

# **AEEM 3042 – Aircraft Performance & Design**

## **Aircraft Weights**

# ***Aircraft Weights & CG***

**Why do we worry about the weight of the aircraft?**

**Sizes the aircraft landing gear**

**Runway restrictions**

**Heavier weight = less performance capability**

**Longer takeoff distance**

**Less rate of climb**

**Less cruise efficiency**

**Lower maneuver capability**

**Higher approach speeds**

**Weight  $\cong$  Acquisition Cost**

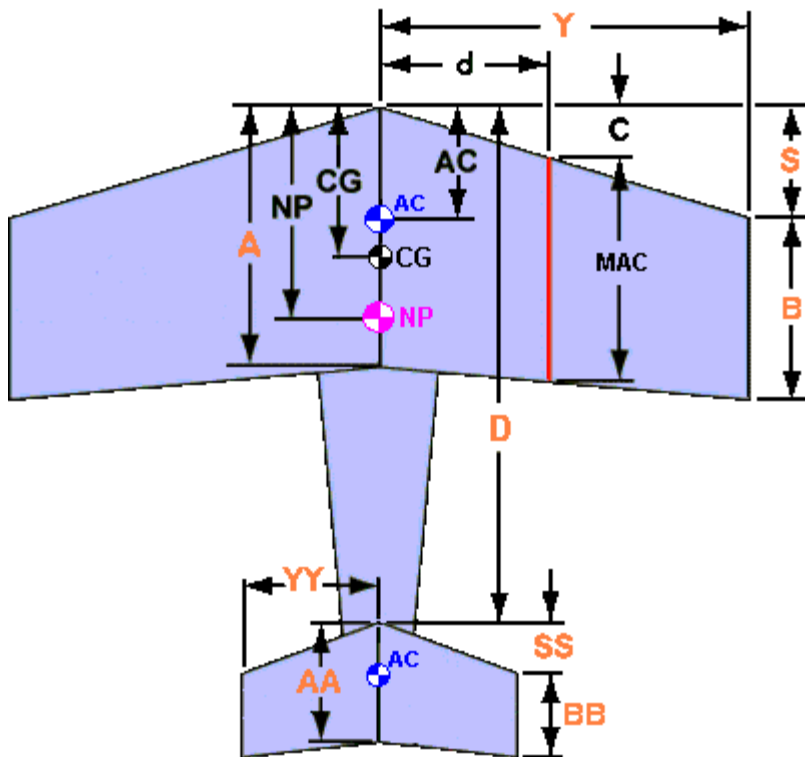
# Aircraft Weights & CG

**Why do we worry about the aircraft's center of gravity?**

**The CG needs to be the right range for stable flight**

**Want the CG near the wing's aerodynamic center**

**Want the CG in front of the Neutral Point**



# ***Aircraft Weights & CG***

**Why do we worry about the aircraft's center of gravity?**

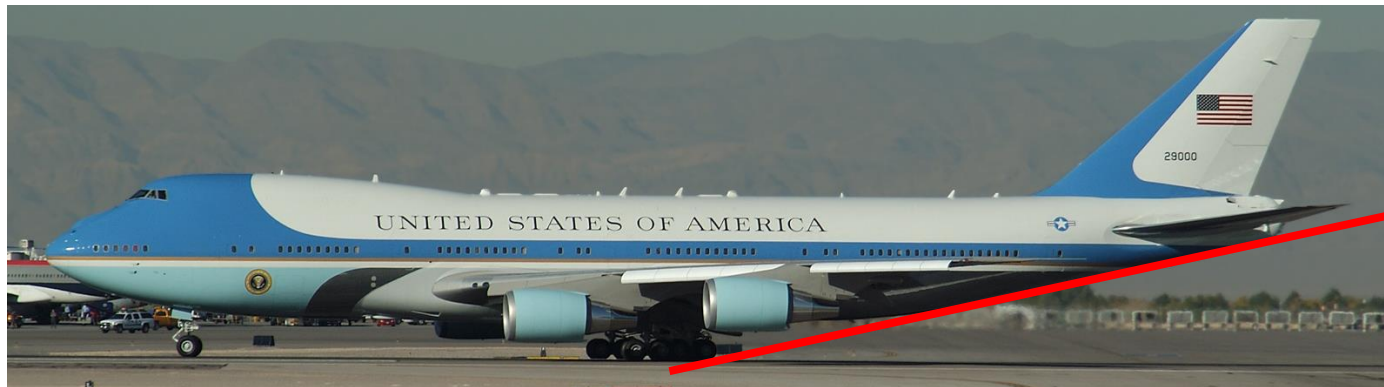
**The CG needs to be the right range on the ground**

**Want the CG between the nose & main landing gear**

**Want the CG just in front of the main landing gear**

**Weight on nose gear  $\cong$  10% to 15%**

**Weight on main landing gear  $\cong$  90% to 85%**



**Tail  
Bump  
Angle**

**10%**  
**CG**  
**90%**

# ***Aircraft Weights & CG***

**Why do we worry about the aircraft's center of gravity?**



# ***Aircraft Weights & CG***

**Society of Allied Weight Engineers**



**Aircraft weight and balance  
Mass properties management  
Weight estimating  
Weight control**

# ***Aircraft Weights***

## **Structure**

***wings***  
***tails***

***fuselage***  
***landing gear***

***engine mounts***  
***air induction***

## **+ Propulsion**

***engines***  
***exhaust***

***fuel tanks***  
***starters***

***engine cooling***  
***engine controls***

## **+ Equipment**

***hydraulics***  
***electrical***  
***instruments***

***avionics***  
***pneumatics***  
***armament***

***flight controls***  
***APU***  
***air conditioning***

**= Weight Empty**

# ***Aircraft Weights***

**Weight Empty**

***structure***

***propulsion***

***equipment***

**+ Operating Items**

***crew***

***unusable fuel***

***fixed items***

***oxygen***

***engine oil***

***crew baggage***

**= Operating Weight**

**+ Payload**

***passengers***

***cargo***

***bombs***

***luggage***

***missiles***

**+ Fuel**

**= Takeoff Gross Weight**



# ***Aircraft Weights***

## **Takeoff Gross Weight**

- Expended Payload  
(bombs, missiles, paratroopers, cargo drop)**
  - Mission Fuel Used**
- = Landing Weight (including reserve fuel)**

# Aircraft Weights

Takeoff Gross Weight (TOGW)

Usable Fuel

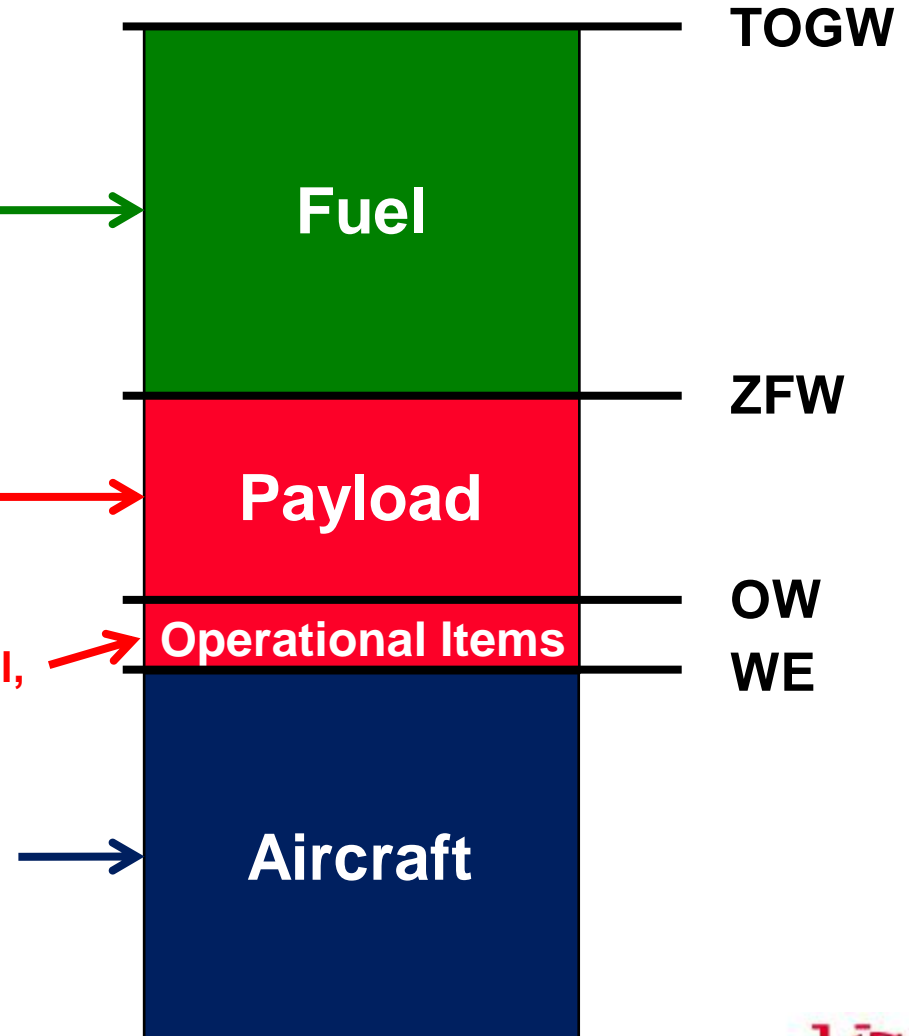
Zero Fuel Weight (ZFW)

Payload – includes stores, bombs, missiles, and other expendables

Operating Weight (OW)

Operational Items– includes unusable fuel, engine oil, crew, suspension equipment

Weight Empty – the weight of the aircraft including structure, propulsion, flight controls, avionics, subsystems, etc



# Aircraft Weights

Basic Mission Takeoff Gross Weight =  
 $OW + \text{Mission Payload} + \text{Mission Fuel}$

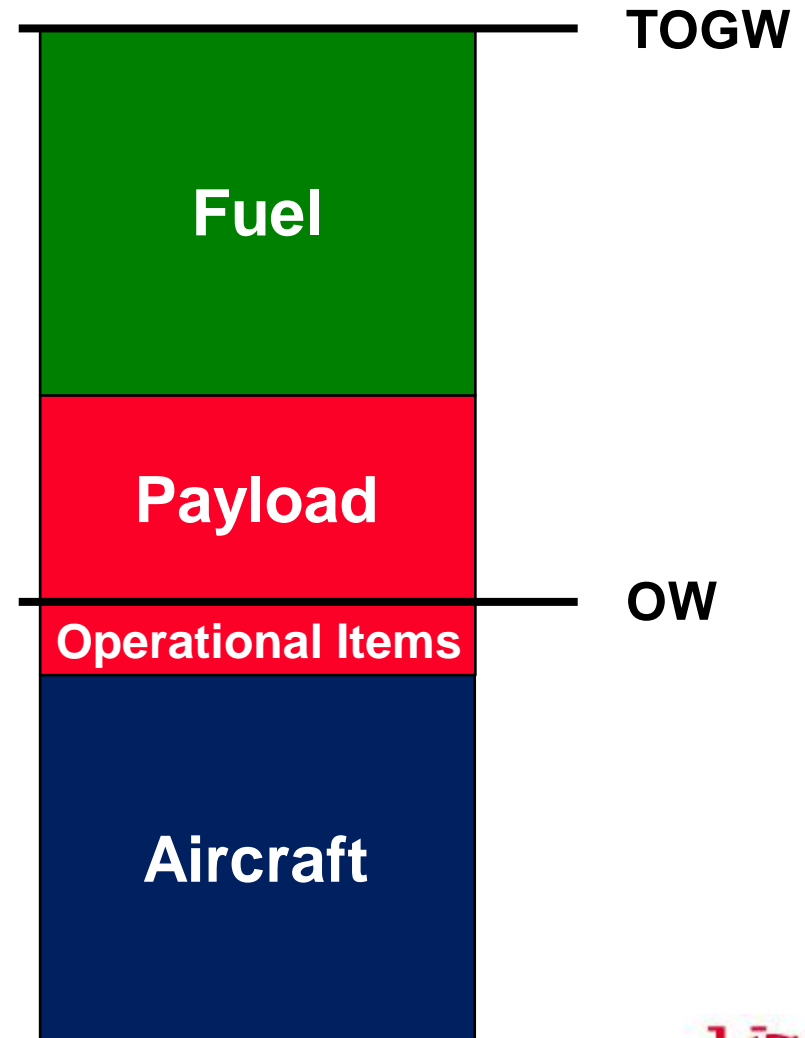
Maximum Fuel – full capacity  
Mission Fuel – specific mission capability

Maximum Payload – full capacity loadout  
Mission Payload – specific mission loadout

Maximum Takeoff Gross Weight =  
 $OW + \text{Maximum Payload} + \text{Maximum Fuel}$

- or -

Maximum Takeoff Gross Weight  
could be set by other factors  
(landing gear limit, c.g. limits, etc)



# ***Aircraft Weights***

## **Common Weight Ratios**

$$\text{Fuel Fraction} = \frac{W_{\text{fuel}}}{W_{\text{TO}}}$$

$$\text{Payload Fraction} = \frac{W_{\text{payload}}}{W_{\text{TO}}}$$

$$\text{Weight Empty Fraction} = \frac{W_{\text{empty}}}{W_{\text{TO}}}$$

$$\text{Crew Weight Fraction} = \frac{W_{\text{crew}}}{W_{\text{TO}}}$$

# *Aircraft Weights*

$$W_{TO} = W_{crew} + W_{payload} + W_{fuel} + W_{empty}$$

$$W_{TO} = \frac{W_{crew} + W_{payload}}{1 - \frac{W_{fuel}}{W_{TO}} - \frac{W_{empty}}{W_{TO}}}$$

# ***Aircraft Weights***



## **Example – 777-200ER**

<b>Operating Weight Empty</b>	<b>304,500 lbs</b>	
<b>Maximum Fuel Capacity</b>	<b>303,000 lbs</b>	
<b>Maximum Payload</b>	<b>96,000 lbs</b>	<b>(400 passengers)</b>
<b>Maximum TOGW</b>	<b>656,000 lbs</b>	

**1 passenger = 180 lbs + 60 lbs of baggage**

**How many passengers can the 777-200ER carry if it carries all of its fuel capacity?**

**How much fuel can the 777-200ER carry if it is a full flight with 400 passengers? How much fuel would it need to offload to not exceed its Maximum TOGW?**

# ***Aircraft Weights***

## **1<sup>st</sup> Example – 777-200ER**



**Operating Weight Empty 304,500 lbs**

**Maximum Fuel Capacity 303,000 lbs**

**Maximum Payload 96,000 lbs (400 passengers)**

**Maximum TOGW 656,000 lbs**

**1 passenger = 180 lbs + 60 lbs of baggage**

**Full Fuel: 656,000 Maximum TOGW**

**- 304,500 Operating Weight**

**- 303,000 Maximum Fuel**

**= 48,500 for passengers + baggage**

**➡ 48,500 / 240 = 202 passengers**

# ***Aircraft Weights***



## **2<sup>nd</sup> Example – 777-200ER**

**Operating Weight Empty 304,500 lbs**

**Maximum Fuel Capacity 303,000 lbs**

**Maximum Payload 96,000 lbs (400 passengers)**

**Maximum TOGW 656,000 lbs**

**1 passenger = 180 lbs + 60 lbs of baggage**

**Full Flight: 656,000 Maximum TOGW**

**- 304,500 Operating Weight**

**- 96,000 400 passengers**

**➡ = 255,500 Fuel Quantity**

**303,000 – 255,500 = 47,500 lbs offloaded ←**

**Convention: all weight values are positive numbers**



# Aircraft Weights

## 2<sup>nd</sup> Example – 777-200ER

Crew Weight	-	Crew Weight	-
Payload Weight	96,000	Payload Fraction	14.6%
Fuel Weight	255,500	Fuel Fraction	38.9%
Empty Weight	304,500	Empty Weight Fraction	46.4%
Sum	656,000	Formula	100.0%



## 2017 UC Aerocats

Crew Weight	-	Crew Weight	-
Payload Weight	37	Payload Fraction	69.8%
Fuel Weight	-	Fuel Fraction	0.0%
Empty Weight	16	Empty Weight Fraction	30.2%
Sum	53	Formula	100.0%

# Aircraft Weights

$$W_{TO} = W_{crew} + W_{payload} + W_{fuel} + W_{empty}$$

$$W_{TO} = \frac{W_{crew} + W_{payload}}{1 - \frac{W_{fuel}}{W_{TO}} - \frac{W_{empty}}{W_{TO}}}$$

**A check of the math ...**

Crew Weight	-	Crew Weight	-
Payload Weight	48,500	Payload Weight	48,500
Fuel Weight	303,000	Fuel Fraction	0.461890
Empty Weight	304,500	Empty Weight Fraction	0.464177
Sum	656,000	Formula	656,000

Crew Weight	-	Crew Weight	-
Payload Weight	96,000	Payload Weight	96,000
Fuel Weight	255,500	Fuel Fraction	0.389482
Empty Weight	304,500	Empty Weight Fraction	0.464177
Sum	656,000	Formula	656,000

# ***Review of Center of Gravity Definitions***

**Weight and Balance** – the process to ensure that the aircraft is within the allowable weight limits and center of gravity limits for safe flight

**Center of gravity limits** – specified longitudinal and lateral boundaries that the center of gravity must be located within during taxi and flight (CG range)

**Ballast** – removable weight specifically used to bring the center of gravity into the allowable range



# ***Review of Center of Gravity Definitions***

**Reference datum** – the reference plane that allows accurate and uniform measurements to any point on the aircraft

**Moment arm** – the chordwise distance from the datum to any point on the aircraft

**Moment** – the measure of force that results from an object's weight acting at a distance from the datum

**Mean Aerodynamic Chord (MAC)** – weighted average chord calculated by:

$$\mathbf{MAC} = \frac{2}{3} c_r \frac{1 + \lambda + \lambda^2}{1 + \lambda}$$

$$y_{\mathbf{MAC}} = \frac{b}{6} \frac{1 + 2\lambda}{1 + \lambda}$$

# ***Homework Assignment***

**HW #5 – Aircraft Weights  
(due by 11:59 pm ET on Monday)**

**HW Help Session  
Monday 1:00 – 2:00 pm ET**

**Posted on Canvas  
HW #5 Assignment with instructions, tips,  
and checklist  
HW #5 Template for data table in Excel**

# Questions?