16

DNA-TR-86-92

CONSERVATION EQUATIONS FOR A NONSTEADY FLOW OF A COMPRESSIBLE VISCOUS SINGLE-PHASE FLUID IN VARIOUS COORDINATE SYSTEMS

Gordon C. K. Yeh Allen L. Kuhl R & D Associates P. O. Box 9695 Marina del Rey, CA 90295-2095

31 December 1984

Technical Report

CONTRACT No. DNA 001-84-C-0006

Approved for public release; distribution is unlimited.

THIS WORK WAS SPONSORED BY THE DEFENSE NUCLEAR AGENCY UNDER RDT&E RMSS CODE B310084466 P990MXDB00006 H2590D.

5

Prepared for Director DEFENSE NUCLEAR AGENCY Washington, DC 20305-1000



Destroy this report when it is no longer needed. Do not return to sender.

PLEASE NOTIFY THE DEFENSE NUCLEAR AGENCY, ATTN: STTI, WASHINGTON, DC 20305-1000, IF YOUR ADDRESS IS INCORRECT. IF YOU WISH IT DELETED FROM THE DISTRIBUTION LIST, OR IF THE ADDRESSEE IS NO LONGER EMPLOYED BY YOUR ORGANIZATION.

THE PROCESS SECTION OF THE PARTY OF THE PART

ECONTIT CLASSIFICATION OF THIS PAGE							
	REPORT DOCUM	ENTATION PA	AGE				
18. REPORT SECURITY CLASSIFICATION UNCLASSIFIED	1b. RESTRICTIVE MARKINGS						
2. SECURITY CLASSIFICATION AUTHORITY		3. DISTRIBUTION/AVAILABILITY OF REPORT					
N/A since Unclassified	DULE	Approved for public release; distribution is unlimited.					
20. DECLASSIFICATION/DOWNGRADING SCHE							
4. PERFORMING ORGANIZATION REPORT NUMBER(S) RDA-TR-130004-001		5. MONITORING ORGANIZATION REPORT NUMBER(S)					
		DNA-TR-86-92					
6. NAME OF PERFORMING ORGANIZATION	7s. NAME OF MONITORING ORGANIZATION						
R & D Associates	(If applicable)	Director					
		Defense Nuclear Agency					
Sc. ADDRESS (City, State and ZIP Code) P.O. BOX 9695	7b. ADDRESS (City, State and ZIP Code)						
Marina del Rey, CA 90295-	2095	Washington, DC 20305-1000					
B. NAM2 OF FUNDING/SPONSORING ORGANIZATION	8b. OFFICE SYMBOL (If applicable)			STRUMENT IDENTIFICATION NUMBER			
		DNA 001-84-C-0006					
AOORESS (City, State and ZIP Coals)			FUNDING NUMB				
		PROGRAM ELEMENT NO.	PROJECT NO.	TASK NO.		WORK UNIT ACCESSION NO.	
			P99QMXD	B		DH007342	
11. TITLE (Include Security Commiscention) CONSERVATION EQUATIONS FOR A NONSTEADY FLOW OF A COMPRESSIBLE VISCOUS SINGLE-PHASE FLUID IN VARIOUS COORDINATE SYSTEMS							
12 PERSONAL AUTHORISH Yeh, Gordon C.K. and Kuhl, Allen L.							
ISA TYPE OF REPORT ISA TIME (14. DATE OF REPORT (Year, Month, Day) 15. PAGE COUNT						
Technical Report FROM 840101 to 841231 841231 46							
This work was sponsored by the Defense Nuclear Agency under RDT&E RMSS Code B310084466 P99QMXDB00006 H2590D.							
17. COSATI CODES	is suspect tenus Nonsteady F	(Continue on reverse if necessary and identify by block number)					
FIELD GROUP SUB-GROUP							
20 11	20 4 Compressible Viscous Fluid 20 11 Momentum Conservation Equation						
IB. ARSTRACT (Centinue on revenue if necessary	and identify by block nu	(75 -1 100)					
"Based on a generalization of the Reynolds transport theorem the mass,							
momentum and energy conservation equations for a nonsteady flow of a com-							
pressible viscous single-phase fluid (expressed in vector and dyadic notations) are first formulated in integral form (in terms of moving							
volume and surface elements), next converted into differential form and							
then transformed in terms of orthogonal curvilinear coordinates. Special-							
ized equations are obtained for three-dimensional flows in Cartesian,							
cylindrical and spherical coordinates. These equations can then be							
reduced to corresponding equations for one and two dimensional flows in							
various coordinates. Equations for the vorticity, entropy and enthalpy and Bernoulli equation are also summarized.							
20. DISTRIBUTION/AVAILABILITY OF ABSTRA		21. ABSTRACT	SECURITY CLASS	FICATI	ON		
DUNCLASSIFIED/UNLINITED SISAME A	S RPT. DOTIC USERS	UNCLASSI		J			

Betty L. Fox DD FORM 1473, 84 MAR

NAME OF THE PARTY OF THE PARTY

224 NAME OF RESPONSIBLE INDIVIDUAL

83 APR edition may be used until exhausted. All other editions are obsolets.

UNCLASSIFIED

225. TELEPHONE Include Area Code! 22c. OFFICE SYMBOL (202) 325-7042 DNA/STTI

18. SUBJECT TERMS (Continued)

Energy Conservation Equation
Cartesian, Cylindrical, Spherical, and Orthogonal Curvilinear Coordinates
One, Two, and Three Dimensional Flows
Vector and Dyadic (Second Order Tensor) Notations
Bernoulli Equation
Vorticity
Entropy
Enthalpy
Kinematic Transport Theorem
Integral and Differential Formulations
Conservative and Nonconservative Forms

Accessi	on For	-,	
NTIS (T.	
DIIC T	13	, <u>I</u>	
Unnon	here.	i,	1
July 1 1 1	r (t. Fou		
guarde n 21 a - 1	•	. W Therefore	-
Ву	e care comme	enter and the second of the second	
Distri	Vite Edge F	<i>!</i> .	***
Lieva	. 1111	· Codes	
graphica .	Arath i	·n/or	
pist	Spect	nl	
	Ì		
101			
H-1	. 1		

TOTAL CONTRACT CONTRACTOR



TABLE OF CONTENTS

Section		Page
1	INTRODUCTION	1
2	INTEGRAL FORMULATION	2
	2.1 Kinematic Transport Theorem	2
	2.2 Conservation Laws	5
	2.3 Constitutive Relations	7
	2.4 Thermodynamic/Transport Properties	8
3	CONSERVATION LAWS IN DIFFERENTIAL FORM	11
	3.1 Conversion by Generalized Definition of Operator V	11
	3.2 Conservative Form	12
	3.3 Nonconservative Form	13
•	3.4 Vorticity Distribution	15
	3.5 Bernoulli Equation	20
4	TRANSFORMATION OF CONSERVATION EQUATIONS IN TERMS OF ORTHOGONAL CURVILINEAR COORDINATES	24
5	CONSERVATION EQUATIONS IN CARTESIAN COORDINATES	27
, 6	CONSERVATION EQUATIONS IN CYLINDRICAL COORDINATES	28
7	CONSERVATION EQUATIONS IN SPHERICAL COORDINATES	30
8	COMPACT ONE-DIMENSIONAL FORMULATION	32
9	CONCLUSIONS	33
10	LIST OF REFERENCES	34

INTRODUCTION

The governing equations in fluid mechanics are usually expressed in terms of Cartesian coordinates. They are not readily convertible into forms useful for practical applications to one-, two- and three-dimensional cylindrical and spherical problems. One of the purposes of this report is to transform the mass, momentum and energy conservation equations for a nonsteady flow of a compressible viscous single-phase fluid from expressions in vector and dyadic notations to those in terms of orthogonal cuvilinear coordinates and then to specialize the equations for Cartesian, cylindrical and spherical coordinates.

Another purpose of this report is to provide a theoretical basis for usage of Eulerian sliding grids in modern computational fluid mechanics by establishing a kinematic transport theorem which is a generalization of the Reynolds transport theorem. Based upon the theorem the conservation equations in integral form are first formulated in terms of moving volume and surface elements and then converted into the corresponding equations in differential form by using the generalized definition of the vector operator V.

In order to present an overall survey of the basic equations this report also summarizes the equations for the vorticity, entropy and enthalpy and Bernoulli equation.

INTEGRAL FORMULATION

2.1 KINEMATIC TRANSPORT THEOREM.

Let $\underline{x} = (x_1, x_2, x_3)$ denote the rectangular spatial ("Eulerian") coordinates which identify a fixed point in space. Let $\underline{X} = (X_1, X_2, X_3)$ denote the rectangular material ("Lagrangian") coordinates which identify a fluid particle in motion. Let $F = F(\underline{x}, t)$ represent any arbitrary single-valued scalar or vector point function (of position \underline{x} and time t) possessing continuous derivatives. The function

$$M_{\overline{V}} = \int_{\overline{V}} F(\underline{x}, t) d\overline{V} = \int_{V_{O}} F[\underline{x}(\underline{x}, t), t] J dV_{O}, \qquad (1)$$

where $\overline{V} = \overline{V}(t)$ denotes a material volume (that is, a volume moving with the fluid), is a well-defined function of time. In Eq. (1) the Jacobian (Ref. 1, p.33)

$$J = \frac{d\overline{V}}{dV_0} = \frac{\partial(x_1, x_2, x_3)}{\partial(x_1, x_2, x_3)} = \det\left(\frac{\partial x_1}{\partial x_\alpha}\right), \quad (i = 1, 2, 3; \ \alpha = 1, 2, 3)$$
 (2)

relates the element $d\vec{V}$ of the moving volume \vec{V} in the x-variables to the element dV_0 of the fixed volume $V_0 = \vec{V}(0) = \vec{V}(t)$ as t = 0 in the X-variables.

Using Euler's expansion formula (Ref. 2)

$$\frac{dJ}{dt} = J\nabla \cdot \underline{y} \tag{3}$$

(where $\underline{\mathbf{v}}$ is the velocity vector of the fluid motion) and the relation between the material time derivative and the spatial derivatives (Ref. 3, Eq. (3.6))

$$\frac{d}{dt} = \frac{\partial}{\partial t} + \underline{\mathbf{v}} \cdot \nabla \tag{4}$$

we can express the material time derivative of $M_{\overline{V}}$ as

$$\frac{d}{dt} M_{\overline{V}} = \int_{V_{O}} \left(J \frac{dF}{dt} + F \frac{dJ}{dt} \right) dV_{O}$$

$$= \int_{V_{O}} \left[\frac{dF}{dt} + F (\overline{V} \cdot \underline{V}) \right] J dV_{O}$$

$$= \int_{\overline{V}} \left[\frac{\partial F}{\partial t} + \underline{V} \cdot \overline{V} F + F (\overline{V} \cdot \underline{V}) \right] d\overline{V}$$

$$= \int_{V_{O}} \left[\frac{\partial F}{\partial t} + div (F \underline{V}) \right] d\overline{V} \qquad (5)$$

Then in view of Green's transformation (Ref. 1, Eq. (7.2)) for any vector or tensor field ϕ

$$\int_{V} \operatorname{div} \phi \, dV = \oint_{A} d\underline{A} \cdot \phi \, , \, (d\underline{A} = \underline{n}dA) \tag{6}$$

we obtain

$$\int_{\mathbf{V}} \frac{\partial \mathbf{F}}{\partial \mathbf{t}} d\mathbf{V} = - \oint_{\mathbf{A}} \mathbf{F} \underline{\mathbf{v}} \cdot \underline{\mathbf{n}} d\mathbf{A} + \frac{d}{d\mathbf{t}} M_{\overline{\mathbf{V}}} , \qquad (7)$$

or

$$\frac{\partial}{\partial t} \int_{V} F dV = - \oint_{A} F \underline{v} \cdot \underline{n} dA + \frac{d}{dt} M_{\overline{V}} , \qquad (7a)$$

where V denotes the volume fixed in space which instantaneously coincides with the material volume \overline{V} , A denotes the surface bounding the volume V and \underline{n} denotes the unit vector along the outward normal to A. Eq. (7a) is known as the transport theorem of Reynolds (Ref. 1, Eq. (25.4) and Ref. 4, § 14). It is a kinematic relation independent of any meaning attached to F. All fluid physics is contained in the $\frac{d}{dt}M_{\overline{V}}$ term.

Now let us consider a volume $\tilde{V} = \tilde{V}(t)$ with bounding surface $\tilde{A} = \tilde{A}(t)$ sliding with respect to the fluid and obeying the relation

$$\frac{\partial}{\partial E} (d\tilde{V}) = \underline{u} \cdot \underline{n} d\tilde{A} , \qquad (8)$$

where u is the local surface velocity. We can write

$$\int_{\hat{V}} F \frac{\partial}{\partial t} (d\hat{V}) = \int_{\hat{A}} F \underline{u} \cdot \underline{n} d\hat{A} . \tag{9}$$

To each time instant t when the fixed volume V coincides with the material volume \tilde{V} there corresponds a time instant t_1 when the moving volume \tilde{V} coincides with the same material volume \tilde{V} instantaneously. At t_1 Eq. (9) becomes

$$\int_{\mathbf{V}} \mathbf{F}_{1} \frac{\partial}{\partial \mathbf{E}_{1}} (\mathbf{d} \hat{\mathbf{V}}) = \int_{\mathbf{A}} \mathbf{F}_{1} \underline{\mathbf{u}}_{1} \cdot \underline{\mathbf{n}} d\hat{\mathbf{A}} , \qquad (10)$$

where
$$F_1 = F(\underline{x}, t_1)$$
, $\underline{u}_1 = \underline{u}(\underline{x}, t_1)$ and $\frac{\partial}{\partial t_1}(d\hat{V}) = \left[\frac{\partial}{\partial t}(d\hat{V})\right]_{t=t_1}$.

Since V, A and \underline{n} in Eq. (10) have the same meaning as those in Eq. (7) and dV and dA (at t) in Eq. (7) equal to $d\hat{V}$ and $d\hat{A}$ (at t_1) in Eq. (10) we may add Eqs. (7) and (10) to obtain

$$\int_{\mathbf{V}} \left[\frac{\partial \mathbf{F}}{\partial \mathbf{t}} d\mathring{\mathbf{V}} + \mathbf{F}_{1} \frac{\partial}{\partial \mathbf{t}_{1}} (d\mathring{\mathbf{V}}) \right] = - \oint_{\mathbf{A}} (\mathbf{F}\underline{\mathbf{v}} - \mathbf{F}_{1}\underline{\mathbf{u}}_{1}) \cdot \underline{\mathbf{n}} d\mathring{\mathbf{A}} + \frac{d}{d\mathbf{t}} M_{\overline{\mathbf{V}}} . \tag{11}$$

In case V and \hat{V} can coincide with \overline{V} at the same time $(t_1 = t)$ (for example, if $\underline{u}_1 = \underline{0}$ or $\underline{u}_1 = \underline{v}$) Eq. (11) becomes

$$\int_{V} \frac{\partial}{\partial t} (F d\hat{V}) = - \oint_{A} F(\underline{v} - \underline{u}) \cdot \underline{n} d\hat{A} + \frac{d}{dt} M_{\overline{V}} , \qquad (11a)$$

or (since V is fixed in space)

$$\frac{\partial}{\partial E} \int_{\mathbf{V}} \mathbf{F} d\vec{\mathbf{V}} = - \oint_{\mathbf{A}} \mathbf{F} (\underline{\mathbf{v}} - \underline{\mathbf{u}}) \cdot \underline{\mathbf{n}} d\hat{\mathbf{A}} + \frac{\mathrm{d}}{\mathrm{d}E} \mathbf{M}_{\overline{\mathbf{V}}} . \tag{11b}$$

This is a generalization of the Reynolds transport theorem and gives the fundamental transport formula for the time variation of the volume integral of any fluid state quantity F over a moving volume which instantaneously coincides with the material volume. As $\underline{u} = \underline{0}$ ($\tilde{V} = V$, hence $t_1 = t$) Eq. (11b) reduces to Eq. (7a) which is the basis of the Eulerian grid representation in numerical algorithm. As $\underline{u} = \underline{v}$ ($\tilde{V} = \overline{V}$, hence $t_1 = t$) Eq. (11b) becomes a statement of the Lagrangian mesh representation. Ref. 5 shows an example of the application of Eq. (11b).

2.2 CONSERVATION LAWS.

The mass, momentum and energy conservation equations for a nonsteady flow of a compressible, viscous, heat-conducting fluid derived and discussed in reference books on gas dynamics (see, for example, Refs. 6 to 13) have been summarized in recent books on computational fluid mechanics (see, for example, Refs. 14 to 16) in compact vector and dyadic (second order tensor) notations. In a fixed Eulerian frame of reference these equations in integral form (which is more basic than the differential form in expressing the physical laws) are as follows:

$$\frac{\text{mass}}{\partial t} \int_{V} \rho dV + \oint_{A} \rho \underline{v} \cdot \underline{n} dA = 0 , \qquad (12)$$

$$\underline{\text{momentum}} : \quad \frac{\partial}{\partial t} \int_{V} \rho \underline{v} dV + \oint_{A} \rho \underline{v} (\underline{v} \cdot \underline{n}) dA = \oint_{A} \underline{\sigma} \cdot \underline{n} dA + \int_{V} \underline{f} dV , \qquad (13)$$

energy:
$$\frac{\partial}{\partial t} \int_{\mathbf{V}} \rho \mathbf{E} d\mathbf{V} + \oint_{\mathbf{A}} \rho \mathbf{E} \underline{\mathbf{v}} \cdot \underline{\mathbf{n}} d\mathbf{A} = \oint_{\mathbf{A}} (\underline{\mathbf{g}} \cdot \underline{\mathbf{v}} - \underline{\mathbf{q}}) \cdot \underline{\mathbf{n}} d\mathbf{A}$$

$$+ \int_{\mathbf{V}} \underline{\mathbf{f}} \cdot \underline{\mathbf{v}} d\mathbf{V} + \int_{\mathbf{V}} \mathbf{G} d\mathbf{V} . \qquad (14)$$

In these equations, ρ is the density, E the total specific energy ("specific" means per unit mass)

$$E = e + \frac{1}{2} v^2$$
 (15)

(where e is the specific internal energy), \underline{g} is the stress tensor, \underline{g} the heat-flux vector, \underline{f} the external force vector per unit volume, and G is the energy generation per unit volume, and V is a control volume fixed in space, A is the surface bounding V, and \underline{n} is the unit vector along the outward normal to A.

In these equations the properties of the fluid need not be continuous functions of space and time.

By comparison of Eq. (7a) with Eqs. (12) to (14) the function F may be identified as ρ , $\rho \underline{v}$ and ρE respectively and the term $dM_{\overline{V}}/dt$ as the right-hand side terms of each of these equations. Then in terms of the moving elements $(d\widehat{V} = dV \text{ and } d\widehat{A} = dA$ instantaneously) the mass, momentum and energy conservation equations in case $t_1 = t$ (exactly or approximately) can be expressed exactly or approximately as a single vector equation according to Eq. (11b):

$$\frac{\partial}{\partial t} \int_{V} \underline{w} \ d\hat{V} = - \oint_{A} \underline{w} (\underline{v} - \underline{u}) \cdot \underline{n} d\hat{A}$$

$$+ \oint_{A} \underline{\Sigma} \cdot \underline{n} d\hat{A} + \int_{V} \underline{Q} d\hat{V} , \qquad (16)$$

where

$$\underline{W} = \begin{pmatrix} \rho \\ \rho \underline{v} \\ \rho \underline{E} \end{pmatrix}, \quad \underline{\Sigma} = \begin{pmatrix} 0 \\ \underline{g} \\ \underline{g} \cdot \underline{v} - \underline{q} \end{pmatrix}, \quad \underline{Q} = \begin{pmatrix} 0 \\ \underline{f} \\ \underline{f} \cdot \underline{v} + G \end{pmatrix} .$$
(17)

Eq. (16) provides a theoretical basis for usage of Eulerian sliding grids in finite-difference numerical algorithm (see, for example, Ref. 5).

2.3 CONSTITUTIVE RELATIONS.

The basic dependent variables in Eqs. (12) to (14) or Eqs. (16) are ρ , \underline{v} and \underline{E} (or e). Constitutive relations for \underline{g} and \underline{q} must be added to these equations in order to obtain a closed system. We are concerned here with the case of Newtonian fluids, i.e., by definition, fluids such that the stress tensor

is a linear function of the velocity gradient. From this definition, excluding the existence of distributed force couples, results Newton's law, also called the Navier-Stokes law, for \underline{q} :

$$\underline{g} = -p\underline{I} + \underline{I} \quad , \tag{18}$$

with

$$\underline{\mathbf{I}} = \lambda \left(\nabla \cdot \underline{\mathbf{v}} \right) \underline{\mathbf{I}} + 2\mu \underline{\mathbf{D}} \tag{19}$$

and

$$\underline{\mathbf{p}} = \frac{1}{2} \left[\nabla \underline{\mathbf{v}} + (\nabla \underline{\mathbf{v}})^{t} \right] , \qquad (20)$$

the superscript t denoting the transpose of a tensor. In Eqs. (18) to (20) p is the pressure, I is the viscous (or deviatoric) stress tensor, I is the unit tensor, μ and λ are the first (shear) and second (dilatational) coefficients of viscosity, and D is the tensor of rates of deformation. Furthermore, the fluid is assumed to obey Fourier's law of heat conduction for q:

$$\mathbf{q} = -\mathbf{k}\nabla\mathbf{T} \quad , \tag{21}$$

where T is the absolute temperature, and k is the thermal conductivity coefficient. Many fluids, in particular air and water, follow Newton's law and Fourier's law.

2.4 THERMODYNAMIC/TRANSPORT PROPERTIES.

The state variables ρ , e, p, T and the specific entropy S are connected by thermodynamic relations (assuming local thermodynamic equilibrium).

We consider the case of a simple fluid such that all its thermodynamic properties can be deduced from a single fundamental relationship which, for a compressible fluid, can be chosen of the type

$$S = S(\rho, e) \quad . \tag{22}$$

From this relationship p and T are obtained in terms of the basic variables ρ and e from

$$p = -\rho^2 T \left(\frac{\partial S}{\partial \rho}\right)_e , T = \frac{1}{(\partial S/\partial e)_\rho} . \qquad (23)$$

An important special case is a perfect gas with constant specific heats c_p and c_v . For such a gas the laws of state are

$$p = (\gamma-1) \rho e, \quad (\gamma = c_p/c_v)$$
 (24)

and

$$e = c_v T$$
 (25)

The viscosity and thermal conductivity coefficients depend on the local thermodynamic state; in most conditions they depend only on the temperature:

$$\mu = \mu(\mathbf{T}), \quad \lambda = \lambda(\mathbf{T}), \quad \mathbf{k} = \mathbf{k}, \mathbf{T}$$
 (26)

The coefficient

$$\kappa = 3\lambda + 2u \tag{27}$$

is called bulk viscosity coefficient In the "Stokes relation" (Ref. 3, p. 238)

$$3\lambda + 2\mu = 0 \tag{2E}$$

it is assumed to be zero. However, except for very special conditions, for example, monatomic gases, there is no reason to assume $3\lambda = -2\mu$. (Ref. 9, p.337 and Ref. 7, p.540.)

CONSERVATION LAWS IN DIFFERENTIAL FORM

3.1 CONVERSION BY GENERALIZED DEFINITION OF OPERATOR V.

If the properties of the fluid are continuous and sufficiently differentiable in some domain of space and time, then the conservation equations in integral form can be converted into an equivalent set of partial differential equations through a generalized definition of the vector operator V (Ref. 17, p.40)

$$\nabla \phi = \lim_{\mathbf{V} \to \mathbf{0}} \frac{1}{\mathbf{V}} \oint_{\mathbf{A}} \phi \underline{\mathbf{n}} d\mathbf{A} , \qquad (29)$$

where ϕ is an unspecified (scalar, vector or dyadic) function of position, V is the volume enclosed by a surface A to which the point P at which $\nabla \phi$ is to be calculated remains interior, while the largest dimension of A tends to zero, and the multiplication in $\nabla \phi$ may be scalar, vector or dyadic. In particular when ϕ = b, a vector, the scalar product becomes

$$\nabla \cdot \underline{\mathbf{b}} = \lim_{\mathbf{V} \to \mathbf{0}} \frac{1}{\mathbf{V}} \oint_{\mathbf{A}} \underline{\mathbf{b}} \cdot \underline{\mathbf{n}} d\mathbf{A} , \qquad (30)$$

which is a scalar, and when $\phi = g$, a dyadic, the dot product becomes

$$\nabla \cdot \underline{\beta} = \lim_{V \to 0} \frac{1}{V} \oint_{\mathbf{A}} \underline{\beta} \cdot \underline{\mathbf{n}} dA , \qquad (31)$$

which is a vector. Eqs. (30) and (31) can be also obtained by dividing Eq. (6) by V and then taking the limits of both sides as V approaches to zero.

THE STATE OF THE S

3.2 CONSERVATIVE FORM.

Dividing Eqs. (12) to (14) by V, taking the limit of every term as V approaches zero and applying Eqs. (30) and (31) we obtain the conservation equations in differential form:

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \underline{\mathbf{v}}) = 0 , \qquad (32)$$

$$\underline{\text{momentum}}: \quad \frac{\partial}{\partial t} (\rho \underline{\mathbf{v}}) + \nabla \cdot (\rho \underline{\mathbf{v}} \underline{\mathbf{v}} - \underline{\mathbf{g}}) = \underline{\mathbf{f}} \quad , \tag{33}$$

energy:
$$\frac{\partial}{\partial t} (\rho E) + \nabla \cdot (\rho E \underline{v} - \underline{g} \cdot \underline{v} + \underline{q}) = \underline{f} \cdot \underline{v} + G$$
. (34)

An alternate form of the energy equation Eq. (34) is in terms of the enthalpy per unit mass

$$\mathbf{h} \doteq \mathbf{e} + \mathbf{p}/\rho \tag{35}$$

Substituting

$$E = h + \frac{1}{2} v^2 - p/\rho = R - p/\rho$$
 (36)

into Eq. (34) we obtain

$$\frac{\partial}{\partial E} (\rho H) + \nabla \cdot (\rho H \underline{v} - p \underline{v} - \underline{g} \cdot \underline{v} + \underline{q}) = \underline{f} \cdot \underline{v} + \frac{\partial p}{\partial E} + G , \quad (37)$$

where

$$H = h + \frac{1}{2} v^2 = E + p/\rho$$
 (38)

is the total enthalpy per unit mass (which is also known as the "specific stagnation enthalpy" or for nonviscous and nonheat-conducting fluid in steady motion the "total energy per unit mass", Trf. 18, p. 33).

Substituting Eq. (36) into Eq. (14) we obtain the integral form of the conservation equation for ρH in terms of fixed elements dV and dA

$$\frac{\partial}{\partial t} \int_{V} \rho H dV + \oint_{A} \rho H \underline{v} \cdot \underline{n} dA$$

$$= \oint_{A} (\underline{g} \cdot \underline{v} - \underline{q} + \underline{p}\underline{v}) \cdot \underline{n} dA + \int_{V} (\underline{f} \cdot \underline{v} + \underline{G} + \frac{\partial \underline{p}}{\partial t}) dV . \qquad (37a)$$

The corresponding conservation equation in terms of moving elements $d\hat{V}$ and $d\hat{A}$ in case t_1 = t is Eq. (16) with

$$\underline{\underline{W}} = (\rho \underline{H}), \ \underline{\Sigma} = (\underline{\underline{g}} \cdot \underline{\underline{v}} - \underline{\underline{q}} + \underline{\underline{p}}\underline{\underline{v}}), \ \underline{Q} = (\underline{\underline{f}} \cdot \underline{\underline{v}} + \underline{G} + \frac{\partial \underline{p}}{\partial \underline{f}})$$
 (17a)

3.3 NONCONSERVATIVE FORM.

Eqs. (32) to (34) are mass, momentum and energy equations in "conservative" or "divergence" form. (See Section III-A-3 of Ref. 14 for the meaning and beneficial effects of using this form). The corresponding equations in nonconservative form in which g follows Newton's law are (Ref. 16, Section 1.1):

$$\frac{\text{mass}}{\text{dt}} + \rho \nabla \cdot \underline{\mathbf{v}} = 0 , \qquad (39)$$

$$\underline{\text{momentum}}: \quad \rho \ \frac{d\underline{v}}{d\underline{t}} + \nabla p = \underline{f} + \mu \nabla^2 \underline{v} + (\lambda + \mu) \nabla (\nabla \cdot \underline{v})$$

$$+ (\nabla \cdot \underline{\mathbf{v}}) \nabla \lambda + 2\underline{\mathbf{p}} \cdot \nabla \mu , \qquad (40)$$

energy:
$$\rho \frac{de}{dt} + p\nabla \cdot \underline{v} = \Phi - \nabla \cdot \underline{q} + G$$
, (41)

where Φ is the dissipation function

$$\Phi = \underline{\mathbf{I}} : \nabla \underline{\mathbf{v}} = \lambda (\nabla \cdot \underline{\mathbf{v}})^2 + 2\mu \underline{\mathbf{D}} : \underline{\mathbf{D}}$$
 (42)

and $\frac{d}{dt}$ is the material derivative given in Eq. (4).

An alternate form of the energy equation Eq. (41) is in terms of the specific entropy S such that (Ref. 3, Eq. (33.1))

$$TdS = de + pd(\frac{1}{\rho}) . (43)$$

By use of Eqs. (39) and (43) we obtain from Eq. (41)

$$\rho T \frac{dS}{dt} = \psi - \nabla \cdot q + G . \qquad (44)$$

Adding Eq. (32) (multiplied by S) and Eq. (44) (divided by T) we have the conservative form of the conservation equation for pS

$$\frac{\partial}{\partial E} (\rho S) + \nabla \cdot (\rho S \underline{v}) = \frac{1}{\pi} (\Phi - \nabla \cdot \underline{q} + G) . \qquad (44a)$$

Taking the integral of Eq. (44a) over a fixed control volume V and applying Green's transformation Eq. (6) we obtain the integral form of the conservation equation for pS in terms of fixed elements dV and dA

$$\frac{\partial}{\partial E} \int_{V} \rho S dV + \int_{A} \rho S \underline{v} \cdot \underline{n} dA$$

$$= \int_{V} \frac{1}{T} (\phi - \nabla \cdot \underline{q} + G) dV \qquad (44b)$$

The corresponding conservation equation in terms of moving elements $d\vec{V}$ and $d\vec{A}$ in case t_1 = t is Eq. (16) with

$$\underline{\mathbf{W}} = (\rho \mathbf{S}), \ \underline{\mathbf{\Sigma}} = (\underline{\mathbf{O}}), \ \underline{\mathbf{Q}} = (\Phi/\mathbf{T} - \nabla \cdot \mathbf{g}/\mathbf{T} + \mathbf{G}/\mathbf{T}) \tag{17b}$$

3.4 VORTICITY DISTRIBUTION.

The vorticity vector

$$\Omega = \nabla \times \mathbf{v} \tag{45}$$

gives the intrinsic rotation of each fluid element.

By taking the curl of both sides of Eq. (40) (divided by ρ) and noting that

$$\nabla \cdot \underline{\Omega} = 0 \tag{46}$$

and

$$\nabla \times \nabla$$
 (scalar function) = 0 (47)

we obtain the general equation of vorticity distribution

$$\frac{d\Omega}{dE} - (\underline{\Omega} \cdot \nabla) \underline{v} + \underline{\Omega} (\nabla \cdot \underline{v}) + \nabla (\frac{1}{\rho}) \times \nabla p$$

$$= \frac{1}{\rho} \nabla \times \underline{E} + \nabla (\frac{1}{\rho}) \times \underline{E} + \frac{\mu}{\rho} \nabla^2 \underline{\Omega} - \nabla (\frac{\mu}{\rho}) \times \nabla \times \underline{\Omega}$$

$$+ \nabla (\frac{\lambda + 2\mu}{\rho}) \times \nabla (\nabla \cdot \underline{v}) - \frac{1}{\rho} \nabla \lambda \times \nabla (\nabla \cdot \underline{v})$$

$$+ (\nabla \cdot \underline{v}) \nabla (\frac{1}{\rho}) \times \nabla \lambda + \frac{2}{\rho} \nabla \times (\underline{D} \cdot \nabla \mu) + \nabla (\frac{2}{\rho}) \times (\underline{D} \cdot \nabla \mu)$$

$$(48)$$

If μ and λ are not functions of space variables Eq. (48) reduces to

$$\frac{d\Omega}{d\dot{\epsilon}} - (\underline{\Omega} \cdot \nabla) \underline{v} + \underline{\Omega} (\nabla \cdot \underline{v}) + \nabla (\frac{1}{\rho}) \times \nabla p$$

$$= \frac{1}{\rho} \nabla \times \underline{\epsilon} + \nabla (\frac{1}{\rho}) \times \underline{\epsilon} + \frac{\mu}{\rho} \nabla^2 \underline{\Omega}$$

$$+ (\lambda + 2\mu) \nabla (\frac{1}{\rho}) \times \nabla (\nabla \cdot \underline{v}) - \mu \nabla (\frac{1}{\rho}) \times \nabla \times \underline{\Omega}$$
(49)

If the flow is incompressible which is characterized by the condition

$$\nabla \cdot \mathbf{v} = 0 \tag{50}$$

and implies that ρ is constant along a fluid particle trajectory but not necessarily independent of space variables (as in stratified flows, Ref. 16, Section 1.3) Eq. (48) reduces to

$$\frac{d\Omega}{dt} - (\underline{\Omega} \cdot \nabla) \underline{v} + \nabla (\frac{1}{\rho}) \times \nabla p$$

$$= \frac{1}{\rho} \nabla \times \underline{t} + \nabla (\frac{1}{\rho}) \times \underline{t} + \frac{\mu}{\rho} \nabla^2 \underline{\Omega}$$

$$-\nabla (\frac{\mu}{\rho}) \times \nabla \times \underline{\Omega} + \frac{2}{\rho} \nabla \times (\underline{p} \cdot \nabla \mu) + \nabla (\frac{2}{\rho}) \times (\underline{p} \cdot \nabla \mu) . \tag{51}$$

If, in addition, ρ is constant everywhere Eq. (51) reduces to

MENTER THE PROPERTY OF THE PRO

$$\frac{d\Omega}{dE} - (\underline{\Omega} \cdot \nabla) \underline{\nu}$$

$$= \frac{1}{\rho} \nabla \times \underline{f} + \frac{\mu}{\rho} \nabla^2 \underline{\Omega}$$

$$- \frac{1}{\rho} \nabla_{\mu} \times \nabla \times \underline{\Omega} + \frac{2}{\rho} \nabla \times (\underline{\Omega} \cdot \nabla_{\mu}) \quad . \tag{52}$$

If, in addition, μ is not a function of space variables Eq. (52) reduces to (Ref. 16, Eq. (1.37))

$$\frac{d\Omega}{dt} - (\underline{\Omega} \cdot \nabla) \underline{\mathbf{v}} = \frac{1}{\rho} \nabla \cdot \mathbf{x} \, \underline{\mathbf{f}} + \frac{\mu}{\rho} \nabla^2 \underline{\Omega} \quad . \tag{53}$$

This equation is usually associated with an equation for a solenoid stream-function vector $\underline{\Psi}$ such that

$$\underline{\mathbf{v}} = \nabla \times \underline{\underline{\mathbf{y}}} \quad , \tag{54}$$

which automatically satisfies the incompressibility condition Eq. (50). The equation satisfied by $\underline{\Psi}$ is derived by applying the curl operator to Eq. (54) and using definition Eq. (45) to obtain

$$\nabla^2 \underline{\Psi} + \underline{\Omega} = \underline{0} \tag{55}$$

since

$$\nabla \cdot \underline{\Psi} = 0 \tag{56}$$

as Y is solenoid.

Let us return now to consider the case of compressible flow. For an ideal fluid ($\mu=\lambda=0$) either Eq. (48) or Eq. (49) reduces to [Ref. 18, Eq. (7-20)]

$$\frac{d\Omega}{dE} - (\underline{\Omega} \cdot \nabla) \underline{v} + \underline{\Omega} (\nabla \cdot \underline{v})$$

$$= \frac{1}{0} \nabla \times \underline{f} + \nabla (\frac{1}{0}) \times \underline{f} - \nabla \times (\frac{1}{0} \nabla p) \quad . \tag{57}$$

Another form of this equation can be obtained by using Eq. (43) Ref. 18, Eq. (7-21):

$$\frac{\mathbf{q}\underline{\mathbf{f}}}{\mathbf{q}\overline{\mathbf{U}}} - (\overline{\mathbf{U}} \cdot \underline{\mathbf{A}})\overline{\mathbf{A}} + \overline{\mathbf{U}}(\underline{\mathbf{A}} \cdot \overline{\mathbf{A}})$$

$$= \frac{1}{\rho} \nabla \times \underline{\mathbf{f}} + \nabla (\frac{1}{\rho}) \times \underline{\mathbf{f}} + \nabla \mathbf{T} \times \nabla \mathbf{S} . \qquad (58)$$

If p is a function of ρ only the last term of Eq. (57) can be written as

$$\nabla \times (\frac{1}{\rho} \nabla p) = \nabla \times \nabla \left(\int \frac{dp}{\rho} \right)$$
, (59)

which is zero according to Eq. (47). If the specific entropy S is a constant, then the last term of Eq. (58) is zero because $\nabla S = Q$. Both conditions are satisfied if the fluid is isentropic, i.e., the fluid has the same entropy everywhere. In the absence of any external force f, if f is zero for one instant at every point of the flow field then df/dt = Q. Therefore the vorticity at every point of the field will remain zero and the flow is irrotational. On the other hand, if the flow is not isentropic, i.e., the fluid has different entropy at different points of the field, the last term of Eq. (58) will cause the vorticity to be different from zero in the next instant. Therefore, nonisentropy flows cannot be irrotational. Hence irrotationality implies isentropy, but isentropy does not imply irrotationality.

Carrendor a minimum and addition of the contract of the contra

Multiplying Eq. (48) by ρ , expanding the term $(\rho\Omega\cdot\nabla)v$ by formula (X) on p. 44 of Ref. 17 and adding Eq. (32) (multiplied by Ω) we have the conservative form of the conservation equation for $\rho\Omega$

$$\frac{\partial}{\partial E} (\rho \underline{\Omega}) + \nabla \cdot (\rho \underline{v} \underline{\Omega}) = -\nabla \cdot (\rho \underline{\Omega} \underline{v}) + \underline{Q}_{1}$$
 (48a)

where

$$\underline{Q}_{1} = -\underline{v}[\nabla \cdot (\rho \underline{\Omega})] - \rho \underline{\Omega}(\nabla \cdot \underline{v}) - \rho \nabla (\frac{1}{\rho}) \times \nabla p + \rho \nabla (\frac{1}{\rho}) \times \underline{\epsilon} + \nabla \times \underline{\epsilon}$$

$$+ \mu \nabla^{2} \underline{\Omega} + \rho \nabla (\frac{\lambda + 2\mu}{\rho}) \times \nabla (\nabla \cdot \underline{v}) - \rho \nabla (\frac{\mu}{\rho}) \times \nabla \times \underline{\Omega}$$

$$+ \rho (\nabla \cdot \underline{v}) \nabla (\frac{1}{\rho}) \times \nabla \lambda - \nabla \lambda \times \nabla (\nabla \cdot \underline{v}) + 2\nabla \times (\underline{p} \cdot \nabla \mu)$$

$$+ \rho \nabla (\frac{2}{\rho}) \times (\underline{p} \cdot \nabla \mu)$$
(48b)

Taking the integral of Eq. (48a) over a fixed control volume V and applying Green's transformation Eq. (6) and Eqs. (6) and (7) on p. 53 of Ref. 17 we obtain the integral form of the conservation equation for $\rho\Omega$ in terms of fixed elements dV and dA

$$\frac{\partial}{\partial t} \int_{V} \rho \underline{n} dV + \int_{A} \rho \underline{n} (\underline{v} \cdot \underline{n}) dA$$

$$= \int_{A} \rho \underline{v} (\underline{n} \cdot \underline{n}) dA + \int_{V} \underline{Q}_{1} dV \qquad (48c)$$

The corresponding conservation equation in terms of moving elements $d\hat{V}$ and $d\hat{A}$ in case t_1 = t is Eq. (16) with

$$\underline{W}=(\rho\underline{\Omega}), \underline{\Sigma}=(\rho\underline{v}\ \underline{\Omega}), \underline{Q}=(\underline{Q}_{\underline{1}}). \tag{17c}$$

Combining Eqs. (17a), (17b) and (17c) with Eqs. (17) we convert Eq. (16) to a vector equation representing six conservation equations in integral form with

3.5 BERNOULLI EQUATION.

As discussed in Ref. 18, Section A,6, aside from the boundary layer, the effects of viscosity and heat conduction can be neglected for the majority of gas dynamics problems. Furthermore, the heat addition, except in problems involving combustion, is either zero or very small. Then under ordinary conditions the gas behaves very much like an ideal fluid (i.e., a nonviscous and nonheat-conducting fluid). Therefore, one of the fundamental problems of gas dynamics is to study the adiabatic flow of an ideal gas.

Let us further assume that the density of the fluid is a function of pressure only (i.e., the fluid is barotropic, Ref. 3, p. 150) and that the external force per unit mass is representable by a potential φ

$$\underline{\mathbf{f}}/\rho = - \nabla \boldsymbol{\varphi} \quad . \tag{60}$$

Then by using the relation (Ref. 17, p. 44)

$$(\underline{\mathbf{v}} \cdot \nabla) \underline{\mathbf{v}} = \frac{1}{2} \nabla \mathbf{v}^2 - \underline{\mathbf{v}} \times (\nabla \times \underline{\mathbf{v}})$$
 (61)

and definition of $\underline{\Omega}$ (Eq. (45)) Eq. (40) can be written as

$$\frac{\partial \underline{v}}{\partial t} - \underline{v} \times \underline{\Omega} = \frac{-1}{\rho} \nabla p - \nabla \left(\frac{1}{2} v^2\right) - \nabla \varphi = -\nabla \chi \quad , \tag{62}$$

where

$$X = \int \frac{\mathrm{d}p}{\rho} + \frac{1}{2} v^2 + \varphi \quad . \tag{63}$$

Let s be a parameter along an arbitrary curve in space. At any point (with position vector \underline{r}) along this curve $\underline{dr}/\underline{ds}$ is a unit tangent vector to the curve. The component of Eq. (§2) in the direction of $\underline{dr}/\underline{ds}$ is

$$\frac{\partial v}{\partial \overline{t}} \cdot \frac{d\underline{r}}{d\overline{s}} - (\underline{v} \times \underline{\Omega}) \cdot \frac{d\underline{r}}{d\overline{s}} = -\frac{d\underline{r}}{d\overline{s}} \cdot \nabla \chi \quad . \tag{64}$$

Since

$$\frac{d\mathbf{r}}{d\mathbf{s}} \cdot \sqrt[9]{\left(\frac{\partial \mathbf{v}}{\partial \mathbf{t}} \cdot \frac{d\mathbf{r}}{d\mathbf{s}}\right)} d\mathbf{s} = \frac{\partial \mathbf{v}}{\partial \mathbf{t}} \cdot \frac{d\mathbf{r}}{d\mathbf{s}}$$
 (65)

we may express Eq. (64) as (Ref.19, p. 121)

$$(\underline{v} \times \underline{\Omega}) \cdot \frac{d\underline{r}}{d\underline{s}} = \frac{d\underline{r}}{d\underline{s}} \cdot \nabla x \quad , \tag{66}$$

where

$$X = \int \frac{dp}{\rho} + \frac{1}{2} v^2 + \varphi + \int \left(\frac{\partial v}{\partial t} \cdot \frac{dr}{ds} \right) ds . \qquad (67)$$

At any point in time and at every point in the flow field let $d\mathbf{r}/ds$ represent a tangent vector to a streamline (in the direction of $\underline{\mathbf{v}}$) or a vortex line (in the direction of Ω).

Then the scalar triple product on the left side of Eq. (66) is zero and the Bernoulli's equation

$$X = a constant$$
 (68)

is valid along any streamline or vortex line but may have different constants for different streamlines and vortex lines.

For an irrotational flow $\underline{\Omega} = \underline{0}$ and there is a velocity potential ϕ :

$$\underline{\mathbf{v}} = \nabla \psi \qquad , \tag{69}$$

Eq. (68) is valid along every curve in the flow field since in Eq. (66) $\nabla X = 0$ everywhere. Furthermore, from Eqs. (62) and (69) we obtain the Bernoulli's equation for a potential flow

$$\int \frac{\mathrm{dp}}{\rho} + \frac{1}{2} \, \mathrm{v}^2 + \varphi + \frac{\partial \phi}{\partial t} = \mathrm{a \ constant} \ , \tag{70}$$

which holds throughout the flow field. The integral in Eq. (70) must be computed with isentropic pressure-density relation because as discussed in the previous section, irrotational motion can generally be maintained only by constant specific entropy throughout the field.

For a perfect gas with constant specific heats, the equation of state can be written in terms of S, p, ρ as [Ref. 18, Eq. (7-16)]

$$p = const \rho^{\gamma} e^{S/c_{\gamma}}$$
 (71)

Then Eq. (70) becomes [Ref. 18, Eq. (8-3)] for $\varphi = 0$ and S = const (isentropic)

$$\frac{\partial \phi}{\partial t} + \frac{1}{2} v^2 + \frac{\gamma}{\gamma - 1} \frac{p}{\rho} = \frac{\partial \phi}{\partial t} + H = a \text{ constant} . \tag{72}$$

Therefore irrotational flows are not necessarily isoenergetic in the sense that the total energy H is a constant throughout the field. Irrotational flows are isoenergetic only if they are steady, i.e., $\partial \phi / \partial t = 0$.

On the other hand for a rotational flow Eqs. (67) and (68) show that when the motion is steady (i.e., $\frac{\partial \mathbf{v}}{\partial t} = \underline{0}$) the total energy H is a constant along any streamline or vortex line [compare with the energy equation Eq. (37)]. The difference is that for steady (rotational) isoenergetic flow the motion is not necessarily isentropic, while for steady irrotational (isoenergetic) flow the motion must be isentropic. An example for the isoenergetic but nonisentropic flow is the problem of steady supersonic flow over a body with curved detached shock.

TRANSFORMATION OF CONSERVATION EQUATIONS IN TERMS OF ORTHOGONAL CURVILINEAR COORDINATES

The vectorial forms of the mass, momentum and energy conservation equations Eqs. (32), (33), and (34) can be expressed in a general curvilinear coordinate system by substituting the expressions for the gradient, divergence, and curl operators in such a system given, for example, in Reference 20, Appendix C. For practically useful orthogonal curvilinear coordinates x_1 , x_2 , x_3 , with local unit vectors \underline{e}_1 , \underline{e}_2 , \underline{e}_3 in the directions of increase of x_1 , x_2 , x_3 the specific expressions can be found in Ref. 21, Appendix 2.

An elementary displacement can be written as

$$d\underline{s} = h_1 dx_1 \underline{e}_1 + h_2 dx_2 \underline{e}_2 + h_3 dx_3 \underline{e}_3$$
, (73)

where h_1 , h_2 and h_3 are the scale factors. The velocity components are v_1 , v_2 and v_3 such that

$$\underline{\mathbf{v}} = \mathbf{v}_1 \underline{\mathbf{e}}_1 + \mathbf{v}_2 \underline{\mathbf{e}}_2 + \mathbf{v}_3 \underline{\mathbf{e}}_3$$
 (74)

Then Eqs. (32), (33) and (34) can be written as follows (Ref. 16, Section 1.2.2):

$$\frac{\text{mass}}{\partial t} : \frac{\partial \rho}{\partial t} + \frac{1}{h_1 h_2 h_3} \frac{\partial}{\partial x_i} \left(\frac{1}{h_i} h_1 h_2 h_3 \rho v_j \right) = 0 , \qquad (75)$$

momentum:
$$\frac{\partial}{\partial t} (\rho v_1) + \frac{1}{h_1 h_2 h_3} \frac{\partial}{\partial x_1} \left(\frac{1}{h_1} h_1 h_2 h_3 f_{1j} \right) +$$

$$+ \frac{1}{h_1 h_2} \left(\mathfrak{I}_{12} \frac{\partial h_1}{\partial x_2} - \mathfrak{I}_{22} \frac{\partial h_2}{\partial x_1} \right)$$

$$+ \frac{1}{h_1 h_3} \left(\mathfrak{I}_{13} \frac{\partial h_1}{\partial x_3} - \mathfrak{I}_{33} \frac{\partial h_3}{\partial x_1} \right) = \underline{\mathbf{f}} \cdot \underline{\mathbf{e}}_1 , \qquad (76)$$

and two morens by cyclic permutation

energy:
$$\frac{\partial}{\partial t} \frac{1}{h_1 h_2 h_3} \frac{\partial}{\partial x_j} \left\{ \frac{1}{h_j} h_1 h_2 h_3 \left[(\rho E + p) v_j \right] \right\}$$

$$\tau_{ij}v_{i}-k\frac{1}{h_{j}}\frac{\partial T}{\partial x_{j}}\bigg\} = \underline{f} \cdot \underline{v} + G , \qquad (77)$$

where the suconvention has been used and \mathfrak{I}_{ij} are the components omsor $\underline{\mathfrak{T}}$:

$$\rho \underline{v}\underline{v} - \underline{g} = \rho \underline{v}\underline{v} + \underline{p}\underline{I} - \underline{I} . \tag{78}$$

They can be ed as

$$_{i} = \rho v_{i}v_{j} + p\delta_{ij} - \tau_{ij}$$
 (79)

with

$$_{i} = \lambda \left(\nabla \cdot \underline{\mathbf{v}} \right) \delta_{\mathbf{i}\mathbf{j}} + 2\mu D_{\mathbf{i}\mathbf{j}} , \qquad (80)$$

where

$$\delta_{ij} = \begin{cases} 1 & (i = j) \\ 0 & (i + j) \end{cases}$$
 (81)

is the Kronecibol.

In Eq. (80)

$$\frac{1}{h_1 h_2 h_3} \frac{\partial}{\partial x_{\ell}} \left(\frac{1}{h_{\ell}} h_1 h_2 h_3 v_{\ell} \right) , \qquad (82)$$

$$\frac{1}{h_1} \left(\frac{\partial v_1}{\partial x_1} + \frac{v_2}{h_2} \frac{\partial h_1}{\partial x_2} + \frac{v_3}{h_3} \frac{\partial h_1}{\partial h_3} \right) , \qquad (83)$$

with \mathbf{D}_{22} and \mathbf{D}_{33} following by cyclic permutation and

$$D_{ij(i\neq j)} = \frac{1}{2} \left[\frac{1}{h_i} \frac{\partial v_j}{\partial x_i} + \frac{1}{h_j} \frac{\partial v_i}{\partial x_j} - \frac{1}{h_i h_j} \left(v_i \frac{\partial h_i}{\partial x_j} + v_j \frac{\partial h_j}{\partial x_i} \right) \right] . \quad (84)$$

CONSE EQUATIONS IN CARTESIAN COORDINATES

To the Carcoordinates

$$x_1 = x, x_2 = y, x_3 = z$$
 (85)

there corrthe scale factors

$$h_1 = h_2 = h_3 = 1$$
 . (86)

The conserequations Eqs. (75), (76) and (77) become when Eq. (used:

$$\underline{\text{mass}}: \quad \frac{\partial}{x_{j}} (\rho v_{j}) = 0 \quad , \tag{87}$$

$$\underline{momentum}: i) + \frac{\partial}{\partial x_{j}} (\rho v_{j} v_{j}) + \frac{\partial p}{\partial x_{i}} - \frac{\partial \tau_{ij}}{\partial x_{j}} = f_{i}, \qquad (88)$$

energy:) +
$$\frac{\partial}{\partial x_{j}} \left[(\rho E + p) v_{j} - \tau_{ij} v_{i} - k \frac{\partial T}{\partial x_{j}} \right] = f_{i} v_{i} + G$$
. (89)

where, sin Eqs. (82) to (84)

$$\nabla \cdot \underline{\mathbf{v}} = \frac{\partial \mathbf{v}_{\ell}}{\partial \mathbf{x}_{\ell}} \tag{90}$$

and

$$D_{ij} = \frac{1}{2} \left(\frac{\partial v_i}{\partial x_j} + \frac{\partial v_j}{\partial x_i} \right) , \qquad (91)$$

we have, fi(80),

$$= \lambda \delta_{ij} \frac{\partial v_{\ell}}{\partial x_{\ell}} + \mu \left(\frac{\partial v_{i}}{\partial x_{j}} + \frac{\partial v_{j}}{\partial x_{i}} \right) . \tag{92}$$

These resue with those usually found in the literature expressed sian tensor notations (see, for example, Ref. 9, E09)).

CONSERVATION EQUATIONS IN CYLINDRICAL COORDINATES

To the cylindrical coordinates

$$x_1 = r, x_2 = \theta, x_3 = z$$
 (93)

there correspond the scale factors (Reference 22, Appendix 4)

$$h_1 = 1, h_2 = r, h_3 = 1$$
 (94)

The conservation equations Eqs. (75), (76) and (77) become

$$\frac{\text{mass}}{\partial t}: \frac{\partial \rho}{\partial t} + \frac{1}{r} \frac{\partial}{\partial r} (r \rho v_r) + \frac{1}{r} \frac{\partial}{\partial \theta} (\rho v_\theta) + \frac{\partial}{\partial z} (\rho v_z) = 0, \quad (95)$$

$$\underline{\text{momentum}} \colon \frac{\partial}{\partial t} (\rho \mathbf{v_r}) + \frac{1}{r} \frac{\partial}{\partial r} (r \mathbf{s_{11}}) + \frac{1}{r} \frac{\partial}{\partial \theta} (\mathbf{s_{12}}) + \frac{\partial}{\partial z} (\mathbf{s_{13}})$$

$$-\frac{1}{r} \, \mathfrak{I}_{22} = f_{c} \quad , \tag{96}$$

$$\frac{\partial}{\partial t} (\rho v_{\theta}) + \frac{1}{r} \frac{\partial}{\partial r} (r \mathfrak{T}_{21}) + \frac{1}{r} \frac{\partial}{\partial \theta} (\mathfrak{T}_{22}) + \frac{\partial}{\partial z} (\mathfrak{T}_{23})$$

$$+\frac{1}{r} \, \mathbf{f}_{21} = \mathbf{f}_{\theta}$$
 (97)

$$\frac{\partial}{\partial t} (\rho v_2) + \frac{1}{r} \frac{\partial}{\partial r} (r \sigma_{31}) + \frac{1}{r} \frac{\partial}{\partial \theta} (\sigma_{32})$$

$$+\frac{\partial}{\partial z}(\mathfrak{I}_{33}) = f_z , \qquad (98)$$

energy:
$$\frac{\partial}{\partial t} (\rho E) + \frac{1}{r} \frac{\partial}{\partial r} \left\{ r \left[(\rho E + p) v_r - (\tau_{11} v_r + \tau_{21} v_\theta + \tau_{31} v_z) - k \frac{\partial T}{\partial r} \right] \right\}$$

$$+ \frac{1}{r} \frac{\partial}{\partial \theta} \left[(\rho E + p) v_{\theta} - (\tau_{12} v_{r} + \tau_{22} v_{\theta} + \tau_{32} v_{z}) \right]$$

$$- \frac{k}{r} \frac{\partial T}{\partial \theta}$$

$$+ \frac{\partial}{\partial z} \left[(\rho E + p) v_{z} - (\tau_{13} v_{r} + \tau_{23} v_{\theta} + \tau_{33} v_{z}) \right]$$

$$- k \frac{\partial T}{\partial z}$$

$$- k \frac{\partial T}{\partial z}$$

$$= f_{r} v_{r} + f_{\theta} v_{\theta} + f_{z} v_{z} + G , \qquad (99)$$

where \mathbf{f}_{ij} is given by Eq. (79) with τ_{ij} given by Eq. (80) in which, according to Eqs. (82) to (84),

$$\nabla \cdot \underline{\mathbf{v}} = \frac{1}{r} \frac{\partial}{\partial r} (\mathbf{r} \mathbf{v}_{\mathbf{r}}) + \frac{1}{r} \frac{\partial \mathbf{v}_{\theta}}{\partial \theta} + \frac{\partial \mathbf{v}_{\mathbf{z}}}{\partial \mathbf{z}} , \qquad (100)$$

and

$$D_{11} = \frac{\partial v_r}{\partial r} , D_{22} = \frac{1}{r} \left(\frac{\partial v_{\theta}}{\partial \theta} + v_r \right) , D_{33} = \frac{\partial v_z}{\partial z} ,$$

$$D_{12} = D_{21} = \frac{1}{2} \left(\frac{\partial v_{\theta}}{\partial r} - \frac{v_{\theta}}{r} + \frac{1}{r} \frac{\partial v_r}{\partial \theta} \right) ,$$

$$D_{23} = D_{32} = \frac{1}{2} \left(\frac{1}{r} \frac{\partial v_z}{\partial \theta} + \frac{\partial v_{\theta}}{\partial z} \right) ,$$

$$D_{31} = D_{13} = \frac{1}{2} \left(\frac{\partial v_r}{\partial z} + \frac{\partial v_z}{\partial z} \right) .$$
(101)

CONSERVATION EQUATIONS IN SPHERICAL COORDINATES

To the spherical coordinates

$$x_1 = r, \quad x_2 = \theta, \quad x_3 = \varphi \tag{102}$$

there correspond the scale factors (Reference 21, Appendix 2)

$$h_1 = 1$$
, $h_2 = r$, $h_3 = r \sin \theta$. (103)

The conservation equations Eqs. (75), (76) and (77) become

$$\frac{\partial \rho}{\partial t} + \frac{1}{r^2 \sin \theta} \left[\frac{\partial}{\partial r} (r^2 \sin \theta \rho v_r) + \frac{\partial}{\partial \theta} (r \sin \theta \rho v_\theta) \right]$$

$$+\frac{\partial}{\partial \varphi} (\mathbf{r} \rho \mathbf{v}_{\varphi}) = 0 , \qquad (104)$$

$$\underline{\text{momentum}} : \quad \frac{\Im}{\partial t} (\rho v_r) + \frac{1}{r^2 \sin \theta} \left[\frac{\partial}{\partial r} (r^2 \sin \theta S_{11}) + \frac{\partial}{\partial \theta} (r \sin \theta S_{12}) \right]$$

$$+\frac{\partial}{\partial \varphi} (r \mathbf{f}_{13}) - \frac{1}{r} \mathbf{f}_{22} - \frac{1}{r} \mathbf{f}_{33} = \mathbf{f}_{r}$$
, (105)

$$\frac{\partial}{\partial t} (\rho v_{\theta}) + \frac{1}{r^2 \sin \theta} \left[\frac{\partial}{\partial r} (r^2 \sin \theta \sigma_{21}) + \frac{\partial}{\partial \theta} (r \sin \theta \sigma_{22}) \right]$$

$$+\frac{\partial}{\partial \varphi} (rf_{23}) - \frac{\cos \theta}{r \sin \theta} f_{33} - \frac{1}{r} f_{21} = f_{\theta}, (106)$$

$$\frac{\partial}{\partial t} (\rho v_{\theta}) + \frac{1}{r^2 \sin \theta} \left[\frac{\partial}{\partial r} (r^2 \sin \theta \theta_{31}) + \frac{\partial}{\partial \theta} (r \sin \theta \theta_{32}) \right]$$

$$+\frac{\partial}{\partial \varphi} (r \mathfrak{F}_{33}) + \frac{1}{r} \mathfrak{F}_{31} + \frac{\cos \theta}{r \sin \theta} \mathfrak{F}_{32} = f_{\varphi}, (107)$$

energy:
$$\frac{\partial}{\partial t} (\rho E) + \frac{1}{r^2 \sin \theta} \frac{\partial}{\partial r} \left\{ r^2 \sin \theta \left[(\rho E + p) v_r - (\tau_{11} v_r + \tau_{21} v_\theta + \tau_{31} v_\varphi) \right] - k \frac{\partial T}{\partial r} \right\} + \frac{1}{r^2 \sin \theta} \frac{\partial}{\partial \theta} \left\{ r \sin \theta \left[(\rho E + p) v_\theta - (\tau_{12} v_r + \tau_{22} v_\theta + \tau_{32} v_\varphi) \right] - \frac{k}{r} \frac{\partial T}{\partial \theta} \right\} + \frac{1}{r^2 \sin \theta} \frac{\partial}{\partial \varphi} \left\{ r \left[(\rho E + p) v_\varphi - (\tau_{13} v_r + \tau_{23} v_\theta + \tau_{33} v_\varphi) \right] - \frac{k}{r \sin \theta} \frac{\partial T}{\partial \varphi} \right\} = f_r v_r + f_\theta v_\theta + f_\varphi v_\varphi + G , \quad (108)$$

where \mathfrak{T}_{ij} is given by Eq. (79) with τ_{ij} given by Eq. (80), in which according to Eqs. (82) to (84),

$$\nabla \cdot \underline{\mathbf{v}} = \frac{1}{\mathbf{r}^2 \sin \theta} \left[\frac{\partial}{\partial \mathbf{r}} \left(\mathbf{r}^2 \sin \theta \mathbf{v}_{\mathbf{r}} \right) + \frac{\partial}{\partial \theta} \left(\mathbf{r} \sin \theta \mathbf{v}_{\theta} \right) + \frac{\partial}{\partial \varphi} \left(\mathbf{r} \mathbf{v}_{\varphi} \right) \right] , \quad (109)$$

and

$$D_{11} = \frac{\partial v_r}{\partial r}, D_{22} = \frac{1}{r} \left(\frac{\partial v_{\theta}}{\partial \theta} + v_r \right), D_{33} = \frac{1}{r \sin \theta} \left(\frac{\partial v_{\theta}}{\partial \theta} + \sin \theta v_r + \cos \theta v_{\theta} \right),$$

$$D_{12} = D_{21} = \frac{1}{2} \left(\frac{\partial v_{\theta}}{\partial r} - \frac{v_{\theta}}{r} + \frac{1}{r} \frac{\partial v_r}{\partial \theta} \right),$$

$$D_{23} = D_{32} = \frac{1}{2r \sin \theta} \left(\sin \theta \frac{\partial v_{\theta}}{\partial \theta} + \frac{\partial v_{\theta}}{\partial \phi} - \cos \theta v_{\phi} \right),$$

$$D_{31} = D_{13} = \frac{1}{2r \sin \theta} \left(\frac{\partial v_r}{\partial \theta} + r \sin \theta \frac{\partial v_{\theta}}{\partial r} - \sin \theta v_{\phi} \right).$$

COMPACT ONE-DIMENSIONAL FORMULATION

The conservation equations in Sections 5, 6 and 7 for three-dimensional Cartesian, cylindrical and spherical problems respectively, can be readily reduced to those for various one-and two-dimensional problems by setting appropriate velocity components, force components and derivatives to zero. In particular, if we consider the one-dimensional problems in which

$$x_1 = r$$
, $\underline{v} = \underline{e}_1 v_r(r)$, $v_2 = v_3 = 0$, $\frac{\partial()}{\partial x_2} = \frac{\partial()}{\partial x_3} = 0$, $f_2 = f_3 = 0$, (111)

the conservation equations in Cartesian, cylindrical and spherical coordinates can be compactly expressed in one single set of equations (with j = 0, 1, 2 representing plane, line and point symmetry respectively):

$$\frac{\partial \rho}{\partial t} + \frac{1}{r^{j}} \frac{\partial}{\partial r} \left(r^{j} \rho v_{r} \right) = 0 , \qquad (112)$$

momentum:
$$\frac{\partial}{\partial t} (\rho v_r) + \frac{1}{r^{\frac{1}{2}}} \frac{\partial}{\partial r} \left(r^{\frac{1}{2}} \rho v_r^2 \right)$$

$$= f_r - \frac{\partial p}{\partial r} + \frac{\partial}{\partial r} \left[\frac{(\lambda + 2\mu)}{r^{\frac{1}{2}}} \frac{\partial (r^{\frac{1}{2}} v_r)}{\partial r} \right] - \frac{2j}{r} v_r \frac{\partial \mu}{\partial r} , \quad (113)$$

energy:
$$\frac{\partial}{\partial t} (\rho E) + \frac{1}{r^{j}} \frac{\partial}{\partial r} \left[r^{j} v_{r} (\rho E + p) \right]$$

$$= f_{r} v_{r} + \frac{1}{r^{j}} \frac{\partial}{\partial r} \left[r^{j} v_{r} \left\{ \frac{\lambda}{r^{j}} \frac{\partial}{\partial r} \left(r^{j} v_{r} \right) + 2\mu \frac{\partial v_{r}}{\partial r} \right\} \right]$$

$$+ \frac{1}{r^{j}} \frac{\partial}{\partial r} \left(kr^{j} \frac{\partial T}{\partial r} \right) + G . \qquad (114)$$

CONCLUSIONS

Following a survey of the basic equations, the mass, momentum and energy conservation equations and the components of the deformation and stress tensors for a nonsteady flow of a compressible viscous single-phase fluid have been expressed in vector and dyadic notations, transformed in terms of orthogonal curvilinear coordinates and specialized for the three-dimensional cases in Cartesian, cylindrical and spherical coordinates. These equations can then be reduced to corresponding equations for one- and two-dimensional flows in various coordinates. They are in a format readily applicable to practical problems.

If desired the procedure can be extended to general nonorthogonal curvilinear coordinate systems.

This report also contains a generalization of the conservation equations in integral form based on an extension of the Reynolds transport theorem.

LIST OF REFERENCES

- Truesdell, C., The Kinematics of Vorticity, Indiana University Press, Bloomington, 1954.
- 2. Euler, C., Principes généraux du mouvement des fluides, Hist. Acad. Berlin 1955, pp. 274-315.
- 3. Serrin, J., "Mathematical Principles of Classical Fluid Mechanics," Encyclopedia of Physics, S. Flügge (ed.), Vol. VIII/1: Fluid Dynamics I, C. Truesdell (co-editor), Springer-Verlag, Berlin, 1959, pp. 125-263.
- 4. Reynolds, O., The Sub-Mechanics of the Universe, Vol. III of "Papers on Mechanical and Physical Subjects," Cambridge University Press, London, 1903.
- 5. Boris, J. P., "Flux-Corrected Transport Modules for Generalized Continuity Equations," NRL Memorandum Report 3237, Naval Research Laboratory, Washington, D.C., 1976. (Also see Finite-Difference Techniques for Vectorized Fluid Dynamics Calculations edited by D. L. Book, Springer-Verlag, New York, N.Y., 1981, pp. 33-40.)
- 6. Schlichting, H., Boundary Layer Theory, 7th Edition, (translated by J. Kestin), McGraw-Hill Book Company, New York, N.Y., 1979.
- 7. Owczarek, J. A., Fundamentals of Gas Dynamics, International Textbook Co., Scranton, Pennsylvania, 1964.
- 8. Shapiro, A. H., The Dynamics and Thermodynamics of Compressible Fluid Flow, Vols. I and II, Ronald Press, New York, N.Y., 1953.
- 9. Liepmann, H. W. and Roshko, A., Elements of Gas Dynamics, John Wiley & Sons, Inc., New York, N.Y., 1957.
- 10. Pai, S. I., <u>Viscous Flow Theory</u>, (I-Laminar Flow, 1956 II-Turbulent Flow, 1957), D. van Nostrand Company, Inc., Princeton, N.J.
- 11. von Mises, R., Mathematical Theory of Compressible Fluid Flow, (completed by H. Geiringer and G. S. S. Ludford), Academic Press, Inc., New York, N.Y. 1958.
- 12. Courant, R. and Friedricks, K. O., <u>Supersonic Flow and Shock Waves</u>, Interscience Publishers, Inc., New York, N.Y., 1948.

- 13. Chapman, A. J. and Walker, W. F., <u>Introductory Gas</u>
 Dynamics, Holt, Rinehart and Winston, New York, N.Y., 1971.
- 14. Roache, P. J., Computational Fluid Dynamics, Hermosa Publishers, Albuquerque, New Mexico, 1972.
- 15. Potter, D., Computational Physics, John Wiley & Sons, New York, N.Y., 1973.
- 16. Peyret, R. and Taylor, T. D., Computational Methods in Fluid Flow, Springer-Verlag, New York, N.Y., 1983.
- 17. Milne-Thomson, L. M., Theoretical Hydrodynamics, 4th ed., The Macmillan Company, New York, N.Y., 1960
- 18. Tsien, H. S., "The Equations of Gas Dynamics," Fundamental of Gas Dynamics, H. W. Emmons (ed.), Vol. III: High Speed Aerodynamics and Jet Propulsion, Princeton University Press, Princeton, N.J., 1958.
- 19. Slattery, J. C., Momentum, Energy and Mass Transfer in Continua, McGraw-Hill Book Company, New York, N.Y., 1972.
- 20. Eringen, A. C., Mechanics of Continua, John Wiley & Sons, Inc., New York, N.Y., 1967.
- 21. Batchelor, G. K., An Introduction to Fluid Dynamics Cambridge University Press, London, 1967.
- 22. Lebedev, N. N., Skalskaya, I. P., and Uflyand, Y. S.,
 Problems of Mathematical Physics, R. A. Silverman (translator and editor), Prentice-Hall, Inc., Englewood Cliffs,
 N.J., 1965.

DISTRIBUTION LIST

DEPARTMENT OF DEFENSE

AIR FORCE SOUTH
ATTN: U S DOC OFFICER

ASST TO THE SECY OF DEFENSE ATOMIC ENERGY ATYN: EXECUTIVE ASSISTANT

DEFENSE INTELLIGENCE AGENCY

ATTN: DT-2 ATTN: RTS ATTN: RTS-2B

DEFENSE NUCLEAR AGENCY

ATTN: SPAS ATTN: STSP 4 CYS ATTN: STTI-CA

DEFENSE TECHNICAL INFORMATION CENTER

12 CYS ATTN: DD

FIELD COMMAND DNA DET 2 LAWRENCE LIVERMORE NATIONAL LAB ATTN: FC-1

FIELD COMMAND DEFENSE NUCLEAR AGENCY

ATTN: FCPR
ATTN: FCT COL J MITCHELL
ATTN: FCTT W SUMMA
ATTN: FCTXE

JOINT CHIEFS OF STAFF
ATTN: J-5 NUC & CHEM DIV

JOINT STRAT TGT PLANNING STAFF ATTN: JLK (ATTN: DNA REP)

ATTN: JLKS ATTN: JPPFM ATTN: JPTP

UNDER SECY OF DEF FOR RSCH & ENGRG

ATTN: STRAT & SPACE SYS (OS)

ATTN: STRAT & THR NUC FOR J THOMPSON

ATTN: STRAT & THEATER NUC FORCES

DEPARTMENT OF THE ARMY

DEP CH OF STAFF FOR OPS & PLANS ATTN: DAMO-NCZ

HARRY DIAMOND LABORATORIES ATTN: SLCHD-NW-RH

U S ARMY MATERIAL COMMAND ATTN: AMCCN

U S ARMY MATERIAL TECHNOLOGY LABORATORY ATTN: DRXMR-HH

U S ARMY STRATEGIC DEFENSE CMD ATTN: DACS-BM/TECHNOLOGY DIV

U S ARMY STRATEGIC DEFENSE COMMAND
ATTN: ATC-D WATTS
ATTN: ATC-R ANDREWS

DEPARTMENT OF THE NAVY

NAVAL RESEARCH LABORATORY
ATTN: CODE 2627 TECH LIB
ATTN: CODE 4040 D BOOK

NAVAL SEA SYSTEMS COMMAND ATTN: SEA-0351

NAVAL SURFACE WEAPONS CENTER ATTN: CODE K82 ATTN: CODE R44 H GLAZ

OFC OF THE DEPUTY CHIEF OF NAVAL OPS ATTN: NOP 654 STRAT EVAL & ANAL BR

STRATEGIC SYSTEMS PROGRAMS(PM-1)
ATTN: SP-272

DEPARTMENT (IF THE AIR FORCE

AIR FORCE SYSTEMS COMMAND ATTN: DLW

AIR FORCE WEAPONS LABORATORY, AFSC

ATTN: NTED J RENICK ATTN: NTED LT KITCH ATTN: NTED R HENNY ATTN: NTEDA ATTN: NTES ATTN: SUL

IR FORCE WRIGHT AERONAUTICAL LAB

ATTN: FIBC ATTN: FIMG

AIR FORCE WRIGHT AERONAUTICAL LAB

ATTN: AFWAL/MLP ATTN: AFWAL/ML/M-

DEPARTMENT OF THE AIR FORCE (CONTINUED)

AIR UNIVERSITY LIBRARY
ATTN: AUL-LSE

BALLISTIC MISSILE OFFICE/DAA

ATTN: CAPT T KING MGEN
ATTN: CC MAJ GEN CASEY

ATTN: ENSR

DEPUTY CHIEF OF STAFF/AF-RDQI

ATTN: AF/RDQI

DEPUTY CHIEF OF STAFF/AFRDS

ATTN: AFRDS SPACE SYS & C3 DIR

STRATEGIC AIR COMMAND/NRI-STINFO

ATTN: NRI/STINFO

STRATEGIC AIR COMMAND/XPQ

ATTN: XPQ

161 ARG ARIZONA ANG

ATTN: LTCOL SHERER

DEPARTMENT OF ENERGY

UNIVERSITY OF CALIFORNIA
LAWRENCE LIVERMORE NATIONAL LAB

ATTN: D BURTON

ATTN: L-10 J CAROTHERS

ATTN: L-122 G GOUDREAU ATTN: L-122 S SACKETT

ATTN: L-203 T BUTKOVICH

ATTN: L-22 D CLARK

ATTN: L-8 P CHRZANOWSKI

ATTN: L-84 H KRUGER

LOS ALAMOS NATIONAL LABORATORY

ATTN: A112 MS R SELDEN

ATTN: M T SANDFORD ATTN: R WHITAKER

SANDIA NATIONAL LABORATORIES

ATTN: DJ RIGALI

ATTN: ORG 7112 A CHABAI

ATTN: R G CLEM

OTHER COVERNMENT

CENTRAL INTELLIGENCE AGENCY

ATTN: OSWR/NED

DEPARTMENT OF THE INTERIOR

ATTN: D RODDY

DEPARTMENT OF DEFENSE CONTRACTORS

ACUREX CORP

ATTN: C WOLF

AEROSPACE CORP

ATTN: H MIRELS

APPLIED RESEARCH ASSOCIATES, INC.

ATTN: N HIGGINS

APPLIED RESEARCH ASSOCIATES, INC

ATTN: S BLOUIN

APPLIED RESEARCH ASSOCIATES, INC

ATTN: D PIEPENBURG

BELL AEROSPACE TEXTRON

ATTN: C TILYOU

CALIFORNIA RESEARCH & TECHNOLOGY, INC.

ATTN: K KREYENHAGEN

ATTN: M ROSENBLATT

CALIFORNIA RESEARCH & TECHNOLOGY, INC.

ATTN: F SAUER

H-TECH LABS, INC

ATTN: B HARTENBAUM

INFORMATION SCIENCE, INC.

ATTN: W DUDZIAK

KAMAN SCIENCES CORP

ATTN: L MENTE

ATTN: R RUETENIK

ATTN: WLEE

MAXWELL LABORATORIES, INC

ATTN: J MURPHY

MERRITT CASES, INC

ATTN: J MERRITT

NEW MEXICO ENGINEERING RESEARCH INSTITUTE

ATTN: G LEIGH

R & D ASSOCIATES

2 CYS ATTN: A KUHL

ATTN: CKBLEE

2 CYS ATTN: G YEN

ATTN: J LEWIS

ATTN: PRAUSCH

R & D ASSOCIATES

ATTN: PMOSTELLER

S-CUBED

ATTN: A WILSON

S-CUBED

ATTN: C NEEDHAM

DEFT OF DEFENSE CONTRACTORS (CONTINUED)

SCIENCE APPLICATIONS INTL CORP ATTN: H WILSON

SCIENCE APPLICATIONS INTL CORP ATTN: J COCKAYNE ATTN: W LAYSON

SCIENCE APPLICATIONS INTL CORP ATTN: A MARTELLUCCI SCIENCE APPLICATIONS INTL CORP ATTN: G BINNINGER

TRW ELECTRONICS & DEFENSE SECTOR
· ATTN: G HULCHER
ATTN: P DAI

WEIDLINGER ASSOC, CONSULTING ENGRG ATTN: P WEIDLINGER