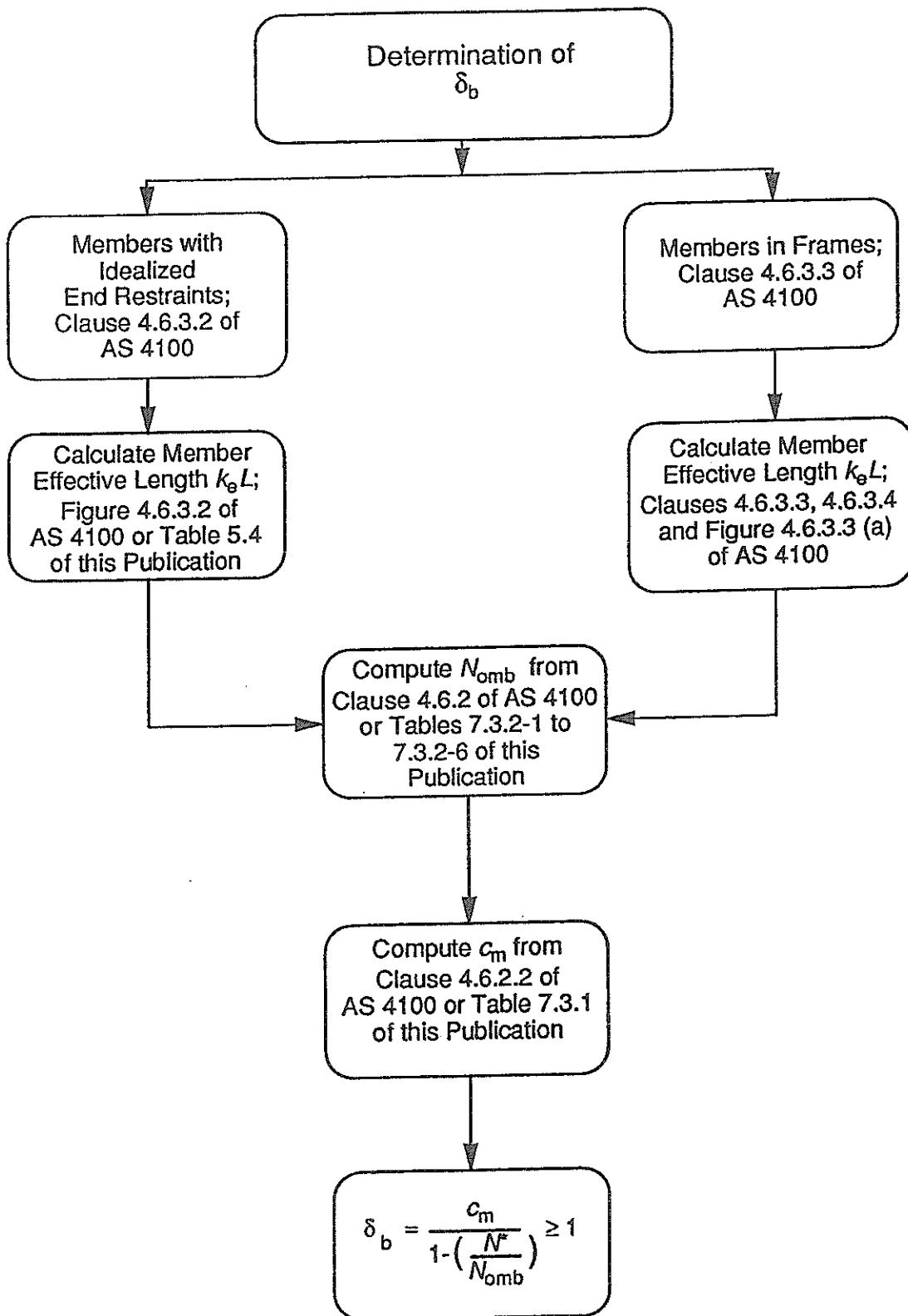
$$N_{om} = \frac{\pi^2 EI}{(k_e L)^2}$$

where  $E = 200 \times 10^3$  MPa

$I$  = second moment of area

$k_e L$  = effective length (Section 5.4 of these Tables)

" $N_{om}$  x-axis" indicates  $N_{om}$  for the member buckling about the x-axis. " $N_{om}$  y-axis" indicates  $N_{om}$  for the member buckling about the y-axis.



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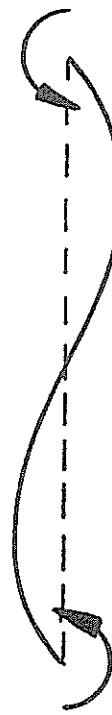
Table 7.3.1: Values of  $c_m$  for braced members

$\beta_m$	$c_m$	$\beta_m$	$c_m$	$\beta_m$	$c_m$	$\beta_m$	$c_m$
-1.00	1.00	-0.50	0.80	+0.05	0.58	+0.55	0.38
-0.95	0.98	-0.45	0.78	+0.10	0.56	+0.60	0.36
-0.90	0.96	-0.40	0.76	+0.15	0.54	+0.65	0.34
-0.85	0.94	-0.35	0.74	+0.20	0.52	+0.70	0.32
-0.80	0.92	-0.30	0.72	+0.25	0.50	+0.75	0.30
-0.75	0.90	-0.25	0.70	+0.30	0.48	+0.80	0.28
-0.70	0.88	-0.20	0.68	+0.35	0.46	+0.85	0.26
-0.65	0.86	-0.15	0.66	+0.40	0.42	+0.95	0.22
-0.55	0.82	-0.05	0.62	+0.50	0.40	+1.00	0.20
		0.00	0.60				

$\beta_m$  is negative for single curvature bending:



$\beta_m$  is positive for reverse curvature bending:



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### 7.4.1.2 Member Capacity

This section only applies to members analysed using an elastic method of analysis. Where there is sufficient restraint to prevent lateral buckling, only the in-plane requirements of Sections 7.4.1.1 and 7.4.1.2 need to be satisfied. If there is insufficient restraint to prevent lateral buckling, then both the in-plane and out-of-plane requirements of Sections 7.4.1.1 and 7.4.1.2 need to be satisfied.

#### a) In-plane capacity

$$\phi M_{ix} = \phi M_{sx} \left( 1 - \frac{N^*}{\phi N_{cx}} \right) \quad (\text{Clause 8.4.2.2 of AS 4100})$$

Note:  $N^* \leq \phi N_{cx}$

#### b) Out-of-plane capacity

$$\phi M_{ox} = \phi M_{bx} \left( 1 - \frac{N^*}{\phi N_{cy}} \right) \quad (\text{Clause 8.4.4.1 of AS 4100})$$

where  $\phi M_{bx}$  = design member moment capacity for bending about the major principal x-axis  
(see Section 4.1.2 and Tables 4.1-1 to 4.1-4)

Note:  $N^* \leq \phi N_{cy}$

### 7.4.2 Uniaxial Bending – about the minor principal y-axis

For a member subject to uniaxial bending about the minor principal y-axis and axial compression, the following condition must be satisfied:

$$M_y^* \leq \min[\phi M_{ry}; \phi M_y]$$

where  $M_y^*$  = design bending moment about the minor principal y-axis

$\phi$  = 0.9 (Table 3.4 of AS 4100)

$\phi M_{ry}$  = design section moment capacity ( $\phi M_s$ ) about the minor principal y-axis reduced by axial force (see section 7.4.2.1)

$\phi M_y$  = nominal in-plane member moment capacity ( $\phi M_i$ ) about the minor principal y-axis (see section 7.4.2.2)

#### 7.4.2.1 Section Capacity

The value of  $\phi M_{ry}$  must be determined at all points along the member and the minimum value is used to satisfy Section 7.4.2.

$$\phi M_{ry} = \phi M_{sy} \left( 1 - \frac{N^*}{\phi N_s} \right) \quad (\text{Clause 8.3.3 of AS 4100})$$

Note:  $N^* \leq \phi N_s$

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## 7.5 Combined Bending and Axial Tension

In this section:

$$\phi = 0.9 \quad (\text{Table 3.4 of AS 4100})$$

$\phi M_{sx}$  = design section moment capacity for bending about the major principal x-axis  
(see Section 3.2.3 and Tables 3.1–1 to 3.1–6)

$\phi M_{sy}$  = design section moment capacity for bending about the minor principal y-axis  
(see Section 3.2.3 and Tables 3.1–1 to 3.1–6)

$N^*$  = design axial compressive force

### 7.5.1 Uniaxial Bending - about the major principal x-axis

For a member subject to uniaxial bending about the major principal x-axis and axial tension, the following condition must be satisfied:

$$M_x^* \leq \min[\phi M_{rx}; \phi M_{ox}]$$

where  $M_x^*$  = design bending moment about the major principal x-axis

$\phi M_{rx}$  = design section moment capacity ( $\phi M_s$ ) for bending about the major principal x-axis reduced by axial force (see section 7.5.1.1)

$\phi M_{ox}$  = design out-of-plane member moment capacity ( $\phi M_o$ ) for bending about the major principal x-axis (see section 7.5.1.2)

#### 7.5.1.1 Section Capacity

The value of  $\phi M_{rx}$  must be determined at all points along the member and the minimum value is used to satisfy Section 7.5.1.

$$\phi M_{rx} = \phi M_{sx} \left( 1 - \frac{N^*}{\phi N_t} \right) \quad (\text{Clause 8.3.2 of AS 4100})$$

where  $\phi N_t$  = design section capacity in tension (see Section 3.2.1 and Tables 3.1–1 to 3.1–6)

Note:  $N^* \leq \phi N_t$

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### 7.5.3.2 Member Capacity

$$\left(\frac{M_x^*}{\phi M_{tx}}\right)^{1.4} + \left(\frac{M_y^*}{\phi M_{ty}}\right)^{1.4} \leq 1 \quad (\text{Clause 8.4.5.2 of AS 4100})$$

where  $M_x^*$  = design bending moment about the major principal x-axis

$\phi M_{tx}$  = lesser of the design section moment capacity ( $\phi M_{nx}$ ) reduced by axial tension and the design out-of-plane member moment capacity ( $\phi M_{ox}$ ) for bending about the major principal x-axis, determined in accordance with Sections 7.5.1.1 and 7.5.1.2 respectively

$M_y^*$  = design bending moment about the minor principal y-axis

$\phi M_{ty}$  = design section moment capacity reduced by axial tension, determined in accordance with Section 7.5.2

Note:  $M_x^* \leq \phi M_{tx}$

$M_y^* \leq \phi M_{ty}$

## 7.6 Biaxial Bending

For a member subject to biaxial bending without any axial force, both the conditions defined in Sections 7.6.1 and 7.6.2 must be satisfied.

### 7.6.1 Section Capacity

$$\frac{M_x^*}{\phi M_{sx}} + \frac{M_y^*}{\phi M_{sy}} \leq 1 \quad (\text{Clause 8.3.4 of AS 4100})$$

where  $M_x^*$  = design bending moment about the major principal x-axis

$\phi = 0.9$  (Table 3.4 of AS 4100)

$\phi M_{sx}$  = design section moment capacity for bending about the major principal x-axis (see Section 3.2.3 and Tables 3.1–1 to 3.1–6)

$M_y^*$  = design bending moment about the minor principal y-axis

$\phi M_{sy}$  = design section moment capacity for bending about the minor principal y-axis (see Section 3.2.3 and Tables 3.1–1 to 3.1–6)

Note:  $M_x^* \leq \phi M_{sx}$

$M_y^* \leq \phi M_{sy}$

### 7.6.2 Member Capacity

$$\left(\frac{M_x^*}{\phi M_{bx}}\right)^{1.4} + \left(\frac{M_y^*}{\phi M_{by}}\right)^{1.4} \leq 1 \quad (\text{Clause 8.4.5 of AS 4100})$$

where  $M_x^*$  = design bending moment about the major principal x-axis

$\phi = 0.9$  (Table 3.4 of AS 4100)

$\phi M_{bx}$  = design member moment capacity for bending about the major principal x-axis (see Section 4.1.2 and Tables 4.1–1 to 4.1–4)

$M_y^*$  = design bending moment about the minor principal y-axis

$\phi M_{by}$  = design section moment capacity for bending about the minor principal y-axis (see Section 3.2.3 and Tables 3.1–1 to 3.1–6)

Note:  $M_x^* \leq \phi M_{bx}$

$M_y^* \leq \phi M_{by}$

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From Figure 7.3(2) the moment amplification factor ( $\delta_b$ ) is given by:

$$\delta_b = \frac{c_m}{1 - \left( \frac{N^*}{N_{omb}} \right)}$$

- Considering flexural buckling about x-axis:  $\delta_{bx} = 0.614 \quad (< 1)$
- ∴ Maximum moment occurs at the ends, ie. at End A  $M_x^* = 50 \text{ kNm}$
- Considering flexural buckling about y-axis:  $\delta_{by} = 1.05 \quad (> 1 \text{ and } < 1.1)$
- ∴ Maximum moment occurs between the ends, ie. in the span  $M_y^* = 1.05 \times 5.0 = 5.25 \text{ kNm}$

2. Considering further Example 1 (Section 7.7), the adequacy of the member under the calculated design action effects is now checked as required by Clauses 8.3 and 8.4 of AS 4100.

### Design Data:

Section: 250 x 150 x 6.0 RHS – Grade C350 steel

Effective lengths: Flexural buckling (x-axis) = 4.0 m  
Flexural buckling (y-axis) = 4.0 m  
Lateral buckling = 6.0 m

Design action effects:  $N^* = 105 \text{ kN}$   
 $M_x^* = 50 \text{ kNm}$   
 $M_y^* = 5.25 \text{ kNm}$

Solution: The example involves biaxial bending and axial compression as defined in Section 7.4.3 of these Tables.

### (i) Section Capacity Check (Section 7.4.3.1)

From Table 3.1–3 we obtain:

$$\begin{aligned}\phi N_s &= 1290 \text{ kN} \\ \phi M_{sx} &= 118 \text{ kNm} \\ \phi M_{sy} &= 65.6 \text{ kNm}\end{aligned}$$

Thus, 
$$\frac{N^*}{\phi N_s} + \frac{M_x^*}{\phi M_{sx}} + \frac{M_y^*}{\phi M_{sy}} = \frac{105}{1290} + \frac{50}{118} + \frac{5.25}{65.6}$$

$$= 0.585 \quad (< 1.0 \therefore \text{O.K.})$$

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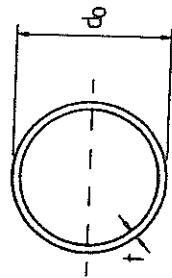
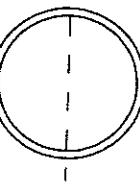


TABLE 7.3.2 – 1 (1)

**ELASTIC BUCKLING LOADS  
CIRCULAR HOLLOW SECTIONS: GRADE C250  
buckling about any axis**

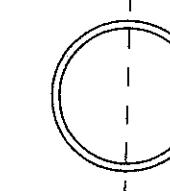


Designation $d_o$ mm	$t$ mm	Mass per m kg/m	Elastic Buckling Loads $N_{om}$ (MN)										
			0.0	1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	9.0	10.0
610.0 x 12.7 CHS	187	~	2100	525	233	131	84.0	58.3	42.8	32.8	25.9	21.0	14.6
9.5 CHS	141	~	1600	399	177	99.7	63.8	44.3	32.6	24.9	19.7	16.0	10.7
6.4 CHS	95.3	~	1090	273	121	68.2	43.6	30.3	22.3	17.0	13.5	10.9	11.1
508.0 x 12.7 CHS	155	~	1200	296	133	74.8	47.9	33.2	24.4	18.7	14.8	12.0	7.57
9.5 CHS	117	~	913	228	101	57.0	36.5	25.3	18.6	14.3	11.3	9.3	6.11
6.4 CHS	79.2	~	623	157	69.6	39.1	25.0	17.4	12.8	9.78	7.73	6.26	4.66
Note: 1. $N_{om} = \pi^2 EI / L_e^2$ (Clause 4.6.2 of AS 4100)													3.19
													2.45

Note: 1.  $N_{om} = \pi^2 EI / L_e^2$  (Clause 4.6.2 of AS 4100)

TABLE 7.3.2 – 1 (2)

**ELASTIC BUCKLING LOADS  
CIRCULAR HOLLOW SECTIONS: GRADE C250  
buckling about any axis**



Designation $d_o$ mm	$t$ mm	Mass per m kg/m	Elastic Buckling Loads $N_{om}$ (kN)											
			0.0	0.5	1.0	1.5	2.0	2.5	3.0	3.5	4.0	5.0	6.0	7.0
165.1 x 5.4 CHS	21.3	~	68300	17100	7590	4270	2730	1900	1390	1070	683	474	348	9.0
5.0 CHS	19.7	~	63700	15900	7080	3980	2550	1770	1300	995	637	442	325	211
139.7 x 5.4 CHS	17.9	~	40600	10200	4510	2540	1620	1130	829	593	406	282	207	197
5.0 CHS	16.6	~	37300	9490	4220	2370	1520	1050	774	539	379	263	194	148
114.3 x 5.4 CHS	14.5	~	21700	5420	2410	1350	867	602	442	339	217	151	111	125
4.5 CHS	12.2	~	18500	4630	2060	1160	740	514	378	289	185	128	94.4	117
101.6 x 5.0 CHS	11.9	~	14000	3500	1560	876	560	389	286	219	140	97.3	71.5	66.9
4.0 CHS	9.63	~	11600	2890	1280	722	462	321	236	180	116	80.2	58.9	57.1
86.9 x 5.9 CHS	12.1	~	10500	2630	1170	657	421	292	215	164	105	73.0	53.6	43.2
5.0 CHS	10.3	~	9190	2300	1020	574	368	255	188	144	91.9	63.8	46.9	35.6
4.0 CHS	8.38	~	7610	1900	845	475	304	211	155	119	76.1	52.8	38.8	29.7
76.1 x 5.9 CHS	10.2	~	6370	1590	708	398	255	177	130	99.6	63.7	44.3	32.5	24.9
4.5 CHS	7.95	~	5340	1290	571	321	206	143	105	80.3	51.4	35.7	26.2	20.1
3.6 CHS	6.44	~	4260	1070	474	267	171	118	87.0	66.6	42.6	29.6	21.8	15.9
Note: 1. $N_{om} = \pi^2 EI / L_e^2$ (Clause 4.6.2 of AS 4100)													16.7	13.2

Note: 1.  $N_{om} = \pi^2 EI / L_e^2$  (Clause 4.6.2 of AS 4100)

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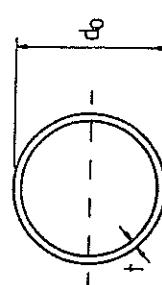
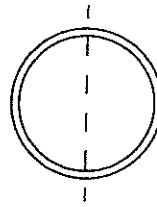


TABLE 7.3.2 - 2 (1)

**ELASTIC BUCKLING LOADS  
CIRCULAR HOLLOW SECTIONS: GRADE C350  
buckling about any axis**



Designation $d_o$ mm	Mass per m kg/m	Elastic Buckling Loads $N_{bm}$ (MN)												
		0.0	1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	9.0	10.0	12.0	14.0
Effective Length ( $L_e$ ) in metres														
457.0 x 12.7 CHS 9.5 CHS 6.4 CHS	139 105 71.1	864 660 454	216 165 113	96.0 73.4 50.4	54.0 41.3 28.4	34.6 26.4 18.2	24.0 18.3 12.6	17.6 13.5 9.26	13.5 10.3 7.09	10.7 8.15 5.60	8.64 6.60 4.54	6.00 4.41 3.37	4.41 3.38 2.58	
406.4 x 12.7 CHS 9.5 CHS 6.4 CHS	123 93.0 63.1	601 461 318	150 115 79.4	66.8 51.2 35.3	37.6 28.8 19.8	24.1 18.4 12.7	16.7 12.8 8.82	12.3 9.40 6.48	9.40 7.20 4.96	7.42 5.69 3.92	6.01 4.61 3.18	4.18 3.20 2.21	3.07 2.35 1.80	
355.6 x 12.7 CHS 9.5 CHS 6.4 CHS	107 81.1 55.1	397 306 211	99.4 76.4 52.8	44.2 33.9 23.5	24.8 19.1 13.2	15.9 12.2 8.45	11.0 8.49 5.87	8.11 6.24 4.31	6.21 4.77 3.30	4.91 3.77 2.61	3.97 3.06 2.12	2.03 1.62 1.47	1.55 1.24 1.08	
323.9 x 12.7 CHS 9.5 CHS 6.4 CHS	97.5 73.7 50.1	297 229 159	74.3 57.3 39.7	33.0 26.5 17.6	18.6 14.3 9.93	11.9 9.16 6.35	8.26 6.36 4.41	6.07 4.67 3.24	4.64 3.58 2.48	3.67 2.83 1.96	2.97 2.29 1.59	2.06 1.59 1.10	1.52 1.17 0.810	
273.1 x 9.3 CHS 6.4 CHS 4.8 CHS	60.5 42.1 31.8	133 94.2 71.9	33.1 23.5 18.0	14.7 10.5 7.99	8.28 5.89 4.49	5.30 3.77 2.88	3.68 2.62 2.00	2.70 1.92 1.47	2.07 1.47 1.12	1.64 1.16 0.887	1.33 0.942 0.719	0.820 0.634 0.499	0.676 0.480 0.367	
219.1 x 8.2 CHS 6.4 CHS 4.8 CHS	42.6 33.6 25.4	59.7 47.8 36.6	14.9 11.9 9.16	6.64 5.31 4.07	3.73 2.99 2.29	2.39 1.91 1.33	1.66 1.33 1.02	1.22 0.975 0.748	0.933 0.737 0.572	0.737 0.590 0.452	0.597 0.478 0.366	0.416 0.332 0.254	0.305 0.244 0.187	
168.3 x 7.1 CHS 6.4 CHS 4.8 CHS	28.2 25.6 19.4	23.1 21.1 16.3	5.77 5.27 4.07	2.57 2.34 1.81	1.44 1.32 1.02	0.924 0.843 0.651	0.642 0.586 0.452	0.471 0.430 0.352	0.361 0.329 0.234	0.285 0.260 0.201	0.231 0.211 0.153	0.160 0.146 0.113	0.118 0.0902 0.0830	

Note: 1.  $N_{bm} = \pi^2 EI / L_e^2$  (Clause 4.6.2 of AS4100)

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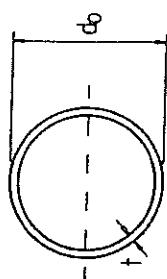
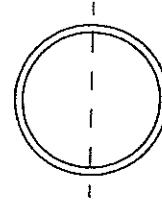


TABLE 7.3.2 – 2 (3)

**ELASTIC BUCKLING LOADS  
CIRCULAR HOLLOW SECTIONS: GRADE C350  
buckling about any axis**



Designation $d_o$ mm	Mass per m kg/m	Elastic Buckling Loads $N_{el}$ (kN)													
		Effective Length ( $L_e$ ) in metres													
		0.0	0.25	0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5	3.0	3.5	4.0	5.0
76.1 x 3.2 CHS 5.75	∞	15400	3850	1710	963	616	428	314	241	154	107	78.6	60.2	38.5	
2.3 CHS 4.19	∞	11500	2870	1280	717	459	319	234	179	115	79.7	58.6	44.8	28.7	
60.3 x 2.9 CHS 4.11	∞	6820	1700	758	426	273	189	139	107	68.2	47.4	34.8	26.6	17.0	
2.3 CHS 3.29	∞	5570	1390	619	348	223	155	114	87.1	55.7	38.7	28.4	21.8	13.9	
48.3 x 2.9 CHS 3.25	∞	3380	845	375	211	135	93.9	69.0	52.8	39.8	23.5	17.2	13.2	8.45	
2.3 CHS 2.61	∞	2780	696	309	174	111	77.3	56.8	43.5	27.8	19.3	14.2	10.9	6.96	
42.4 x 2.6 CHS 2.55	∞	2040	510	227	128	81.7	56.7	41.7	31.9	20.4	14.2	10.4	7.98	5.10	
2.0 CHS 1.99	∞	1640	410	182	102	65.6	45.5	33.5	25.6	16.4	11.4	8.37	6.40	4.10	
33.7 x 2.6 CHS 1.99	∞	977	244	109	61.0	39.1	27.1	19.9	15.3	9.77	6.78	4.98	3.82	2.44	
2.0 CHS 1.56	∞	793	198	88.1	49.6	31.7	22.0	16.2	12.4	7.93	5.51	4.05	3.10	1.98	
26.9 x 2.3 CHS 1.40	∞	428	107	47.6	26.8	17.1	11.9	8.74	6.69	4.28	2.97	2.19	1.67	1.07	
2.0 CHS 1.23	∞	385	96.4	42.8	24.1	15.4	10.7	7.87	6.02	3.85	2.68	1.97	1.51	0.984	
21.3 x 2.0 CHS 0.932	∞	180	45.1	20.0	11.3	7.21	5.01	3.68	2.82	1.80	1.25	0.920	0.704	0.451	

Note: 1.  $N_{el} = \pi^2 EI / L_e^2$  (Clause 4.6.2 of AS 4100)

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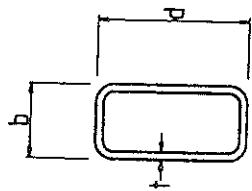
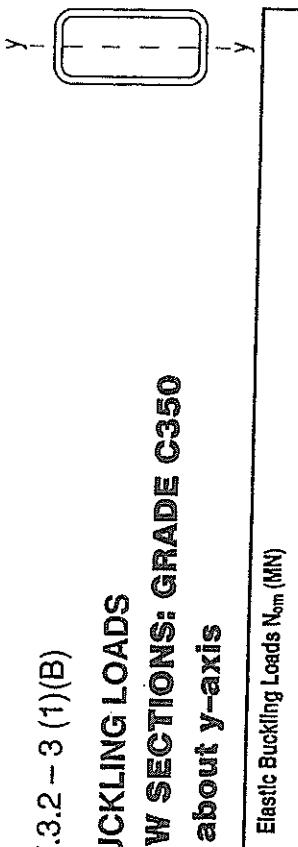


TABLE 7.3.2 – 3 (1)(B)

**ELASTIC BUCKLING LOADS  
RECTANGULAR HOLLOW SECTIONS: GRADE C350  
buckling about y-axis**



Designation	Mass per m	d mm	b mm	t mm	Elastic Buckling Loads N <sub>bm</sub> (MN)								
					0.0	0.5	1.0	1.5	2.0	2.5	3.0	3.5	4.0
250 x 150 x 9.0 RHS	51.8	~	192	48.0	21.3	12.0	7.68	5.34	3.92	3.00	1.92	1.33	0.980
6.0 RHS	35.6	~	138	34.8	15.4	8.64	5.53	3.84	2.62	2.16	1.38	0.950	0.750
5.0 RHS	29.9	~	118	29.5	13.1	7.38	4.72	3.28	2.41	1.85	1.18	0.820	0.540
200 x 100 x 9.0 RHS	37.7	~	60.4	15.1	6.71	3.77	2.41	1.68	1.23	0.943	0.604	0.419	0.236
6.0 RHS	26.2	~	44.9	11.2	4.99	2.81	1.80	1.25	0.917	0.702	0.449	0.312	0.176
5.0 RHS	22.1	~	36.8	9.70	4.31	2.43	1.55	1.08	0.792	0.606	0.398	0.270	0.152
4.0 RHS	17.9	~	32.2	8.04	3.57	2.01	1.29	0.893	0.656	0.503	0.322	0.223	0.126
150 x 100 x 6.0 RHS	21.4	~	34.5	8.61	3.83	2.15	1.38	0.957	0.703	0.538	0.345	0.239	0.176
5.0 RHS	18.2	~	28.9	7.47	3.32	1.87	1.20	0.830	0.610	0.467	0.299	0.208	0.153
4.0 RHS	14.8	~	24.9	6.22	2.76	1.55	0.985	0.691	0.508	0.389	0.249	0.173	0.127
150 x 50 x 5.0 RHS	14.2	~	6.04	1.51	0.671	0.378	0.242	0.168	0.123	0.0944	0.0604	0.0420	0.0236
4.0 RHS	11.6	~	5.16	1.29	0.573	0.322	0.205	0.143	0.105	0.0806	0.0516	0.0359	0.0151
3.0 RHS	8.96	~	4.16	1.04	0.462	0.260	0.166	0.115	0.0948	0.0650	0.0416	0.0289	0.0129
125 x 75 x 6.0 RHS	16.7	~	14.8	3.70	1.64	0.925	0.592	0.411	0.302	0.231	0.148	0.103	0.0578
5.0 RHS	14.2	~	13.0	3.25	1.44	0.812	0.520	0.361	0.255	0.203	0.130	0.093	0.0658
4.0 RHS	11.6	~	11.0	2.74	1.22	0.694	0.438	0.304	0.224	0.171	0.110	0.0761	0.0428
3.0 RHS	8.96	~	8.73	2.18	0.970	0.555	0.349	0.242	0.178	0.136	0.0873	0.0606	0.0341
100 x 50 x 6.0 RHS	12.0	~	4.48	1.12	0.498	0.280	0.179	0.124	0.0914	0.0700	0.0448	0.0311	0.0229
5.0 RHS	10.3	~	4.04	1.01	0.448	0.252	0.161	0.112	0.0824	0.0631	0.0404	0.0280	0.0216
4.0 RHS	8.49	~	3.48	0.870	0.387	0.218	0.139	0.0967	0.0710	0.0544	0.0348	0.0242	0.0158
3.5 RHS	7.53	~	3.18	0.790	0.351	0.197	0.126	0.0878	0.0645	0.0494	0.0316	0.0219	0.0136
3.0 RHS	6.60	~	2.85	0.712	0.316	0.178	0.114	0.0791	0.0581	0.0445	0.0285	0.0193	0.0123
2.5 RHS	5.56	~	2.45	0.613	0.222	0.153	0.0981	0.0600	0.0383	0.0245	0.0170	0.0115	0.00712
2.0 RHS	4.50	~	2.03	0.507	0.225	0.127	0.0811	0.0563	0.0317	0.0203	0.0141	0.0103	0.00613

Note: 1. N<sub>bm</sub> =  $\pi^2 EI / l_e^2$  (Clause 4.6.2 of AS4100)

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**ELASTIC BUCKLING LOADS  
RECTANGULAR HOLLOW SECTIONS: GRADE C450  
buckling about x-axis**

Designation d b t	Mass per m kg/m	Elastic Buckling Loads $N_{\text{om}}$ (kN)					
		0.0	0.5	1.0	1.5	2.0	2.5
mm mm mm							
125 x 75 x 3.8 RHS	11.1	~	23100	5780	1440	925	642
3.3 RHS	9.73	~	20550	5140	2280	822	571
2.8 RHS	8.39	~	18000	4510	2000	1130	722
2.3 RHS	6.95	~	15100	3780	1680	945	605
100 x 50 x 3.8 RHS	7.14	~	8910	2230	990	557	356
2.8 RHS	6.19	~	7830	1980	881	496	317
2.3 RHS	5.14	~	6700	1670	744	419	268

Note: 1.  $N_{\text{om}} = \pi^2 EI / L_e^2$  (Clause 4.6.2 of AS 4100)

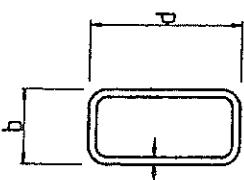
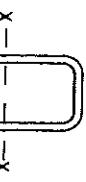
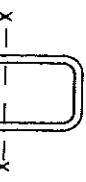


TABLE 7.3.2 - 4 (2)(A)

**ELASTIC BUCKLING LOADS  
RECTANGULAR HOLLOW SECTIONS: GRADE C450  
buckling about x-axis**

Designation d b t	Mass per m kg/m	Elastic Buckling Loads $N_{\text{om}}$ (kN)					
		0.0	0.25	0.5	0.75	1.0	1.25
mm mm mm							
100 x 50 x 3.3 RHS	7.14	~	35600	8910	3960	2230	1420
2.8 RHS	6.19	~	31700	7390	3550	1980	1270
2.3 RHS	5.14	~	26800	6700	2980	1670	1070
75 x 50 x 2.8 RHS	5.09	~	15600	3900	1730	974	623
2.3 RHS	4.24	~	13200	3310	1470	827	529
65 x 35 x 2.8 RHS	3.99	~	8450	2110	937	527	337
2.3 RHS	3.34	~	7230	1810	803	452	289
50 x 25 x 2.8 RHS	2.89	~	3370	844	375	211	135
2.3 RHS	2.44	~	2940	735	327	184	118
50 x 20 x 2.8 RHS	2.67	~	2830	720	320	180	115
2.3 RHS	2.25	~	2530	632	281	158	101

Note: 1.  $N_{\text{om}} = \pi^2 EI / L_e^2$  (Clause 4.6.2 of AS 4100)



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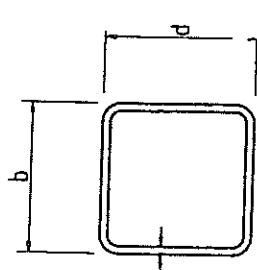
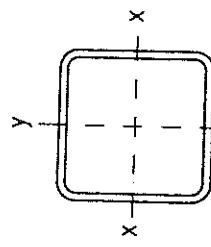


TABLE 7.3.2 - 5 (1)

**ELASTIC BUCKLING LOADS  
SQUARE HOLLOW SECTIONS: GRADE C350  
buckling about x- and y-axis**



Designation d b t mm mm mm	Mass kg/m	Elastic Buckling Loads Nom (kN) ( $\times 10^3$ )									
		Effective Length ( $L_e$ ) in metres									
250 x 250 x 9.0 SHS	65.9	0.0	1.0	1.5	2.0	2.5	3.0	3.5	4.0	5.0	6.0
250 x 250 x 9.0 SHS	65.9	88	158	70.0	39.4	25.2	17.5	12.9	9.85	6.30	4.38
250 x 250 x 9.0 SHS	60 SHS	45.0	88	111	49.3	27.8	17.8	12.3	9.06	6.94	4.44
200 x 200 x 9.0 SHS	51.8	88	77.3	34.4	19.3	12.4	8.59	6.31	4.83	3.09	2.15
200 x 200 x 9.0 SHS	50 SHS	35.6	88	55.3	24.6	13.8	8.85	6.14	4.51	3.46	2.21
150 x 150 x 9.0 SHS	37.7	88	30.3	13.5	7.59	4.86	3.37	2.48	1.90	1.21	0.843
150 x 150 x 9.0 SHS	50 SHS	26.2	88	22.3	9.90	5.57	3.56	2.47	1.82	1.39	0.891
125 x 125 x 9.0 SHS	30.6	88	16.5	7.35	4.13	2.65	1.84	1.35	1.03	0.662	0.459
125 x 125 x 9.0 SHS	60 SHS	21.4	88	12.4	5.52	3.10	1.99	1.38	1.01	0.776	0.496
125 x 125 x 9.0 SHS	50 SHS	18.2	88	10.7	4.77	2.69	1.72	1.19	0.877	0.671	0.430
125 x 125 x 9.0 SHS	40 SHS	14.8	88	8.92	3.96	2.23	1.48	0.991	0.728	0.557	0.387
100 x 100 x 9.0 SHS	23.5	88	7.71	3.43	1.93	1.23	0.857	0.629	0.482	0.308	0.214
100 x 100 x 9.0 SHS	60 SHS	16.7	88	5.99	2.66	1.50	0.959	0.666	0.489	0.375	0.240
100 x 100 x 9.0 SHS	50 SHS	14.2	88	5.24	2.33	1.31	0.839	0.583	0.428	0.328	0.210
100 x 100 x 9.0 SHS	40 SHS	11.6	88	4.40	1.96	1.10	0.704	0.489	0.359	0.275	0.176
100 x 100 x 9.0 SHS	30 SHS	8.96	88	3.49	1.55	0.874	0.559	0.388	0.285	0.218	0.140
89 x 89 x 6.0 SHS	14.6	88	4.06	1.80	1.01	0.649	0.451	0.331	0.254	0.162	0.113
89 x 89 x 6.0 SHS	5.0 SHS	12.5	88	3.58	1.59	0.894	0.572	0.397	0.292	0.223	0.143
89 x 89 x 6.0 SHS	3.5 SHS	9.06	88	2.71	1.21	0.678	0.434	0.301	0.221	0.170	0.109
75 x 75 x 6.0 SHS	12.0	88	2.29	1.02	0.571	0.366	0.254	0.187	0.143	0.0914	0.0635
75 x 75 x 6.0 SHS	5.0 SHS	10.3	88	2.04	0.906	0.510	0.326	0.227	0.166	0.127	0.0815
75 x 75 x 6.0 SHS	4.0 SHS	8.49	88	1.74	0.774	0.435	0.279	0.194	0.142	0.109	0.0697
75 x 75 x 6.0 SHS	3.5 SHS	7.53	88	1.57	0.699	0.393	0.252	0.175	0.128	0.0984	0.0629
75 x 75 x 6.0 SHS	3.0 SHS	6.60	88	1.41	0.628	0.353	0.226	0.157	0.115	0.0884	0.0555
75 x 75 x 6.0 SHS	2.5 SHS	5.55	88	1.21	0.539	0.303	0.194	0.135	0.0989	0.0757	0.0485

Note: 1.  $N_{\text{nom}} = \pi^2 EI / L_e^2$  (Clause 4.6.2 of AS 4100)

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**ELASTIC BUCKLING LOADS**  
**SQUARE HOLLOW SECTIONS: GRADE C450**  
 buckling about x- and y-axis

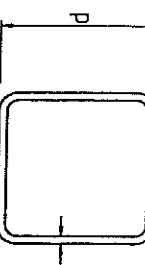
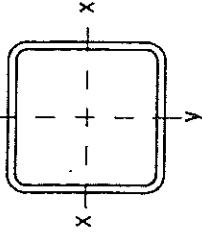


TABLE 7.3.2 - 6 (1)



Designation d b i	Mass perm kg/m	Elastic Buckling Loads $N_{om}$ (kN)							
		Effective Length ( $L_e$ ) in metres							
mm mm mm	0.0	0.5	1.0	1.5	2.0	2.5	3.0	3.5	4.0
100 x 100 x 3.8 SHS	11.1	~	16900	4220	1880	1050	675	469	344
3.3 SHS 9.73	~	18000	3750	1670	937	600	417	306	234
2.8 SHS 8.39	~	13200	3290	1460	822	526	365	288	208
2.3 SHS 6.95	~	11060	2760	1230	680	441	306	225	150
75 x 75 x 3.3 SHS	7.14	~	6010	1500	688	376	240	167	123
2.8 SHS 6.19	~	5340	1330	593	334	214	148	109	74.2
2.3 SHS 5.14	~	4510	1130	501	282	180	125	92.0	65.9

Note: 1.  $N_{om} = \pi^2 EI / L_e^2$  (Clause 4.6.2 of AS4100)

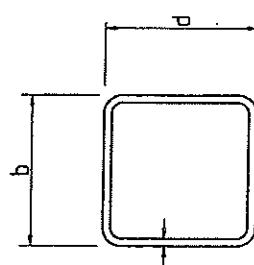
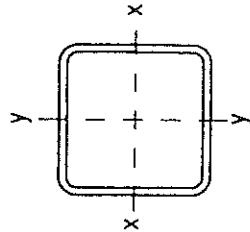


TABLE 7.3.2 - 6 (2)

**ELASTIC BUCKLING LOADS**  
**SQUARE HOLLOW SECTIONS: GRADE C450**  
 buckling about x- and y-axis



Designation d b i	Mass perm kg/m	Elastic Buckling Loads $N_{om}$ (kN)							
		Effective Length ( $L_e$ ) in metres							
mm mm mm	0.0	0.25	0.5	0.75	1.0	1.25	1.50	1.75	2.0
75 x 75 x 3.3 SHS	7.14	~	24000	6010	2670	1500	962	668	491
2.8 SHS 6.19	~	21400	5340	2370	1330	854	593	436	376
2.3 SHS 5.14	~	18000	4510	2000	1130	721	501	368	240
65 x 65 x 2.8 SHS	5.31	~	13800	3390	1510	848	543	377	277
2.3 SHS 4.42	~	11500	2880	1280	719	460	320	235	180
50 x 50 x 2.8 SHS	3.99	~	5840	1460	649	365	234	162	119
2.3 SHS 3.34	~	5010	1250	557	313	200	139	102	78.3
40 x 40 x 2.8 SHS	3.11	~	2810	703	312	176	112	78.1	57.4
2.3 SHS 2.62	~	2440	610	271	152	97.6	67.8	49.8	38.1
35 x 35 x 2.8 SHS	2.67	~	1800	450	200	112	72.0	50.0	36.7
2.3 SHS 2.25	~	1580	394	175	98.5	63.1	43.8	32.2	24.6
30 x 30 x 2.8 SHS	2.23	~	1070	266	118	66.6	42.6	29.6	21.8
2.3 SHS 1.89	~	945	236	105	59.1	37.8	26.3	19.3	14.8
25 x 25 x 2.8 SHS	1.53	~	510	127	56.6	31.9	20.4	14.2	10.4

Note: 1.  $N_{om} = \pi^2 EI / L_e^2$  (Clause 4.6.2 of AS4100)

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## PART 8 MAXIMUM DESIGN LOADS FOR BEAMS

	PAGE
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## TABLES

### TABLES 8.1–1 to 8.1–6

Maximum Design Loads for single span, simply supported beams

- with full lateral restraint
- deflection limited.....

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### TABLES 8.2–1 to 8.2–6

Maximum Design Loads for continuous, two span, simply supported beams

- with full lateral restraint
- deflection limited.....

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### TABLES 8.3–1 to 8.3–6

Maximum Design Loads for single span, fixed end beams

- with full lateral restraint
- deflection limited.....

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### TABLES 8.4–1 to 8.4–6

Maximum Design Loads for cantilever beams

- with full lateral restraint
- deflection limited.....

8-76

**NOTE: SEE PAGE (ix) FOR THE SPECIFIC MATERIAL  
STANDARD REFERRED TO BY THE SECTION TYPE  
AND STEEL GRADE IN THESE TABLES**

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**PART 8 MAXIMUM DESIGN LOADS FOR BEAMS****8.1 Scope**

PART 8 gives values of maximum design loads distributed uniformly along the length of the beam with full lateral restraint for different beam support conditions.

- Tables 8.1 for single span, simply supported beams
- Tables 8.2 for continuous two span, simply supported beams
- Tables 8.3 for single span, fixed end beams
- Tables 8.4 for cantilever beams

Each group of tables is separated into two series:

- the (A) series (e.g. Table 8.1–1(A)) for the strength limit state
- the (B) series (e.g. Table 8.1–1(B)) for the serviceability limit state

For each group of tables, the (A) series tables are immediately followed by the (B) series tables. The design load ( $W^*$  = total design load) is assumed to be uniformly distributed and applied through the shear centre in the direction of the principal  $y$ -axis.

**NOTE – BEAM SELF WEIGHT:** For all tables, the self weight of the beam has NOT been deducted. The designer must include the self weight as part of the dead load when determining the maximum design load  $W_L^*$  or  $W_S^*$ .

**8.2 Method**

The maximum design load is the lesser of the strength limit state design load given in the (A) series tables determined in accordance with Section 8.2.1, and the serviceability limit state design load given in the (B) series tables determined and adjusted if necessary in accordance with Section 8.2.2.

**8.2.1 Strength Limit State Design**

The value of the maximum design load ( $W_L^*$ ) given in the tables is the lesser of the maximum design load ( $W_{L1}^*$ ) associated with the design section moment capacity ( $\phi M_{sx}$ ) and the maximum design load ( $W_{L2}^*$ ) associated with the design shear capacity ( $\phi V_{vx}$ ).

$$W_L^* = \text{Min.}[W_{L1}^* ; W_{L2}^*]$$

The method is illustrated in Section 8.2.1.1 for the case of a simply supported beam.

**Note:** the interaction of shear and bending has not been included in the tables.  
If  $V^* < 0.6 \phi V_v$  or if  $M^* < 0.75 \phi M_b$  then no interaction check is necessary.  
Otherwise reference should be made to Section 8.4.

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## 8.2.2 Serviceability Limit State Design

The value of serviceability load ( $W_s^*$ ) given in the tables is the maximum design load which will achieve a calculated total elastic deflection of  $L / 250$  (where  $L$  is the span of the beam).

For deflection limits other than  $span / 250$ , the value of  $W_{s2}^*$  for the alternative deflection limit may be calculated from the tabulated value  $W_{s1}^*$  using the formula:

$$W_{s2}^* = \frac{250W_{s1}^*}{D}$$

where  $D$  = the denominator value in the deflection limit incorporating the span term  
(e.g.  $D = 500$  for the  $L / 500$  deflection limit)

For sections with a high shape factor (ratio of plastic moment to the yield moment of a beam) it may be possible for the maximum stresses in a member to reach the yield stress at serviceability loads without exceeding the strength limit state. This will invalidate the deflection calculations based on the assumption of elastic behaviour. However it has been found that for the hollow sections contained in these tables, using the load factors in AS 1170 and AS 4100, the strength limit state will always be exceeded before first yield occurs. Therefore values of the load at which first yield occurs have not been included in the tables.

The method is illustrated below for the case of a simply supported beam.

### $W_{s1}^*$ – based on a Deflection Limit of $L / 250$ (simply supported beam)

For a simply supported beam subject to a uniformly distributed load, the maximum deflection ( $\Delta_{max}$ ) is given by:

$$\Delta_{max} = \frac{5WL^3}{384EI_x}$$

where  $W$  = total uniformly distributed load

$L$  = length of span

$E$  =  $200 \times 10^3$  MPa

$I_x$  = second moment of area about the major principal x-axis

Therefore, substituting  $\Delta_{max} = L / 250$  and rearranging the equation gives the serviceability limit state maximum design load ( $W_s^*$ ):

$$W_s^* = \frac{384EI_x}{1250L^2}$$

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## 8.5 Other Load Conditions

The values given in Tables 8.1–1 to 8.1–6 are for single span, simply supported beams subject to uniformly distributed loads. However, the information presented in these tables may be used for other loading situations, for beams with full lateral restraint and  $\alpha_m = 1.0$ , using the equivalent uniform design loads given in Table 8.5 and in conjunction with the following procedure:

- (1) Calculate equivalent uniformly distributed maximum design load for moment ( $W_{EM}^*$ ) using Table 8.5.
- (2) Based on  $W_{EM}^*$  select a section with an adequate maximum design load ( $W_{L1}^*$ ) associated with the design shear capacity from Tables 8.1–1(A) to 8.1–6(A).
- (3) Calculate equivalent uniformly distributed maximum design load for shear ( $W_{EV}^*$ ) using Table 8.5.
- (4) Check that the section selected in (2) has an adequate maximum design load ( $W_{L2}^*$ ) associated with the design shear capacity to resist  $W_{EV}^*$ . If not, select a new section size which can resist  $W_{EV}^*$ .
- (5) Check shear and bending interaction in accordance with Section 4.2.3. A check is necessary if the design shear is less than 60% of the design shear capacity ( $V^* < 0.6 \phi V_v$ ) or if  $M^* < 0.75 \phi M_b$ .
- (6) Calculate equivalent uniformly distributed serviceability load ( $W_{ES}^*$ ) using Table 8.5.
- (7) Check that the section selected in (4) has an adequate maximum serviceability design load ( $W_{S1}^*$ ) to resist  $W_{ES}^*$ . If not, select a new section size which can resist  $W_{ES}^*$ .

## 8.6 Examples

1. A simply supported beam of 4 metres is subjected to uniformly distributed loads of:

$$G \text{ (Dead Load)} = 50 \text{ kN} \quad (\text{total load})$$

$$Q \text{ (Live Load)} = 64 \text{ kN} \quad (\text{total load})$$

The beam is continuously laterally supported. The total deflection of the beam under serviceability load must not exceed  $L / 250$ .

### Solution:

- (a) Calculation of maximum design loads:

$$\begin{aligned} \text{Strength limit state} \quad W_L^* &= 1.25G + 1.5Q \\ &= 159 \text{ kN} \end{aligned}$$

$$\begin{aligned} \text{Serviceability limit state} \quad W_S^* &= G + 0.7Q \quad (\text{short term live load}) \\ &= 94.8 \text{ kN} \end{aligned}$$

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## (b) Use of the Tables:

Strength Limit State – Select a section from the Tables such that the maximum design loads  $W_{L1}^*$  and  $W_{L2}^*$  are greater than or equal to  $W_L^*$ . It can be seen from Table 8.1–3(1)(A) that for a 250 x 150 x 5.0 RHS – Grade C350, the maximum design loads are:

$$W_{L1}^* = 189 \text{ kN}$$

$$W_{L2}^* = 873 \text{ kN}$$

$$\therefore W_L^* = 189 \text{ kN} \quad (> 159 \text{ kN})$$

Therefore, a 250 x 150 x 5.0 RHS has adequate strength.

Serviceability Limit State – From Table 8.1–3(1)(B) it can be seen that for a 250 x 150 x 5.0 RHS – Grade C350, the serviceability load is:

$$W_S^* = 126 \text{ kN} \quad (> 94.8 \text{ kN})$$

The most efficient and practical hollow sections for this application are RHS and SHS. The alternative sections which satisfy the above strength and serviceability limit states are listed below:

250 x 150 x 5.0 RHS – Grade C350	29.9 kg/m
200 x 200 x 6.0 SHS – Grade C350	35.6 kg/m

Therefore, based on mass, select 250 x 150 x 5.0 RHS – Grade C350.

2. A beam which is simply supported has a span of 6.0 metres with full lateral restraint. The beam is subjected to nominal central dead and short term live loads of 10 kN and 25 kN respectively. Design a suitable RHS in Grade C350 steel with no limit on deflection of span / 250.

Solution:

- (1) Calculate equivalent uniformly distributed maximum design load for moment ( $W_{EM}^*$ )

From Table 8.5 ( $W_{EM}^*$ ) associated with the central dead and live loads is:

$$\begin{aligned} W_{EM}^* &= 2P \\ &= 2(1.25 \times 10 + 1.5 \times 25) \\ &= 100 \text{ kN} \end{aligned}$$

- (2) Based on  $W_{EM}^*$  select a section with an adequate maximum design load ( $W_L^*$ ) based on design moment capacity

From Table 8.1–3(1)(A), a 250 x 150 x 5.0 RHS has adequate maximum design load ( $W_{L1}^* = 126 \text{ kN}$ ).

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**FOR SIMPLY SUPPORTED BEAMS WITH FULL LATERAL RESTRAINT  
CIRCULAR HOLLOW SECTIONS: GRADE C250  
bending about any axis**

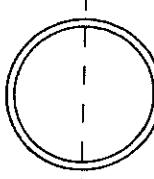


TABLE 8.1 - 1 (1)(A)

$W_{L1}$  = Maximum Design Load based on Design Moment Capacity  
 $W_{L2}$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L^*$  is LESSER of  $W_{L1}$  and  $W_{L2}$

Designation $d_0$ mm	Mass per m kg/m	Span of Beam (L) in metres						$W_L^*$ (kN)
		1.0	2.0	3.0	4.0	5.0	6.0	
610.0 x 12.7 CHS	187	8160	4080	2220	2040	1630	1170	1020
9.5 CHS	141	5880	2940	1960	1470	1180	980	840
6.4 CHS	95.3	3590	1800	1200	898	718	599	513
508.0 x 12.7 CHS	155	5610	2800	1870	1400	1120	935	801
9.5 CHS	117	4200	2100	1400	1050	840	700	600
6.4 CHS	79.2	2630	1310	875	656	525	438	375

Notes: Refer to TABLE 8.1-1 (3)(A)

TABLE 8.1 - 1 (2)(A)

$W_{L1}$  = Maximum Design Load based on Design Moment Capacity  
 $W_{L2}$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L^*$  is LESSER of  $W_{L1}$  and  $W_{L2}$

Designation $d_0$ mm	Mass per m kg/m	Span of Beam (L) in metres						$W_L^*$ (kN)
		0.5	1.0	1.5	2.0	2.5	3.0	
165.1 x 5.4 CHS	21.3	496	248	165	124	99.2	82.7	62.0
5.0 CHS	19.7	462	231	154	115	92.3	76.9	65.9
139.7 x 5.4 CHS	17.9	351	175	117	87.7	70.2	58.5	50.1
5.0 CHS	16.6	327	163	109	81.7	65.3	54.5	46.7
114.3 x 5.4 CHS	14.5	231	115	76.9	57.7	46.1	38.5	32.7
4.5 CHS	12.2	195	97.7	65.1	48.9	39.1	32.6	27.9
101.6 x 5.0 CHS	11.9	168	84.1	56.0	42.0	33.6	28.0	24.0
4.0 CHS	9.63	137	68.6	45.7	34.3	27.4	22.9	19.6
88.9 x 5.9 CHS	12.1	147	73.3	48.9	36.6	29.3	24.4	20.9
5.0 CHS	10.3	127	63.4	42.3	31.7	25.4	21.1	18.1
4.0 CHS	8.38	104	51.9	34.6	26.0	20.8	17.3	14.8
76.1 x 5.9 CHS	10.2	105	52.5	35.0	26.2	21.0	17.5	15.0
4.5 CHS	7.95	89.2	41.6	27.7	20.8	16.6	13.9	11.9
3.6 CHS	6.44	68.2	34.1	22.7	17.0	13.6	11.4	9.74

Notes: Refer to TABLE 8.1-1 (3)(A)

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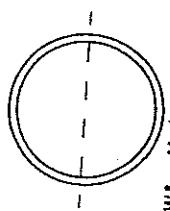
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**TABLE 8.1 – 1 (3)(A)**

**FOR SIMPLY SUPPORTED BEAMS WITH FULL LATERAL RESTRAINT**

**CIRCULAR HOLLOW SECTIONS: GRADE C250**

bending about any axis



$W_1^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_2^*$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_1^*$  LESSER of  $W_1^*$  and  $W_2^*$

Designation $d_o$	Mass per m	mm mm	Span of Beam (L) in metres						$W_1^*$					
			0.25	0.5	0.75	1.0	1.25	1.5						
76.1 x 59 CHS	10.2	210	105	69.9	52.5	42.0	35.0	30.0	26.2	21.0	17.5	13.1	10.5	5.0
4.5 CHS	7.95	166	83.2	55.4	41.6	33.3	27.7	23.8	20.8	16.6	13.9	11.9	10.4	21.1
3.6 CHS	6.44	136	68.2	45.5	34.1	27.3	22.7	19.5	17.0	13.6	11.4	9.74	8.32	16.4
60.3 x 54 CHS	7.31	118	58.8	39.2	29.4	23.5	19.6	16.8	14.7	11.8	9.80	8.40	7.35	6.82
4.5 CHS	6.19	101	50.6	33.7	25.3	20.2	16.9	14.4	12.6	10.1	8.43	7.22	6.32	5.06
3.6 CHS	5.03	83.4	41.7	27.8	20.9	16.7	13.9	11.9	10.4	8.34	6.95	5.96	5.22	128
48.3 x 54 CHS	5.71	71.9	36.0	24.0	18.0	14.4	12.0	10.3	8.99	7.19	5.99	5.14	4.50	104
4.0 CHS	4.37	56.7	28.3	18.9	14.2	11.3	9.45	8.10	7.08	5.67	4.72	4.05	3.54	151
3.2 CHS	3.86	46.9	23.5	15.6	11.7	9.39	7.82	6.71	5.87	4.69	3.91	3.35	2.93	235
42.4 x 4.9 CHS	4.53	49.9	24.9	16.6	12.5	9.98	8.32	7.13	6.24	4.99	4.16	3.56	3.60	118
4.0 CHS	3.79	42.6	21.3	14.2	10.7	8.52	7.10	6.09	5.33	4.26	3.55	3.04	2.66	90.2
3.2 CHS	3.09	35.5	17.7	11.8	8.87	7.10	5.91	5.07	4.44	3.55	2.96	2.53	2.22	73.4
33.7 x 4.5 CHS	3.24	27.8	13.9	9.28	6.96	5.57	4.64	3.98	3.48	2.78	2.32	1.99	1.74	2.49
4.0 CHS	2.93	25.6	12.8	8.52	6.39	5.11	4.25	3.65	3.19	2.56	2.13	1.83	1.60	93.5
3.2 CHS	2.41	21.5	10.8	7.17	5.38	4.30	3.59	3.07	2.69	2.15	1.79	1.54	1.34	60.5
26.9 x 4.0 CHS	2.26	15.3	7.83	5.09	3.81	3.05	2.54	2.18	1.91	1.53	1.27	1.09	0.954	49.7
3.2 CHS	1.87	13.0	6.51	4.34	3.25	2.60	2.17	1.86	1.63	1.30	1.08	0.930	0.814	78.2
2.6 CHS	1.56	11.1	5.55	3.70	2.77	2.22	1.85	1.59	1.39	1.11	0.925	0.793	0.694	63.8
21.3 x 3.6 CHS	1.57	8.23	4.12	2.74	2.06	1.65	1.37	1.18	1.03	0.823	0.686	0.588	0.515	32.4
3.2 CHS	1.43	7.63	3.81	2.54	1.91	1.53	1.27	1.09	0.953	0.763	0.636	0.545	0.477	29.5
2.6 CHS	1.20	6.59	3.29	2.20	1.65	1.32	1.10	0.941	0.824	0.659	0.549	0.471	0.412	24.7

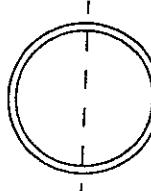
- Notes:
- $\phi = 0.9$
  - $\alpha_m = 1.0$
  - $\alpha_s = 1.0$
  - $W_1^* = 8\phi M_s/L$
  - $W_2^* = 2\phi V_y$

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TABLE 8.1 – 2 (1)(A)

**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR SIMPLY SUPPORTED BEAMS WITH FULL LATERAL RESTRAINT  
CIRCULAR HOLLOW SECTIONS: GRADE C350**



$W_L^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_2^*$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L^*$  is LESSER of  $W_L^*$  and  $W_2^*$

Designation $d_o$ mm	Mass per m kg/m	Span of Beam ( $L$ ) in metres								$W_2^*$ kN				
		1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	9.0	10.0	12.0	14.0	16.0
457.0 x 12.7 CHS 9.5 CHS 6.4 CHS	139	6310	3160	2100	1580	1260	1050	902	769	701	631	526	451	394
	105	4520	2260	1510	1130	904	753	646	565	502	452	377	323	283
	71.1	2730	1370	916	687	550	458	393	343	305	275	229	196	172
406.4 x 12.7 CHS 9.5 CHS 6.4 CHS	123	4960	2480	1650	1240	992	827	709	620	551	496	414	354	310
	93.0	3650	1820	1220	912	730	608	521	456	405	365	304	261	228
	63.1	2260	1130	752	564	451	376	322	282	251	226	188	161	141
355.6 x 12.7 CHS 9.5 CHS 6.4 CHS	107	3760	1880	1250	941	753	627	538	471	418	376	314	269	235
	81.1	2850	1420	948	711	569	474	406	356	316	285	237	203	178
	55.1	1790	894	596	447	358	298	255	224	199	179	149	128	112
323.9 x 12.7 CHS 9.5 CHS 6.4 CHS	97.5	3100	1550	1030	775	620	517	443	388	345	310	258	222	194
	73.7	2370	1180	789	592	473	395	338	296	263	237	197	169	148
	50.1	1510	757	505	379	303	252	216	189	168	151	125	108	94.7
273.1 x 9.3 CHS 6.4 CHS 4.8 CHS	60.5	1630	816	544	408	326	272	233	204	181	163	136	117	102
	42.1	1110	555	370	278	222	185	159	139	123	111	92.5	79.3	69.4
	31.8	787	393	262	197	157	131	112	98.3	87.4	78.7	65.6	56.2	49.2
219.1 x 8.2 CHS 6.4 CHS 4.8 CHS	42.6	920	460	307	230	184	153	131	115	102	92.0	76.6	65.7	57.5
	33.6	730	365	243	182	146	122	104	91.2	81.1	73.0	60.8	52.1	45.6
	25.4	530	265	177	133	106	88.3	75.7	66.3	58.9	53.0	44.2	37.9	33.1
168.3 x 7.1 CHS 6.4 CHS 4.8 CHS	28.2	465	233	155	116	93.0	77.5	66.5	58.2	51.7	46.5	38.8	33.2	29.1
	25.6	423	211	141	106	84.6	70.5	60.4	52.9	47.0	42.3	35.2	30.2	26.4
	19.4	323	162	108	80.9	64.7	53.9	46.2	40.4	35.9	32.3	27.0	23.1	20.2

Notes:

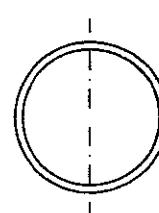
1.  $\phi = 0.9$
2.  $\alpha_m = 1.0$
3.  $\alpha_c = 1.0$
4.  $W_L^* = 8\phi M_s/L$
5.  $W_2^* = 2\phi V_y$

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TABLE 8.1 – 2 (2)(A)

**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR SIMPLY SUPPORTED BEAMS WITH FULL LATERAL RESTRAINT  
CIRCULAR HOLLOW SECTIONS: GRADE C350**



$W_L^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_L^2$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L^*$  is LESSER of  $W_L^*$  and  $W_L^2$

Designation $d_o$ mm	Mass per m kg/m	Span of Beam (L) In metres							$W_L^2$ kN
		0.5	1.0	1.5	2.0	2.5	3.0	3.5	
168.3 x 7.1 CHS	28.2	930	465	310	233	186	155	133	93.0
6.4 CHS	25.6	846	423	282	211	169	141	121	84.6
4.8 CHS	19.4	647	323	216	162	129	108	92.4	80.9
165.1 x 3.5 CHS	13.9	436	218	145	109	87.3	72.7	62.3	54.5
3.0 CHS	12.0	362	181	121	90.6	72.4	60.4	51.7	45.3
139.7 x 3.5 CHS	11.8	321	160	107	80.2	64.2	53.5	45.8	40.1
3.0 CHS	10.1	268	134	89.5	67.1	53.7	44.7	38.3	33.6
114.3 x 6.0 CHS	16.0	365	178	118	88.8	71.0	59.2	50.7	44.4
4.8 CHS	13.0	280	145	96.8	72.6	58.1	48.4	41.5	36.3
3.6 CHS	9.83	222	111	74.1	55.6	44.5	37.1	31.8	27.8
3.2 CHS	8.77	199	99.6	66.4	49.8	39.8	33.2	28.4	24.9
101.6 x 3.2 CHS	7.77	156	78.1	52.1	39.1	31.2	26.0	22.3	19.5
2.6 CHS	6.35	126	63.2	42.1	31.6	25.3	21.1	18.1	15.8
88.9 x 5.5 CHS	11.3	193	96.5	64.4	48.3	38.6	32.2	27.6	24.1
4.8 CHS	9.96	171	85.6	57.1	42.8	34.3	28.5	24.5	21.4
3.2 CHS	6.76	119	59.3	39.5	29.6	23.7	19.8	16.9	14.8
2.6 CHS	5.53	97.6	48.8	32.5	24.4	19.5	16.3	13.9	12.2
76.1 x 3.2 CHS	5.75	85.8	42.9	28.6	21.4	17.2	14.3	12.3	10.7
2.3 CHS	4.19	63.2	31.6	21.1	15.8	12.6	10.5	9.02	7.89

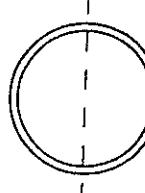
- Notes:  
 1.  $\phi = 0.9$   
 2.  $c_m = 1.0$   
 3.  $c_s = 1.0$   
 4.  $W_L^* = 8\phi M_s/L$   
 5.  $W_L^2 = 2\phi V_y$

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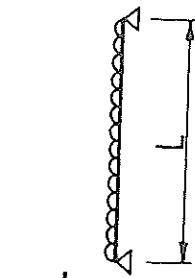
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TABLE 8.1 – 2 (3)(A)

**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR SIMPLY SUPPORTED BEAMS WITH FULL LATERAL RESTRAINT  
CIRCULAR HOLLOW SECTIONS: GRADE C350  
bending about any axis**



$W_L^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_2$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_2^*$  is LESSER of  $W_L^*$  and  $W_2$



Designation $d_0$ mm	Mass per m kg/m	Span of Beam ( $L$ ) In metres	$W_L^*$ (kN)						
			0.25	0.5	0.75	1.0	1.25	1.5	1.75
76.1 x 3.2 CHS	5.75	172	85.8	57.2	42.9	34.3	28.6	24.5	21.4
2.3 CHS 4.19	126	63.2	42.1	31.6	25.3	21.1	18.0	15.8	12.6
60.3 x 2.9 CHS	4.11	96.4	48.2	32.1	24.1	19.3	16.1	13.8	12.0
2.3 CHS 3.29	78.0	39.0	26.0	19.5	15.6	13.0	11.1	9.75	7.80
48.3 x 2.9 CHS	3.25	60.3	30.2	20.1	15.1	12.1	10.1	8.62	7.54
2.3 CHS 2.61	49.1	24.5	16.4	12.3	9.62	8.18	7.01	6.14	4.91
42.4 x 2.6 CHS	2.55	41.6	20.8	13.9	10.4	8.31	6.93	5.94	5.20
2.0 CHS 1.99	32.9	16.5	11.0	8.23	6.59	5.49	4.70	4.12	3.29
33.7 x 2.6 CHS	1.99	25.4	12.7	8.47	6.35	5.08	4.23	3.63	3.18
2.0 CHS 1.56	20.3	10.1	6.76	5.07	4.06	3.38	2.90	2.54	2.03
26.9 x 2.3 CHS	1.40	14.1	7.04	4.69	3.52	2.81	2.35	2.01	1.76
2.0 CHS 1.23	12.5	6.28	4.18	3.13	2.51	2.09	1.79	1.57	1.25
21.3 x 2.0 CHS	0.952	7.54	3.77	2.51	1.88	1.51	1.26	1.08	0.92

- Notes:  
 1.  $\phi = 0.9$   
 2.  $\alpha_m = 1.0$   
 3.  $\alpha_s = 1.0$   
 4.  $W_L^* = 8\phi M_s / L$   
 5.  $W_2^* = 2\phi V_y$

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TABLE 8.1-3 (1)(A)

**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR SIMPLY SUPPORTED BEAMS WITH FULL LATERAL RESTRAINT  
RECTANGULAR HOLLOW SECTIONS: GRADE C350  
bending about x-axis**

$W_{L1}^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_{L2}^*$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L$  is LESSER of  $W_{L1}$  and  $W_{L2}$

Designation	Mass per m	W <sub>L1</sub> (kN)										W <sub>L2</sub>	FLR				
		Span of Beam (L) in metres	1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	9.0	10.0	11.0	12.0	13.0	14.0	
250 x 150 x 9.0 RHS	51.8	1340	672	448	336	269	224	192	168	149	134	122	112	103	96.0	1520	15.6
6.0 RHS	35.6	943	471	314	236	189	157	135	118	105	94.3	85.7	78.6	72.5	67.4	1040	16.0
5.0 RHS	29.9	755	378	252	189	151	126	108	94.4	83.9	75.5	68.7	62.9	58.1	53.9	873	16.1
200 x 100 x 9.0 RHS	37.7	739	370	246	185	148	123	106	92.4	82.2	73.9	67.2	61.6	56.9	52.8	1180	8.55
6.0 RHS	26.2	530	265	177	132	106	88.3	75.7	66.2	58.8	53.0	46.1	44.1	40.7	37.8	812	8.86
5.0 RHS	22.1	452	226	151	113	90.3	75.3	64.5	56.5	50.2	45.2	41.1	37.6	34.7	32.3	684	8.96
4.0 RHS	17.9	370	185	123	92.4	73.9	61.6	52.8	46.2	41.1	37.0	33.6	30.8	28.4	26.4	553	9.06
150 x 100 x 6.0 RHS	21.4	339	169	113	84.6	67.7	56.4	48.4	42.3	37.6	33.9	30.8	28.2	26.0	24.2	606	11.4
5.0 RHS	18.2	290	145	96.7	72.5	58.0	48.3	41.4	36.3	32.2	29.0	26.4	24.2	22.3	20.7	512	11.6
4.0 RHS	14.8	238	119	79.5	59.6	47.7	39.7	34.1	29.8	26.5	23.8	21.7	19.9	18.3	17.0	416	11.7
150 x 50 x 5.0 RHS	14.2	199	99.4	66.2	49.7	39.7	33.1	28.4	24.8	22.1	19.9	18.1	16.6	15.3	14.2	492	2.93
4.0 RHS	11.6	165	82.4	55.0	41.2	33.0	27.5	23.6	20.6	18.3	16.5	15.0	13.7	12.7	11.8	399	3.00
3.0 RHS	8.96	130	64.8	43.2	32.4	25.9	21.6	18.5	16.2	14.4	13.0	11.8	10.8	9.97	9.26	304	3.07
125 x 75 x 6.0 RHS	16.7	212	106	70.7	53.1	42.4	35.4	30.3	26.5	23.6	21.2	19.3	17.7	16.3	15.2	493	7.62
5.0 RHS	14.2	183	91.7	61.1	45.8	36.7	30.6	26.2	22.9	20.4	18.3	16.7	15.3	14.1	13.1	418	7.75
4.0 RHS	11.6	152	75.9	50.6	38.0	30.4	25.3	21.7	19.0	16.9	15.2	13.8	12.7	11.7	10.8	340	7.87
3.0 RHS	8.96	119	59.5	39.7	29.8	23.8	19.8	17.0	14.9	13.2	11.9	10.8	9.92	9.16	8.51	260	8.00
100 x 50 x 6.0 RHS	12.0	114	57.1	38.1	28.6	22.9	19.0	16.3	14.3	12.7	11.4	10.4	9.52	8.79	8.16	380	4.12
5.0 RHS	10.3	100	50.1	33.4	25.0	20.0	16.7	14.3	12.5	11.1	10.0	9.11	8.35	7.71	7.16	324	4.23
4.0 RHS	8.49	84.2	42.1	28.1	21.0	16.8	14.0	12.0	10.5	9.35	8.42	7.65	7.02	6.48	6.01	265	4.33
3.5 RHS	7.53	75.4	37.7	25.1	18.9	15.1	12.6	10.8	9.43	8.38	7.54	6.86	6.29	5.80	5.39	234	4.38
3.0 RHS	6.60	67.2	33.6	22.4	16.8	13.4	11.2	9.60	8.40	7.46	6.72	6.11	5.60	5.17	4.80	203	4.44
2.5 RHS	5.56	57.1	28.6	19.0	14.3	11.4	9.52	8.16	7.14	6.35	5.71	5.19	4.76	4.39	4.08	171	4.49
2.0 RHS	4.50	46.6	23.3	15.5	11.7	9.33	7.77	6.66	5.83	5.18	4.66	4.24	3.89	3.59	3.38	138	4.53

Notes:  
 1. FLR = segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -0.8$

2.  $\phi = 0.9$

3.  $\alpha_m = 1.0$

4.  $\alpha_s = 1.0$

5.  $W_{L1}^* = 8\phi M_s / L$

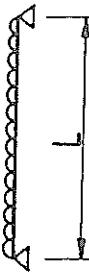
6.  $W_{L2}^* = 2\phi V_v$

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**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS**  
**FOR SIMPLY SUPPORTED BEAMS WITH FULL LATERAL RESTRAINT**  
**RECTANGULAR HOLLOW SECTIONS: GRADE C350**  
 bending about x-axis

$W_{L1}^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_{L2}^*$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L$  is LESSER of  $W_{L1}^*$  and  $W_{L2}^*$

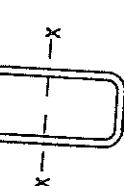


Designation	Mass per m	Span of Beam (L) in metres										$W_{L2}^*$	FLR					
		0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5	3.0	3.5	4.0	4.5	5.0	6.0			
100 x 50 x 6.0 RHS	12.0	229	152	114	91.4	76.2	65.3	57.1	45.7	38.1	32.6	28.6	25.4	22.9	19.0	360	412	
5.0 RHS	10.3	200	134	100	80.2	66.8	57.3	50.1	40.1	33.4	28.6	25.0	22.3	20.0	16.7	324	423	
4.0 RHS	8.49	168	112	84.2	67.4	56.1	48.1	42.1	33.7	28.1	24.1	21.0	18.7	16.8	14.0	255	433	
3.5 RHS	7.53	151	101	75.4	60.4	50.3	43.1	37.7	30.2	25.1	21.6	18.9	16.8	15.1	12.6	234	438	
3.0 RHS	6.60	134	89.6	67.2	53.7	44.8	38.4	33.6	26.9	22.4	19.2	16.8	14.9	13.4	11.2	203	444	
2.5 RHS	5.56	114	76.2	57.1	45.7	38.1	32.6	28.6	22.9	19.0	16.3	14.3	12.7	11.4	9.52	171	449	
2.0 RHS	4.50	93.3	62.2	46.6	37.3	31.1	26.6	23.3	18.7	15.5	13.3	11.7	10.4	9.33	7.77	198	453	
75 x 50 x 6.0 RHS	9.87	141	94.3	70.7	56.6	47.1	40.4	35.4	31.4	28.3	23.6	20.2	17.7	15.7	14.1	11.8	277	528
5.0 RHS	8.35	125	83.6	62.7	50.2	41.8	35.8	30.4	26.6	21.3	20.9	17.9	15.7	13.9	12.5	10.5	238	543
4.0 RHS	6.92	107	71.0	53.3	42.6	34.4	28.7	24.6	21.5	17.8	15.2	13.3	11.8	10.7	8.88	196	557	
3.0 RHS	5.42	86.1	57.4	43.1	34.4	28.7	24.5	21.0	18.4	14.7	12.3	10.8	9.57	8.61	7.18	151	573	
2.5 RHS	4.58	73.5	49.0	36.8	29.4	24.5	21.0	17.2	15.1	12.1	10.5	9.19	8.17	7.35	6.13	128	579	
2.0 RHS	3.72	60.3	40.2	30.1	24.1	20.1	17.2	15.1	12.1	10.0	8.61	7.53	6.70	6.03	5.02	104	586	
75 x 25 x 2.5 RHS	3.60	50.7	33.8	25.4	20.3	16.9	14.5	12.7	10.1	8.45	7.24	6.34	5.63	5.07	4.23	123	147	
2.0 RHS	2.93	41.9	27.9	20.9	16.8	14.0	12.0	10.5	8.38	6.98	5.98	5.23	4.65	4.19	3.48	99.9	150	
1.6 RHS	2.38	34.3	22.9	17.2	13.7	11.4	9.81	8.59	6.87	5.72	4.91	4.29	3.82	3.43	2.86	80.8	153	
65 x 35 x 3.0 RHS	4.25	55.4	36.9	27.7	22.2	18.5	15.8	13.8	11.1	9.23	7.91	6.92	6.16	5.54	4.62	128	223	
2.5 RHS	3.60	47.6	31.8	23.8	19.1	15.9	13.6	11.9	9.53	7.94	6.81	5.96	5.29	4.76	3.97	108	328	
2.0 RHS	2.93	39.3	26.2	19.7	15.7	13.1	11.2	9.83	7.86	6.55	5.62	4.91	4.37	3.93	3.28	88.2	333	
50 x 25 x 3.0 RHS	3.07	29.5	19.7	14.8	11.8	9.85	8.44	7.39	6.91	4.92	4.22	3.69	3.28	2.95	2.46	95.0	208	
2.5 RHS	2.62	25.7	17.2	12.9	10.3	8.58	7.35	6.43	5.15	4.29	3.68	3.22	2.86	2.57	2.14	81.0	212	
2.0 RHS	2.15	21.5	14.3	10.7	8.69	7.16	6.14	5.37	4.30	3.58	3.07	2.69	2.39	2.15	1.79	66.2	217	
1.6 RHS	1.75	17.8	11.8	8.88	7.11	5.92	5.08	4.44	3.65	2.96	2.54	2.22	1.97	1.78	1.48	53.9	221	
50 x 20 x 3.0 RHS	2.83	26.0	17.3	13.0	10.4	8.66	7.43	6.50	5.20	4.33	3.71	3.25	2.89	2.60	2.17	93.8	131	
2.5 RHS	2.42	22.7	16.2	11.4	9.10	7.58	6.50	5.68	4.55	3.79	3.25	2.84	2.53	2.27	1.89	80.0	135	
2.0 RHS	1.99	19.1	12.7	9.53	7.63	6.35	5.45	4.77	3.81	3.18	2.72	2.38	2.12	1.91	1.59	65.4	139	
1.6 RHS	1.63	15.8	10.5	7.91	6.39	5.27	4.52	3.95	3.16	2.64	2.26	1.98	1.76	1.58	1.32	53.2	142	

Notes:  
 1. FLR = segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -0.8$   
 2.  $\phi = 0.9$   
 3.  $\alpha_m = 1.0$   
 4.  $\alpha_s = 1.0$   
 5.  $W_{L1}^* = 8\phi M_s / L$   
 6.  $W_{L2}^* = 2\phi V_u$

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## FOR SIMPLY SUPPORTED BEAMS WITH FULL LATERAL RESTRAINT

### RECTANGULAR HOLLOW SECTIONS: GRADE C450 bending about x-axis

TABLE 8.1 – 4 (1)(A)

Designation	Mass per m	W <sub>1</sub> (kN)						W <sub>2</sub>	FLR								
		d mm	b mm	t mm	1.0	2.0	3.0	4.0									
125 x 75 x 3.8 RHS	11.1	187	93.4	62.3	46.7	37.4	31.1	26.7	23.3	20.8	18.7	17.0	11.0	12.0	13.0	14.0	14.0
3.3 RHS	9.73	165	82.5	55.0	41.3	33.0	27.5	23.6	20.6	18.3	16.5	15.0	13.8	12.7	11.3	10.4	14.4
2.8 RHS	8.39	136	67.8	45.2	33.9	27.1	22.6	19.4	16.3	15.1	13.6	12.3	11.3	10.4	9.68	10.4	13.3
2.3 RHS	6.95	98.0	49.0	32.7	24.5	19.6	16.3	14.0	12.3	10.9	9.80	8.91	8.17	7.54	7.00	7.00	11.3
100 x 50 x 3.3 RHS	7.14	92.3	46.2	30.8	23.1	18.5	15.4	13.2	11.5	10.3	9.23	8.39	7.69	7.10	6.59	417	6.14
2.8 RHS	6.19	81.3	40.6	27.1	20.3	16.3	13.5	11.6	10.2	9.03	8.13	7.39	6.77	6.25	5.81	365	6.19
2.3 RHS	5.14	68.1	34.1	22.7	17.0	13.6	11.4	9.73	8.52	7.57	6.81	6.19	5.68	5.24	4.87	313	6.24
Notes: Refer to TABLE 8.1–4 (2)(A)																259	6.29
																203	3.90

W<sub>1</sub> = Maximum Design Load based on Design Moment Capacity  
W<sub>2</sub> = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load W<sub>1</sub> is LESSER of W<sub>1</sub> and W<sub>2</sub>

Notes: Refer to TABLE 8.1–4 (2)(A)

Designation	Mass per m	W <sub>1</sub> (kN)						W <sub>2</sub>	FLR								
		d mm	b mm	t mm	0.5	0.75	1.0	1.25									
100 x 50 x 3.3 RHS	7.14	185	123	92.3	73.9	61.5	52.8	46.2	36.9	30.8	26.4	23.1	20.5	18.5	15.4	15.4	15.4
2.8 RHS	6.19	163	108	81.3	65.0	54.2	46.4	40.6	32.5	27.1	23.2	20.3	18.1	16.3	13.5	285	3.42
2.3 RHS	5.14	136	90.8	68.1	54.5	45.4	38.9	34.1	27.3	22.7	19.5	17.0	15.1	13.6	11.4	245	3.47
75 x 50 x 2.8 RHS	5.09	104	63.6	52.2	41.7	34.8	29.8	26.1	20.9	17.4	14.9	13.0	11.6	10.4	8.70	203	3.50
2.3 RHS	4.24	87.8	58.6	43.9	35.1	29.3	25.1	22.0	17.6	14.6	12.5	11.0	9.76	8.78	7.32	183	4.48
65 x 35 x 2.8 RHS	3.99	67.3	44.9	33.7	26.9	22.4	19.2	16.8	13.5	11.2	9.62	8.42	7.48	6.73	5.61	155	4.53
2.3 RHS	3.34	57.1	38.0	28.5	22.8	19.0	16.3	14.3	11.4	9.51	8.15	7.13	6.34	5.71	4.76	129	2.53
50 x 25 x 2.8 RHS	2.89	36.1	24.1	18.0	14.4	12.0	10.3	9.02	7.22	6.02	5.16	4.51	4.01	3.61	3.01	115	1.67
2.3 RHS	2.44	31.0	20.6	15.5	12.4	10.3	8.85	7.74	6.19	5.16	4.42	3.87	3.44	3.10	2.58	95.7	1.67
50 x 20 x 2.8 RHS	2.67	31.8	21.2	15.9	12.7	10.6	9.09	7.95	6.36	5.30	4.54	3.98	3.53	3.18	2.65	114	1.03
2.3 RHS	2.25	27.4	18.3	13.7	11.0	9.14	7.83	6.85	5.48	4.57	3.92	3.43	3.05	2.74	2.28	95.4	1.06
Notes:																	
1. FLR = Segment Length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with $\beta_m = -0.8$																	
2. $\phi = 0.9$																	
3. $\alpha_m = 1.0$																	
4. $\alpha_s = 1.0$																	
5. $W_1 = 8\phi M_s / L$																	
6. $W_2 = 8\phi M_s / L$																	

TABLE 8.1 – 4 (2)(A)

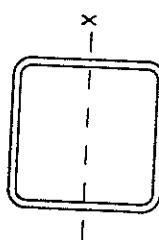
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TABLE 8.1-5 (1)(A)

**FOR SIMPLY SUPPORTED BEAMS** WITH FULL LATERAL RESTRAINT

**SQUARE HOLLOW SECTIONS: GRADE C350**  
bending about x-axis



W<sub>L1</sub> = Maximum Design Load based on Design Moment Capacity  
W<sub>L2</sub> = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load W<sub>f</sub> is LESSER of W<sub>L1</sub> and W<sub>L2</sub>

Designation	Mass per m	W <sub>L1</sub> (kN)						W <sub>L2</sub>	FLR							
		Span of Beam (L) in metres														
d mm	b mm	t mm	1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	9.0	10.0	11.0	12.0	13.0	14.0
250 x 250 x 9.0 SHS	65.9	1880	938	625	469	375	313	268	234	208	188	171	156	144	134	120
6.0 SHS	45.0	1030	515	343	257	206	172	147	129	114	103	93.6	85.8	79.2	73.5	60
200 x 200 x 9.0 SHS	51.8	1170	586	391	283	235	195	168	147	130	117	107	97.7	90.2	83.8	79.2
6.0 SHS	35.6	740	370	247	185	148	123	106	92.5	82.2	74.0	67.3	61.7	56.9	52.9	49.0
5.0 SHS	29.9	561	280	187	140	112	93.4	80.1	70.1	62.3	56.1	51.0	46.7	43.1	40.0	37.7
150 x 150 x 9.0 SHS	37.7	625	313	208	156	125	104	89.4	78.2	69.5	62.5	56.9	52.1	48.1	44.7	41.8
6.0 SHS	26.2	447	224	149	112	89.5	74.6	63.9	55.9	49.7	44.7	40.7	37.3	34.4	32.0	42.4
5.0 SHS	22.1	384	182	121	91.0	72.8	60.7	52.0	45.5	40.5	36.4	33.1	30.3	28.0	26.0	20.0
125 x 125 x 9.0 SHS	30.6	416	208	139	104	83.1	69.3	59.4	52.0	46.2	41.6	37.8	34.6	32.0	29.7	23.9
6.0 SHS	21.4	302	151	101	75.6	60.4	50.4	43.2	37.8	33.6	30.2	27.5	25.2	23.2	21.6	24.3
5.0 SHS	18.2	259	129	86.3	64.7	51.8	43.2	37.0	32.4	28.8	25.9	23.5	21.6	19.9	18.5	24.9
4.0 SHS	14.8	199	99.4	66.2	49.7	39.7	33.1	28.4	24.8	22.1	19.9	18.1	16.6	15.3	14.2	25.2
100 x 100 x 9.0 SHS	23.5	248	124	82.8	62.1	49.7	41.4	35.5	31.0	27.6	24.8	22.6	20.7	19.1	17.7	19.9
6.0 SHS	16.7	185	92.7	61.8	46.3	37.1	30.9	26.5	23.2	20.6	18.5	16.8	15.4	14.3	13.2	20.6
5.0 SHS	14.2	160	80.0	53.3	40.0	32.0	26.7	22.9	20.0	17.8	16.0	14.5	13.3	12.3	11.4	20.8
4.0 SHS	11.6	133	66.3	44.2	33.1	26.5	22.1	18.9	16.6	14.7	13.3	12.1	11.0	10.2	9.47	21.0
3.0 SHS	8.96	93.5	46.8	31.2	23.4	18.7	15.6	13.4	11.7	10.4	9.35	8.50	7.79	7.19	6.68	34.9
89 x 89 x 6.0 SHS	14.6	143	71.3	47.5	35.6	28.5	23.8	20.4	17.8	15.8	14.3	13.0	11.9	11.0	10.2	15.5
5.0 SHS	12.5	124	61.9	41.2	30.9	24.7	20.6	17.7	15.5	13.7	12.4	11.2	10.3	9.52	8.84	9.6
3.5 SHS	9.06	91.9	45.9	30.6	23.0	18.4	15.3	13.1	11.5	10.2	9.19	8.35	7.66	7.07	6.56	21.4
75 x 75 x 6.0 SHS	12.0	96.8	48.4	32.3	24.2	19.4	16.1	13.8	12.1	10.8	9.68	8.80	8.07	7.45	6.91	14.2
5.0 SHS	10.3	84.8	42.4	28.3	21.2	17.0	14.1	12.1	10.6	9.42	8.48	7.71	7.07	6.52	6.05	14.5
4.0 SHS	8.49	71.2	35.6	23.7	17.8	14.2	11.9	10.2	8.90	7.91	7.12	6.47	5.93	5.48	5.08	12.0
3.5 SHS	7.53	63.8	21.3	15.9	12.8	10.6	9.11	7.97	7.08	6.38	5.80	5.31	4.90	4.55	4.05	12.2
3.0 SHS	6.60	56.7	28.3	18.9	14.2	11.3	9.44	8.09	7.08	6.30	5.67	5.15	4.72	4.36	4.05	12.4
2.5 SHS	5.56	46.0	23.0	15.3	11.5	9.20	7.67	6.57	5.75	5.11	4.60	4.18	3.83	3.54	3.29	12.5

Notes:  
1. FLR = segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -0.8$   
2.  $\phi = 0.9$   
3.  $\alpha_m = 1.0$   
4.  $\alpha_s = 1.0$   
5.  $W_{L1}^* = 8\phi M_s/L$   
6.  $W_{L2} = 2\phi V_y$

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TABLE 8.1 – 5 (2)(A)

**FOR SIMPLY SUPPORTED BEAMS**

**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
WITH FULL LATERAL RESTRAINT  
SQUARE HOLLOW SECTIONS: GRADE C350  
bending about x-axis**

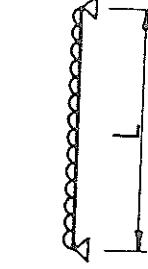
W<sub>1</sub> = Maximum Design Load based on Design Moment Capacity  
W<sub>2</sub> = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load W<sub>1</sub> is LESSER of W<sub>1</sub> and W<sub>2</sub>

Designation	Mass per m	d mm	b mm	t mm	W <sub>1</sub> (kN)								W <sub>2</sub> (kN)	FLR	
					0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5	3.0	3.5	
<b>Span of Beam (L) in metres</b>															
75 x 75 x 6.0 SHS	12.0	194	129	96.8	77.4	64.5	55.3	48.4	42.4	38.7	32.3	27.7	24.2	21.5	19.4
5.0 SHS	10.3	170	113	84.8	67.8	56.5	49.0	42.4	37.5	33.9	28.3	24.2	21.2	18.8	16.1
4.0 SHS	8.49	142	94.9	71.2	56.9	47.5	40.7	35.6	30.6	26.0	20.8	17.3	13.6	10.2	7.77
3.5 SHS	7.53	128	85.0	63.8	51.0	42.5	36.4	31.9	27.8	23.9	20.3	17.8	13.0	10.4	6.18
3.0 SHS	6.60	113	75.6	56.7	45.3	37.8	32.4	28.3	22.7	21.3	18.2	15.8	12.3	10.2	4.23
2.5 SHS	5.56	92.0	61.4	46.0	36.8	30.7	26.3	23.0	18.4	15.3	13.1	11.5	10.6	12.8	12.0
65 x 65 x 6.0 SHS	10.1	139	92.4	69.3	55.5	46.2	39.6	34.7	27.7	23.1	19.8	17.3	13.9	11.6	10.6
5.0 SHS	8.75	123	81.7	61.3	49.0	40.9	34.6	30.6	24.5	20.4	17.5	15.3	12.3	10.2	7.77
4.0 SHS	7.23	104	69.2	51.9	41.5	34.6	29.7	26.0	20.8	17.3	14.8	13.0	11.5	10.4	6.18
3.0 SHS	5.66	83.5	55.7	41.8	33.4	27.8	23.9	20.9	16.7	13.9	11.9	10.4	9.28	8.35	7.20
2.5 SHS	4.78	71.3	47.5	35.6	28.5	23.8	20.4	17.8	14.3	11.9	10.2	8.91	7.92	7.13	6.35
2.0 SHS	3.88	59.3	35.6	26.7	21.3	17.8	15.2	13.3	10.7	8.89	7.62	6.67	5.93	5.94	5.33
50 x 50 x 5.0 SHS	6.39	66.3	44.2	33.2	26.5	22.1	18.9	16.6	13.3	11.1	9.47	8.29	7.37	6.63	5.53
4.0 SHS	5.35	57.4	38.2	28.7	22.9	19.1	16.4	14.3	11.5	9.56	8.19	7.17	6.37	5.74	4.49
3.0 SHS	4.25	47.3	31.5	23.7	18.9	15.8	13.5	11.8	9.46	7.89	6.76	5.91	5.26	4.73	3.27
2.5 SHS	3.60	40.7	27.1	20.3	16.3	13.6	11.6	10.2	8.14	6.78	5.81	5.09	4.52	4.07	2.05
2.0 SHS	2.93	33.6	22.4	16.8	13.4	11.2	9.59	8.39	7.35	6.72	5.60	4.80	4.20	3.39	1.03
1.6 SHS	2.38	25.7	17.1	12.9	10.3	8.57	7.35	6.43	5.14	4.29	3.67	3.21	2.31	2.08	8.40
40 x 40 x 4.0 SHS	4.09	33.9	22.6	17.0	13.6	11.3	9.70	8.49	6.79	5.66	4.85	4.24	3.77	3.39	2.80
3.0 SHS	3.30	28.8	19.2	14.4	11.5	9.62	8.24	7.21	5.77	4.81	4.12	3.61	3.21	2.88	1.49
2.5 SHS	2.82	25.0	16.7	12.5	10.3	8.34	7.15	6.26	5.01	4.17	3.58	3.13	2.78	2.40	7.61
2.0 SHS	2.31	20.8	13.9	10.4	8.33	6.94	5.95	5.21	4.17	3.47	2.98	2.60	2.50	2.09	7.85
1.6 SHS	1.88	17.2	11.5	8.60	6.88	5.73	4.91	4.30	3.44	2.87	2.46	2.15	1.91	1.74	9.86
35 x 35 x 3.0 SHS	2.83	21.3	14.2	10.7	8.33	7.10	6.09	5.33	4.26	3.55	3.04	2.66	2.37	2.14	5.59
2.5 SHS	2.42	18.6	12.4	9.31	7.45	6.21	5.32	4.66	3.72	3.10	2.66	2.33	2.07	1.86	5.50
2.0 SHS	1.99	15.6	10.4	7.80	6.24	5.20	4.46	3.90	3.12	2.60	2.23	1.95	1.56	1.30	4.63
1.6 SHS	1.63	12.9	8.62	6.47	5.17	4.31	3.70	3.23	2.59	2.16	1.85	1.55	1.33	1.08	6.48
30 x 30 x 3.0 SHS	2.36	14.9	9.94	7.46	5.97	4.97	4.26	3.73	2.98	2.49	2.13	1.86	1.66	1.49	5.58
2.5 SHS	2.03	13.2	8.77	6.58	5.26	4.38	3.76	3.29	2.63	2.19	1.88	1.64	1.46	1.24	5.71
2.0 SHS	1.68	11.1	7.41	5.56	4.45	3.70	3.18	2.78	2.22	1.85	1.59	1.39	1.23	1.08	6.67
1.6 SHS	1.38	9.28	6.19	4.64	3.71	3.09	2.65	2.32	1.86	1.55	1.33	1.16	1.03	0.928	5.80
25 x 25 x 3.0 SHS	1.89	9.65	6.43	4.83	3.86	3.22	2.76	2.41	1.93	1.61	1.38	1.21	1.07	0.965	5.50
2.5 SHS	1.64	8.63	5.75	4.32	3.45	2.88	2.47	2.16	1.73	1.44	1.23	1.08	0.959	0.804	4.63
2.0 SHS	1.36	7.39	4.93	3.69	2.96	2.46	2.11	1.85	1.48	1.23	1.06	0.924	0.821	0.759	4.73
1.6 SHS	1.12	6.23	4.15	3.11	2.49	2.08	1.78	1.56	1.25	1.04	0.890	0.779	0.632	0.623	3.75
20 x 20 x 1.6 SHS	0.873	3.78	2.52	1.89	1.51	1.26	1.08	0.946	0.757	0.631	0.540	0.473	0.420	0.378	0.315
															20.1
															3.17

Note: Refer to TABLE 8.1-5 (1)(A)

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**FOR SIMPLY SUPPORTED BEAMS WITH FULL LATERAL RESTRAINT**  
**SQUARE HOLLOW SECTIONS: GRADE C450**  
 bending about x-axis

TABLE 8.1 – 6 (1)(A)

$W_{L1}^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_{L2}$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L$  is LESSER of  $W_{L1}^*$  and  $W_{L2}$

Designation	Mass per m	Span of Beam (L) in metres								$W_{L1}^*$ (kN)	$W_{L2}$	FLR
		1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0			
d mm	b mm	t mm	kg/m									
100 x 100 x 3.8 SHS	11.1	157	78.3	52.2	39.2	31.3	26.1	22.4	19.6	17.4	15.7	14.2
3.3 SHS	9.73	127	63.7	42.4	31.8	25.5	21.2	18.2	15.9	14.1	12.7	13.1
2.8 SHS	8.39	100	50.1	33.4	25.1	20.1	16.7	14.3	12.5	11.1	10.6	9.80
2.3 SHS	6.95	74.8	37.4	24.9	18.7	15.0	12.5	10.7	9.34	8.31	7.48	8.36
75 x 75 x 3.3 SHS	7.14	78.0	39.0	26.0	19.5	15.6	13.0	11.1	9.75	8.67	7.80	7.09
2.8 SHS	6.19	65.3	32.6	21.8	16.3	13.1	10.9	9.32	8.16	7.25	6.53	5.93
2.3 SHS	5.14	52.5	26.3	17.5	13.1	10.5	8.75	7.50	6.56	5.83	5.25	4.77

Notes: Refer to TABLE 8.1-5 (1)(A)

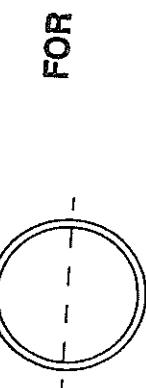
TABLE 8.1 – 6 (2)(A)

Designation	Mass per m	Span of Beam (L) in metres								$W_{L1}^*$ (kN)	$W_{L2}$	FLR
		0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5			
d mm	b mm	t mm	kg/m									
75 x 75 x 3.3 SHS	7.14	156	104	78.0	62.4	52.0	44.6	39.0	31.2	26.0	22.3	19.5
2.8 SHS	6.19	131	87.0	65.3	52.2	43.5	37.3	32.6	26.1	21.8	18.6	16.3
2.3 SHS	5.14	97.1	64.7	48.6	38.8	32.4	27.7	24.3	19.4	16.2	13.9	12.1
65 x 65 x 2.8 SHS	5.31	101	67.5	50.6	40.5	33.7	28.9	25.3	20.2	16.9	14.5	12.6
2.3 SHS	4.42	78.6	52.4	39.3	31.4	26.2	22.5	19.6	15.7	13.1	11.2	9.82
50 x 50 x 2.8 SHS	3.99	57.5	38.3	28.8	23.0	19.2	16.4	14.4	11.5	9.58	8.21	7.19
2.3 SHS	3.34	48.7	32.5	24.4	19.5	16.2	13.9	12.2	9.75	8.12	6.96	6.09
40 x 40 x 2.8 SHS	3.11	35.2	23.5	17.6	14.1	11.7	10.1	8.80	7.04	5.86	5.03	4.40
2.3 SHS	2.62	30.1	20.1	15.0	12.0	10.0	8.80	7.52	6.02	5.01	4.30	3.76
35 x 35 x 2.8 SHS	2.67	28.1	17.4	13.0	10.4	8.69	7.45	6.52	5.21	4.34	3.72	3.26
2.3 SHS	2.25	22.4	15.0	11.2	8.98	7.48	6.41	5.61	4.49	3.74	3.21	2.80
30 x 30 x 2.8 SHS	2.23	18.3	12.2	9.16	7.33	6.10	5.23	4.58	3.66	3.05	2.62	2.29
2.3 SHS	1.89	15.9	10.6	7.95	6.36	5.30	4.54	3.98	3.18	2.65	2.27	1.99
25 x 25 x 2.8 SHS	1.53	10.5	7.00	5.25	4.20	3.50	3.00	2.62	2.10	1.75	1.50	1.31

Notes: Refer to TABLE 8.1-5 (1)(A)

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**FOR CONTINUOUS BEAMS WITH FULL LATERAL RESTRAINT  
CIRCULAR HOLLOW SECTIONS: GRADE C250  
bending about any axis**

$W_{L1}^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_{L2}^*$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L^*$  is LESSER of  $W_{L1}^*$  and  $W_{L2}^*$

TABLE 8.2 - 1 (1)(A)

Designation	Mass per m	Span of Beam (L) in metres	$W_L^*$ (kN)						
			1.0	2.0	3.0	4.0	5.0	6.0	7.0
610.0 x 12.7 CHS	187	8160	4080	2720	2040	1630	1360	1170	1020
9.5 CHS 141	5880	2940	1960	1470	1180	980	840	735	654
6.4 CHS 95.3	3590	1850	1200	898	718	599	513	449	399
508.0 x 12.7 CHS	155	5610	2800	1870	1400	1120	935	801	701
9.5 CHS 117	4200	2100	1400	1050	840	700	600	525	467
6.4 CHS 79.2	2630	1310	875	656	525	438	375	328	292

Notes: Refer to TABLE 8.2-1 (3)(A)

TABLE 8.2 - 1 (2)(A)

Designation	Mass per m	Span of Beam (L) in metres	$W_L^*$ (kN)						
			0.5	1.0	1.5	2.0	2.5	3.0	3.5
165.1 x 5.4 CHS	21.3	496	248	165	124	99.2	82.7	70.9	62.0
5.0 CHS 19.7	462	231	154	115	92.3	76.9	65.9	57.7	46.2
139.7 x 5.4 CHS	351	175	117	87.7	70.2	58.5	50.1	43.9	35.1
5.0 CHS 16.6	327	163	109	81.7	65.3	54.5	46.7	40.8	32.7
114.3 x 5.4 CHS	231	115	76.9	57.7	46.1	38.5	33.0	28.8	23.3
4.5 CHS 12.2	195	97.7	65.1	48.9	39.1	32.6	27.9	24.4	20.4
101.6 x 5.0 CHS	168	84.1	56.0	42.0	33.6	28.0	24.0	21.0	19.2
4.0 CHS 9.63	137	68.6	45.7	34.3	27.4	22.9	19.6	17.2	13.7
88.9 x 5.9 CHS	12.1	147	73.3	48.9	36.6	29.3	24.4	20.9	18.3
5.0 CHS 10.3	127	63.4	42.3	31.7	25.4	21.1	18.1	15.9	14.7
4.0 CHS 8.38	104	51.9	34.6	26.0	20.8	17.3	14.8	13.0	12.7
76.1 x 5.9 CHS	10.2	105	52.5	35.0	26.2	21.0	17.5	15.0	13.1
4.5 CHS 7.95	83.2	41.6	27.7	20.8	16.6	13.9	11.9	10.4	9.74
3.6 CHS 6.44	68.2	34.1	22.7	17.0	13.6	11.4	9.74	8.52	8.82

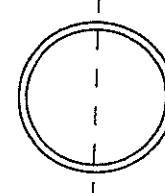
Notes: Refer to TABLE 8.2-1 (3)(A)

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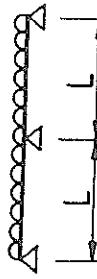
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TABLE 8.2 - 1 (3)(A)



**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR CONTINUOUS BEAMS WITH FULL LATERAL RESTRAINT  
CIRCULAR HOLLOW SECTIONS; GRADE C250**

W<sub>L1</sub>\* = Maximum Design Load based on Design Moment Capacity  
W<sub>L2</sub>\* = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load W<sub>L</sub> is LESSER of W<sub>L1</sub> and W<sub>L2</sub>\*



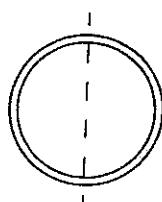
Designation d <sub>b</sub> t	Mass per m	W <sub>L1</sub> (kN)						W <sub>L2</sub>
		Span of Beam (L) in metres						
mm mm	kg/m	0.25	0.5	0.75	1.0	1.25	1.5	
76.1 x 5.9 CHS	10.2	210	105	69.9	52.5	42.0	35.0	30.0
4.5 CHS	7.95	166	83.2	55.4	41.6	33.3	27.7	26.2
3.6 CHS	6.44	136	68.2	45.5	34.1	27.3	22.7	20.8
60.3 x 5.4 CHS	118	58.8	39.2	28.4	23.5	19.6	16.8	14.7
4.5 CHS	6.19	101	50.6	33.7	25.3	20.2	16.9	14.4
3.6 CHS	5.03	83.4	41.7	27.8	20.9	16.7	13.9	11.9
48.3 x 5.4 CHS	5.71	71.9	36.0	24.0	18.0	14.4	12.0	10.3
4.0 CHS	4.37	56.7	28.3	18.9	14.2	11.3	9.45	8.10
3.2 CHS	3.56	46.9	23.5	15.6	11.7	9.39	7.82	6.71
42.4 x 4.9 CHS	4.53	49.9	24.9	16.6	12.5	9.98	8.32	7.13
4.0 CHS	3.79	42.6	21.3	14.2	10.7	8.52	7.10	6.09
3.2 CHS	3.09	35.5	17.7	11.8	8.87	7.10	5.91	5.07
35.7 x 4.5 CHS	3.24	27.8	13.9	9.28	6.96	5.57	4.64	3.98
4.0 CHS	2.93	25.6	12.8	8.52	6.39	5.11	4.26	3.65
3.2 CHS	2.41	21.5	10.8	7.17	5.38	4.30	3.59	3.07
26.9 x 4.0 CHS	2.26	15.3	7.63	5.09	3.81	3.06	2.54	2.18
3.2 CHS	1.87	13.0	6.51	4.34	3.25	2.60	2.17	1.86
2.6 CHS	1.56	11.1	5.55	3.70	2.77	2.22	1.85	1.59
21.3 x 3.6 CHS	1.57	8.23	4.12	2.74	2.06	1.65	1.37	1.18
3.2 CHS	1.43	7.63	3.81	2.54	1.91	1.53	1.27	1.03

- Notes:  
 1.  $\phi = 0.9$   
 2.  $\alpha_m = 1.0$   
 3.  $\alpha_s = 1.0$   
 4.  $W_{L1}^* = 8\phi M_s/L$   
 5.  $W_{L2}^* = 1.6\phi V_v$

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TABLE 8.2 - 2 (1)(A)



**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR CONTINUOUS BEAMS WITH FULL LATERAL RESTRAINT  
CIRCULAR HOLLOW SECTIONS: GRADE C350  
bending about any axis**

$W_1^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_2^*$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_1^*$  is LESSER of  $W_1$  and  $W_2^*$

Designation	Mass per m	$d_o$	$t$	Span of Beam (L) in metres						$W_2^*$						
				1.0	2.0	3.0	4.0	5.0	6.0							
457.0 x 12.7 CHS	139	6310	3160	2100	1580	1260	1050	902	789	701	631	526	451	394	3220	16.0
9.5 CHS	105	4520	2260	1510	1130	904	763	646	565	502	452	377	323	283	2420	
6.4 CHS	71.1	2750	1370	916	687	550	458	393	343	305	275	229	196	172	1640	
406.4 x 12.7 CHS	123	4960	2480	1650	1240	992	827	709	620	551	496	414	354	310	2850	
9.5 CHS	93.0	3850	1820	1220	912	730	608	521	456	405	365	304	261	228	2150	
6.4 CHS	63.1	2260	1130	752	584	451	376	322	282	251	226	188	161	141	1460	
355.6 x 12.7 CHS	107	3760	1880	1250	941	753	627	538	471	418	376	314	269	235	2450	
9.5 CHS	81.1	2850	1420	948	711	569	474	406	356	316	285	237	203	178	1870	
6.4 CHS	55.1	1790	894	596	447	358	288	255	224	199	179	149	128	112	1270	
323.9 x 12.7 CHS	97.5	3100	1560	1030	775	620	517	443	388	345	310	258	222	194	2250	
9.5 CHS	73.7	2370	1180	789	592	473	395	338	296	263	237	197	169	148	1700	
6.4 CHS	50.1	1510	757	505	379	303	252	216	189	168	151	126	108	94.7	1160	
273.1 x 9.3 CHS	60.5	1630	816	544	408	326	272	233	204	181	163	136	117	102	1400	
6.4 CHS	42.1	1110	565	370	278	222	185	159	139	123	111	92.5	79.3	69.4	973	
4.8 CHS	31.8	787	393	262	197	157	131	112	98.3	87.4	78.7	65.6	56.2	49.2	734	
219.1 x 8.2 CHS	42.6	920	460	307	230	184	153	131	115	102	92.0	76.6	65.7	57.5	986	
6.4 CHS	33.6	730	365	243	182	146	122	104	91.2	81.1	73.0	60.8	52.1	45.6	776	
4.8 CHS	25.4	530	285	177	133	106	83.3	75.7	66.3	56.9	53.0	44.2	37.9	33.1	586	
168.3 x 7.1 CHS	28.2	465	233	155	116	93.0	77.5	66.5	58.2	51.7	46.5	38.8	33.2	29.1	652	
6.4 CHS	25.6	423	211	141	106	84.6	70.5	60.4	52.9	47.0	42.3	35.2	30.2	26.4	591	
4.8 CHS	19.4	323	162	108	80.9	64.7	53.9	46.2	40.4	35.9	32.3	27.0	23.1	20.2	447	

- Notes:  
 1.  $\phi = 0.9$   
 2.  $\alpha_m = 1.0$   
 3.  $\alpha_s = 1.0$   
 4.  $W_1^* = 8\phi M_s/L$   
 5.  $W_2^* = 1.6\phi V_v$

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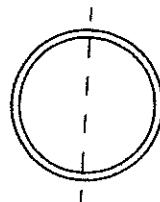
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**TABLE 8.2 – 2 (2)(A)**

**FOR CONTINUOUS BEAMS WITH FULL LATERAL RESTRAINT**

**CIRCULAR HOLLOW SECTIONS: GRADE C350**

bending about any axis



$W_L^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_2$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L$  is LESSER of  $W_L^*$  and  $W_2$

Designation $d_o$ mm	$t$ mm	Mass per m kg/m	Span of Beam (L) in metres						$W_L^*$ (kN)	
			0.5	1.0	1.5	2.0	2.5	3.0		
168.3 x 7.1 CHS	28.2	930	465	310	233	186	155	133	116	92.0
6.4 CHS	25.6	846	423	282	211	169	141	121	106	84.6
4.8 CHS	19.4	647	323	216	162	129	108	92.4	80.9	77.5
165.1 x 3.5 CHS	13.9	436	218	145	109	87.8	72.7	62.3	54.5	93.0
3.0 CHS	12.0	362	181	121	90.6	72.4	60.4	51.7	45.3	70.5
139.7 x 3.5 CHS	11.8	321	160	107	80.2	64.2	53.5	45.8	36.4	66.5
3.0 CHS	10.1	268	134	89.5	67.1	53.7	44.7	38.3	32.1	60.4
114.3 x 6.0 CHS	16.0	355	178	118	88.8	71.0	59.2	50.7	44.4	84.7
4.8 CHS	13.0	290	145	96.8	72.6	58.1	48.4	41.5	35.5	75.9
3.6 CHS	9.83	222	111	74.1	55.6	44.5	37.1	31.8	29.0	65.2
3.2 CHS	8.77	199	99.6	66.4	49.8	39.8	33.2	28.4	27.8	52.9
101.6 x 3.2 CHS	7.77	156	78.1	52.1	39.1	31.2	26.0	22.3	19.5	46.2
2.6 CHS	6.35	126	63.2	42.1	31.6	25.3	21.1	18.1	15.8	40.4
88.9 x 5.5 CHS	11.3	193	96.5	64.4	48.3	38.6	32.2	27.6	24.1	80.4
4.8 CHS	9.96	171	85.6	57.1	42.8	34.3	28.5	24.5	21.4	73.9
3.2 CHS	6.76	119	59.3	39.5	29.6	23.7	19.8	16.9	14.8	60.4
2.6 CHS	5.53	97.6	48.8	32.5	24.4	19.5	16.3	13.9	12.2	52.9
76.1 x 3.2 CHS	5.75	85.8	42.9	28.6	21.4	17.2	14.3	12.3	9.76	45.3
2.3 CHS	4.19	63.2	31.6	21.1	15.8	12.6	10.5	9.02	7.89	39.5

- Notes:
1.  $\phi = 0.9$
  2.  $c_m = 1.0$
  3.  $c_s = 1.0$
  4.  $W_{L1}^* = 8 \phi M_s / L$
  5.  $W_2^* = 16 \phi V_v$

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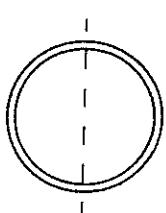
**TABLE 8.2 – 2 (3)(A)**

**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS**

**FOR CONTINUOUS BEAMS WITH FULL LATERAL RESTRAINT**

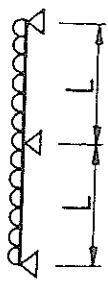
**CIRCULAR HOLLOW SECTIONS: GRADE C350**

bending about any axis



$W_{L1}^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_{L2}^*$  = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load  $W_L$  is LESSER of  $W_{L1}^*$  and  $W_{L2}^*$

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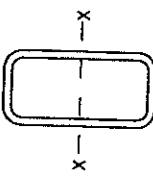
Designation $d_o$ mm	Mass per m kg/m	Span of Beam (L) in metres						$W_{L1}$ (kN)	$W_{L2}$
		0.25	0.5	0.75	1.0	1.25	1.5		
76.1x3.2 CHS	5.75	172	65.8	57.2	42.9	34.3	28.6	24.5	21.4
2.3 CHS	4.19	126	63.2	42.1	31.6	25.3	21.1	18.0	15.8
60.3x2.9 CHS	4.11	96.4	48.2	32.1	24.1	19.3	16.1	13.8	12.0
2.3 CHS	3.29	78.0	39.0	26.0	19.5	15.6	13.0	11.1	9.75
48.3x2.9 CHS	3.25	60.3	30.2	20.1	15.1	12.1	10.1	8.62	7.54
2.3 CHS	2.61	49.1	24.5	16.4	12.3	9.82	8.18	7.01	6.14
42.4x2.6 CHS	2.55	41.6	20.8	13.9	10.4	8.31	6.93	5.94	5.20
2.0 CHS	1.99	32.9	16.5	11.0	8.23	6.59	5.49	4.70	4.12
33.7x2.6 CHS	1.99	25.4	12.7	8.47	6.35	5.08	4.23	3.63	3.18
2.0 CHS	1.56	20.3	10.1	6.76	5.07	4.06	3.38	2.90	2.54
26.9x2.3 CHS	1.40	14.1	7.04	4.69	3.52	2.81	2.35	2.01	1.76
2.0 CHS	1.23	12.5	6.26	4.16	3.19	2.51	2.09	1.79	1.57
21.3x2.0 CHS	0.952	7.54	3.77	2.51	1.88	1.51	1.26	1.08	0.942

- Notes:
1.  $\phi = 0.9$
  2.  $c_m = 1.0$
  3.  $\alpha_s = 1.0$
  4.  $W_{L1}^* = 8 \phi M_s / L$
  5.  $W_{L2}^* = 1.6 \phi V_v$

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TABLE 8.2 – 3 (1)(A)



**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR CONTINUOUS BEAMS WITH FULL LATERAL RESTRAINT  
RECTANGULAR HOLLOW SECTIONS: GRADE C350  
bending about x-axis**

W<sub>1</sub><sup>\*</sup> = Maximum Design Load based on Design Moment Capacity  
W<sub>2</sub><sup>\*</sup> = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load W<sub>1</sub><sup>\*</sup> is LESSER of W<sub>1</sub> and W<sub>2</sub><sup>\*</sup>

Designation	Mass per m	d mm	b mm	t mm	W <sub>1</sub> <sup>*</sup> (kN)						W <sub>2</sub> <sup>*</sup>	FLR					
					1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	9.0	10.0	11.0	12.0	13.0
250 x 150 x 9.0 RHS	51.8	1340	672	448	336	269	224	192	168	149	134	122	112	103	96.0	1220	15.6
6.0 RHS	35.6	943	471	314	236	189	157	135	118	105	94.3	85.7	78.6	72.5	67.4	83.1	16.0
5.0 RHS	29.9	755	378	252	189	151	126	108	94.4	83.9	75.5	68.7	62.9	58.1	53.9	69.8	16.1
200 x 100 x 9.0 RHS	37.7	739	370	246	185	148	123	106	92.4	82.2	73.9	67.2	61.6	56.9	52.8	94.3	8.55
6.0 RHS	26.2	639	265	177	132	106	88.3	75.7	66.2	58.8	53.0	48.1	44.1	40.7	37.8	65.0	8.86
5.0 RHS	22.1	452	226	151	113	90.3	75.3	64.5	56.5	50.2	45.2	41.1	37.6	34.7	32.3	54.7	8.96
4.0 RHS	17.9	370	185	123	92.4	73.9	61.6	52.8	46.2	41.1	37.0	33.6	30.8	28.4	26.4	44.2	9.06
150 x 100 x 6.0 RHS	21.4	339	169	113	84.6	67.7	56.4	48.4	42.3	37.6	33.9	30.8	28.2	26.0	24.2	48.5	11.4
5.0 RHS	18.2	290	145	96.7	72.5	58.0	48.3	41.4	36.3	32.2	29.0	26.4	24.2	22.3	20.7	41.0	11.6
4.0 RHS	14.8	238	119	79.5	59.6	47.7	39.7	34.1	29.8	26.5	23.8	21.7	19.9	18.3	17.0	33.2	11.7
150 x 50 x 5.0 RHS	14.2	199	99.4	66.2	49.7	39.7	33.1	28.4	24.8	22.1	19.9	18.1	16.6	15.3	14.2	39.4	2.93
4.0 RHS	11.6	165	82.4	55.0	41.2	33.0	27.5	23.6	20.6	18.3	16.5	15.0	13.7	12.7	11.8	32.0	3.00
3.0 RHS	8.96	130	64.8	43.2	32.4	25.9	21.6	18.5	16.2	14.4	13.0	11.8	10.8	9.97	9.26	24.3	3.07
125 x 75 x 6.0 RHS	16.7	212	106	70.7	53.1	42.4	35.4	30.3	26.5	23.6	21.2	19.3	17.7	16.3	15.2	39.5	7.62
5.0 RHS	14.2	183	91.7	61.1	45.8	36.7	30.6	26.2	22.9	20.4	18.3	16.7	15.3	14.1	13.1	33.5	7.75
4.0 RHS	11.6	152	75.9	50.6	38.0	30.4	25.3	21.7	19.0	16.9	15.2	13.8	12.7	11.7	10.8	27.2	7.87
3.0 RHS	8.96	119	59.5	39.7	29.8	23.8	19.8	17.0	14.9	13.2	11.9	10.8	9.92	9.16	8.51	20.8	8.90
100 x 50 x 6.0 RHS	12.0	114	57.1	38.1	28.6	22.9	19.0	16.3	14.3	12.7	11.4	10.4	9.52	8.79	8.16	30.4	4.12
5.0 RHS	10.3	100	50.1	33.4	25.0	20.0	16.7	14.3	12.5	11.1	10.0	9.11	8.35	7.71	7.16	25.9	4.23
4.0 RHS	8.49	84.2	42.1	28.1	21.0	16.8	14.0	12.0	10.5	9.35	8.42	7.65	7.02	6.48	6.01	21.2	4.33
3.5 RHS	7.53	75.4	37.7	25.1	18.9	15.1	12.6	10.8	9.43	8.38	7.54	6.86	6.29	5.80	5.39	18.7	4.38
3.0 RHS	6.60	67.2	33.6	22.4	16.8	13.4	11.2	9.60	8.40	7.46	6.72	6.11	5.60	5.17	4.80	16.2	4.44
2.5 RHS	5.56	57.1	28.6	19.0	14.3	11.4	9.52	8.16	7.14	6.35	5.71	4.76	4.39	4.06	3.37	4.49	4.49
2.0 RHS	4.50	46.6	23.3	15.5	11.7	9.38	7.77	6.66	5.83	5.18	4.66	4.24	3.89	3.59	3.33	111	4.53

Notes: 1. FLR = segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -0.8$

2.  $\phi = 0.9$

3.  $\alpha_m = 1.0$

4.  $\alpha_s = 1.0$

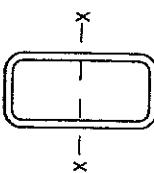
5.  $W_1^* = 8\phi M_s / L$

6.  $W_2^* = 1.6\phi V_y$

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TABLE 8.2 - 3 (2)(A)



**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR CONTINUOUS BEAMS WITH FULL LATERAL RESTRAINT  
RECTANGULAR HOLLOW SECTIONS: GRADE C350  
bending about x-axis**

W<sub>1</sub><sup>\*</sup> = Maximum Design Load based on Design Moment Capacity  
W<sub>2</sub><sup>\*</sup> = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load W<sub>1</sub>' is LESSER of W<sub>1</sub> and W<sub>2</sub>'

Designation	Mass per m	d mm	b mm	t mm	Span of Beam (L) in metres								W <sub>1</sub> ' (kN)	W <sub>2</sub> ' (kN)	FLR	
					0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5				
100 x 50 x 6.0 RHS	12.0	229	152	114	91.4	76.2	65.3	57.1	45.7	38.1	32.6	28.6	25.4	22.9	19.0	304
5.0 RHS	10.3	200	134	100	80.2	66.8	57.3	50.1	40.1	33.4	28.6	25.0	22.3	20.0	16.7	4.12
4.0 RHS	8.49	168	112	84.2	67.4	56.1	48.1	42.1	33.7	28.1	24.1	21.0	18.7	16.8	14.0	4.23
3.5 RHS	7.53	151	101	75.4	60.4	50.3	43.1	37.7	30.2	25.1	21.6	18.9	15.1	12.6	10.7	4.33
3.0 RHS	6.60	134	89.6	67.2	53.7	44.8	38.4	33.6	26.9	22.4	19.2	16.8	14.9	13.4	11.2	4.38
2.5 RHS	5.56	114	76.2	57.1	45.7	38.1	32.6	28.6	22.9	19.0	16.3	14.3	12.7	11.4	10.2	4.44
2.0 RHS	4.50	93.3	62.2	46.6	37.3	31.1	26.6	23.3	18.7	15.5	13.3	11.7	10.4	9.33	7.77	4.49
75 x 50 x 6.0 RHS	9.67	141	94.3	70.7	56.6	47.1	40.4	35.4	28.3	23.6	20.2	17.7	15.7	14.1	11.8	221
5.0 RHS	8.35	125	83.6	62.7	50.2	41.8	35.8	31.4	25.1	20.9	17.9	15.7	13.9	12.5	10.5	5.28
4.0 RHS	6.92	107	71.0	53.3	42.6	35.5	30.4	26.6	21.3	17.8	15.2	13.3	11.8	10.7	9.0	5.43
3.0 RHS	5.42	86.1	57.4	43.1	34.4	28.7	24.6	21.5	17.2	14.4	12.3	10.8	9.57	8.61	7.88	5.57
2.5 RHS	4.58	73.5	49.0	36.8	29.4	24.5	21.0	18.4	14.7	12.3	10.5	9.19	8.17	7.35	6.13	5.73
2.0 RHS	3.72	60.3	40.2	30.1	24.1	20.1	17.2	15.1	12.1	10.0	8.61	7.53	6.70	6.03	5.02	5.79
75 x 25 x 2.5 RHS	3.60	50.7	33.8	25.4	20.3	16.9	14.5	12.7	10.1	8.45	7.24	6.34	5.63	5.07	4.23	5.86
2.0 RHS	2.93	41.9	27.9	20.9	16.8	14.0	12.0	10.5	8.38	6.98	5.98	5.23	4.65	4.19	3.49	1.47
1.6 RHS	2.38	34.3	22.9	17.2	13.7	11.4	9.81	8.59	6.87	5.72	4.91	4.29	3.82	3.43	2.86	1.50
65 x 35 x 3.0 RHS	4.25	55.4	36.9	27.7	22.2	18.5	15.8	13.8	11.1	9.23	7.91	6.92	6.16	5.54	4.62	102
3.60	47.6	31.8	23.8	19.1	15.9	13.6	11.9	9.53	7.94	6.81	5.96	5.29	4.76	3.97	86.8	9.9
2.93	39.3	26.2	19.7	15.7	13.1	11.2	9.83	7.86	6.55	5.62	4.91	4.37	3.93	3.28	70.6	1.53
50 x 25 x 3.0 RHS	3.07	28.5	19.7	14.8	11.8	9.85	8.44	7.39	5.91	4.92	4.22	3.69	3.28	2.95	2.46	76.0
2.5 RHS	2.62	25.7	17.2	12.9	10.3	8.58	7.35	6.43	5.15	4.29	3.68	3.22	2.86	2.57	2.14	64.8
2.0 RHS	2.15	21.5	14.3	10.7	8.59	7.16	6.14	5.37	4.30	3.58	3.07	2.69	2.39	2.15	1.79	53.0
1.6 RHS	1.75	17.8	11.8	8.88	7.11	5.92	5.08	4.44	3.65	2.96	2.54	2.22	1.97	1.78	1.48	43.1
50 x 20 x 3.0 RHS	2.83	26.0	17.3	13.0	10.4	8.66	7.43	6.50	5.20	4.33	3.71	3.25	2.89	2.60	2.17	75.1
2.5 RHS	2.42	22.7	15.2	11.4	9.10	7.58	6.50	5.68	4.55	3.79	3.25	2.84	2.53	2.27	1.89	64.0
2.0 RHS	1.99	19.1	12.7	9.53	7.63	6.35	5.45	4.77	3.81	3.18	2.72	2.38	2.12	1.91	1.59	52.3
1.6 RHS	1.63	15.8	10.5	7.91	6.33	5.27	4.52	3.95	3.16	2.64	2.26	1.98	1.76	1.58	1.32	42.6

Notes: 1. FLR = Segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -0.8$

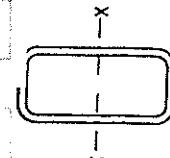
2.  $\phi = 0.9$
3.  $\alpha_m = 1.0$
4.  $\alpha_s = 1.0$
5.  $W_1' = 0.8 M_s / L$
6.  $W_2' = 1.6 \phi V_v$

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**FOR CONTINUOUS BEAMS**

**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
RECTANGULAR HOLLOW SECTIONS: GRADE C450**



W<sub>1</sub> = Maximum Design Load based on Design Moment Capacity  
W<sub>2</sub> = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load W<sub>1</sub> is LESSER of W<sub>1</sub> and W<sub>2</sub>

Designation	Mass per m	W <sub>1</sub> (kN)						W <sub>2</sub>	FLR									
		d mm	b mm	t mm	Span of Beam (L) In metres	W <sub>1</sub> (kN)												
125 x 75 x 3.8 RHS	1.0 kg/m	1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	9.0	10.0	11.0	12.0	13.0	14.0	kN	m	
165 x 82.5 RHS	1.1 kg/m	1.1	1.87	3.4	62.3	46.7	37.4	31.1	26.7	23.3	20.8	18.7	16.5	15.6	14.4	13.3	334	6.14
136 x 87.8 RHS	0.973 kg/m	0.973	1.65	55.0	41.3	33.0	27.5	23.6	20.6	18.3	16.3	15.0	13.8	12.7	11.8	292	6.19	
49.0 x 67.8 RHS	0.839 kg/m	0.839	1.36	45.2	33.9	27.1	22.6	19.4	16.9	15.1	13.6	12.3	11.3	10.4	9.68	250	6.24	
19.6 x 69.0 RHS	0.695 kg/m	0.695	2.3	24.5	19.6	16.3	14.0	12.3	10.9	9.80	8.91	8.17	7.54	7.00	207	6.29		
100 x 50 x 3.3 RHS	0.714 kg/m	0.714	92.3	46.2	30.8	23.1	18.5	15.4	13.2	11.5	10.3	9.23	8.39	7.69	7.10	6.59	228	
20.3 x 51.4 RHS	0.619 kg/m	0.619	40.6	27.1	20.3	16.3	13.5	11.6	10.2	9.03	8.13	7.39	6.77	6.25	5.81	196		
17.0 x 51.4 RHS	0.681 kg/m	0.681	34.1	22.7	17.0	13.6	11.4	9.73	8.52	7.57	6.81	6.19	5.68	5.24	4.87	162		
																3.50		

Notes: Refer to TABLE 8.2-4 (2)(A)

W<sub>1</sub> = Maximum Design Load based on Design Moment Capacity  
W<sub>2</sub> = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load W<sub>1</sub> is LESSER of W<sub>1</sub> and W<sub>2</sub>

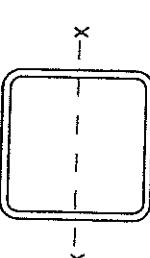
Designation	Mass per m	W <sub>1</sub> (kN)						W <sub>2</sub>	FLR								
		d mm	b mm	t mm	Span of Beam (L) In metres	W <sub>1</sub> (kN)											
185 x 123 RHS	0.5 kg/m	7.14	185	92.3	73.9	61.5	52.8	46.2	36.9	30.8	26.4	23.1	20.5	18.5	15.4	228	3.42
163 x 108 RHS	0.714 kg/m	6.19	163	81.3	65.0	54.2	46.4	40.6	32.5	27.1	23.2	20.3	18.1	16.3	13.5	196	3.47
90.8 x 90.8 RHS	0.514 kg/m	5.14	136	68.1	54.5	45.4	38.9	34.1	27.3	22.7	19.5	17.0	15.1	13.6	11.4	162	3.50
104 x 69.6 RHS	0.909 kg/m	5.09	104	52.2	41.7	34.8	29.8	26.1	20.9	17.4	14.9	13.0	11.6	10.4	8.70	146	4.48
87.8 x 58.6 RHS	0.714 kg/m	4.24	87.8	43.9	35.1	29.3	25.1	22.0	17.6	14.6	12.5	11.0	9.76	8.78	7.32	122	4.53
57.1 x 36.0 RHS	0.514 kg/m	3.34	57.1	38.0	28.5	22.8	19.0	16.3	14.3	11.4	9.51	8.15	7.13	6.34	5.71	4.76	103
24.1 x 18.0 RHS	0.399 kg/m	3.99	67.3	44.9	33.7	26.9	22.4	19.2	16.8	13.5	11.2	9.62	8.42	7.48	6.73	5.61	124
15.5 x 12.4 RHS	0.244 kg/m	2.44	31.0	20.6	15.5	12.4	10.3	8.85	7.74	6.19	5.16	4.51	4.01	3.61	3.01	92.1	1.63
18.3 x 13.7 RHS	0.267 kg/m	2.25	31.8	21.2	15.9	12.7	10.6	9.09	7.95	6.36	5.30	4.54	3.98	3.53	3.18	2.65	77.3
11.0 x 9.14 RHS	0.225 kg/m	2.25	27.4	18.3	13.7	11.0	9.14	7.83	6.85	5.48	4.57	3.92	3.43	3.05	2.74	2.28	1.67
																1.06	

Notes: 1. FLR = segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -0.8$   
2.  $\phi = 0.9$   
3.  $\alpha_m = 1.0$   
4.  $\alpha_s = 1.0$   
5.  $W_1^* = 8\phi M_s/L$   
6.  $W_2^* = 16\lambda V_v$

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TABLE 8.2 – 5 (1)(A)



**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR CONTINUOUS BEAMS WITH FULL LATERAL RESTRAINT  
SQUARE HOLLOW SECTIONS: GRADE C350  
bending about X-axis**

$W_1$  = Maximum Design Load based on Design Moment Capacity  
 $W_2$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_1$  is LESSER of  $W_1^1$  and  $W_1^2$

Designation	Mass per m	Span of Beam (L) In metres										$W_2^1$	FLR				
		1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	9.0	10.0	11.0	12.0	13.0	14.0	KN	m
250 x 250 x 9.0 SHS	65.9	1880	938	625	469	375	313	268	234	208	188	171	156	144	134	1250	41.8
6.0 SHS 45.0	1030	515	343	257	206	172	147	129	114	103	93.6	85.8	79.2	73.5	653	42.4	
200 x 200 x 9.0 SHS	51.8	1170	586	391	293	235	195	168	147	130	117	107	97.7	90.2	83.8	978	33.0
6.0 SHS 35.6	740	370	247	185	148	123	106	92.5	82.2	74.0	67.3	61.7	56.9	52.9	674	33.7	
5.0 SHS 29.9	561	280	187	140	112	93.4	80.1	70.1	62.3	56.1	51.0	46.7	43.1	40.0	567	33.9	
150 x 150 x 9.0 SHS	37.7	625	313	208	156	125	104	89.4	78.2	69.5	62.5	56.9	52.1	48.1	44.7	710	24.3
6.0 SHS 26.2	447	224	149	112	89.5	74.6	63.9	55.9	49.7	44.7	40.7	37.3	34.4	32.0	495	24.9	
5.0 SHS 22.1	364	182	121	91.0	72.8	60.7	52.0	45.5	40.5	36.4	33.1	30.3	28.0	26.0	418	25.2	
125 x 125 x 9.0 SHS	30.6	416	208	139	104	83.1	69.3	59.4	52.0	46.2	41.6	37.8	34.6	32.0	29.7	575	19.9
6.0 SHS 21.4	302	151	101	75.6	60.4	50.4	43.2	37.8	33.6	30.2	27.5	25.2	23.2	21.6	405	20.6	
5.0 SHS 18.2	259	129	86.3	64.7	51.8	43.2	37.0	32.4	28.8	25.9	23.5	21.6	19.9	18.5	333	20.8	
4.0 SHS 14.8	199	99.4	66.2	49.7	39.7	33.1	28.4	24.8	22.1	19.9	18.1	16.6	15.3	14.2	280	21.0	
100 x 100 x 9.0 SHS	23.5	248	124	82.8	62.1	49.7	41.4	35.5	31.0	27.6	24.8	22.6	20.7	19.1	17.7	441	15.5
6.0 SHS 16.7	185	92.7	61.8	46.3	37.1	30.9	26.5	23.2	20.6	18.5	16.8	15.4	14.3	13.2	315	16.2	
5.0 SHS 14.2	160	80.0	53.3	40.0	32.0	26.7	22.9	20.0	17.8	16.0	14.5	13.3	12.3	11.4	289	16.4	
4.0 SHS 11.6	133	66.3	44.2	33.1	26.5	22.1	18.9	16.6	14.7	13.3	12.1	11.0	10.2	9.47	220	16.6	
3.0 SHS 8.96	93.5	46.8	31.2	23.4	18.7	15.6	13.4	11.7	10.4	9.35	8.50	7.79	7.19	6.68	168	16.9	
89 x 89 x 6.0 SHS	14.6	143	71.3	47.5	35.6	28.5	23.8	20.4	17.8	15.8	14.3	13.0	11.9	11.0	10.2	276	14.2
5.0 SHS 12.5	124	61.9	41.2	30.9	24.7	20.6	17.7	15.5	13.7	12.4	11.2	10.3	9.52	8.84	236	14.5	
3.5 SHS 9.06	91.9	45.9	30.6	23.0	18.4	15.3	13.1	11.5	10.2	9.19	8.35	7.65	7.07	6.56	171	14.8	
75 x 75 x 6.0 SHS	12.0	96.8	48.4	32.3	24.2	19.4	16.1	13.8	12.1	10.8	9.68	8.80	8.07	7.45	6.91	226	11.8
5.0 SHS 10.3	84.8	42.4	28.3	21.2	17.0	14.1	12.1	10.6	9.42	8.48	7.71	7.07	6.52	6.06	194	12.0	
4.0 SHS 8.49	71.2	35.6	23.7	17.8	14.2	11.9	10.2	8.90	7.91	7.12	6.47	5.93	5.48	5.08	160	12.2	
3.5 SHS 7.53	63.8	31.9	21.3	15.9	12.8	10.6	9.11	7.97	7.08	6.38	5.80	5.31	4.90	4.55	142	12.4	
3.0 SHS 6.60	56.7	28.3	18.9	14.2	11.3	9.44	8.09	7.08	6.30	5.67	5.15	4.72	4.36	4.05	124	12.5	
2.5 SHS 5.56	46.0	23.0	15.3	11.5	9.20	7.67	6.57	5.75	5.11	4.60	4.18	3.83	3.54	3.29	105	12.6	

Notes:

1. FLR = segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -0.8$
2.  $\phi = 0.9$
3.  $c_m = 1.0$
4.  $c_s = 1.0$
5.  $W_1^1 = 8\phi M_s/L$
6.  $W_2^1 = 2\phi V_y$

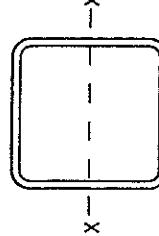
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**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS**  
**FOR CONTINUOUS BEAMS WITH FULL LATERAL RESTRAINT**  
**SQUARE HOLLOW SECTIONS: GRADE C350**

bending about x-axis

W<sub>1</sub> = Maximum Design Load based on Design Moment Capacity  
 W<sub>2</sub> = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load, W<sub>1</sub> is LESSER of W<sub>1</sub> and W<sub>2</sub>



Designation d b t mm mm mm	Mass per m kg/m	W <sub>1</sub> ' (kN)							W <sub>2</sub> kN	FLR m
		0.5	0.75	1.0	1.25	1.5	1.75	2.0		
75 x 75 x 6.0 SHS	12.0	194	129	96.8	77.4	64.5	55.3	48.4	38.7	27.7
5.0 SHS	10.3	170	113	84.8	67.8	56.5	48.4	42.4	33.9	24.2
4.0 SHS	8.49	142	94.9	71.2	56.9	47.5	40.7	35.6	30.5	23.3
3.5 SHS	7.53	128	85.0	63.8	51.0	42.5	36.4	31.9	25.5	21.3
3.0 SHS	6.60	113	75.6	56.7	45.3	37.8	32.4	28.3	22.7	18.2
2.5 SHS	5.56	92.0	61.4	46.0	36.8	30.7	26.3	23.0	18.4	15.3
65 x 65 x 6.0 SHS	10.1	139	92.4	69.3	55.5	46.2	39.6	34.7	27.7	23.1
5.0 SHS	8.75	123	81.7	61.3	49.0	40.9	35.0	30.6	24.5	20.4
4.0 SHS	7.23	104	69.2	51.9	41.5	34.6	29.7	26.0	20.8	17.3
3.0 SHS	5.66	83.5	55.7	41.8	33.4	27.8	23.9	20.9	16.7	13.0
2.5 SHS	4.78	71.3	47.5	35.6	28.5	23.8	20.4	17.8	14.3	11.9
2.0 SHS	3.88	53.3	35.6	26.7	21.3	17.8	15.2	13.3	10.7	8.9
50 x 50 x 5.0 SHS	6.39	66.3	44.2	33.2	26.5	22.1	18.9	16.6	13.3	11.1
4.0 SHS	5.35	57.4	38.2	28.7	22.9	19.1	16.4	14.3	11.5	9.5
3.0 SHS	4.25	47.3	31.5	23.7	18.9	15.8	13.5	11.8	9.4	7.8
2.5 SHS	3.60	40.7	27.1	20.3	16.3	13.6	11.6	10.2	8.14	6.78
2.0 SHS	2.93	33.6	22.4	16.8	13.4	11.2	9.59	8.39	6.72	5.60
1.6 SHS	2.38	25.7	17.1	12.9	10.3	8.57	7.35	6.43	5.14	4.29
40 x 40 x 4.0 SHS	4.09	33.9	22.6	17.0	13.6	11.3	9.70	8.49	6.79	5.66
3.0 SHS	3.30	28.8	19.2	14.4	11.5	9.62	8.24	7.21	5.77	4.81
2.5 SHS	2.82	25.0	16.7	12.5	10.0	8.34	7.15	6.28	5.01	4.17
2.0 SHS	2.31	20.8	13.9	10.4	8.33	6.94	5.95	5.21	4.17	3.47
1.6 SHS	1.88	17.2	11.5	8.60	6.88	5.73	4.91	4.30	3.44	2.87
35 x 35 x 3.0 SHS	2.83	21.3	14.2	10.7	8.53	7.10	6.09	5.33	4.26	3.55
2.5 SHS	2.42	18.6	12.4	9.31	7.45	6.21	5.32	4.66	3.72	3.10
2.0 SHS	1.99	15.6	10.4	7.80	6.24	5.20	4.46	3.90	3.12	2.60
1.6 SHS	1.63	12.9	8.62	6.47	5.17	4.31	3.70	3.23	2.59	2.16
30 x 30 x 3.0 SHS	2.36	14.9	9.94	7.46	5.97	4.97	4.26	3.73	2.98	2.49
2.5 SHS	2.03	13.2	8.77	6.58	5.26	4.38	3.76	3.29	2.63	2.19
2.0 SHS	1.68	11.1	7.41	5.56	4.45	3.70	3.18	2.73	2.22	1.85
1.6 SHS	1.38	9.28	6.19	4.64	3.71	3.09	2.65	2.32	1.86	1.55
25 x 25 x 3.0 SHS	1.89	9.65	6.43	4.83	3.86	3.22	2.76	2.41	1.93	1.61
2.5 SHS	1.64	8.63	5.75	4.32	3.45	2.88	2.47	2.16	1.73	1.44
2.0 SHS	1.36	7.39	4.93	3.69	2.96	2.46	2.11	1.85	1.48	1.23
1.6 SHS	1.12	6.23	4.15	3.11	2.49	2.08	1.78	1.56	1.25	1.04
20 x 20 x 3.0 SHS	0.873	3.78	2.52	1.89	1.51	1.26	1.08	0.946	0.757	0.631

Notes:  
 1. To Table 2-5 (1)



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## FOR CONTINUOUS BEAMS

## SQUARE HOLLOW SECTIONS: GRADE C450

$W_{L1}^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_{L2}^*$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L^*$  is LESSER of  $W_{L1}^*$  and  $W_{L2}^*$

$W_{L1}$  = Maximum Design Load based on Design Moment Capacity  
 $W_{L2}$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L$  is LESSER of  $W_{L1}$  and  $W_{L2}$

TABLE 8.2 – 6 (1)(A)

Designation	Mass per m		$W_{L1}$ (kN)						$W_L$	FLR												
			Span of Beam (L) in metres																			
d b t	mm mm mm	kg/m	1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	9.0	10.0	11.0	12.0	13.0	14.0	kN	m				
100 x 100 x 3.8 SHS	11.1	157	78.3	52.2	39.2	31.3	26.1	22.4	19.6	17.4	15.7	14.2	13.1	12.0	11.0	12.0	13.0	14.0	kN	m		
3.3 SHS	9.73	127	63.7	42.4	31.8	25.5	21.2	18.2	15.9	14.1	12.7	11.6	10.6	9.80	9.10	10.6	11.2	12.0	27.0	13.0		
2.8 SHS	8.39	100	50.1	33.4	25.1	20.1	16.7	14.3	12.5	11.1	10.0	9.12	8.36	7.71	7.16	8.36	7.71	8.36	9.10	23.7	13.1	
2.3 SHS	6.95	74.8	37.4	24.9	18.7	15.0	12.5	10.7	9.34	8.31	7.48	6.80	6.23	5.75	5.34	6.80	5.75	6.23	7.16	20.3	13.2	
75 x 75 x 3.3 SHS	7.14	78.0	39.0	26.0	19.5	15.6	13.0	11.1	9.75	8.67	7.80	7.09	6.50	6.00	5.57	6.50	6.00	5.57	6.00	16.9	13.2	
2.8 SHS	6.19	65.3	32.6	21.8	16.3	13.1	10.9	9.32	8.16	7.25	6.53	5.93	5.44	5.02	4.66	5.93	5.44	5.02	4.66	14.9	9.65	
2.3 SHS	5.14	48.6	24.3	16.2	12.1	9.71	8.09	6.94	6.07	5.39	4.86	4.41	4.05	3.73	3.47	4.86	4.41	4.05	3.73	3.47	12.4	9.76

Notes: Refer to TABLE 8.2-5(1)(A)

TABLE 8.2 – 6 (2)(A)

Designation	Mass per m		$W_{L1}$ (kN)						$W_L$	FLR										
			Span of Beam (L) in metres																	
d b t	mm mm mm	kg/m	0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5	3.0	3.5	4.0	4.5	5.0	6.0	kN	m		
75 x 75 x 3.3 SHS	7.14	156	104	78.0	62.4	52.0	44.6	39.0	31.2	26.0	22.3	19.5	17.3	15.6	13.0	17.3	14.9	17.3	9.65	9.76
2.8 SHS	6.19	131	87.0	65.3	52.2	43.5	37.3	32.6	26.1	21.8	18.6	16.3	14.5	13.1	10.9	10.9	14.9	14.9	12.4	9.84
2.3 SHS	5.14	97.1	64.7	48.6	38.8	32.4	27.7	24.3	19.4	16.2	13.9	12.1	10.8	9.71	9.71	9.71	10.9	10.9	12.4	9.84
65 x 65 x 2.8 SHS	5.31	101	67.5	50.6	40.5	33.7	28.9	25.3	20.2	16.9	14.5	12.6	11.2	10.1	8.43	8.43	8.43	10.1	12.4	8.40
2.3 SHS	4.42	78.6	52.4	39.3	31.4	26.2	22.5	19.6	15.7	13.1	11.2	9.82	8.73	7.86	6.55	6.55	6.55	10.9	10.9	8.40
50 x 50 x 2.8 SHS	3.99	57.5	38.3	26.8	23.0	19.2	16.4	14.4	11.5	9.58	8.21	7.19	6.39	5.75	4.79	4.79	5.75	5.75	5.75	8.40
2.3 SHS	3.34	48.7	32.5	24.4	19.5	16.2	13.9	12.2	9.75	8.12	6.96	6.09	5.42	4.87	4.87	4.87	5.42	5.42	5.42	8.40
40 x 40 x 2.8 SHS	3.11	35.2	23.5	17.6	14.1	11.7	10.1	8.80	7.04	5.86	5.03	4.40	3.91	3.52	2.93	3.01	3.01	2.51	2.51	8.40
35 x 35 x 2.8 SHS	2.67	26.1	17.4	13.0	10.4	8.69	7.45	6.52	5.21	4.34	3.72	3.26	2.90	2.61	2.17	2.17	2.17	2.17	2.17	8.40
2.3 SHS	2.25	22.4	15.0	11.2	8.98	7.48	6.41	5.61	4.49	3.74	3.21	2.80	2.49	2.24	1.87	1.87	1.87	1.87	1.87	8.40
30 x 30 x 2.8 SHS	2.23	18.3	12.2	9.16	7.33	6.10	5.23	4.58	3.66	3.05	2.62	2.29	2.03	1.83	1.53	1.53	1.53	1.53	1.53	8.40
2.3 SHS	1.89	15.9	10.6	7.95	6.36	5.30	4.54	3.98	3.18	2.65	2.27	1.99	1.77	1.59	1.33	1.33	1.33	1.33	1.33	8.40
25 x 25 x 2.8 SHS	1.53	10.5	7.00	5.25	4.20	3.50	3.00	2.62	2.10	1.75	1.50	1.31	1.17	1.05	0.874	0.874	0.874	0.874	0.874	8.40

Notes: Refer to TABLE 8.2-5(1)(A)

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# FOR FIXED END BEAMS WITH FULL LATERAL RESTRAINT

## CIRCULAR HOLLOW SECTIONS: GRADE C250 bending about any axis

TABLE 8.3 – 1 (1)(A)

$W_L^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_L^{\dagger}$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L^*$  is LESSER of  $W_L^*$  and  $W_L^{\dagger}$

Designation	$d_o$ mm	$t$ mm	Mass per m kg/m	Span of Beam (L) in metres						$W_L^{\dagger}$ (kN)	$W_L^*$ (kN)	
				1.0	2.0	3.0	4.0	5.0	6.0			
610.0 x 12.7 CHS	187	12200	6120	4080	3060	2450	2040	1750	1530	1360	1020	874
9.5 CHS	141	8820	4410	2940	2210	1760	1470	1260	1100	980	735	765
6.4 CHS	95.3	5390	2890	1800	1350	1080	893	770	673	599	449	552
508.0 x 12.7 CHS	155	8410	4210	2800	2100	1680	1400	1200	1050	935	701	601
9.5 CHS	117	6300	3150	2100	1580	1260	1050	900	788	700	630	525
6.4 CHS	79.2	3940	1970	1310	985	788	635	583	492	438	394	328
												246
												1630

Notes: Refer to TABLE 8.3-1 (3)(A)

TABLE 8.3 – 1 (2)(A)

$W_L^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_L^{\dagger}$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L^*$  is LESSER of  $W_L^*$  and  $W_L^{\dagger}$

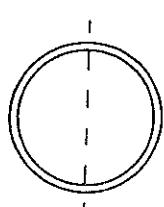
Designation	$d_o$ mm	$t$ mm	Mass per m kg/m	Span of Beam (L) in metres						$W_L^{\dagger}$ (kN)	$W_L^*$ (kN)	
				0.5	1.0	1.5	2.0	2.5	3.0			
165.1 x 5.4 CHS	21.3	744	372	248	186	149	124	106	93.0	74.4	62.0	53.1
5.0 CHS	19.7	692	346	231	173	138	115	98.9	86.5	69.2	57.7	49.4
139.7 x 5.4 CHS	17.9	526	263	175	132	105	87.7	75.2	65.8	52.6	43.9	34.3
5.0 CHS	16.6	490	245	163	123	98.0	81.7	70.0	61.3	49.0	40.8	36.5
114.3 x 5.4 CHS	14.5	346	173	115	86.5	69.2	57.7	49.4	43.3	37.6	30.6	27.2
4.5 CHS	12.2	293	147	97.7	73.3	58.6	48.9	41.9	36.6	29.3	24.4	20.9
101.6 x 5.0 CHS	11.9	252	126	84.1	63.0	50.4	42.0	36.0	31.5	25.2	21.0	18.3
4.0 CHS	9.63	206	103	68.6	51.5	41.2	34.3	29.4	25.7	20.6	17.2	14.7
88.9 x 5.9 CHS	12.1	220	110	73.3	55.0	44.0	36.6	31.4	27.5	22.0	18.3	15.7
5.0 CHS	10.3	190	95.1	63.4	47.6	38.1	31.7	27.2	23.8	19.0	15.9	13.6
4.0 CHS	8.38	156	77.9	51.9	39.0	31.2	26.0	22.3	19.5	15.6	13.0	11.1
76.1 x 5.9 CHS	10.2	157	78.7	52.5	39.3	31.5	26.2	22.5	19.7	15.7	13.7	12.2
4.5 CHS	7.95	125	62.4	41.6	31.2	24.9	20.8	17.8	15.6	12.5	10.4	9.0
3.6 CHS	6.44	102	51.1	34.1	25.6	20.5	17.0	14.6	12.8	10.2	8.52	7.30
												173
												133

Notes: Refer to TABLE 8.3-1 (3)(A)

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TABLE 8.3 - 1 (3)(A)



**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR FIXED END BEAMS WITH FULL LATERAL RESTRAINT  
CIRCULAR HOLLOW SECTIONS: GRADE C250**

$W_{L1}^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_{L2}^*$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L^*$  is LESSER of  $W_{L1}^*$  and  $W_{L2}^*$

Designation	Mass per m	$d_0$	$t$	Span of Beam (L) in metres								$W_L^* (kN)$	
				0.25	0.5	1.0	1.25	1.5	1.75	2.0	2.5	3.0	
76.1 x 5.9 CHS	10.2	315	157	105	187	63.0	52.5	45.0	39.3	31.5	26.2	22.5	19.7
4.5 CHS	7.95	249	125	83.2	62.4	49.9	41.6	35.6	31.2	24.9	20.8	17.8	15.6
3.6 CHS	6.44	205	102	68.2	51.1	40.9	34.1	29.2	25.6	20.5	17.0	14.6	12.8
60.3 x 5.4 CHS	7.31	176	88.2	58.8	44.1	35.3	29.4	25.2	22.0	17.6	14.7	12.6	11.0
4.5 CHS	6.19	152	75.8	50.6	37.9	30.3	25.3	21.7	19.0	15.2	12.6	10.8	9.48
3.6 CHS	5.03	125	62.6	41.7	31.3	25.0	20.9	17.9	15.6	12.5	10.4	8.94	7.82
48.3 x 5.4 CHS	5.71	108	63.9	36.0	27.0	21.6	18.0	16.4	13.5	10.8	8.99	7.71	6.74
4.0 CHS	4.37	85.0	42.5	28.3	21.3	17.0	14.2	12.1	10.6	8.50	7.08	6.07	5.31
3.2 CHS	3.56	70.4	35.2	23.5	17.6	14.1	11.7	10.1	8.80	7.04	5.87	5.03	4.40
42.4 x 4.9 CHS	4.53	74.8	37.4	24.9	18.7	15.0	12.6	10.7	9.36	7.48	6.24	5.35	4.40
4.0 CHS	3.79	63.9	32.0	21.3	16.0	12.8	10.7	9.13	7.99	6.39	5.33	4.57	3.74
3.2 CHS	3.09	53.2	26.8	17.7	13.3	10.6	8.87	7.60	6.65	5.32	4.44	3.80	3.33
33.7 x 4.5 CHS	3.24	41.8	20.9	13.9	10.4	8.35	6.96	5.97	5.22	4.18	3.48	2.98	2.61
4.0 CHS	2.93	38.3	19.2	12.8	9.58	7.67	6.39	5.48	4.79	3.83	3.19	2.74	2.40
3.2 CHS	2.41	32.3	16.1	10.8	8.07	6.45	5.38	4.61	4.03	3.23	2.69	2.30	2.02
26.9 x 4.0 CHS	2.26	22.9	11.4	7.63	5.72	4.58	3.81	3.27	2.86	2.29	1.91	1.63	1.43
3.2 CHS	1.87	19.5	9.76	6.51	4.88	3.91	3.25	2.79	2.44	1.95	1.63	1.39	1.22
2.6 CHS	1.56	16.6	8.32	5.55	4.16	3.33	2.77	2.38	2.08	1.66	1.39	1.19	1.04
21.3 x 3.6 CHS	1.57	12.3	6.17	4.12	3.09	2.47	2.06	1.76	1.64	1.23	1.03	0.882	0.772
3.2 CHS	1.43	11.4	5.72	3.81	2.86	2.29	1.91	1.63	1.43	1.14	0.953	0.817	0.715
2.6 CHS	1.20	9.88	4.94	3.29	2.47	1.98	1.65	1.41	1.24	0.988	0.824	0.706	0.618

Notes:

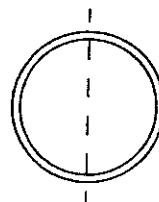
1.  $\phi = 0.9$
2.  $\alpha_m = 1.0$
3.  $\alpha_s = 1.0$
4.  $W_{L1}^* = 12 \phi M_s / L$
5.  $W_{L2}^* = 2 \phi V_v$

 $W_L^*$

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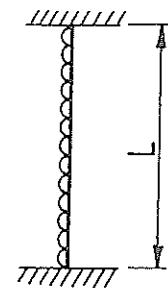
@Seismicisolation

TABLE 8.3 - 2 (1)(A)



**STRENGTH/LIMIT STATE MAXIMUM DESIGN LOADS  
FOR FIXED END BEAMS WITH FULL LATERAL RESTRAINT  
CIRCULAR HOLLOW SECTIONS: GRADE C350  
bending about any axis**

$W_L^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_2^*$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L$  is LESSER of  $W_L^*$  and  $W_2^*$



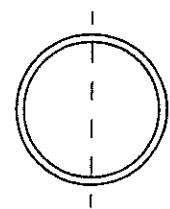
Designation $d_o$ mm	$t$ mm	Mass per m kg/m	Span of Beam (L) in metres						$W_L^*$ (kN)	$W_2^*$ kN
			1.0	2.0	3.0	4.0	5.0	6.0		
457.0 x 12.7 CHS	139	9470	4730	3160	2370	1890	1580	1350	1180	1050
9.5 CHS	105	6780	3390	2260	1700	1350	1130	969	848	947
6.4 CHS	71.1	4120	2060	1370	1030	824	687	589	515	753
406.4 x 12.7 CHS	123	7440	3720	2480	1880	1490	1240	1060	930	827
9.5 CHS	93.0	5470	2740	1820	1370	1090	912	782	684	744
6.4 CHS	63.1	3380	1690	1130	846	677	564	483	423	620
355.6 x 12.7 CHS	107	5650	2820	1880	1410	1130	941	807	706	627
9.5 CHS	81.1	4270	2130	1420	1070	854	711	610	533	565
6.4 CHS	55.1	2680	1340	894	671	537	447	383	335	474
323.9 x 12.7 CHS	97.5	4650	2330	1560	1160	930	775	665	581	517
9.5 CHS	73.7	3550	1780	1180	888	710	592	507	444	465
6.4 CHS	50.1	2270	1140	737	568	454	379	325	284	395
273.1 x 9.3 CHS	60.5	2450	1220	816	612	489	408	350	306	272
6.4 CHS	42.1	1670	833	555	416	333	278	238	208	185
4.8 CHS	31.8	1180	560	393	295	236	197	169	147	131
219.1 x 8.2 CHS	42.6	1380	690	460	345	276	230	197	172	153
6.4 CHS	33.6	1090	547	365	274	219	182	156	137	122
4.8 CHS	25.4	795	398	285	199	159	133	114	99.4	88.3
168.3 x 7.1 CHS	28.2	698	349	233	174	140	116	99.7	87.2	77.5
6.4 CHS	25.6	634	317	211	159	127	106	90.6	79.3	69.8
4.8 CHS	19.4	465	243	162	121	97.0	80.9	69.3	60.6	53.9

- Notes:  
 1.  $\phi = 0.9$   
 2.  $c_m = 1.0$   
 3.  $c_s = 1.0$   
 4.  $W_L^* = 12\phi M_s / L$   
 5.  $W_2^* = 2\phi V_v$

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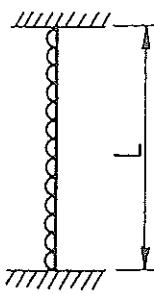
@Seismicisolation

TABLE 8.3 - 2 (2)(A)



**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR **FIXED END BEAMS** WITH FULL LATERAL RESTRAINT  
CIRCULAR HOLLOW SECTIONS: GRADE C350  
bending about any axis**

$W_{L1}^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_{L2}^*$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L^*$  is LESSER of  $W_{L1}^*$  and  $W_{L2}^*$



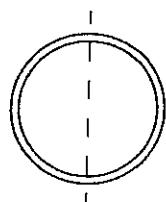
Designation $d_o$ mm	Mass per m kg/m		Span of Beam (L) in metres								$W_{L2}^*$				
			0.5	1.0	1.5	2.0	2.5	3.0	3.5	4.0	5.0	6.0	7.0	8.0	9.0
168.3 x 7.1 CHS	28.2	1400	698	465	349	279	233	199	174	140	116	99.7	87.2	77.5	61.5
6.4 CHS	25.6	1270	634	423	317	254	211	181	159	127	106	90.6	79.3	70.5	73.8
4.8 CHS	19.4	970	485	323	243	194	162	139	121	97.0	80.9	69.3	60.6	53.9	55.9
165.1 x 3.5 CHS	13.9	655	327	218	164	131	109	93.5	81.8	65.5	54.5	46.8	40.9	36.4	40.3
3.0 CHS	12.0	543	272	181	136	109	90.6	77.6	67.9	54.3	45.3	38.8	34.0	30.2	34.6
139.7 x 3.5 CHS	11.8	481	241	160	120	96.3	80.2	68.8	60.2	48.1	40.1	34.4	30.1	26.7	34.0
3.0 CHS	10.1	403	201	134	101	80.5	67.1	57.5	50.3	40.3	33.6	28.8	25.2	22.4	29.2
114.3 x 6.0 CHS	16.0	533	286	178	133	107	88.8	76.1	66.6	53.3	44.4	38.0	33.3	29.6	46.3
4.8 CHS	13.0	435	218	145	109	87.1	72.6	62.2	54.4	43.5	36.3	31.1	27.2	24.2	37.4
3.6 CHS	9.63	334	167	111	83.4	66.7	55.6	47.7	41.7	33.4	27.8	23.8	20.9	18.5	28.4
3.2 CHS	8.77	299	149	99.6	74.7	59.7	49.8	42.7	37.3	29.9	24.9	21.3	18.7	16.6	25.3
101.6 x 3.2 CHS	7.77	234	117	78.1	58.6	46.9	39.1	33.5	29.3	23.4	19.5	16.7	14.6	13.0	22.4
2.6 CHS	6.35	190	94.8	63.2	47.4	37.9	31.6	27.1	23.7	19.0	15.8	13.5	11.9	10.5	18.3
88.9 x 5.5 CHS	11.3	290	145	96.5	72.4	57.9	48.3	41.4	36.2	29.0	24.1	20.7	18.1	16.1	32.7
4.8 CHS	9.96	257	128	85.6	64.2	51.4	42.8	36.7	32.1	25.7	21.4	18.4	16.1	14.3	28.8
3.2 CHS	6.76	178	88.9	59.3	44.4	35.6	29.6	25.4	22.2	17.8	14.8	12.7	11.1	9.88	19.5
2.6 CHS	5.53	146	73.2	48.8	36.6	29.3	24.4	20.9	18.3	14.6	12.2	10.5	9.15	8.14	16.0
76.1 x 3.2 CHS	5.75	129	64.3	42.9	32.2	25.7	21.4	18.4	16.1	12.9	10.7	9.19	8.04	7.15	16.6
2.3 CHS	4.19	94.7	47.4	31.6	23.7	18.9	15.8	13.5	11.8	9.47	7.89	6.77	5.92	5.26	12.1

- Notes:
1.  $\phi = 0.9$
  2.  $\alpha_m = 1.0$
  3.  $\alpha_s = 1.0$
  4.  $W_{L1}^* = 12 \phi M_s / L$
  5.  $W_{L2}^* = 2 \phi V_v$

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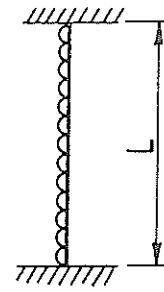
@Seismicisolation

TABLE 8.3 – 2 (3)(A)



**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR FIXED END BEAMS WITH FULL LATERAL RESTRAINT  
CIRCULAR HOLLOW SECTIONS: GRADE C350  
bending about any axis**

$W_L^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_L^t$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L^*$  is LESSER of  $W_L^*$  and  $W_L^t$

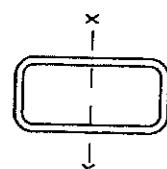


Designation $d_o$ t	Mass per m kg/m	Span of Beam (L) In metres								$W_L^t$ (kN)				
		0.25	0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5	3.0	3.5	4.0	5.0
76.1x3.2 CHS	5.75	257	129	85.8	64.3	51.5	42.9	36.8	32.2	25.7	21.4	18.4	16.1	12.9
2.3 CHS	4.19	189	94.7	63.2	47.4	37.9	31.6	27.1	23.7	18.9	15.8	13.5	11.8	9.47
60.3x2.9 CHS	4.11	145	72.3	48.2	36.1	28.9	24.1	20.7	18.1	14.5	12.0	10.3	9.04	7.23
2.3 CHS	3.29	117	58.5	39.0	28.3	23.4	19.5	16.7	14.6	11.7	9.75	8.36	7.32	6.65
48.3x2.9 CHS	3.25	90.5	45.3	30.2	22.6	18.1	15.1	12.9	11.3	9.05	7.54	6.46	5.66	4.53
2.3 CHS	2.61	73.6	36.8	24.5	18.4	14.7	12.3	10.5	9.21	7.36	6.14	5.26	4.60	3.68
42.4x2.6 CHS	2.55	62.4	31.2	20.8	15.6	12.5	10.4	8.91	7.80	6.24	5.20	4.45	3.90	3.12
2.0 CHS	1.99	49.4	24.7	16.5	12.3	9.86	8.23	7.06	6.17	4.94	4.12	3.53	3.09	2.47
33.7x2.6 CHS	1.99	38.1	19.1	12.7	9.53	7.62	6.35	5.44	4.76	3.81	3.18	2.72	2.38	1.91
2.0 CHS	1.56	30.4	15.2	10.1	7.61	6.09	5.07	4.35	3.80	3.04	2.54	2.17	1.90	1.52
26.9x2.3 CHS	1.40	21.1	10.6	7.04	5.28	4.22	3.52	3.02	2.64	2.11	1.76	1.51	1.32	1.06
2.0 CHS	1.23	16.8	9.39	6.26	4.70	3.76	3.13	2.68	2.35	1.86	1.57	1.34	1.17	0.939
21.3x2.0 CHS	0.952	11.3	5.65	3.77	2.83	2.26	1.88	1.61	1.41	1.13	0.942	0.807	0.707	0.565
														27.5

- Notes: 1.  $\phi = 0.9$   
 2.  $\alpha_m = 1.0$   
 3.  $\alpha_s = 1.0$   
 4.  $W_L^* = 12\phi M_s / L$   
 5.  $W_L^t = 2\phi V_y$

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**TABLE 8.3 – 3 (1)(A)**

**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR FIXED END BEAMS WITH FULL LATERAL RESTRAINT  
RECTANGULAR HOLLOW SECTIONS: GRADE C350  
bending about x-axis**

W<sub>L1</sub> = Maximum Design Load based on Design Moment Capacity  
W<sub>L2</sub> = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load W<sub>L</sub> is LESSER of W<sub>L1</sub> and W<sub>L2</sub>

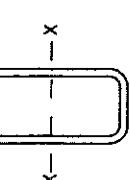
Designation	Mass per m	d mm	b mm	t mm	W <sub>L1</sub> (kN)								W <sub>L2</sub>	FLR														
					Span of Beam (L) in metres				1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	9.0	10.0	11.0	12.0	13.0	14.0						
250 x 150 x 9.0 RHS	51.8	2920	1010	672	504	403	336	288	252	224	202	183	168	155	144	130	120	11.0	12.0	13.0	14.0	KN	m					
6.0 RHS	35.6	1410	707	471	354	283	236	202	177	157	141	129	118	109	101	94.4	87.1	80.9	79.2	72.2	66.2	61.1	56.7	1520	15.6			
5.0 RHS	29.9	1130	566	378	283	227	189	162	142	126	113	103	94.4	87.1	80.9	79.2	72.2	66.2	61.1	56.7	52.1	48.4	44.7	41.0	1040	16.0		
200 x 100 x 9.0 RHS	37.7	1110	555	370	277	222	185	158	139	123	111	101	92.4	89.3	84.7	79.4	75.3	70.7	67.7	63.5	59.3	55.4	50.4	46.2	42.6	39.6	873	16.1
6.0 RHS	26.2	794	397	265	199	159	132	113	99.3	88.3	79.4	72.2	66.2	61.1	56.7	52.1	48.4	44.7	41.0	37.3	33.5	30.8	27.5	24.2	20.9	18.6	1180	8.55
5.0 RHS	22.1	677	339	226	169	135	113	96.8	84.7	75.3	67.7	61.6	56.5	52.1	48.4	44.7	41.0	37.3	33.5	30.8	27.5	24.2	20.9	18.6	1180	8.55		
4.0 RHS	17.9	554	277	185	139	111	92.4	79.2	69.3	61.6	55.4	50.4	46.2	42.6	39.6	35.8	33.5	30.8	27.5	24.2	20.9	18.6	1180	8.55				
150 x 100 x 6.0 RHS	21.4	508	254	169	127	102	84.6	72.5	63.5	56.4	50.8	46.2	42.3	39.1	36.3	33.5	30.8	27.5	24.2	20.9	18.6	16.0	13.5	11.4	10.0	873	9.06	
5.0 RHS	18.2	435	218	145	109	87.0	72.5	62.2	54.4	48.3	43.5	39.6	36.3	33.5	30.8	27.5	24.2	20.9	18.6	16.0	13.5	11.4	10.0	873	9.06			
4.0 RHS	14.8	358	179	119	89.4	71.5	59.6	51.1	44.7	39.7	35.8	32.5	29.8	27.5	25.6	23.2	20.9	18.6	16.0	13.5	11.4	10.0	873	9.06				
150 x 50 x 5.0 RHS	14.2	298	149	99.4	74.5	59.6	48.7	42.6	37.3	33.1	29.8	27.1	24.8	22.9	21.3	19.0	17.7	16.2	15.0	13.9	12.6	11.4	10.0	873	11.6			
4.0 RHS	11.6	247	124	82.4	61.8	49.5	41.2	35.3	30.9	27.5	24.7	22.5	20.6	19.0	17.7	16.2	15.0	13.9	12.6	11.4	10.0	873	11.6					
3.0 RHS	8.96	194	97.2	64.8	48.6	38.9	32.4	27.8	24.3	21.6	19.4	17.7	16.2	15.0	13.9	12.6	11.4	10.0	873	11.6								
125 x 75 x 6.0 RHS	16.7	318	159	106	79.6	63.7	53.1	45.5	39.8	35.4	31.8	28.9	26.5	24.5	22.7	20.6	19.0	17.7	16.2	15.0	13.9	12.6	11.4	10.0	873	3.07		
5.0 RHS	14.2	275	137	91.7	68.7	55.0	45.8	39.3	34.4	30.6	27.5	25.0	22.9	21.2	19.6	17.7	16.2	15.0	13.9	12.6	11.4	10.0	873	3.07				
4.0 RHS	11.6	228	114	75.9	57.0	45.6	38.0	32.5	28.5	25.3	22.8	20.7	19.0	17.5	16.3	15.0	13.9	12.6	11.4	10.0	873	3.07						
3.0 RHS	8.96	179	89.3	59.5	44.7	35.7	29.8	25.5	22.3	19.8	17.9	16.2	14.9	13.7	12.8	11.4	10.0	873	3.07									
100 x 75 x 6.0 RHS	12.0	171	85.7	57.1	42.8	34.3	28.6	24.5	21.4	19.0	17.1	15.6	14.3	13.2	12.2	11.6	10.7	9.71	9.02	8.00	7.07	6.00	4.93	3.99	2.93	1.00		
5.0 RHS	10.3	150	75.1	50.1	37.6	30.1	25.0	21.5	18.8	16.7	15.0	13.7	12.5	11.6	10.7	9.71	9.02	8.00	7.07	6.00	4.93	3.99	2.93	1.00				
4.0 RHS	8.49	126	63.1	42.1	31.6	25.3	21.0	18.0	15.8	14.0	12.6	11.5	10.5	9.71	9.02	8.00	7.07	6.00	4.93	3.99	2.93	1.00						
3.5 RHS	7.53	113	56.6	37.7	28.3	22.6	18.9	16.2	14.1	12.6	11.3	10.3	9.43	8.71	8.08	7.07	6.00	4.93	3.99	2.93	1.00							
3.0 RHS	6.60	101	50.4	33.6	25.2	20.2	16.8	14.4	12.6	11.2	10.1	9.16	8.40	7.75	7.07	6.00	4.93	3.99	2.93	1.00								
2.5 RHS	5.56	85.7	42.8	28.6	21.4	14.3	12.2	10.7	9.52	8.57	7.79	7.14	6.59	6.12	5.49	4.44	3.48	2.03	1.00									
2.0 RHS	4.50	69.9	35.0	23.3	17.5	14.0	11.7	9.99	8.74	7.77	6.99	6.36	5.83	5.38	5.00	4.49	3.48	2.03	1.00									

Notes:

- FLR = segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -0.8$
- $\phi = 0.9$
- $c_m = 1.0$
- $a_s = 1.0$
- $W_{L1} = 12\phi M_s / L$
- $W_{L2} = 2\phi V_y$

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**TABLE 8.3 – 3 (2)(A)**

**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR FIXED END BEAMS WITH FULL LATERAL RESTRAINT  
RECTANGULAR HOLLOW SECTIONS: GRADE C350  
bending about x-axis**

W<sub>L1</sub>' = Maximum Design Load based on Design Moment Capacity  
W<sub>L2</sub>' = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load W<sub>L</sub>' is LESSER of W<sub>L1</sub>' and W<sub>L2</sub>'

Designation d b t mm mm mm	Mass per m kg/m	Span of Beam (L) in metres										W <sub>L2</sub> ' kN	FLR m				
		0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5	3.0	3.5	4.0	4.5	5.0	6.0		
100 x 100 x 6.0 RHS	343	229	171	137	114	97.9	85.7	68.6	57.1	49.0	42.8	38.1	34.3	28.6	360	4.12	
5.0 RHS	301	200	150	120	100	85.9	75.1	60.1	50.1	42.9	37.6	33.4	30.1	25.0	324	4.23	
4.0 RHS	263	168	126	101	84.2	72.2	63.1	50.5	42.1	36.1	31.6	28.1	25.3	21.0	265	4.33	
3.5 RHS	226	151	113	90.5	75.4	64.7	56.6	45.3	37.7	32.3	26.3	25.1	22.6	18.9	234	4.38	
3.0 RHS	202	134	101	80.6	67.2	57.6	50.4	40.3	33.6	28.8	25.2	22.4	20.2	16.8	203	4.44	
2.5 RHS	171	114	85.7	68.6	57.1	49.0	42.8	34.3	28.6	24.5	21.4	19.0	17.1	14.3	171	4.49	
2.0 RHS	140	93.3	69.9	58.0	46.6	40.0	35.0	28.0	23.3	20.0	17.5	15.5	14.0	11.7	138	4.53	
75 x 50 x 6.0 RHS	9.67	212	141	106	84.9	70.7	60.6	53.0	42.4	35.4	30.3	26.5	23.6	21.2	17.7	277	5.28
5.0 RHS	8.35	188	125	94.1	75.3	62.7	53.8	47.1	37.6	31.4	26.9	23.5	20.9	18.8	15.7	238	5.43
4.0 RHS	6.92	160	107	79.9	63.9	53.3	45.7	40.0	32.0	26.6	22.8	20.0	17.8	16.0	13.3	196	5.57
3.0 RHS	5.42	129	86.1	64.6	51.7	43.1	36.9	32.3	25.8	21.5	18.5	16.1	14.4	12.9	10.8	151	5.73
2.5 RHS	4.58	110	73.5	55.2	44.1	36.8	31.5	27.6	22.1	18.4	15.8	13.8	12.3	11.0	9.19	128	5.79
2.0 RHS	3.72	90.4	60.3	45.2	36.2	30.1	25.8	22.6	18.1	15.1	12.9	11.3	10.0	9.04	7.53	104	5.86
75 x 25 x 2.5 RHS	3.60	76.1	50.7	38.0	30.4	25.4	21.7	19.0	15.2	12.7	10.9	9.51	8.45	7.61	6.34	123	1.47
2.0 RHS	2.93	62.8	41.9	31.4	25.1	20.9	17.9	15.7	12.6	10.5	8.97	7.85	6.98	6.28	5.23	99.9	1.50
1.6 RHS	2.38	51.5	34.3	25.8	20.6	17.2	14.7	12.9	10.3	8.59	7.36	6.44	5.72	5.15	4.29	80.8	1.53
65 x 35 x 3.0 RHS	4.25	83.1	55.4	41.5	33.2	27.7	23.7	20.8	16.6	13.8	11.9	10.4	9.23	8.31	6.92	128	3.23
2.5 RHS	3.60	71.5	47.6	35.7	28.6	23.8	20.4	17.9	14.3	11.9	10.2	8.93	7.94	7.15	5.96	108	3.28
2.0 RHS	2.93	59.0	39.3	29.5	23.6	19.7	16.8	14.7	11.8	9.83	8.42	7.37	6.55	5.90	4.91	88.2	3.33
50 x 25 x 3.0 RHS	3.07	44.3	29.5	22.2	17.7	14.8	12.7	11.1	8.86	7.39	6.33	5.54	4.92	4.43	3.69	95.0	2.08
2.5 RHS	2.62	38.6	25.7	19.3	15.4	12.9	11.0	9.65	7.72	6.43	5.51	4.82	4.29	3.86	3.22	81.0	2.12
2.0 RHS	2.15	32.2	21.5	16.1	12.9	10.7	9.21	8.06	6.44	5.37	4.60	4.03	3.58	3.22	2.69	66.2	2.17
1.6 RHS	1.75	26.7	17.8	13.3	10.7	8.86	7.61	6.66	5.33	4.44	3.81	3.33	2.96	2.67	2.22	53.9	2.21
50 x 20 x 3.0 RHS	2.83	39.0	26.0	19.5	15.6	13.0	11.1	9.75	7.80	6.50	5.57	4.87	4.33	3.90	3.25	93.8	1.31
2.5 RHS	2.42	34.1	22.7	17.1	13.6	11.4	9.74	8.53	6.82	5.68	4.87	4.26	3.79	3.41	2.84	80.0	1.35
2.0 RHS	1.99	28.6	19.1	14.3	11.4	9.53	8.17	7.15	5.72	4.77	4.08	3.57	3.18	2.86	2.38	65.4	1.39
1.6 RHS	1.63	23.7	15.8	11.9	9.49	7.91	6.78	5.93	4.74	3.95	3.39	2.97	2.64	2.37	1.98	53.2	1.42

Notes: 1. FLR - segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m \approx 0.8$

2.  $\phi = 0.9$

3.  $a_m = 1.0$

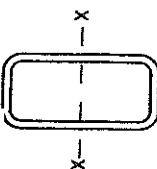
4.  $\alpha_e = 1.0$

5.  $W_L' = 12\phi M_e/L$

6.  $W_L'' = 2\phi V_v$

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**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR FIXED END BEAMS WITH FULL LATERAL RESTRAINT  
RECTANGULAR HOLLOW SECTIONS: GRADE C450  
bending about x-axis**

TABLE 8.3 - 4 (1)(A)

Designation	Mass per m	W <sub>1</sub> <sup>*</sup> (kN)						W <sub>2</sub>	FLR
		d	b	t	Span of Beam (L) in metres	W <sub>1</sub> <sup>*</sup> (kN)	W <sub>2</sub>		
125 x 75 x 3.8 RHS	11.1	280	140	93.4	70.0	56.0	46.7	40.0	35.0
3.3 RHS	9.73	248	124	82.5	61.9	49.5	41.3	35.4	31.1
2.8 RHS	8.39	203	102	67.8	50.8	40.7	33.9	27.5	24.8
2.3 RHS	6.95	147	73.5	49.0	36.8	29.4	24.5	22.6	20.6
100 x 50 x 3.3 RHS	7.14	138	69.2	46.2	34.6	27.7	23.1	19.8	16.3
2.8 RHS	6.19	122	61.0	40.6	30.5	24.4	20.3	17.4	15.2
2.3 RHS	5.14	102	51.1	34.1	25.5	20.4	17.0	14.6	12.8

Notes: Refer to 8.3-4 (2)(A)

TABLE 8.3 - 4 (2)(A)

Designation	Mass per m	W <sub>1</sub> <sup>*</sup> (kN)						W <sub>2</sub>	FLR
		d	b	t	Span of Beam (L) in metres	W <sub>1</sub> <sup>*</sup> (kN)	W <sub>2</sub>		
100 x 50 x 3.3 RHS	7.14	277	185	138	111	92.3	79.1	69.2	55.4
2.8 RHS	6.19	244	163	122	97.5	81.3	69.7	61.0	46.2
2.3 RHS	5.14	204	136	102	81.8	68.1	58.4	51.1	40.6
75 x 50 x 2.8 RHS	5.09	157	104	78.3	62.6	52.2	44.7	39.1	31.3
2.3 RHS	4.24	132	87.8	65.9	52.7	43.9	37.6	32.9	26.4
65 x 35 x 2.8 RHS	3.99	101	67.3	50.5	40.4	33.7	28.9	25.2	20.2
2.3 RHS	3.34	85.6	57.1	42.8	34.2	28.5	24.5	21.4	17.1
50 x 25 x 2.8 RHS	2.89	54.1	36.1	27.1	21.7	18.0	15.5	13.5	10.8
2.3 RHS	2.44	46.4	31.0	23.2	18.6	15.5	13.3	11.6	9.29
50 x 20 x 2.8 RHS	2.67	47.7	31.8	23.9	19.1	15.9	13.6	11.9	9.54
2.3 RHS	2.25	41.1	27.4	20.6	16.4	13.7	11.7	10.3	8.22

Notes:

1. FLR = segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -0.8$
2.  $\phi = 0.9$
3.  $c_m = 1.0$
4.  $c_3 = 1.0$
5.  $W_1^* = 12 \phi M_s / L$
6.  $W_2 = 2 \phi V_v$

1. FLR – segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -0.8$

2.  $\phi = 0.9$

3.  $c_m = 1.0$

4.  $c_3 = 1.0$

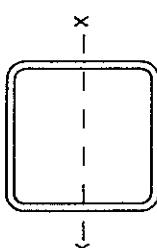
5.  $W_1^* = 12 \phi M_s / L$

6.  $W_2 = 2 \phi V_v$

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**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR **FIXED END BEAMS** WITH FULL LATERAL RESTRAINT  
SQUARE HOLLOW SECTIONS: GRADE C350  
bending about x-axis**



$W_{U1}$  = Maximum Design Load based on Design Moment Capacity  
 $W_{U2}$  = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load  $W_U$  is LESSER of  $W_{U1}$  and  $W_{U2}$

Designation d b t	Mass per m	Span of Beam (L) in metres								$W_{U2}$	FLR
		1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0		
250 x 250 x 9.0 SHS	65.9	2810	1410	938	703	563	469	402	352	313	281
6.0 SHS	45.0	1540	772	515	386	309	257	221	193	172	154
200 x 200 x 9.0 SHS	51.8	1760	880	586	440	352	293	251	220	195	176
6.0 SHS	35.6	1110	555	370	278	222	185	159	139	123	111
5.0 SHS	29.9	841	420	280	210	168	140	120	105	93.4	84.1
150 x 150 x 9.0 SHS	37.7	938	469	313	235	188	156	134	117	104	93.8
6.0 SHS	26.2	671	336	224	168	134	112	95.9	83.9	74.6	67.1
5.0 SHS	22.1	546	273	182	137	109	91.0	78.0	68.3	60.7	54.6
125 x 125 x 9.0 SHS	30.6	623	312	208	156	125	104	89.1	77.9	69.3	62.3
6.0 SHS	21.4	453	227	151	113	90.7	75.6	64.8	56.7	50.4	45.3
5.0 SHS	18.2	388	194	129	97.1	77.7	64.7	55.5	48.5	43.2	38.8
4.0 SHS	14.8	298	149	99.4	74.5	59.6	49.7	42.6	37.3	33.1	29.8
100 x 100 x 9.0 SHS	23.5	373	186	124	93.4	74.5	62.1	53.2	46.6	41.4	37.3
6.0 SHS	16.7	278	139	92.7	69.5	55.6	46.3	39.7	34.7	30.9	27.8
5.0 SHS	14.2	240	120	80.0	60.0	48.0	40.0	34.3	30.0	26.7	24.0
4.0 SHS	11.6	199	99.4	66.3	49.7	39.8	33.1	28.4	24.9	22.1	19.9
3.0 SHS	8.96	140	70.1	46.8	35.1	28.1	23.4	20.0	17.5	15.6	14.0
89 x 89 x 6.0 SHS	14.6	214	107	71.3	53.4	42.8	35.6	30.5	26.7	23.8	21.4
5.0 SHS	12.5	186	92.8	61.9	46.4	37.1	30.9	26.5	23.2	20.6	18.6
3.5 SHS	9.06	138	68.9	45.9	34.5	27.6	23.0	19.7	17.2	15.3	13.8
75 x 75 x 6.0 SHS	12.0	145	72.6	48.4	36.3	29.0	24.2	20.7	18.1	16.1	14.5
5.0 SHS	10.3	127	63.6	42.4	31.8	25.4	21.2	18.2	15.9	14.1	12.7
4.0 SHS	8.49	107	53.4	35.6	26.7	21.4	17.8	15.3	13.3	11.9	10.7
3.5 SHS	7.53	95.6	47.8	31.9	23.9	19.1	15.9	13.7	12.0	10.6	9.56
3.0 SHS	6.60	85.0	42.5	28.3	21.2	17.0	14.2	12.1	10.6	9.44	8.50
2.5 SHS	5.56	69.0	34.5	23.0	17.3	13.8	11.5	9.86	8.63	7.67	6.90

Notes:  
1. FLR = segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -0.8$

2.  $\phi = 0.9$
3.  $a_m = 1.0$
4.  $a_s = 1.0$
5.  $W_{U1} = 12\phi M_s / L$
6.  $W_{U2} = 2\phi V_r$

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## FOR FIXED END BEAMS WITH FULL LATERAL RESTRAINT

### SQUARE HOLLOW SECTIONS: GRADE C350

bending about x-axis

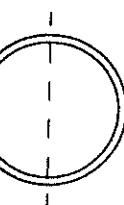
$W_{L1}$  = Maximum Design Load based on Design Moment Capacity  
 $W_{L2}$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L$  is LESSER of  $W_{L1}$  and  $W_{L2}$

Designation	Mass per m	Span of Beam (l) in metres										$W_{L1}$ (kN)	FLR					
		d mm	b mm	t mm	0.5	0.75	1.0	1.25	1.5	1.75	2.0			4.5	5.0	6.0		
75 x 75 x 6.0 SHS	12.0	290	194	145	116	96.8	83.0	72.6	58.1	46.4	41.5	36.3	32.3	29.0	24.2	282	11.8	
5.0 SHS	10.3	254	170	127	102	84.8	72.7	63.6	50.9	42.4	36.3	31.8	28.3	25.4	21.2	243	12.0	
4.0 SHS	8.49	214	142	107	85.4	71.2	61.0	53.4	42.7	35.6	30.6	26.7	23.7	21.4	17.8	200	12.2	
3.5 SHS	7.53	191	126	95.6	76.5	63.8	54.7	47.8	36.3	31.9	27.3	23.9	21.3	19.1	15.9	178	12.4	
3.0 SHS	6.60	170	113	85.0	68.0	56.7	48.6	42.5	34.0	28.3	24.3	21.2	18.9	17.0	14.2	155	12.5	
2.5 SHS	5.56	138	92.0	69.0	55.2	46.0	39.4	34.5	27.6	23.0	19.7	17.3	15.3	13.8	11.5	131	12.6	
65 x 65 x 6.0 SHS	10.1	208	139	104	83.2	69.3	59.4	52.0	41.6	34.7	28.7	26.0	23.1	20.8	17.3	237	10.0	
5.0 SHS	8.75	184	123	91.9	73.6	61.3	52.5	46.0	36.8	30.6	26.3	23.0	20.4	18.4	15.3	205	10.3	
4.0 SHS	7.23	156	104	77.9	62.3	51.9	44.5	38.9	31.1	26.0	22.2	19.5	17.3	15.6	13.0	176	10.5	
3.0 SHS	5.66	125	82.5	62.6	50.1	41.8	35.8	31.3	25.1	20.9	17.9	15.7	13.9	12.5	10.4	132	10.8	
2.5 SHS	4.78	107	71.3	53.5	42.8	35.6	30.5	26.7	21.4	17.8	15.3	13.4	11.9	10.7	8.91	112	10.9	
2.0 SHS	3.88	80.0	53.3	40.0	32.0	26.7	22.9	20.0	16.0	13.3	11.4	10.0	8.89	8.00	6.67	91.1	11.0	
50 x 50 x 5.0 SHS	6.39	99.5	66.3	49.7	39.8	33.2	28.4	24.9	19.9	16.6	14.2	12.4	11.1	9.95	8.29	149	7.61	
4.0 SHS	5.35	86.0	57.4	43.0	34.4	28.7	24.6	21.5	17.2	14.3	12.3	10.8	9.56	8.60	7.17	125	7.85	
3.0 SHS	4.25	71.0	47.3	35.5	28.4	23.7	20.3	17.7	14.2	11.8	10.1	8.87	7.89	7.10	5.91	98.6	8.13	
2.5 SHS	3.60	61.0	40.7	30.5	24.4	20.3	17.4	15.3	12.2	10.2	8.72	7.63	6.78	6.10	5.09	84.0	8.24	
2.0 SHS	2.93	50.4	33.6	25.2	20.1	16.8	14.4	12.6	10.1	8.39	7.20	6.30	5.60	5.04	4.20	68.7	8.34	
1.6 SHS	2.38	38.6	25.7	19.3	15.4	12.9	11.0	9.64	7.71	6.43	5.51	4.82	4.29	3.86	3.21	55.9	8.42	
40 x 40 x 4.0 SHS	4.09	50.9	32.9	25.5	20.4	17.0	14.5	12.7	10.2	8.49	7.27	6.37	5.66	5.09	4.24	95.6	6.09	
3.0 SHS	3.30	43.3	28.8	21.6	17.3	14.4	12.4	10.8	8.65	7.21	6.18	5.41	4.81	4.33	3.61	76.2	6.38	
2.5 SHS	2.82	37.5	25.0	18.8	15.0	12.5	10.7	9.39	7.51	6.26	5.35	4.69	4.17	3.75	3.13	65.3	6.48	
2.0 SHS	2.31	31.3	20.8	15.6	12.5	10.4	8.93	7.81	6.25	4.25	4.6	3.91	3.47	3.13	2.60	53.8	6.59	
1.6 SHS	1.88	25.8	17.2	12.9	10.3	8.60	7.37	6.45	5.16	4.30	3.68	3.22	2.87	2.58	2.15	44.0	6.67	
35 x 35 x 3.0 SHS	2.83	32.0	21.3	16.0	12.8	10.7	9.13	7.99	6.39	5.33	4.57	4.00	3.55	3.20	2.66	65.0	5.50	
2.5 SHS	2.42	27.9	18.6	14.0	11.2	9.31	7.98	6.98	5.59	4.66	3.98	3.49	3.10	2.79	2.33	56.0	5.61	
2.0 SHS	1.99	23.4	15.6	11.7	9.36	7.80	6.68	5.85	4.76	4.17	3.33	2.92	2.60	2.34	1.95	46.3	5.71	
1.6 SHS	1.63	19.4	12.9	9.70	7.76	6.47	5.54	4.85	3.98	3.48	2.78	2.92	2.42	2.16	1.94	1.62	38.0	5.80
30 x 30 x 3.0 SHS	2.36	22.4	14.9	11.2	8.95	7.46	6.39	5.59	4.47	3.73	3.20	2.80	2.49	2.24	1.86	53.8	4.83	
2.5 SHS	2.03	19.7	13.2	9.87	7.89	6.58	5.64	4.93	3.95	3.29	2.92	2.47	2.19	1.97	1.64	46.7	4.73	
2.0 SHS	1.68	16.7	11.1	8.34	6.67	5.56	4.76	4.17	3.33	2.78	2.38	2.08	1.85	1.67	1.39	38.3	4.84	
1.6 SHS	1.38	13.9	9.28	6.96	5.57	4.64	3.98	3.48	2.78	2.34	1.99	1.74	1.55	1.39	1.16	32.0	4.92	
25 x 25 x 3.0 SHS	1.89	14.5	9.65	7.24	5.79	4.83	4.14	3.62	2.90	2.41	2.07	1.81	1.61	1.45	1.21	42.6	3.75	
2.5 SHS	1.64	12.9	8.63	6.47	5.18	4.32	3.70	3.24	2.58	2.16	1.95	1.62	1.44	1.29	1.08	37.3	3.85	
2.0 SHS	1.36	11.1	7.39	5.54	4.43	3.69	3.17	2.77	2.22	1.85	1.58	1.39	1.23	1.11	0.924	31.4	3.96	
1.6 SHS	1.12	9.34	6.23	4.67	3.74	3.11	2.67	2.34	1.87	1.56	1.33	1.17	1.04	0.934	0.779	26.0	4.04	
20 x 20 x 1.6 SHS	0.873	5.67	3.78	2.84	2.27	1.89	1.62	1.42	1.13	0.946	0.811	0.709	0.631	0.567	0.473	20.1	3.17	

Notes: Refer to TABLE 8.3-5 (1)(A)

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**FOR CANTILEVER BEAMS**

**CIRCULAR HOLLOW SECTIONS: GRADE C250**

bending about any axis

TABLE 8.4 - 1 (1)(A)

$W_t^*$  = Maximum Design Load based on Design Moment Capacity  
 $W_2$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_t$  is LESSER of  $W_t^*$  and  $W_2$

Designation $d_o$ mm	Mass per m kg/m	Span of Beam (L) in metres								$W_t^*$ (kN)
		1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	
610.0 x 12.7 CHS	187	2040	1020	680	510	408	340	291	255	227
9.5 CHS	141	1470	735	490	368	294	245	210	184	147
6.4 CHS	95.3	898	449	299	224	180	150	128	112	99.8
508.0 x 12.7 CHS	155	1400	701	467	361	280	234	200	175	156
9.5 CHS	117	1050	525	350	263	210	175	150	131	117
6.4 CHS	79.2	656	328	219	164	131	109	93.8	82.1	72.9

Notes: Refer to TABLE 8.4-1 (3)(A)

TABLE 8.4 - 1 (2)(A)

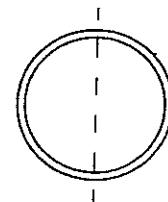
Designation $d_o$ mm	Mass per m kg/m	Span of Beam (L) in metres								$W_t^*$ (kN)
		0.5	1.0	1.5	2.0	2.5	3.0	3.5	4.0	
165.1 x 5.4 CHS	21.3	124	62.0	41.3	31.0	24.8	20.7	17.7	15.5	12.4
5.0 CHS	19.7	115	57.7	38.5	28.8	23.1	19.2	16.5	14.4	11.5
139.7 x 5.4 CHS	17.9	87.7	43.9	29.2	21.9	17.5	14.6	12.5	11.0	8.77
5.0 CHS	16.6	81.7	40.8	27.2	20.4	16.3	13.6	11.7	10.2	8.17
114.3 x 5.4 CHS	14.5	57.7	28.8	19.2	14.4	11.5	9.61	8.24	7.21	6.26
4.5 CHS	12.2	48.9	24.4	16.3	12.2	9.77	8.14	6.98	6.11	5.77
101.6 x 5.0 CHS	11.9	42.0	21.0	14.0	10.5	8.41	7.00	6.00	5.25	4.20
4.0 CHS	9.63	34.3	17.2	11.4	8.58	6.86	5.72	4.90	4.29	3.43
88.9 x 5.9 CHS	12.1	36.6	18.3	12.2	9.16	7.33	6.11	5.23	4.58	3.66
5.0 CHS	10.3	31.7	15.9	10.6	7.93	6.34	5.29	4.53	3.96	3.17
4.0 CHS	8.38	26.0	13.0	8.66	6.49	5.19	4.33	3.71	3.25	2.60
76.1 x 5.9 CHS	10.2	26.2	13.1	8.74	6.56	5.25	4.37	3.75	3.28	2.62
4.5 CHS	7.95	20.8	10.4	6.93	5.20	4.16	3.46	2.97	2.60	2.08
3.6 CHS	6.44	17.0	8.52	5.68	4.26	3.41	2.84	2.43	2.13	1.70

Notes: Refer to TABLE 8.4-1 (3)(A)

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TABLE 8.4 - 2 (2)(A)



**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR CANTILEVER BEAMS WITH FULL LATERAL RESTRAINT  
CIRCULAR HOLLOW SECTIONS: GRADE C350  
bending about any axis**

$W_1^*$  = Maximum Design Load based on Design Moment Capacity

$W_2^*$  = Maximum Design Load based on Design Shear Capacity

Maximum Design Load  $W_1^*$  is LESSER of  $W_1^*$  and  $W_2^*$

Designation. $d_o$ mm	Mass per m kg/m	$W_1^*/(kN)$								$W_2^*$ kN					
		0.5	1.0	1.5	2.0	2.5	3.0	3.5	4.0						
168.3 x 7.1 CHS 6.4 CHS 4.8 CHS	28.2 25.6 19.4	233 211 162	116 106 80.9	77.5 70.5 53.9	58.2 52.9 40.4	46.5 42.3 32.3	36.8 35.2 27.0	33.2 30.2 23.1	29.1 26.4 20.2	23.3 21.1 16.2	19.4 17.6 13.5	16.6 15.1 11.6	14.5 13.2 10.1	12.9 11.7 8.98	408 369 280
165.1 x 3.5 CHS 3.0 CHS	13.9 12.0	109 90.6	54.5 45.3	36.4 30.2	27.3 22.6	21.8 18.1	18.2 15.1	15.6 12.9	13.6 11.3	10.9 9.06	9.09 7.55	7.79 6.47	6.82 5.66	6.05 5.03	201 173
139.7 x 3.5 CHS 3.0 CHS	11.8 10.1	80.2 67.1	40.1 33.6	26.7 22.4	20.1 16.8	16.0 13.4	13.4 11.2	11.5 9.59	10.0 8.39	8.02 6.71	6.68 5.59	5.73 4.79	5.01 4.19	4.46 3.73	170 146
114.3 x 6.0 CHS 4.8 CHS 3.6 CHS 3.2 CHS	16.0 13.0 9.83 8.77	88.8 72.6 55.6 49.8	44.4 36.3 27.8 24.9	29.6 24.2 18.5 16.6	22.2 18.1 13.9 12.4	17.8 14.5 11.1 9.96	14.8 10.4 9.27 8.30	12.7 9.07 7.94 7.11	11.1 7.26 6.95 6.22	8.88 7.40 6.05 4.98	6.34 5.55 5.18 4.63	5.73 5.59 4.79 4.15	5.01 4.46 4.03 3.48	4.46 3.73 3.09 3.11	201 173 146 127
101.6 x 3.2 CHS 2.6 CHS	7.77 6.35	39.1 31.6	19.5 15.8	13.0 10.5	9.76 7.90	7.81 6.32	6.51 5.27	5.58 4.52	4.88 3.95	3.91 3.16	3.25 2.63	2.79 2.26	2.44 1.98	2.17 1.76	112 91.7
88.9 x 5.5 CHS 4.8 CHS 3.2 CHS 2.6 CHS	11.3 9.96 6.76 5.53	48.3 42.8 29.6 24.4	24.1 21.4 14.8 12.2	16.1 14.3 9.88 8.14	12.1 10.7 7.41 6.10	9.65 8.56 5.93 4.88	8.05 7.14 4.94 4.07	6.90 6.12 4.23 3.49	4.83 4.28 3.70 3.05	4.02 3.57 2.96 2.44	3.45 3.06 2.47 2.03	3.02 2.68 2.38 1.74	2.68 2.38 1.85 1.53	2.68 2.38 1.65 1.36	163 144 97.7 79.9
76.1 x 3.2 CHS 2.3 CHS	5.75 4.19	21.4 15.8	10.7 7.89	7.15 5.26	5.36 3.95	4.29 3.16	3.57 2.63	3.06 2.26	2.68 1.97	2.14 1.58	1.79 1.32	1.53 1.13	1.34 0.987	1.19 0.877	83.1 60.5

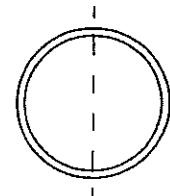
Notes:

1.  $\phi = 0.9$
2.  $\alpha_m = 1.0$
3.  $\alpha_s = 1.0$
4.  $W_1^* = 2\phi M_s / L$
5.  $W_2^* = \phi V_v$

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TABLE 8.4 – 2 (3)(A)



**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR CANTILEVER BEAMS WITH FULL LATERAL RESTRAINT  
CIRCULAR HOLLOW SECTIONS: GRADE C350  
bending about any axis**

W<sub>L1</sub>\* = Maximum Design Load based on Design Moment Capacity  
W<sub>L2</sub>\* = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load W<sub>L1</sub>\* is LESSER of W<sub>L1</sub>\* and W<sub>L2</sub>\*

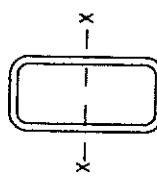
Designation <i>d<sub>0</sub></i> mm	Mass per m kg/m	Span of Beam (L) in metres							W <sub>L2</sub> * kN
		0.25	0.5	0.75	1.0	1.25	1.5	1.75	
76.1x3.2 CHS	5.75	42.9	21.4	14.3	10.7	8.58	7.15	6.13	5.36
2.3 CHS 4.19	31.6	15.8	10.5	7.89	6.32	5.26	4.51	3.95	3.57
60.3x2.9 CHS 4.11	24.1	12.0	8.03	6.02	4.82	4.02	3.44	3.01	2.41
2.3 CHS 3.29	19.5	9.75	6.50	4.88	3.90	3.25	2.79	2.44	1.95
48.3x2.9 CHS 3.25	15.1	7.54	5.03	3.77	3.02	2.51	2.15	1.89	1.51
2.3 CHS 2.61	12.3	6.14	4.09	3.07	2.45	2.05	1.75	1.53	1.23
42.4x2.6 CHS 2.55	10.4	5.20	3.46	2.60	2.08	1.73	1.48	1.20	1.04
2.0 CHS 1.99	8.23	4.12	2.74	2.06	1.65	1.37	1.18	1.03	0.823
33.7x2.6 CHS 1.99	6.35	3.18	2.12	1.59	1.27	1.06	0.907	0.794	0.635
2.0 CHS 1.56	5.07	2.54	1.69	1.27	1.01	0.845	0.724	0.634	0.507
26.9x2.3 CHS 1.40	3.52	1.76	1.17	0.879	0.704	0.586	0.503	0.440	0.352
2.0 CHS 1.23	3.13	1.57	1.04	0.783	0.626	0.522	0.447	0.391	0.313
21.3x2.0 CHS 0.952	1.88	0.942	0.628	0.471	0.377	0.314	0.269	0.236	0.188

- Notes:  
 1.  $\phi = 0.9$   
 2.  $\alpha_m = 1.0$   
 3.  $\alpha_s = 1.0$   
 4.  $W_{L1}^* = 2\phi M_s / L$   
 5.  $W_{L2}^* = \phi V_v$

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TABLE 8.4 - 3 (1)(A)



**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR CANTILEVER BEAMS WITH FULL LATERAL RESTRAINT  
RECTANGULAR HOLLOW SECTIONS: GRADE C350  
bending about X-axis**

W<sub>1</sub>' = Maximum Design Load based on Design Moment Capacity  
W<sub>2</sub>' = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load W<sub>1</sub>' is LESSER of W<sub>1</sub>' and W<sub>2</sub>'

Designation	Mass per m	Span of Beam (L) in metres						W <sub>1</sub> ' (kN)	W <sub>2</sub> '	FLR
		1.0	2.0	3.0	4.0	5.0	6.0			
250 x 150 x 9.0 RHS	51.8	336	168	112	84.0	67.2	56.0	48.0	37.3	33.6
6.0 RHS	35.6	236	118	78.6	58.9	47.1	39.3	33.7	26.2	21.4
5.0 RHS	29.9	189	94.4	82.9	47.2	37.8	31.5	27.0	23.6	18.9
200 x 100 x 9.0 RHS	37.7	185	92.4	61.6	46.2	37.0	30.8	26.4	23.1	20.5
6.0 RHS	26.2	132	66.2	44.1	33.1	26.5	22.1	18.9	16.6	14.7
5.0 RHS	22.1	113	56.5	37.6	28.2	22.6	18.8	16.1	14.1	12.5
4.0 RHS	17.9	92.4	46.2	30.8	23.1	18.6	15.4	13.2	11.5	10.3
150 x 100 x 6.0 RHS	21.4	84.6	42.3	28.2	21.2	16.9	14.1	12.1	10.6	9.40
5.0 RHS	18.2	72.5	36.3	24.2	18.1	14.5	12.1	10.4	9.07	8.06
4.0 RHS	14.8	59.6	29.8	19.9	14.9	11.9	9.94	8.52	7.45	6.62
150 x 50 x 5.0 RHS	14.2	49.7	24.8	16.6	12.4	9.94	8.28	7.10	6.21	5.52
4.0 RHS	11.6	41.2	20.6	13.7	10.3	8.24	6.87	5.89	5.15	4.58
3.0 RHS	8.96	32.4	16.2	10.8	8.10	6.48	5.40	4.63	4.05	3.60
125 x 75 x 6.0 RHS	16.7	53.1	26.5	17.7	13.3	10.6	8.84	7.58	6.63	5.90
5.0 RHS	14.2	45.8	22.9	15.3	11.5	9.17	7.64	6.55	5.73	5.09
4.0 RHS	11.6	38.0	19.0	12.7	9.49	7.59	6.33	5.42	4.75	4.22
3.0 RHS	8.96	29.8	14.9	9.92	7.44	5.95	4.96	4.25	3.72	3.31
100 x 50 x 6.0 RHS	12.0	28.6	14.3	9.52	7.14	5.71	4.76	4.08	3.57	3.17
5.0 RHS	10.3	25.0	12.5	8.35	6.26	5.01	4.17	3.58	3.13	2.78
4.0 RHS	8.49	21.0	10.5	7.02	5.26	4.21	3.51	3.01	2.63	2.34
3.5 RHS	7.53	18.9	9.43	6.29	4.72	3.77	3.14	2.69	2.36	2.10
3.0 RHS	6.60	16.8	8.40	5.60	4.20	3.36	2.80	2.40	2.10	1.87
2.5 RHS	5.56	14.3	7.14	4.76	3.57	2.86	2.38	2.04	1.79	1.59
2.0 RHS	4.50	11.7	5.83	3.89	2.91	2.33	1.94	1.67	1.46	1.30

Notes:  
1. FLR = segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -0.8$

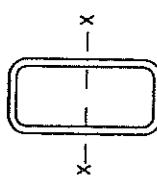
2.  $\phi = 0.9$
3.  $\alpha_m = 1.0$
4.  $\alpha_s = 1.0$
5.  $W_{1}' = 2 \phi M_s / L$
6.  $W_{2}' = \phi V_r$

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TABLE 8.4 - 3 (2)(A)

**STRENGTH LIMIT STATE MAXIMUM DESIGN LOADS  
FOR CANTILEVER BEAMS WITH FULL LATERAL RESTRAINT  
RECTANGULAR HOLLOW SECTIONS: GRADE C350  
bending about x-axis**



W<sub>L1</sub> = Maximum Design Load based on Design Moment Capacity  
W<sub>L2</sub> = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load W<sub>L</sub> is LESSER of W<sub>L1</sub> and W<sub>L2</sub>

Designation	Mass per m	W <sub>L1</sub> (kN)										Span of Beam (L) in metres										W <sub>L2</sub>	FLR
d mm	b mm	t mm	0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5	3.0	3.5	4.0	4.5	5.0	6.0	kN	m					
100 x 50 x 6.0 RHS	12.0	57.1	38.1	29.6	22.9	19.0	16.3	14.3	11.4	9.52	8.16	7.14	6.35	5.71	4.76	190	4.12						
5.0 RHS	10.3	50.1	33.4	25.0	20.0	16.7	14.3	12.5	10.0	8.35	7.16	6.26	5.57	5.01	4.17	162	4.23						
4.0 RHS	8.49	42.1	28.1	21.0	16.8	14.0	12.0	10.5	8.42	7.02	6.01	5.26	4.68	4.21	3.51	132	4.33						
3.5 RHS	7.53	37.7	25.1	18.9	15.1	12.6	10.8	9.43	7.54	6.29	5.39	4.72	4.19	3.77	3.14	117	4.38						
3.0 RHS	6.60	33.6	22.4	16.8	13.4	11.2	9.60	8.40	6.72	5.60	4.80	4.20	3.73	3.36	2.80	102	4.44						
2.5 RHS	5.56	28.6	19.0	14.3	11.4	9.52	8.16	7.14	5.71	4.76	4.08	3.57	3.17	2.86	2.38	85.5	4.49						
2.0 RHS	4.50	23.3	15.5	11.7	9.33	7.77	6.66	5.83	4.66	3.89	3.33	2.91	2.59	2.33	1.94	68.1	4.53						
75 x 50 x 6.0 RHS	9.67	35.4	23.6	17.7	14.1	11.8	10.1	8.84	7.07	5.89	5.05	4.42	3.93	3.54	2.95	138	5.28						
5.0 RHS	8.35	31.4	20.9	15.7	12.5	10.5	8.96	7.84	6.27	5.23	4.48	3.92	3.49	3.14	2.61	119	5.43						
4.0 RHS	6.92	26.6	17.8	13.3	10.7	8.8	7.61	6.66	5.33	4.44	3.81	3.33	2.96	2.66	2.22	98.0	5.57						
3.0 RHS	5.42	21.5	14.4	10.8	8.61	7.18	6.15	5.38	4.31	3.59	3.08	2.69	2.39	2.15	1.79	75.7	5.73						
2.5 RHS	4.58	18.4	12.3	9.19	7.35	6.13	5.25	4.60	3.68	3.06	2.63	2.30	2.04	1.84	1.53	64.0	5.79						
2.0 RHS	3.72	15.1	10.0	7.53	6.03	5.02	4.31	3.77	3.01	2.51	2.15	1.88	1.67	1.51	1.26	51.9	5.86						
75 x 25 x 25 RHS	3.60	12.7	8.45	6.34	5.07	4.23	3.62	3.17	2.54	2.11	1.81	1.58	1.41	1.27	1.06	61.5	1.47						
2.0 RHS	2.93	10.5	6.98	5.23	4.19	3.49	2.99	2.62	2.09	1.74	1.50	1.31	1.16	1.05	0.872	49.9	1.50						
1.6 RHS	2.38	8.59	5.72	4.29	3.43	2.86	2.45	2.15	1.72	1.43	1.23	1.07	0.944	0.859	0.716	40.4	1.53						
65 x 35 x 30 RHS	4.25	13.8	9.23	6.92	5.54	4.62	3.96	3.46	2.77	2.31	1.98	1.73	1.54	1.38	1.15	64.0	1.41						
2.5 RHS	3.60	11.9	7.94	5.96	4.76	3.97	3.40	2.98	2.38	1.99	1.70	1.49	1.32	1.19	0.993	54.2	1.47						
2.0 RHS	2.93	9.83	6.55	4.91	3.93	3.28	2.81	2.46	1.97	1.64	1.40	1.23	1.09	0.93	0.819	44.1	1.50						
50 x 25 x 30 RHS	3.07	7.39	4.92	3.69	2.95	2.46	2.11	1.85	1.48	1.23	1.06	0.923	0.821	0.739	0.616	47.5	2.08						
2.5 RHS	2.62	4.43	4.29	3.22	2.57	2.14	1.84	1.61	1.29	1.07	0.919	0.804	0.715	0.643	0.536	40.5	2.12						
2.0 RHS	2.15	5.37	3.58	2.69	2.15	1.79	1.53	1.34	1.07	0.895	0.767	0.671	0.597	0.537	0.448	33.1	2.17						
1.6 RHS	1.75	4.44	2.96	2.22	1.78	1.48	1.27	1.11	0.888	0.740	0.635	0.555	0.494	0.444	0.370	27.0	2.21						
50 x 20 x 30 RHS	2.83	6.50	4.33	3.25	2.60	2.17	1.86	1.62	1.30	1.08	0.928	0.812	0.722	0.650	0.542	46.9	1.31						
2.5 RHS	2.42	5.68	3.79	2.84	2.27	1.89	1.62	1.42	1.14	0.947	0.812	0.711	0.632	0.568	0.474	40.0	1.35						
2.0 RHS	1.99	4.77	3.18	2.38	1.91	1.59	1.36	1.19	0.933	0.794	0.681	0.596	0.500	0.477	0.397	32.7	1.39						
1.6 RHS	1.63	3.95	2.64	1.98	1.58	1.32	1.13	0.989	0.791	0.659	0.565	0.494	0.439	0.395	0.330	26.6	1.42						

Notes: 1. FLR = segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -0.8$

2.  $\phi = 0.9$

3.  $\alpha_m = 1.0$

4.  $\alpha_s = 1.0$

5.  $W_{L1} = 2 \phi M_s / L$

6.  $W_{L2} = \phi V_v$

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# RECTANGULAR HOLLOW SECTIONS: GRADE C450

**bending about x-axis**

**TABLE 8.4 – 4 (1)(A)**

Designation	Mass per m	d b t	W <sub>1</sub> (kN)						W <sub>2</sub>	FLR								
			1.0	2.0	3.0	4.0	5.0	6.0	Span of Beam (L) in metres									
125 x 75 x 3.8 RHS	11.1	46.7	23.3	15.6	11.7	9.34	7.78	6.67	5.84	5.19	4.67	4.25	3.89	3.59	3.34	209	6.14	
3.3 RHS	9.72	41.3	20.6	13.8	10.3	8.25	6.88	5.89	5.16	4.58	4.13	3.75	3.44	3.17	2.95	2.42	183	6.19
2.8 RHS	8.39	33.9	16.9	11.3	8.47	6.78	5.65	4.84	4.24	3.76	3.39	3.08	2.82	2.61	2.42	156	6.24	
2.3 RHS	6.95	24.3	12.3	8.17	6.13	4.90	4.08	3.50	3.06	2.72	2.45	2.23	2.04	1.88	1.75	129	6.29	
100 x 50 x 3.3 RHS	7.14	23.1	11.5	7.69	5.77	4.62	3.85	3.30	2.88	2.56	2.31	2.10	1.92	1.78	1.65	143	3.42	
2.8 RHS	6.19	20.3	10.2	6.77	5.08	4.06	3.39	2.90	2.54	2.26	2.03	1.85	1.69	1.56	1.45	122	3.47	
2.3 RHS	5.14	17.0	8.52	5.68	4.26	3.41	2.84	2.43	2.13	1.89	1.70	1.55	1.42	1.31	1.22	102	3.50	

Notes: Refer to TABLE 8.4-4 (2)(A)

**TABLE 8.4 – 4 (2)(A)**

Designation	Mass per m	d b t	W <sub>1</sub> (kN)						W <sub>2</sub>	FLR						
			0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5	3.0	3.5	4.0	4.5	5.0	6.0
100 x 50 x 3.3 RHS	7.14	46.2	30.8	23.1	18.5	15.4	13.2	11.5	9.23	7.69	6.59	5.77	5.13	4.62	3.85	3.42
2.8 RHS	6.19	40.6	27.1	20.3	16.3	13.5	11.6	10.2	8.13	6.77	5.81	5.06	4.52	4.06	3.39	143
2.3 RHS	5.14	34.1	22.7	17.0	13.6	11.4	9.73	8.52	6.81	5.68	4.87	4.26	3.78	3.41	2.84	102
75 x 50 x 2.8 RHS	5.09	26.1	17.4	13.0	10.4	8.70	7.45	6.52	5.22	4.35	3.73	3.26	2.90	2.61	2.17	91.4
2.3 RHS	4.24	22.0	14.6	11.0	8.78	7.32	6.27	5.49	4.39	3.66	3.14	2.74	2.44	2.20	1.88	76.2
65 x 35 x 2.8 RHS	3.99	16.8	11.2	8.42	6.73	5.61	4.81	4.21	3.37	2.81	2.40	2.10	1.87	1.68	1.40	77.3
2.3 RHS	3.34	14.3	9.51	7.13	5.71	4.76	4.08	3.57	2.85	2.38	2.04	1.78	1.59	1.43	1.19	64.6
50 x 25 x 2.8 RHS	2.89	9.02	6.02	4.51	3.61	3.01	2.58	2.26	1.80	1.50	1.29	1.13	1.00	0.902	0.752	57.5
2.3 RHS	2.44	7.74	5.16	3.87	3.10	2.58	2.21	1.94	1.55	1.29	1.11	0.968	0.860	0.774	0.645	48.3
50 x 20 x 2.8 RHS	2.67	7.95	5.30	3.98	3.18	2.65	2.27	1.99	1.59	1.33	1.14	0.984	0.884	0.795	0.663	56.8
2.3 RHS	2.25	6.85	4.57	3.43	2.74	2.28	1.96	1.71	1.37	1.14	0.979	0.857	0.761	0.685	0.571	47.7

Notes:

- 1. FLR = segment length for Full Lateral Restraint (Clause 5.3.2.4 of AS 4100) with  $\beta_m = -0.8$
- 2.  $\phi = 0.9$
- 3.  $\alpha_m = 1.0$
- 4.  $\alpha_s = 1.0$
- 5.  $W_{l1} = 2 \phi M_s / L$
- 6.  $W_{l2} = \phi V_v$

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TABLE 8.4 – 5 (2)(A)

**FOR CANTILEVER BEAMS** WITH FULL LATERAL RESTRAINT

**SQUARE HOLLOW SECTIONS: GRADE C350**  
bending about x-axis

W<sub>L1</sub> = Maximum Design Load based on Design Moment Capacity  
W<sub>L2</sub> = Maximum Design Load based on Design Shear Capacity  
Maximum Design Load W<sub>L</sub> is LESSER of W<sub>L1</sub> and W<sub>L2</sub>

Designation	d mm	b mm	Mass per m kg/m.	W <sub>L1</sub> (kN)								W <sub>L2</sub> kN	FLR m					
				0.5	0.75	1.0	1.25	1.50	1.75	2.0	2.5	3.0	3.5	4.0				
75 x 75 x 6.0 SHS	12.0	48.4	32.3	24.2	19.4	16.1	13.8	12.1	9.68	8.07	6.91	6.05	5.38	4.84	4.03	141	11.8	
5.0 SHS	10.3	42.4	28.3	21.2	17.0	14.1	12.1	10.6	8.48	7.07	6.06	5.30	4.71	4.24	3.53	121	12.0	
4.0 SHS	8.49	35.6	23.7	17.8	14.2	11.9	10.2	8.90	7.12	5.93	5.08	4.45	3.95	3.56	2.97	100	12.2	
3.5 SHS	7.53	31.9	21.3	15.9	12.8	10.6	9.11	7.97	6.38	5.31	4.55	3.98	3.54	3.19	2.66	88.9	12.4	
3.0 SHS	6.60	28.3	18.9	14.2	11.3	9.44	8.09	7.08	5.67	4.72	4.05	3.54	3.15	2.83	2.36	77.3	12.5	
2.5 SHS	5.56	23.0	15.3	11.5	9.20	7.67	6.57	5.75	4.60	3.83	3.29	2.88	2.56	2.30	1.92	65.3	12.6	
65 x 65 x 6.0 SHS	10.1	34.7	23.1	17.3	13.9	11.6	9.90	8.67	6.93	5.78	4.95	4.33	3.65	3.47	2.89	119	10.0	
5.0 SHS	8.75	30.6	20.4	15.3	12.3	10.2	8.76	7.66	6.13	5.11	4.38	3.41	3.06	2.55	2.08	109	10.3	
4.0 SHS	7.23	26.0	17.3	13.0	10.4	8.65	7.42	6.49	5.19	4.33	3.71	3.24	2.88	2.60	2.16	85.1	10.5	
3.0 SHS	5.66	20.9	13.9	10.4	8.35	6.96	5.97	5.22	4.18	3.48	2.98	2.32	2.09	1.74	1.48	66.1	10.8	
2.5 SHS	4.78	17.8	11.9	8.91	7.13	5.94	5.09	4.45	3.56	2.97	2.55	2.23	1.98	1.78	1.48	56.0	10.9	
2.0 SHS	3.88	13.3	8.89	6.67	5.33	4.44	3.81	3.33	2.67	2.22	1.90	1.67	1.48	1.33	1.11	45.5	11.0	
50 x 50 x 5.0 SHS	6.39	16.6	11.1	8.29	6.63	5.53	4.74	4.14	3.32	2.76	2.37	2.07	1.84	1.66	1.38	74.7	7.61	
4.0 SHS	5.35	14.3	9.56	7.17	5.74	4.78	4.10	3.59	2.87	2.39	2.05	1.79	1.59	1.43	1.20	62.7	7.85	
3.0 SHS	4.25	11.8	7.89	5.91	4.73	3.94	3.38	2.96	2.37	1.97	1.69	1.48	1.31	1.18	0.986	49.3	8.13	
2.5 SHS	3.60	10.2	6.78	5.09	4.07	3.39	2.91	2.54	2.03	1.70	1.45	1.27	1.13	1.02	0.848	42.0	8.24	
2.0 SHS	2.93	8.39	5.60	4.20	3.36	2.80	2.40	2.10	1.68	1.40	1.20	1.05	0.933	0.839	0.700	34.3	8.34	
1.6 SHS	2.38	6.43	4.29	3.21	2.57	2.14	1.84	1.61	1.29	1.07	0.918	0.804	0.714	0.643	0.536	28.0	8.42	
40 x 40 x 4.0 SHS	4.09	8.49	5.66	4.24	3.39	2.83	2.42	2.12	1.70	1.41	1.21	1.06	0.93	0.849	0.707	47.8	6.09	
3.0 SHS	3.30	7.21	4.81	3.61	2.88	2.40	2.06	1.80	1.44	1.20	1.03	0.901	0.721	0.601	0.501	38.1	6.38	
2.5 SHS	2.82	6.26	4.17	3.13	2.50	2.09	1.79	1.56	1.25	1.04	0.894	0.782	0.695	0.626	0.522	32.7	6.48	
2.0 SHS	2.31	5.21	3.47	2.60	2.08	1.74	1.49	1.30	1.04	0.868	0.744	0.651	0.579	0.521	0.434	26.9	6.59	
1.6 SHS	1.88	4.30	2.87	2.16	1.62	1.72	1.43	1.23	1.07	0.860	0.716	0.614	0.537	0.478	0.430	0.358	22.0	6.67
35 x 35 x 3.0 SHS	2.83	5.33	3.55	2.66	2.13	1.78	1.52	1.33	1.07	0.888	0.761	0.666	0.592	0.533	0.444	32.5	5.50	
2.5 SHS	2.42	4.66	3.10	2.33	1.86	1.55	1.33	1.16	0.931	0.776	0.665	0.582	0.517	0.466	0.388	28.0	5.61	
2.0 SHS	1.99	3.90	2.60	1.95	1.56	1.30	1.11	0.975	0.780	0.650	0.557	0.487	0.433	0.390	0.325	23.1	5.71	
1.6 SHS	1.63	3.23	2.16	1.62	1.29	1.08	0.944	0.808	0.647	0.539	0.462	0.404	0.359	0.323	0.268	19.0	5.80	
30 x 30 x 3.0 SHS	2.36	3.73	2.49	1.86	1.49	1.24	1.07	0.932	0.746	0.621	0.533	0.466	0.414	0.373	0.311	26.9	4.63	
2.5 SHS	2.03	3.29	2.19	1.64	1.32	1.10	0.940	0.822	0.658	0.548	0.470	0.411	0.365	0.329	0.274	23.3	4.73	
2.0 SHS	1.68	2.78	1.85	1.39	1.11	0.926	0.794	0.695	0.556	0.463	0.397	0.347	0.309	0.278	0.222	19.4	4.84	
1.6 SHS	1.38	2.32	1.55	1.16	0.928	0.773	0.683	0.580	0.464	0.387	0.331	0.290	0.258	0.232	0.193	16.0	4.92	
25 x 25 x 3.0 SHS	1.89	2.41	1.61	1.21	0.985	0.804	0.689	0.603	0.483	0.402	0.345	0.302	0.288	0.241	0.201	21.3	3.75	
2.5 SHS	1.64	2.16	1.44	1.08	0.863	0.719	0.617	0.539	0.432	0.360	0.270	0.216	0.180	0.154	0.1301	15.7	3.85	
2.0 SHS	1.36	1.85	1.23	0.924	0.739	0.616	0.528	0.462	0.369	0.308	0.264	0.231	0.205	0.173	0.156	13.0	4.04	
1.6 SHS	1.12	1.56	1.04	0.779	0.623	0.519	0.445	0.389	0.311	0.260	0.222	0.195	0.173	0.156	0.1301	13.0	3.17	

Note: Refer to TABLE 8.1-5 (1)(A)

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X—  
— X

**FOR CANTILEVER BEAMS WITH FULL LATERAL RESTRAINT**

**SQUARE HOLLOW SECTIONS: GRADE C450**

Bending about x-axis

$W_{L1}$  = Maximum Design Load based on Design Moment Capacity  
 $W_{L2}$  = Maximum Design Load based on Design Shear Capacity  
 Maximum Design Load  $W_L$  is LESSER of  $W_{L1}$  and  $W_{L2}$

TABLE 8.4 – 6 (1)(A)

Designation	Mass per m	$W_{L1}$ (kN)										$W_{L2}$	FLR			
		Span of Beam (L) in metres														
d mm	b mm	t mm	1.0	2.0	3.0	4.0	5.0	6.0	7.0	8.0	9.0	10.0	11.0	12.0	13.0	14.0
100 x 100 x 3.8 SHS	11.1	39.2	19.6	13.1	9.79	7.83	6.53	5.59	4.89	4.35	3.92	3.56	3.26	3.01	2.80	2.60
3.3 SHS	9.73	31.8	15.9	10.6	7.96	6.37	5.31	4.55	3.98	3.54	3.18	2.89	2.65	2.45	2.27	2.07
2.8 SHS	8.39	25.1	12.5	8.36	6.27	5.01	4.18	3.58	3.13	2.79	2.51	2.28	2.09	1.93	1.79	1.63
2.3 SHS	6.95	18.7	9.34	6.23	4.67	3.74	3.11	2.67	2.34	2.08	1.87	1.70	1.56	1.44	1.33	1.22
75 x 75 x 3.3 SHS	7.14	19.5	9.75	6.50	4.86	3.90	3.25	2.79	2.44	2.17	1.95	1.77	1.63	1.50	1.39	1.28
2.8 SHS	6.19	16.3	8.16	5.44	4.08	3.26	2.72	2.33	2.04	1.81	1.63	1.48	1.36	1.25	1.17	1.06
2.3 SHS	5.14	12.1	6.07	4.05	3.03	2.43	2.02	1.73	1.52	1.35	1.21	1.10	1.01	0.934	0.867	0.796

Notes: Refer to TABLE 8.4-5 (1)(A)

TABLE 8.4 – 6 (2)(A)

Designation	Mass per m	$W_{L1}$ (kN)										$W_{L2}$	FLR			
		Span of Beam (L) in metres														
d mm	b mm	t mm	0.5	0.75	1.0	1.25	1.5	1.75	2.0	2.5	3.0	3.5	4.0	4.5	5.0	6.0
75 x 75 x 3.3 SHS	7.14	39.0	26.0	19.5	15.6	13.0	11.1	9.75	7.80	6.50	5.57	4.88	4.33	3.90	3.25	2.72
2.8 SHS	6.19	32.6	21.8	16.3	13.1	10.9	9.32	8.16	6.53	5.44	4.66	4.08	3.63	3.26	2.72	2.22
2.3 SHS	5.14	24.3	16.2	12.1	9.71	8.09	6.94	6.07	4.86	4.05	3.47	3.03	2.70	2.43	2.02	1.63
65 x 65 x 2.8 SHS	5.31	25.3	16.9	12.6	10.1	8.43	7.23	6.32	5.06	4.22	3.61	3.16	2.81	2.53	2.11	1.71
2.3 SHS	4.42	19.6	13.1	9.82	7.86	6.55	5.61	4.91	3.93	3.27	2.81	2.46	2.18	1.96	1.64	1.34
50 x 50 x 2.8 SHS	3.99	14.4	9.58	7.19	5.75	4.79	4.11	3.59	2.88	2.40	2.05	1.80	1.60	1.44	1.20	1.00
2.3 SHS	3.34	12.2	8.12	6.09	4.87	4.06	3.48	3.05	2.44	2.03	1.74	1.52	1.35	1.22	1.02	0.84
40 x 40 x 2.8 SHS	3.11	8.80	5.86	4.40	3.52	2.93	2.51	2.20	1.76	1.47	1.26	1.10	0.977	0.880	0.738	0.627
2.3 SHS	2.82	7.52	5.01	3.76	3.01	2.51	2.15	1.88	1.50	1.25	1.07	0.940	0.836	0.752	0.627	0.508
35 x 35 x 2.8 SHS	2.67	6.52	4.34	3.26	2.61	2.17	1.86	1.63	1.30	1.09	0.931	0.815	0.724	0.652	0.543	0.431
2.3 SHS	2.25	5.61	3.74	2.80	2.24	1.87	1.60	1.40	1.12	0.935	0.801	0.701	0.623	0.561	0.467	0.336
30 x 30 x 2.8 SHS	2.23	4.58	3.05	2.29	1.83	1.53	1.31	1.14	0.916	0.763	0.654	0.572	0.509	0.458	0.382	0.328
2.3 SHS	1.89	3.98	2.65	1.99	1.59	1.33	1.14	0.994	0.795	0.663	0.568	0.497	0.442	0.398	0.331	0.280
25 x 25 x 2.8 SHS	1.53	2.62	1.75	1.31	1.05	0.874	0.750	0.656	0.525	0.437	0.375	0.328	0.291	0.262	0.219	0.203

Notes: Refer to TABLE 8.4-5 (1)(A)

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**NOTES**

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