

DPW-VII

Opening Remarks



**John Vassberg
Chairman, DPW-OC**

**Aviation 2022
Chicago, IL**

June 25-26, 2022

DPW-VII: “*Expanding The Envelope*”

- Introduction
- Organizing Committee
- Agenda
- DPW History
- Test Cases 1-6
- Participant Demographics
- CRM WB Geometry
- Measured Wing Deflections
- CRM Reference Quantities
- Gridding Guidelines & Family Plan
- Baseline Grid Families
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 - NLR Multi-Block
 - JAXA Unstructured
 - DLR Unstructured



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DPW-VII: Organizing Committee

- Olaf Brodersen, DLR
- Ed Feltrop, Cessna
- David Hue, ONERA
- Stefan Keye, DLR
- Kelly Laflin, Cessna
- Dimitri Mavriplis, UWy
- Joe Morrison, NASA
- Mitsuhiro Murayama, JAXA
- Raj Nanjia, RAeS
- Ben Rider, Boeing
- Melissa Rivers, NASA
- Ed Tinoco, Retired
- Chris Toomer, UWE Bristol
- John Vassberg, JetZero
- Rich Wahls, NASA

15 Members

2 Charter Members

3 New Members

10 Institutions

5 Countries

3 Continents

DPW-VII Agenda – Day 1

Day 1 - Saturday June 25, 2022					
8:00 AM	9:00 AM	Arrive and Registration			
Session 1 - Opening Remarks Agenda, Purpose, CRM Geometry/Design, & Test Cases				Rich Wahls	NASA
9:00 AM	9:15 AM	Opening Remarks Agenda, Purpose, CRM Geometry/Design, Test Cases	John Vassberg	Jet Zero	
9:15 AM	9:35 AM	CRM Baseline Grids:	John Vassberg Stefan Keye (remote) Mitsuhiko Murayama	Jet Zero DLR JAXA	
9:35 AM	9:50 AM	CRM Experimental Data (NTF, 11'TWT, ETW)	Melissa Rivers Rich Wahls	NASA	
9:50 AM	10:00 AM	Aeroelastic Deflections Overview	Stefan Keye (remote)	DLR	
10:00 AM	10:20 AM	Break			
Session 2 - Participant Presentations I				David Hue	ONERA
10:40 AM	11:00 AM	TAU	unstructured	Stefan Keye, Olaf Brodersen	DLR (Aerodynamics)
10:20 AM	10:40 AM	USM3D	structured/unstructured	Brent Pomeroy	NASA
11:00 AM	11:15 AM	M-Edge	structured/unstructured	Peter Eliasson	SAAB/VZLU/FOI
11:15 AM	11:30 AM	ENSOLV	structured	Michel van Rooij	NLR
11:30 AM	1:00 PM	LUNCH (on your own)			
Session 3 - Participant Presentations II				Dimitri Mavriplis	U of Wyoming
1:00 PM	1:25 PM	OVERFLOW 2.3e + GGNS	structured/unstructured	Ben Rider	Boeing
1:25 PM	1:40 PM	CHAMPS	structured	Frédéric Plante	Polytechnique Montreal
1:40 PM	1:55 PM	OVERFLOW	structured	Lawton Shoemaker	Univ. Tennessee (Knoxville)
1:55 PM	2:10 PM	ASOP	structured	Yalu Zhu	Nanjing Xfluids Aerospace Tech. Ltd
2:10 PM	2:30 PM	Break			
Session 4 - Participant Presentations III				Mitsuhiko Murayama	JAXA
2:30 PM	2:45 PM	TAS	unstructured	Mitsuhiko Murayama	JAXA
2:45 PM	3:10 PM	FaSTAR	unstructured	Sansica/Abe	JAXA
3:10 PM	3:25 PM	FLOW360	unstructured	Thomas Fitzgibbon	FlexCompute
3:25 PM	3:45 PM	CFD++20.1	unstructured	Amarnatha Potturi	Metacomp
3:45 PM	4:00 PM	Closing of Day 1, Review Day 2 Agenda	Mitsuhiko Murayama		

DPW-VII Agenda – Day 2

Day 2 - Sunday June 26, 2022					
8:30 AM	9:00 AM	Arrive			
Session 5 - Participant Presentations IV			Ben Rider	Boeing	
9:00 AM	9:05 AM	Welcome to Day 2, Agenda Review, etc	Ben Rider	Boeing	
9:05 AM	9:20 AM	Ansys Fluent	unstructured	Krishna Zore, Cristhian Aliaga	Ansys
9:20 AM	9:35 AM	zCFD	unstructured	Oliver Derbyshire	ZenoTech
9:35 AM	9:50 AM	TAU	unstructured	Diliana Friedewald	DLR (Aeroelasticity)
9:50 AM	10:05 AM			Yannick Hoarau	Univ. Strasbourg
10:05 AM	10:25 AM	Break			
Session 6 - Participant Presentations V			Kelly Laflin	Textron	
10:25 AM	10:50 AM	elsA	structured	David Hue	ONERA
10:50 AM	11:15 AM	Kestrel 12.1	structured/unstructured	Brent Pomeroy	NASA
11:15 AM	12:45 PM	LUNCH (on your own)			
Session 7 - Summary Presentations			Ed Tinoco	Retired	
12:45 PM	1:30 PM	DPW-VII Summary of Participant Data - Force/Moment/CP	Ed Tinoco	Retired	
1:30 PM	2:15 PM	DPW-VI Summary of Participant Data - CRM Case 6 SOB Separation TE Separation	Stefan Keye (remote) Kelly Laflin Ben Rider/Olaf Brodersen	DLR Textron Boeing/DLR	
2:15 PM	2:30 PM	Break			
Session 8 - Summary/Closing			John Vassberg	Boeing	
2:30 PM	3:15 PM	Open Discussion	All Attendees		
3:15 PM	3:25 PM	Next Steps	All Attendees		
3:25 PM	3:30 PM	Closing Remarks/Thanks!	John Vassberg		

DPW Series – Brief History

- DPW Charter Formalized Jan 2000
 - State of the Art/Practice CFD Drag Prediction
- DPW-I, Anaheim, CA Jun 2001
 - DLR-F4 WB, Fixed CL & Drag Polar Studies
 - Scatter > 100 Counts → SOP Worse Than Expected
- DPW-II, Orlando, FL Jun 2003
 - DLR-F6 WB & WBNP, Fixed-CL Grid Convergence
 - Scatter > 50 Counts, Drag Deltas, Juncture Flow Issues
- DPW-III, San Francisco, CA Jun 2006
 - DLR-F6 WB & WBF, DPW-W1/W2 Wing-Only Fixed AoA
- DPW-IV, San Antonio, TX Jun 2009
 - CRM WBT, Trim-Drag Study, Blind CFD Predictions
- DPW-V, New Orleans, LA Jun 2012
 - CRM WB, Common Grid Study, TMR Verification Case
- DPW-VI, Washington, DC Jun 2016
 - CRM WB & WBNP, Aero-Elastic Deflection Study
- DPW-VII, Chicago, IL Jun 2022
 - CRM WB, “Expanding the Envelope Beyond RANS”



DPW-VII: “*Expanding The Envelope*”

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DPW-VII: Case 1 “Halfway to Buffet”

- **Case 1: CRM Wing-Body Grid Convergence Study**
 - Use 3.00-deg LoQ AE CRM Geometry/Grids
 - Use at least 4 Grids in Family
 - Plot [CD, CM, AoA] .vs. $N^{(2/3)}$
 - Case 1a (Requested)
 - M = 0.85, Re = 20 million, Fixed CL = 0.58, T = -250°F
 - Case 1b (Optional)
 - M = 0.85, Re = 05 million, Fixed CL = 0.58, T = 100°F

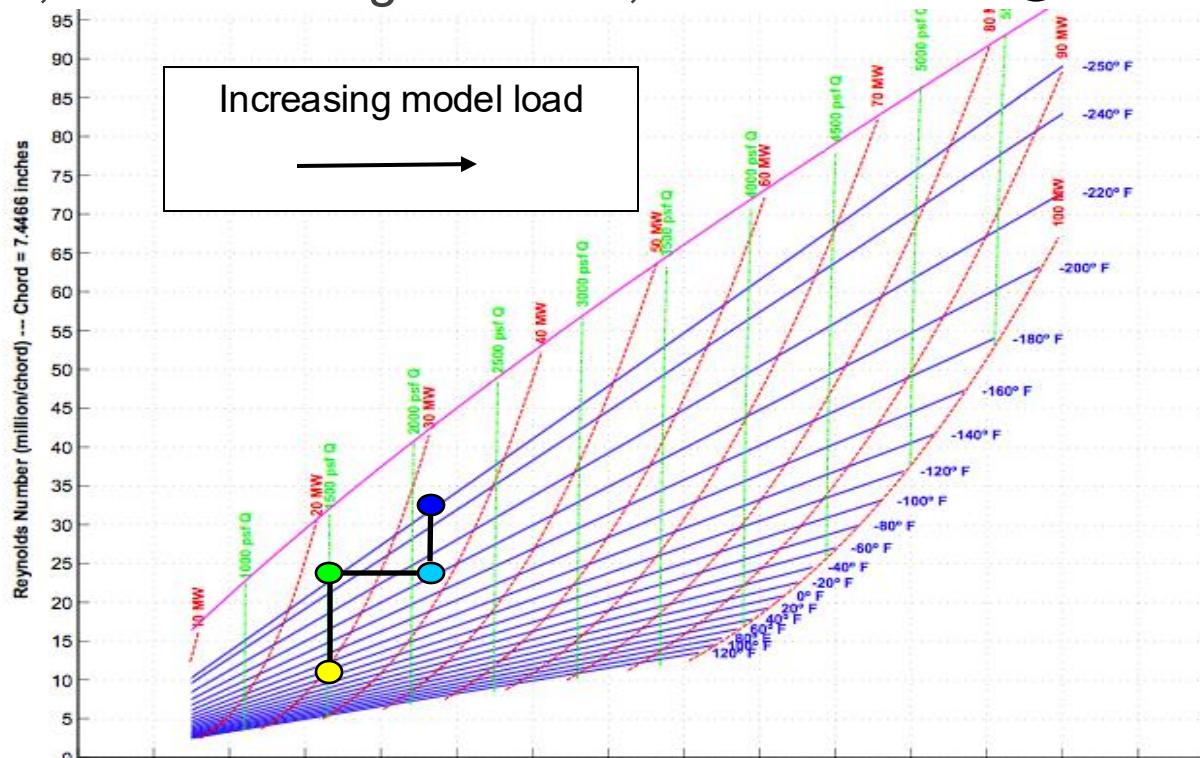
DPW-VII: Case 2 “Polar with AE Effects”

- **Case 2: CRM Wing-Body Static LoQ Aero-Elastic Effect**
 - AoA Sweep with ETW Deflections
 - CL = 0.50 on 2.50-deg LoQ Geometry
 - AoA = [2.50, 2.75, 3.00, 3.25, 3.50, 3.75, 4.00, 4.25] degrees
 - Medium Baseline Grids: [9 Solutions on 8 Grids/Geometries]
 - Case 2a (Requested)
 - Mach=0.85, Re=20 million, T=-250°F
 - Case 2b (Optional)
 - Mach=0.85, Re=05 million, T=100°F

DPW-VII: Case 3 “Ren & AE Effects”

- Case 3: CRM WB Ren-Sweep at Fixed CL

- M = 0.85, CL = 0.50, Medium Grids
- Re=05M, LoQ 2.50-deg R05 Grid, T= 100°F
- Re=20M, LoQ 2.50-deg R30 Grid, T= -250°F
- Re=20M, HiQ 2.50-deg R30 Grid, T = -182°F
- Re=30M, HiQ 2.50-deg R30 Grid, T = -250°F



DPW-VII: Case 4 “Grid Adaptation”

- **Case 4: CRM WB Grid Adaptation – Alpha Sweep**
 - Optional Test Case
 - $M = 0.85$, $Re = 20$ million, $T = -250^{\circ}\text{F}$
 - $CL = 0.50$ on 2.50-deg LoQ Geometry
 - $\text{AoA} = [2.50, 2.75, 3.00, 3.25, 3.50, 3.75, 4.00, 4.25]$ degrees
 - Start Adaptation from Appropriate Baseline LoQ Mesh or AE Geometry
 - Participants to Document Adaptation Process
 - Additional cases: $M = 0.85$; $Re = 5$ million; $T = 100^{\circ}\text{F}$

DPW-VII: Case 5 “Beyond RANS”

- **Case 5: Beyond RANS**
 - Optional Test Case
 - URANS, DDES, WMLES, Lattice Boltzmann, etc.
 - $M = 0.85$, $Re = 20$ million, $T = -250^{\circ}\text{F}$
 - $CL \sim 0.58$ on 3.00-deg LoQ Geometry
 - $\text{AoA} = [2.50, 2.75, 3.00, 3.25, 3.50, 3.75, 4.00, 4.25]$ degrees
 - Use Appropriate LoQ AE Geometry

DPW-VII: Case 6 “Couple AE Simulation”

- **Case 6: CRM WB Coupled Aero-Elastic Simulation**
 - Optional Test Case
 - $M = 0.85$, $Re = 20$ million, $T = -250^{\circ}\text{F}$
 - $CL = 0.58$
 - $\text{AoA} = [2.50, 2.75, 3.00, 3.25, 3.50, 3.75, 4.00, 4.25]$ degrees
 - Start AE Process with NoQ AE Geometry or NoQ R30 Medium Grid

DPW-VII: Participant Demographics

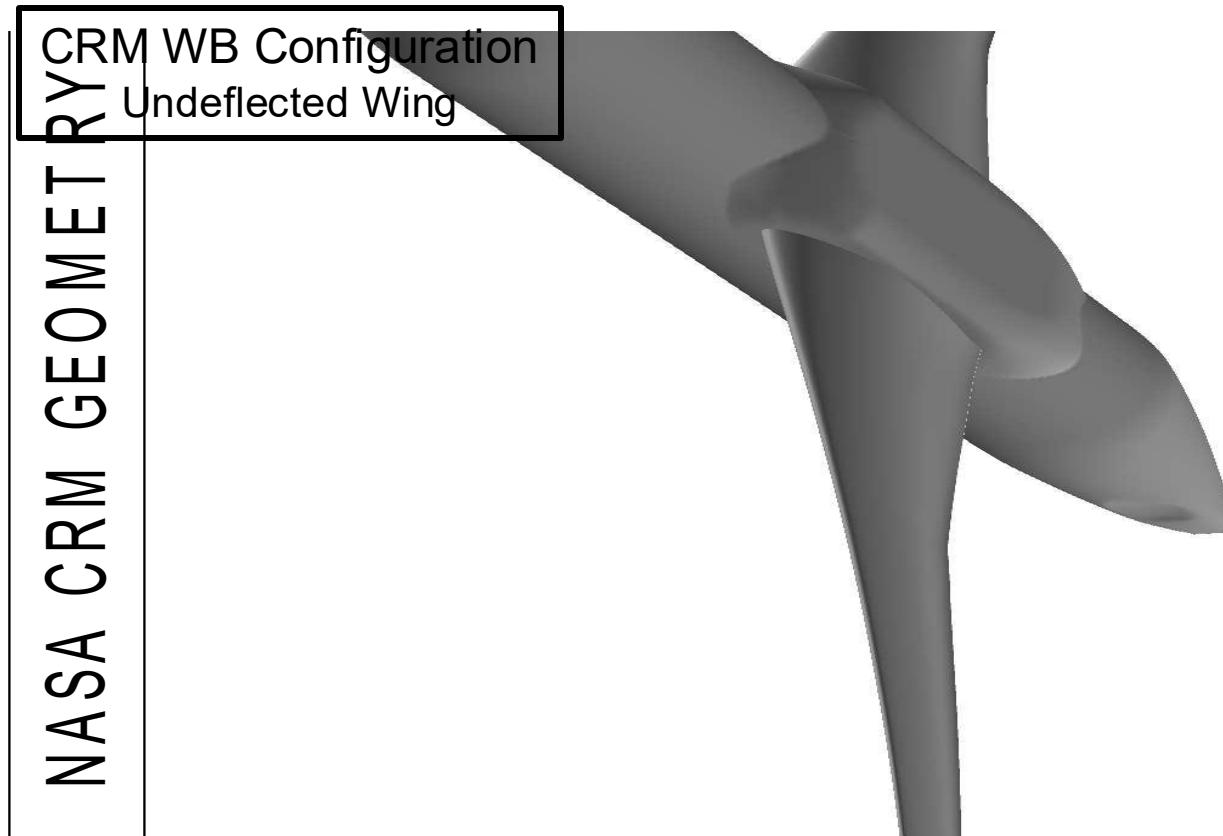
- **18 Teams/Organizations**
 - 7 N. America, 7 Europe, 4 Asia
 - 7 Government, 3 Industry, 3 Academia, 5 Commercial
- **30 Total Data Submittals**
- **Grid Types:**
 - 16 Unstructured
 - 3 Overset
 - 3 Structured Multi-Block
 - 1 Custom Cartesian
- **Turbulence Models:**
 - 14 SA (w/ & w/o QCR), 4 SST,
2 EARSM, 1 SSG/LRR, 1 AMM-QCR, 1 RSW-In(w)

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Common Research Model (CRM)

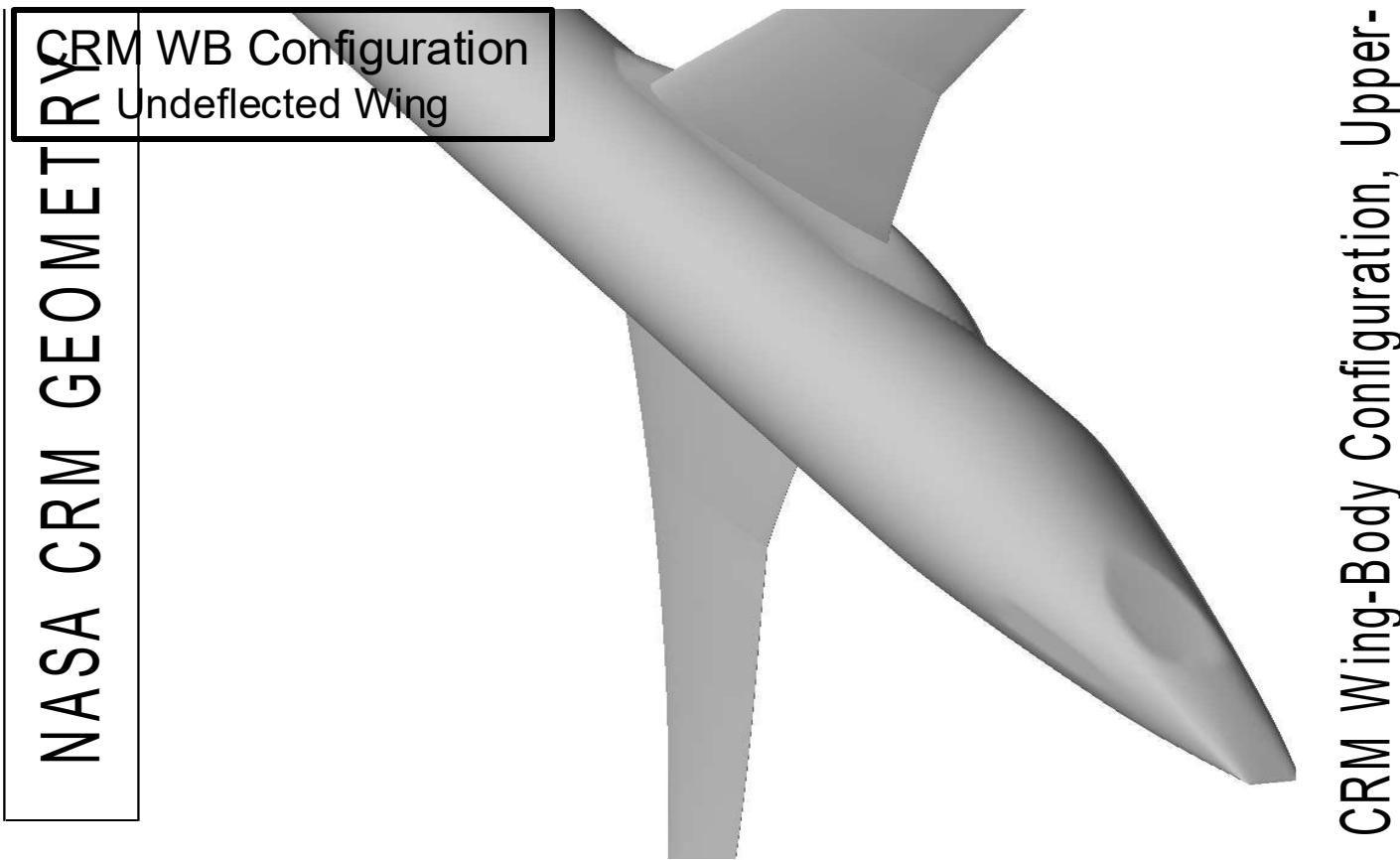


CRM Wing-Body Configuration, Lower-F

W-V, New Orleans LA, June 2012

Vassberg AIAA 2008-6919

Common Research Model (CRM)



Vassberg AIAA 2008-6919

Common Research Model (CRM)



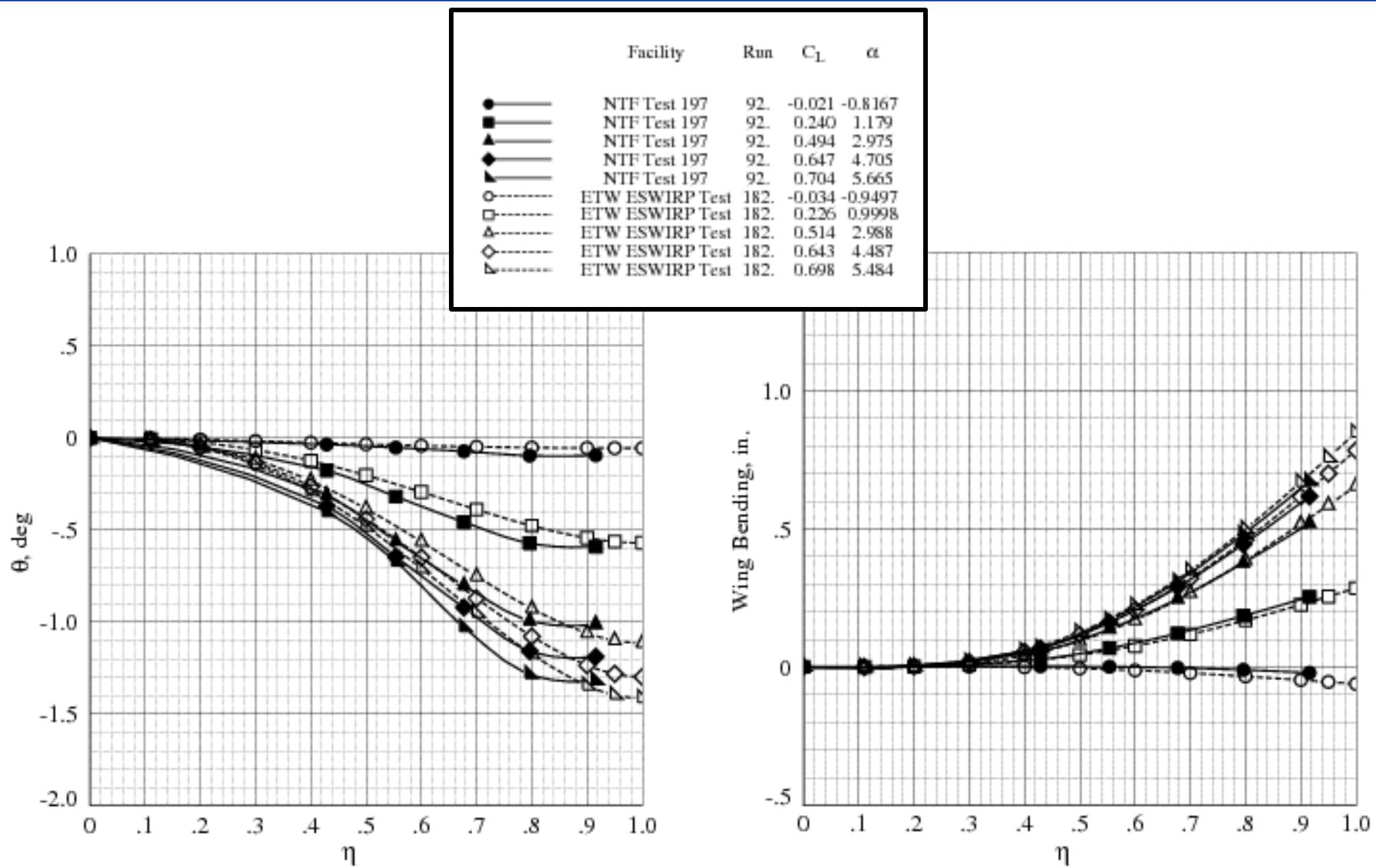
<http://commonresearchmodel.larc.nasa.gov/>

Common Research Model (CRM)

Table 1: Reference Quantities for the CRM.

S_{ref}	594,720.0 in ²	4,130.0 ft ²
<i>Trap-Wing Area</i>	576,000.0 in ²	4,000.0 ft ²
C_{ref}	275.80 in	
$Span$	2,313.50 in	192.8 ft
X_{ref}	1,325.90 in	
Y_{ref}	468.75 in	
Z_{ref}	177.95 in	
λ	0.275	
$\Lambda_{C/4}$	35°	
AR	9.0	

CRM Wing Deflections in NTF, ETW



DPW-7 Utilizes ETW Deflections

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DPW-VII: Gridding Guidelines (1/2)

- **Tiny Grid (L1)**

- Viscous Wall Spacing: $Y^+ \sim 1.0 \rightarrow \Delta y_1 = 0.0002332"$
 - Based on local C_f @ 10% Cref for $Re_c = 30$ million
 - $C_f \sim 0.455 / \ln^2(0.06 * Re_x) = 0.003107$, where $Re_x = 0.1 * Re_c = 3$ million
 - $\Delta y_1 = Cref / [Re_c * \sqrt{C_f/2}] = 0.0002332"$
- At Least 2 Constantly-Spaced Cells at Viscous Walls, $\Delta y_2 = \Delta y_1$
- Growth Rates < 1.2X Normal to Viscous Walls
- Wing Spanwise Spacing < 0.1%*Semispan at Root & Tip
- Wing Chordwise Spacing < 0.1%*C (Local Chord) at LE & TE
- Wing TE Base $\gg 8$ Cells
- Spacing Near Fuselage Nose & End-of-Body < 1%*Cref

- **Grow Next-Finer Grid in Family by $\sim [(L+2)/(L+1)]^3$ in Size**

- Scale Dimensions in All Three Directions by $\sim [(L+2)/(L+1)]$
- Grid Spacings Should Reduce as follows, (0.1% in Tiny Grid)
 - $[T,C,M,F,X,U] = [0.100, 0.067, 0.050, 0.040, 0.033, 0.029]\%$

DPW-VII: Gridding Guidelines (2/2)

- **Farfield Boundary > 100*Semispans**
- **Miscellaneous Notes:**
 - Try to be Multigrid Friendly on Structured Meshes
 - Store Grid Coordinates in 64-bit Precision
 - If Storing Grids in Plot3D Format, Keep Zones < 38M Nodes
 - Itemize Surface Elements by Components [W, B, Sym, Far]
 - Itemize Element Count for Unstructured Meshes
 - Volume: Tetrahedra, Prisms, Pyramids, Hexahedra
 - Surface: Triangles, Quads
 - Total of 15 Grids Needed per Grid Type
 - Subtotal of 8 AE Medium Grids @ Low-Q for Alpha Sweep
 - Subtotal of 1 AE Medium Grid @ High-Q for Q Effect
 - Subtotal of 1 Medium Grid on Undeflected Geometry for Case 6
 - Subtotal of 6 Grids in Grid Family for Grid Convergence
 - AE3.00degLoQ Geometry, CL = 0.58, Re = 20M, (Re = 5M Optional)

DPW-VII: Baseline RANS Grid Family Plan

Name	L	WB	Δy_1	γ^+	# Δy_1 s
Tiny (T)	1	~5	0.0002332"	~1.00	2
Coarse (C)	2	~17	0.0001555"	~0.67	3
Medium (M)	3	~40	0.0001166"	~0.50	4
Fine (F)	4	~78	0.0000933"	~0.40	5
Extra Fine (X)	5	~135	0.0000777"	~0.33	6
Ultra Fine (U)	6	~215	0.0000666"	~0.29	7

Rough Nominal Size of Grid System in M-DOF

At Least 4 Sequential Mesh Levels & Bias Towards Finest

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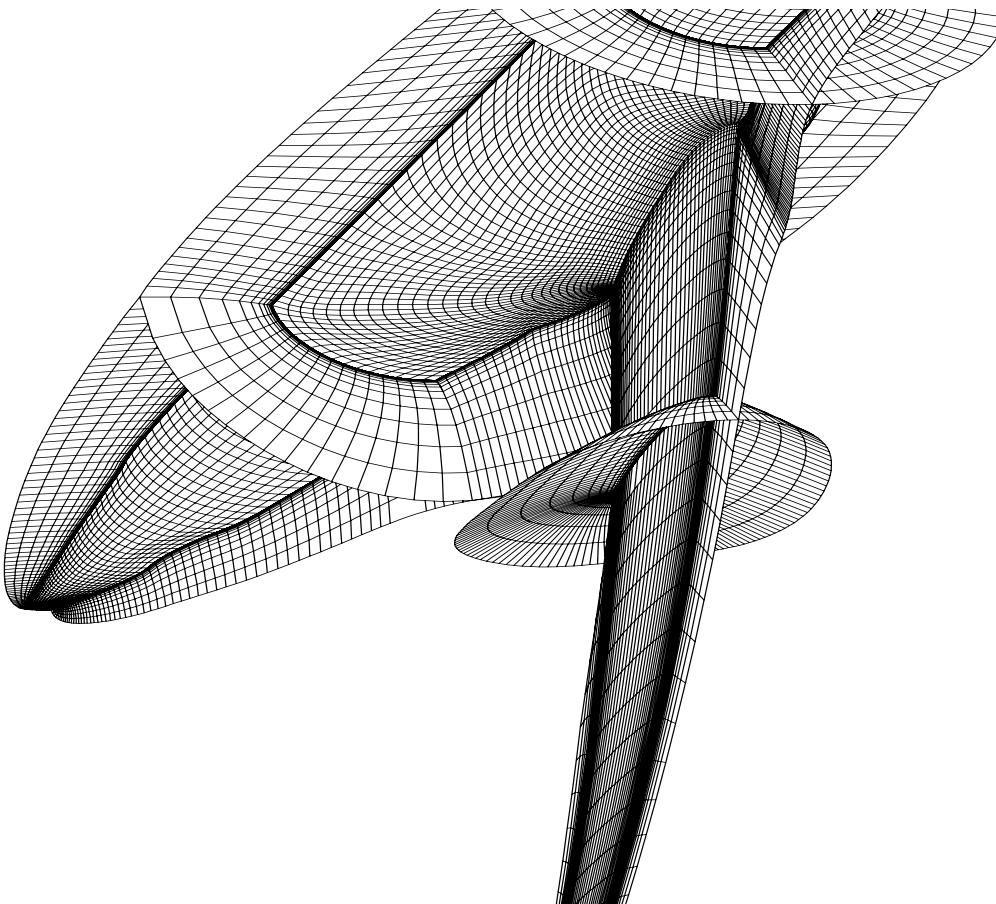


DPW-VII: Vassberg Grid Family Data

Name	L	WB	Δy_1	Y^+	# Δy_1 s
Tiny (T)	1	5,286,597	0.0002332"	~1.00	2
Coarse (C)	2	17,644,325	0.0001555"	~0.67	3
Medium (M)	3	41,590,149	0.0001166"	~0.50	4
Fine (F)	4	80,957,925	0.0000933"	~0.40	5
Extra Fine (X)	5	139,581,509	0.0000777"	~0.33	6
Ultra Fine (U)	6	221,294,757	0.0000666"	~0.29	7

DPW-VII: Vassberg Grid Topology

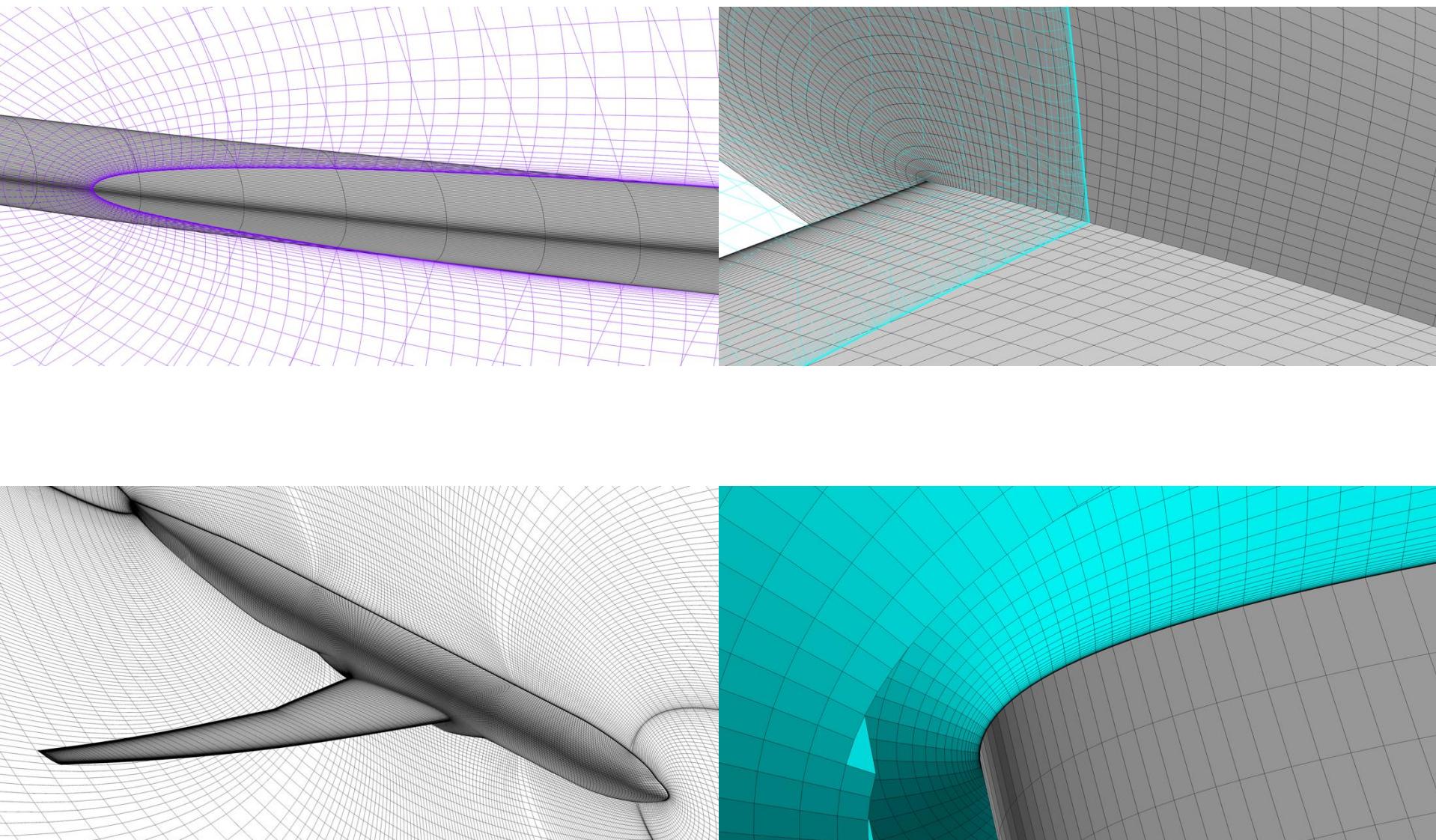
L1 TINY GRID



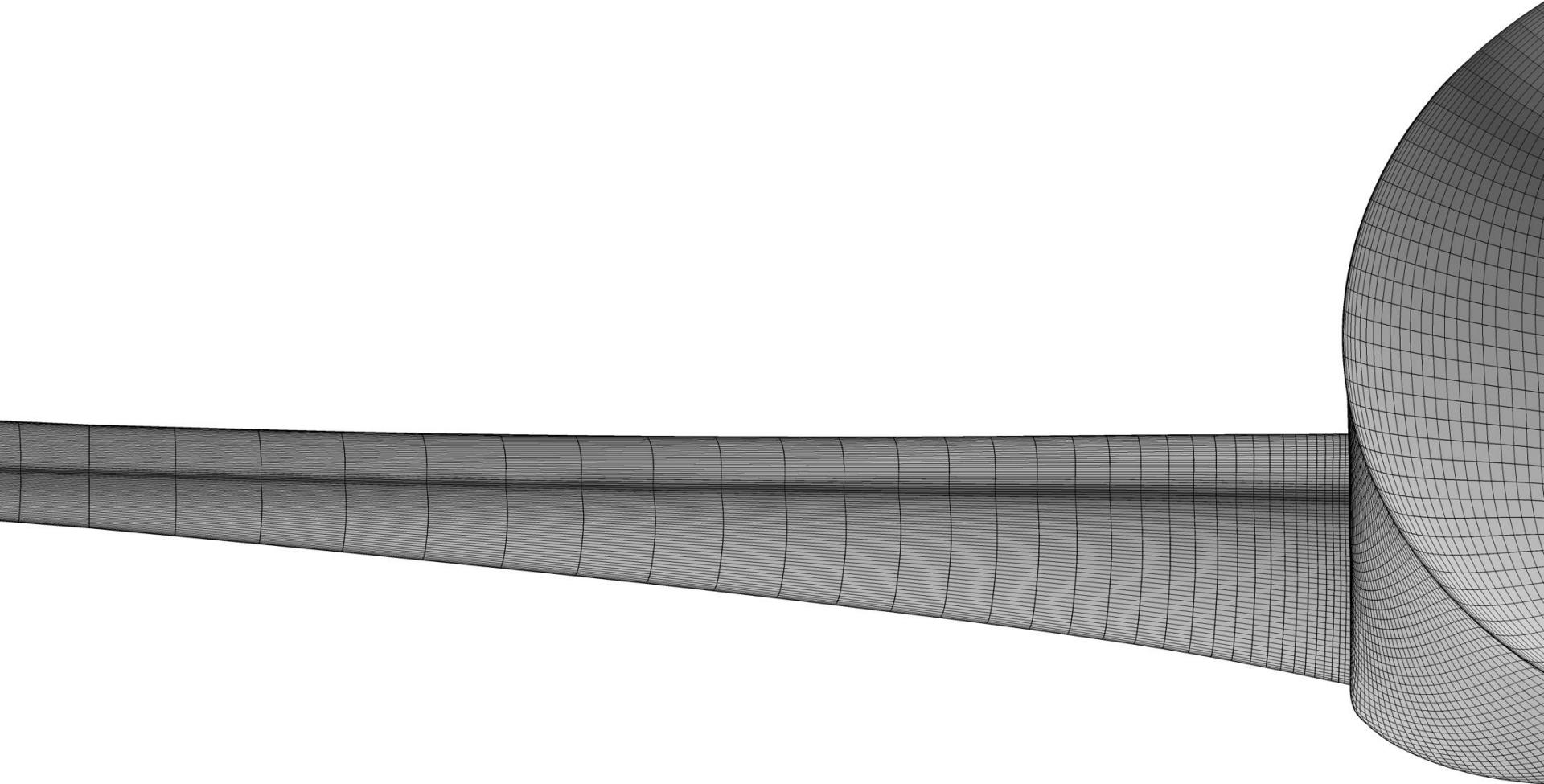
Oblique View, Field Grid Extends To K-N

, DPW-V, New Orleans LA, June 2012

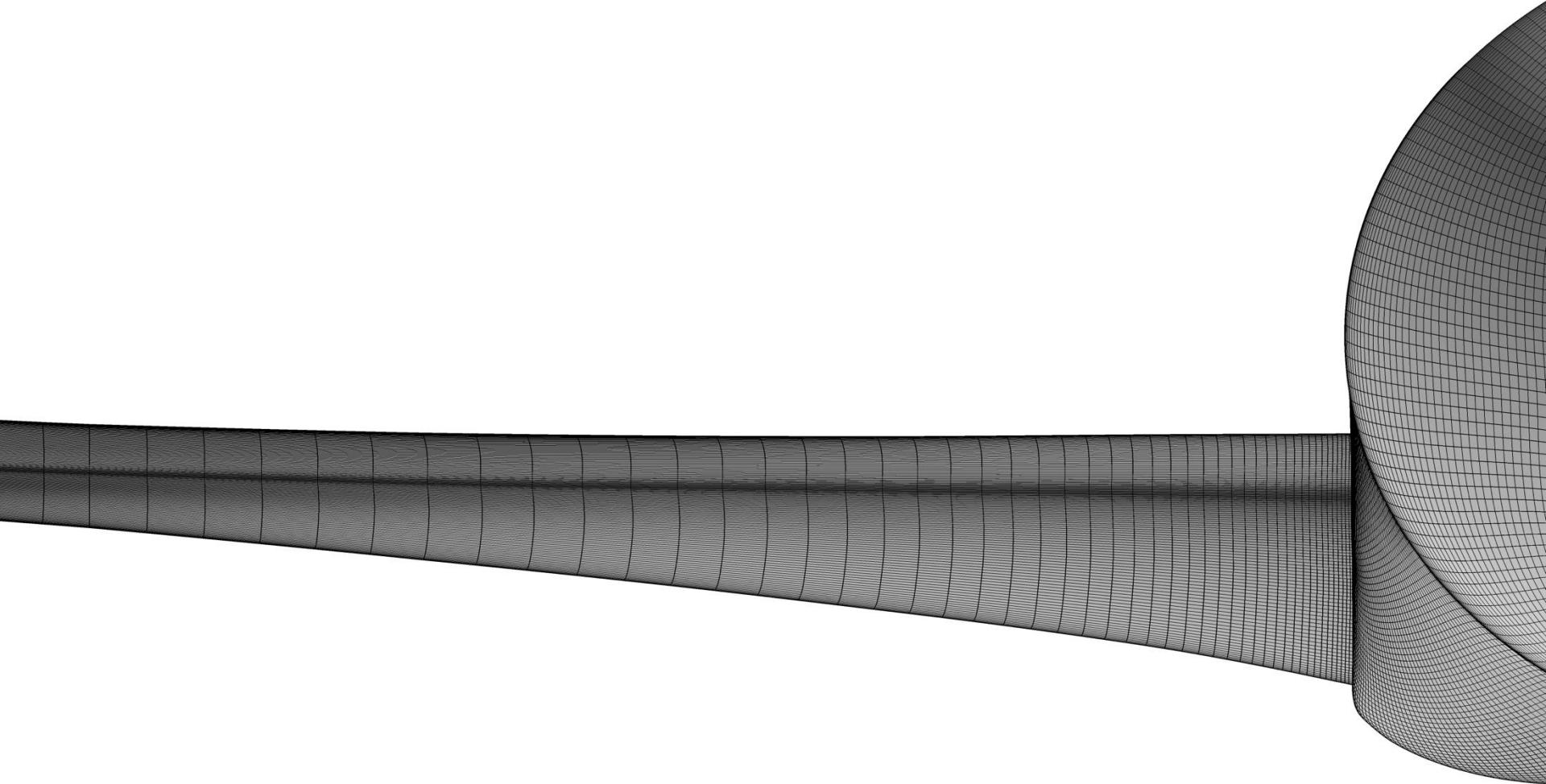
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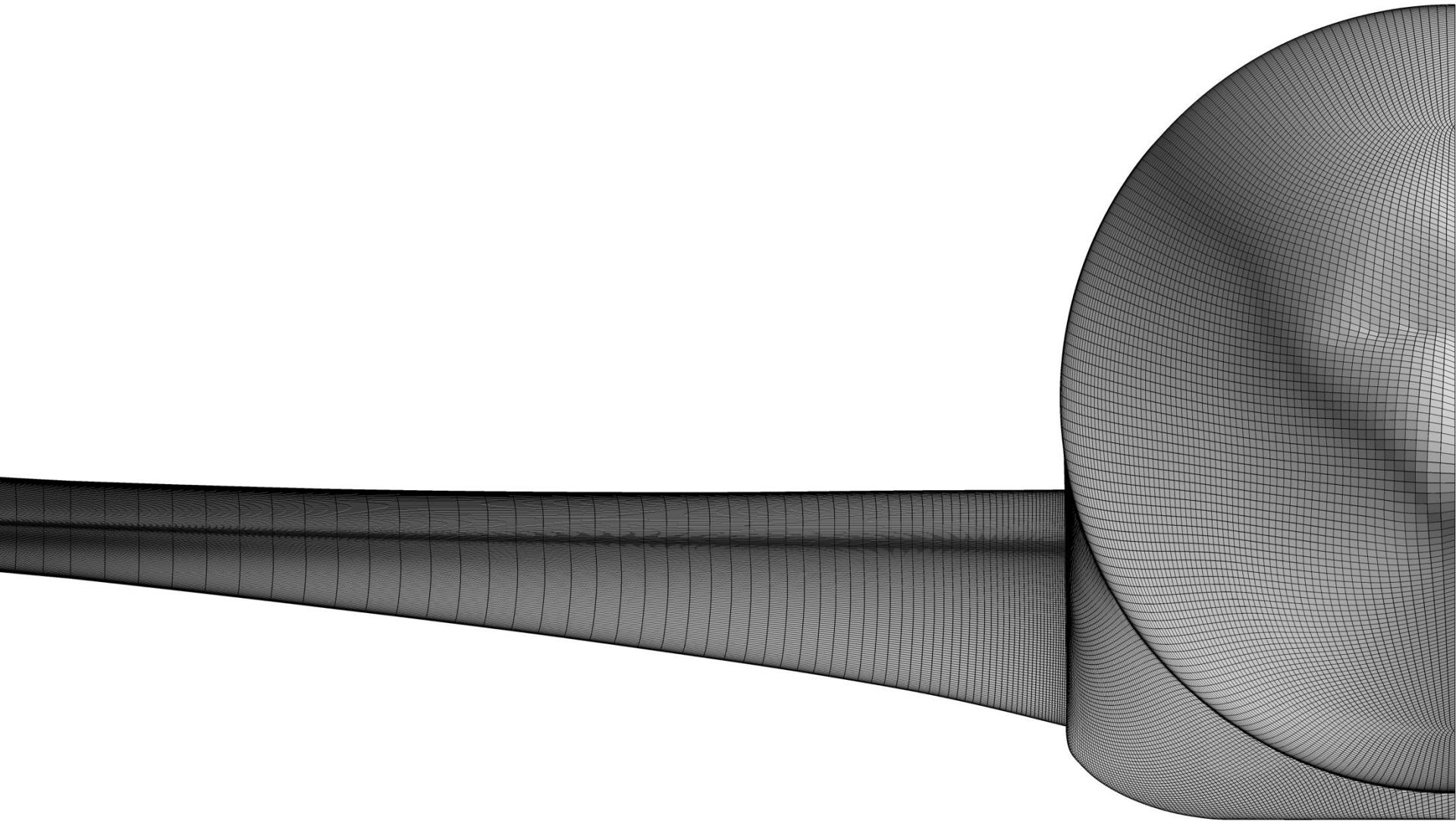
DPW-VII: Vassberg L1 Tiny Grid



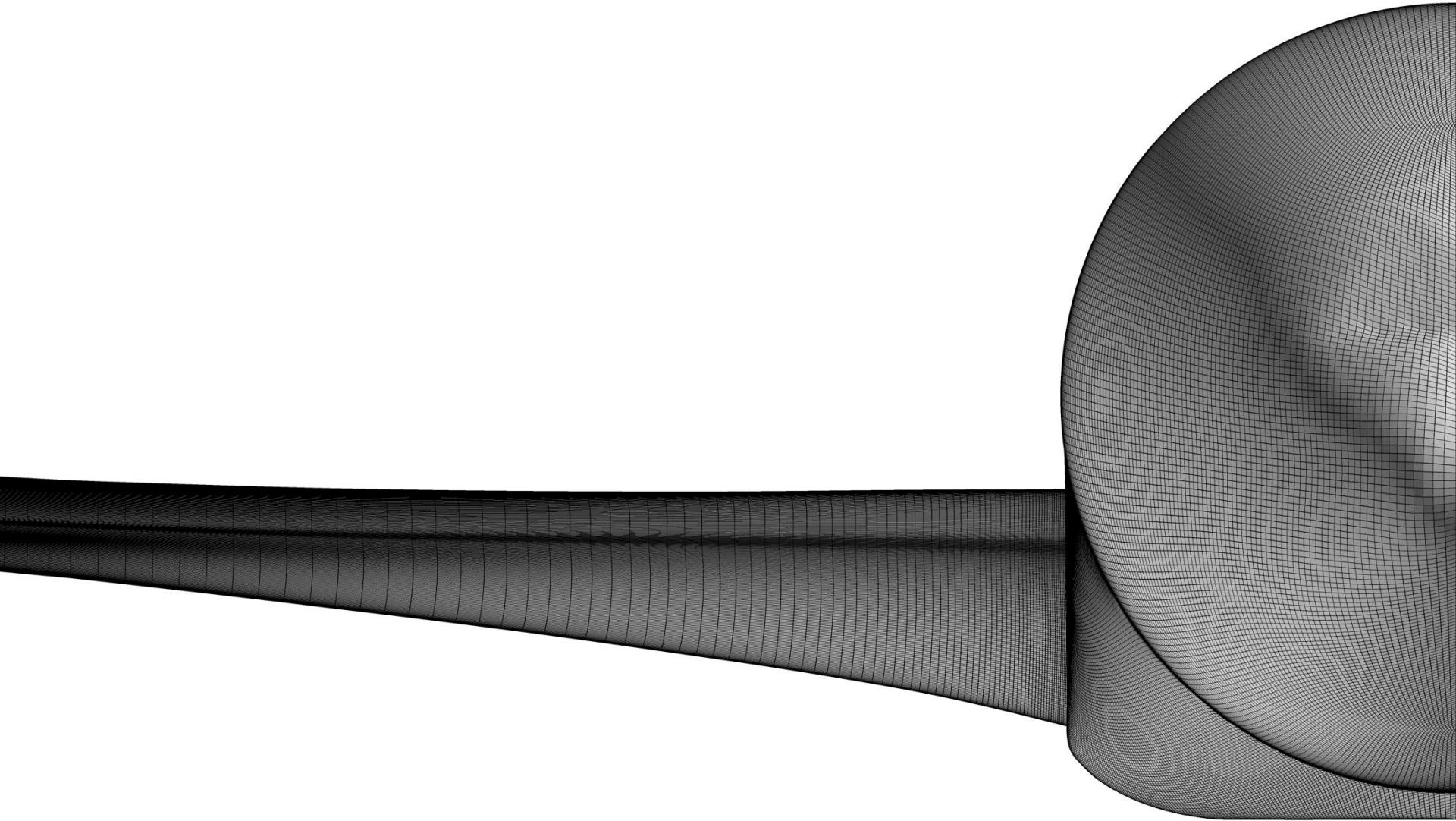
DPW-VII: Vassberg L2 Coarse Grid



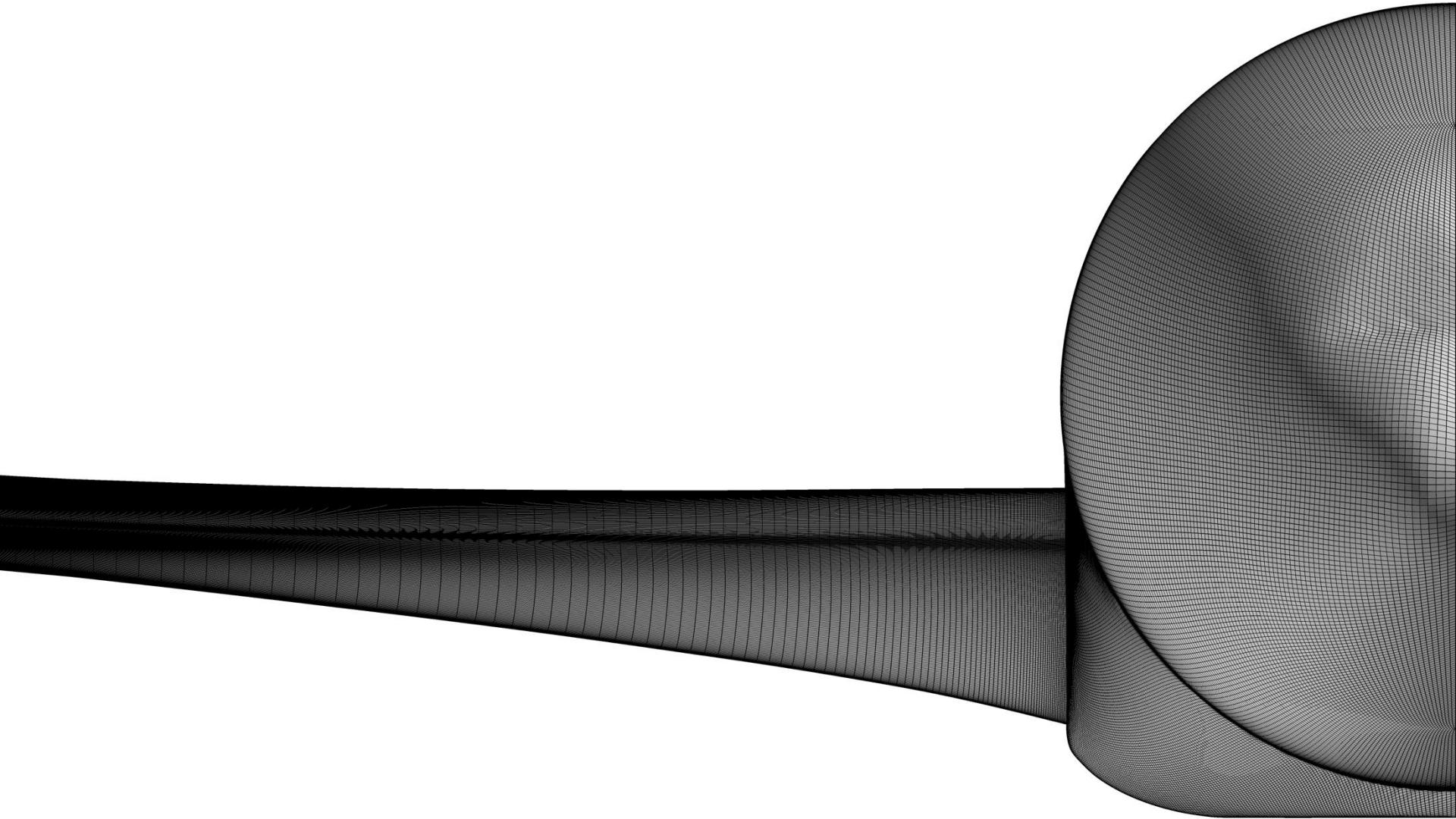
DPW-VII: Vassberg L3 Medium Grid



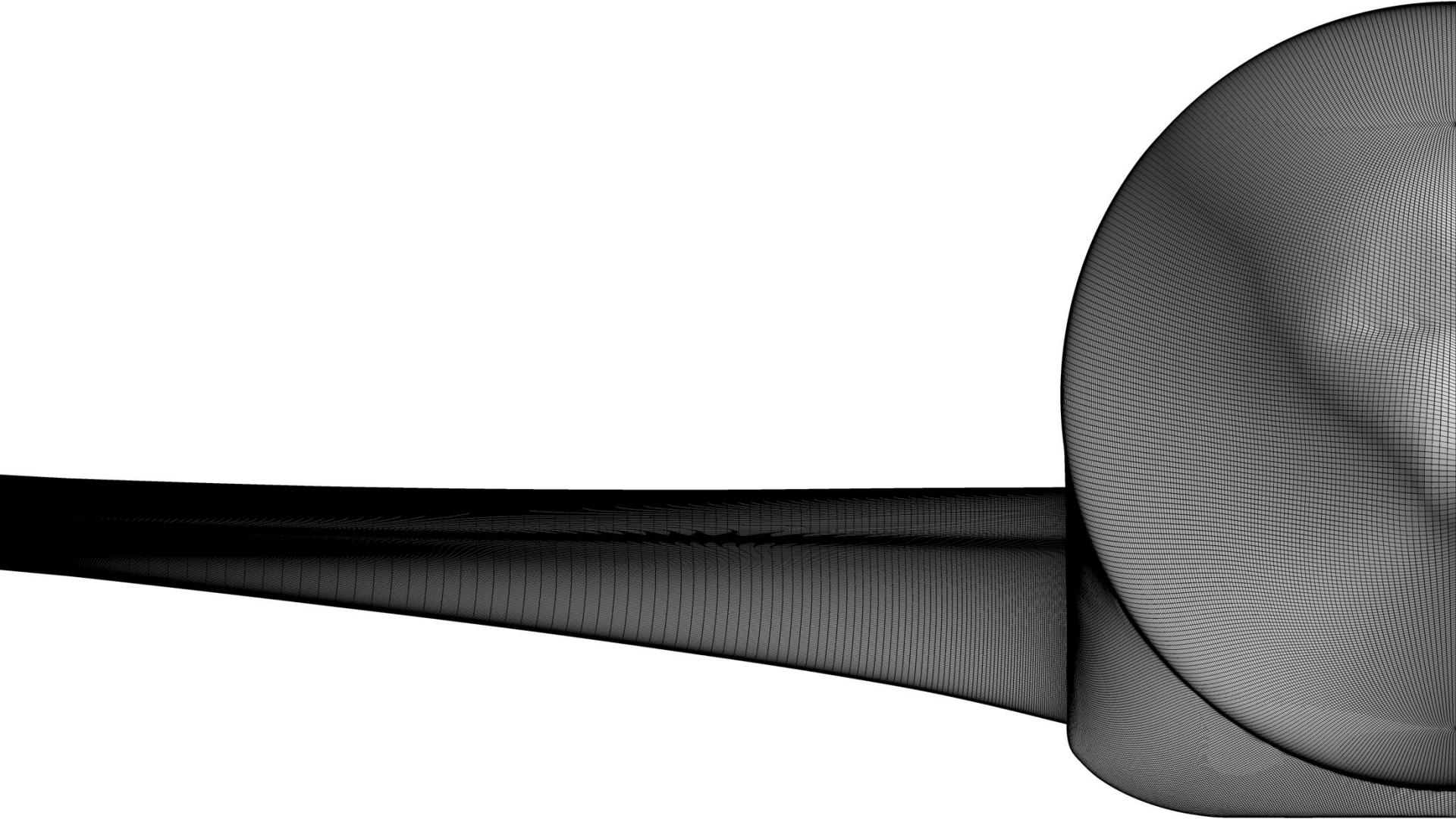
DPW-VII: Vassberg L4 Fine Grid



DPW-VII: Vassberg L5 Extra-Fine Grid



DPW-VII: Vassberg L6 Ultra-Fine Grid

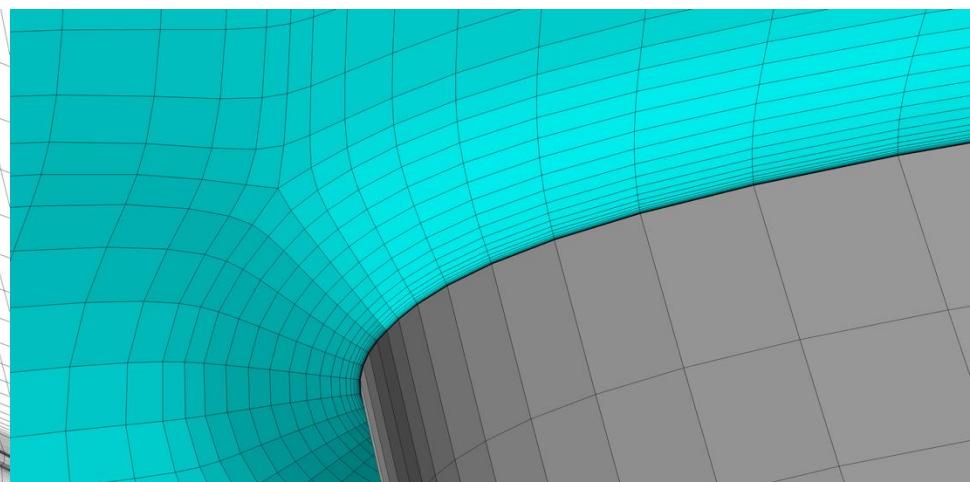
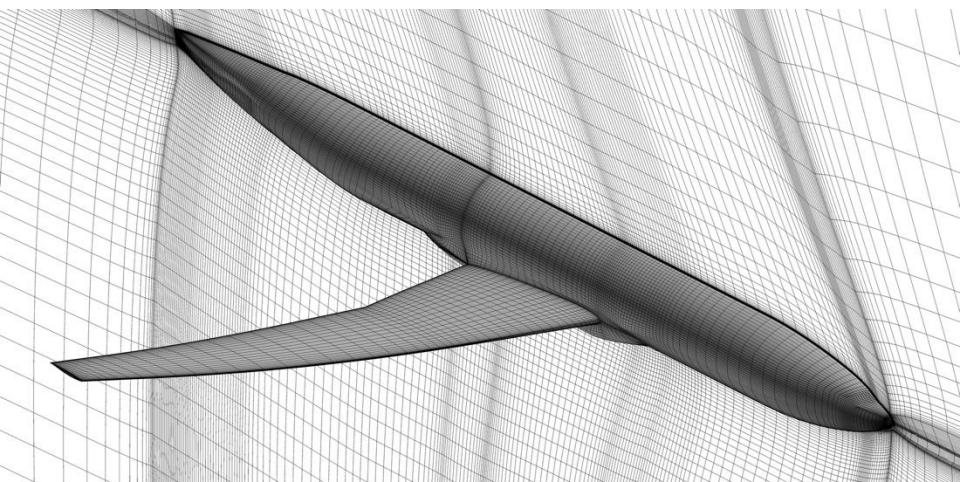
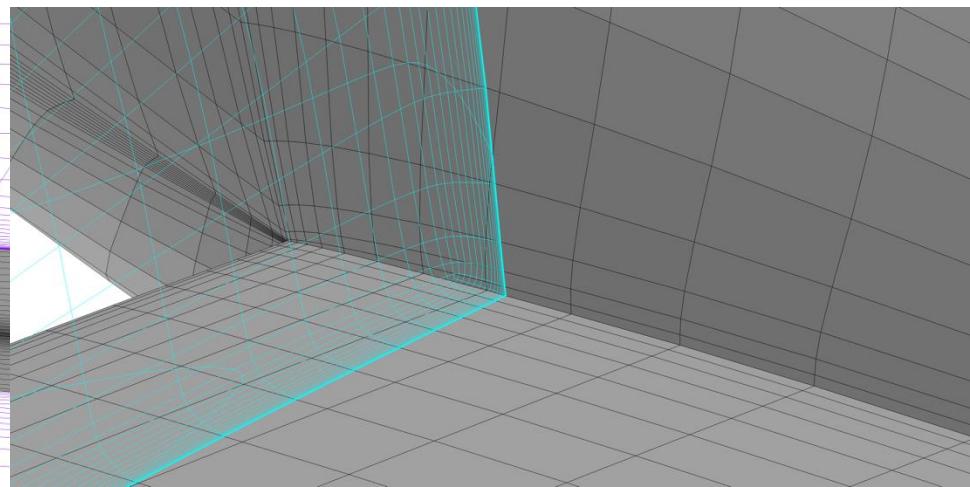
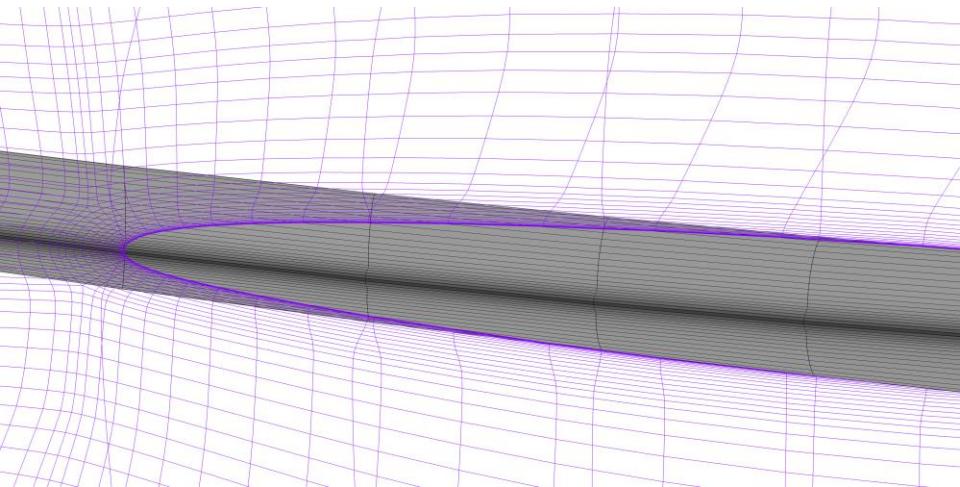


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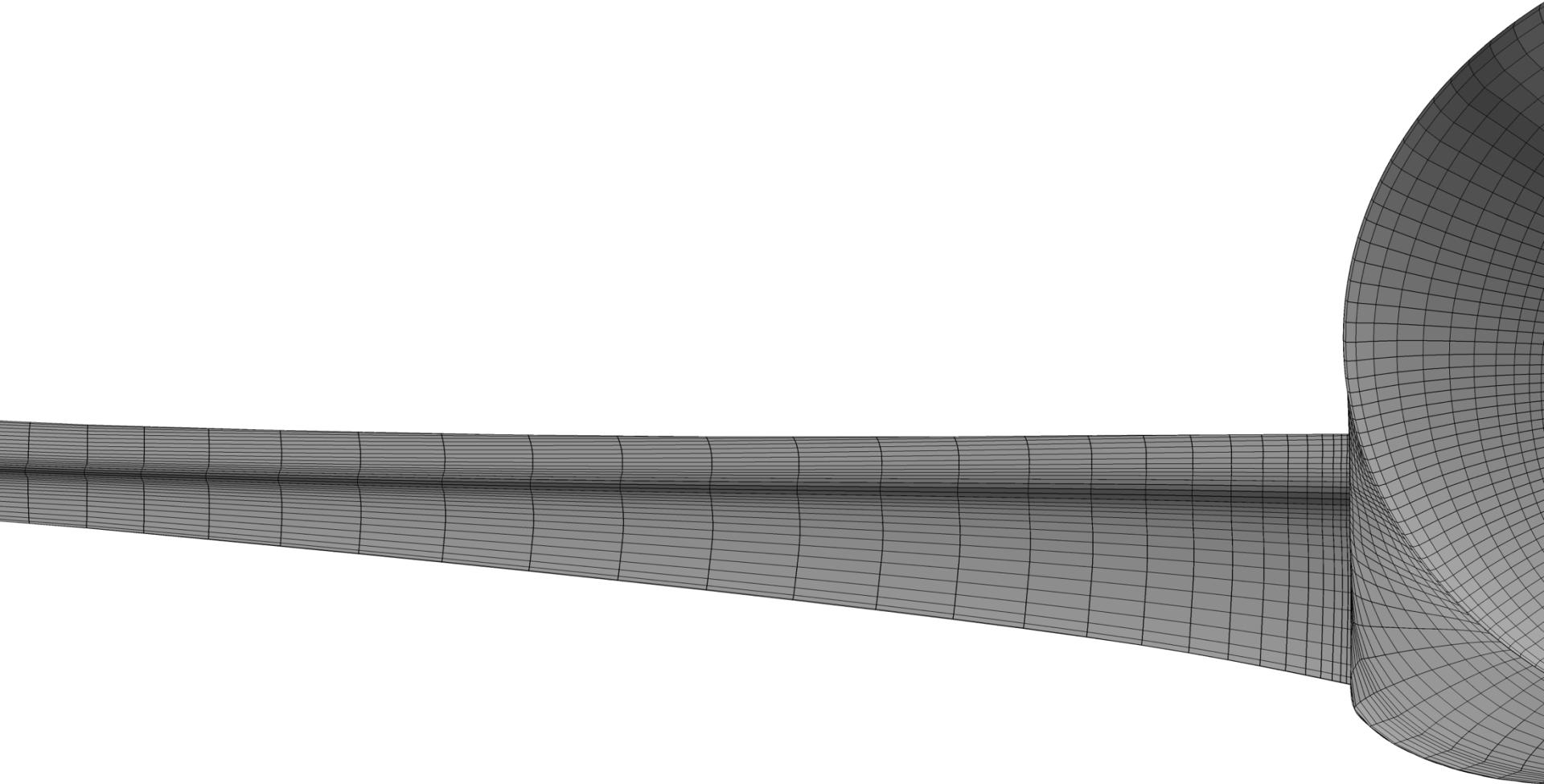
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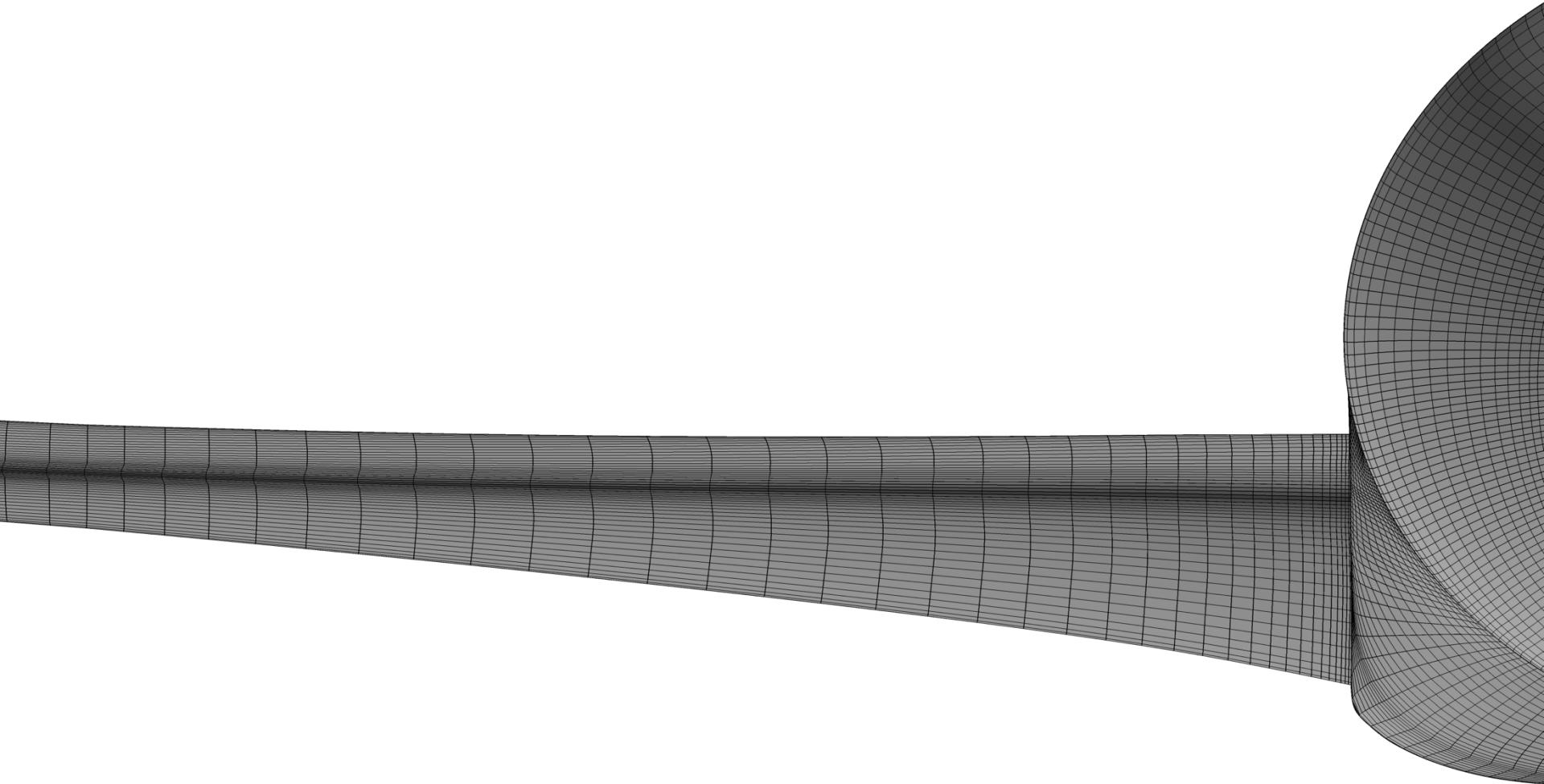
DPW-VII: NLR Grid Topology



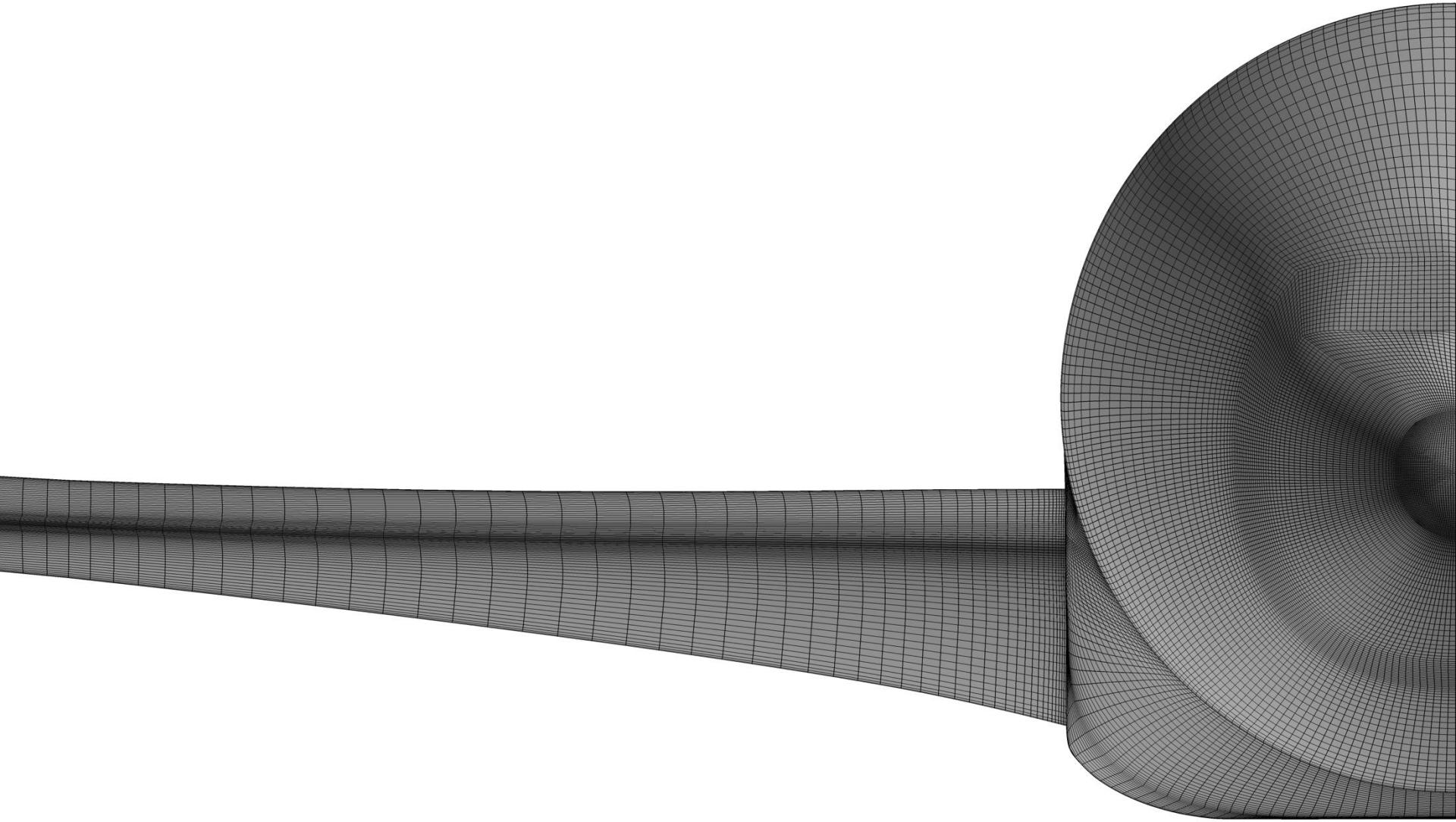
DPW-VII: NLR L1 Tiny Grid



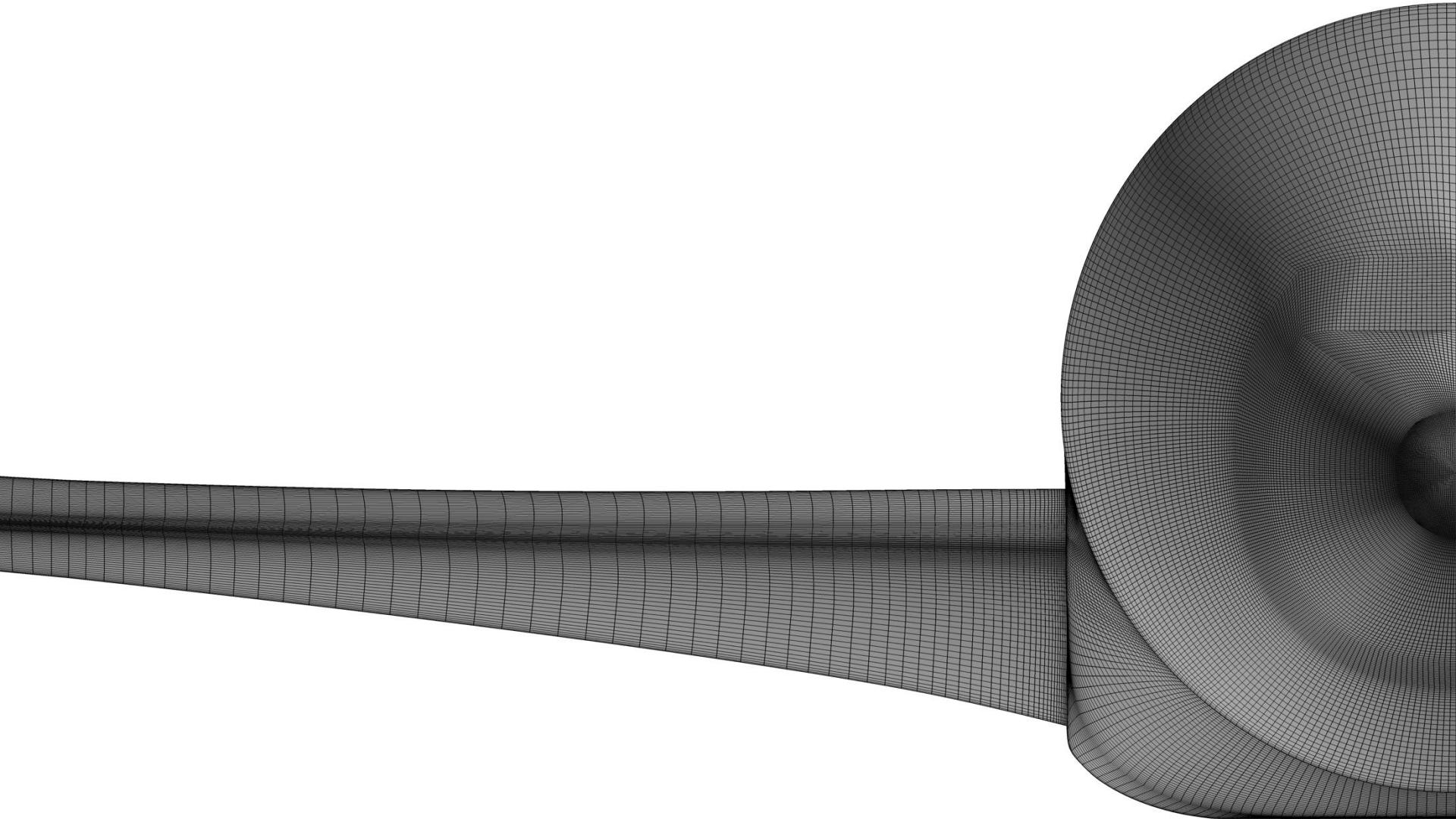
DPW-VII: NLR L2 Coarse Grid



DPW-VII: NLR L3 Medium Grid



DPW-VII: NLR L4 Fine Grid

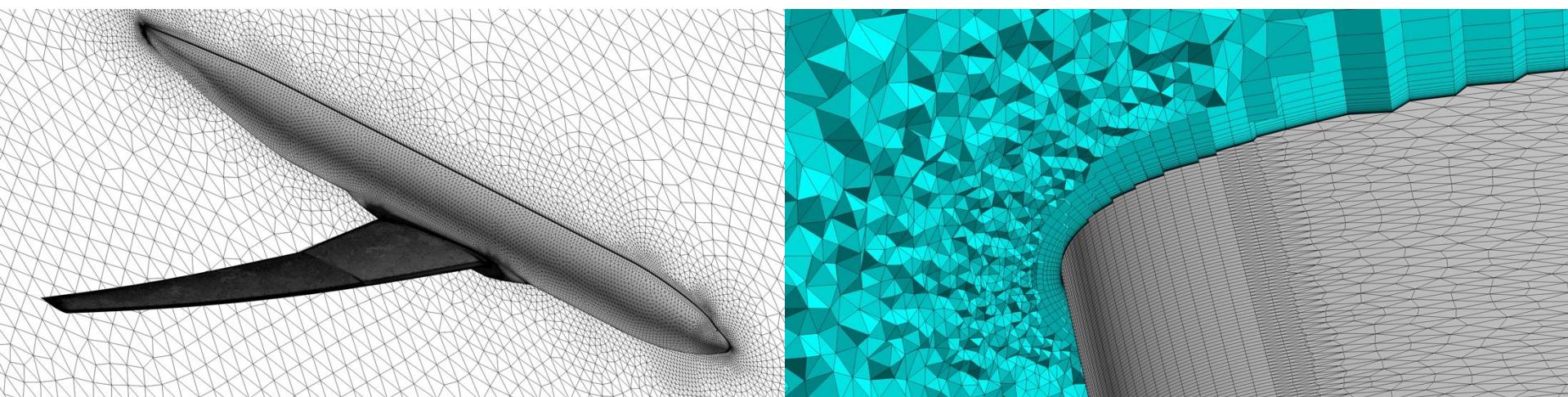
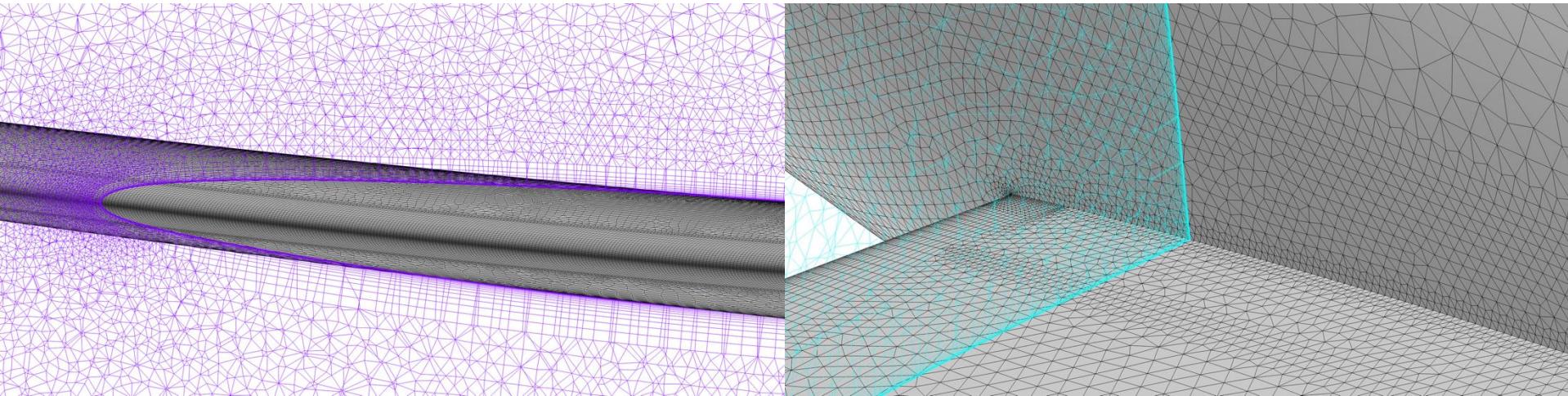


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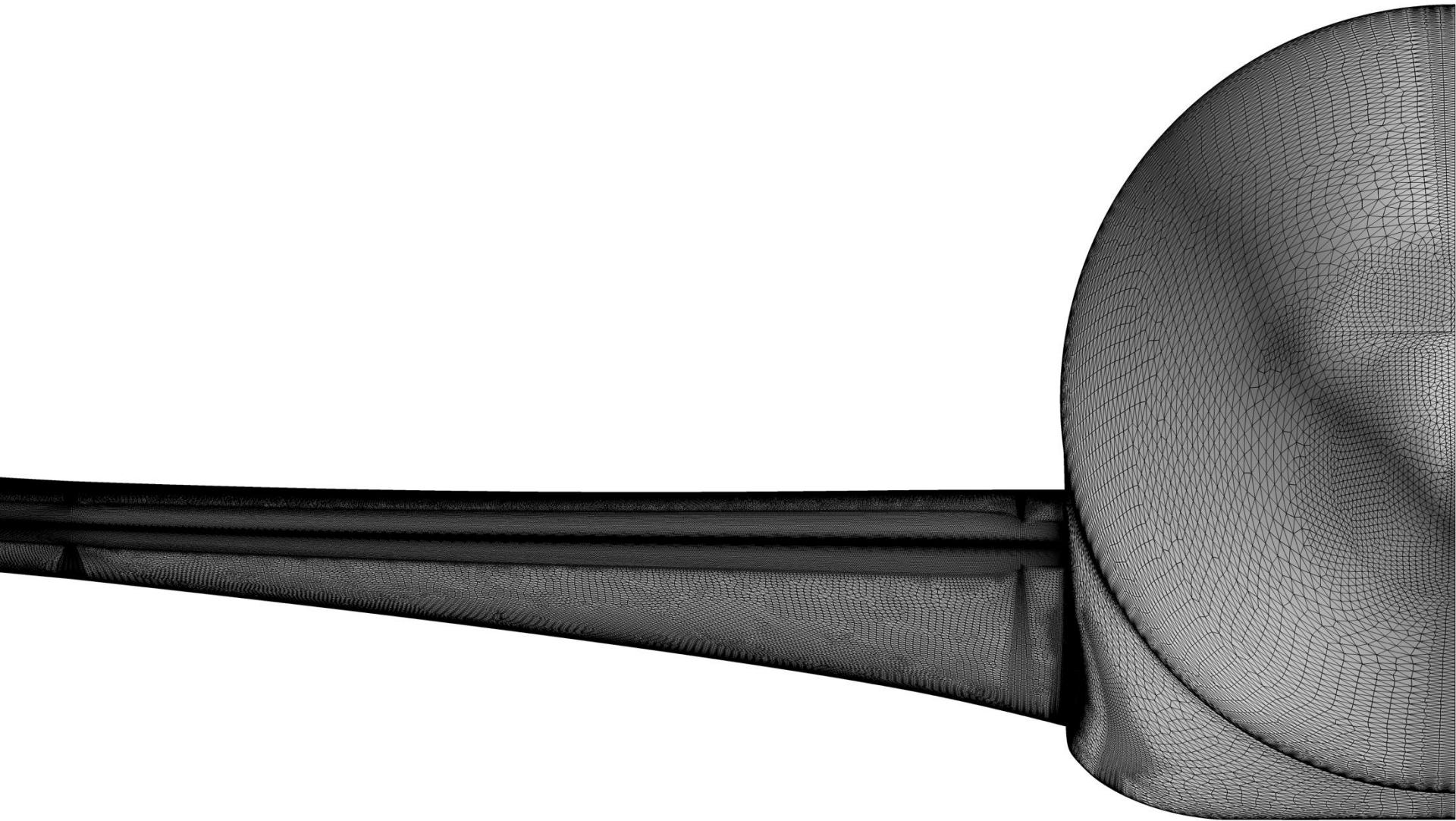
DPW-VII: JAXA Grid Topology



DPW-VII: JAXA L1 Tiny Grid



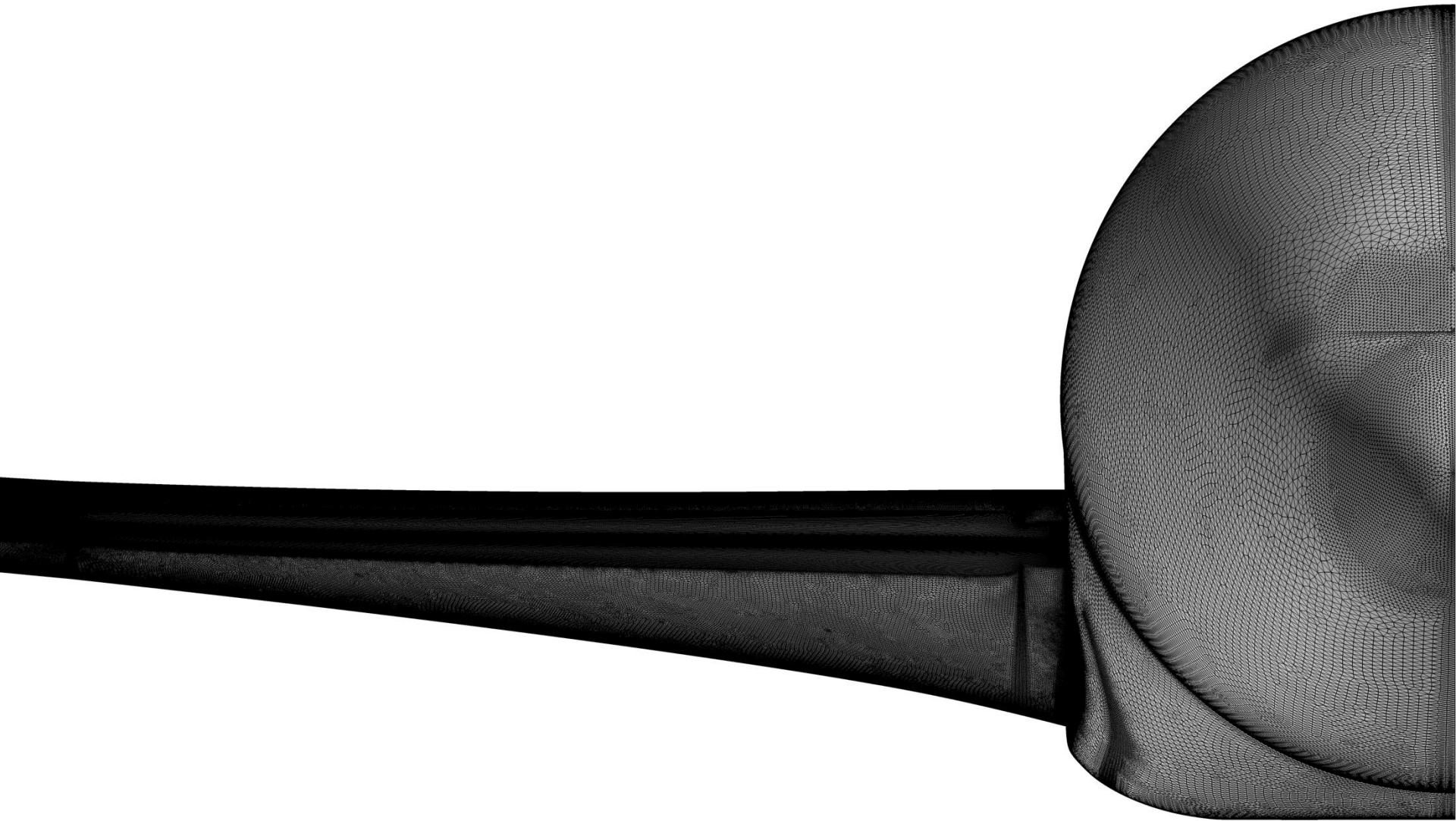
DPW-VII: JAXA L3 Medium Grid



DPW-VII: JAXA L5 Extra-Fine Grid



DPW-VII: JAXA L6 Ultra-Fine Grid



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DPW-VII: DLR Grid Family Statistics

Grid	Points /[10 ⁶]	Elements /[10 ⁶]	Surface Points /[10 ⁶]	y ₁ /[μm]	# Layers	# constant Cells @ Wall	Stretching Ratio in BL
T	11.70	31.59	0.506	6.560	53	2	1.2
C	25.01	64.33	0.837	4.374	53	3	1.2
M	47.06	130.7	1.317	3.280	52	4	1.2
F	76.51	224.1	1.931	2.624	52	5	1.2
X	118.8	367.9	2.726	2.187	52	6	1.2
U	164.5	534.2	3.485	1.874	53	7	1.2

Additional Grid Family built using Hybrid Meshing Approach:

- SOLAR for Surface Mesh
- ANSA (BETA CAE Systems, USA) for Boundary Layer / Farfield Mesh

Grids available on DPW-7 Website

Tiny

0.51×10^6 Points

Coarse

0.84×10^6 Points

Medium

1.3×10^6 Points

Fine

1.9×10^6 Points

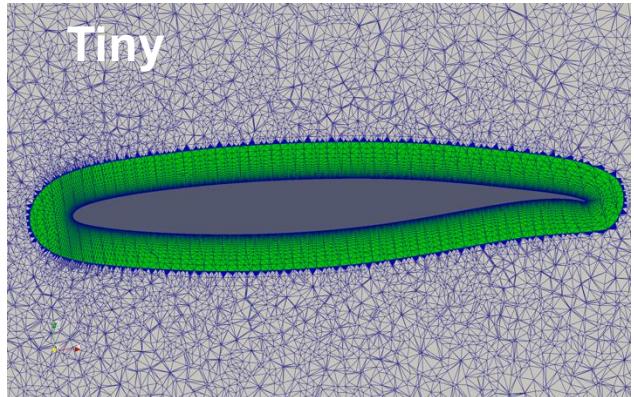
Extra Fine

2.7×10^6 Point

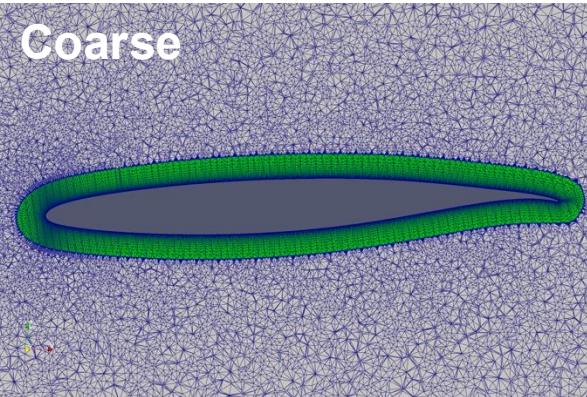
Ultra Fine

3.5×10^6 Point

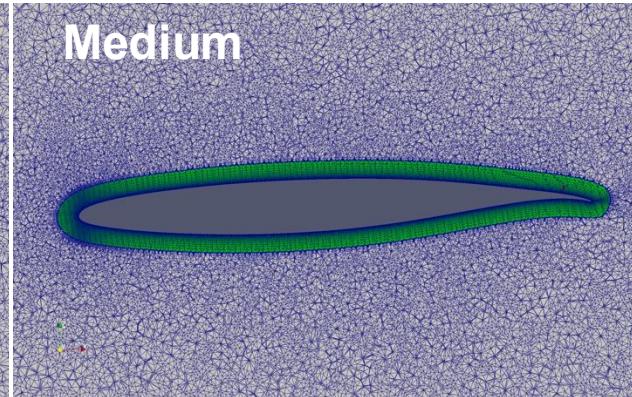
Tiny



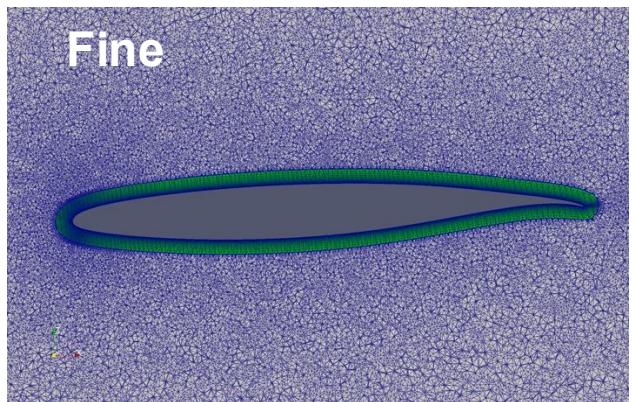
Coarse



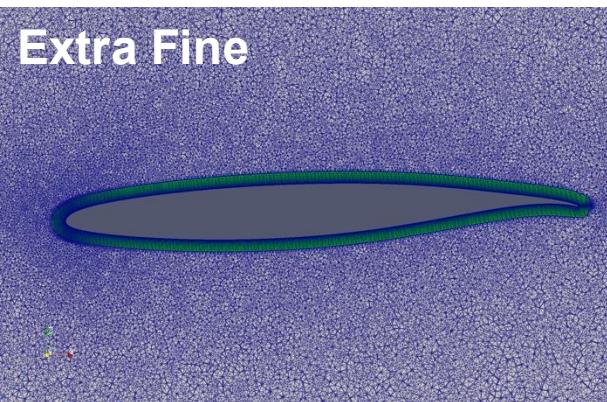
Medium



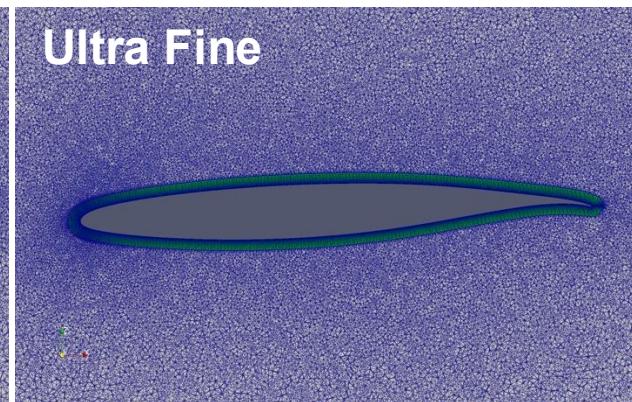
Fine



Extra Fine

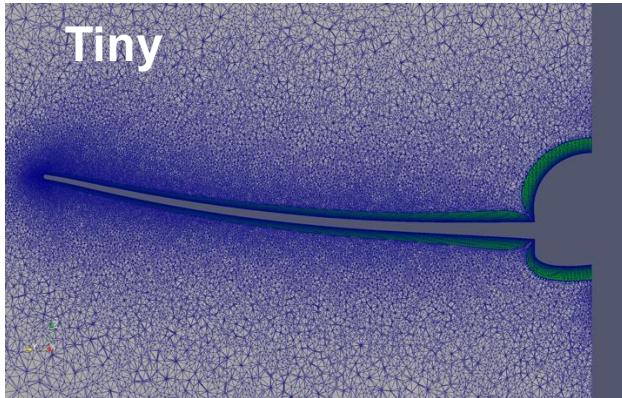


Ultra Fine

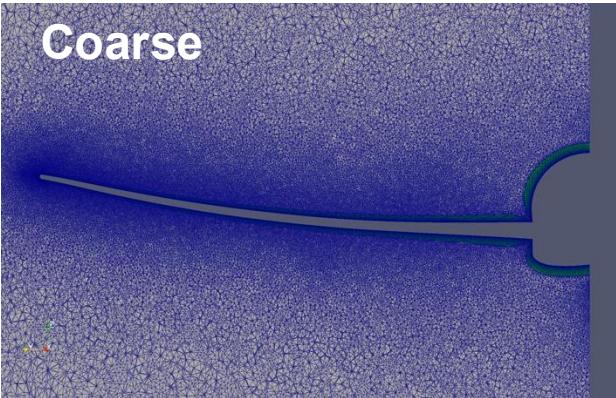


Boundary Layer Height reduced from coarser to finer Meshes
due to automatic Stopping when Cell Aspect Ratio reaches 1

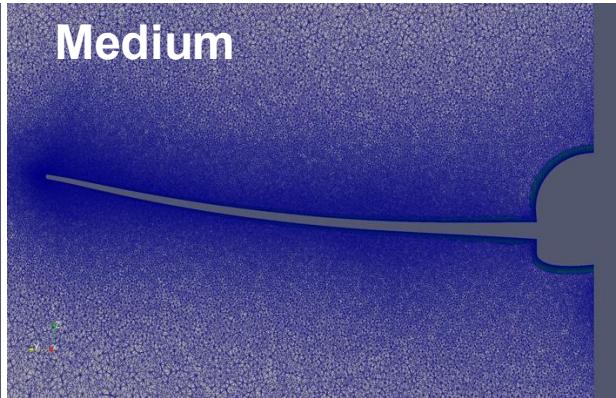
Tiny



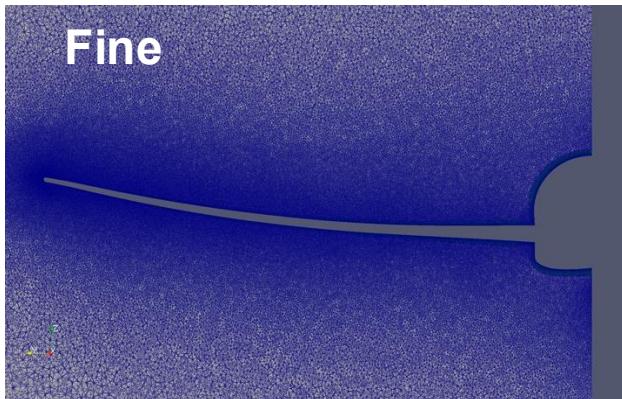
Coarse



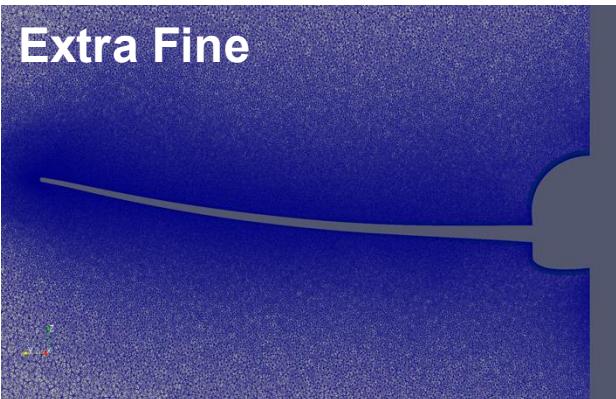
Medium



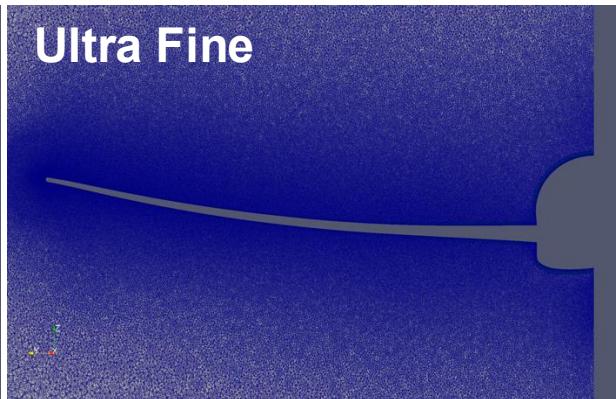
Fine



Extra Fine



Ultra Fine



Questions?

DPW-VII Workshop
25-26 June 2022

AIAA Aviation 2022
Applied Aerodynamics Conference

Chicago, IL

