α=0.999, C_L=0.999, C_D=0.09999, C_M=-0.999, M=0.73, Re=3e6, MyAirfoil.iges, Upper Surface
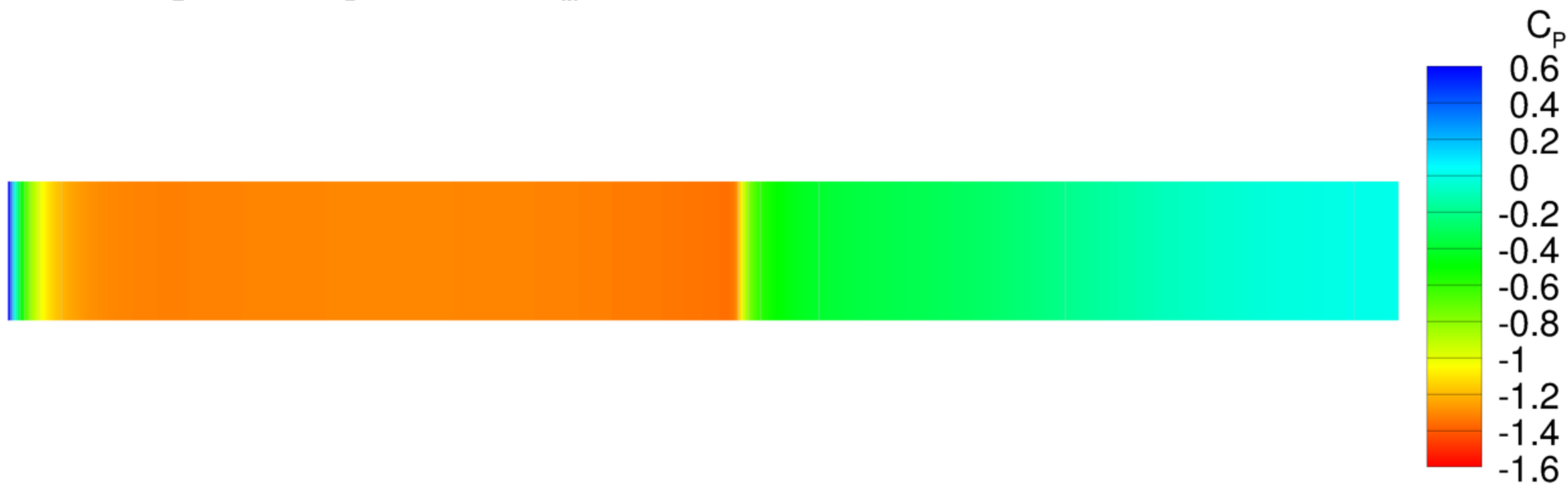
α=0.999, C_L=0.999, C_D=0.09999, C_M=-0.999, M=0.73, Re=3e6, MyAirfoil.iges, Upper Surface

α=0.999, C_L=0.999, C_D=0.09999, C_M=-0.999, M=0.73, Re=3e6, MyAirfoil.iges, Upper Surface

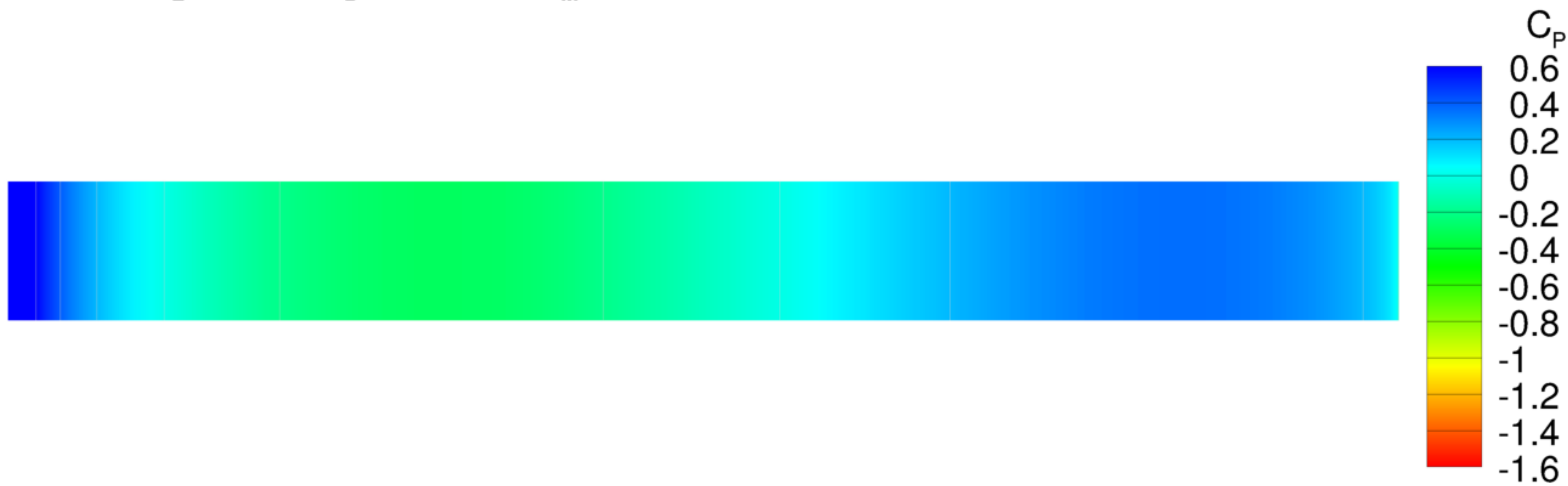α=0.999, C_L=0.999, C_D=0.09999, C_M=-0.999, M=0.73, Re=3e6, MyAirfoil.iges, Lower Surface

α=0.999, C_L=0.999, C_D=0.09999, C_M=-0.999, M=0.73, Re=3e6, MyAirfoil.iges, Lower Surface

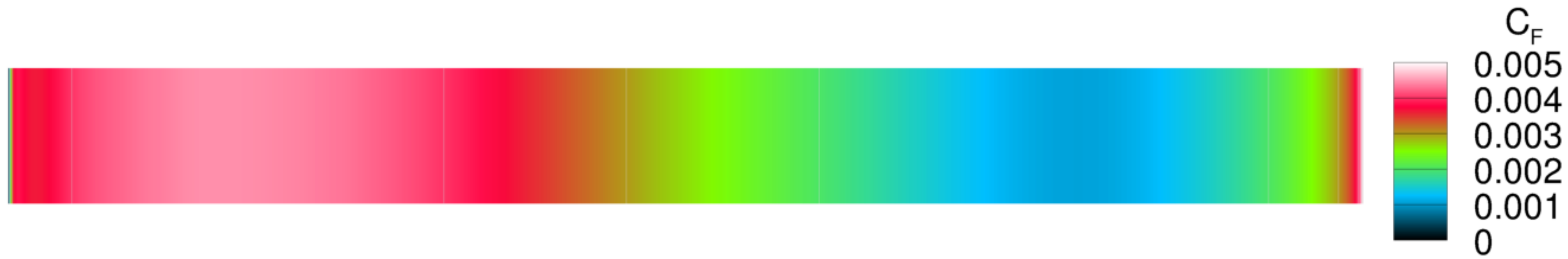α=0.999, C_L=0.999, C_D=0.09999, C_M=-0.999, M=0.73, Re=3e6, MyAirfoil.iges, Lower Surface

C_{Fx}

0.005
0.004
0.003
0.002
0.001
0
-0.001
-0.002