Homework 3 : Aircraft Collision Avoidance Analyses using Reachability

David McPherson

May 7, 2016

I Linear Velocity Control

Consider a system of two airplanes (Dubin's vehicles) careening through the sky on a collision course. Our two planes can only control their linear velocity and cannot escape into a veering mode. This no-turning system defined in terms of relative coordinates is:

$$\dot{x}_r = -u + d\cos(\psi_r)$$

$$\dot{y}_r = d\sin(\psi_r)$$

$$\dot{\psi}_r = 0$$

We can use Ian Mitchell's "Toolbox of Level Set Methods" to perform reachability analysis on this system. We will modify the default example Matlab script "air3D" that comes with his ToolboxLS. We will need to change the functions defining the Hamiltonian ("air3DHamFunc") and the Hamiltonian's maximum partials ("air3DPartialFunc").

We need to find the optimal Hamiltonian so we can program it into the "air3DHamFunc" function. Let us recast the bounds on the controls u and d as:

$$\frac{v_1}{\bar{v_1}} = c_u - o_u$$

$$\frac{v_1}{\bar{v_1}} = c_u + o_u$$

$$\frac{v_1}{\bar{v_1}} = c_d - o_d$$

$$\frac{v_1}{\bar{v_1}} = c_d + o_d$$

So that c_u is the center of the region U and o_u is the offset from the center to either end of the interval U. Accordingly, c_d is the center of D and o_d is the offset to either end of D.

Now, note that the system dynamics can be re-cast into a control-affine form:

$$\dot{z} = \begin{bmatrix} \dot{x}_r \\ \dot{y}_r \\ \dot{\psi}_r \end{bmatrix} = \begin{bmatrix} -u + d\cos(\psi_r) \\ d\sin(\psi_r) \\ 0 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} -1 \\ 0 \\ 0 \end{bmatrix} u + \begin{bmatrix} \cos(\psi_r) \\ \sin(\psi_r) \\ 0 \end{bmatrix} d$$

As the time horizon $T\to\infty$ the reachable set reached a fixed point and ceased to expand.

2 Mode Switching Control