

ECE-6320 HWK 7

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Problem 1

2.3

$$\dot{x} = \begin{bmatrix} x_2 \\ \frac{g}{l} \sin(\theta) - \frac{b}{ml^2} + \frac{u}{ml^2} \end{bmatrix} \quad (1)$$

a)

$$\dot{\delta x} = \begin{bmatrix} 0 & 1 \\ \frac{g}{l} & -\frac{b}{ml^2} \end{bmatrix} \delta x + \begin{bmatrix} 0 \\ \frac{1}{ml^2} \end{bmatrix} \quad (2)$$

$$= \begin{bmatrix} 0 & 1 \\ 4g & -16g \end{bmatrix} \delta x + \begin{bmatrix} 0 \\ 16g \end{bmatrix} \quad (3)$$

We know the eigenvalues of A are $0.2496, -158.0496$. Checking for controllability with the unstable eigenvalue.

$$\text{rank}([0.2496I - A, B]) = 2 \quad (4)$$

$$\text{rank}([-158.0496I - A, B]) = 2 \quad (5)$$

Thus we know this linearized system is both controllable and stabilizable

b)

$$\dot{\delta x} = \begin{bmatrix} 0 & 1 \\ -4g & -16g \end{bmatrix} \delta x + \begin{bmatrix} 0 \\ 16g \end{bmatrix} \quad (6)$$

For this matrix we know the eigenvalues are $-0.25, -156.55$ which is LAS by itself. However when we check the results of controllability we get,

$$\text{rank}([-0.25I - A, B]) = 2 \quad (7)$$

$$\text{rank}([-156.55I - A, B]) = 2 \quad (8)$$

$$(9)$$

We see that the system is controllable and stabilizable.

c)

$$\dot{\delta x} = \begin{bmatrix} 0 & 1 \\ \frac{4g}{\sqrt{2}} & -16g \end{bmatrix} \delta x + \begin{bmatrix} 0 \\ 16g \end{bmatrix} \quad (10)$$

For this system we see the eigenvalues are $0.1766, -156.9766$. So checking the the eigenvalues we get,

$$\text{rank}([0.1766I - A, B]) = 2 \quad (11)$$

$$\text{rank}([-156.9766I - A, B]) = 2 \quad (12)$$

$$(13)$$

Showing that the system is both controllable and stabilizable.

2.4

$$\dot{x} = \begin{bmatrix} x_2 \\ -\sin(x_1) + u \end{bmatrix} \quad (14)$$

c)

For the system we have,

$$\dot{\delta x} = \begin{bmatrix} 0 & 1 \\ -1 & 0 \end{bmatrix} \delta x + \begin{bmatrix} 0 \\ 1 \end{bmatrix} \delta u \quad (15)$$

From this we know the eigenvalues are $\pm 1i$. So checking the controllability of the matrix we see,

$$\text{rank}([-iI - A, B]) = 2 \quad (16)$$

$$\text{rank}([iI - A, B]) = 2 \quad (17)$$

$$(18)$$

From this we see that the system is both Controllable and stabilizable.

2.6)

$$\dot{x} = \begin{bmatrix} x_2 \\ -\frac{gm}{l} \cos(x_1) - \frac{b}{l}x_2 + \frac{1}{l}u \end{bmatrix} \quad (19)$$

b)

$$\delta \dot{x} = \begin{bmatrix} 0 & 1 \\ 1 & -2 \end{bmatrix} \delta x + \begin{bmatrix} 0 \\ 1 \end{bmatrix} \delta u \quad (20)$$

We know the system has eigenvalues $-2.4142, 0.4142$. Now checking if the system is controllable,

$$\text{rank}([-2.4142I - A, B]) = 2 \quad (21)$$

$$\text{rank}([0.4142I - A, B]) = 2 \quad (22)$$

$$(23)$$

Showing the system is both controllable and stabilizable.

2.7

$$\dot{x} = \begin{bmatrix} x_2 u_2 + u_1 \\ -u_2 x_1 \\ u_2 \end{bmatrix} \quad (24)$$

b)

Finding the linearization around the equilibrium point,

$$\delta \dot{x} = \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} \delta x + \begin{bmatrix} 1 & 0 \\ 0 & 0 \\ 0 & 1 \end{bmatrix} \delta u \quad (25)$$

From this system we know the eigenvalues are all 0's. So checking if the system is controllable,

$$\text{rank}([0I - A, B]) = 2 \quad (26)$$

Thus the system is neither controllable or stabilizable.

d)

Finding the linearization around the trajectory,

$$\delta \dot{x} = \begin{bmatrix} 0 & 1 & 0 \\ 1 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} \delta x + \begin{bmatrix} 1 & -1 \\ 0 & 0 \\ 0 & 1 \end{bmatrix} \delta u \quad (27)$$

From this we see that the eigenvalues are $\pm i, 0$. So checking whether the system is controllable,

$$\text{rank}([-iI - A, B]) = 3 \quad (28)$$

$$\text{rank}([iI - A, B]) = 3 \quad (29)$$

$$\text{rank}([0I - A, B]) = 3 \quad (30)$$

$$(31)$$

Thus showing the system is both controllable and stabilizable.

Controllability Proof

For this we need to prove,

$$\mathcal{C}[t_0, t_1] = \text{Im}(W_C(t_0, t_1))$$

Where the definition of Controllability is

$$\mathcal{C}[t_0, t_1] = \{x_0 \in \mathcal{R} \mid \exists u(\cdot) \in \mathcal{U}, 0 = \Phi(t_1, t_0)x_0 + \int_{t_0}^{t_1} \Phi(t_1, \tau)B(\tau)u(\tau) d\tau\} \quad (32)$$

The controllability Grammian

$$W_C(t_0, t_1) = \int_{t_0}^{t_1} \Phi(t_0, \tau)B(\tau)B^T(\tau)\Phi^T(t_0, \tau) d\tau \quad (33)$$

1)

First proving $x_0 \in \text{Im}(W_C(t_0, t_1)) \implies x_0 \in \mathcal{C}(t_0, t_1)$. Starting with the definition of the image space for the Controllability Grammian,

$$\exists \eta_1 \text{ s.t. } x_1 = W_C(t_0, t_1)\eta_1 \quad (34)$$

Now plugging in the our control $u = -B(\tau)\Phi^T(t_0, \tau)\eta_1$ into the controllability definition,

$$0 = \Phi(t_1, t_0)x_1 + \int_{t_0}^{t_1} -\Phi(t_1, \tau)B(\tau)B(\tau)\Phi^T(t_0, \tau)\eta_1 d\tau \quad (35)$$

Now left multiplying by $\Phi(t_0, t_1)$ and using the properties of the state transition matrix,

$$0 = \Phi(t_0, t_0)x_1 + \int_{t_0}^{t_1} -\Phi(t_0, \tau)B(\tau)B(\tau)\Phi^T(t_0, \tau)\eta_1 d\tau \quad (36)$$

$$0 = Ix_1 + \int_{t_0}^{t_1} -\Phi(t_0, \tau)B(\tau)B(\tau)\Phi^T(t_0, \tau)\eta_1 d\tau \quad (37)$$

$$x_1 = \int_{t_0}^{t_1} \Phi(t_0, \tau)B(\tau)B(\tau)\Phi^T(t_0, \tau)\eta_1 d\tau \quad (38)$$

$$x_1 = W_C(t_0, t_1)\eta_1 \quad (39)$$

Which is the definition for the image space of the Controllability Grammian and thus showing that $x_0 \in \mathcal{C}(t_0, t_1)$

2)

Second proving $x_0 \in \mathcal{C}(t_0, t_1) \implies x_0 \in W_C(t_0, t_1)$. Starting with the null space of the controllability grammian.

$$\exists \eta_0 \text{ s.t. } W_C(t_0, t_1)\eta_0 = 0 \quad (40)$$

Now looking at $\eta_0^T W_C \eta_0$,

$$0 = \int_{t_0}^{t_1} \eta_0^T \Phi(t_0, \tau)B(\tau)B^T(\tau)\Phi^T(t_0, \tau)\eta_0 d\tau \quad (41)$$

Now defining $B^T(\tau)\Phi^T(t_0, \tau)\eta_0$ as V we get,

$$0 = \int_{t_0}^{t_1} V^T V d\tau \quad (42)$$

$$0 = \int_{t_0}^{t_1} ||V||^2 d\tau \quad (43)$$

Then seeing that the norm cannot be zero unless V is zero we get,

$$B^T(\tau)\Phi^T(t_0, \tau)\eta_0 = 0 \quad (44)$$

Now using the definition of the orthogonal compliment,

$$x_0^T \eta_0 = 0 \implies x_0 \in (N(W_C))^\perp \implies x_0 \in \text{Im}(W_C) \quad (45)$$

To make this equivalence we start with our definition of controllability and left multiply by $\Phi(t_0, t_1)$

$$0 = \Phi(t_0, t_0)x_0 + \int_{t_0}^{t_1} \Phi(t_0, \tau)B(\tau)u(\tau) d\tau \quad (46)$$

$$-x_0 = \int_{t_0}^{t_1} \Phi(t_0, \tau)B(\tau)u(\tau) d\tau \quad (47)$$

Now using our definition for orthogonal compliment,

$$-x_0^T \eta_0 = \int_{t_0}^{t_1} u^T(\tau)B^T(\tau)\Phi^T(t_0, \tau)\eta_0 d\tau \quad (48)$$

Now seeing we proved $B^T(\tau)\Phi^T(t_0, \tau)\eta_0 = 0$ earlier we know that $x_0^T \eta_0 = 0$. Showing x_0 is in the image space of the controllability grammian.

Control Design

Simple System Design

S1

$$\dot{x} = \begin{bmatrix} 1 & 2 \\ 3 & 4 \end{bmatrix} x + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u \quad (49)$$

Checking the controllability of the system we find,

$$\text{rank}([-0.3722I - A, B]) = 2 \quad (50)$$

$$\text{rank}([0.3722I - A, B]) = 2 \quad (51)$$

So the system is controllable. Now defining u as $-kx$ we get,

$$\dot{x} = \begin{bmatrix} 1 & 2 \\ 3 - k_1 & 4 - k_2 \end{bmatrix} x \quad (52)$$

Now finding the eigenvalues of the system,

$$0 = \det \left(\begin{bmatrix} \lambda - 1 & -2 \\ -(3 - k_1) & \lambda - (4 - k_2) \end{bmatrix} \right) \quad (53)$$

$$0 = (\lambda - 1)(\lambda - 4 + k_2) - 2(3 - k_1) \quad (54)$$

$$0 = \lambda^2 + (-5 + k_2)\lambda + 4 - k_2 - 6 + 2k_1 \quad (55)$$

$$0 = \lambda^2 + (-5 + k_2)\lambda + (-k_2 - 2 + 2k_1) \quad (56)$$

Now we know a GAS system with poles at -1, -2 has the form, $\lambda^2 + 3\lambda + 2$. So solving for the k's, we get $k_1 = 6, k_2 = 8$ so the control law is $u = - \begin{bmatrix} 6 & 8 \end{bmatrix} x$

S2

$$\dot{x} = \begin{bmatrix} -1 & -2 \\ 6 & 7 \end{bmatrix} x + \begin{bmatrix} -0.5 \\ 1 \end{bmatrix} u \quad (57)$$

Checking the controllability of the system we see,

$$\text{rank}([5I - A, B]) = 2 \quad (58)$$

$$\text{rank}([I - A, B]) = 2 \quad (59)$$

So we see the system is controllable

Now adding Bk

$$\dot{x} = \begin{bmatrix} -1 + .5k_1 & -2 + .5k_2 \\ 6 - k_1 & 7 - k_2 \end{bmatrix} x \quad (60)$$

Now finding the eigenvalues of the matrix,

$$0 = \det \left(\begin{bmatrix} \lambda + 1 - .5k_1 & 2 - .5k_2 \\ k_1 - 6 & \lambda + k_2 - 7 \end{bmatrix} \right) \quad (61)$$

$$0 = \lambda^2 + (-.5k_1 + k_2 - 6)\lambda + (1.5k_1 - 2k_2 + 5) \quad (62)$$

Now using the coefficients for eigenvalues of -1, -2 we get $k_1 = 30$ and $k_2 = 24$. Where the control law is $u = - \begin{bmatrix} 30 & 24 \end{bmatrix} x$

S3

$$\dot{x} = \begin{bmatrix} 1 & 2 & 3 \\ 4 & 5 & 6 \\ 7 & 8 & 9 \end{bmatrix} x + \begin{bmatrix} 1 & 0 \\ 0 & 1 \\ 1 & 1 \end{bmatrix} u \quad (63)$$

Checking the controllability of the system we find,

$$\text{rank}([-1.117I - A, B]) = 3 \quad (64)$$

$$\text{rank}([1.117I - A, B]) = 3 \quad (65)$$

$$\text{rank}([-A, B]) = 3 \quad (66)$$

So the system is controllable. Now defining our k as $\begin{bmatrix} k_1 & k_2 & k_3 \\ k_4 & k_5 & k_6 \end{bmatrix}$ we get,

$$\dot{x} = \begin{bmatrix} 1 - k_1 & 2 - k_2 & 3 - k_3 \\ 4 - k_4 & 5 - k_5 & 6 - k_6 \\ -k_1 - k_4 + 7 & -k_2 - k_5 + 8 & -k_3 - k_6 + 9 \end{bmatrix} x \quad (67)$$

$$0 = \det \left(\begin{bmatrix} \lambda + k_1 - 1 & k_2 - 2 & k_3 - 3 \\ k_4 - 4 & \lambda + k_5 - 5 & k_6 - 6 \\ (k_1 + k_4 - 7) & k_2 + k_5 - 8 & \lambda + k_3 + k_6 - 9 \end{bmatrix} \right) \quad (68)$$

$$(69)$$

Using the place function for eigenvalues of -1, -2, and -3 we get a control law of, $u = - \begin{bmatrix} 3.50 & 1.66 & 4.16 \\ 3.99 & 6.66 & 6.65 \end{bmatrix} x$.

Orbit-plane Motion

1)

Starting with our dynamics,

$$\dot{x} = \begin{bmatrix} x_3 \\ x_4 \\ x_4^2 x_1 - \frac{\mu}{x_1^2} + u_1 \\ -2 \frac{x_3 x_4}{x_1} + \frac{u_2}{x_1} \end{bmatrix} \quad (70)$$

Solving for a trajectory around R we know that $x_1 = R$, $x_3 = 0$, and $\dot{x}_3 = 0$. We also assume inputs are 0

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ 0 \\ \dot{x}_4 \end{bmatrix} = \begin{bmatrix} 0 \\ x_4 \\ x_4^2 R - \frac{\mu}{R^2} \\ 0 \end{bmatrix} \quad (71)$$

$$x_4^2 R = \frac{\mu}{R^2} \quad (72)$$

$$x_4 = \sqrt{\frac{\mu}{R^3}} \quad (73)$$

Now we know we have a trajectory around $x_1 = R$, $x_2 = \sqrt{\frac{\mu}{R^3}}t$, $x_3 = 0$, $x_4 = \sqrt{\frac{\mu}{R^3}}$ and $a_i = a_r = 0$

Finding the jacobians we get,

$$\frac{\partial f}{\partial x} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ x_4^2 + 2\frac{\mu}{x_1^3} & 0 & 0 & 2x_4x_1 \\ 2\frac{x_3x_4}{x_1^2} - \frac{a_i}{x_1^2} & 0 & -2\frac{x_4}{x_1} & -2\frac{x_3}{x_1} \end{bmatrix} \quad (74)$$

$$\frac{\partial f}{\partial u} = \begin{bmatrix} 0 & 0 \\ 0 & 0 \\ 1 & 0 \\ 0 & \frac{1}{x_1} \end{bmatrix} \quad (75)$$

Now evaluating about the trajectory,

$$\delta \dot{x} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 3\frac{\mu}{R^3} & 0 & 0 & 2\sqrt{\frac{\mu}{R}} \\ 0 & 0 & -2\sqrt{\frac{\mu}{R^5}} & 0 \end{bmatrix} \delta x + \begin{bmatrix} 0 & 0 \\ 0 & 0 \\ 1 & 0 \\ 0 & \frac{1}{R} \end{bmatrix} \delta u \quad (76)$$

$$\text{Where } \delta x = \begin{bmatrix} x_1 - R \\ x_2 - \sqrt{\frac{\mu}{R^3}}t \\ x_3 \\ x_4 - \sqrt{\frac{\mu}{R^3}} \end{bmatrix} \text{ and } \delta u = \begin{bmatrix} a_r \\ a_i \end{bmatrix}$$

2)

After substituting in μ and R we get eigenvalues of $0, 0, \pm 2.3189 \cdot 10^{-5}i$. So since we have all eigenvalues at $\text{Re}(0)$ we know the system is unstable.

3)

Using the definition for controllability we check the rank of the following matrices,

$$\text{Rank}([0I - A, B]) = 4 \quad (77)$$

$$\text{Rank}([\pm 2.3189 \cdot 10^{-5}iI - A, B]) = 4 \quad (78)$$

This makes sense because the the last 2 columns contain the remaining directions in the space that the B matrix is missing.

4)

For this question we define our K as $\begin{bmatrix} k_1 & k_2 & k_3 & k_4 \\ k_5 & k_6 & k_7 & k_8 \end{bmatrix}$. Using a control law $\delta u = -BK$ we get,

$$\dot{\delta x} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 3\frac{\mu}{R^3} - k_1 & -k_2 & k_3 & 2\sqrt{\frac{\mu}{R}} - k_4 \\ -k_5 & -k_6 & -2\sqrt{\frac{\mu}{R^5}} - k_7 & -k_8 \end{bmatrix} \quad (79)$$

Using the place function with eigenvalues of -1, -1, -2, and -2 we find the control law to be

$$u = - \begin{bmatrix} 2 & 0 & 3 & .00927 \\ 4.44e-14 & 400 & -4.6378e-05 & 600 \end{bmatrix} \delta x$$

5)

