



Applied Aerodynamics
Technical Committee

7th CFD Drag Prediction Workshop

Chicago, IL – June 2022



Trailing Edge Separation



- **Trailing Edge Separation**
 - **17 Submitted Data Sets (*as of June 24*)**

Grids

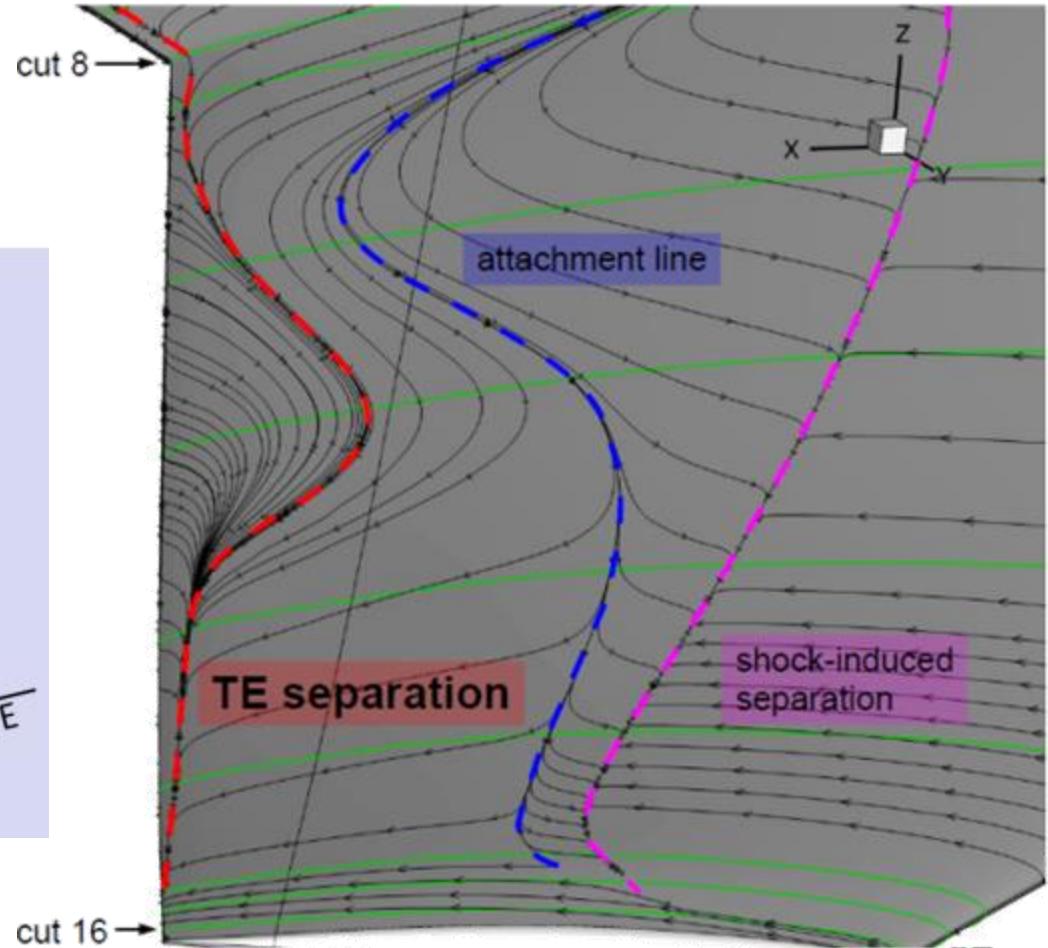
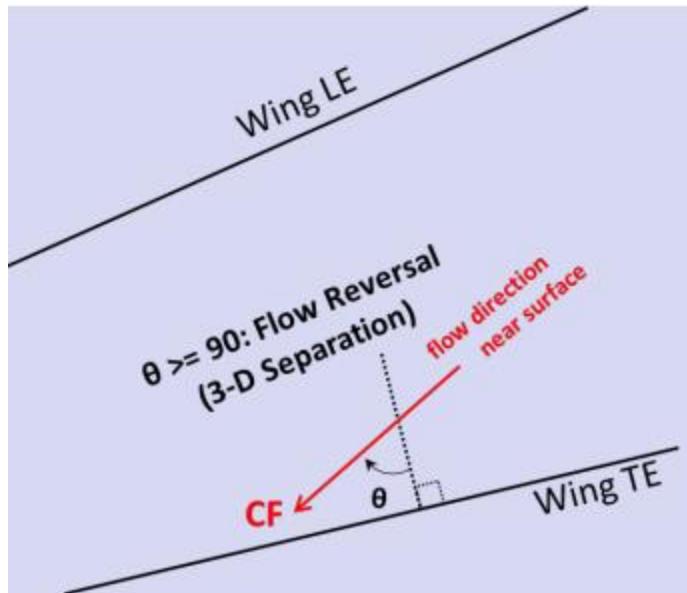
11 Unstructured
3 Overset
2 Multi-block
1 Hybrid Unstr/Cart

Turbulence Models

3 SA (RC)
10 SA QCR (RC)
1 AMM QCR
1 SST
2 EARSM

Trailing Edge Separation

- Criteria: C_f normal to TE < 0
 - difficult to detect at higher α





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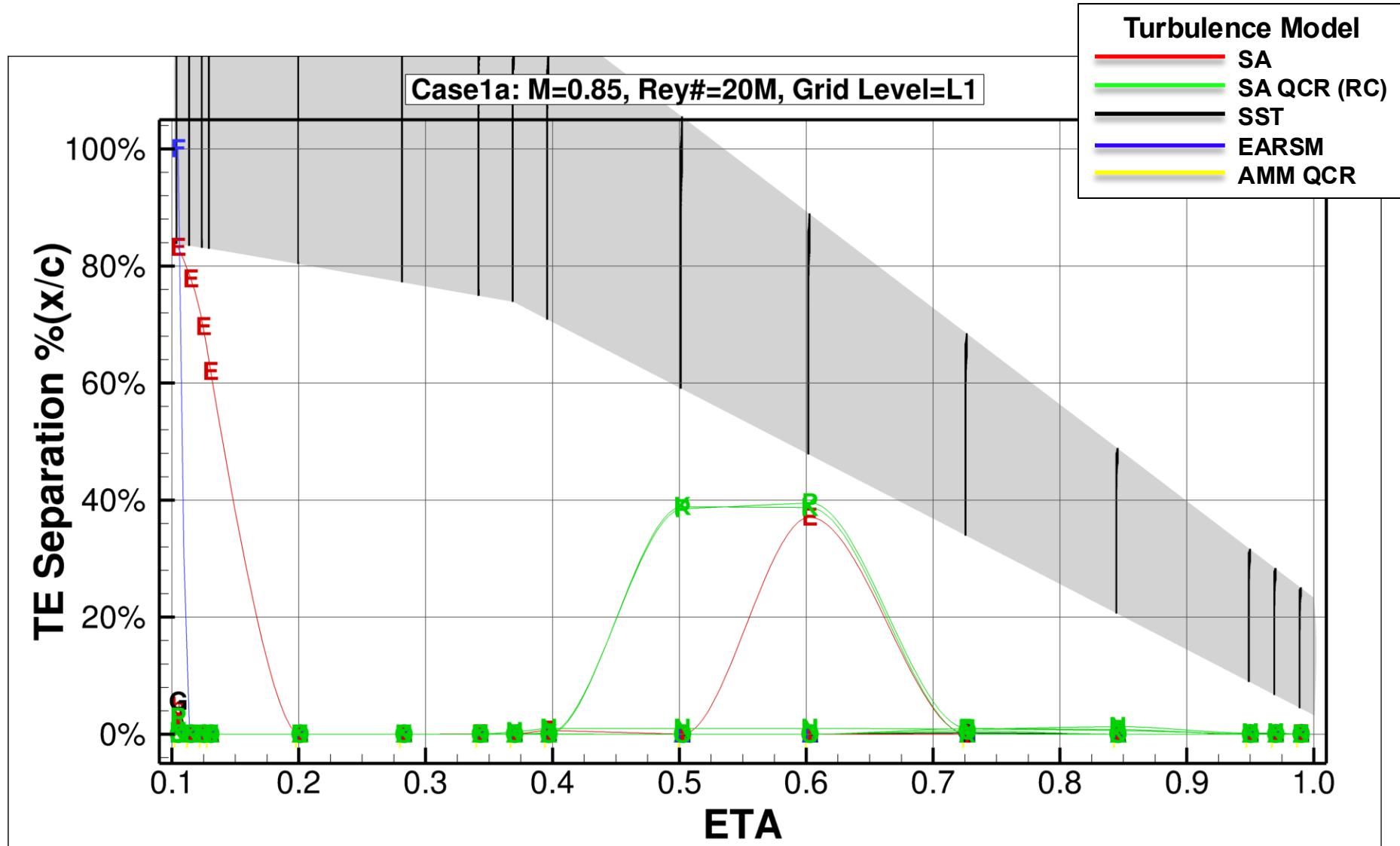
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Case 1a

CRM Wing-Body Grid Convergence Study

$M = 0.85 \mid C_L = 0.58 \mid Rey\# = 20M$

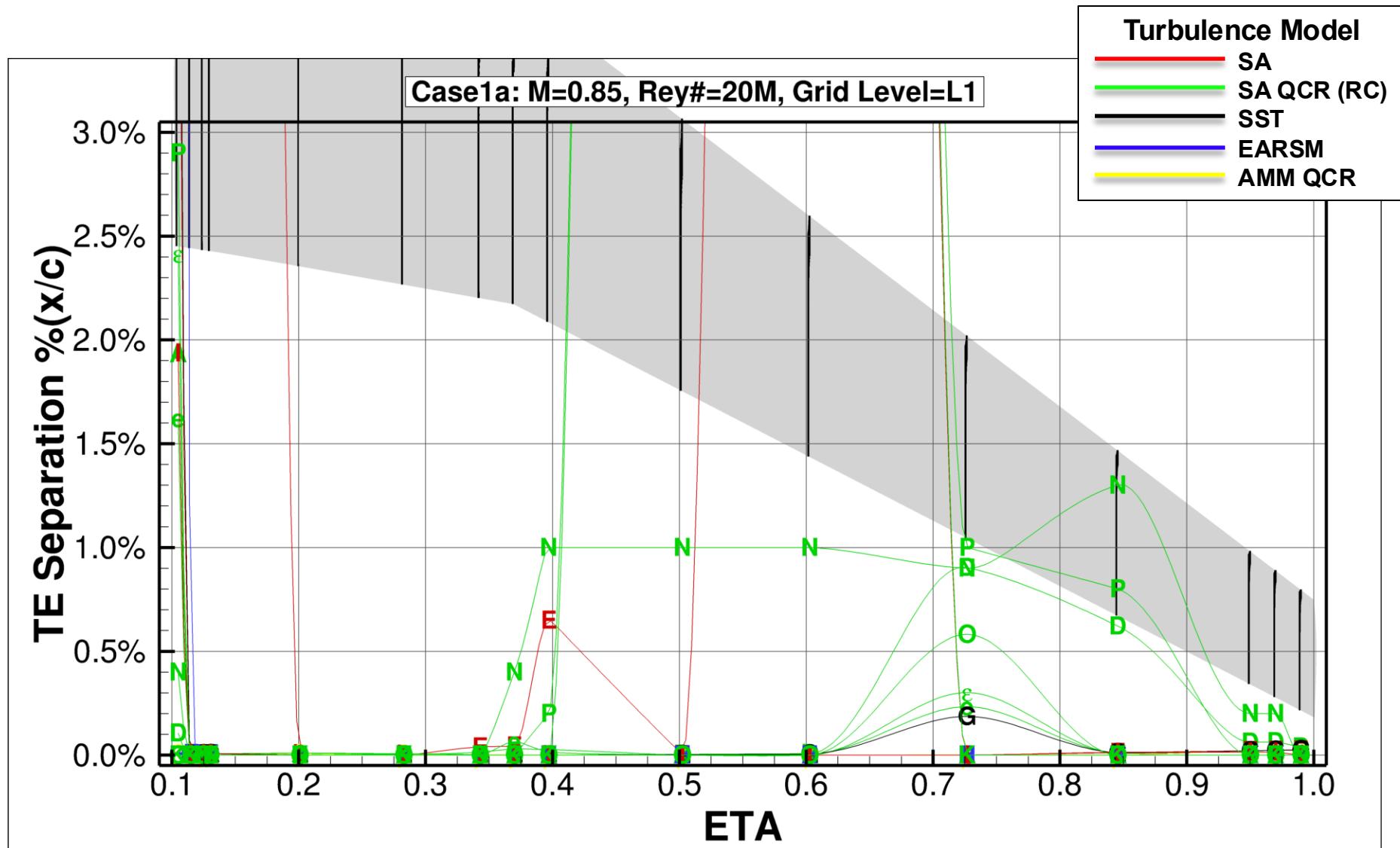


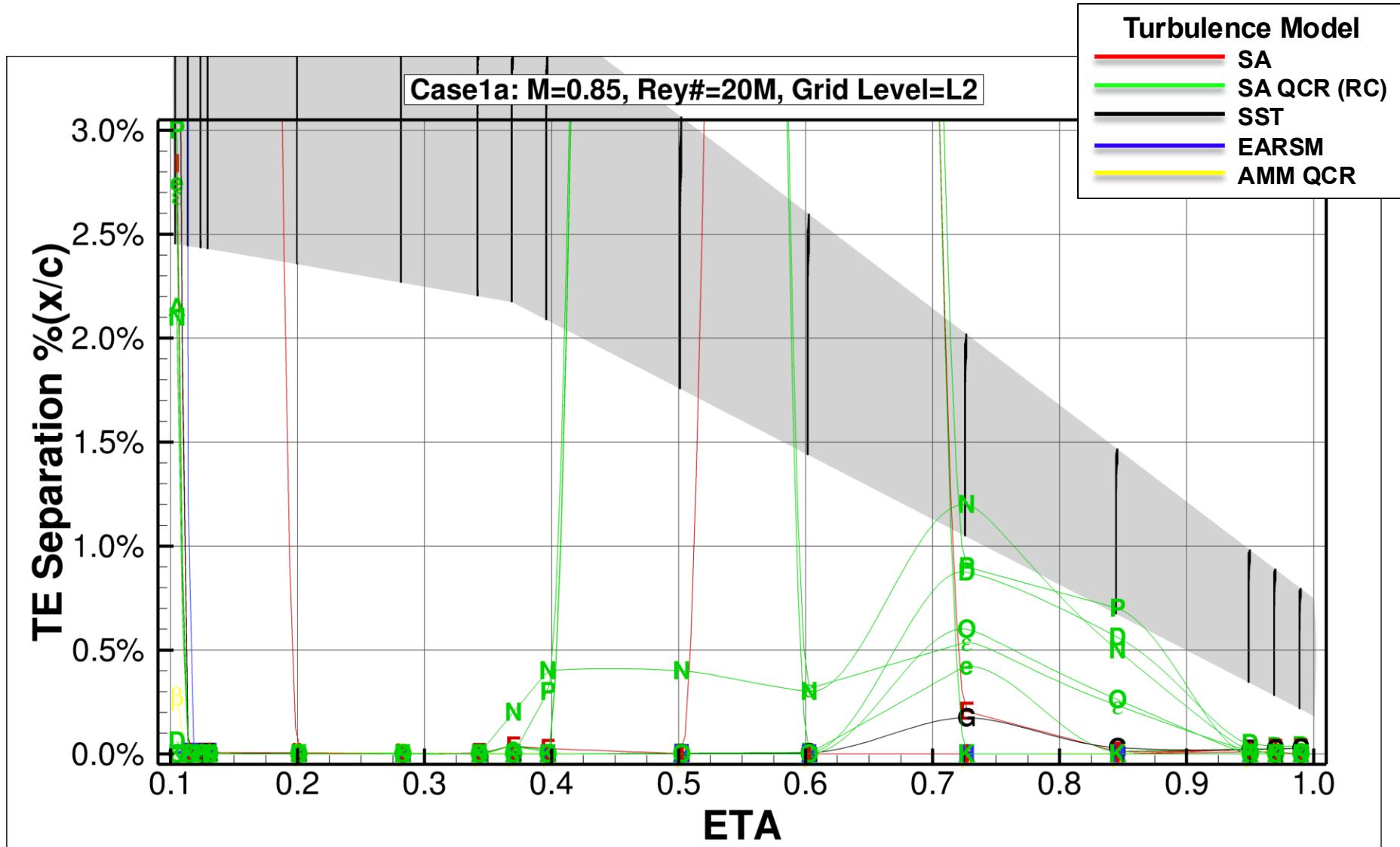
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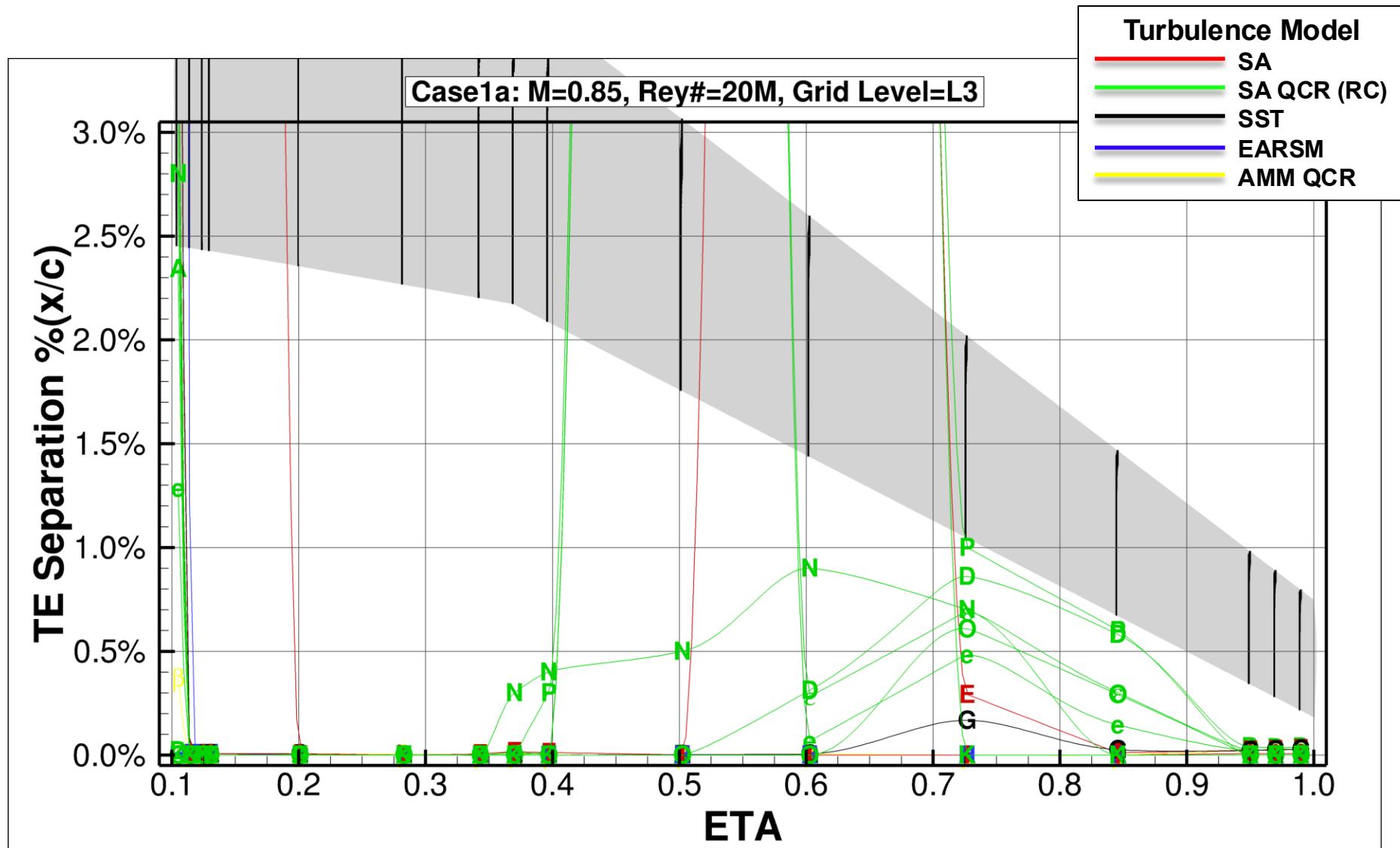


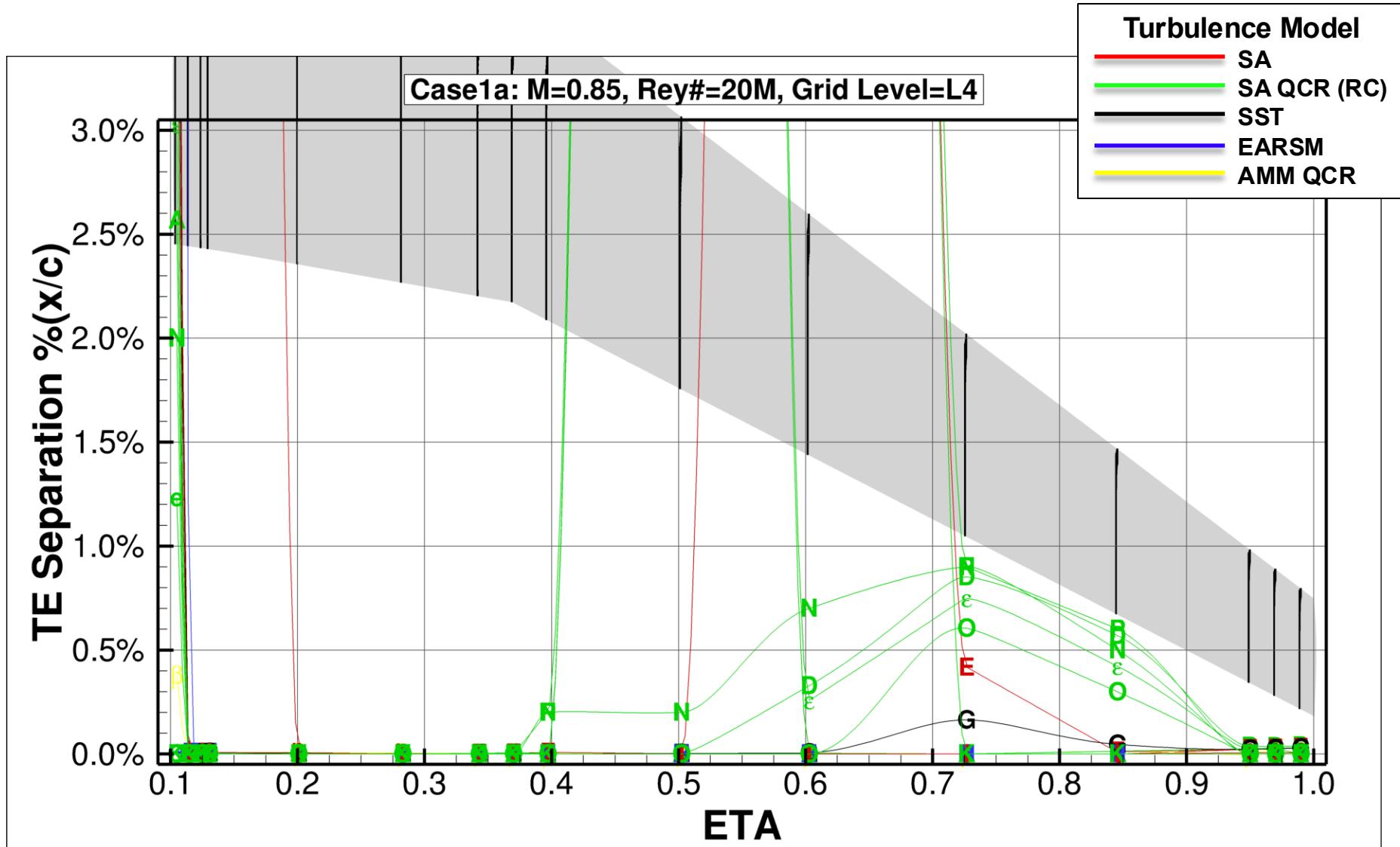
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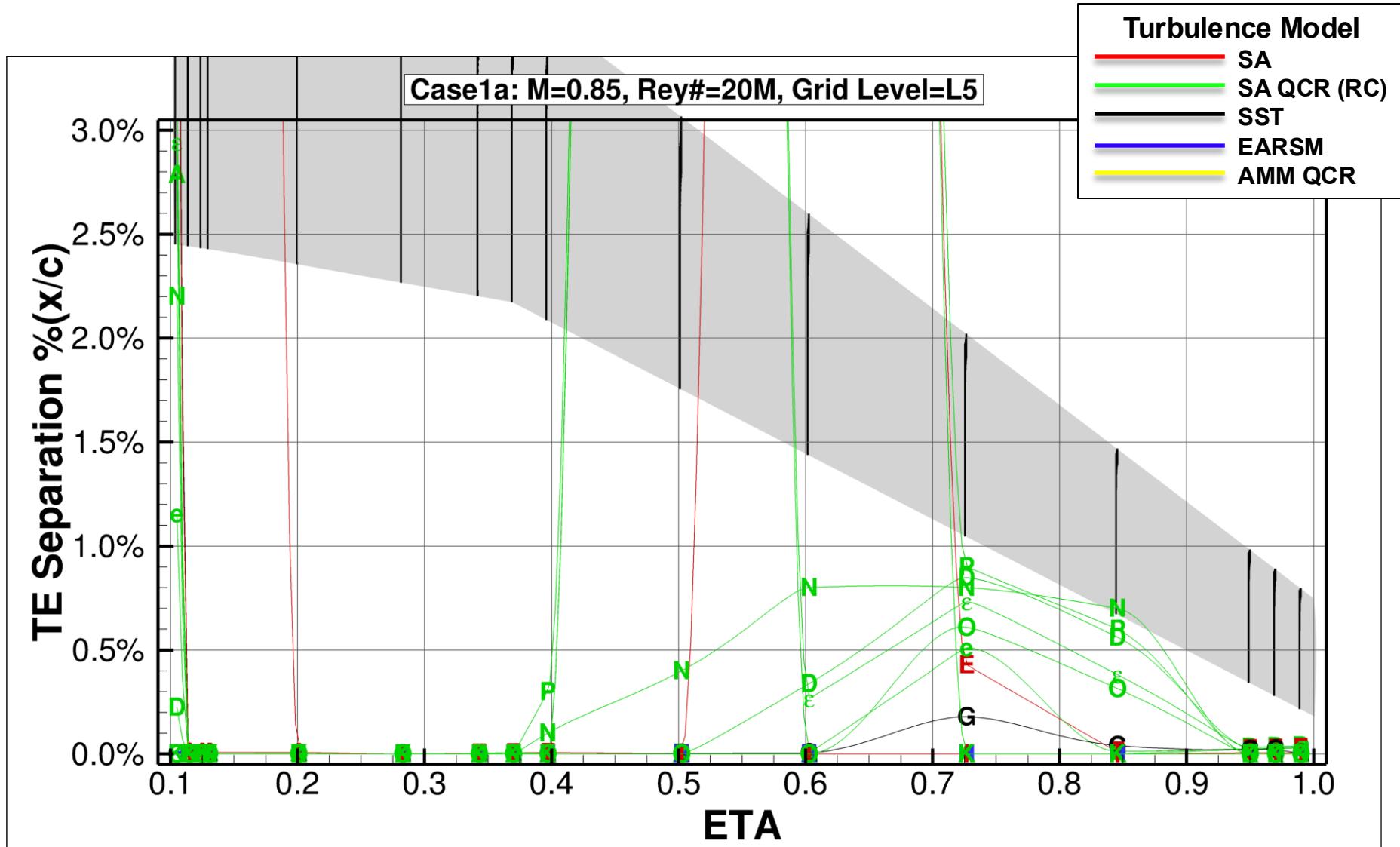
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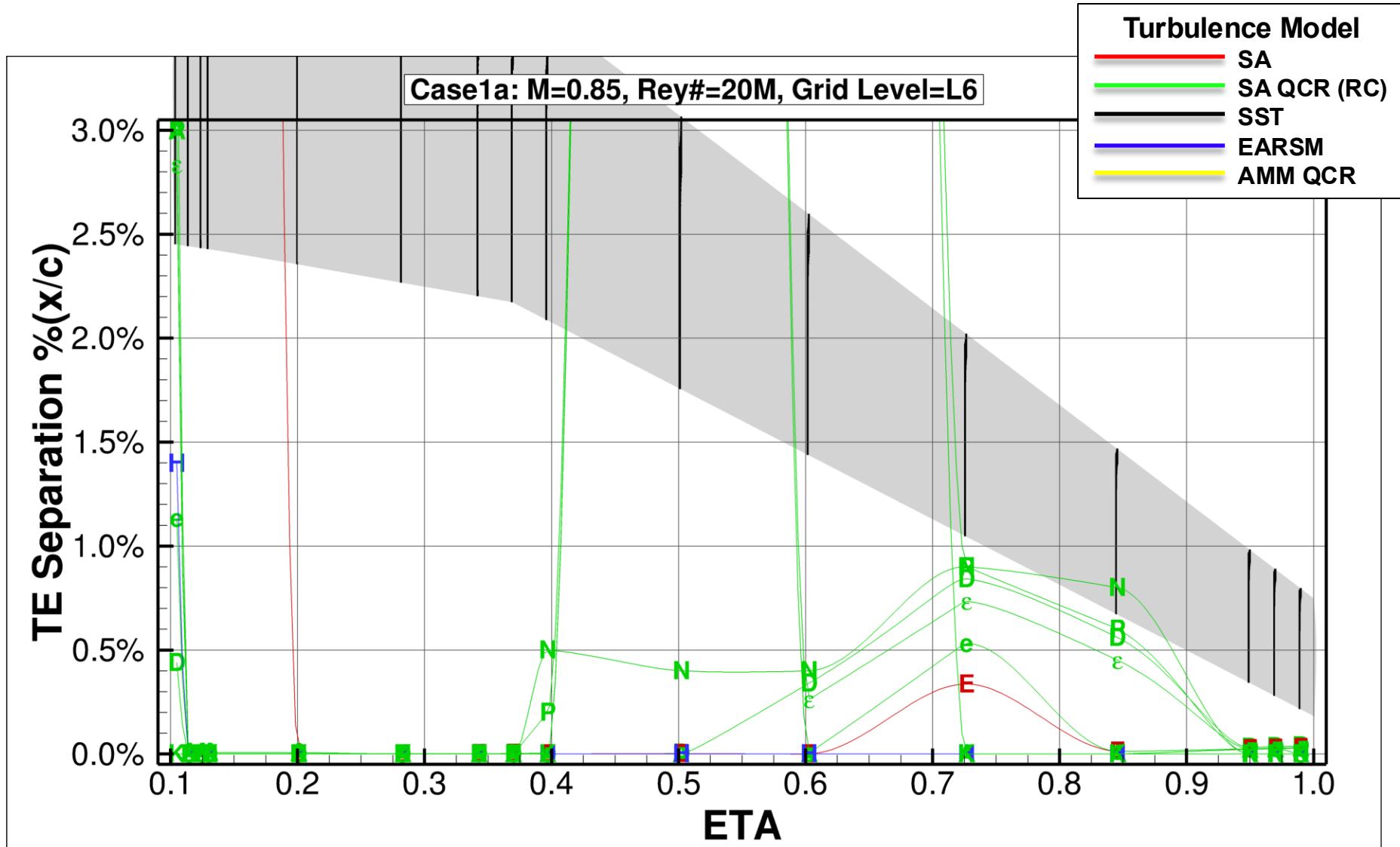


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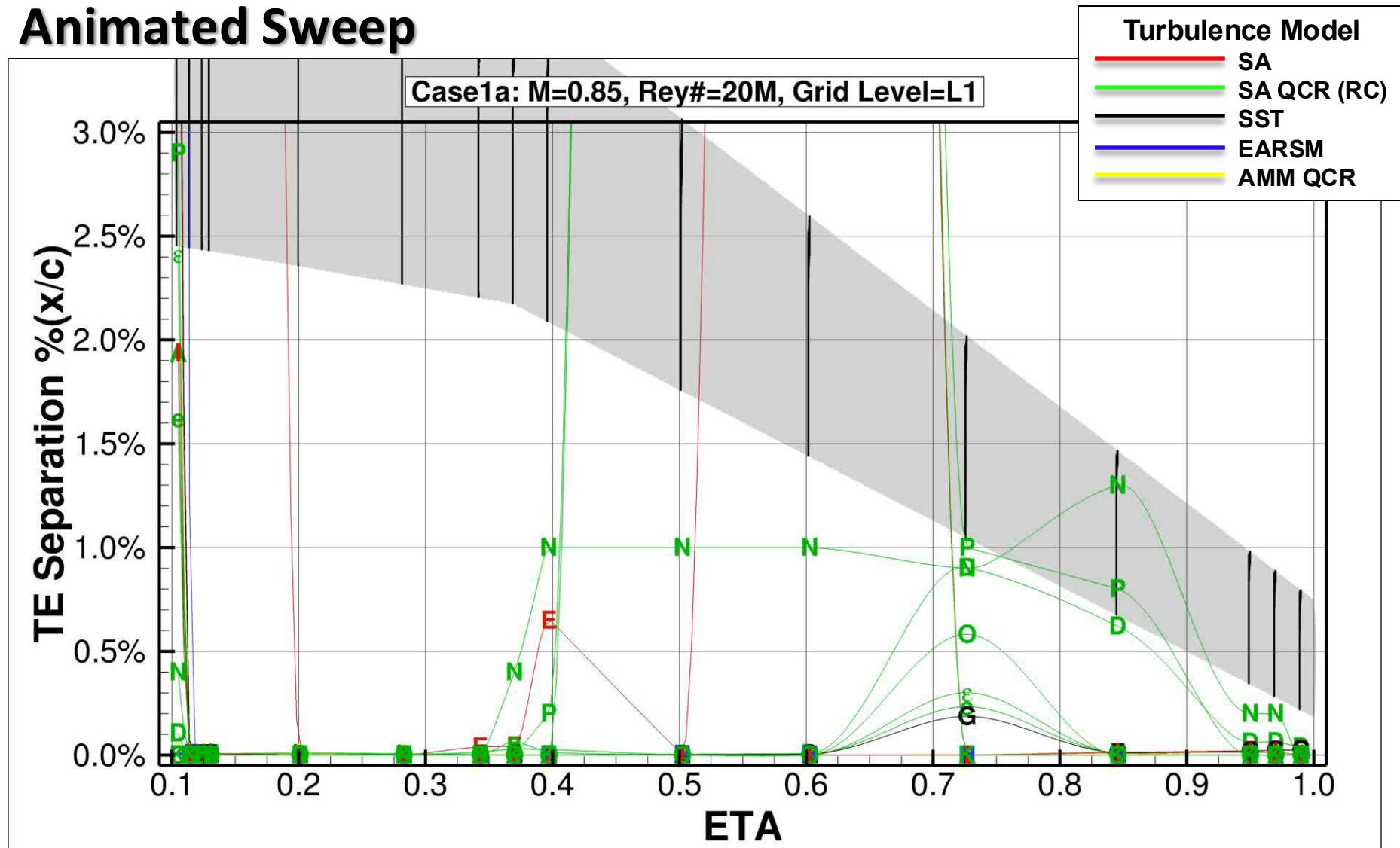








Animated Sweep





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Case 1b

CRM Wing-Body Grid Convergence Study

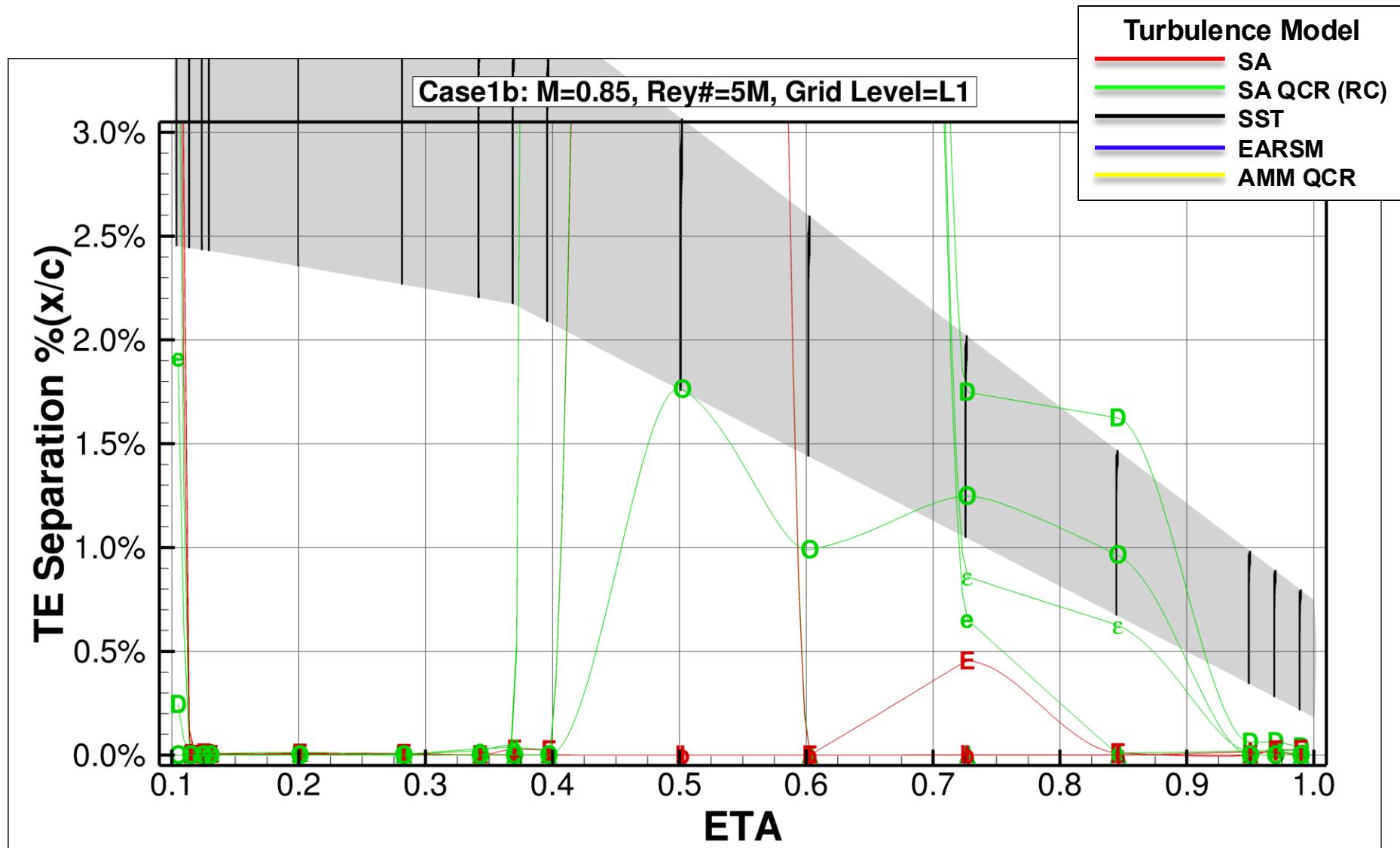
$M = 0.85$ | $C_L = 0.58$ | $Rey\# = 5M$

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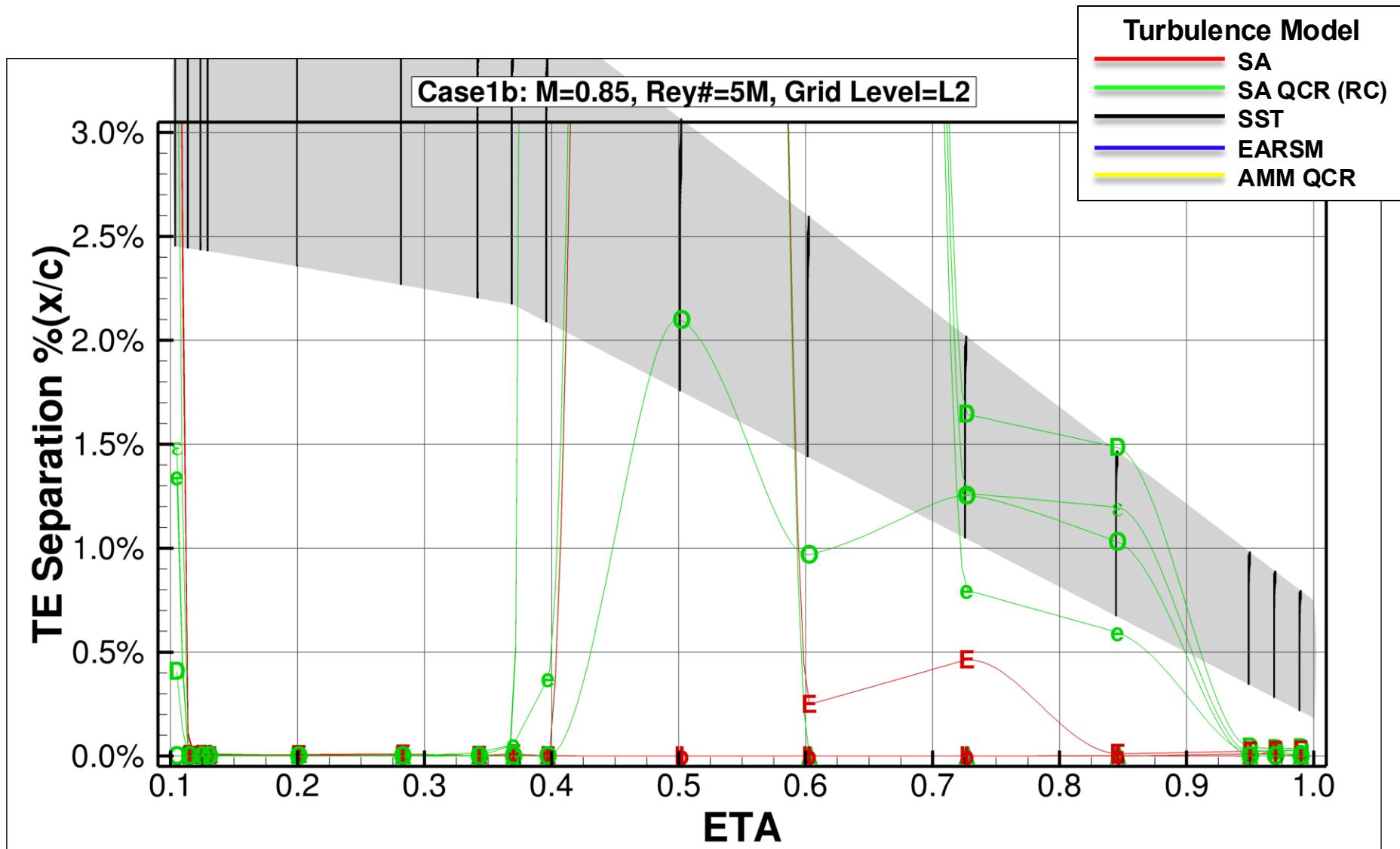


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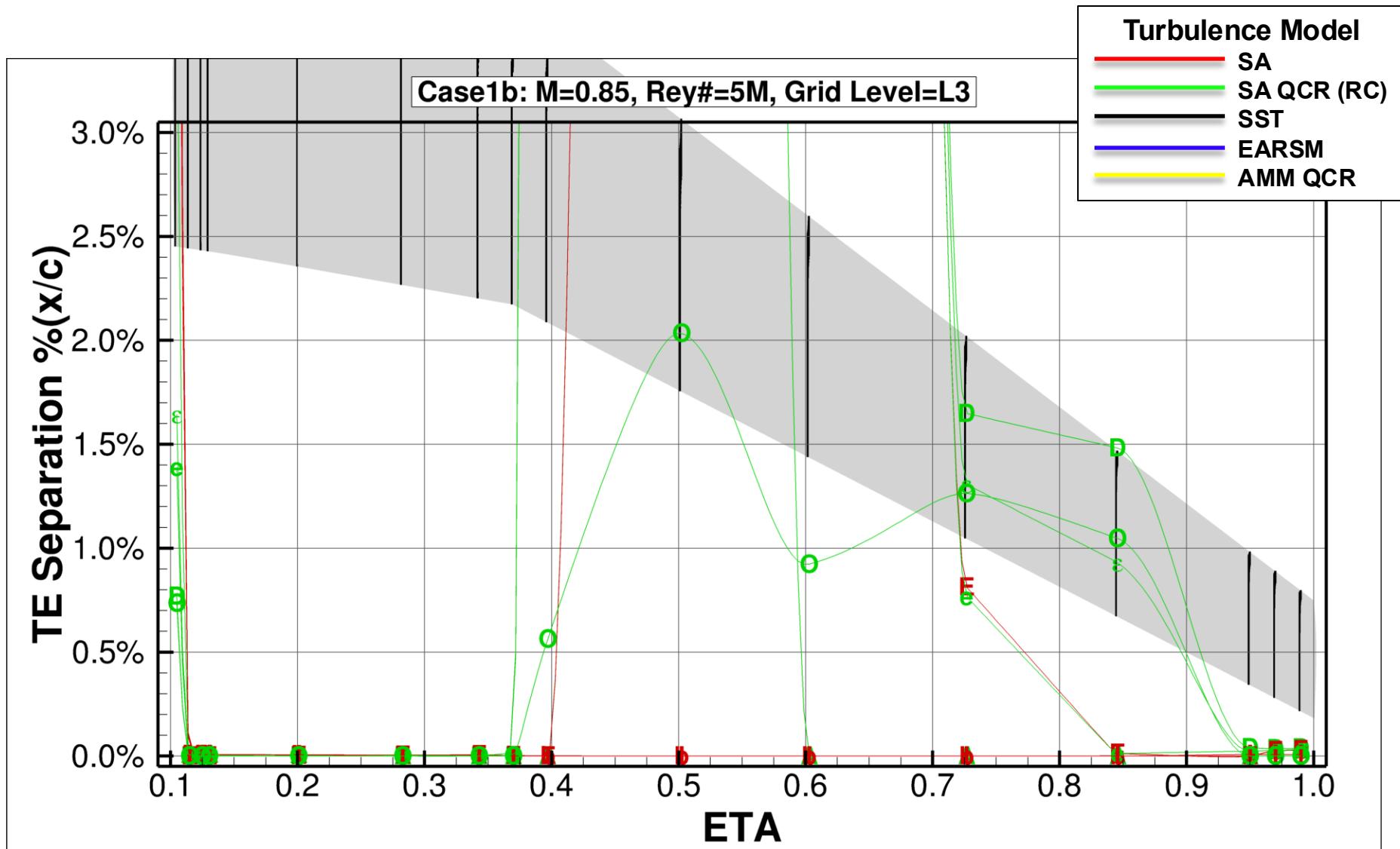


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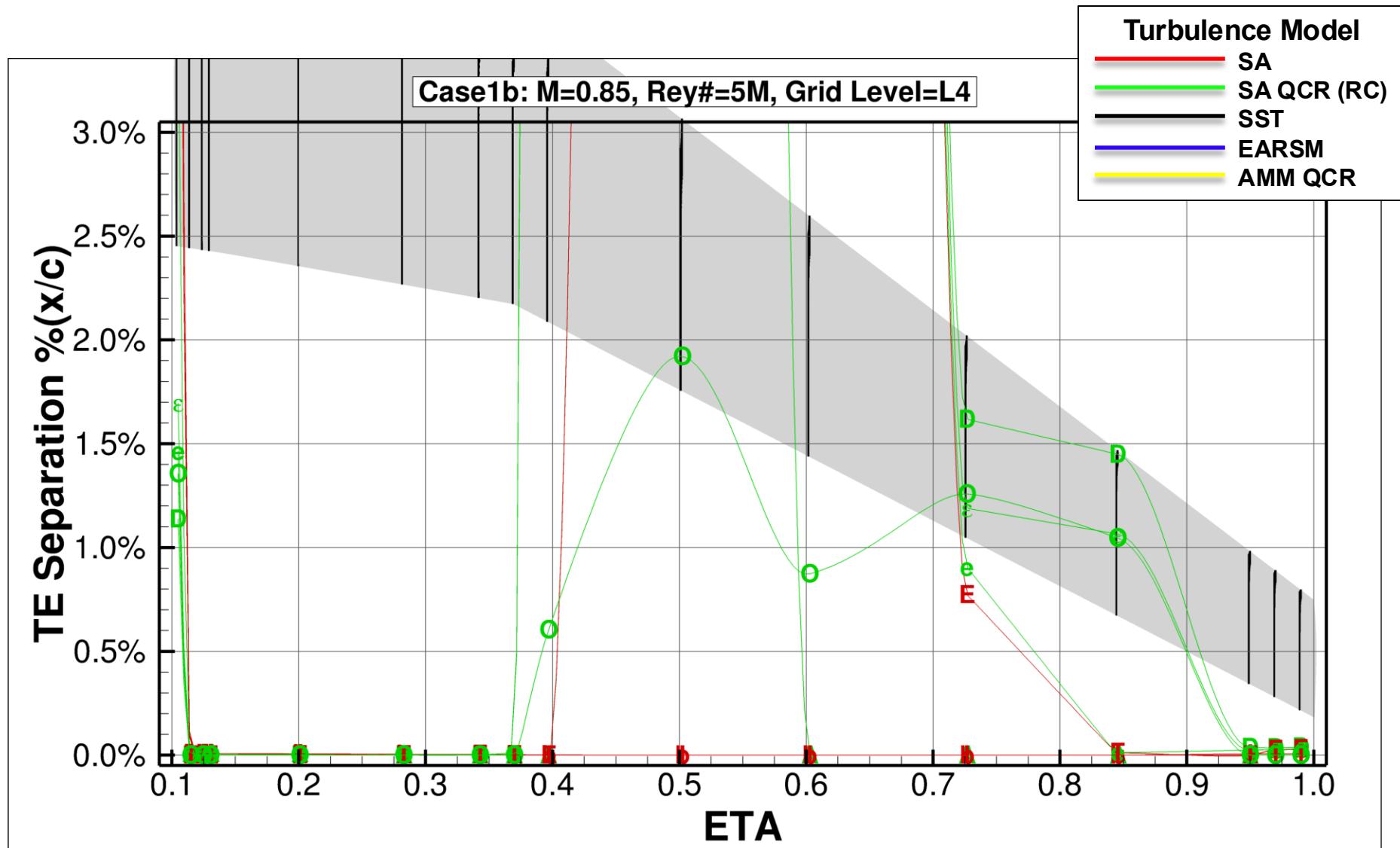


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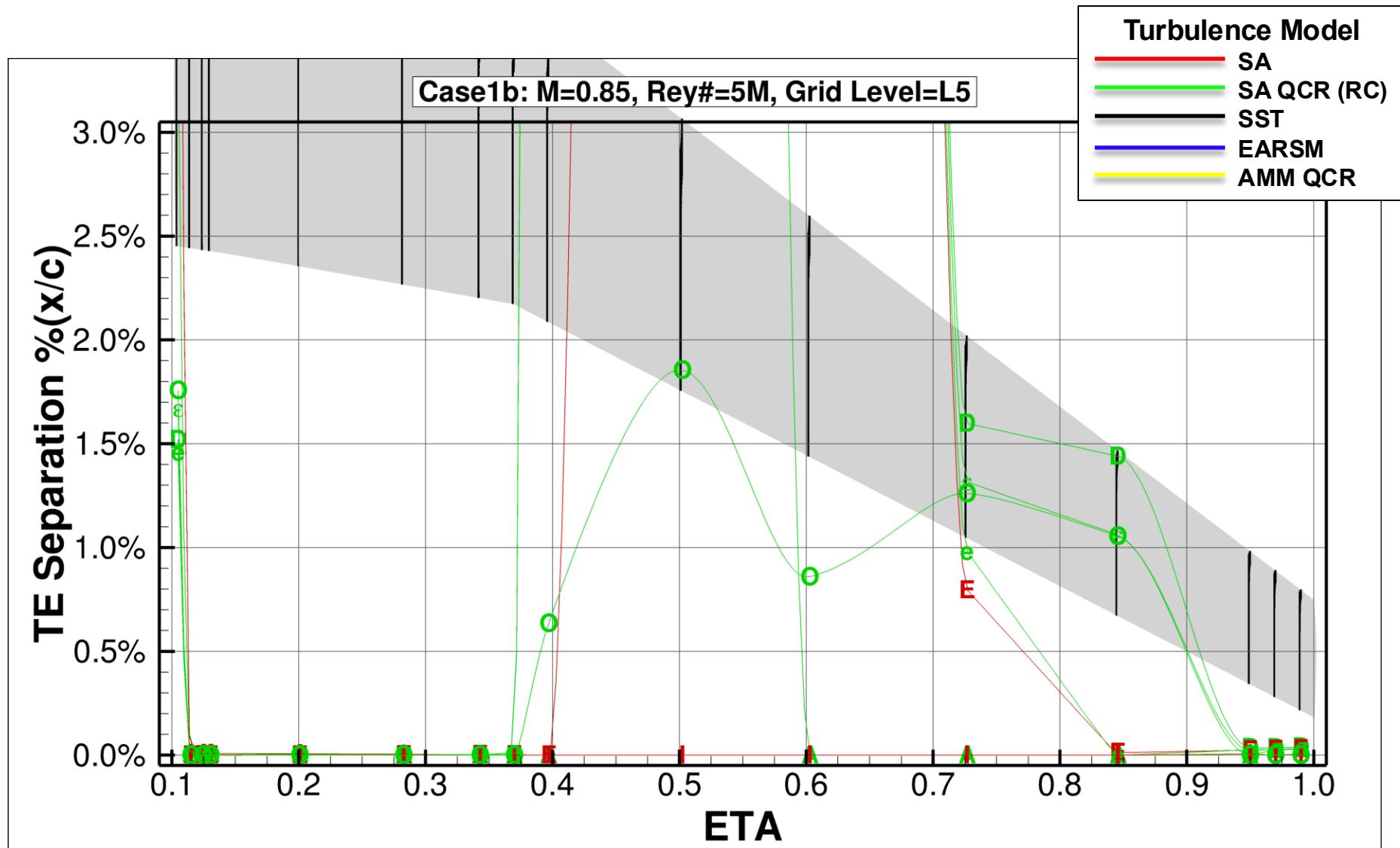


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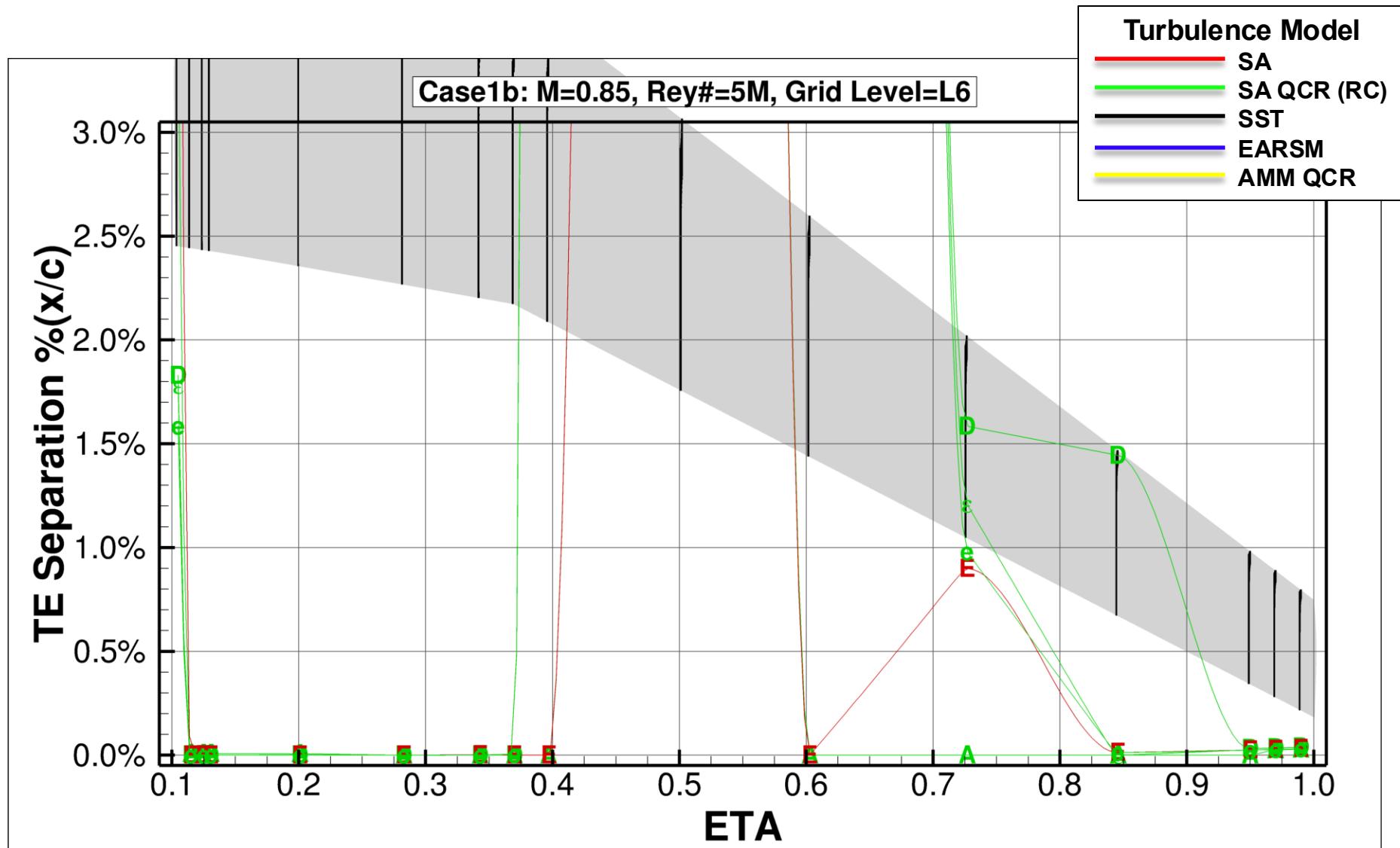


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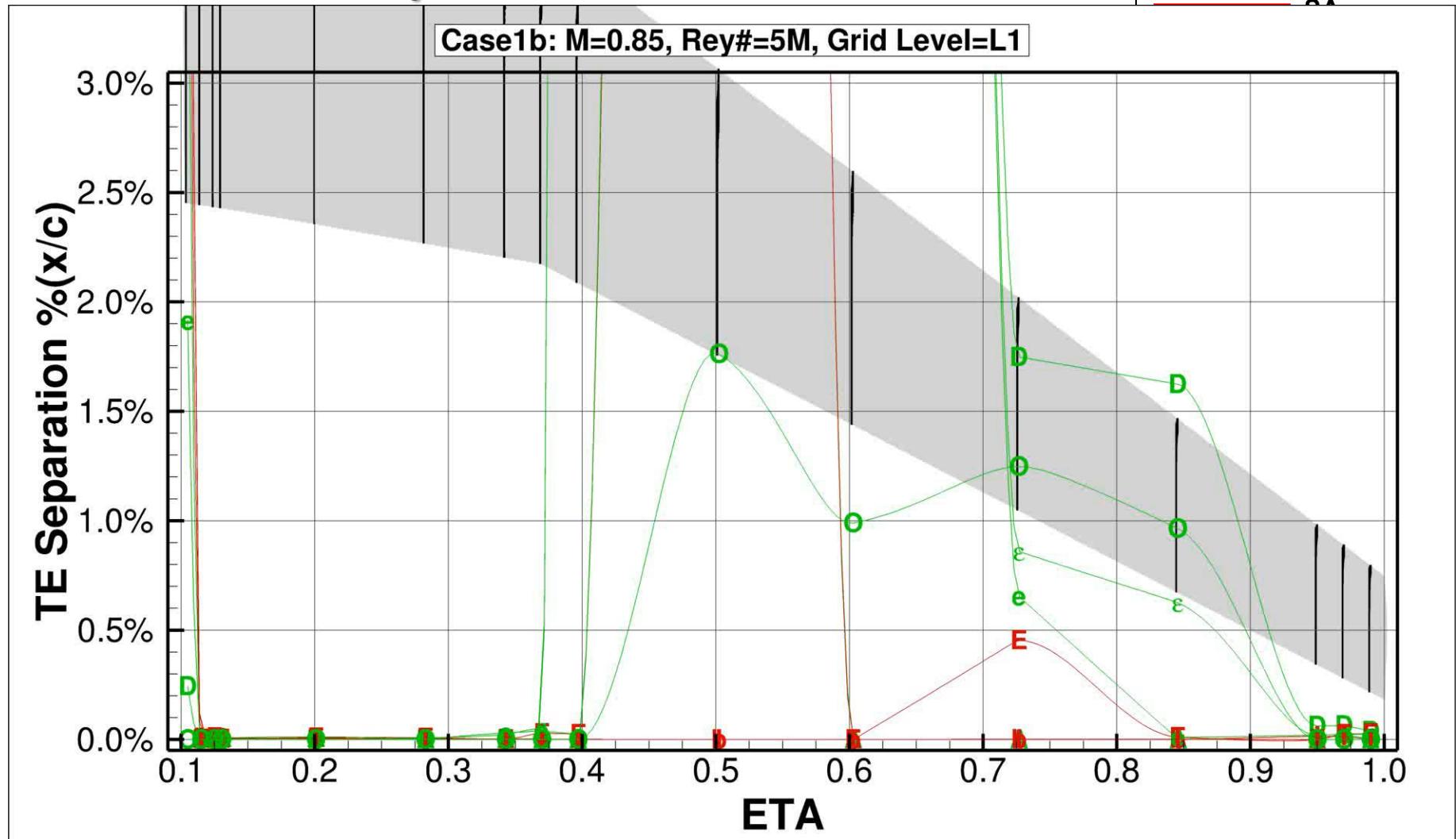


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Turbulence Model





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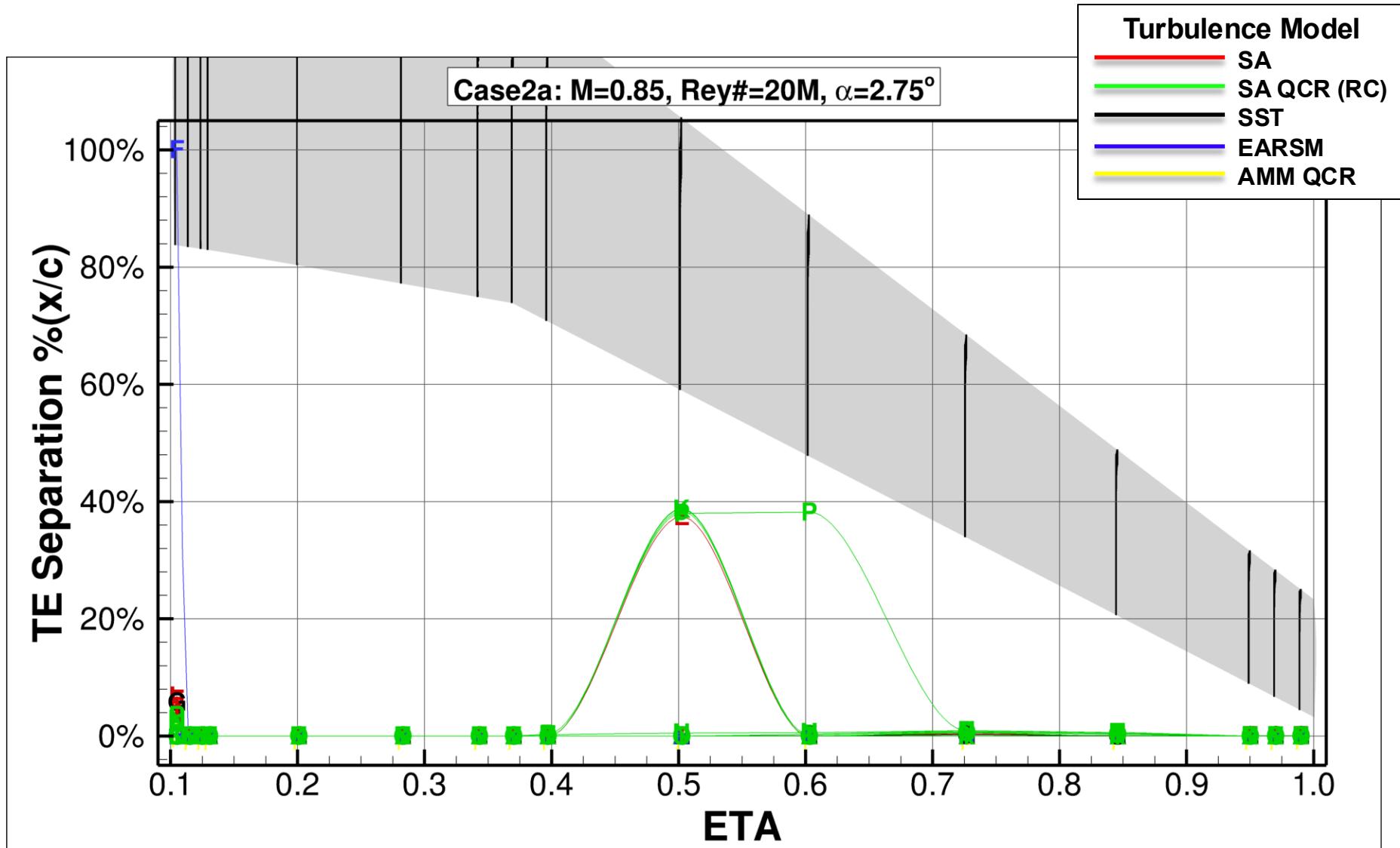
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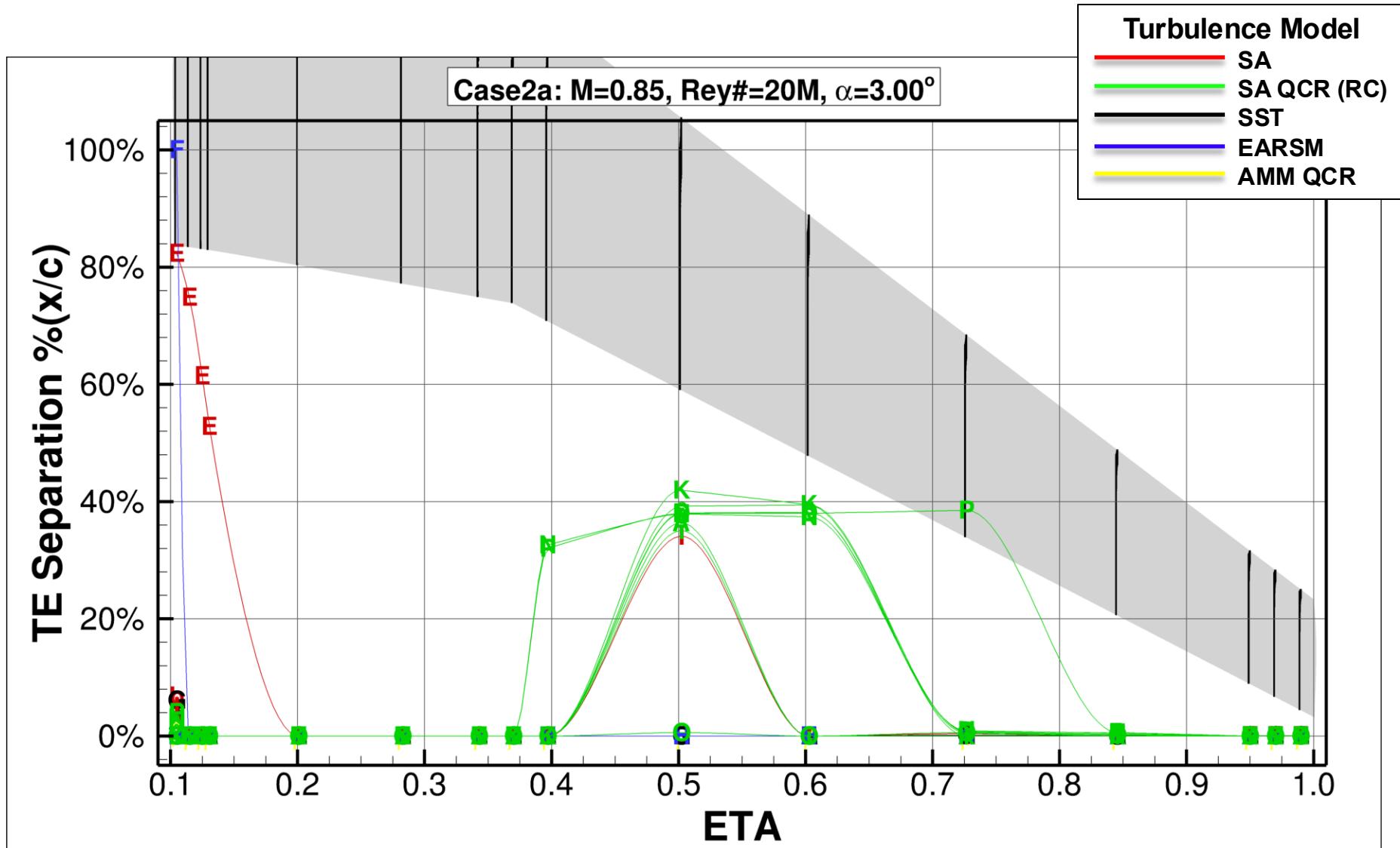


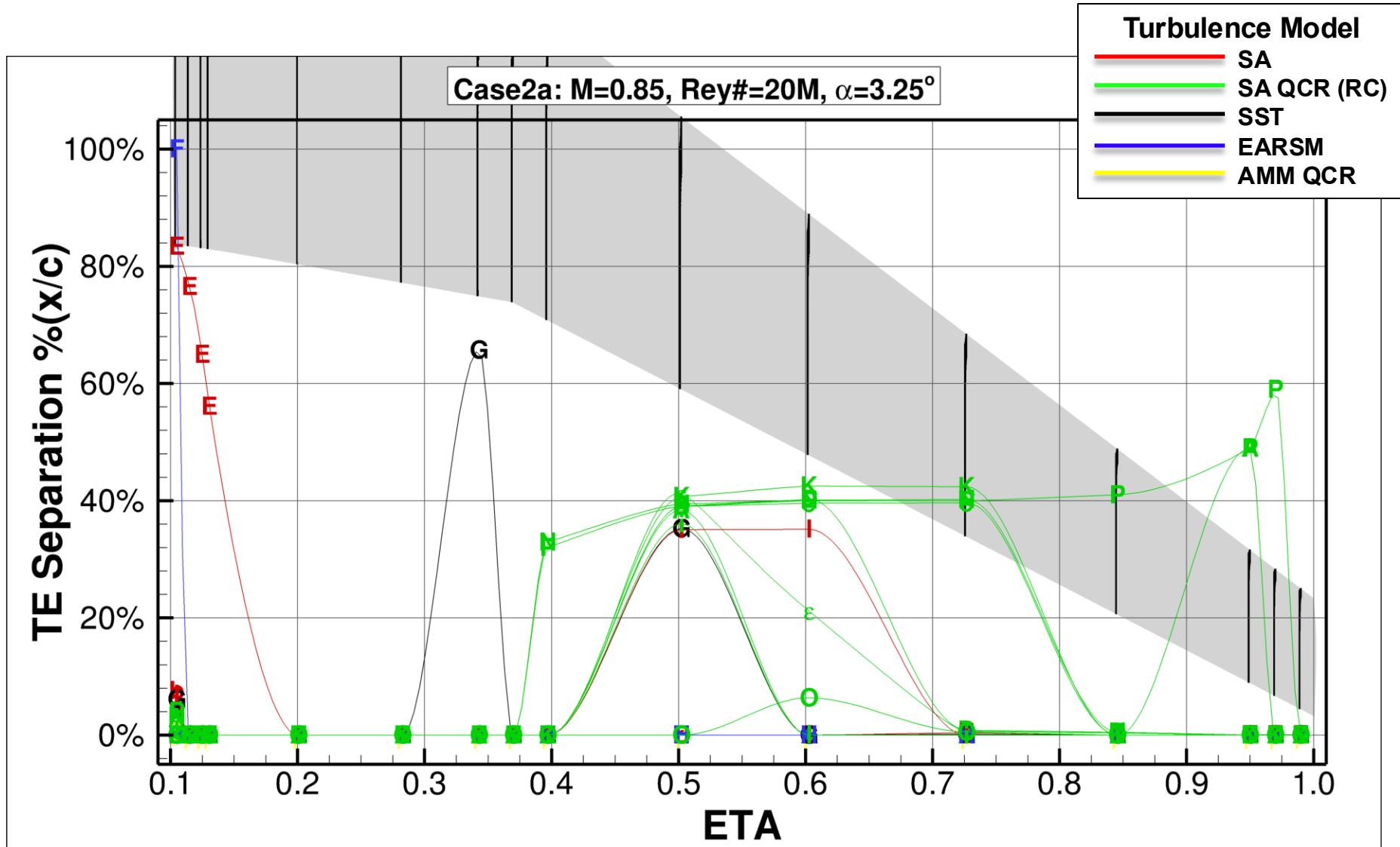
Case 2a

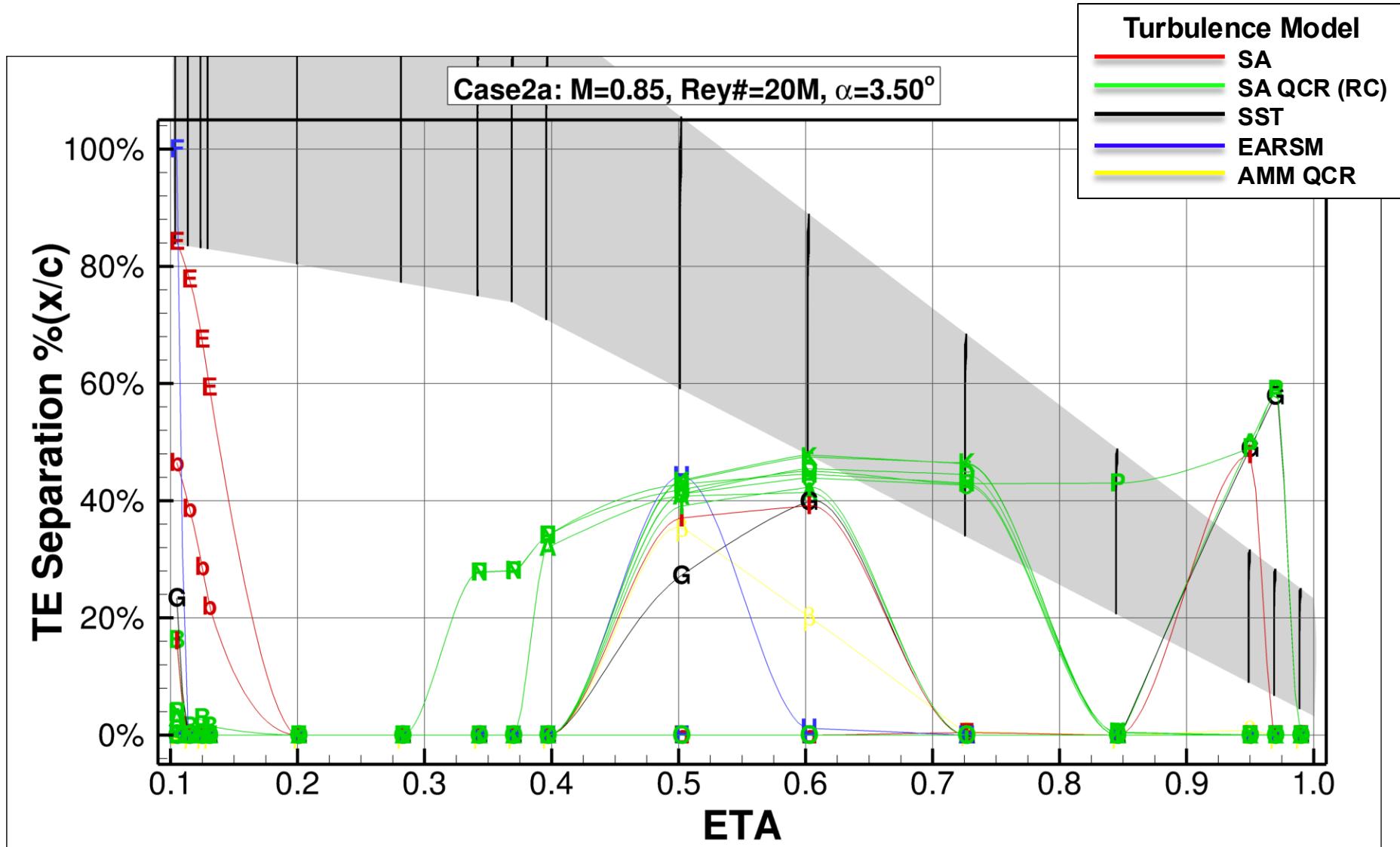
CRM Wing-Body Alpha Sweep

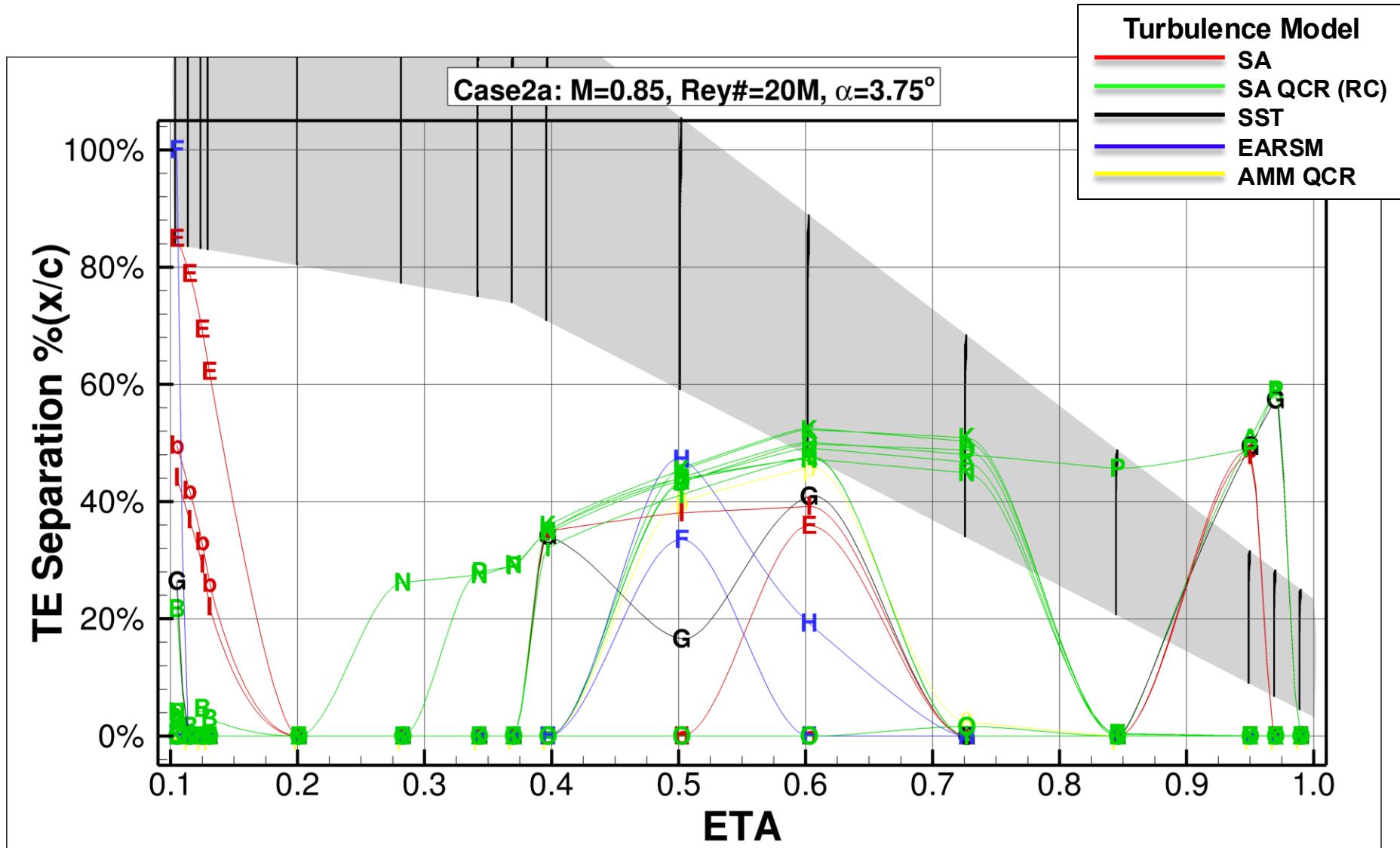
$M = 0.85$ | $\alpha = 2.75\text{--}4.25^\circ$ | $\text{Rey\#} = 20M$

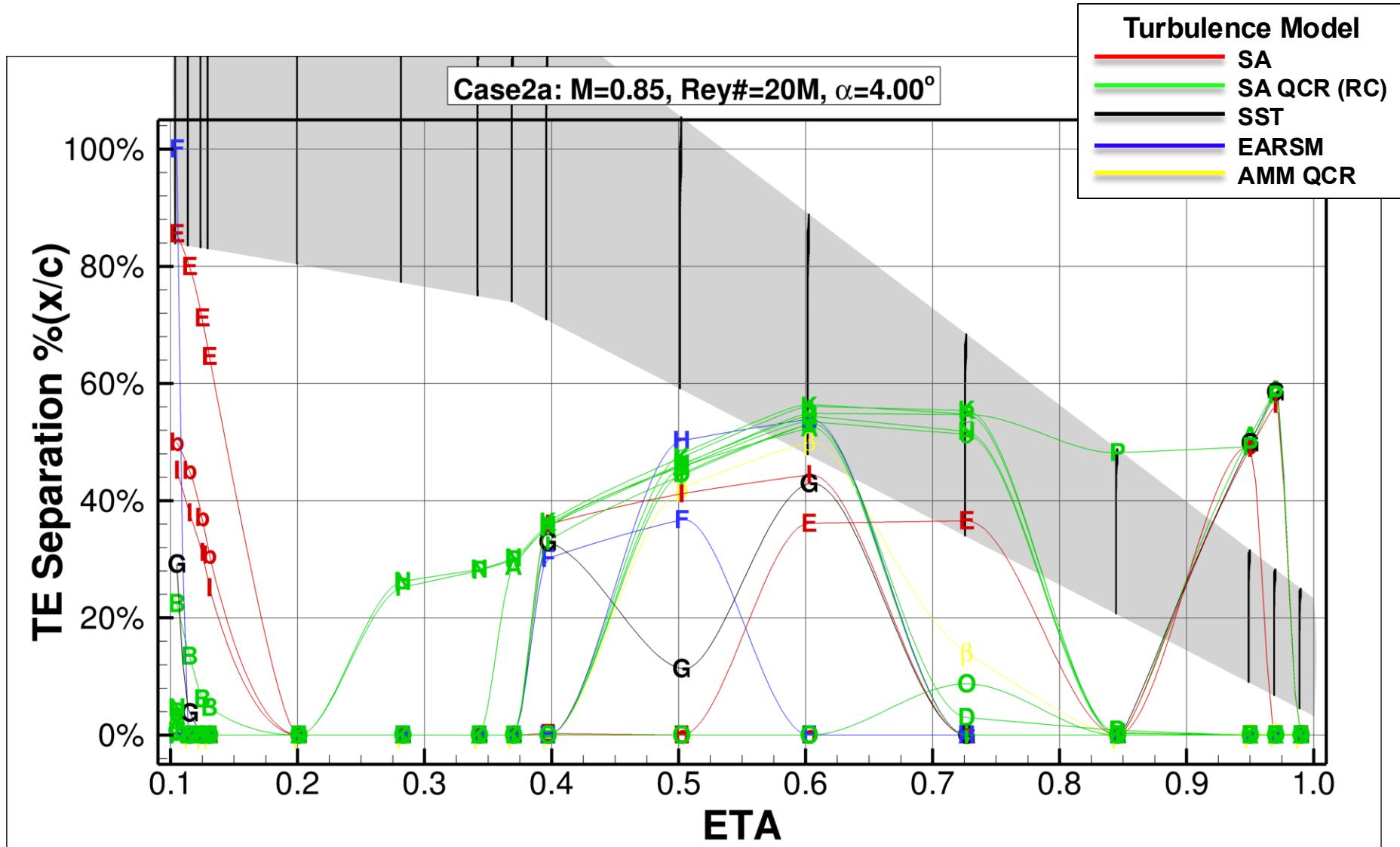


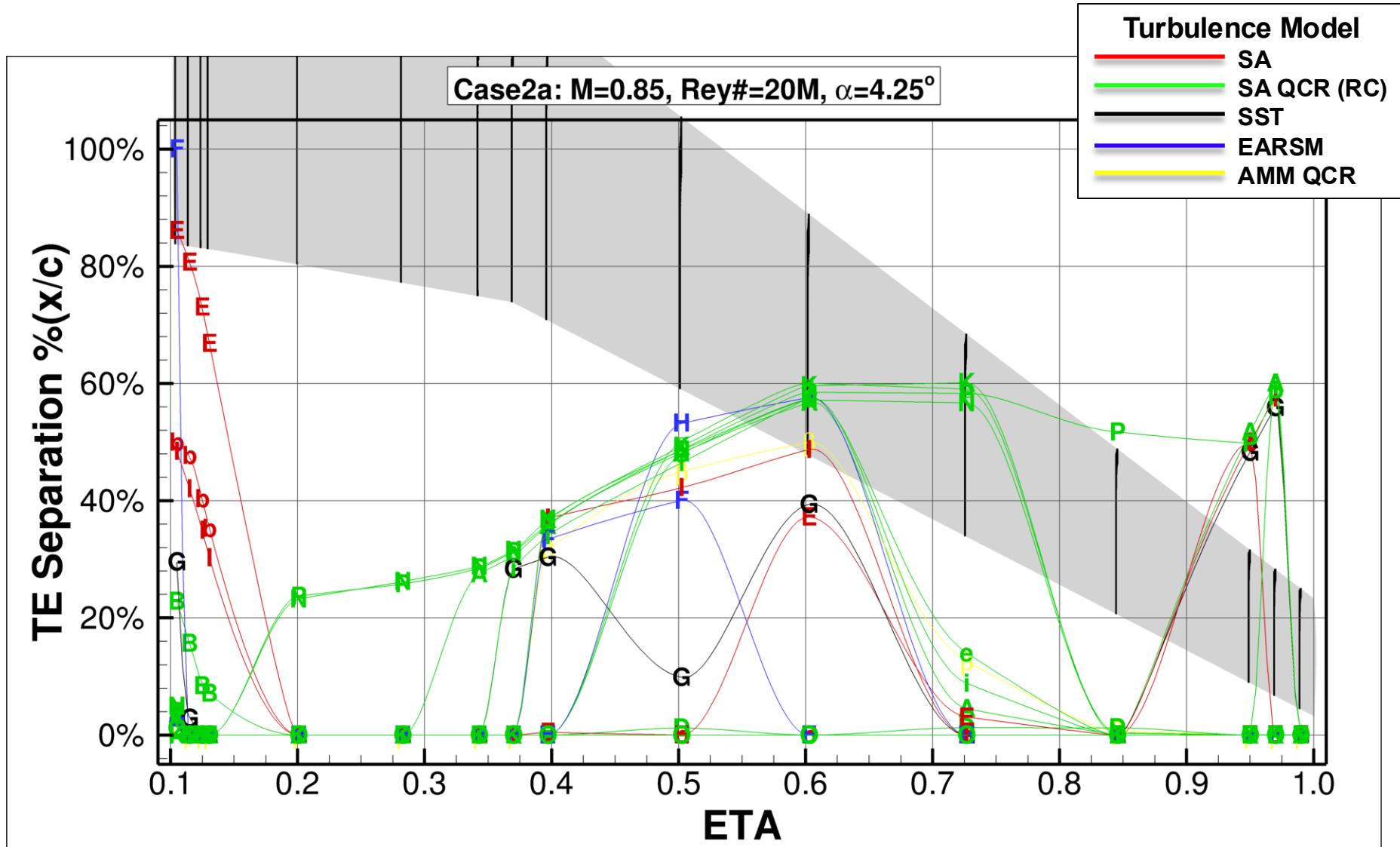




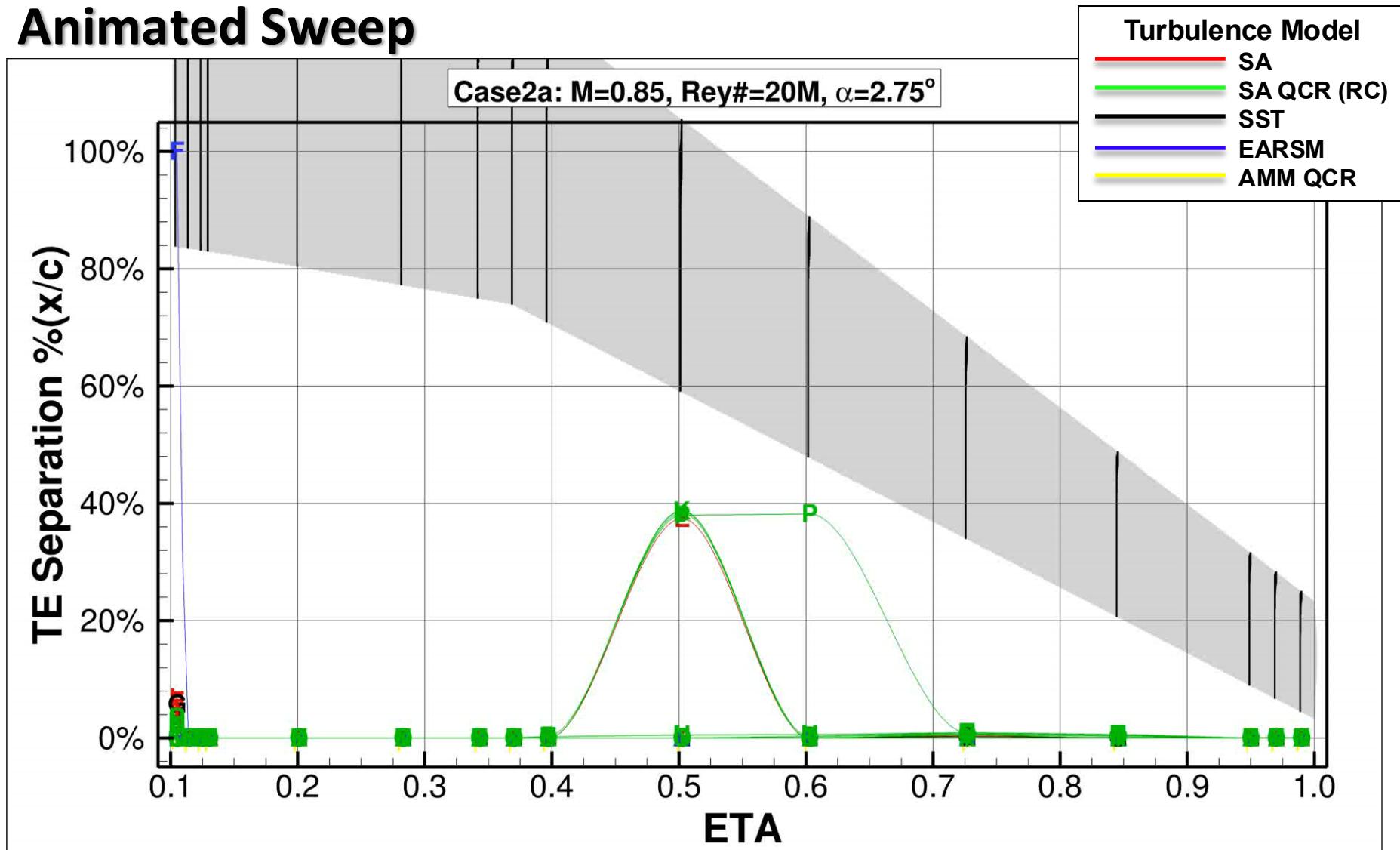








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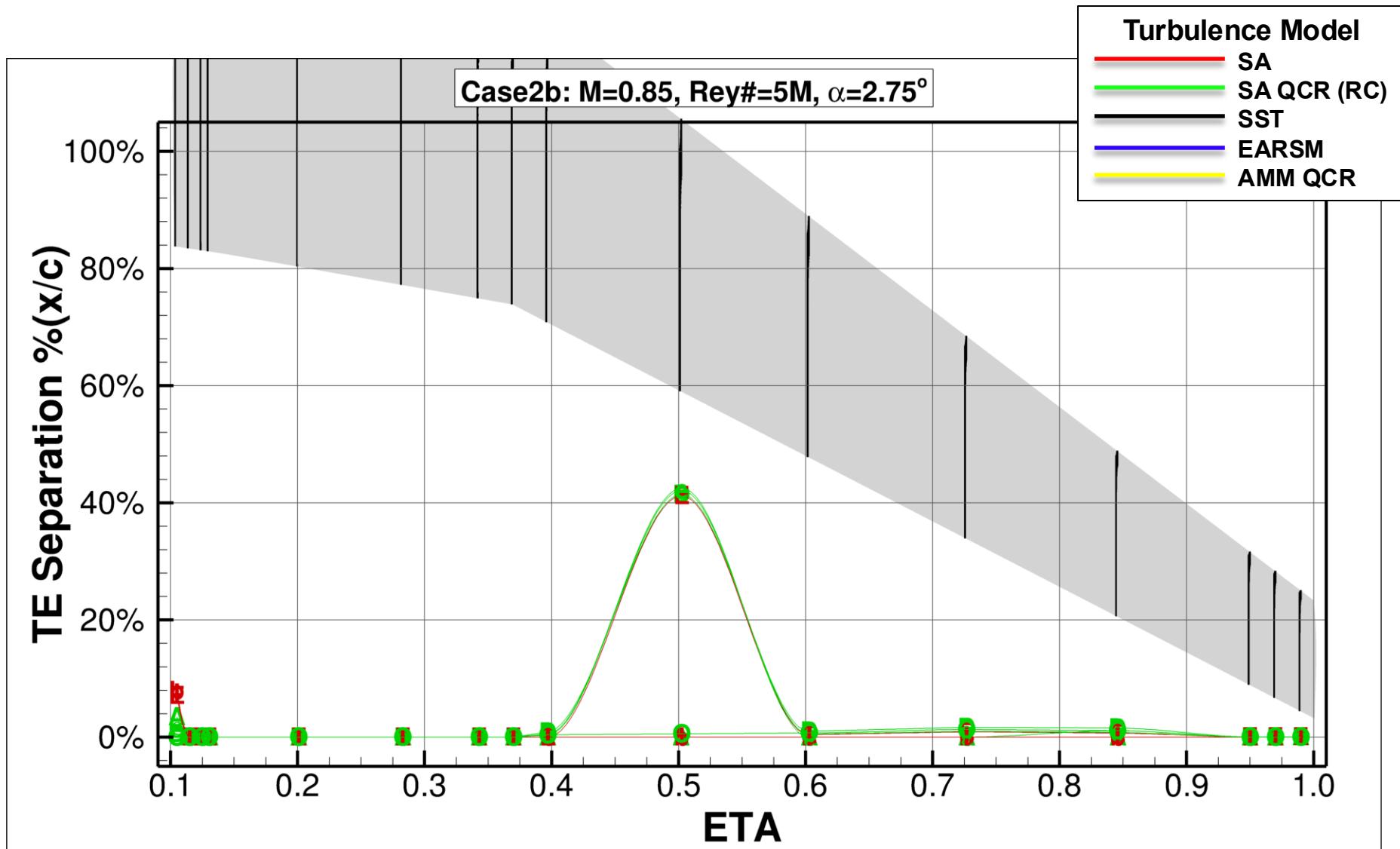
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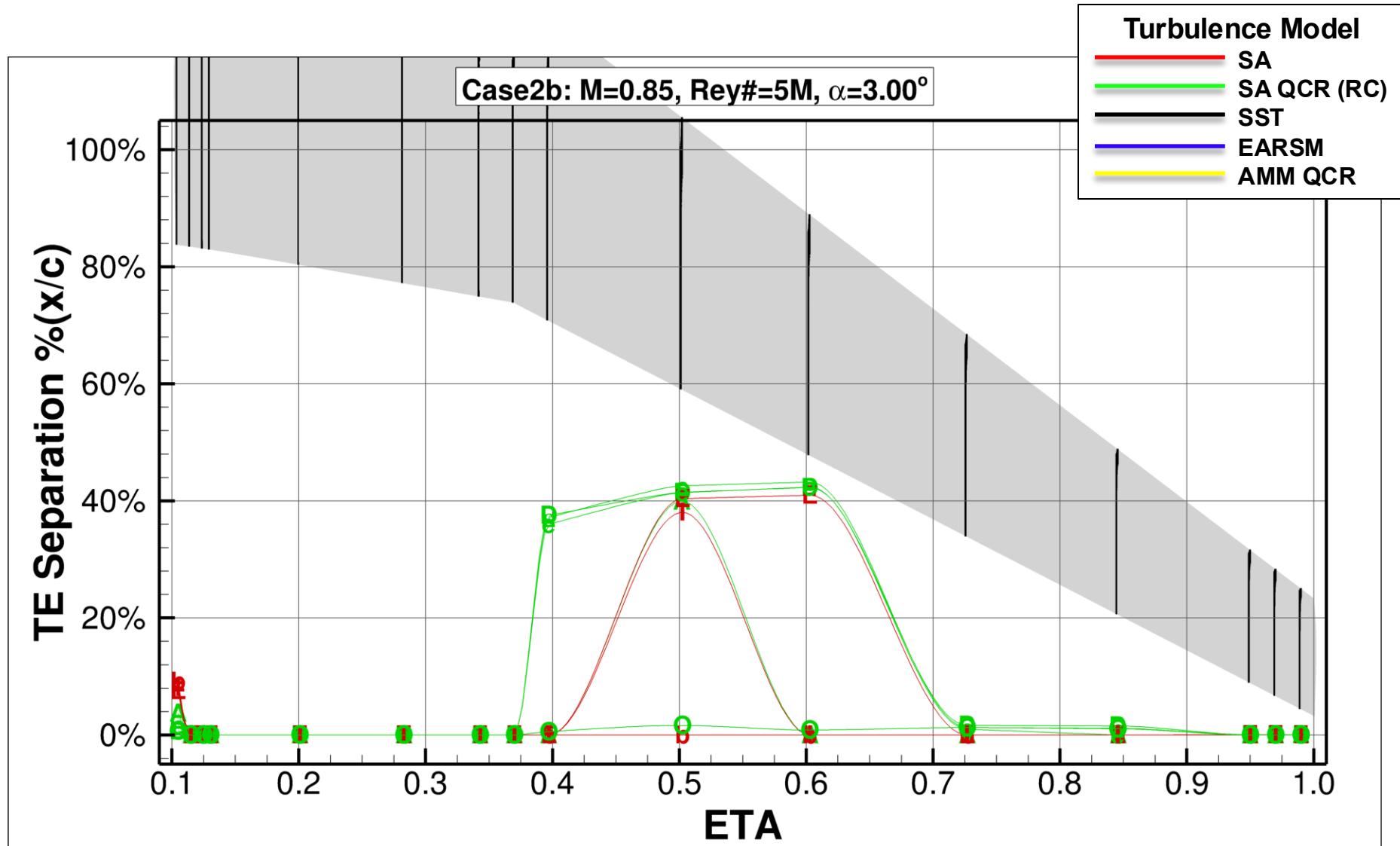


Case 2b

CRM Wing-Body Alpha Sweep

$M = 0.85$ | $\alpha = 2.75\text{--}4.25^\circ$ | $\text{Rey\#} = 5M$



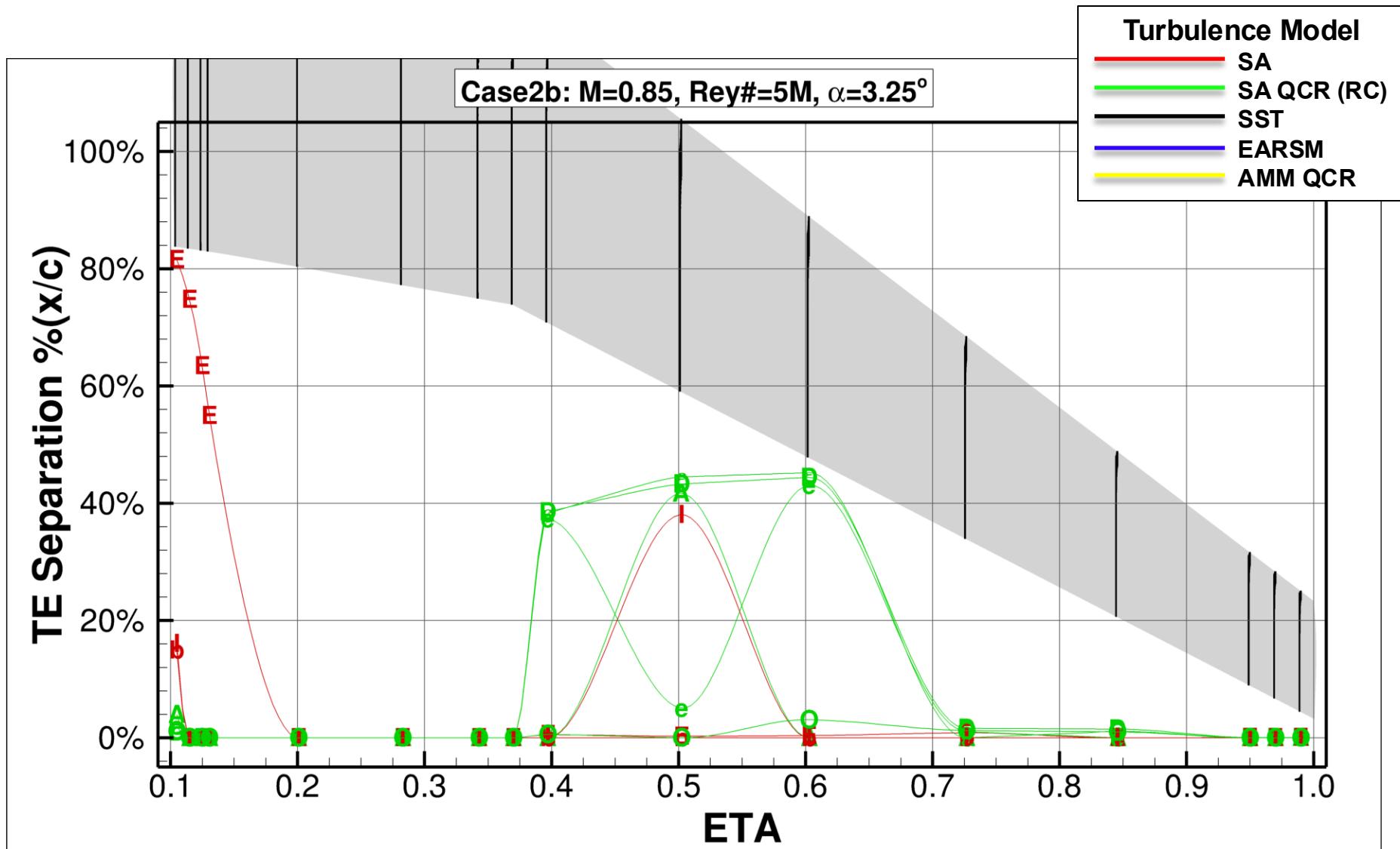


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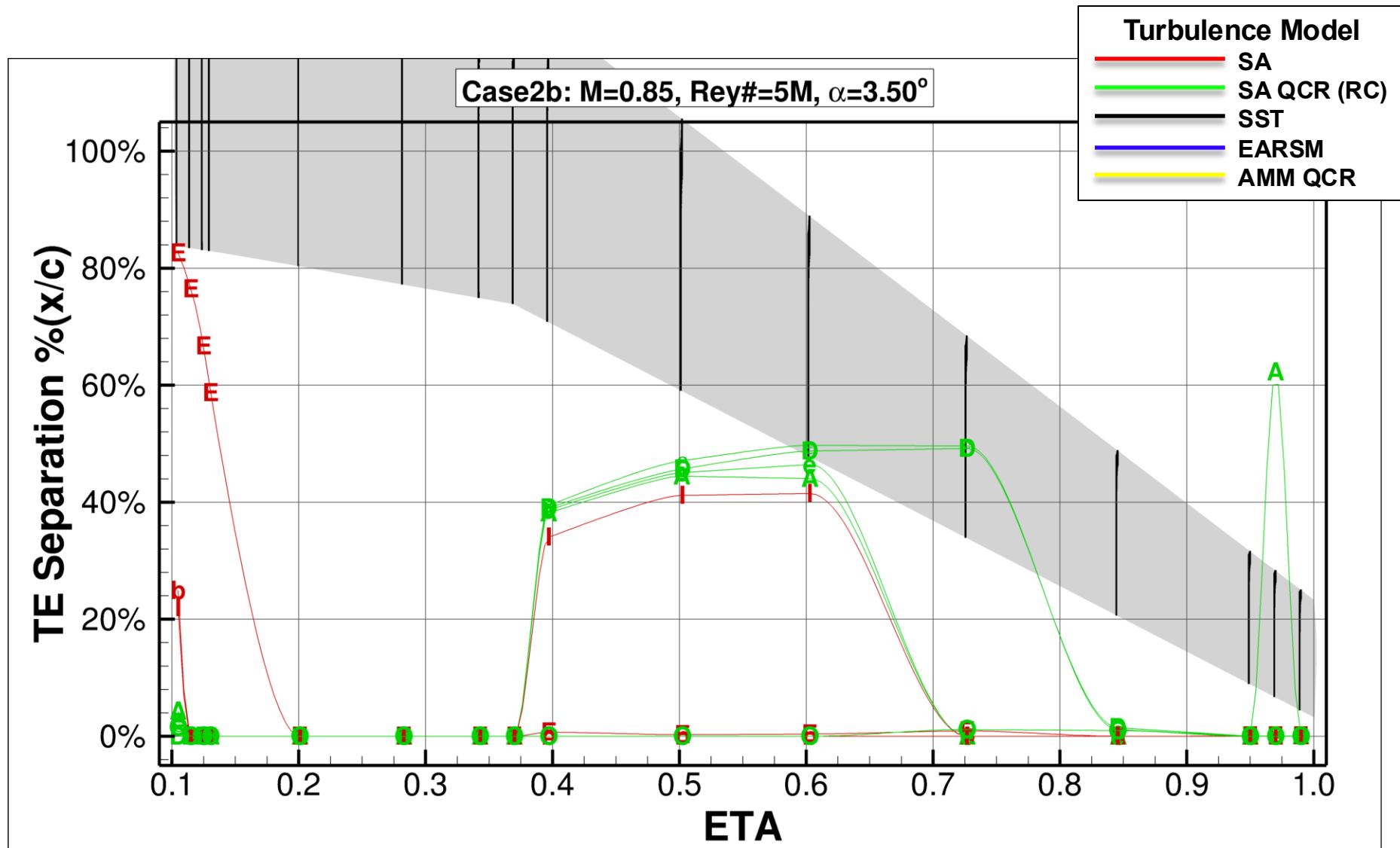


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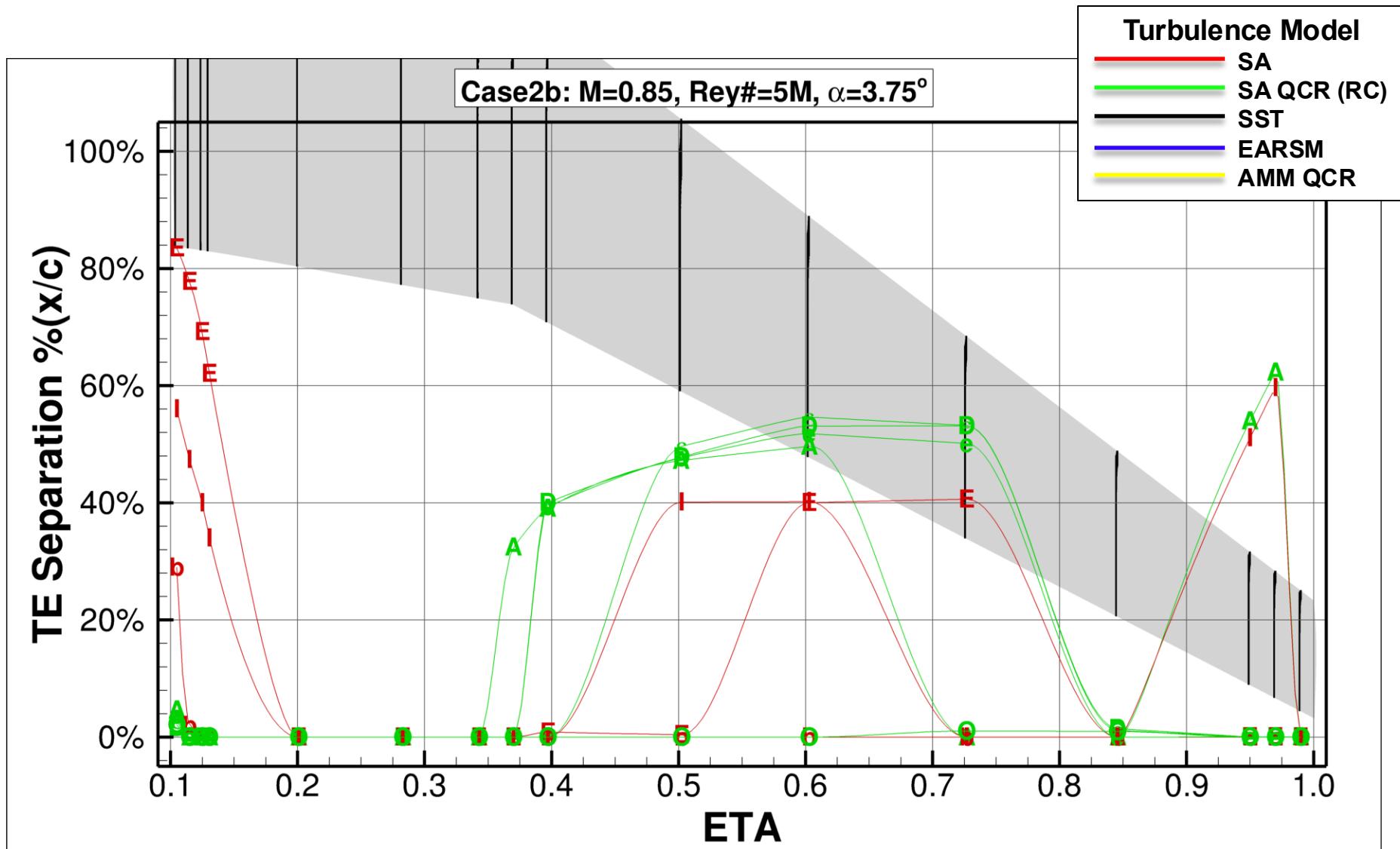


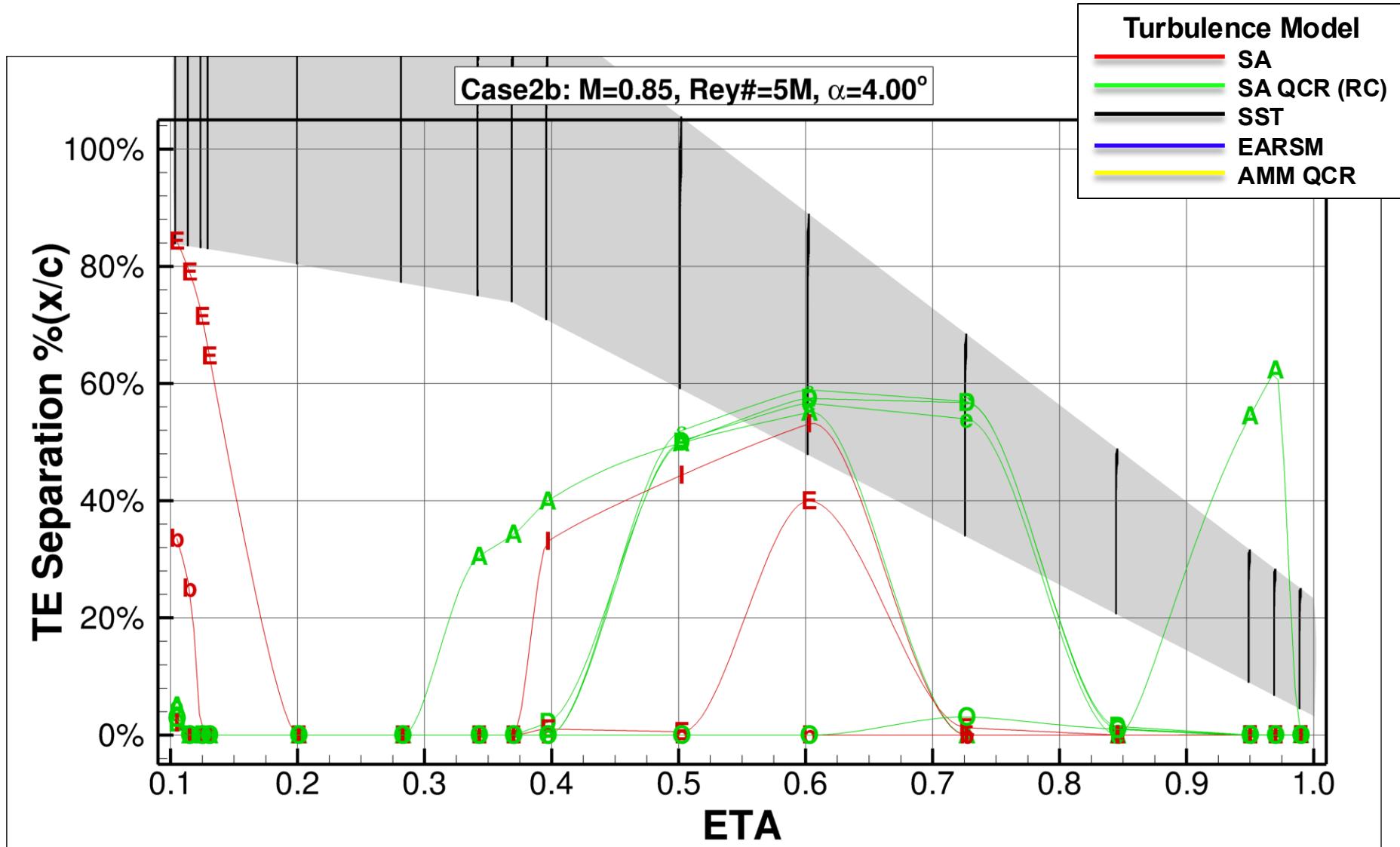
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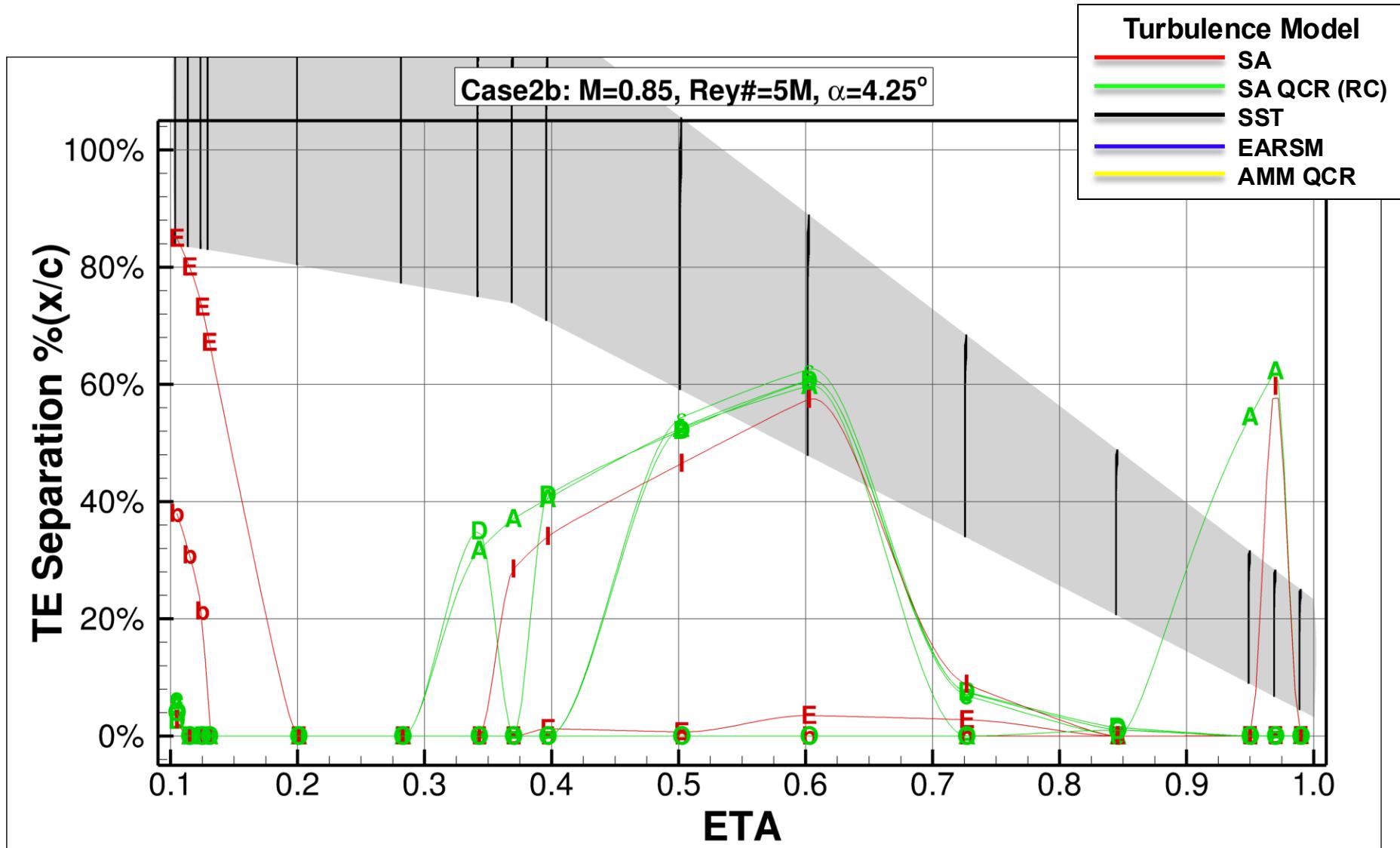
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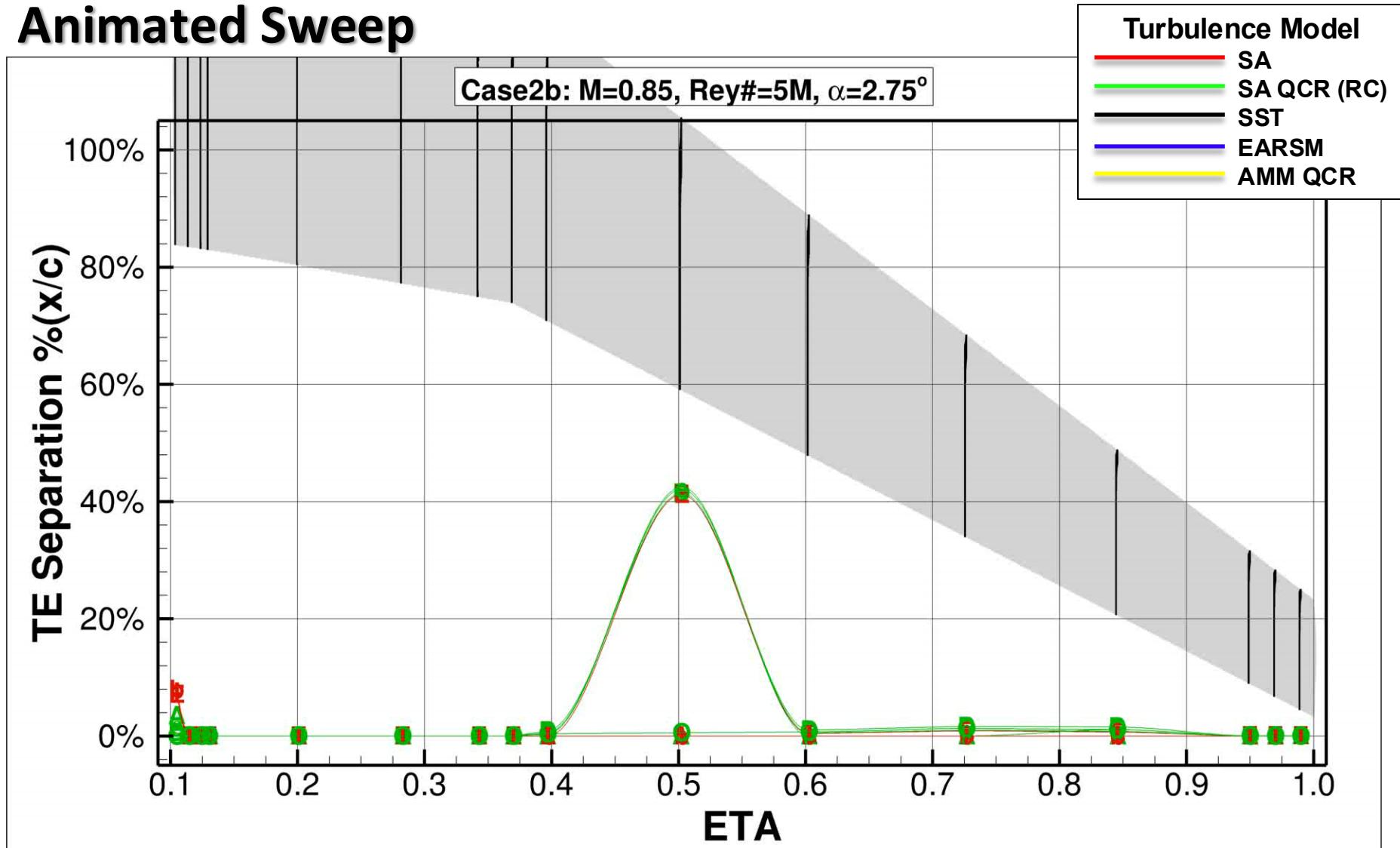
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Animated Sweep





Conclusions:

- Extracting “TE” Separation is difficult to identify generally
 - Requires judgement about how to handle sections with a separation bubble at the shock extends far enough to be called TE separation and
- Not enough data set for turbulence models other than SA-QCR to make judgments
- Case 1
 - Reduced spread (within matching turbulence model) as grid density is increased
- Case 2
 - Fairly good agreement between Yehudi break and $\eta=0.70$.
 - Much more disagreement on inboard wing and wing tip