

# DPW-3 DLR F6/FX2B Results

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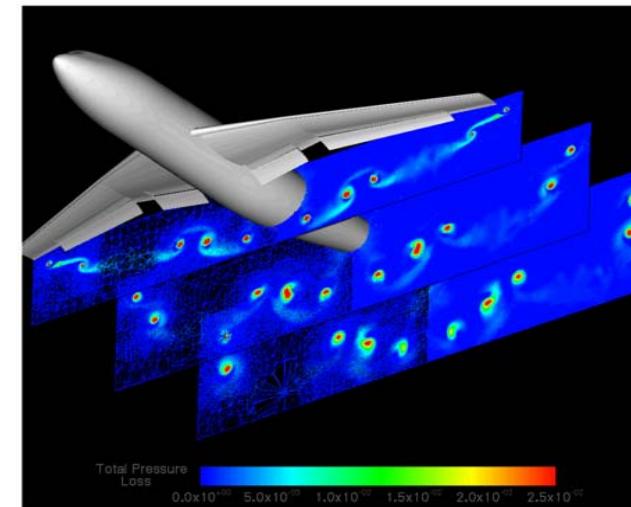
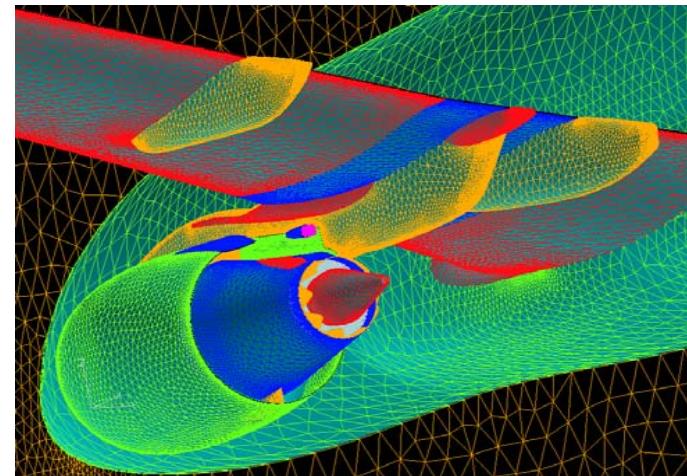


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## Method

- RANS solver DLR TAU
- Unstructured database, node based
- State-of-the-Art algorithms
- 1- and 2-eq. turbulence models
- Fluid-Structure coupling
- Overlapping grids
- Grid adaptation
- Hypersonic extensions
- C code and Python scripting
- High performance on parallel machines
- Applied in European aircraft industry and research



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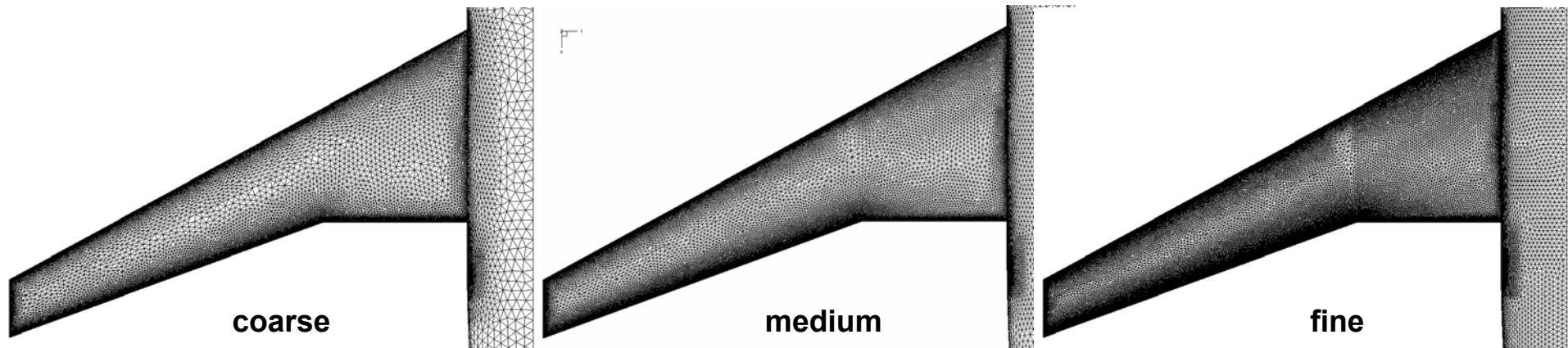




## Grids

- Unstructured hybrid grids generated with Centaur from Centaursoft
- Prismatic elements for BL
- 3 grid densities
- Specification of sources for wing surface
- Constant refinement factor

	Coarse	Medium	Fine
Nodes	2.46 (f6wb) 2.87 (fx2b)	5.10 6.11	8.53 10.30
Boundary nodes	45065 60423	85769 112672	121588 167434
Prismatic layers	21	31	40



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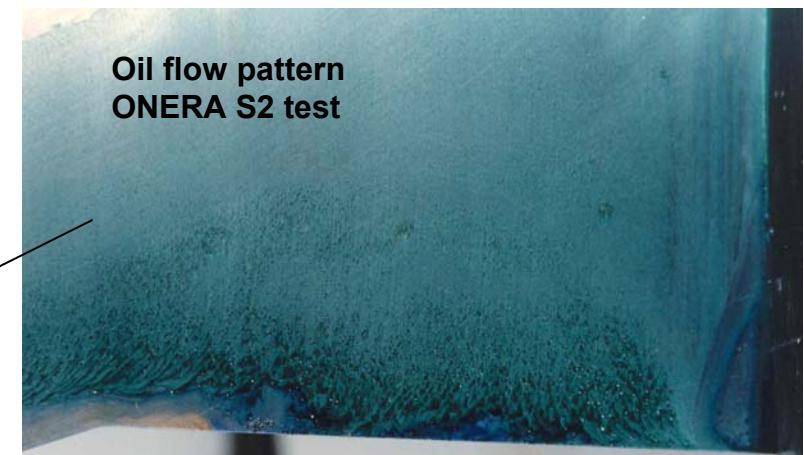
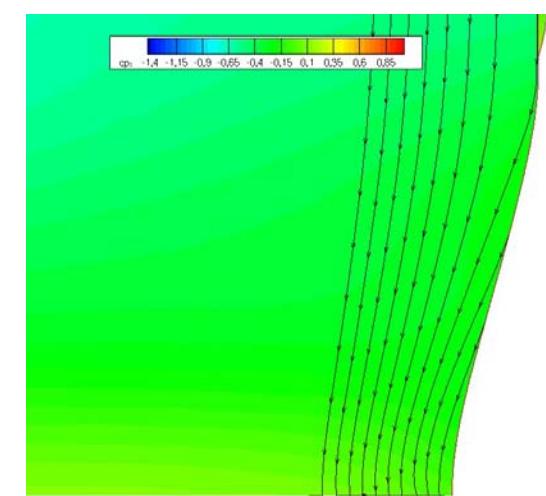
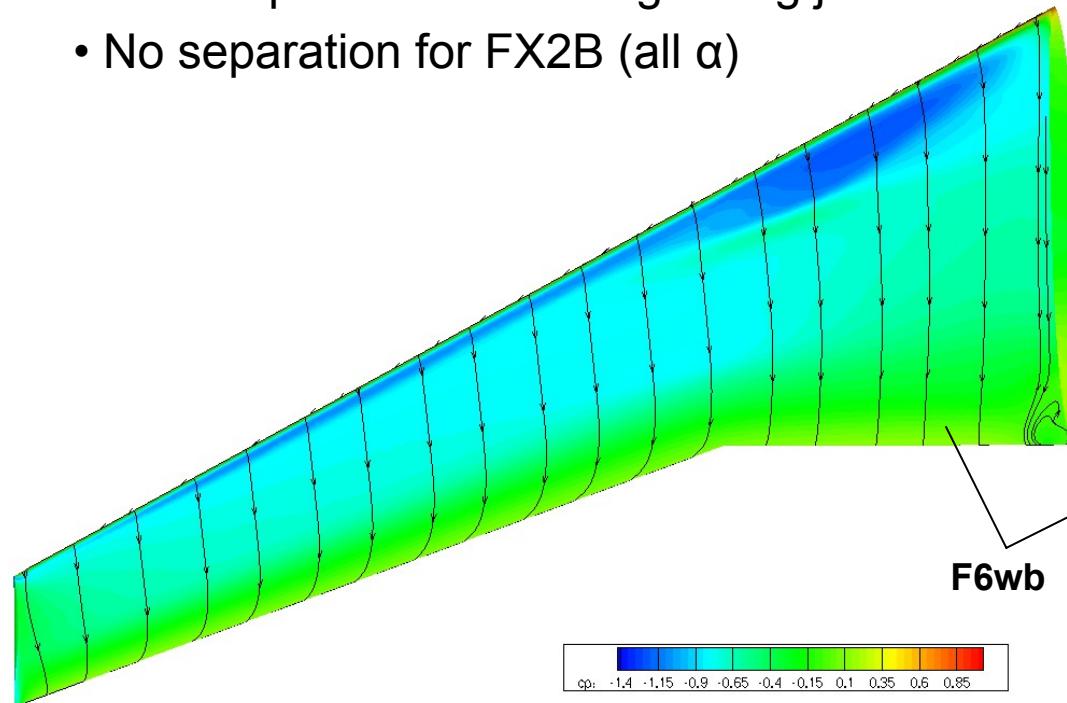


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## Results

- TAU sae, medium prismatic grids, CL=0.5
- DPW-2:
  - separations Wing/fuselage, pylon/wing found
  - no trailing edge flow separation
- Flow separation at fuselage wing junction
- No separation for FX2B (all  $\alpha$ )



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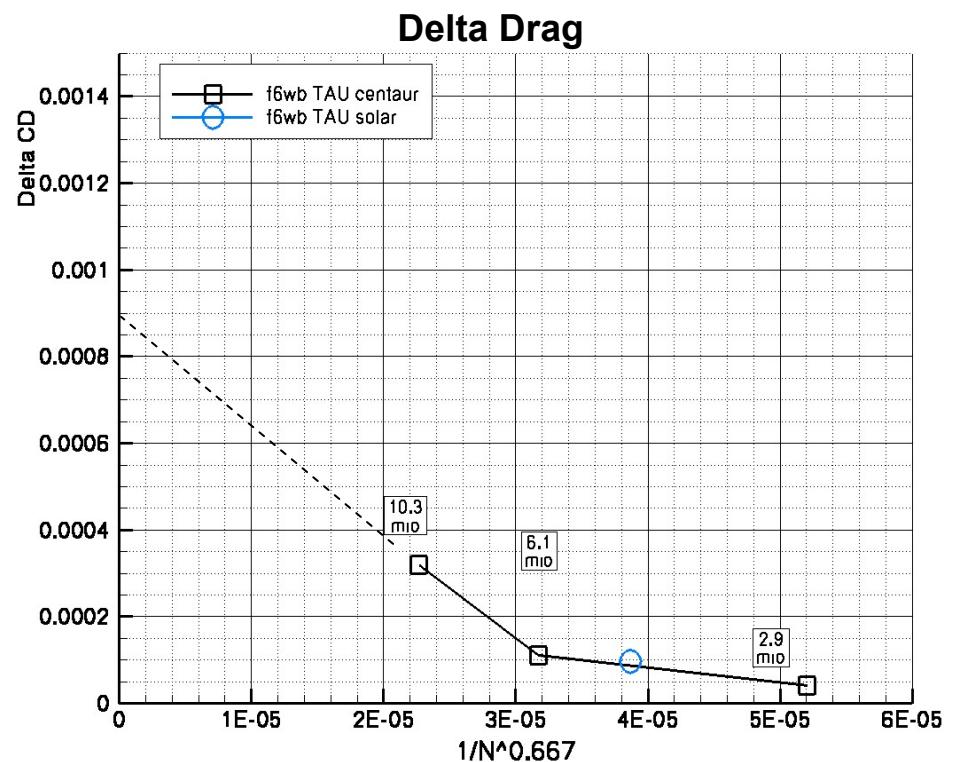
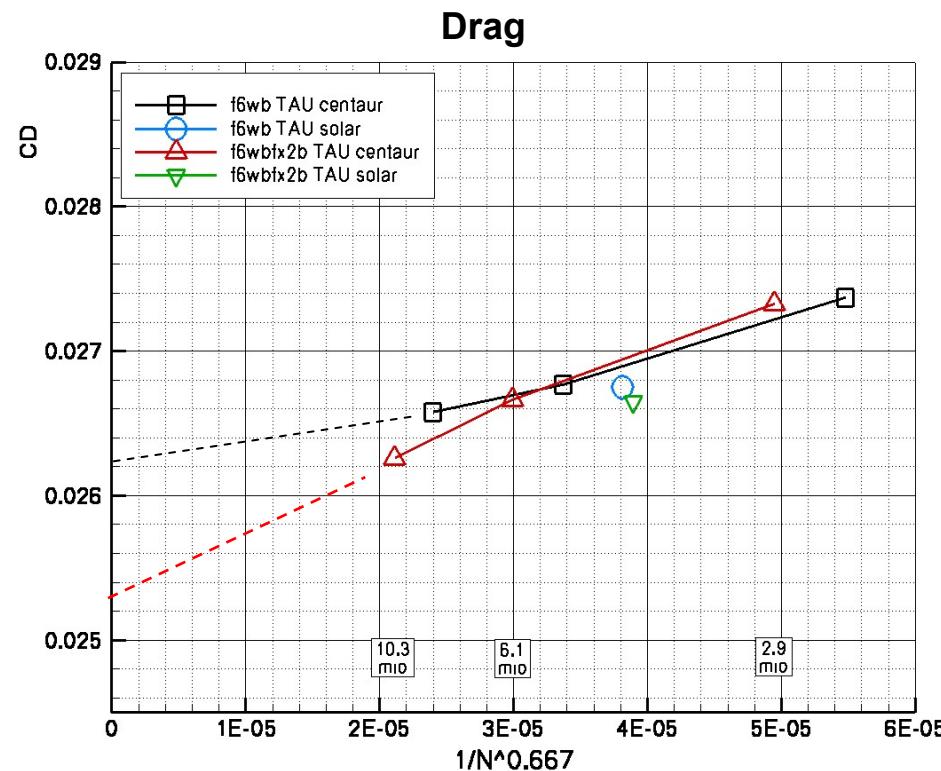


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## Results: Grid Influence

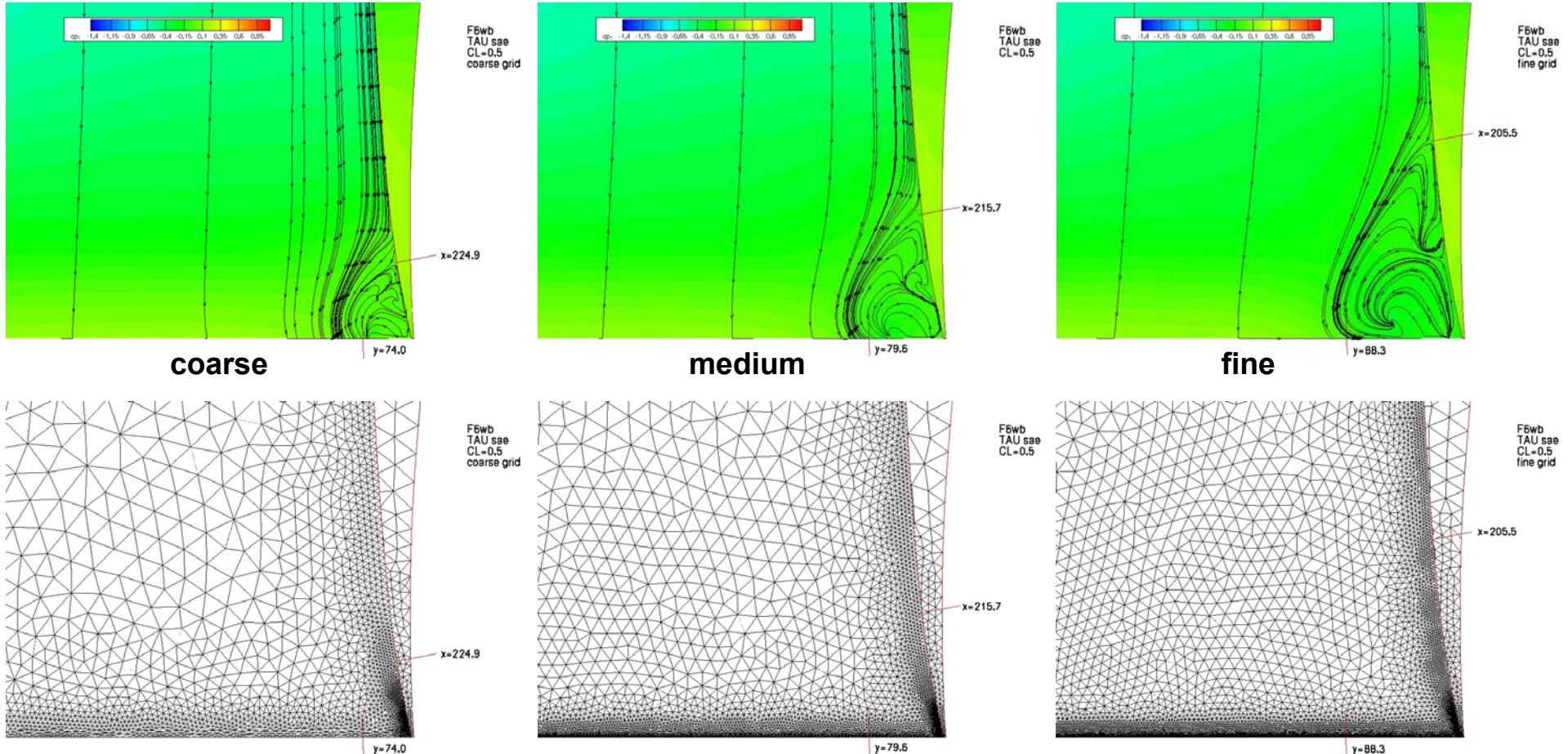
### Convergence Study





## Results: Grid Influence

Influence on Flow Separation F6wb for CL=0.5



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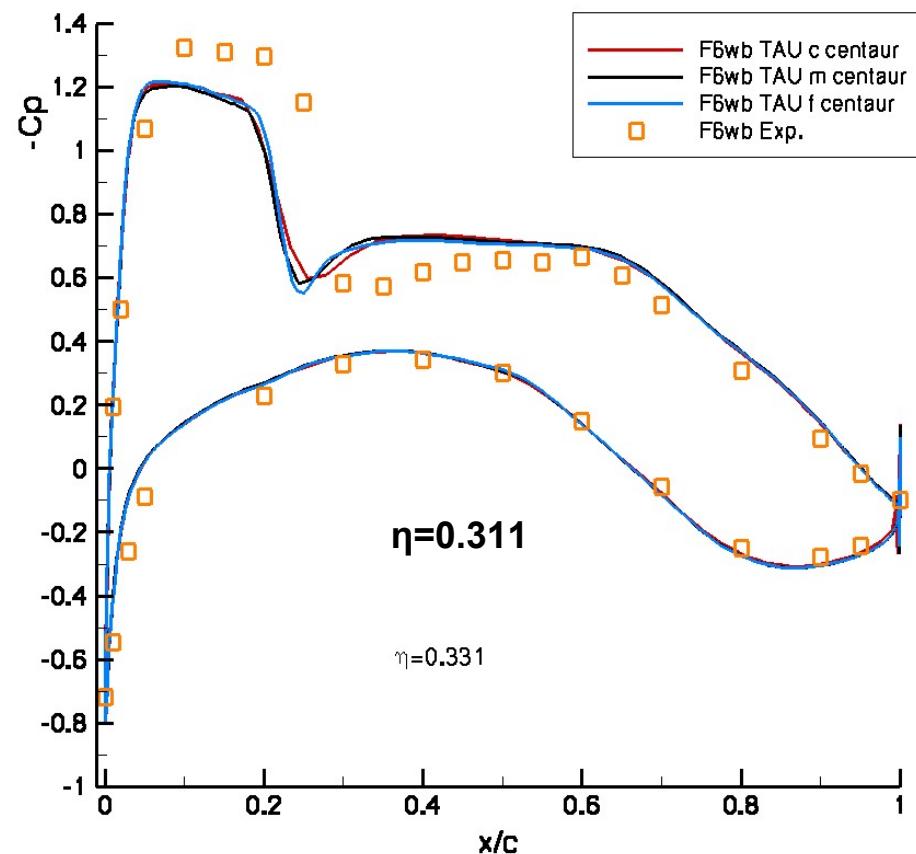
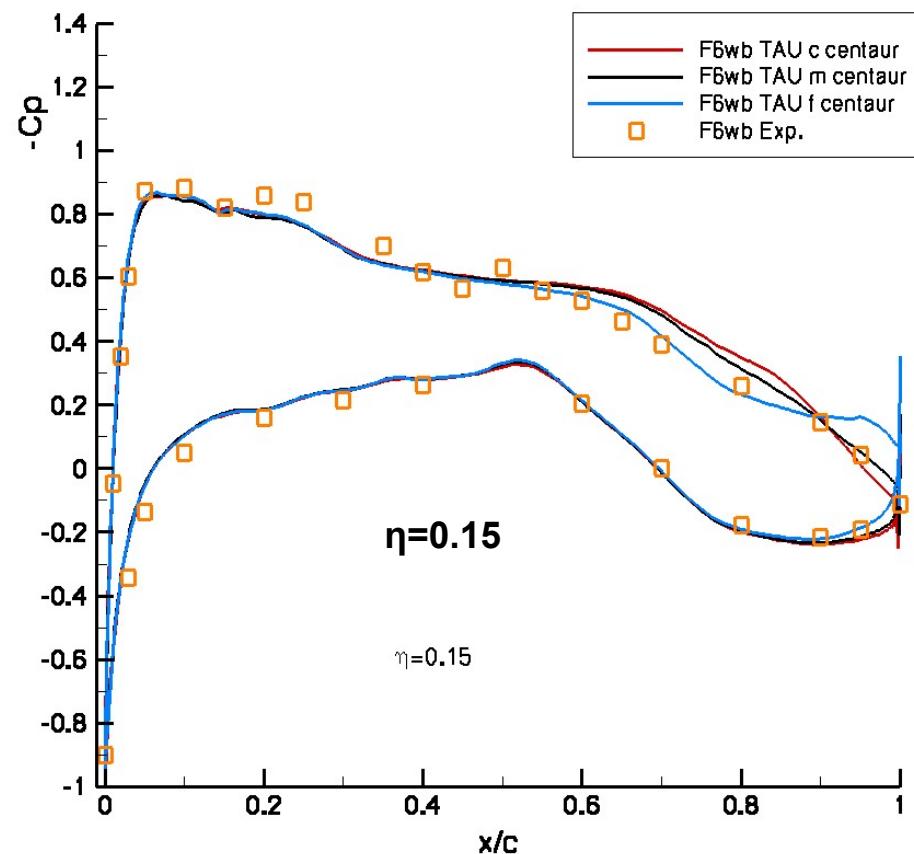


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## Results: Grid Influence

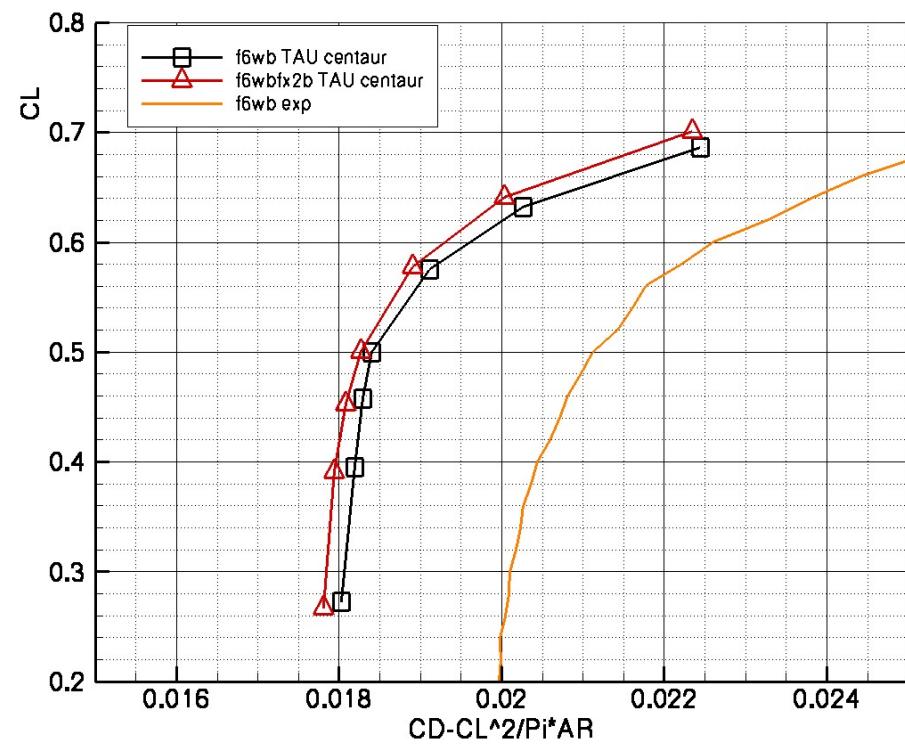
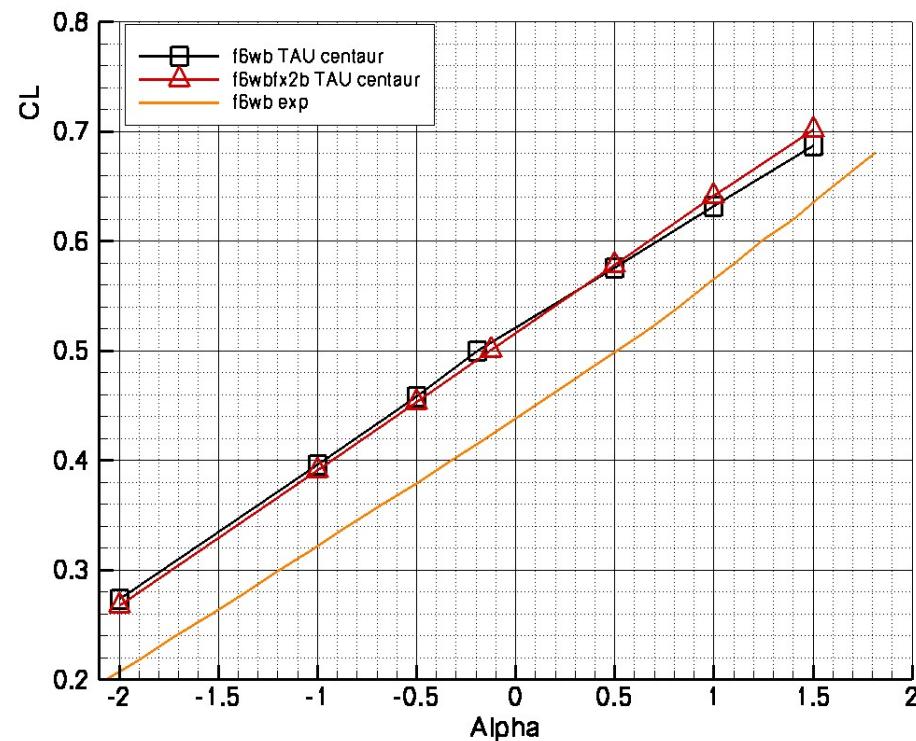
Influence on  $C_p$  for  $CL=0.5$





## Results: F6 / F6FX2B

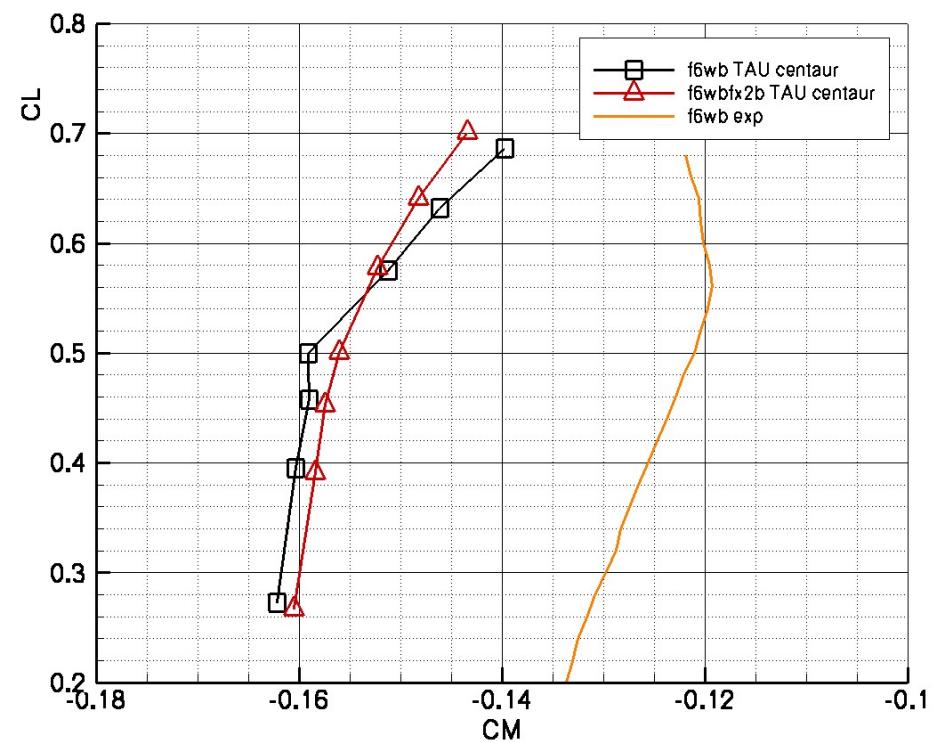
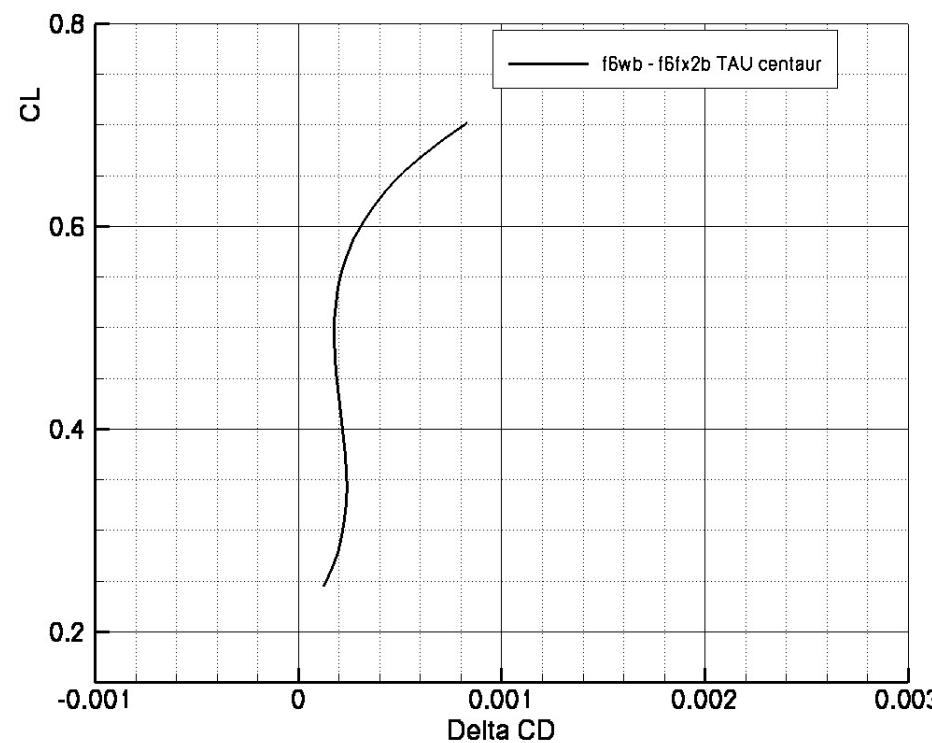
Medium Grid: lift and drag





## Results: F6 / F6FX2B

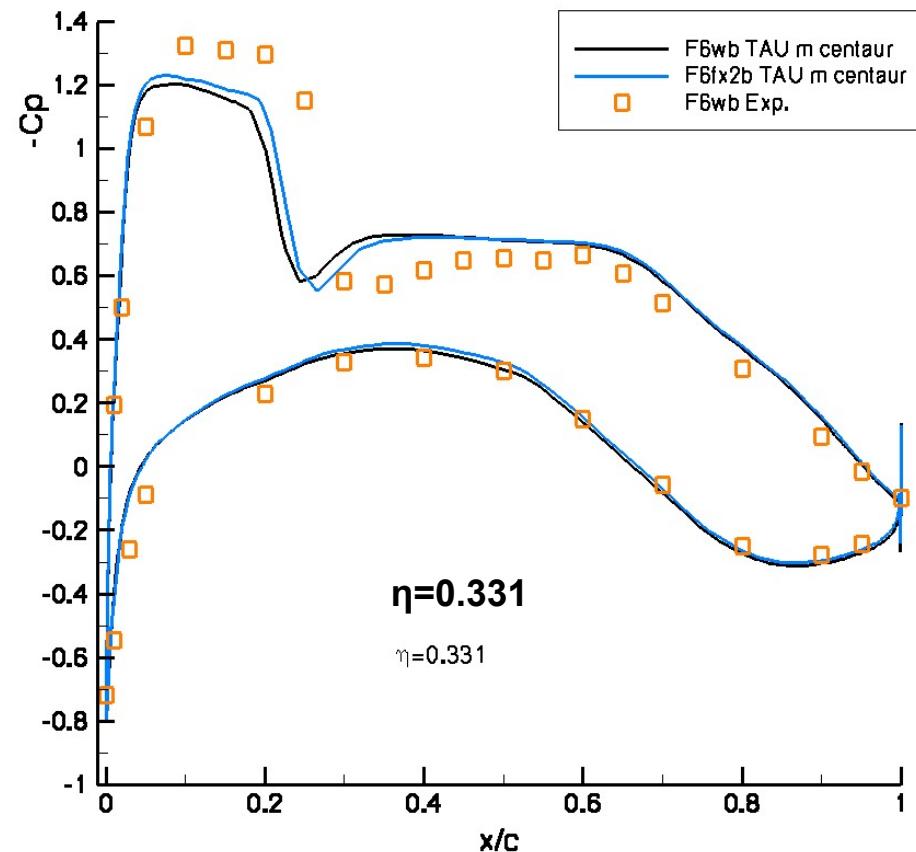
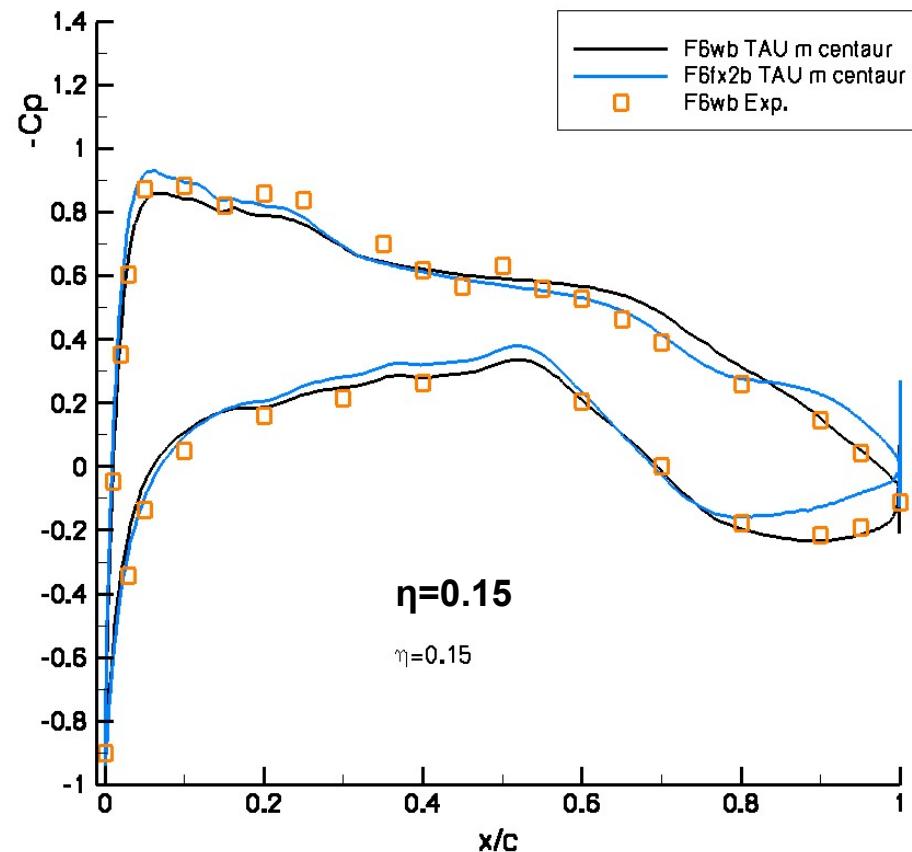
Medium Grid: delta drag , moments





## Results: F6 / F6FX2B

Medium Grid: influence on  $C_p$  for  $CL=0.5$

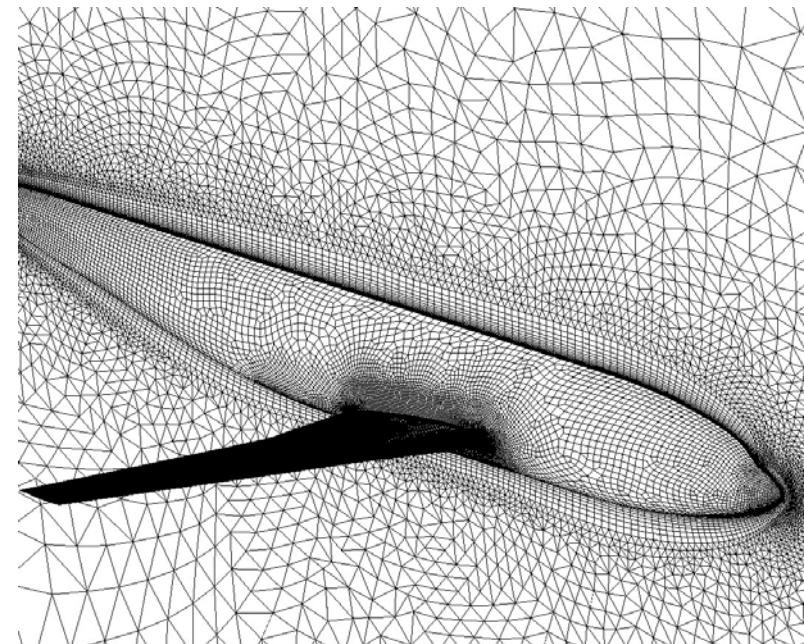
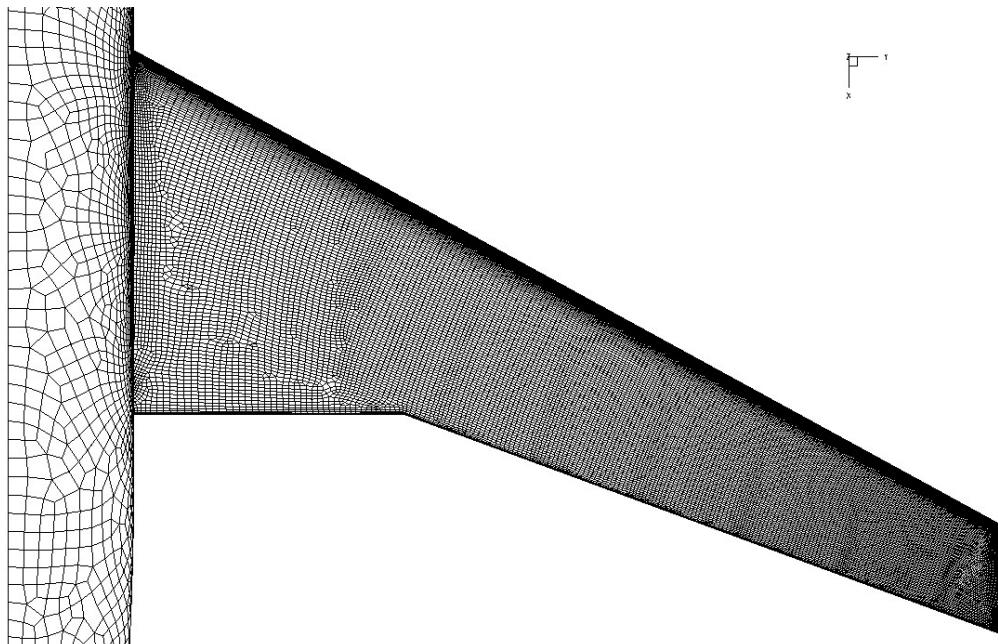




## Grids

- Well-known: grid quality and density is important
- Different approach: hexahedrals and tetrahedrals
- Hexahedral elements for BL resolution
- Solar grid generator from Airbus, ARA, BAES
- Grids generated by QinetiQ

	Medium
Nodes	4.24 4.11



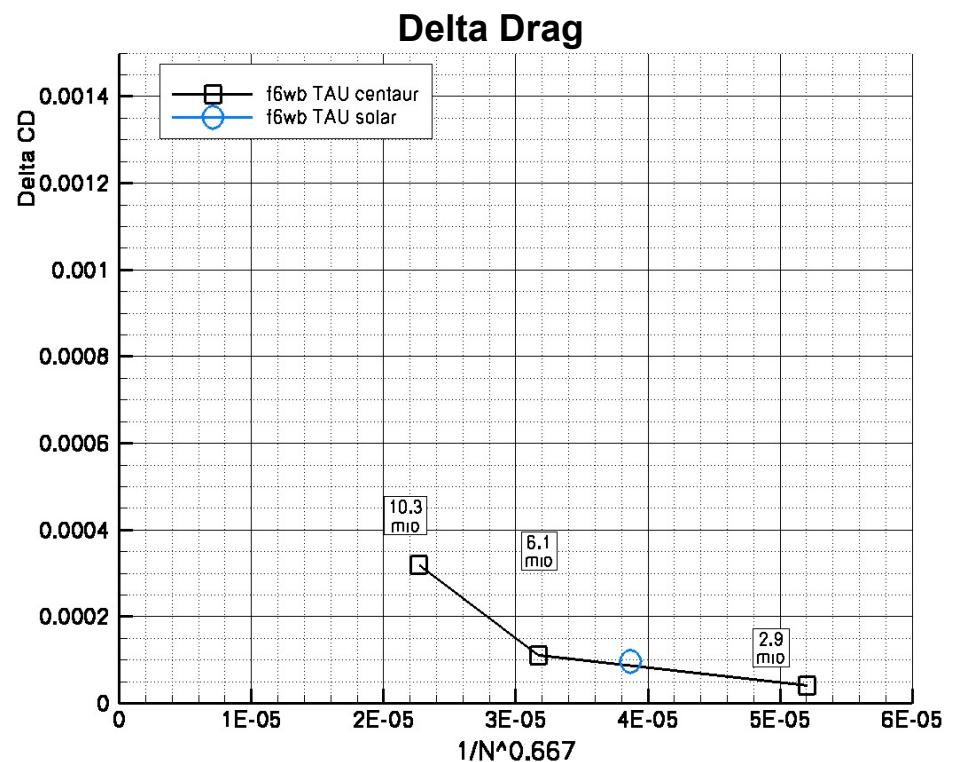
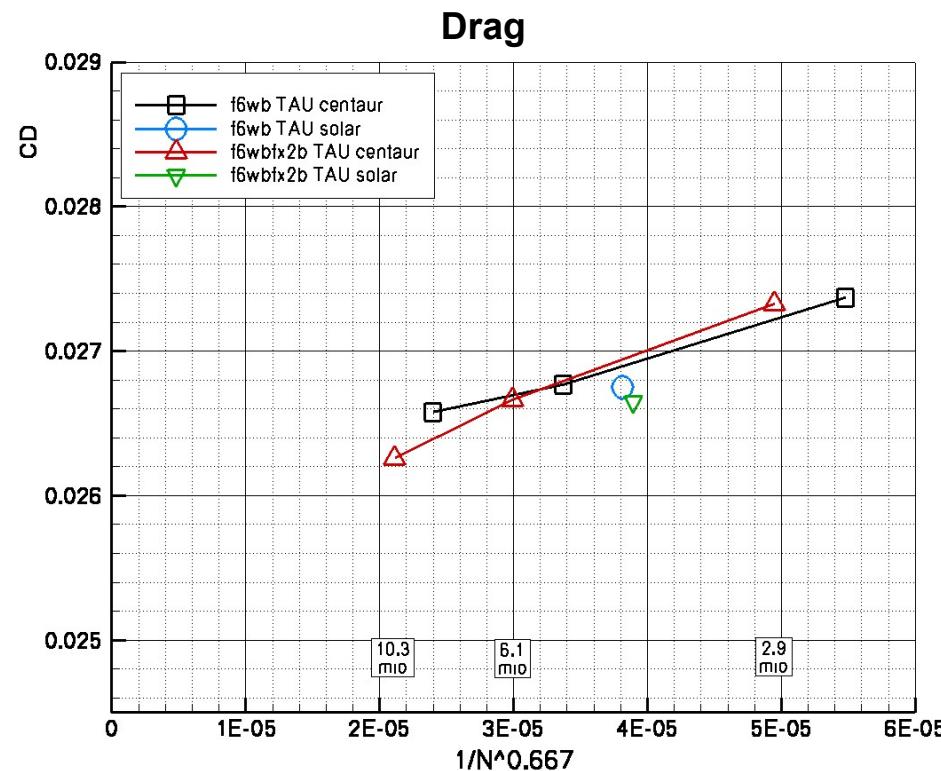
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## Results: Grid Influence

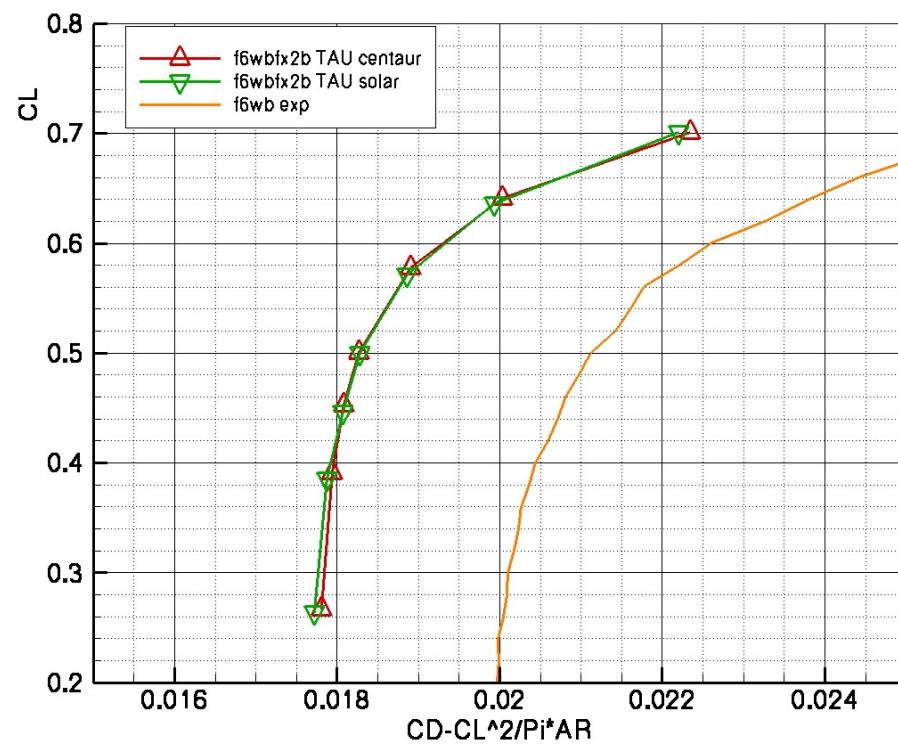
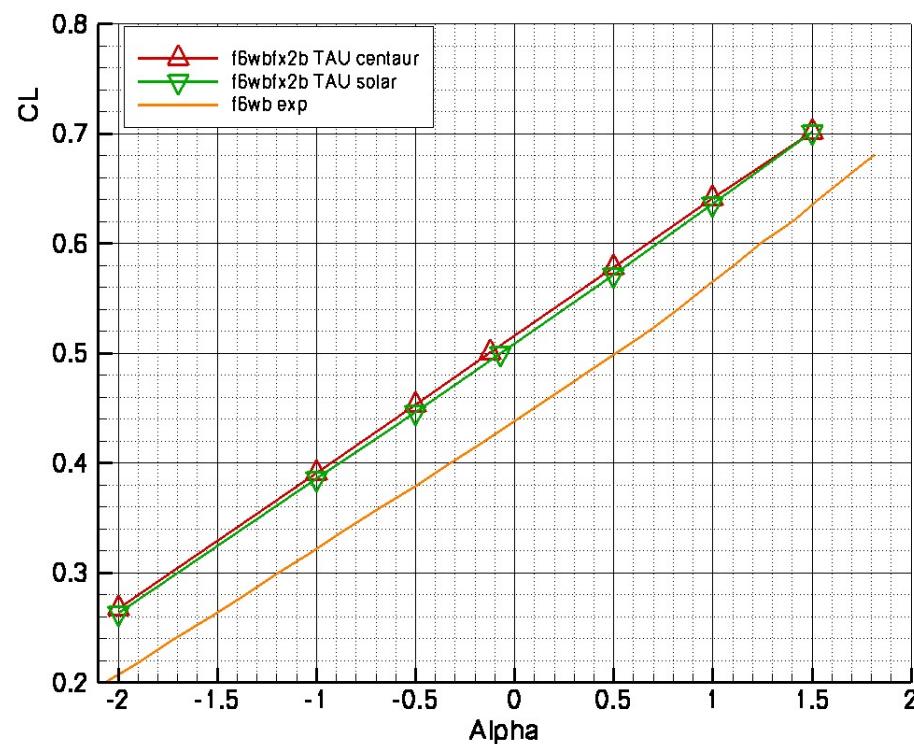
### Convergence Study





## Results: Grid Type

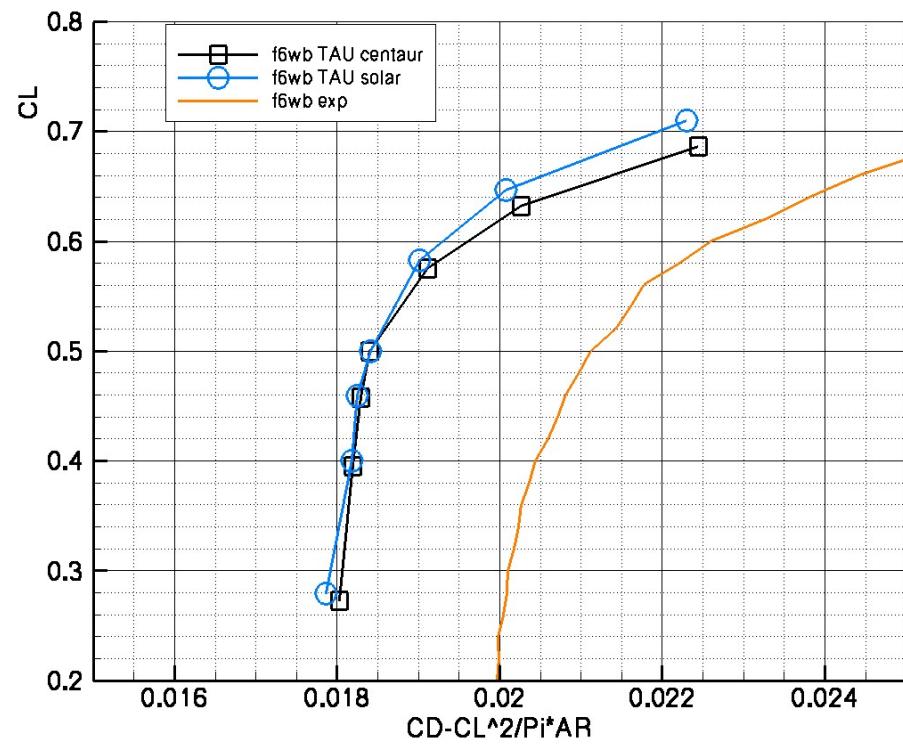
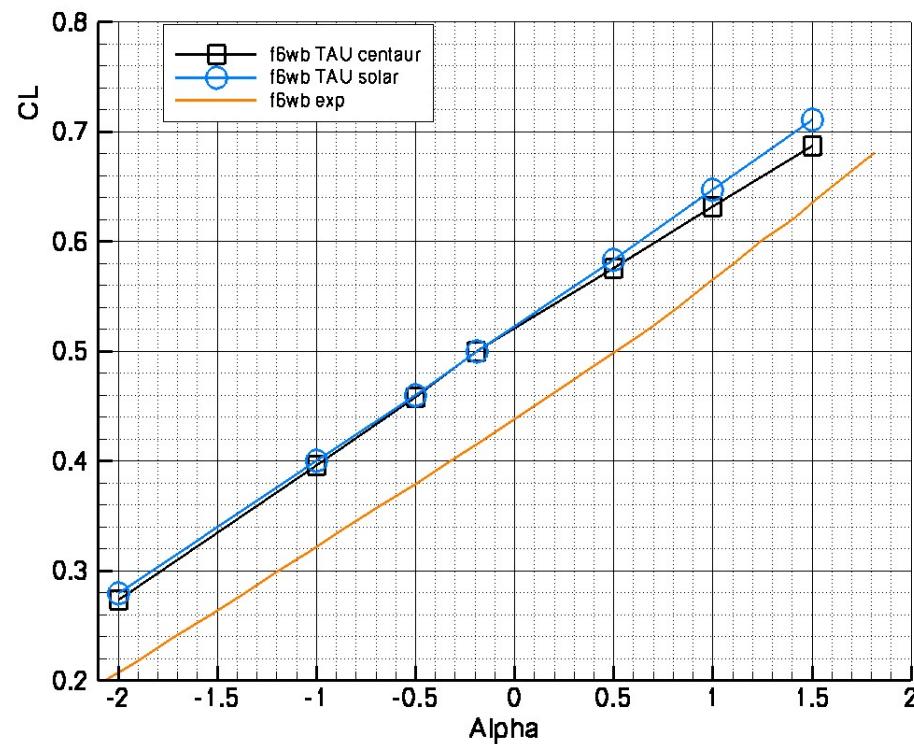
F6fx2b: Prismatic / Hexahedral hybrid grids





## Results: Grid Type

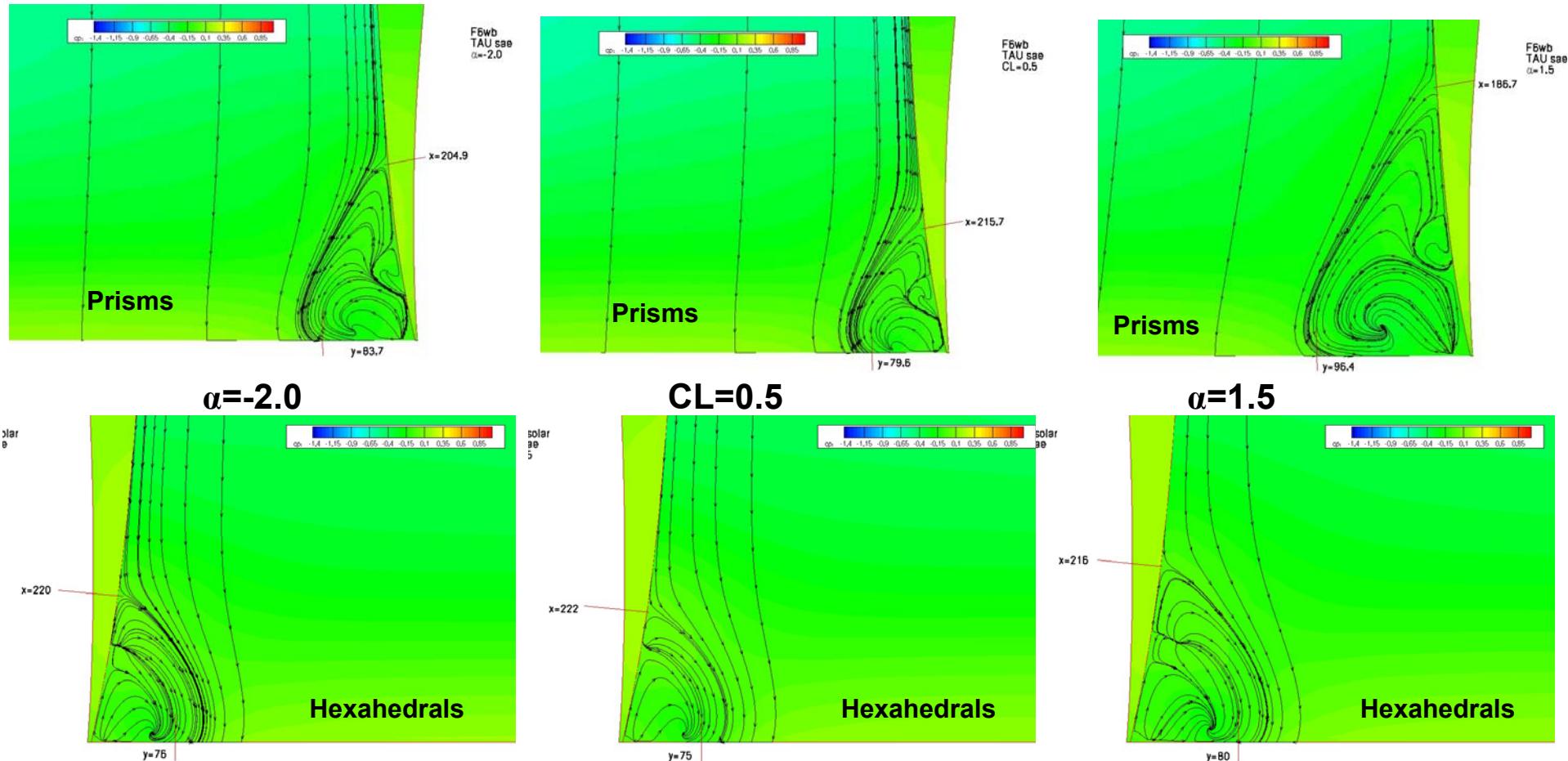
F6wb: Prismatic / Hexahedral hybrid grids





## Results: Grid Type

Influence of alpha on flow separation



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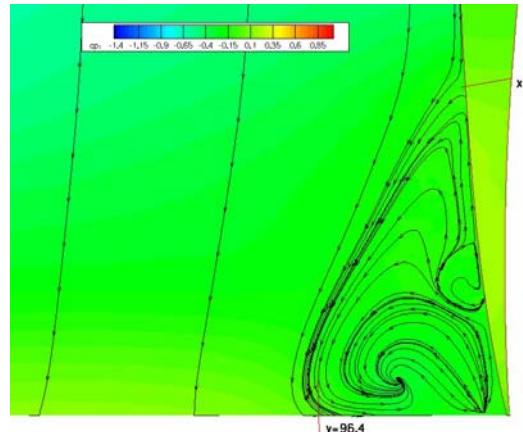


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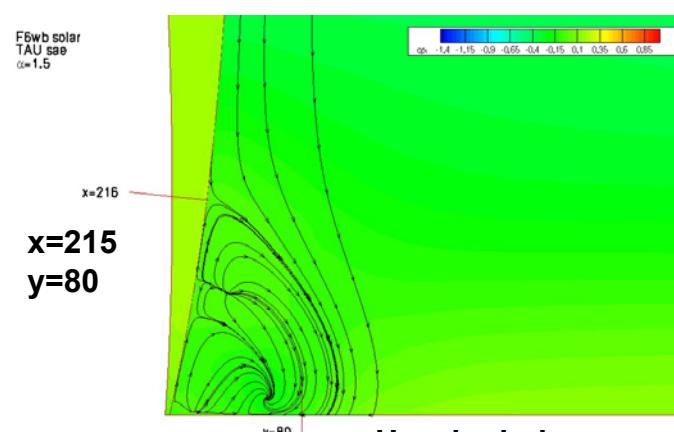


## Results: Grid Type

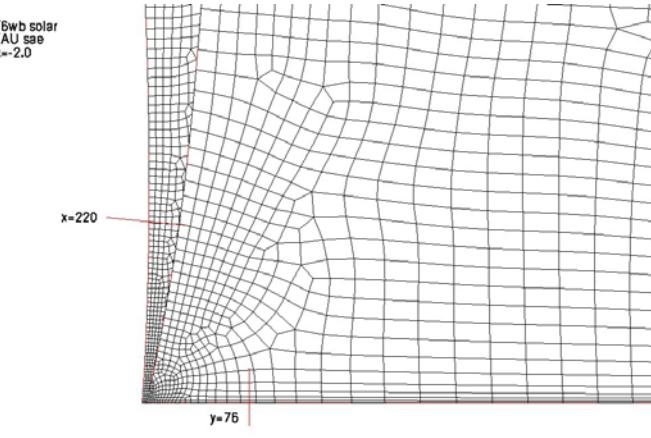
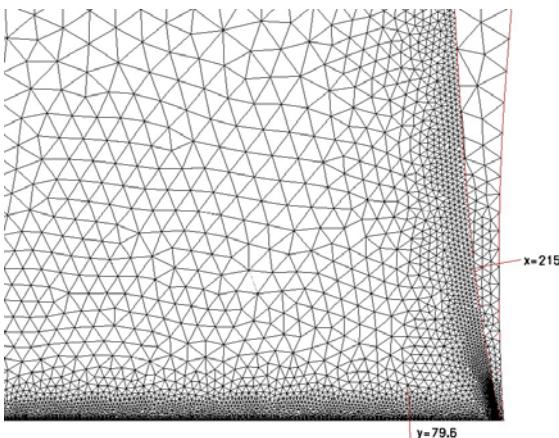
Influence on flow separation F6wb for  $\alpha=1.5$



Triangles



Hexahedrals



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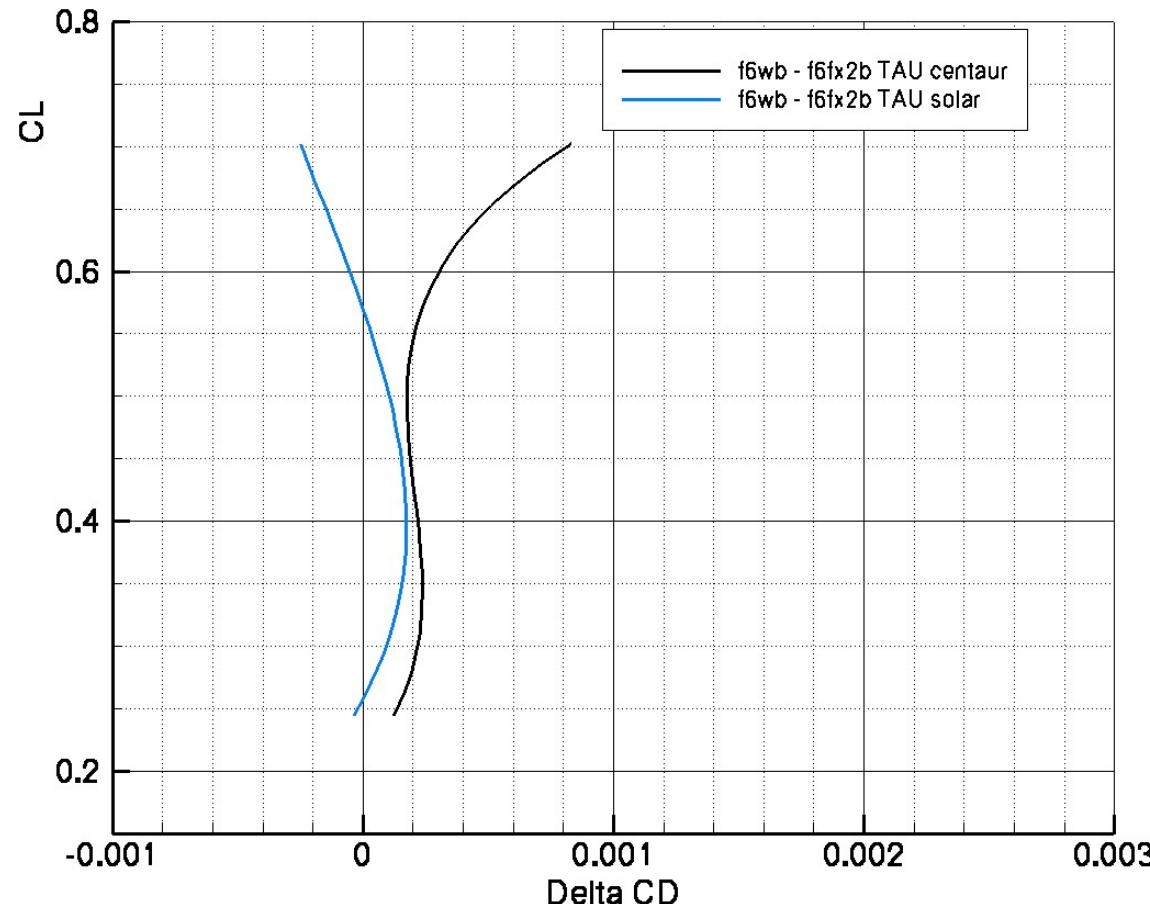


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## Results: Grid Type

F6wb / F6fx2b Delta Drag





## Summary

- F6fx2b shows improved junction flow
- Delta drag at design point approx.:
  - $\approx 1.5$  dc for medium grid
  - $\approx 9$  dc for extrapolation
- Grid density in wing fuselage junction has major effect on size of separation bubble at high alpha (SAE model)
- Delta drag of F6wb-F6fx2b can switch its sign for high / low alpha
- Wing trailing edge separation can not be found with SAE (see also DPW-2)
- Hybrid hexahedral grids are very promising  
(improved convergence, less nodes)



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