

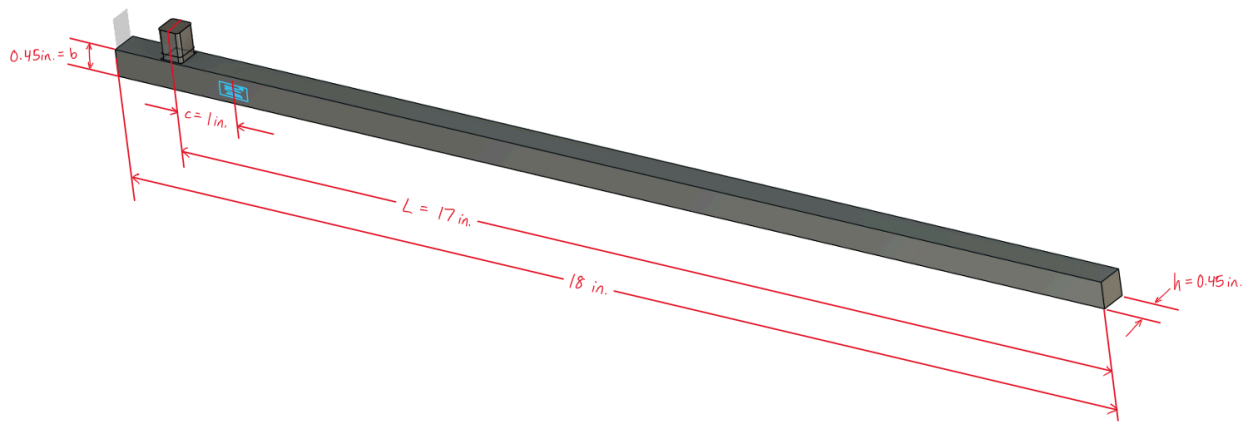
# MAE 3270 - Torque Wrench Design

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## 5.2 Your Design:

### 5.2.1 Results:

#### 1. CAD Model



#### 2. Material used and its relevant mechanical properties

Chosen material: Intermediate alloy, Fe-5Cr-Mo-V aircraft steel, quenched & tempered

```
% dimensions to determine:
L = 17;                               % (inches)
h = 0.45;                             % (inches)
b = 0.45;                             % (inches)
c = 1.0;                              % (inches)
I = (b*h^3)/12;                       % moment of inertia (inches^4) - based on chosen geometry

% chosen material:
name = 'Fe-5Cr-Mo-V';
E = 31.E6;                           % Young's modulus (psi)
nu = .375;                           % Poisson's ration (dimensionless)
su = 240.E3;                          % tensile strength -- use yield or ultimate strength depending
on material (psi)
KIC = 120.E3;                         % fracture toughness (psi sqrt(in))
sfatigue = 125.E3;                   % fatigue strength from Granta for 10^6 cycles
```

Mechanical properties of Fe-5Cr-Mo-V and our chosen dimensions

```

1  T = 600;                                % max torque (in-lbf) FIXED
2
3  % dimensions to determine:
4  L = 17;                                % (inches)
5  h = 0.45;                              % (inches)
6  b = 0.45;                              % (inches)
7  c = 1.0;                               % (inches)
8  I = (b*h^3)/12;                        % moment of inertia (inches^4) - based on chosen geometry
9
10 % chosen material:
11 name = 'Fe-5Cr-Mo-V';
12 E = 31.E6;                             % Young's modulus (psi)
13 nu = .375;                             % Poisson's ration (dimensionless)
14 su = 240.E3;                            % tensile strength -- use yield or ultimate strength depending
    on material (psi)
15 KIC = 120.E3;                          % fracture toughness (psi sqrt(in))
16 sfatigue = 125.E3;                     % fatigue strength from Granta for 10^6 cycles
17
18 % Stress and deflection analysis
19 smax = T * (h/2) / I;                  % max normal stress (psi)
20 Mg = T * (1 - c/L);                   % Moment at gauge
21 sgauge = Mg * (h/2) / I;              % stress at gauge
22 X0 = su / smax;                       % safety factor for yield or brittle failure
23 delta = (T*L^2)/(3*E*I);              % load point deflection (inches)
24
25 % Crack analysis
26 a = .04;                              % crack depth (inches)
27 F = 1.12;                             % geometric factor
28 KI = F * sqrt(pi*a) * smax;           % stress intensity factor (psi*sqrt(in))
29 XK = KIC / KI;                        % safety factor for crack growth
30
31
32 % Fatigue analysis
33 Xs = sfatigue / smax;                  % safety factor for fatigue
34
35 % Strain gauge
36 strain = sgauge / E;
37 output = 1000 * (2 * strain / 2);
38
39
40 % Print values
41 disp("Chosen material: " + name);
42
43 fprintf('\nStress and deflection analysis:\n');
44 fprintf('load point deflection = %.3f in\n', delta);
45 fprintf('max normal stress = %.2f ksi\n', (smax*10^-3));
46
47 fprintf('\nSafety factor results:\n');
48 fprintf('safety factor for strength = %.2f\n', X0);
49 fprintf('safety factor for crack growth = %.2f\n', XK);
50 fprintf('safety factor for fatigue = %.2f\n', Xs);
51
52 fprintf('\nStrain gauge results:\n');
53 fprintf('strain at gauge = %.f microstrain\n', (strain*10^6));
54 fprintf('output = %.2f mV/V at 600 in-lbf using half bridge\n', output);
55

```

Full code used to compute hand calculations and confirm requirements

```

>> FinalHW
Chosen material: Fe-5Cr-Mo-V

Stress and deflection analysis:
load point deflection = 0.546 in
max normal stress = 39.51 ksi

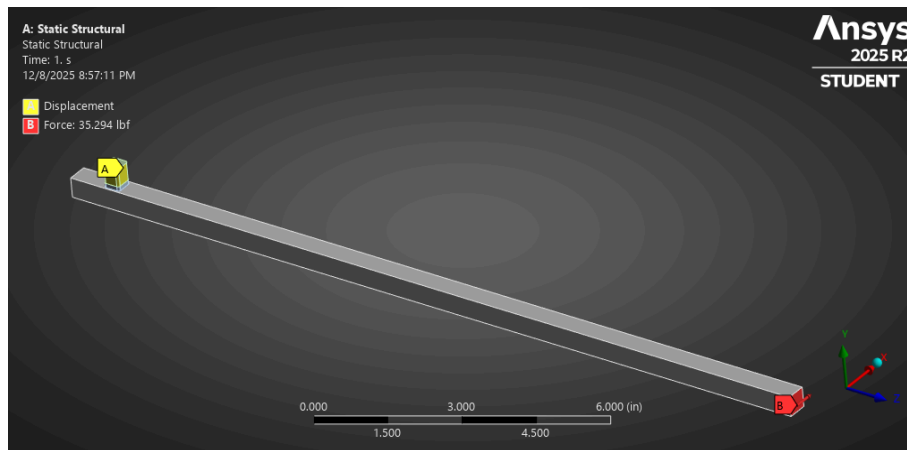
Safety factor results:
safety factor for strength = 6.08
safety factor for crack growth = 7.65
safety factor for fatigue = 3.16

Strain gauge results:
strain at gauge = 1199 microstrain
output = 1.20 mV/V at 600 in-lbf using half bridge

```

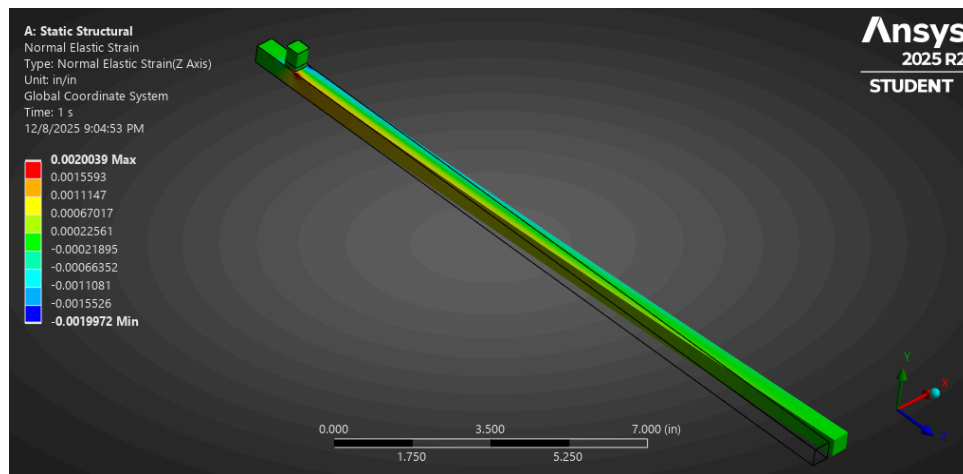
Results of hand calculations with our chosen dimensions and material

- Diagram of how loads and boundary conditions were applied to the FEM model

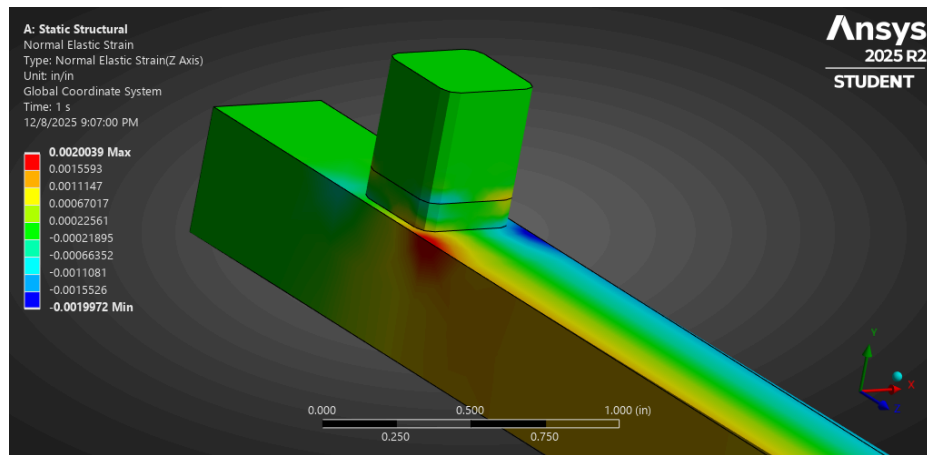


Note: “A” displacement is clamped boundary condition and “B” is applied force

- Normal strain contours from FEM

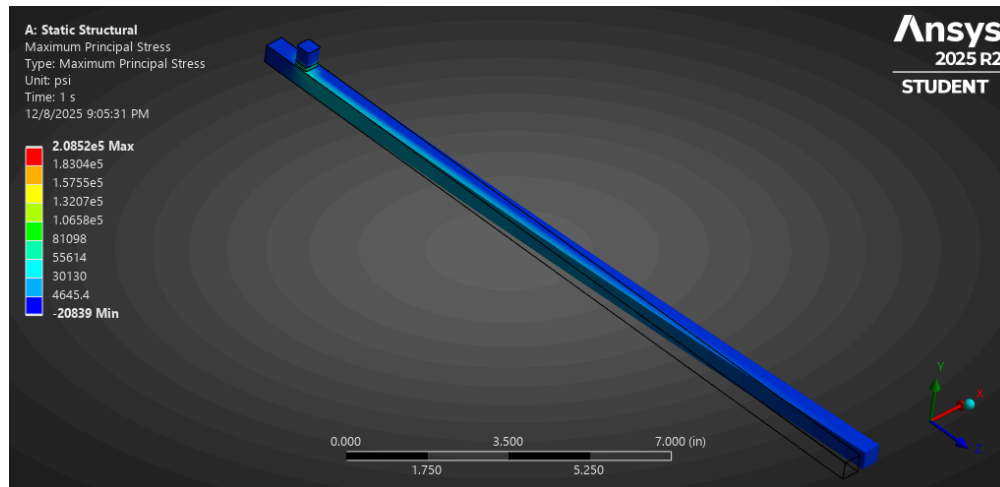


Zoomed out normal strain contour

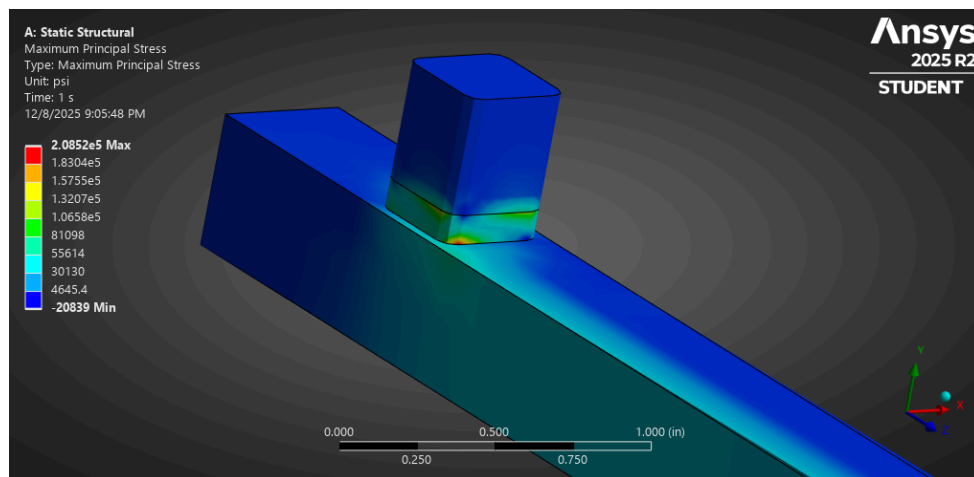


Zoomed in normal strain contour

##### 5. Contour plot of maximum principal stress from FEM

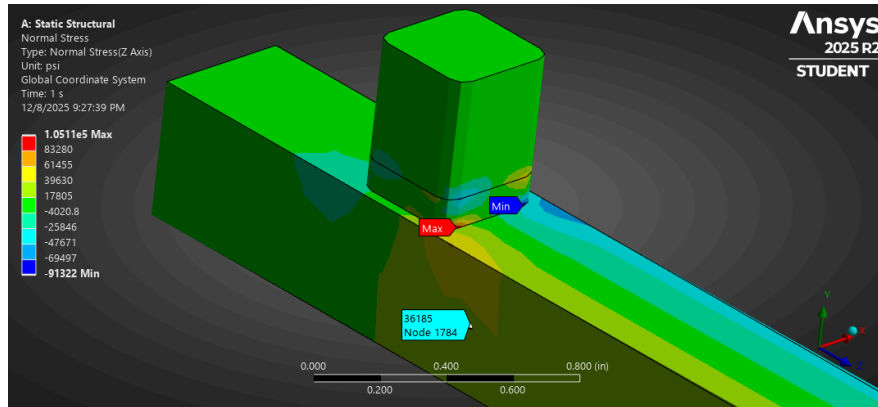


Zoomed out maximum principal stress contour

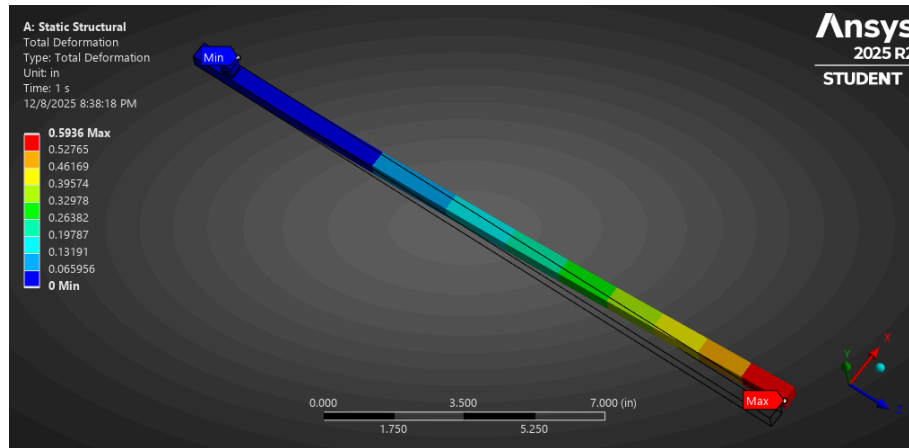


Zoomed in maximum principal stress contour

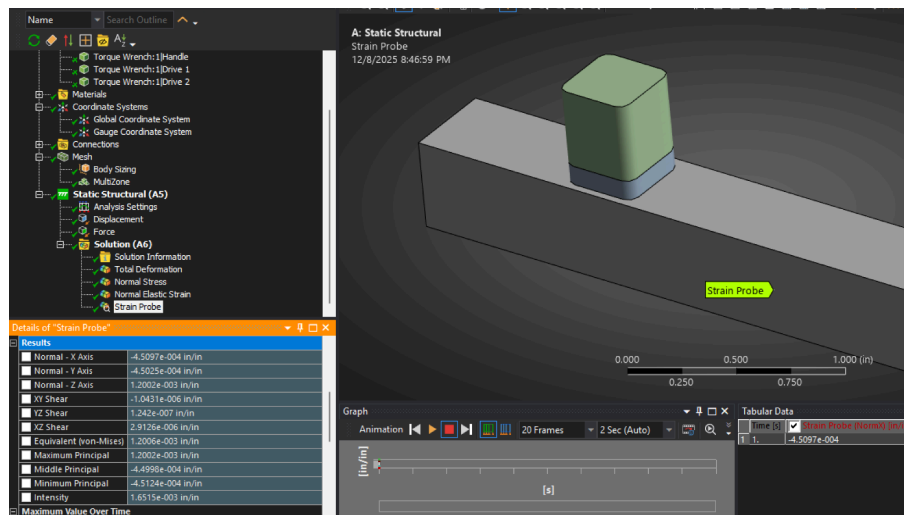
## 6. Results from FEM calculation



Maximum normal stress = 105.11 ksi



Load point deflection = 0.5936"



Strain at strain gauge = 1200.2 microstrain

7. Torque wrench sensitivity in mV/V using strain from FEM analysis

```
% Strain gauge  
strain = 1.2002e-3;  
output = 1000 * (2 * strain / 2);
```

```
Strain gauge results:  
strain at gauge = 1200 microstrain  
output = 1.20 mV/V at 600 in-lbf using half bridge
```

Torque wrench sensitivity = 1.20 mV/V at 600 in-lbf

8. Strain gauge selected (type and dimensions)

Selected strain gauge: [SGD-2/350-LY11 Linear Strain Gage on DwyerOmega](#), 350  $\Omega$  Resistance

Gauge Type:

Miniature linear pattern, measurement of stress concentration

Dimensions:

Grid length = 6 mm

Grid width = 6.3 mm = 0.248 in.

(Note: The grid width is less than the width of the torque wrench, b)