

## Problem 1

$f(t) := |\sin(t)|$  is a periodic function with  $T = \pi$   
 Fourier series of  $f(t)$  is  $f(t+T) = f(t)$

$$f(t) = \sum_{n=-\infty}^{\infty} a_n \cos(2nt)$$

$\because f(t)$  is an even function  $\Rightarrow$  use  $\cos$  to expand  
 Also,  $\cos(2nt)$  has period of  $T = \pi$

①

$$n=0$$

$$\begin{aligned} a_0 &= \frac{1}{\pi} \int_0^\pi |\sin(t)| \cdot \cos(0t) dt \\ &= \frac{1}{\pi} (-\cos(t)) \Big|_{t=0}^\pi = \frac{2}{\pi} \end{aligned}$$

②  $n=k$

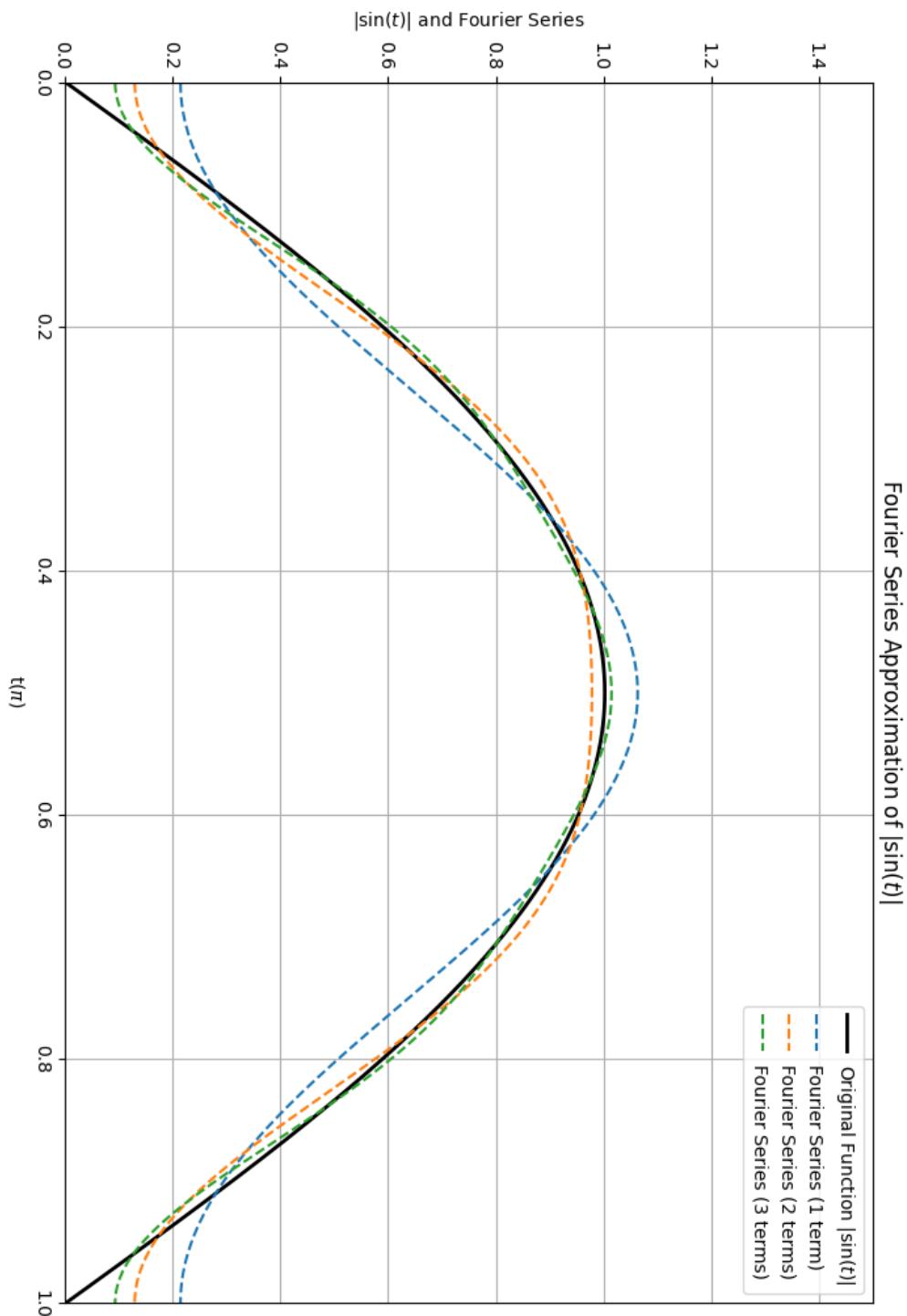
$$\begin{aligned} a_k &= \frac{1}{\pi} \int_0^\pi \sin t \cdot \cos(2kt) dt \\ &= \frac{1}{\pi} \int_0^\pi \frac{1}{2} [\sin((1+2k)t) - \sin((1-2k)t)] dt \\ &= \frac{1}{2\pi} \left\{ \frac{\cos((1+2k)t)}{1+2k} + \frac{\cos((1-2k)t)}{1-2k} \right\} \Big|_0^\pi \\ &= \frac{1}{\pi} \left( \frac{1}{1+2k} + \frac{1}{1-2k} \right) = -\frac{2}{\pi} \frac{1}{4k^2-1} \end{aligned}$$

$$\begin{aligned} \therefore f(t) &= \sum_{n=-\infty}^{\infty} \left( \frac{-2}{\pi} \right) \frac{1}{4k^2-1} \cos(2nt) \\ &= \frac{2}{\pi} + \sum_{n=1}^{\infty} \left( \frac{-4}{\pi} \right) \frac{1}{4k^2-1} \cos(2nt) \end{aligned}$$

We plot the first three terms in the following.

### Problem 1

The figure is the first three fourier terms of the fourier series of the full rectifier



## Problem 2

(a)

$$\mathcal{L}\{x(t)\} = (\partial_t^2 + 2\beta\partial_t + \omega_0^2)x = F(t)/m$$

$$\mathcal{L}\{g(t, t')\} = \delta(t - t')$$

$$\Rightarrow g(t, t') = \frac{A}{\omega_1} e^{-\beta(t-t')} \sin[\omega_1(t-t')]$$

$$\omega_1 = \sqrt{\omega_0^2 - \beta^2} \quad \text{From the result of HWQ5}$$

$$\therefore x(t) = \int_0^{NT_1} g(t, t') \cdot \frac{F_{ext}(t')}{m} dt'$$

$$= \frac{A}{\omega_1} \int_0^{NT_1} e^{-\beta(t-t')} \sin[\omega_1(t-t')] \cdot \sin(\omega_1 t') dt'$$

$$= \frac{A}{\omega_1} \int_0^{NT_1} e^{-\beta(t-t')} \left( \frac{-1}{4} \right) \left[ e^{i\omega_1(t-t')} - e^{-i\omega_1(t-t')} \right]$$

$$\left( T_1 := \frac{2\pi}{\omega_1} \right) \cdot \left[ (e^{i\omega_1 t'} - e^{-i\omega_1 t'}) \right] dt'$$

$$= \frac{-A}{4\omega_1} \int_0^{NT_1} e^{-\beta(t-t')} \left( e^{i\omega_1 t} - e^{-i\omega_1 t} - \frac{i\omega_1(t-2t')}{2} - \frac{i\omega_1(t-2t')}{2} \right) dt'$$

$$= \frac{-A}{2\omega_1} \int_0^{NT_1} e^{-\beta(t-t')} \{ \cos(\omega_1 t) - \cos[\omega_1(t-2t')] \} dt'$$

For large  $N$ , the anti-rotating term  $\cos[\omega_1(t-2t')]$  will have negligible contribution

$$\approx \frac{-A}{2\omega_1} \cos(\omega_1 t) e^{-\beta t} \left( e^{+\beta NT_1} - 1 \right) / \beta$$

$$= \frac{-A}{2\beta\omega_1} \cos(\omega_1 t) e^{-\beta t} \cdot \left( e^{\beta NT_1} - 1 \right)$$

(b) As  $\beta \rightarrow 0$

$$\lim_{\beta \rightarrow 0} x(t) = \frac{-A}{2\omega_1} - \lim_{\beta \rightarrow 0} e^{-\beta t} \cdot \frac{e^{j\beta NT_1}}{\beta} \cos(\omega_1 t)$$
$$= \frac{-A}{2\omega_1} \cos(\omega_0 t) \cdot (NT_1) \quad (\because \omega_1 \rightarrow \omega_0 \text{ as } \beta \rightarrow 0)$$

We can see that  $x(t)$ 's amplitude is proportional to  $N$ , which means, if the  $F_{ext}$  continues forever ( $N \rightarrow \infty$ ), the amplitude will also go to infinity

$\Rightarrow$  Same results as frictionless resonance  
for equation of motion

$$\frac{d^2}{dt^2} x + \omega_0^2 x = A_m \cos(\omega t)$$

$$\Rightarrow \text{Amp}(x) = \infty \text{ as } \omega \rightarrow \omega_0$$

For the phase delay  $\phi$ ,

$$\tan \phi = \left( \frac{2\beta \omega}{\omega_0^2 - \omega^2} \right), \text{ as } \omega \rightarrow \omega_0,$$

$$\tan \phi \rightarrow +\infty, \phi \rightarrow \frac{\pi}{2}$$

And  $x(t) \propto \cos(\omega t)$  as  $\beta \rightarrow 0$

has a phase delay of  $\frac{\pi}{2}$  compared with the  
 $F_{ext} \propto \sin(\omega_0 t)$  as  $\beta \rightarrow 0$

### Problem 3

$$\begin{aligned}
 \text{(a)} \quad T_1 &= \frac{1}{2} m_1 r^2 \cdot \omega^2 + \frac{1}{2} m_1 \dot{r}^2 \\
 &= \frac{1}{2} m_1 (r^2 \omega^2 + \dot{r}^2) \\
 T_2 &= \frac{1}{2} m_2 (\dot{x}_2^2 + \dot{y}_2^2) \\
 &= \frac{1}{2} m_2 \left[ \frac{d}{dt} [r \cos(\omega t) + l \cos(\omega t + \theta)] \right]^2 \\
 &\quad + \left[ \frac{d}{dt} [r \sin(\omega t) + l \sin(\omega t + \theta)] \right]^2 \\
 &= \frac{1}{2} m_2 (r^2 \omega^2 + \dot{r}^2) \\
 &\quad + \frac{1}{2} m_2 l^2 (\omega + \dot{\theta})^2 \\
 &\quad + m_2 [-\dot{r} \cos(\omega t) l \sin(\omega t + \theta) (\omega + \dot{\theta}) \\
 &\quad \quad + \dot{r} \sin(\omega t) l \cos(\omega t + \theta) (\omega + \dot{\theta}) \\
 &\quad \quad + \omega r \sin(\omega t) l \sin(\omega t + \theta) (\omega + \dot{\theta}) \\
 &\quad \quad + \omega r \cos(\omega t) l \cos(\omega t + \theta) (\omega + \dot{\theta})] \\
 &= m_2 l (\omega + \dot{\theta}) \left[ \dot{r} \sin(\omega t - \omega t - \theta) \right. \\
 &\quad \quad \left. + \omega r \cos(\omega t - \omega t - \theta) \right] \\
 &\quad + \frac{1}{2} m_2 [r^2 \omega^2 + \dot{r}^2 + l^2 (\omega + \dot{\theta})^2] \\
 &= \frac{1}{2} m_2 [r^2 \omega^2 + \dot{r}^2 + l^2 (\omega + \dot{\theta})^2] \\
 &\quad + m_2 l (\omega + \dot{\theta}) [-\dot{r} \sin \theta + \omega r \cos \theta] \\
 V &= \frac{1}{2} k (r - r_0)^2 \\
 L &= T_1 + T_2 - V
 \end{aligned}$$

$$(b) \rho_r = \frac{\partial L}{\partial \dot{r}}$$

$$= m_1 \dot{r} + m_2 \dot{r} - m_2 l (\omega + \dot{\theta}) \sin \theta$$

$$\rho_\theta = \frac{\partial L}{\partial \dot{\theta}}$$

$$= m_2 l^2 (\omega + \dot{\theta}) - m_2 l [r \sin \theta + \omega r \cos \theta]$$

$$(c) H = \dot{r} \rho_r + \dot{\theta} \rho_\theta - L$$

$$= \frac{1}{2} (m_1 + m_2) (\dot{r}^2 - \omega^2 r^2)$$

$$+ \frac{1}{2} m_2 l^2 (-\omega^2 + \dot{\theta}^2)$$

$$- m_2 l \omega [r \sin \theta + \omega r \cos \theta]$$

$$+ \frac{1}{2} k (r - r_0)^2$$

$$(d) E = T_1 + T_2 + V$$

$H$  is clearly not  $E$ ; rather,  $H$  is some very weird form.

$$\frac{dH}{dt} = -\frac{\partial L}{\partial t} = 0$$

( $\because L$  is no explicit function of  $t$ , unless  $w = w(t)$  is not const.)

$E$  is not conserved because we must exert some external torque to maintain the rotational speed of  $m_1$  as  $w$  (if  $m_2$  and the rigid rod will apply torque on  $m_1$ ).

$$\therefore \frac{dE}{dt} = T_{ext} \cdot w \neq 0 \Rightarrow E \text{ is not conserved}$$

(e) For  $r = r_0 = \text{const.}$ ,

$$L = \frac{1}{2} m_2 [r^2 \omega^2 + l^2 (\omega + \dot{\theta})^2]$$

$$+ m_2 \omega l r (\omega + \dot{\theta}) \cos \theta$$

$$\rightarrow L = \frac{1}{2} m_2 l^2 (\omega + \dot{\theta})^2$$

$$+ m_2 \omega l r_0 (\omega + \dot{\theta}) \cos \theta$$

$$\frac{\partial L}{\partial \theta} = \frac{d}{dt} \left( \frac{\partial L}{\partial \dot{\theta}} \right)$$

$$\Rightarrow -m_2 \omega (\omega + \dot{\theta}) l r_0 \sin \theta$$

$$= \frac{d}{dt} [m_2 l^2 (\omega + \dot{\theta}) + m_2 \omega l r_0 \cos \theta]$$

$$\Rightarrow -m_2 \omega (\omega + \dot{\theta}) l r_0 \sin \theta$$

$$= m_2 l^2 \ddot{\theta} - m_2 \omega \dot{\theta} l r_0 \sin \theta$$

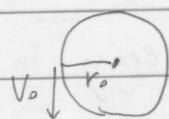
$$\Rightarrow l \ddot{\theta} + \omega^2 r_0 \sin \theta = \dot{\theta}$$

$$\Rightarrow g = \omega^2 r_0 \Rightarrow \text{Centrifugal acceleration}$$

## Problem 4.

(a)

In the old orbit,



$$m \frac{v_0^2}{r_0} = \frac{GMm}{r_0^2} \Rightarrow v_0^2 = \frac{GM}{r_0} = \frac{k}{mr_0}$$

$$\text{Angular M.m. } L = mr_0^2\omega$$

$$= m v_0 r_0 = mr_0 \sqrt{\frac{k}{mr_0}} = \sqrt{mk} r_0$$

$$\begin{aligned} \text{New } E_{\text{new}} &= \frac{1}{2}mv_0^2 - \frac{k}{r} + \frac{1}{2}\frac{I^2}{m} \\ &= E_{\text{old}} + \frac{1}{2}\frac{I^2}{m} \end{aligned}$$

In the new orbit, the angular momentum is not changed by the impulse  $\vec{I}$  because  $\vec{I}$  is along the radial direction.

$$\therefore \alpha_{\text{new}} = \frac{L^2}{mk} = \alpha_{\text{old}} := \alpha = r_0$$

$$E_{\text{new}} = \sqrt{1 + \frac{2E_{\text{new}}L^2}{mk^2}} = \sqrt{1 + \frac{2(E_{\text{old}} + \frac{I^2}{2m})L^2}{mk^2}}$$

$$= \sqrt{1 + \frac{2E_{\text{old}}L^2}{mk^2} + \frac{I^2L^2}{m^2k^2}}$$

$$= \sqrt{0 + \frac{IL^2}{m^2k^2}} \quad (\because E_{\text{old}} = \sqrt{1 + \frac{2E_{\text{old}}L^2}{mk^2}} = 0)$$

$$= \frac{IL}{mk} = \frac{I}{mk} \sqrt{mk} r_0$$

$$= I \sqrt{\frac{r_0}{mk}}$$

### Problem 4(b)

We plotted the old and new orbits. For the new orbits, we plotted the case for  $\epsilon = 0.8$ , for both cases of applying impulse outward and inward.

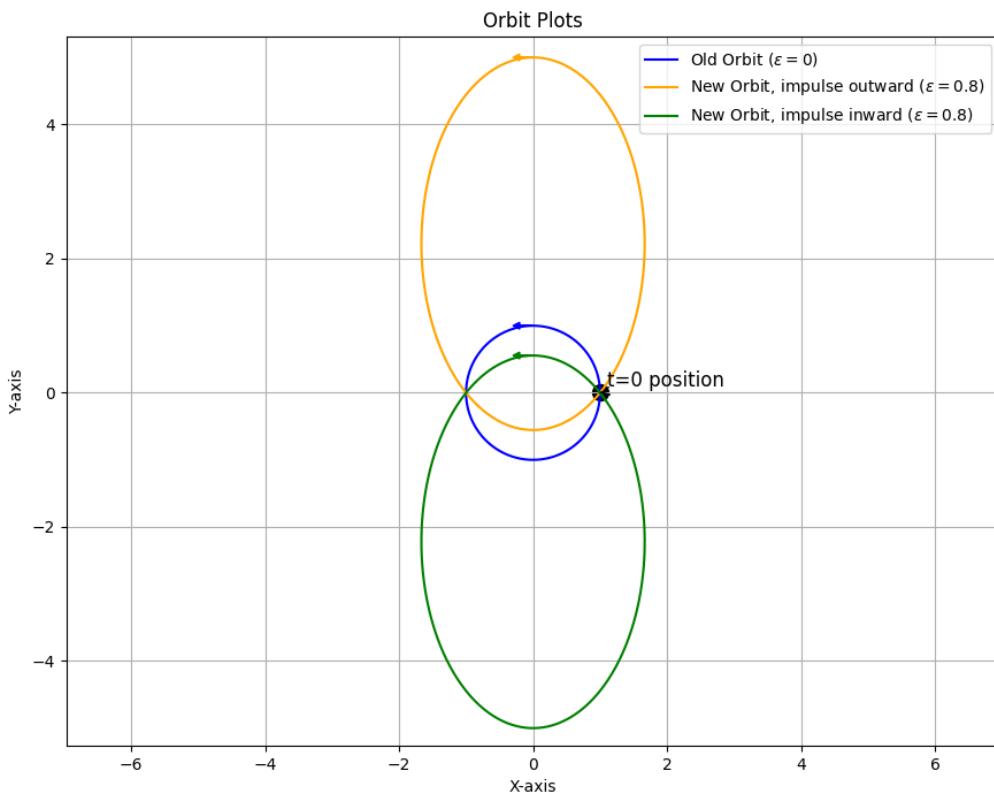
The orbit equation for the new orbit is:

$$r(\theta) = \frac{\alpha}{1 + \epsilon \cos(\theta + \theta_0)}$$

Because at  $t=0$ ,  $\theta = 0$ , and  $r(\theta = 0) = \alpha = r_0$ ,  $\theta_0$  must be  $\pm\pi/2$ .

If  $\theta_0 = \pi/2$ , we will have the case that impulse is applied outward.

If  $\theta_0 = -\pi/2$ , we will have the case that impulse is applied inward.



(C)

$$P_{\text{star}} := P_o \frac{r_o^2}{r^2}$$

$E_o = P_o \cdot T_o$  is total energy in one period  
 $T_o$  (Period of the old orbit)

In the new orbit,

$$E_{\text{new}} = \int_0^{T_{\text{new}}} P(r) dt = \int_0^{T_{\text{new}}} P_o r_o \frac{1}{r^2} dt$$

$$= \int_0^{T_{\text{new}}} P_o r_o \frac{m}{mr^2} \frac{d\theta}{dt} \frac{1}{r^2} dt$$

$$= \frac{P_o r_o m}{L_{\text{new}}} \int_0^{2\pi} d\theta \quad (L_{\text{new}} = mr^2 \frac{d\theta}{dt})$$

$$= \frac{2\pi P_o r_o m}{L_{\text{new}}} = P_o \left( \frac{2\pi}{T_o} \cdot m r_o^2 \right) \cdot \frac{1}{L_{\text{new}}} \cdot T_o$$

$$= P_o \frac{L_{\text{old}} T_o}{L_{\text{new}}} = P_o T_o = E_o \quad (\because L_{\text{new}} = L_{\text{old}}) \quad (\text{C.1})$$

∴ The total energy for one period is the same for both old and new orbits.

$\bar{P}_{\text{new}}$  is the average power in one period in the new orbit

$$\bar{P}_{\text{new}} = \frac{E_o}{T_{\text{new}}} < \frac{E_o}{T_o} = P_o = \bar{P}_{\text{old}}$$

$$\because T_{\text{new}} > T_o$$

For old orbit

$$\frac{k}{r_0^2} = \frac{m}{r_0} \left( \frac{2\pi r_0}{T_0} \right)^2 \Rightarrow T_0 = 2\pi r_0 \sqrt{\frac{mr_0}{k}} = 2\pi \sqrt{\frac{mr_0^3}{k}}$$

For new orbit, due to Kepler's 3rd law, we know

$$T_{\text{new}} = 2\pi \sqrt{\frac{ma^3}{k}}, \text{ where } \frac{x}{1-\varepsilon} + \frac{x}{1+\varepsilon} = 2a$$

$$\Rightarrow a_{\text{new}} = \frac{x_{\text{new}}}{1-\varepsilon_{\text{new}}} = \frac{r_0}{1-\varepsilon_{\text{new}}^2} > r_0$$

$$\Rightarrow T_{\text{new}} > T_0$$

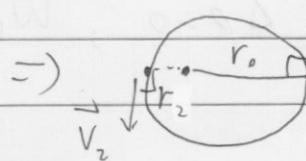
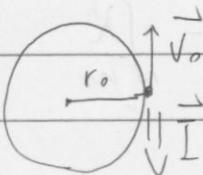
$\therefore (C.1)$

$\therefore$  To increase the average power, one should

① increase  $E_{\text{new}} = E_{\text{old}} \cdot \frac{L_{\text{old}}}{L_{\text{new}}} \rightarrow$  decrease  $L_{\text{new}}$

② decrease  $T_{\text{new}} \rightarrow$  decrease  $a_{\text{new}} = \frac{x_{\text{new}}}{1-\varepsilon_{\text{new}}^2}$

We can achieve both ① & ② by reducing  $L_{\text{new}}$  by giving  $\vec{I}$  opposite to the current velocity  $\vec{V}_0$ .



$\Rightarrow$  Weaker centrifugal force,

$$a_{\text{new}} = \frac{r_0 + r_2}{2} < r_0 \quad \text{the ship is pulled inside by the gravitational force more}$$

$$\Rightarrow r_2 < r_0$$

$\Rightarrow T_{\text{new}}$  is decreased

$$L_{\text{new}} \text{ decreased} \Rightarrow E_{\text{new}} = E_{\text{old}} \cdot \frac{L_{\text{old}}}{L_{\text{new}}} > E_{\text{old}} \text{ increased}$$

### Problem 5

(a)

$$m \frac{d^2r}{dt^2} = F(r) + F_{\text{centrifugal}}$$

$$= -\frac{k}{r} e^{-\frac{r}{a}} + \frac{\ell^2}{mr^3}$$

For stable orbit,  $r = \rho = \text{const.}$

$$m \frac{d^2r}{dt^2} = 0 = -\frac{\ell^2}{mr^3} - \frac{k}{\rho^2} e^{-\frac{\rho}{a}} \Rightarrow \frac{\ell^2}{mr^3} = \frac{k}{\rho^2} e^{-\frac{\ell}{a}}$$

For small perturbation  $\rho' = \rho + \Delta\rho$

$$(|\Delta\rho| \ll \rho)$$

$$\Rightarrow m \frac{d^2\Delta\rho}{dt^2} = -\frac{3\ell^2}{mr^4} \Delta\rho + \left( \frac{2k}{\rho^3} + \frac{k}{a\rho^2} \right) \Delta\rho e^{-\frac{\ell}{a}}$$

$$= \left[ \frac{k}{\rho^2} \left( \frac{2}{\rho} + \frac{1}{a} \right) e^{-\frac{\ell}{a}} - \frac{3\ell^2}{mr^4} \right] \Delta\rho$$

$$\Rightarrow K := \frac{3\ell^2}{mr^4} - \frac{k}{\rho^2} \left( \frac{2}{\rho} + \frac{1}{a} \right) e^{-\frac{\ell}{a}} = \frac{k}{\rho^2} \left( \frac{1}{\rho} - \frac{1}{a} \right)$$

$$\therefore m \frac{d^2\Delta\rho}{dt^2} + K \Delta\rho = 0, \quad \omega_{\text{osc}} = \sqrt{\frac{K}{m}}$$

$K > 0$  for

$$\frac{3k}{\rho^3} e^{-\frac{\ell}{a}} \frac{3\ell^2}{mr^4} > \frac{k}{\rho^2} \left( \frac{2}{\rho} + \frac{1}{a} \right) e^{-\frac{\ell}{a}} \Rightarrow 3 > 2 + \frac{\ell}{a}$$

$$\Rightarrow \rho < \rho_{\text{stab}} = a$$

$$(b) \frac{\ell^2}{m\rho^3} = \frac{k}{\rho^2} e^{-\frac{\rho}{a}}$$

$$\Rightarrow \ell^2 = \frac{mk\rho}{m\rho^3} e^{-\frac{\rho}{a}} = \frac{k^2}{\rho^4} \omega^2$$

$$\Rightarrow \omega = \sqrt{\frac{k}{m\rho^3}} e^{-\frac{\rho}{a}} = \sqrt{\frac{k}{m\rho^3}} e^{-\frac{\rho}{2a}}$$

$$\theta_{Aps} = \frac{2\pi}{\omega_{osc}} \times \omega$$

$$= \frac{2\pi}{\sqrt{\frac{k}{m\rho^3}}} \times \sqrt{\frac{k}{m\rho^3}} e^{-\frac{\rho}{2a}}$$

$$\sqrt{\frac{k}{m\rho^3} \left( \frac{1}{\rho} - \frac{1}{a} \right)}$$

$$= 2\pi e^{-\frac{\rho}{2a}} \cdot \sqrt{\frac{1}{\left(1 - \frac{\rho}{a}\right)}}$$

### Problem 5(c)

Here we made the plot for different values of initial radius:

$$r_0 = \rho = 0.75, 0.95, \text{ and } 1.2$$

Here we set  $a = 1$ .

We can see that when  $r_0$  is smaller than 1, the orbit is very stable in a circle (like the 0.75 and 0.95 case), but when  $r_0 = 1.2 > 1$ , it becomes quite unstable after several turns.

