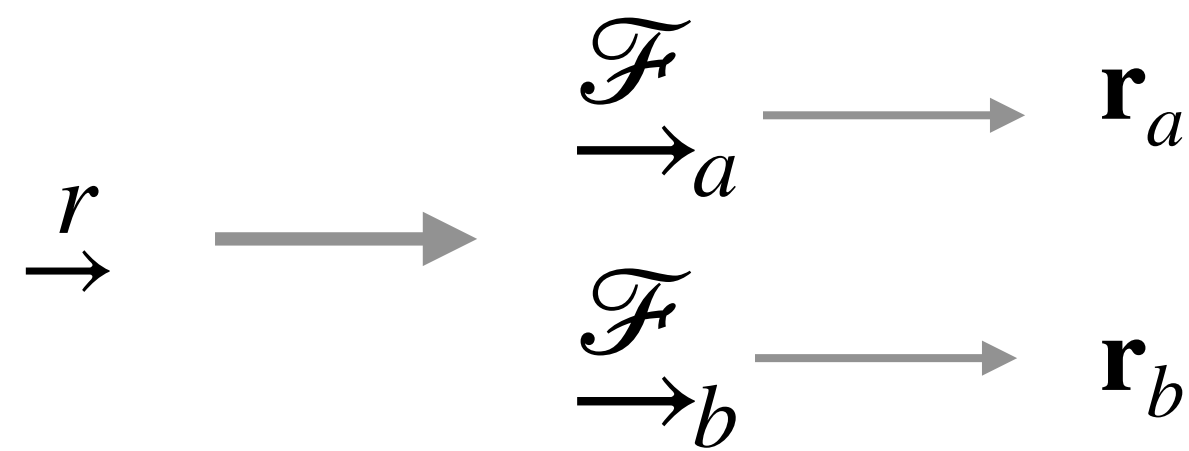


Module 3 | Lesson 1

# 3D GEOMETRY AND REFERENCE FRAMES

# Coordinate Rotations

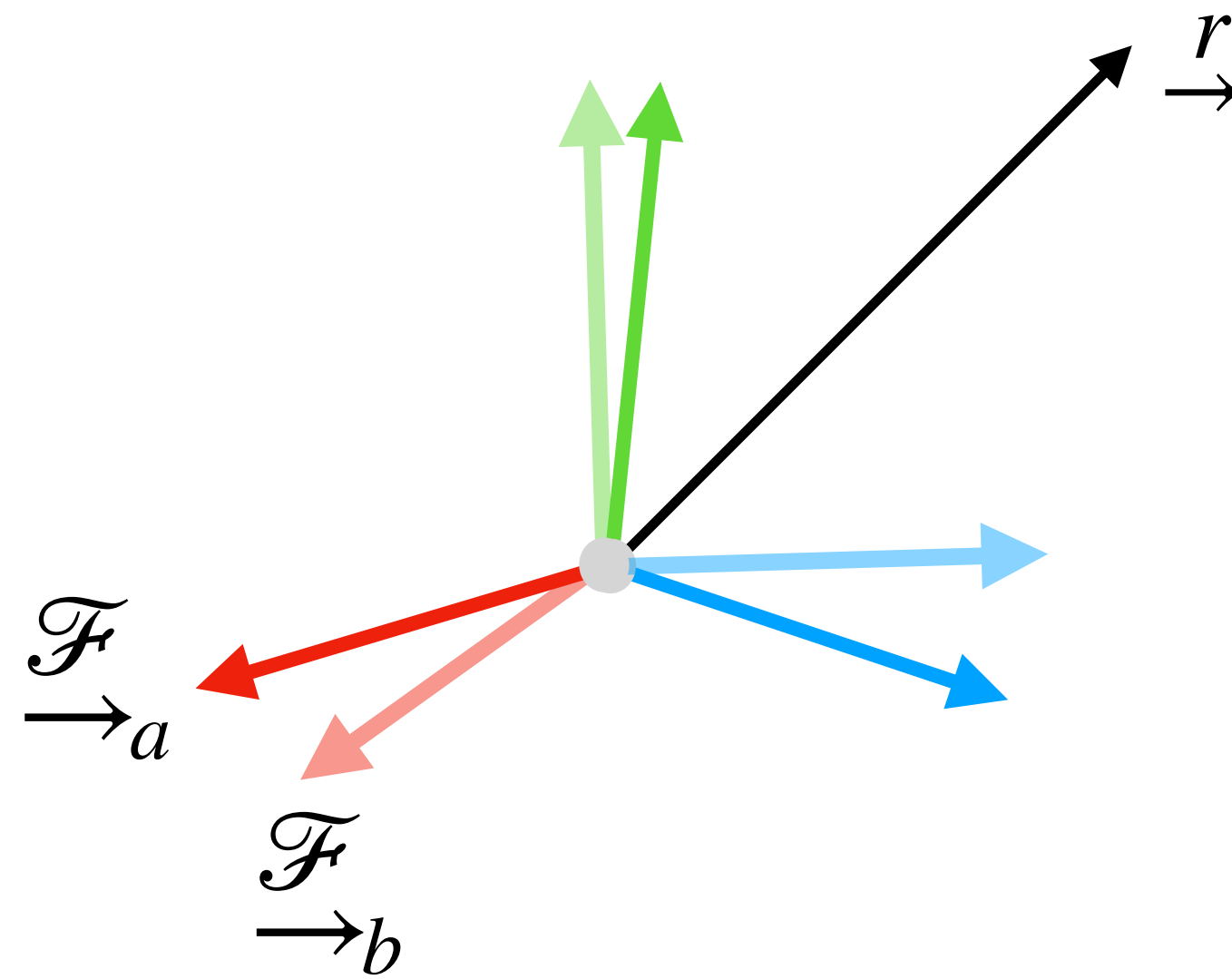
Vectors can be expressed in different coordinate frames:



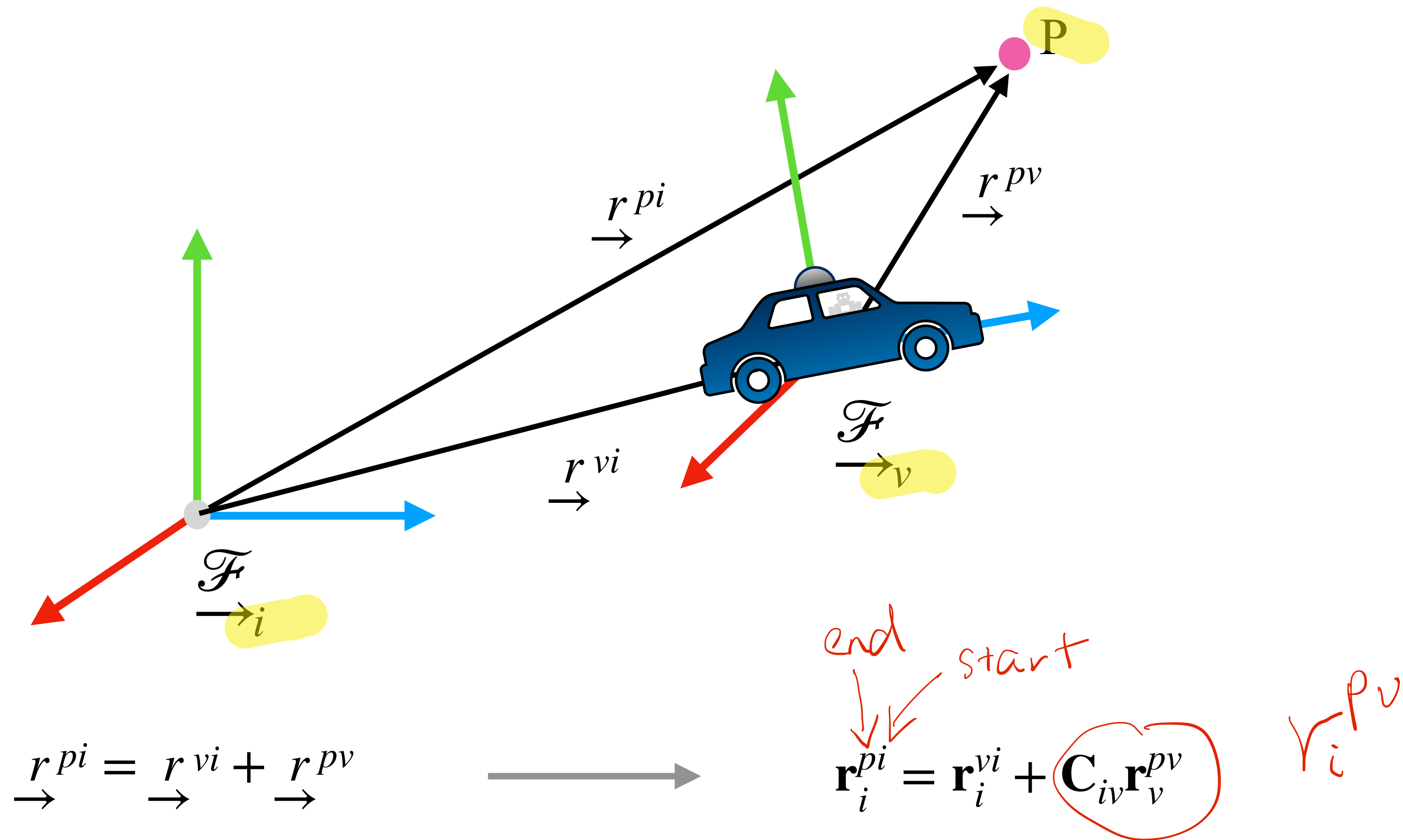
The coordinates of the vector are related through a *rotation matrix*:

$$\mathbf{r}_b = \mathbf{C}_{ba} \mathbf{r}_a$$

takes coordinates in frame  $a$   
and rotates them into frame  $b$



# Transformations



# How Can We Represent a Rotation?

1

Rotation matrix

$$\mathbf{C}_{ba} = \begin{bmatrix} \vec{b}_{\rightarrow 1} \\ \vec{b}_{\rightarrow 2} \\ \vec{b}_{\rightarrow 3} \end{bmatrix} \begin{bmatrix} \vec{a}_{\rightarrow 1} & \vec{a}_{\rightarrow 2} & \vec{a}_{\rightarrow 3} \end{bmatrix}$$
$$= \begin{bmatrix} \vec{b}_{\rightarrow 1} \cdot \vec{a}_{\rightarrow 1} & \vec{b}_{\rightarrow 1} \cdot \vec{a}_{\rightarrow 2} & \vec{b}_{\rightarrow 1} \cdot \vec{a}_{\rightarrow 3} \\ \vec{b}_{\rightarrow 2} \cdot \vec{a}_{\rightarrow 1} & \vec{b}_{\rightarrow 2} \cdot \vec{a}_{\rightarrow 2} & \vec{b}_{\rightarrow 2} \cdot \vec{a}_{\rightarrow 3} \\ \vec{b}_{\rightarrow 3} \cdot \vec{a}_{\rightarrow 1} & \vec{b}_{\rightarrow 3} \cdot \vec{a}_{\rightarrow 2} & \vec{b}_{\rightarrow 3} \cdot \vec{a}_{\rightarrow 3} \end{bmatrix}$$

$$\mathbf{C}_{ba} \in \mathcal{R}^{3 \times 3}$$

$$\mathbf{r}_b = \mathbf{C}_{ba} \mathbf{r}_a$$

$$\mathbf{C}_{ba} \mathbf{C}_{ba}^T = \mathbf{C}_{ba} \mathbf{C}_{ab} = \mathbf{1}$$

$$\mathbf{C}_{ba}^T = \mathbf{C}_{ba}^{-1}$$

“direction  
cosine matrix”  
(DCM)

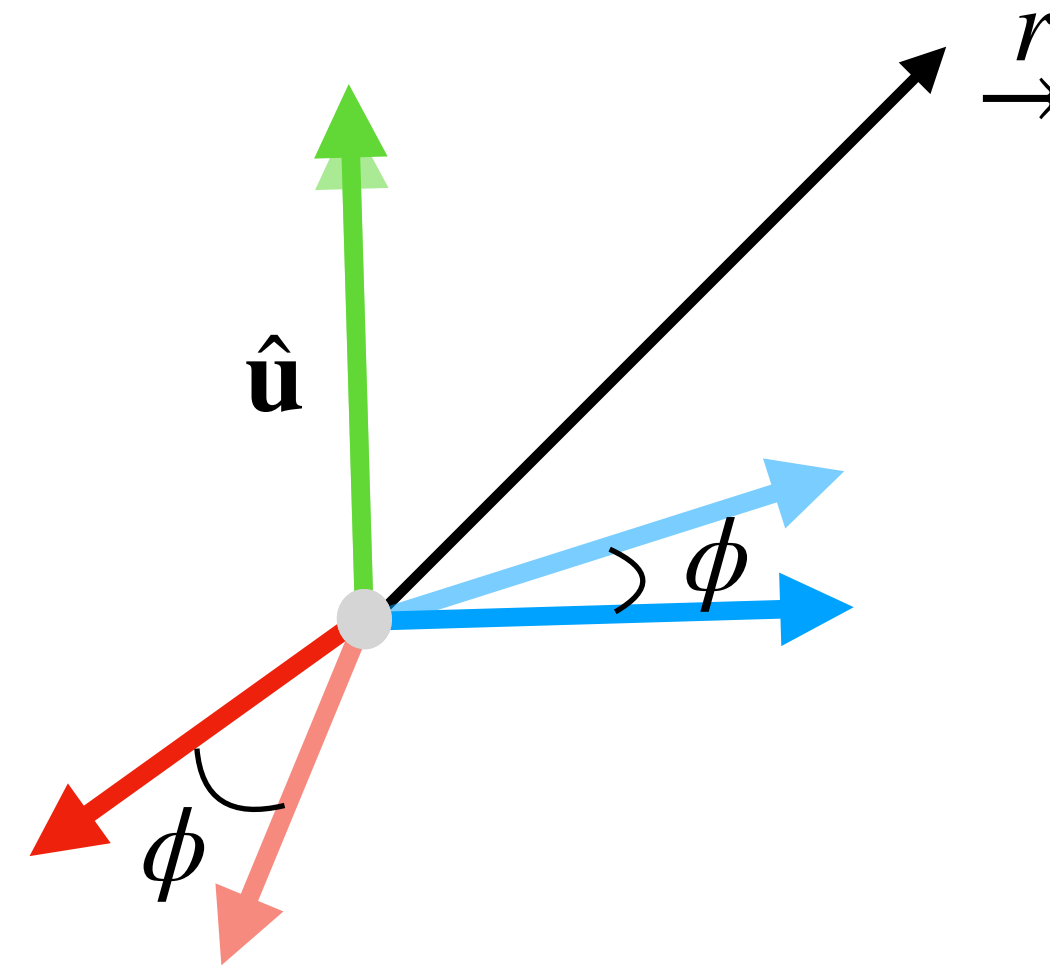
# How Can We Represent a Rotation?

2

Unit quaternions

$$\mathbf{q} = \begin{bmatrix} q_w \\ \mathbf{q}_v \end{bmatrix} = \begin{bmatrix} \cos \frac{\phi}{2} \\ \hat{\mathbf{u}} \sin \frac{\phi}{2} \end{bmatrix}$$

$$\|\mathbf{q}\| = 1$$



$$\mathbf{r}_b = \mathbf{C}(\mathbf{q}_{ba})\mathbf{r}_a$$

$$\mathbf{C}(\mathbf{q}) = (q_w^2 - \mathbf{q}_v^T \mathbf{q}_v)\mathbf{1} + 2\mathbf{q}_v \mathbf{q}_v^T + 2q_w[\mathbf{q}_v]_{\times}$$

where

$$[\mathbf{a}]_{\times} = \begin{bmatrix} 0 & -a_z & a_y \\ a_z & 0 & -a_x \\ -a_y & a_x & 0 \end{bmatrix}$$

# Quaternion Multiplication and Rotations

Quaternions multiplication is a special operation that is *associative* but is not commutative in general (just like matrix multiplication!):

*The order  
matters*

$$\mathbf{p} \otimes \mathbf{q} = \begin{bmatrix} p_w q_w - \mathbf{p}_v^T \mathbf{q}_v \\ p_w \mathbf{q}_v + q_w \mathbf{p}_v + [\mathbf{p}_v]_{\times} \mathbf{q}_v \end{bmatrix}$$

quaternion product  
operator

Sequential rotation operations can also be performed by taking advantage of quaternion multiplication:

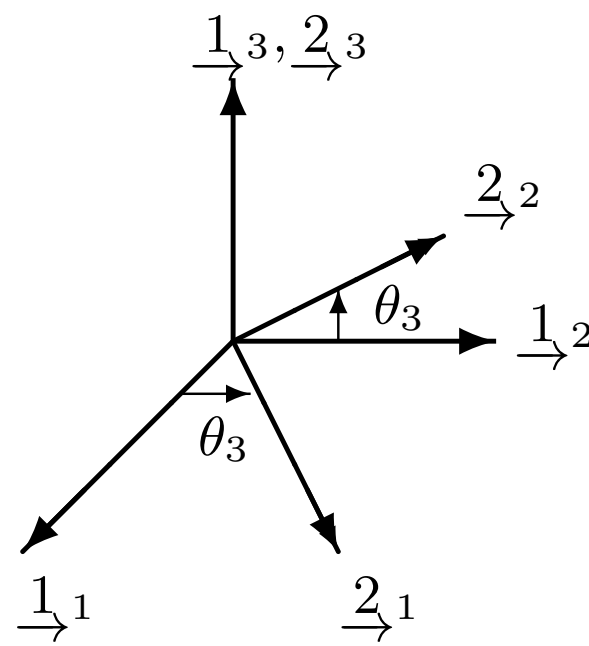
$$\mathbf{C}(\mathbf{p} \otimes \mathbf{q}) = \mathbf{C}(\mathbf{p}) \mathbf{C}(\mathbf{q})$$

# How Can We Represent a Rotation?

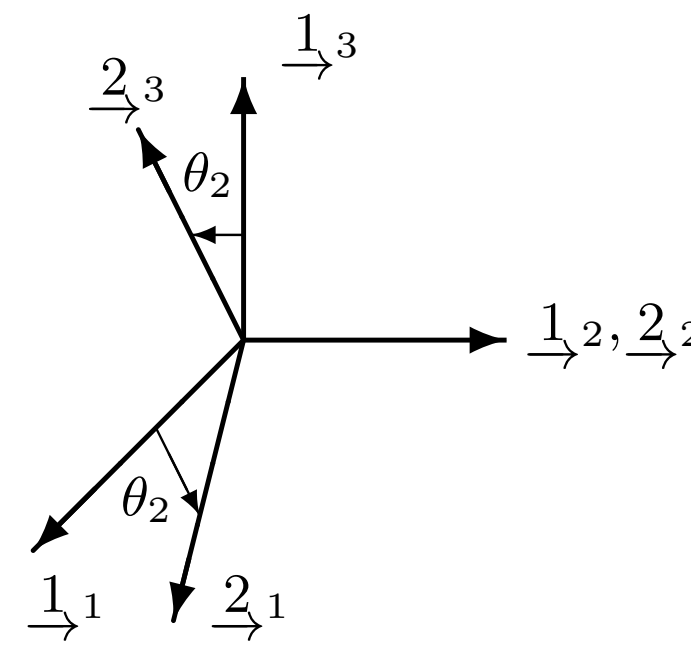
3

## Euler angles

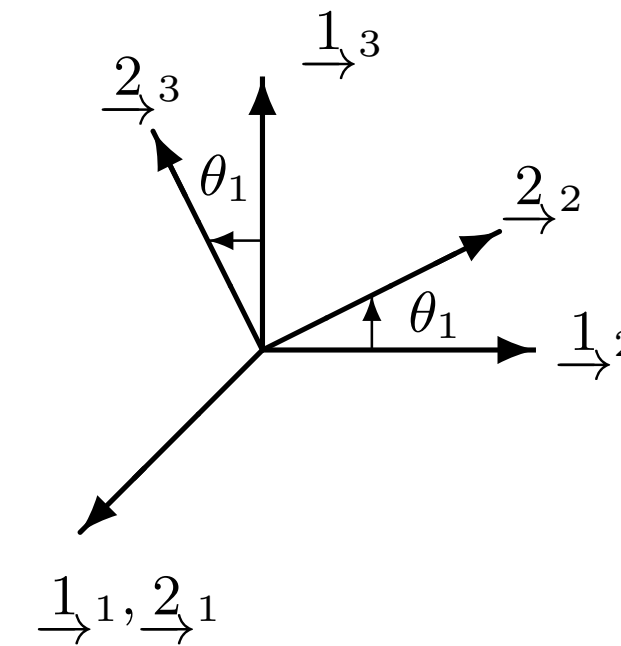
$$\mathbf{C}(\theta_3, \theta_2, \theta_1) = \mathbf{C}_3(\theta_3) \mathbf{C}_2(\theta_2) \mathbf{C}_1(\theta_1)$$



about the 3-axis



about the 2-axis



about the 1-axis

$$\mathbf{C}_3 = \begin{bmatrix} \cos \theta_3 & -\sin \theta_3 & 0 \\ \sin \theta_3 & \cos \theta_3 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

$$\mathbf{C}_2 = \begin{bmatrix} \cos \theta_2 & 0 & \sin \theta_2 \\ 0 & 1 & 0 \\ -\sin \theta_2 & 0 & \cos \theta_2 \end{bmatrix}$$

$$\mathbf{C}_1 = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \theta_1 & -\sin \theta_1 \\ 0 & \sin \theta_1 & \cos \theta_1 \end{bmatrix}$$

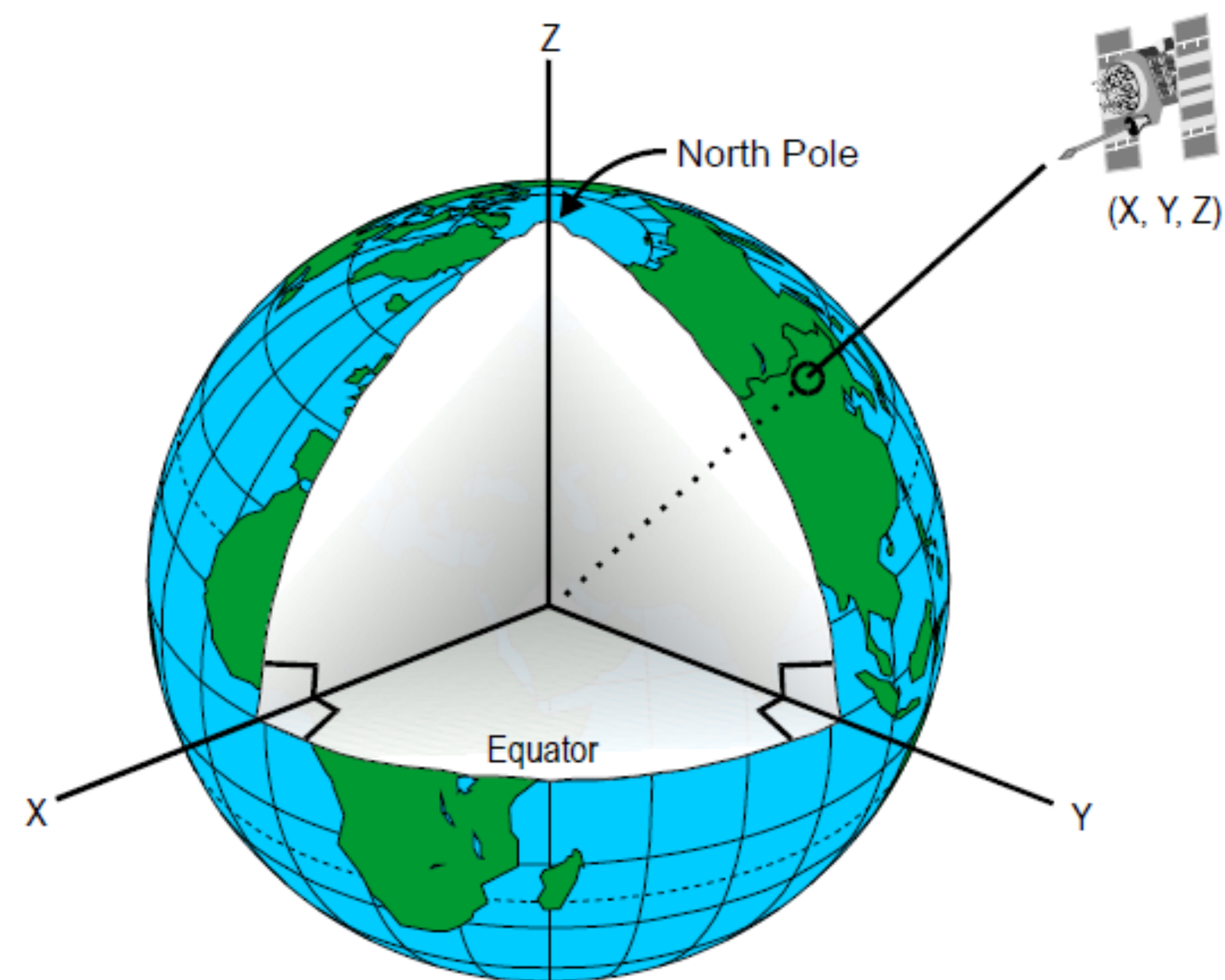
# Which Rotation Representation Should I Use?

	Rotation Matrix	Unit quaternion	Euler angles
<i>Expression</i>	$\mathbf{C}$	$\mathbf{q} = \begin{bmatrix} \cos \frac{\phi}{2} \\ \hat{\mathbf{u}} \sin \frac{\phi}{2} \end{bmatrix}$	$\{\theta_3, \theta_2, \theta_1\}$
<i>Parameters</i>	9	4	3
<i>Constraints</i>	$\mathbf{C}\mathbf{C}^T = \mathbf{1}$	$ \mathbf{q}  = 1$	None*
<i>Singularities?</i>	No	No	Yes!

For particular rotations,  
two Euler angles are  
indistinguishable.



# Reference Frames | ECIF



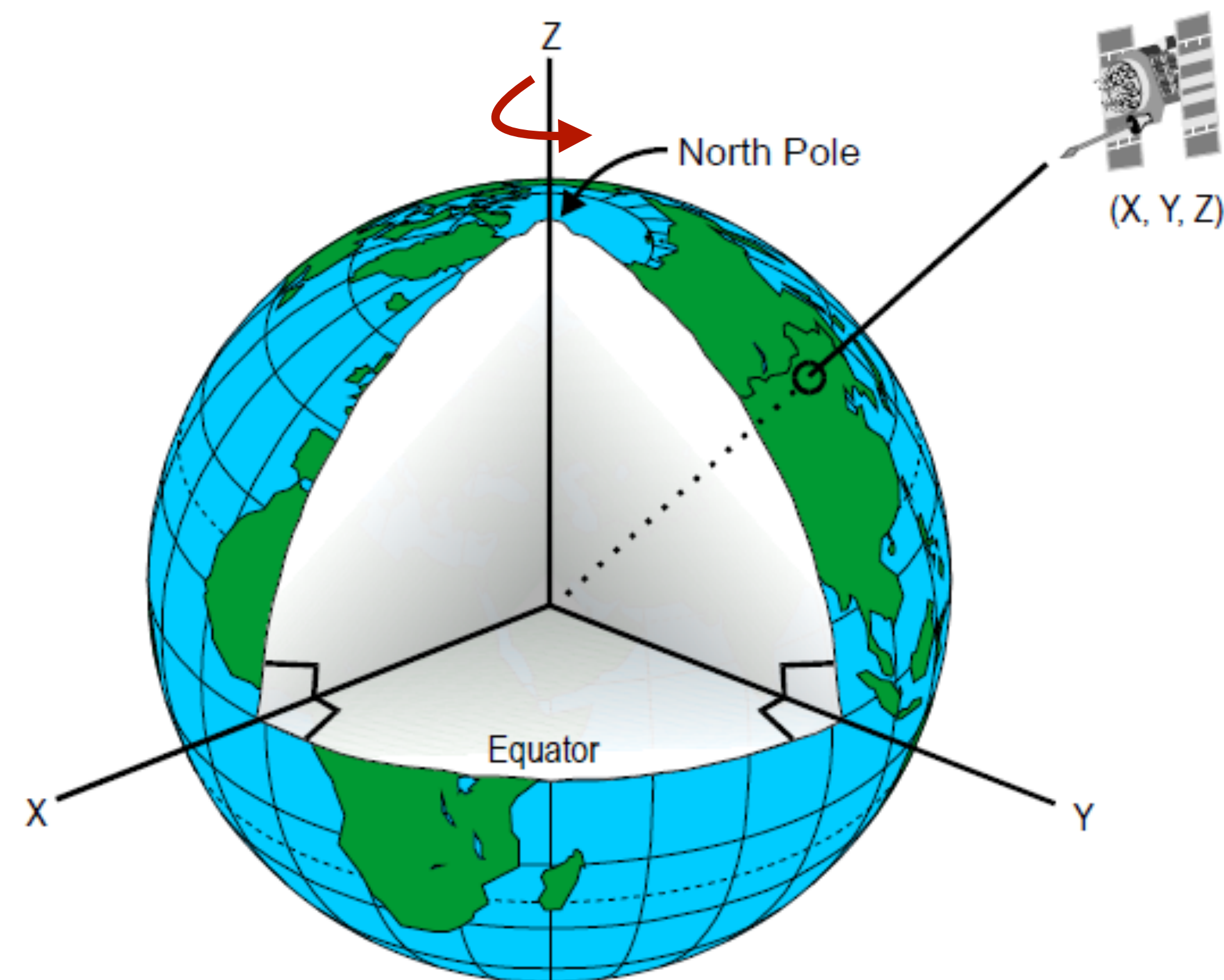
Earth-Centred Inertial Frame

ECIF coordinate frame is fixed,  
Earth rotates about the z axis.

$$\mathcal{F} \rightarrow \text{ECIF}$$

$x$	<i>fixed w.r.t. stars</i>
$y$	<i>fixed w.r.t. stars</i>
$z$	true north

# Reference Frames | ECEF



Earth-Centred Earth-Fixed Frame

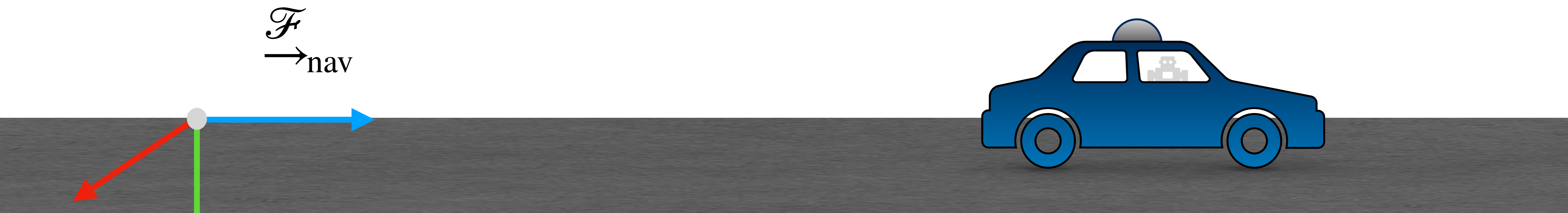
ECEF coordinate frame rotates with the Earth.

$\mathcal{F}$   
 $\rightarrow$ ECEF

x	prime meridian (on equator)
y	RHR
z	true north

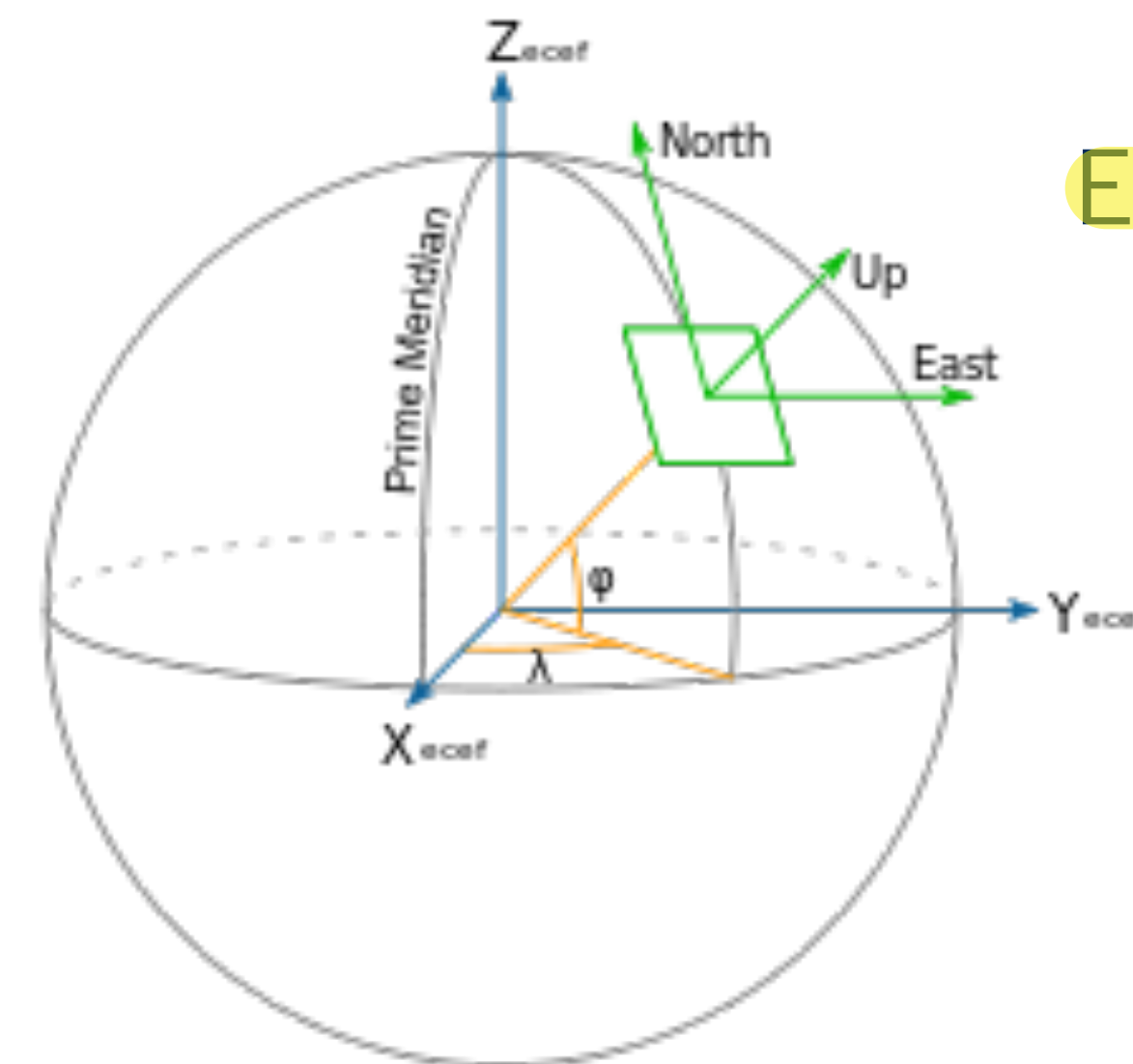
right hand rule

# Reference Frames | Navigation



NED Frame

$x$	True North
$y$	True East
$z$	Down (along gravity)



ENU Frame

# Reference Frames | Sensor & Vehicle

