Modern Optimal Control

Matthew M. Peet Arizona State University

Lecture 22: H_2 , LQG and LGR

Conclusion

To solve the H_{∞} -optimal state-feedback problem, we solve

$$\begin{aligned} & \min_{\gamma, X_1, Y_1, A_n, B_n, C_n, D_n} \gamma & \text{ such that} \\ & \begin{bmatrix} X_1 & I \\ I & Y_1 \end{bmatrix} > 0 \\ & \begin{bmatrix} AY_1 + Y_1 A^T + B_2 C_n + C_n^T B_2^T & *^T & *^T & *^T \\ A^T + A_n + [B_2 D_n C_2]^T & X_1 A + A^T X_1 + B_n C_2 + C_2^T B_n^T & *^T & *^T \\ & [B_1 + B_2 D_n D_{21}]^T & [X B_1 + B_n D_{21}]^T & -\gamma I \\ & C_1 Y_1 + D_{12} C_n & C_1 + D_{12} D_n C_2 & D_{11} + D_{12} D_n D_{21} & -\gamma I \end{bmatrix} < 0 \end{aligned}$$

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Conclusion

Then, we construct our controller using

$$\begin{split} D_K &= (I + D_{K2}D_{22})^{-1}D_{K2} \\ B_K &= B_{K2}(I - D_{22}D_K) \\ C_K &= (I - D_KD_{22})C_{K2} \\ A_K &= A_{K2} - B_K(I - D_{22}D_K)^{-1}D_{22}C_K. \end{split}$$

where

$$\begin{bmatrix} A_{K2} & B_{K2} \\ \hline C_{K2} & D_{K2} \end{bmatrix} = \begin{bmatrix} X_2 & X_1 B_2 \\ 0 & I \end{bmatrix}^{-1} \begin{bmatrix} \begin{bmatrix} A_n & B_n \\ C_n & D_n \end{bmatrix} - \begin{bmatrix} X_1 A Y_1 & 0 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} Y_2^T & 0 \\ C_2 Y_1 & I \end{bmatrix}^{-1}.$$

and where X_2 and Y_2 are any matrices which satisfy $X_2Y_2 = I - X_1Y_1$.

- e.g. Let $Y_2 = I$ and $X_2 = I X_1 Y_1$.
- The optimal controller is NOT uniquely defined.
- Don't forget to check invertibility of $I-D_{22}D_K$

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Conclusion

The H_{∞} -optimal controller is a dynamic system.

• Transfer Function $\hat{K}(s) = \begin{bmatrix} A_K & B_K \\ \hline C_K & D_K \end{bmatrix}$

Minimizes the effect of external input (w) on external output (z).

$$\|z\|_{L_2} \leq \|\underline{\mathsf{S}}(P,K)\|_{H_\infty} \|w\|_{L_2}$$

Minimum Energy Gain

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Motivation

 H_2 -optimal control minimizes the H_2 -norm of the transfer function.

• The H_2 -norm has no direct interpretation.

$$\|G\|_{H_2}^2 = \frac{1}{2\pi} \int_{-\infty}^{\infty} \mathrm{Trace}(\hat{G}^*(\imath \omega) \hat{G}(\imath \omega)) d\omega$$

Motivation: Assume external input is Gaussian noise with spectral density S_w

$$E[w(t)^{2}] = \frac{1}{2\pi} \int_{-\infty}^{\infty} \operatorname{Trace}(\hat{S}_{w}(\imath \omega)) d\omega$$

Theorem 1.

For an LTI system P, if w is noise with spectral density $\hat{S}_w(\imath\omega)$ and z=Pw, then z is noise with density

$$\hat{S}_z(i\omega) = \hat{P}(i\omega)\hat{S}(i\omega)\hat{P}(i\omega)^*$$

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Motivation

Then the output z = Pw has signal variance (Power)

$$\begin{split} E[z(t)^2] &= \frac{1}{2\pi} \int_{-\infty}^{\infty} \mathrm{Trace}(\hat{G}^*(\imath \omega) S(\imath \omega) \hat{G}(\imath \omega)) d\omega \\ &\leq \|S\|_{H_{\infty}} \|G\|_{H_2}^2 \end{split}$$

If the input signal is white noise, then $\hat{S}(\imath\omega)=I$ and

$$E[z(t)^2] = ||G||_{H_2}^2$$

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Colored Noise

If the noise is colored, then we can still use the same approach to H_2 optimal control. Let $\hat{H}(\imath\omega)\hat{H}(\imath\omega)^* = \hat{S}_w(\imath\omega)$. Then design an H_2 -optimal controller for the plant $\hat{P}_{\omega}(\imath\omega)\hat{H}(\imath\omega) = \hat{P}_{\omega}(\imath\omega) \hat{P$

 $\hat{P}_s(s) = \begin{bmatrix} \hat{P}_{11}(s)\hat{H}(s) & \hat{P}_{12}(s) \\ \hat{P}_{21}(s)\hat{H}(s) & \hat{P}_{22}(s) \end{bmatrix}$

Now, using the controller and filtered plant,

$$\underline{S}(P_s, K)(s) = P_{11}H + P_{12}(I - KP_{22})^{-1}KP_{21}H = \underline{S}(P, K)H$$

For noise with spectral density S_w , let $z = \underline{S}(P,K)w$. Then if S_z is the spectral density of z, we have

$$S_z(s) = \underline{S}(P, K)(s)\hat{S}_w(s)\underline{S}(P, K)(s)^*$$

= $\underline{S}(P, K)(s)\hat{H}(s)\hat{H}(s)^*\underline{S}(P, K)(s)^* = \underline{S}(P_s, K)(s)\hat{S}(P_s, K)(s)^*$

and hence $E[z(t)^2] = \|\hat{S}(P_s,K)\|_{H_2}^2 = \gamma$. Thus minimization of $\|\hat{S}(P_s,K)\|_{H_2}^2$ achieves the optimal system response to noise with density \hat{S}_w .

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Theorem 2.

Suppose $\hat{P}(s) = C(sI - A)^{-1}B$. Then the following are equivalent.

- 1. A is Hurwitz and $\|\hat{P}\|_{H_2} < \gamma$.
- 2. There exists some X > 0 such that

$$\label{eq:condition} \operatorname{trace} CXC^T < \gamma$$

$$AX + XA^T + BB^T < 0$$

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Proof.

Suppose A is Hurwitz and $\|\hat{P}\|_{H_2} < \gamma$. Then the Controllability Grammian is defined as

$$X_c = \int_0^\infty e^{At} B B^T e^{A^T} dt$$

Now recall the Laplace transform

$$(\Lambda e^{At})(s) = \int_0^\infty e^{At} e^{-ts} dt$$

$$= \int_0^\infty e^{-(sI-A)t} dt$$

$$= -(sI-A)^{-1} e^{-(sI-A)t} dt \Big|_{t=0}^{t=-\infty}$$

$$= (sI-A)^{-1}$$

Hence $(\Lambda Ce^{At}B)(s) = C(sI - A)^{-1}B$.

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Proof.

$$\begin{split} \left(\Lambda C e^{At} B\right)(s) &= C(sI-A)^{-1} B \text{ implies} \\ \|\hat{P}\|_{H_2}^2 &= \|C(sI-A)^{-1} B\|_{H_2}^2 \\ &= \frac{1}{2\pi} \int_0^\infty \operatorname{Trace}((C(\imath \omega I-A)^{-1} B)^* (C(\imath \omega I-A)^{-1} B)) d\omega \\ &= \frac{1}{2\pi} \int_0^\infty \operatorname{Trace}((C(\imath \omega I-A)^{-1} B)(C(\imath \omega I-A)^{-1} B)^*) d\omega \\ &= \operatorname{Trace} \int_{-\infty}^\infty C e^{At} B B^* e^{A^* t} C^* dt \\ &= \operatorname{Trace} C X_c C^T \end{split}$$

Thus
$$X_c \geq 0$$
 and $\text{Trace}CX_cC^T = \|\hat{P}\|_{H_2}^2 < \gamma$.

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Proof.

Likewise ${\rm Trace} B^T X_o B = \|\hat{P}\|_{H_2}^2.$ To show that we can take strict the inequality X>0, we simply let

$$X = \int_0^\infty e^{At} \left(BB^T + \epsilon I \right) e^{A^T} dt$$

for sufficiently small $\epsilon>0$. Furthermore, we already know the controllability grammian X_c and thus X_ϵ satisfies the Lyapunov inequality.

$$A^T X_{\epsilon} + X_{\epsilon} A + B B^T < 0$$

These steps can be reversed to obtain necessity.

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Full-State Feedback

Lets consider the full-state feedback problem

$$\hat{G}(s) = \begin{bmatrix} A & B_1 & B_2 \\ \hline C_1 & 0 & D_{12} \\ I & 0 & 0 \end{bmatrix}$$

- D_{12} is the weight on control effort.
- $D_{11} = 0$ is neglected as the feed-through term.
- $C_2 = I$ as this is state-feedback.

$$\hat{K}(s) = \begin{bmatrix} 0 & 0 \\ \hline 0 & K \end{bmatrix}$$

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Theorem 3.

The following are equivalent.

- 1. $||S(K,P)||_{H_2} < \gamma$.
- 2. $K = ZX^{-1}$ for some Z and X > 0 where

$$\begin{bmatrix} A & B_2 \end{bmatrix} \begin{bmatrix} X \\ Z \end{bmatrix} + \begin{bmatrix} X & Z^T \end{bmatrix} \begin{bmatrix} A^T \\ B_2^T \end{bmatrix} + B_1 B_1^T < 0$$

$$\textit{Trace} \left[C_1 X + D_{12} Z \right] X^{-1} \left[C_1 X + D_{12} Z \right] < \gamma$$

However, this is nonlinear, so we need to reformulate using the Schur Complement.

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Applying the Schur Complement gives the alternative formulation convenient for control.

Theorem 4.

Suppose $\hat{P}(s) = C(sI - A)^{-1}B$. Then the following are equivalent.

- 1. A is Hurwitz and $\|\hat{P}\|_{H_2} < \gamma$.
- 2. There exists some X, Z > 0 such that

$$\begin{bmatrix} A^TX + XA & XB \\ B^TX & -\gamma I \end{bmatrix} < 0, \qquad \begin{bmatrix} X & C^T \\ C & Z \end{bmatrix} > 0, \qquad \textit{Trace} Z < \gamma$$

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Theorem 5.

The following are equivalent.

- 1. $||S(K,P)||_{H_2} < \gamma$.
- 2. $K = ZX^{-1}$ for some Z and X > 0 where

$$\begin{split} \left[A \quad B_2 \right] \begin{bmatrix} X \\ Z \end{bmatrix} + \left[X \quad Z^T \right] \begin{bmatrix} A^T \\ B_2^T \end{bmatrix} + B_1 B_1^T < 0 \\ \begin{bmatrix} X & (C_1 X + D_{12} Z)^T \\ C_1 X + D_{12} Z & W \end{bmatrix} > 0 \\ \end{split}$$

$$TraceW < \gamma$$

Thus we can solve the H_2 -optimal static full-state feedback problem.

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Relationship to LQR

Thus minimizing the H_2 -norm minimizes the effect of white noise on the power of the output noise.

• This is why H_2 control is often called Least-Quadratic-Gaussian (LQG).

LQR:

- Full-State Feedback
- Choose K to minimize the cost function

$$\int_0^\infty x(t)^T Qx(t) + u(t)^T Ru(t) dt$$

subject to dynamic constraints

$$\dot{x}(t) = Ax(t) + Bu(t)$$

$$u(t) = Kx(t), \qquad x(0) = x_0$$

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Relationship to LQR

To solve the LQR problem, let

•
$$C_1 = \begin{bmatrix} Q^{\frac{1}{2}} \\ 0 \end{bmatrix}$$

$$D_{12} = \begin{bmatrix} 0 \\ R^{\frac{1}{2}} \end{bmatrix}$$

• $B_2 = B$ and $B_1 = I$.

So that

$$\underline{S}(\hat{P}, \hat{K}) = \begin{bmatrix} A + B_2 K & B_1 \\ \hline C_1 + D_{12} K & D_{11} \end{bmatrix} = \begin{bmatrix} A + B K & I \\ \hline Q^{\frac{1}{2}} & 0 \\ \hline R^{\frac{1}{2}} K & 0 \end{bmatrix}$$

And solve the H_2 full-state feedback problem. Then if

$$\dot{x}(t) = A_{CL}x(t) = (A + BK)x(t) = Ax(t) + Bu(t)$$

 $u(t) = Kx(t), \qquad x(0) = x_0$

Then
$$x(t) = e^{A_{CL}t}x_0$$

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Relationship to LQR

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$$\dot{x}(t) = A_{CL}x(t) = (A + BK)x(t) = Ax(t) + Bu(t)$$

 $u(t) = Kx(t), \qquad x(0) = x_0$

then $x(t) = e^{A_{CL}t}x_0$ and

$$\begin{split} &\int_{0}^{\infty} x(t)^{T}Qx(t) + u(t)^{T}Ru(t)dt = \int_{0}^{\infty} x_{0}^{T}e^{A_{CL}^{T}t}(Q + K^{T}RK)e^{A_{CL}t}x_{0}dt \\ &= \operatorname{Trace} \int_{0}^{\infty} x_{0}^{T}e^{A_{CL}^{T}t} \begin{bmatrix} Q^{\frac{1}{2}} \\ R^{\frac{1}{2}}K \end{bmatrix}^{T} \begin{bmatrix} Q^{\frac{1}{2}} \\ R^{\frac{1}{2}}K \end{bmatrix} e^{A_{CL}t}x_{0}dt \\ &= \|x_{0}\|^{2}\operatorname{Trace} \int_{0}^{\infty} B_{1}e^{A_{CL}^{T}t}(C_{1} + D_{12}K)^{T}(C_{1} + D_{12}K)e^{A_{CL}t}B_{1}^{T}dt \\ &= \|x_{0}\|^{2}\|S(K, P)\|_{H_{2}}^{2} \end{split}$$

Thus LQR reduces to a special case of H_2 static state-feedback.

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H_2 -optimal output feedback control

Theorem 6 (Lall).

The following are equivalent.

- There exists a $\hat{K}=\begin{bmatrix} A_K & B_K \\ C_K & D_K \end{bmatrix}$ such that $\|S(K,P)\|_{H_2}<\gamma$.
- There exist $X_1,Y_1,Z,A_n,B_n,C_n,D_n$ such that $\begin{bmatrix} X_1 & I \\ I & Y_1 \end{bmatrix}>0$

$$\begin{bmatrix} AY_1 + Y_1A^T + B_2C_n + C_n^TB_2^T & *^T & *^T \\ A^T + A_n + [B_2D_nC_2]^T & X_1A + A^TX_1 + B_nC_2 + C_2^TB_n^T & *^T \\ [B_1 + B_2D_nD_{21}]^T & [XB_1 + B_nD_{21}]^T & -\gamma I \end{bmatrix} < 0,$$

$$\begin{bmatrix} X_1 & I & *^T \\ I & Y_1 & *^T \\ C_1Y_1 + D_{12}C_n & C_1 + D_{12}D_nC_2 & Z \end{bmatrix} > 0,$$

$$D_{11} + D_{12}D_nD_{21} = 0, \quad \operatorname{trace}(Z) < \gamma$$

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H_2 -optimal output feedback control

As before, the controller can be recovered as

$$\begin{bmatrix} A_{K2} & B_{K2} \\ \hline C_{K2} & D_{K2} \end{bmatrix} = \begin{bmatrix} X_2 & X_1B_2 \\ 0 & I \end{bmatrix}^{-1} \begin{bmatrix} \begin{bmatrix} A_n & B_n \\ C_n & D_n \end{bmatrix} - \begin{bmatrix} X_1AY_1 & 0 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} Y_2^T & 0 \\ C_2Y_1 & I \end{bmatrix}^{-1}$$

for any full-rank X_2 and Y_2 such that

$$\begin{bmatrix} X_1 & X_2 \\ X_2^T & X_3 \end{bmatrix} = \begin{bmatrix} Y_1 & Y_2 \\ Y_2^T & Y_3 \end{bmatrix}^{-1}$$

To find the actual controller, we use the identities:

$$D_K = (I + D_{K2}D_{22})^{-1}D_{K2}$$

$$B_K = B_{K2}(I - D_{22}D_K)$$

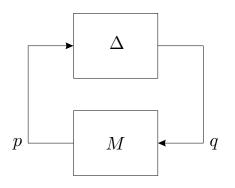
$$C_K = (I - D_KD_{22})C_{K2}$$

$$A_K = A_{K2} - B_K(I - D_{22}D_K)^{-1}D_{22}C_K$$

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Robust Control

Before we finish, let us briefly touch on the use of LMIs in Robust Control.



Questions:

- Is $\underline{\mathsf{S}}(\Delta,M)$ stable for all $\Delta\in\mathbf{\Delta}$?
- Determine

$$\sup_{\Delta \in \mathbf{\Delta}} \|\underline{\mathbf{S}}(\Delta, M)\|_{H_{\infty}}.$$

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