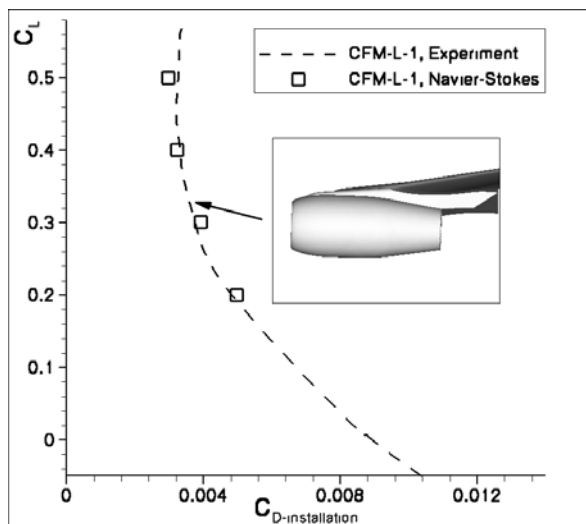


CFD Drag Prediction Workshop

Objectives

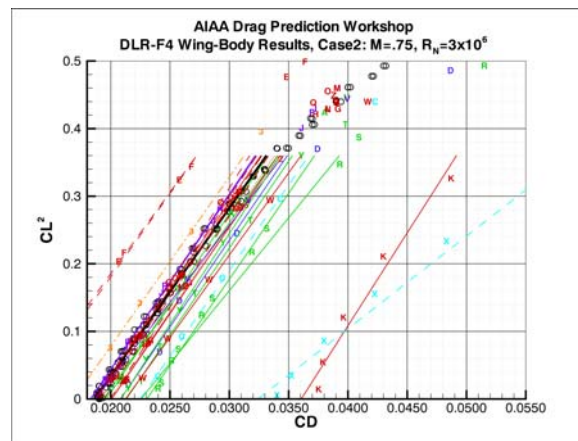
- To build on the results of the first AIAA Drag Prediction Workshop by investigating incremental drag as well as total drag.
- To assess the state-of-the-art computational methods as practical aerodynamic tools for aircraft force and moment prediction of increasingly complex geometries.
- To provide an impartial forum for evaluating the effectiveness of existing computer codes and modeling techniques using Navier-Stokes solvers.
- To identify areas needing additional research and development.



CFD Drag Prediction Workshop

General Information

- This workshop is open to participants worldwide. Efforts will be made to ensure representation from all areas of industry, academia and government labs.
- The trip locations from the wind tunnel tests will be provided to better match the wind tunnel drag data. Computations may be fully turbulent if necessary.
- An open forum will be included in the workshop to discuss the solutions and modeling techniques.
- The results will be made available after the workshop in a report and on the web page.
- A nominal registration fee will be required for attendance.



CFD Drag Prediction Workshop

Test Cases

Required Cases

1. Single Point Grid Sensitivity Study
Mach = 0.75
 $C_L = 0.500 \pm 0.001$
2. Drag Polar
Mach = 0.75
 α (Deg) = $-3^\circ, -2^\circ, -1.5^\circ, -1^\circ, 0^\circ, 1^\circ, 1.5^\circ$

Optional Cases

3. Single Points
Mach = 0.75
 $C_L = 0.400 \pm 0.001$
 $C_L = 0.600 \pm 0.001$
4. Constant C_L Mach Sweep
M = .50, .60, .70, .72, .74, .75, .76, .77
 $C_L = 0.500 \pm 0.001$

Test cases must be run for both configurations. Upon acceptance of the entry request, files will be provided for Point-Matched Structured-Grid, Over-Set Structured-Grid and Unstructured Grids for case (1).

All participants are encouraged to build their own grids. Grids submitted for consideration as baseline grids must include two additional grids for the Grid Sensitivity Analysis. The DPW Organizing committee will coordinate the distribution of the provided grids. An IGES model and CATIA model will be provided for building grids. As many grids as possible will be made available.

The geometry and additional information about the workshop will be available on the web site as it becomes available.

<http://ad-www.larc.nasa.gov/tsab/cfdlarc/aiaa-dpw/index.html>

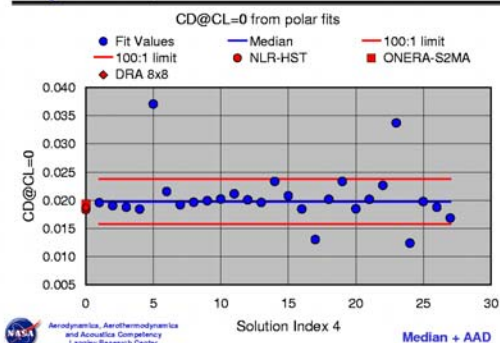
CFD Drag Prediction Workshop

Focus

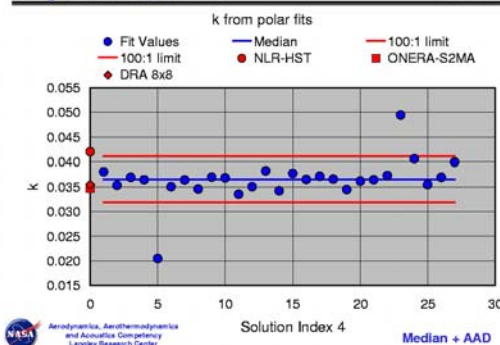
The focus of this workshop will be on drag prediction accuracy and component drag increments. The DLR-F6 model has been chosen to compare a transport configuration with nacelles on and off. The interference drag as well as the prediction of the drag polar and drag rise will be particularly emphasized.

A statistical framework will be used to assess the results.

Drag Polar Fits, $M=0.75$ DPW



Drag Polar Fits, $M=0.75$ DPW



CFD Drag Prediction Workshop

Workshop Committee

John Vassberg

Ed Tinoco

The Boeing Company

Mark Rakowitz

Olaf Brodersen

Jean-Luc Godard

DLR, ONERA

Rich Wahls

Joe Morrison

NASA Langley Research Center

Tom Zickuhr

Kelly Laflin

Cessna Aircraft Company

Steve Klausmeyer

Raytheon Aircraft Company

Dates

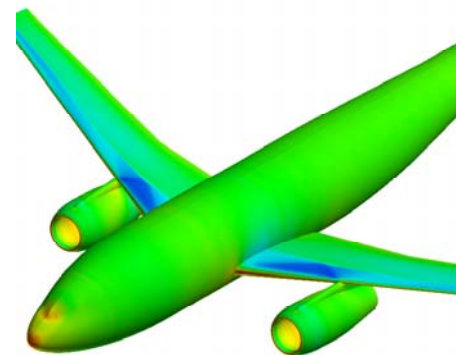
Geometry Available	Aug 5, 2002
Grid Submissions	Oct 4, 2002
Entry Requests	Nov 8, 2002
Abstracts Due	Dec 20, 2002
Notification	Feb 14, 2003
Registration	Apr 11, 2003
Data Submittal	May 16, 2003
Workshop	June 21-22, 2003

The Workshop presentations will not be official AIAA papers, however, several participants will be invited to support a special session on drag prediction to be held during the AIAA Aerospace Sciences Meeting, January 2004.

2nd AIAA CFD Drag Prediction Workshop

Sponsored
by the
Applied Aerodynamics TC

2-Day Workshop
Preceding
21ST APA Conference
Orlando, FL
June 21-22, 2003



For more information and results from
the first workshop:

[http://ad-www.larc.nasa.gov/tsab/
cfdlarc/aiaa-dpw/index.html](http://ad-www.larc.nasa.gov/tsab/cfdlarc/aiaa-dpw/index.html)

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