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## Properties of a standard atmosphere

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#### THE PREPARATION OF THIS DATA ITEM

The work on this particular Data Item which supersedes ESDU Aero 00.01.02 to 04, was monitored and guided by the Aerodynamics Committee. This Committee first met in 1942 and now has the following membership:

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### PROPERTIES OF A STANDARD ATMOSPHERE

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### PROPERTIES OF A STANDARD ATMOSPHERE

#### 1. NOTATION AND UNITS

		SI	British
a	speed of sound	m/s	ft/s
g	acceleration due to gravity	$m/s^2$	$ft/s^2$
H	geopotential altitude	m	ft
k	thermal conductivity	W/m K	ft lbf/ft s K
L	mean free path	m	ft
l	characteristic length	m	ft
m	mean molecular mass*	kg/kmol	slug/slug-mol
p	pressure	$N/m^2$	lbf/ft <sup>2</sup>
$\mathscr{R}$	universal gas constant	N m/kmol K	ft lbf/slug-mol K
$Re^*$	Reynolds number at Mach 1, (= $\rho a l/\mu$ )		
$r_e$	radius of earth	m	ft
T	temperature	K	K
$ar{V}$	particle mean speed	m/s	ft
Z	geometric altitude	m	ft
μ	dynamic viscosity	$N s/m^2$	lbf s/ft <sup>2</sup>
v	kinematic viscosity	$m^2/s$	$ft^2/s$
ρ	density	$kg/m^3$	slug/ft <sup>3</sup>
Subscript			

<sup>0</sup> denotes sea-level values

Based upon the carbon-12 isotope scale for which the atomic mass of the nuclide <sup>12</sup>C = 12. This scale was adopted in 1961 by the Conference of the International Union of Pure and Applied Chemistry.

#### **Constants and Sea-level Values**

Quantity	SI	British
$a_0$	340.294 m/s	1116.45 ft/s
$g_0$	9.80665 m/s <sup>2</sup>	$32.174049 \text{ ft/s}^2$
$k_0$	$2.5343 \times 10^{-2} \text{ W/m K}$	$5.6973 \times 10^{-3}$ ft lbf/ft s K
$m_0^*$	28.96442 kg/kmol	28.96442 slug/slug-mol
$p_0$	$1.01325 \times 10^5 \text{ N/m}^2$	$2.11622 \times 10^3  \text{lbf/ft}^2$
R	$8.31432 \times 10^3 \text{ N m/kmol K}$	$8.94946 \times 10^4$ ft lbf/slug-mol K
$r_e$	$6.356766 \times 10^6 \mathrm{m}$	$2.085553 \times 10^7 \text{ ft}$
$T_0$	288.15 K	288.15 K
$\mu_0$	$1.7894 \times 10^{-5} \text{ N s/m}^2$	$3.7372 \times 10^{-7} \text{ lbf s/ft}^2$
$v_0$	$1.4607 \times 10^{-5} \mathrm{m}^2/\mathrm{s}$	$1.5723 \times 10^{-4} \text{ ft}^2/\text{s}$
$\rho_0$	$1.225 \text{ kg/m}^3$	$2.3768924 \times 10^{-3} \text{ slug/ft}^3$

The value adopted for  $m_0$  is identical with that quoted in Derivation 1, and is derived by substituting the SI sea-level values for  $\rho_0$ ,  $p_0$  and  $T_0$  into the perfect gas law. Derivation 2 quotes a value of 28.9644. For all practical purposes both values of  $m_0$  will give the same results.

#### 2. INTRODUCTION

This Item gives properties of a standard atmosphere for altitudes up to 1000 km. The information given is extracted from Derivations 1 and 2. The data in this Item, both in the intervals of altitude and the number of significant figures quoted are much less comprehensive than those in Derivation 2. Users who require more detailed information should refer to that Derivation or, for altitudes below 50 km, to Item Nos 68046 and 72018 (Reference 9) which give data for standard and non-standard atmospheres. For altitudes up to 80 km, Appendix A of this Item gives the equations needed to construct the atmospheric model, in a manner suitable for computer programming.

The atmosphere considered in the Item is an idealised, mean-annual, steady-state model assumed to exist in a period of moderate solar activity at a latitude of about 45°N and to rotate with the earth. The effects of dust and moisture in the atmosphere are neglected.

Up to 80 km the atmosphere is represented by the ideal gas law, a linearly segmented temperature-altitude profile (see Sketch 7.1) and the assumption of hydrostatic equilibrium in which the air is treated as a homogeneous mixture of the several constituent gases. At greater heights, where dissociation and diffusion processes produce significant departures from homogeneity the definitions governing the atmosphere are more sophisticated, involving the treatment of individual gas species (see Derivation 2). The temperature-altitude profile between 80 km and 1000 km is not expressed in linear segments but by functions chosen to fit the observed data and provide a continuous first derivative with respect to altitude for the entire altitude regime.

#### 3. STATUS OF ATMOSPHERE

The data presented in Derivations 1 and 2 are the culmination of many years' work and represent the latest stage in a continuing process of defining a standard atmosphere. Standard atmospheres were originally developed in the 1920's in the United States and in Europe to satisfy a need for standardisation of aircraft instruments and aircraft performance. The US atmosphere was generated by the National Advisory Committee on Aeronautics (NACA), Reference 3 (and later supplements), while the European atmosphere was generated by the International Commission for Aerial Navigation (ICAN), Reference 4. There were slight differences between these two independently derived atmospheres. These differences were reconciled and international uniformity was achieved through adoption by the International Civil Aviation Organization (ICAO) in 1952 of a new international standard atmosphere for altitudes up to 20 km, Reference 5. Work on extending this atmosphere to an altitude of 300 km was undertaken by the US Committee on Extension of the Standard Atmosphere (COESA) and reported in Reference 6. The gathering of more data by rockets and satellites enabled a further extension to 700 km to be made by COESA, which was published in 1962, Reference 7. This atmosphere was adopted in 1964 as a new standard by ICAO for altitudes up to 32 km, Reference 8, superseding the 1952 ICAO atmosphere. Derivation 2 is a revision and extension to 1000 km of Reference 7 as a result of more experimental data being gathered. It is identical with Reference 7 up to an altitude of 51 km and therefore still agrees with the ICAO standard atmosphere. In 1975 the International Organization for Standardization (ISO) adopted a standard atmosphere, Derivation 1, which covers heights up to 80 km. For heights below 50 km the atmosphere is termed "International Standard Atmosphere" while for heights between 50 and 80 km it is termed "Interim Standard Atmosphere". For all practical purposes, the ISO atmosphere is identical with the data in Derivation 2. The World Meteorological Organization Standard Atmosphere, defined between -2 km and 32 km, is identical with the data in Derivation 1.

#### 4. ALTITUDE SCALE

For atmosphere modelling it is common practice to work in terms of geopotential altitude, *H*. This is related to geometric altitude, *Z*, by the equation

$$H = \int_0^Z \frac{g}{g_0} \, \mathrm{d}Z. \tag{4.1}$$

By neglecting centrifugal acceleration and using only Newton's law of gravity, this reduces to

$$H = \frac{r_e Z}{r_o + Z}. ag{4.2}$$

The advantage of working with H is that it enables a constant (sea-level) value of g to be assumed when solving the hydrostatic equation. That is

$$dp = -g\rho dZ \tag{4.3}$$

becomes

$$dp = -g_0 \rho dH. (4.4)$$

No early revision of the "Interim" atmosphere is envisaged.

The basic equations defining the atmospheric temperature profile are defined in terms of H for altitudes up to 80 km, see Appendix A. Figures 1 (feet) and 2 (metres) show the difference between Z and H plotted against Z. At low altitudes the difference is very small. In this Item, atmospheric properties are tabulated against geopotential altitude up to H = 80 km, and then against geometric altitude. (See Item Nos 68046 and 77022 (Reference 9 and 10) for altitude scales in non-standard atmospheres.)

#### 5. PRESENTATION

Table 8.1 presents atmospheric data for altitudes up to  $2.5 \times 10^5$  ft at intervals of 1000 ft. Table 8.1 presents corresponding data for altitudes up to 80 km at intervals of 0.5 km. Table 8.3 gives atmospheric data from 80 km to 1000 km, at intervals of 5 km to 300 km and then at intervals of 10 km. In these tables, a one or two digit number (preceded by a plus or minus sign) following the initial entry of a block indicates the power of ten by which that and each succeeding entry in the block should be multiplied. A change of power occurring within a block is indicated by a similar notation. Sketches 7.1 to 7.11 indicate, roughly, the variation with altitude of the more important atmospheric parameters. The equations necessary to derive the data in Tables 8.1 and 8.1 are given in Appendix A in a manner suitable for programming.

#### 6. DERIVATION AND REFERENCES

#### **Derivation**

1. –	Standard Atmosphere. International Organization for Standardization, ISO-2533-1975(E), 1975.
2. –	US Standard Atmosphere, 1976. US Committee on Extension to the Standard Atmosphere. US Government Printing Office, 1976.

### References

10.

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3.	GREGG, W.R.	Standard Atmosphere. NACA Rep. 147, 1922.							
4.	_	ICAN Standard Atmosphere. Air Publication 1275, 1924. (See also ARC R & M 1891 (1941) Pankhurst R.C and Conn J.F.C. Physical properties of the Standard Atmosphere.)							
5.	-	Manual of the ICAO Standard Atmosphere. Document 7488, International Civil Aviation Organization, 1954.							
6.	MINZER, R.A. RIPLEY, W.S. CONDRON, T.P.	US extension to the ICAO Standard Atmosphere Tables and Data to 300 standard geopotential kilometres. Geophys. Res. Dir. and U.S. Weather Bureau, 1958.							
7.	-	US Standard Atmosphere, 1962. US Committee on Extension to the Standard Atmosphere. US Government Printing Office, 1962.							
8.	_	Manual of the ICAO Standard Atmosphere extended to 32 km. Document 7488/2, International Civil Aviation Organization, 1964.							
9.	ESDU	Atmospheric data for performance calculations. ESDU Item No. 68046 (Performance Series), 1968. (See also Addendum Item 72018, height in feet data in SI units.)							

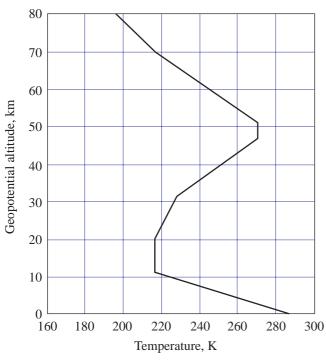
Equations for calculation of international standard atmosphere and

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associated off-standard atmospheres.

ESDU Item No. 77022 (Performance Series), 1977.

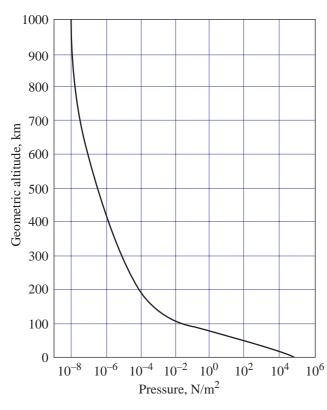
#### 7.

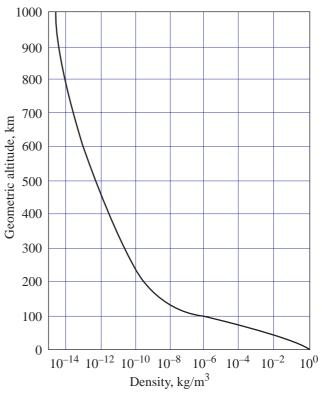


500 400 Geometric altitude, km 300 200 100 800 1000 1200 1400 1600 200 400 Temperature, K



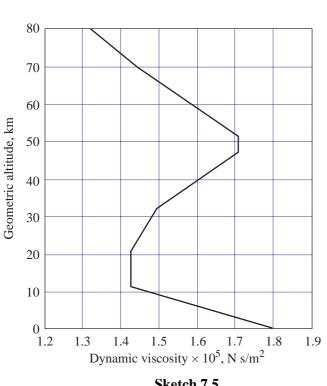


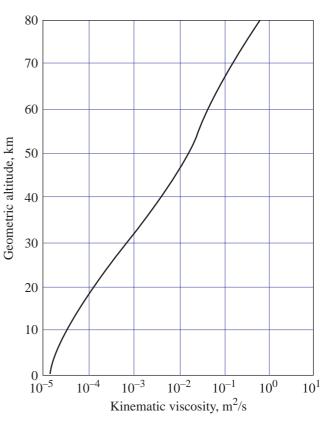




Sketch 7.3

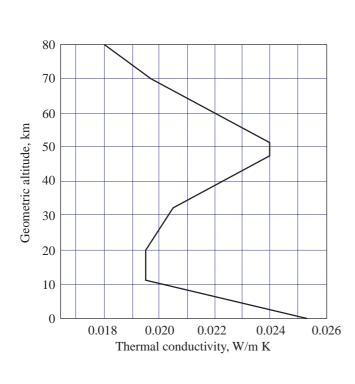
Sketch 7.4

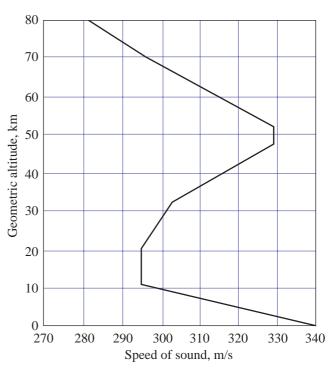




Sketch 7.5



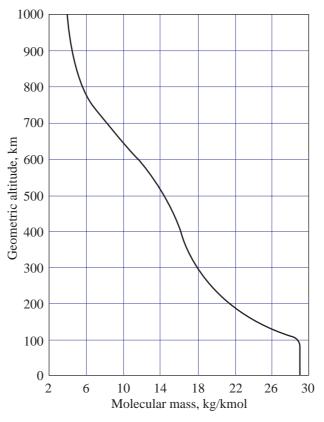




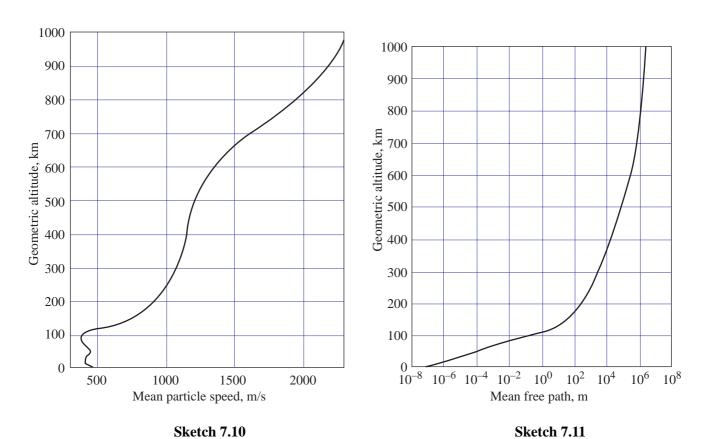
Sketch 7.7

Sketch 7.8

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Sketch 7.9



#### 8. TABLES

TABLE 8.1 Properties of a Standard Atmosphere: Geopotential Altitude in Feet (-2000 ft to 250 000 ft)

Н	T	,	,	,	,	,		$Re^*/l$
(ft)	(K)	$a/a_0$	$p/p_0$	$\rho/\rho_0$	<i>v</i> / <i>v</i> <sub>0</sub>	μ/μ <sub>0</sub>	$k/k_0$	$(ft^{-1})$
-2.00 +3 -1.00	292.1 290.1	1.007 0 1.003	1.074 0 1.037	1.060 0 1.030	9.536 -1 9.764	1.011 0 1.005	1.012 0 1.006	7.497 +6 7.297
0.00 0 1.00 +3 2.00 3.00 4.00 5.00 6.00 7.00 8.00 9.00	288.2 286.2 284.2 282.2 280.2 278.2 276.3 274.3 272.3 270.3	1.000 0 9.966 -1 9.931 9.896 9.862 9.827 9.792 9.756 9.721 9.686	1.000 0 9.644 -1 9.298 8.962 8.637 8.320 8.014 7.716 7.428 7.148	1.000 0 9.711 -1 9.428 9.151 8.881 8.617 8.359 8.106 7.860 7.620	1.000 0 1.024 1.049 1.075 1.102 1.129 1.158 1.187 1.217 1.248	1.000 0 9.946 -1 9.893 9.839 9.785 9.731 9.676 9.622 9.567 9.512	1.000 0 9.938 -1 9.877 9.815 9.753 9.691 9.629 9.567 9.504 9.442	7.101 +6 6.909 6.720 6.536 6.355 6.179 6.006 5.837 5.671 5.509
1.00 +4 1.10 1.20 1.30 1.40 1.50 1.60 1.70 1.80 1.90	268.3 266.4 264.4 262.4 260.4 258.4 256.5 254.5 252.5 250.5	9.650 -1 9.614 9.579 9.543 9.507 9.470 9.434 9.397 9.361 9.324	6.877 -1 6.614 6.360 6.113 5.875 5.643 5.420 5.203 4.994 4.791	7.385 -1 7.156 6.932 6.713 6.500 6.292 6.090 5.892 5.699 5.511	1.281 0 1.314 1.348 1.384 1.421 1.459 1.498 1.539 1.581 1.624	9.457 -1 9.401 9.346 9.290 9.234 9.178 9.122 9.065 9.008 8.951	9.379 -1 9.316 9.253 9.190 9.127 9.064 9.001 8.937 8.873 8.810	5.351 +6 5.196 5.045 4.897 4.752 4.610 4.472 4.337 4.205 4.076
2.00 +4 2.10 2.20 2.30 2.40 2.50 2.60 2.70 2.80 2.90	248.5 246.5 244.6 242.6 240.6 238.6 236.6 234.7 232.7 230.7	9.287 -1 9.250 9.213 9.175 9.138 9.100 9.062 9.024 8.986 8.948	4.595 -1 4.406 4.223 4.046 3.876 3.711 3.552 3.398 3.250 3.107	5.328 -1 5.150 4.976 4.807 4.642 4.481 4.325 4.173 4.025 3.881	1.669 0 1.716 1.764 1.814 1.866 1.920 1.976 2.034 2.094 2.157	8.894 -1 8.837 8.779 8.721 8.663 8.605 8.547 8.488 8.429 8.370	8.746 -1 8.682 8.618 8.554 8.489 8.425 8.360 8.295 8.231 8.166	3.950 +6 3.828 3.708 3.591 3.476 3.365 3.256 3.150 3.047 2.946
3.00 +4 3.10 3.20 3.30 3.40 3.50 3.60 3.70 3.80 3.90	228.7 226.7 224.8 222.8 220.8 218.8 216.7 216.7 216.7	8.909 -1 8.870 8.832 8.793 8.753 8.714 8.675 8.671 8.671	2.970 -1 2.837 2.709 2.586 2.467 2.353 2.243 2.138 2.038 1.942	3.741 -1 3.605 3.473 3.345 3.220 3.099 2.981 2.844 2.710 2.583	2.221 0 2.289 2.359 2.431 2.507 2.585 2.667 2.794 2.932 3.076	8.311 -1 8.251 8.192 8.132 8.071 8.011 7.950 7.945 7.945 7.945	8.100 -1 8.035 7.970 7.904 7.839 7.773 7.707 7.701 7.701 7.701	2.848 +6 2.752 2.659 2.568 2.480 2.394 2.310 2.204 2.100 2.002
4.00 +4 4.10 4.20 4.30 4.40 4.50 4.60 4.70 4.80 4.90	216.7 216.7 216.7 216.7 216.7 216.7 216.7 216.7 216.7 216.7	8.671 -1 8.671 8.671 8.671 8.671 8.671 8.671 8.671 8.671 8.671	1.851 -1 1.764 1.681 1.602 1.527 1.455 1.387 1.322 1.260 1.201	2.462 -1 2.346 2.236 2.131 2.031 1.936 1.845 1.758 1.676 1.597	3.227 0 3.386 3.553 3.728 3.911 4.104 4.306 4.518 4.741 4.974	7.945 -1 7.945 7.945 7.945 7.945 7.945 7.945 7.945 7.945 7.945 7.945	7.701 -1 7.701 7.701 7.701 7.701 7.701 7.701 7.701 7.701 7.701 7.701 7.701	1.908 +6 1.818 1.733 1.652 1.574 1.500 1.430 1.363 1.299 1.238
5.00 +4 5.10 5.20 5.30 5.40 5.50 5.60 5.70 5.80 5.90	216.7 216.7 216.7 216.7 216.7 216.7 216.7 216.7 216.7 216.7	8.671 -1 8.671 8.671 8.671 8.671 8.671 8.671 8.671 8.671 8.671	1.145 -1 1.091 1.040 9.909 -2 9.444 9.001 8.578 8.176 7.792 7.426	1.522 -1 1.451 1.383 1.318 1.256 1.197 1.141 1.087 1.036 9.877 -2	5.219 0 5.476 5.745 6.028 6.325 6.637 6.963 7.306 7.666 8.043	7.945 -1 7.945 7.945 7.945 7.945 7.945 7.945 7.945 7.945 7.945 7.945	7.701 -1 7.701 7.701 7.701 7.701 7.701 7.701 7.701 7.701 7.701 7.701 7.701	1.180 +6 1.124 1.072 1.021 9.734 +5 9.277 8.842 8.427 8.032 7.655

TABLE 8.1 Properties of a Standard Atmosphere: Geopotential Altitude in Feet (continued)

77	T	ı						<i>Re</i> * / <i>l</i>
H (ft)	(K)	a/a <sub>0</sub>	$p/p_0$	$\rho/\rho_0$	$v/v_0$	μ/μ <sub>0</sub>	$k/k_0$	$(ft^{-1})$
6.00 +4 6.10 6.20 6.30 6.40 6.50 6.60 6.70 6.80 6.90	216.7 216.7 216.7 216.7 216.7 216.7 216.8 217.1 217.4 217.7	8.671 -1 8.671 8.671 8.671 8.671 8.671 8.673 8.679 8.686 8.692	7.078 -2 6.746 6.429 6.127 5.840 5.566 5.305 5.056 4.819 4.594	9.414 -2 8.972 8.551 8.150 7.767 7.403 7.052 6.712 6.389 6.081	8.439 0 8.855 9.291 9.748 1.023 +1 1.073 1.127 1.186 1.247 1.312	7.945 -1 7.945 7.945 7.945 7.945 7.945 7.945 7.948 7.958 7.967 7.976	7.701 -1 7.701 7.701 7.701 7.701 7.701 7.705 7.715 7.726 7.736	7.296 +5 6.953 6.627 6.316 6.020 5.737 5.464 5.198 4.945 4.705
7.00 +4 7.10 7.20 7.30 7.40 7.50 7.60 7.70 7.80 7.90	218.0	8.698 -1	4.380 -2	5.789 -2	1.379 +1	7.986 -1	7.746 -1	4.477 +5
	218.3	8.704	4.176	5.512	1.451	7.995	7.756	4.261
	218.6	8.710	3.981	5.248	1.525	8.004	7.766	4.055
	218.9	8.716	3.796	4.997	1.604	8.014	7.776	3.859
	219.2	8.722	3.620	4.758	1.686	8.023	7.786	3.673
	219.5	8.728	3.452	4.531	1.773	8.032	7.796	3.496
	219.8	8.734	3.292	4.316	1.863	8.042	7.807	3.328
	220.1	8.740	3.140	4.110	1.959	8.051	7.817	3.169
	220.4	8.746	2.995	3.915	2.059	8.060	7.827	3.017
	220.7	8.752	2.857	3.730	2.164	8.069	7.837	2.872
8.00 +4	221.0	8.758 -1	2.725 -2	3.553 -2	2.274 +1	8.079 -1	7.847 -1	2.735 +5
8.10	221.3	8.764	2.600	3.385	2.389	8.088	7.857	2.604
8.20	221.6	8.770	2.481	3.225	2.511	8.097	7.867	2.480
8.30	221.9	8.776	2.367	3.073	2.638	8.107	7.877	2.362
8.40	222.3	8.782	2.258	2.928	2.772	8.116	7.887	2.250
8.50	222.6	8.788	2.155	2.790	2.912	8.125	7.897	2.143
8.60	222.9	8.794	2.057	2.659	3.059	8.134	7.907	2.041
8.70	223.2	8.800	1.963	2.534	3.213	8.144	7.918	1.945
8.80	223.5	8.806	1.873	2.416	3.375	8.153	7.928	1.853
8.90	223.8	8.812	1.788	2.303	3.545	8.162	7.938	1.765
9.00 +4	224.1	8.818 -1	1.707 -2	2.195 -2	3.723 +1	8.171 -1	7.948 -1	1.682 +5
9.10	224.4	8.824	1.630	2.093	3.909	8.181	7.958	1.603
9.20	224.7	8.830	1.556	1.995	4.105	8.190	7.968	1.527
9.30	225.0	8.836	1.485	1.902	4.310	8.199	7.978	1.456
9.40	225.3	8.842	1.418	1.814	4.526	8.208	7.988	1.387
9.50	225.6	8.848	1.354	1.730	4.751	8.217	7.998	1.322
9.60	225.9	8.854	1.293	1.649	4.988	8.227	8.008	1.261
9.70	226.2	8.860	1.235	1.573	5.236	8.236	8.018	1.202
9.80	226.5	8.866	1.179	1.500	5.496	8.245	8.028	1.146
9.90	226.8	8.872	1.126	1.431	5.768	8.254	8.038	1.092
1.00 +5	227.1	8.878 -1	1.076 -2	1.365 -2	6.054 +1	8.263 -1	8.048 -1	1.041 +5
1.01	227.4	8.884	1.028	1.302	6.353	8.272	8.058	9.929 +4
1.02	227.7	8.890	9.818 -3	1.242	6.667	8.282	8.068	9.469
1.03	228.0	8.896	9.379	1.185	6.996	8.291	8.078	9.030
1.04	228.3	8.902	8.961	1.131	7.340	8.300	8.088	8.612
1.05	228.7	8.908	8.562	1.079	7.702	8.309	8.099	8.213
1.06	229.5	8.925	8.181	1.027	8.115	8.335	8.127	7.809
1.07	230.4	8.941	7.819	9.780 -3	8.548	8.360	8.155	7.427
1.08	231.2	8.958	7.474	9.314	9.003	8.386	8.183	7.065
1.09	232.1	8.974	7.145	8.872	9.481	8.411	8.211	6.721
1.10 +5 1.11 1.12 1.13 1.14 1.15 1.16 1.17 1.18 1.19	232.9	8.991 -1	6.832 -3	8.452 -3	9.982 +1	8.437 -1	8.239 -1	6.396 +4
	233.8	9.007	6.534	8.054	1.051 +2	8.462	8.267	6.087
	234.6	9.024	6.250	7.676	1.106	8.488	8.295	5.795
	235.5	9.040	5.979	7.316	1.164	8.513	8.323	5.517
	236.3	9.057	5.721	6.975	1.224	8.538	8.350	5.254
	237.2	9.073	5.475	6.651	1.287	8.563	8.378	5.004
	238.0	9.089	5.240	6.343	1.354	8.588	8.406	4.767
	238.9	9.105	5.016	6.050	1.424	8.614	8.434	4.542
	239.8	9.122	4.803	5.772	1.497	8.639	8.462	4.328
	240.6	9.138	4.599	5.508	1.573	8.664	8.489	4.125

TABLE 8.1 Properties of a Standard Atmosphere: Geopotential Altitude in Feet (continued)

H (ft)	(K)	$a/a_0$	$p/p_0$	$\rho/\rho_0$	$v/v_0$	$\mu/\mu_0$	$k/k_0$	$Re^*/l$ (ft <sup>-1</sup> )
1.20 +5 1.21 1.22 1.23 1.24 1.25 1.26 1.27 1.28 1.29	241.5 242.3 243.2 244.0 244.9 245.7 246.6 247.4 248.3 249.1	9.154 -1 9.170 9.186 9.203 9.219 9.235 9.251 9.267 9.283 9.299	4.405 -3 4.219 4.042 3.873 3.711 3.557 3.410 3.269 3.134 3.006	5.256 -3 5.017 4.789 4.573 4.367 4.171 3.984 3.807 3.638 3.476	1.653 +2 1.737 1.825 1.916 2.012 2.113 2.218 2.328 2.443 2.564	8.689 -1 8.714 8.739 8.764 8.788 8.813 8.838 8.863 8.863 8.887 8.912	8.517 -1 8.545 8.573 8.600 8.628 8.656 8.683 8.711 8.738 8.766	3.932 +4 3.749 3.575 3.410 3.253 3.103 2.961 2.826 2.698 2.576
1.30 +5 1.31 1.32 1.33 1.34 1.35 1.36 1.37 1.38 1.39	250.0	9.314 -1	2.883 -3	3.323 -3	2.689 +2	8.937 -1	8.793 -1	2.459 +4
	250.9	9.330	2.766	3.177	2.821	8.961	8.821	2.349
	251.7	9.346	2.653	3.038	2.958	8.986	8.848	2.243
	252.6	9.362	2.546	2.905	3.102	9.010	8.876	2.143
	253.4	9.378	2.443	2.778	3.252	9.035	8.903	2.048
	254.3	9.394	2.345	2.658	3.409	9.059	8.931	1.957
	255.1	9.409	2.251	2.543	3.572	9.084	8.958	1.870
	256.0	9.425	2.161	2.433	3.744	9.108	8.985	1.788
	256.8	9.441	2.075	2.328	3.922	9.132	9.013	1.709
	257.7	9.456	1.993	2.229	4.109	9.157	9.040	1.634
1.40 +5	258.5	9.472 -1	1.914 -3	2.133 -3	4.303 +2	9.181 -1	9.067 -1	1.563 +4
1.41	259.4	9.488	1.839	2.043	4.506	9.205	9.094	1.495
1.42	260.2	9.503	1.766	1.956	4.719	9.229	9.122	1.430
1.43	261.1	9.519	1.697	1.873	4.940	9.253	9.149	1.368
1.44	261.9	9.534	1.631	1.794	5.171	9.277	9.176	1.309
1.45	262.8	9.550	1.568	1.719	5.412	9.301	9.203	1.253
1.46	263.7	9.565	1.507	1.647	5.663	9.325	9.230	1.199
1.47	264.5	9.581	1.449	1.578	5.925	9.349	9.257	1.148
1.48	265.4	9.596	1.393	1.512	6.198	9.373	9.285	1.099
1.49	266.2	9.612	1.339	1.450	6.483	9.397	9.312	1.053
1.50 +5 1.51 1.52 1.53 1.54 1.55 1.56 1.57 1.58 1.59	267.1 267.9 268.8 269.6 270.5 270.7 270.7 270.7 270.7 270.7	9.627 -1 9.643 9.658 9.673 9.689 9.692 9.692 9.692 9.692 9.692 9.692	1.288 -3 1.239 1.192 1.146 1.103 1.061 1.021 9.828 -4 9.457 9.100	1.390 -3 1.332 1.277 1.225 1.175 1.130 1.087 1.046 1.007 9.688 -4	6.780 +2 7.090 7.412 7.748 8.099 8.426 8.756 9.100 9.457 9.828	9.421 -1 9.445 9.469 9.493 9.516 9.521 9.521 9.521 9.521 9.521	9.339 -1 9.366 9.393 9.420 9.447 9.452 9.452 9.452 9.452 9.452	1.008 +4 9.658 +3 9.252 8.865 8.495 8.167 7.859 7.563 7.277 7.002
1.60 +5 1.61 1.62 1.63 1.64 1.65 1.66 1.67 1.68 1.69	270.7 270.7 270.7 270.7 270.7 270.7 270.7 270.7 270.7 270.1 269.2	9.692 -1 9.692 9.692 9.692 9.692 9.692 9.692 9.692 9.681 9.666	8.756 -4 8.426 8.108 7.802 7.507 7.224 6.951 6.436 6.192	9.322 -4 8.971 8.632 8.306 7.993 7.691 7.401 7.121 6.867 6.628	1.021 +3 1.061 1.103 1.146 1.191 1.238 1.287 1.337 1.384 1.431	9.521 -1 9.521 9.521 9.521 9.521 9.521 9.521 9.521 9.521 9.505 9.481	9.452 -1 9.452 9.452 9.452 9.452 9.452 9.452 9.452 9.434 9.407	6.738 +3 6.484 6.239 6.004 5.777 5.559 5.349 5.147 4.967 4.798
1.70 +5	268.4	9.651 -1	5.957 -4	6.396 -4	1.479 +3	9.457 -1	9.380 -1	4.635 +3
1.71	267.5	9.635	5.730	6.172	1.528	9.434	9.353	4.476
1.72	266.7	9.620	5.511	5.955	1.580	9.410	9.326	4.323
1.73	265.8	9.604	5.300	5.745	1.634	9.386	9.299	4.174
1.74	265.0	9.589	5.096	5.542	1.689	9.362	9.272	4.031
1.75	264.1	9.574	4.899	5.345	1.747	9.338	9.245	3.891
1.76	263.2	9.558	4.709	5.155	1.807	9.314	9.217	3.756
1.77	262.4	9.543	4.526	4.971	1.869	9.290	9.190	3.626
1.78	261.5	9.527	4.350	4.793	1.933	9.266	9.163	3.499
1.79	260.7	9.511	4.180	4.620	2.000	9.242	9.136	3.376
1.80 +5 1.81 1.82 1.83 1.84 1.85 1.86 1.87 1.88 1.89	259.8	9.496 -1	4.016 -4	4.454 -4	2.070 +3	9.218 -1	9.109 -1	3.258 +3
	259.0	9.480	3.858	4.293	2.142	9.193	9.081	3.143
	258.1	9.465	3.706	4.137	2.217	9.169	9.054	3.032
	257.3	9.449	3.559	3.986	2.294	9.145	9.027	2.925
	256.4	9.433	3.417	3.840	2.375	9.121	9.000	2.820
	255.6	9.418	3.281	3.700	2.459	9.096	8.972	2.720
	254.7	9.402	3.150	3.564	2.546	9.072	8.945	2.622
	253.9	9.386	3.024	3.432	2.636	9.048	8.917	2.528
	253.0	9.370	2.902	3.305	2.730	9.023	8.890	2.437
	252.1	9.354	2.785	3.182	2.828	8.999	8.863	2.349

TABLE 8.1 Properties of a Standard Atmosphere: Geopotential Altitude in Feet (concluded)

H (ft)	(K)	$a/a_0$	$p/p_0$	$\rho/\rho_0$	$v/v_0$	$\mu/\mu_0$	$k/k_0$	$Re^*/l$ (ft <sup>-1</sup> )
1.90 +5 1.91 1.92 1.93 1.94 1.95 1.96 1.97 1.98 1.99	251.3	9.339 -1	2.672 -4	3.064 -4	2.929 +3	8.974 -1	8.835 -1	2.264 +3
	250.4	9.323	2.563	2.949	3.035	8.949	8.808	2.181
	249.6	9.307	2.459	2.839	3.144	8.925	8.780	2.102
	248.7	9.291	2.358	2.732	3.258	8.900	8.753	2.025
	247.9	9.275	2.261	2.629	3.377	8.876	8.725	1.950
	247.0	9.259	2.168	2.529	3.500	8.851	8.698	1.879
	246.2	9.243	2.078	2.433	3.628	8.826	8.670	1.809
	245.3	9.227	1.992	2.340	3.761	8.801	8.642	1.742
	244.5	9.211	1.909	2.250	3.900	8.776	8.615	1.677
	243.6	9.195	1.829	2.164	4.044	8.752	8.587	1.614
2.00 +5 2.01 2.02 2.03 2.04 2.05 2.06 2.07 2.08 2.09	242.8	9.179 -1	1.753 -4	2.081 -4	4.195 +3	8.727 -1	8.559 -1	1.554 +3
	241.9	9.163	1.679	2.000	4.351	8.702	8.532	1.495
	241.1	9.146	1.608	1.922	4.513	8.677	8.504	1.439
	240.2	9.130	1.540	1.848	4.683	8.652	8.476	1.384
	239.3	9.114	1.475	1.775	4.859	8.627	8.448	1.332
	238.5	9.098	1.412	1.706	5.043	8.602	8.421	1.281
	237.6	9.081	1.351	1.639	5.234	8.576	8.393	1.232
	236.8	9.065	1.293	1.574	5.433	8.551	8.365	1.185
	235.9	9.049	1.238	1.511	5.641	8.526	8.337	1.139
	235.1	9.032	1.184	1.451	5.857	8.501	8.309	1.095
2.10 +5 2.11 2.12 2.13 2.14 2.15 2.16 2.17 2.18 2.19	234.2	9.016 -1	1.133 -4	1.393 -4	6.083 +3	8.475 -1	8.281 -1	1.053 +3
	233.4	8.999	1.083	1.338	6.317	8.450	8.253	1.012
	232.5	8.983	1.036	1.284	6.562	8.425	8.225	9.720 +2
	231.7	8.967	9.905 -5	1.232	6.818	8.399	8.197	9.339
	230.8	8.950	9.469	1.182	7.084	8.374	8.169	8.971
	230.0	8.933	9.050	1.184	7.361	8.348	8.141	8.617
	229.1	8.917	8.649	1.088	7.651	8.323	8.113	8.276
	228.3	8.900	8.264	1.043	7.953	8.297	8.085	7.946
	227.4	8.884	7.895	1.000	8.268	8.271	8.057	7.629
	226.5	8.867	7.541	9.591 -5	8.597	8.246	8.029	7.324
2.20 +5	225.7	8.850 -1	7.201 -5	9.194 -5	8.940 +3	8.220 -1	8.001 -1	7.029 +2
2.21	224.8	8.833	6.876	8.812	9.299	8.194	7.973	6.745
2.22	224.0	8.817	6.564	8.445	9.673	8.168	7.945	6.472
2.23	223.1	8.800	6.266	8.091	1.006 +4	8.143	7.916	6.209
2.24	222.3	8.783	5.979	7.751	1.047	8.117	7.888	5.956
2.25	221.4	8.766	5.705	7.424	1.090	8.091	7.860	5.712
2.26	220.6	8.749	5.443	7.110	1.134	8.065	7.832	5.477
2.27	219.7	8.732	5.191	6.808	1.181	8.039	7.803	5.251
2.28	218.9	8.715	4.950	6.518	1.229	8.013	7.775	5.034
2.29	218.0	8.698	4.720	6.238	1.280	7.986	7.747	4.825
2.30 +5 2.31 2.32 2.33 2.34 2.35 2.36 2.37 2.38 2.39	217.2	8.681 -1	4.499 -5	5.970 -5	1.333 +4	7.960 -1	7.718 -1	4.623 +2
	216.3	8.664	4.288	5.713	1.389	7.934	7.690	4.430
	215.5	8.647	4.086	5.465	1.447	7.908	7.662	4.243
	214.6	8.630	3.893	5.227	1.508	7.882	7.634	4.064
	214.0	8.618	3.709	4.994	1.575	7.863	7.613	3.886
	213.4	8.606	3.532	4.770	1.645	7.844	7.593	3.715
	212.8	8.593	3.364	4.555	1.718	7.826	7.573	3.552
	212.2	8.581	3.203	4.350	1.795	7.807	7.552	3.395
	211.6	8.569	3.049	4.153	1.875	7.788	7.532	3.245
	211.0	8.556	2.903	3.965	1.959	7.769	7.512	3.101
2.40 +5 2.41 2.42 2.43 2.44 2.45 2.46 2.47 2.48 2.49		8.544 -1 8.532 8.519 8.507 8.494 8.482 8.469 8.457 8.444 8.432	2.763 -5 2.629 2.501 2.380 2.264 2.153 2.047 1.947 1.851 1.759	3.784 -5 3.612 3.447 3.289 3.137 2.993 2.854 2.722 2.595 2.474	2.048 +4 2.140 2.237 2.339 2.446 2.558 2.675 2.798 2.927 3.063	7.750 -1 7.731 7.712 7.693 7.674 7.655 7.636 7.617 7.597 7.578	7.491 -1 7.471 7.450 7.430 7.410 7.389 7.369 7.348 7.328 7.307	2.963 +2 2.830 2.704 2.582 2.466 2.355 2.248 2.146 2.048 1.955
2.50 +5	204.2	8.419 -1	1.672 -5	2.358 -5	3.205 +4	7.559 -1	7.287 -1	1.865 +2

2.50 +5 | 204.2 | 8.419 -1 | 1.672 -5 | 2.358 -5 | 3.205 +4 | 7.559 -1 | 7.287 -1 | 1.865 +2 | In this table a one or two digit number (preceded by a plus or minus sign) following the initial entry of a block indicates the power of ten by which that and each succeeding entry in the block should be multiplied. A change of power occurring within a block is indicated by a similar notation.

TABLE 8.2 Properties of a Standard Atmosphere: Geopotential Altitude in Metres (-500 m to 80 000 m)

<i>H</i> (m)	(K)	$a/a_0$	$p/p_0$	$\rho/\rho_0$	$v/v_0$	$\mu/\mu_0$	$k/k_0$	$Re^*/l \ (m^{-1})$
-5.00 +2	291.4	1.006 0	1.061 0	1.049 0	9.617 -1	1.009 0	1.010 0	2.436 +7
0.00 0 5.00 +2 1.00 +3 1.50 2.00 2.50 3.00 3.50 4.00 4.50	288.2 284.9 281.7 278.4 275.2 271.9 268.7 265.4 262.2 258.9	1.000 0 9.943 -1 9.887 9.829 9.772 9.714 9.656 9.597 9.538 9.479	1.000 0 9.421 -1 8.870 8.345 7.846 7.371 6.919 6.490 6.083 5.697	1.000 0 9.529 -1 9.075 8.637 8.216 7.811 7.421 7.047 6.687 6.341	1.000 0 1.040 1.083 1.127 1.174 1.223 1.275 1.330 1.388 1.449	1.000 0 9.912 -1 9.824 9.735 9.645 9.556 9.465 9.374 9.283 9.191	1.000 0 9.899 -1 9.798 9.696 9.594 9.491 9.389 9.286 9.183 9.079	2.330 +7 2.227 2.128 2.032 1.939 1.850 1.764 1.681 1.601 1.523
5.00 +3 5.50 6.00 6.50 7.00 7.50 8.00 8.50 9.00 9.50	255.7 252.4 249.2 245.9 242.7 239.4 236.2 232.9 229.7 226.4	9.419 -1 9.359 9.299 9.238 9.177 9.115 9.053 8.990 8.927 8.864	5.331 -1 4.985 4.656 4.346 4.052 3.775 3.513 3.267 3.034 2.815	6.009 -1 5.691 5.385 5.093 4.812 4.544 4.287 4.042 3.807 3.583	1.514 0 1.583 1.655 1.732 1.813 1.899 1.990 2.087 2.190 2.300	9.099 -1 9.006 8.912 8.818 8.723 8.628 8.532 8.436 8.339 8.241	8.975 -1 8.871 8.766 8.661 8.556 8.450 8.344 8.238 8.131 8.024	1.449 +7 1.378 1.309 1.243 1.179 1.118 1.060 1.003 9.495 +6 8.977
1.00 +4 1.05 1.10 1.15 1.20 1.25 1.30 1.35 1.40 1.45	223.2 219.9 216.7 216.7 216.7 216.7 216.7 216.7 216.7 216.7	8.800 -1 8.736 8.671 8.671 8.671 8.671 8.671 8.671 8.671 8.671	2.609 -1 2.415 2.234 2.064 1.908 1.763 1.629 1.506 1.392 1.286	3.369 -1 3.165 2.971 2.746 2.537 2.345 2.167 2.003 1.851 1.711	2.417 0 2.542 2.674 2.894 3.131 3.388 3.666 3.967 4.292 4.644	8.143 -1 8.044 7.945 7.945 7.945 7.945 7.945 7.945 7.945 7.945	7.917 -1 7.809 7.701 7.701 7.701 7.701 7.701 7.701 7.701 7.701 7.701	8.482 +6 8.007 7.553 6.981 6.452 5.962 5.510 5.093 4.706 4.350
1.50 +4 1.55 1.60 1.65 1.70 1.75 1.80 1.85 1.90 1.95	216.7 216.7 216.7 216.7 216.7 216.7 216.7 216.7 216.7 216.7	8.671 -1 8.671 8.671 8.671 8.671 8.671 8.671 8.671 8.671 8.671	1.189 -1 1.099 1.015 9.383 -2 8.672 8.014 7.407 6.845 6.326 5.847	1.581 -1 1.461 1.350 1.248 1.153 1.066 9.851 -2 9.104 8.414 7.776	5.025 0 5.437 5.883 6.366 6.888 7.453 8.065 8.726 9.442 1.022 +1	7.945 -1 7.945 7.945 7.945 7.945 7.945 7.945 7.945 7.945 7.945 7.945	7.701 -1 7.701 7.701 7.701 7.701 7.701 7.701 7.701 7.701 7.701 7.701	4.020 +6 3.715 3.433 3.173 2.933 2.710 2.505 2.315 2.139 1.977
2.00 +4 2.05 2.10 2.15 2.20 2.25 2.30 2.35 2.40 2.45	216.7 217.2 217.7 218.2 218.7 219.2 219.7 220.2 220.7 221.2	8.671 -1 8.681 8.691 8.701 8.711 8.721 8.731 8.741 8.751 8.761	5.403 -2 4.994 4.617 4.269 3.947 3.651 3.378 3.125 2.892 2.677	7.187 -2 6.627 6.112 5.638 5.202 4.801 4.431 4.091 3.777 3.488	1.105 +1 1.201 1.305 1.417 1.539 1.671 1.814 1.968 2.136 2.317	7.945 -1 7.960 7.975 7.991 8.006 8.021 8.037 8.052 8.067 8.082	7.701 -1 7.718 7.735 7.751 7.768 7.784 7.801 7.818 7.834 7.851	1.827 +6 1.684 1.552 1.430 1.319 1.216 1.121 1.034 9.544 +5 8.808

TABLE 8.2 Properties of a Standard Atmosphere: Geopotential Altitude in Metres (continued)

Н	T	$a/a_0$	n/n	0/0	/	/	$k/k_0$	<i>Re</i> * / <i>l</i>
(m)	(K)	<i>a</i> / <i>a</i> <sub>0</sub>	$p/p_0$	$\rho/\rho_0$	$v/v_0$	$\mu/\mu_0$	κ/κ <sub>0</sub>	$(m^{-1})$
2.50 +4	221.7	8.771 -1	2.478 -2	3.222 -2	2.513 +1	8.097 -1	7.867 -1	8.129 +5
2.55	222.2	8.780	2.295	2.976	2.726	8.113	7.884	7.504
2.60	222.7	8.790	2.125	2.750	2.956	8.128	7.900	6.929
2.65	223.2	8.800	1.968	2.541	3.204	8.143	7.917	6.398
2.70	223.7	8.810	1.823	2.349	3.473	8.158	7.933	5.910
2.75	224.2	8.820	1.689	2.172	3.764	8.173	7.950	5.459
2.80	224.7	8.830	1.566	2.008	4.078	8.188	7.966	5.044
2.85	225.2	8.839	1.451	1.857	4.417	8.204	7.983	4.662
2.90	225.7	8.849	1.345	1.718	4.785	8.219	7.999	4.309
2.95	226.2	8.859	1.247	1.589	5.181	8.234	8.016	3.983
3.00 +4	226.7	8.869 -1	1.157 -2	1.470 -2	5.610 +1	8.249 -1	8.032 -1	3.683 +5
3.05	227.2	8.879	1.073	1.361	6.073	8.264	8.049	3.406
3.10	227.7	8.888	9.950 -3	1.259	6.573	8.279	8.065	3.150
3.15	228.2	8.898	9.232	1.166	7.113	8.294	8.082	2.914
3.20	228.7	8.908	8.567	1.080	7.696	8.309	8.098	2.696
3.25	230.1	8.935	7.952	9.960 -3	8.384	8.351	8.144	2.483
3.30	231.5	8.962	7.384	9.193	9.129	8.393	8.190	2.287
3.35	232.9	8.989	6.861	8.490	9.935	8.434	8.236	2.108
3.40	234.3	9.016	6.377	7.844	1.081 +2	8.476	8.282	1.944
3.45	235.7	9.043	5.930	7.251	1.175	8.518	8.328	1.793
3.50 +4 3.55 3.60 3.65 3.70 3.75 3.80 3.85 3.90 3.95	237.1	9.070 -1	5.516 -3	6.705 -3	1.276 +2	8.559 -1	8.373 -1	1.655 +5
	238.5	9.097	5.134	6.204	1.386	8.600	8.419	1.529
	239.9	9.123	4.780	5.742	1.505	8.641	8.465	1.412
	241.3	9.150	4.452	5.318	1.633	8.682	8.510	1.306
	242.7	9.177	4.149	4.927	1.771	8.723	8.556	1.207
	244.1	9.203	3.867	4.566	1.919	8.764	8.601	1.117
	245.5	9.229	3.607	4.234	2.079	8.805	8.646	1.034
	246.9	9.256	3.365	3.928	2.252	8.846	8.692	9.575 +4
	248.3	9.282	3.141	3.645	2.438	8.886	8.737	8.870
	249.7	9.308	2.932	3.385	2.637	8.927	8.782	8.222
4.00 +4	251.1	9.334 -1	2.739 -3	3.144 -3	2.852 +2	8.967 -1	8.827 -1	7.623 +4
4.05	252.5	9.360	2.559	2.921	3.083	9.007	8.872	7.072
4.10	253.9	9.386	2.392	2.715	3.332	9.047	8.917	6.563
4.15	255.3	9.412	2.237	2.525	3.598	9.087	8.962	6.093
4.20	256.7	9.438	2.093	2.349	3.885	9.127	9.007	5.659
4.25	258.1	9.463	1.958	2.187	4.192	9.167	9.052	5.259
4.30	259.5	9.489	1.833	2.036	4.522	9.207	9.096	4.888
4.35	260.9	9.515	1.717	1.896	4.876	9.246	9.141	4.546
4.40	262.3	9.540	1.608	1.767	5.256	9.286	9.186	4.229
4.45	263.7	9.565	1.507	1.647	5.662	9.325	9.230	3.936
4.50 +4 4.55 4.60 4.65 4.70 4.75 4.80 4.85 4.90 4.95	265.1 266.5 267.9 269.3 270.7 270.7 270.7 270.7 270.7 270.7	9.591 -1 9.616 9.641 9.666 9.692 9.692 9.692 9.692 9.692 9.692 9.692	1.413 -3 1.325 1.243 1.166 1.095 1.028 9.648 -4 9.058 8.504 7.983	1.536 -3 1.433 1.337 1.248 1.165 1.094 1.027 9.643 -4 9.053 8.500	6.098 +2 6.564 7.064 7.598 8.170 8.702 9.269 9.873 1.052 +3 1.120	9.365 -1 9.404 9.443 9.482 9.521 9.521 9.521 9.521 9.521 9.521	9.275 -1 9.319 9.364 9.408 9.452 9.452 9.452 9.452 9.452 9.452 9.452	3.664 +4 3.413 3.180 2.964 2.763 2.594 2.436 2.287 2.147 2.016

TABLE 8.2 Properties of a Standard Atmosphere: Geopotential Altitude in Metres (concluded)

<i>Н</i> (m)	T (K)	$a/a_0$	$p/p_0$	$\rho/\rho_0$	$v/v_0$	$\mu/\mu_0$	$k/k_0$	$Re^*/l \ (m^{-1})$
5.00 +4	270.7	9.692 -1	7.495 -4	7.980 -4	1.193 +3	9.521 -1	9.452 -1	1.892 +4
5.05	270.7	9.692	7.037	7.492	1.271	9.521	9.452	1.777
5.10	270.7	9.692	6.606	7.033	1.354	9.521	9.452	1.668
5.15	269.3	9.666	6.201	6.637	1.429	9.482	9.408	1.576
5.20	267.9	9.641	5.819	6.260	1.508	9.443	9.364	1.489
5.25	266.5	9.616	5.459	5.903	1.593	9.404	9.319	1.406
5.30	265.1	9.591	5.119	5.565	1.683	9.365	9.275	1.328
5.35	263.7	9.565	4.799	5.244	1.778	9.325	9.230	1.253
5.40	262.3	9.540	4.497	4.941	1.879	9.286	9.186	1.183
5.45	260.9	9.515	4.212	4.653	1.987	9.246	9.141	1.115
5.50 +4	259.5	9.489 -1	3.945 -4	4.381 -4	2.101 +3	9.207 -1	9.096 -1	1.052 +4
5.55	258.1	9.463	3.693	4.123	2.223	9.167	9.052	9.916 +3
5.60	256.7	9.438	3.456	3.880	2.353	9.127	9.007	9.346
5.65	255.3	9.412	3.232	3.649	2.490	9.087	8.962	8.805
5.70	253.9	9.386	3.023	3.431	2.637	9.047	8.917	8.292
5.75	252.5	9.360	2.825	3.225	2.793	9.007	8.872	7.807
5.80	251.1	9.334	2.640	3.030	2.959	8.967	8.827	7.348
5.85	249.7	9.308	2.466	2.846	3.136	8.927	8.782	6.914
5.90	248.3	9.282	2.302	2.672	3.325	8.886	8.737	6.503
5.95	246.9	9.256	2.149	2.508	3.526	8.846	8.692	6.115
6.00 +4	245.5	9.229 -1	2.005 -4	2.354 -4	3.741 +3	8.805 -1	8.646 -1	5.747 +3
6.05	244.1	9.203	1.870	2.208	3.970	8.764	8.601	5.400
6.10	242.7	9.177	1.743	2.070	4.215	8.723	8.556	5.072
6.15	241.3	9.150	1.624	1.940	4.476	8.682	8.510	4.763
6.20	239.9	9.123	1.513	1.817	4.755	8.641	8.465	4.470
6.25	238.5	9.097	1.409	1.702	5.053	8.600	8.419	4.194
6.30	237.1	9.070	1.311	1.593	5.371	8.559	8.373	3.934
6.35	235.7	9.043	1.219	1.491	5.712	8.518	8.328	3.688
6.40	234.3	9.016	1.134	1.395	6.076	8.476	8.282	3.457
6.45	232.9	8.989	1.054	1.304	6.467	8.434	8.282	3.238
6.50 +4	231.5	8.962 -1	9.792 -5	1.219 -4	6.884 +3	8.393 -1	8.190 -1	3.033 +3
6.55	230.1	8.935	9.094	1.139	7.332	8.351	8.144	2.839
6.60	228.7	8.908	8.441	1.064	7.811	8.309	8.098	2.657
6.65	227.3	8.881	7.831	9.930 -5	8.325	8.267	8.052	2.485
6.70	225.9	8.853	7.263	9.266	8.876	8.225	8.006	2.324
6.75	224.5	8.826	6.732	8.643	9.467	8.182	7.960	2.172
6.80	223.1	8.798	6.237	8.058	1.010 +4	8.140	7.914	2.029
6.85	221.7	8.771	5.776	7.509	1.078	8.097	7.867	1.895
6.90	220.3	8.743	5.346	6.994	1.152	8.055	7.821	1.769
6.95	218.9	8.715	4.946	6.512	1.230	8.012	7.775	1.650
7.00 +4	217.5	8.687 -1	4.574 -5	6.061 -5	1.315 +4	7.969 -1	7.728 -1	1.539 +3
7.05	216.1	8.659	4.227	5.638	1.406	7.926	7.681	1.435
7.10	214.7	8.631	3.905	5.242	1.504	7.883	7.635	1.337
7.15	213.7	8.611	3.605	4.862	1.615	7.852	7.601	1.242
7.20	212.7	8.591	3.328	4.509	1.735	7.821	7.568	1.154
7.25	211.7	8.570	3.070	4.180	1.864	7.790	7.535	1.071
7.30	210.7	8.550	2.832	3.873	2.003	7.759	7.501	9.943 +2
7.35	209.7	8.530	2.611	3.588	2.154	7.728	7.468	9.226
7.40	208.7	8.509	2.406	3.323	2.317	7.697	7.434	8.557
7.45	207.7	8.489	2.216	3.075	2.493	7.666	7.401	7.934
7.50 +4 7.55 7.60 7.65 7.70 7.75 7.80 7.85 7.90 7.95	206.7 205.7 204.7 203.7 202.7 201.7 200.7 199.7 197.7	8.469 -1 8.448 8.427 8.407 8.386 8.365 8.345 8.324 8.303 8.282	2.041 -5 1.879 1.728 1.590 1.462 1.343 1.234 1.133 1.040 9.538 -6	2.846 -5 2.632 2.434 2.249 2.078 1.919 1.772 1.635 1.508 1.391	2.683 +4 2.888 3.111 3.352 3.613 3.896 4.202 4.534 4.894 5.285	7.634 -1 7.603 7.572 7.540 7.509 7.477 7.445 7.413 7.382 7.350	7.367 -1 7.334 7.300 7.266 7.233 7.199 7.165 7.132 7.098 7.064	7.354 +2 6.813 6.311 5.843 5.407 5.002 4.626 4.277 3.952 3.650
8.00 +4	196.7	8.261 -1	8.747 -6	1.282 -5	5.710 +4	7.318 -1	7.030 -1	3.371 +2

TABLE 8.3 Properties of a Standard Atmosphere: Geopotential Altitude in Metres (80 000 m to 1 000 000 m)

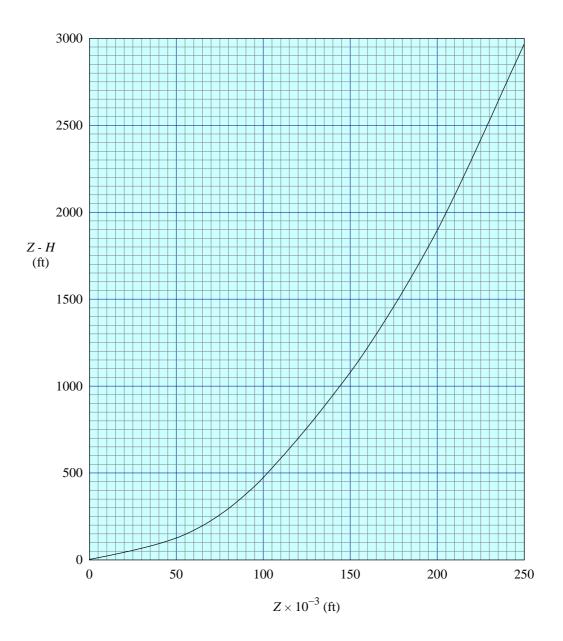
Z	T	(m:m)	,	,	$\overline{V}$	L	m
(m)	(K)	$\sqrt{(T/T_0)}$	$p/p_0$	$\rho/\rho_0$	(m/s)	(m)	(kg/kmol)
8.00 +4	198.6	8.303 -1	1.039 -5	1.507 -5	3.811 +2	4.40 -3	28.96
8.50	188.8	8.095	4.399 -6	6.710 -6	3.716	9.88	28.95
9.00	186.9	8.053	1.812	2.789	3.699	2.37 -2	28.91
9.50	188.4	8.086	7.497 -7	1.137	3.726	5.79	28.73
1.00 +5	195.1	8.228 -1	3.159 -7	4.575 -7	3.814 +2	1.42 -1	28.40
1.05	208.8	8.513	1.429	1.898	3.982	3.36	27.88
1.10	240.0	9.126	7.011 -8	7.925 -8	4.317	7.88	27.27
1.15	300.0	1.020 0	3.957	3.501	4.879	1.75 +0	26.68
1.20	360.0	1.118	2.505	1.814	5.393	3.31	26.20
1.25	417.2	1.203	1.713	1.054	5.851	5.6	25.80
1.30	469.3	1.276	1.234	6.655 -9	6.250	8.8	25.44
1.35	516.6	1.339	9.235 -9	4.461	6.603	1.3 +1	25.09
1.40	559.6	1.394	7.109	3.128	6.919	1.8	24.75
1.45	598.8	1.442	5.595	2.270	7.205	2.5	24.42
1.50 +5	634.4	1.484 0	4.483 -9	1.694 -9	7.465 +2	3.3 +1	24.10
1.55	666.8	1.521	3.645	1.294	7.703	4.2	23.79
1.60	696.3	1.554	3.000	1.007	7.922	5.3	23.49
1.65	723.1	1.584	2.495	7.959 -10	8.125	6.7	23.19
1.70	747.6	1.611	2.093	6.380	8.313	8.2	22.90
1.75	769.8	1.634	1.770	5.174	8.489	1.0 +2	22.62
1.80	790.1	1.656	1.507	4.240	8.653	1.2	22.34
1.85	808.5	1.675	1.291	3.506	8.806	1.4	22.07
1.90	825.3	1.692	1.112	2.923	8.951	1.7	21.81
1.95	840.6	1.708	9.622 -10	2.454	9.087	2.0	21.55
2.00 +5	854.6	1.722 0	8.363 -10	2.074 -10	9.216 +2	2.4 +2	21.30
2.05	867.3	1.735	7.298	1.763	9.337	2.7	21.06
2.10	878.8	1.746	6.391	1.507	9.452	3.2	20.83
2.15	889.4	1.757	5.616	1.294	9.562	3.6	20.60
2.20	899.0	1.766	4.950	1.116	9.665	4.2	20.37
2.25	907.8	1.775	4.374	9.665 -11	9.764	4.8	20.16
2.30	915.8	1.783	3.876	8.402	9.858	5.4	19.95
2.35	923.1	1.790	3.443	7.329	9.947	6.2	19.75
2.40	929.7	1.796	3.065	6.415	1.003 +3	7.0	19.56
2.45	935.8	1.802	2.735	5.631	1.011	7.9	19.37
2.50 +5	941.3	1.807 0	2.444 -10	4.957 -11	1.019 +3	8.9 +2	19.19
2.55	946.4	1.812	2.189	4.375	1.027	1.0 +3	19.02
2.60	951.0	1.817	1.963	3.871	1.034	1.1	18.85
2.65	955.2	1.821	1.764	3.433	1.040	1.2	18.69
2.70	959.0	1.824	1.587	3.052	1.047	1.4	18.53
2.75	962.5	1.828	1.430	2.718	1.053	1.5	18.38
2.80	965.8	1.831	1.291	2.425	1.059	1.7	18.24
2.85	968.7	1.833	1.166	2.168	1.064	1.9	18.11
2.90	971.3	1.836	1.055	1.941	1.070	2.1	17.97
2.95	973.8	1.838	9.549 -11	1.741	1.075	2.3	17.85
3.00 +5	976.0	1.840 0	8.656 -11	1.564 -11	1.080 +3	2.6 +3	17.73
3.10	979.9	1.844	7.134	1.267	1.089	3.2	17.50
3.20	983.2	1.847	5.901	1.032	1.097	3.8	17.29
3.30	985.9	1.850	4.898	8.447 -12	1.105	4.6	17.09
3.40	988.2	1.852	4.078	6.941	1.112	5.6	16.91
3.50	990.1	1.854	3.405	5.725	1.119	6.7	16.74
3.60	991.7	1.855	2.850	4.739	1.126	8.0	16.57
3.70	993.0	1.856	2.392	3.934	1.132	9.6	16.42
3.80	994.1	1.857	2.012	3.276	1.137	1.1 +4	16.27
3.90	995.0	1.858	1.696	2.734	1.143	1.4	16.13
4.00 +5 4.10 4.20 4.30 4.40 4.50 4.60 4.70 4.80 4.90	995.8 996.5 997.0 997.5 997.9 998.2 998.5 998.7 998.9	1.859 0 1.860 1.860 1.861 1.861 1.861 1.862 1.862 1.862 1.862 1.862	1.433 -11 1.213 1.029 8.749 -12 7.453 6.363 5.443 4.667 4.011 3.455	2.288 -12 1.918 1.612 1.357 1.144 9.668 -13 8.180 6.932 5.884 5.001	1.419 +3 1.154 1.160 1.165 1.171 1.177 1.184 1.191 1.198 1.206	1.6 +4 1.9 2.2 2.6 3.1 3.6 4.2 4.9 5.7 6.7	15.98 15.84 15.70 15.55 15.40 15.25 15.08 14.91 14.73 14.54

In this table a one or two digit number (preceded by a plus or minus sign) following the initial entry of a block indicates the power of ten by which that and each succeeding entry in the block should be multiplied. A change of power occurring within a block is indicated by a similar notation.

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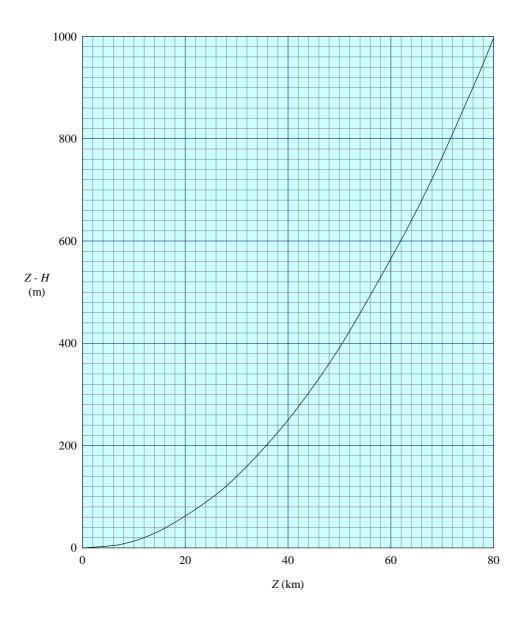
TABLE 8.3 Properties of a Standard Atmosphere: Geopotential Altitude in Metres (concluded)

Z (m)	T (K)	$\sqrt{(T/T_0)}$	$p/p_0$	$\rho/\rho_0$	\( \overline{V} \) (m/s)	<i>L</i> (m)	m (kg/kmol)
5.00 +5 5.10 5.20 5.30 5.40 5.50 5.60 5.70 5.80 5.90	999.2 999.4 999.5 999.5 999.6 999.7 999.7 999.8 999.8	1.862 0 1.862 1.862 1.862 1.863 1.863 1.863 1.863 1.863 1.863	2.984 -12 2.584 2.243 1.953 1.706 1.494 1.313 1.158 1.024 9.095 -13	4.257 -13 3.629 3.099 2.650 2.269 1.946 1.672 1.439 1.241 1.072	1.215 +3 1.225 1.235 1.246 1.258 1.272 1.286 1.302 1.319 1.337	7.7 +4 8.9 1.0 +5 1.2 1.3 1.5 1.8 2.0 2.2 2.5	14.33 14.11 13.88 13.63 13.37 13.09 12.80 12.49 12.18 11.85
6.00 +5 6.10 6.20 6.30 6.40 6.50 6.60 6.70 6.80 6.90	999.9 999.9 999.9 999.9 999.9 999.9 1000 1000	1.863 0 1.863 1.863 1.863 1.863 1.863 1.863 1.863 1.863	8.106 -13 7.251 6.511 5.868 5.310 4.823 4.397 4.023 3.695 3.405	9.279 -14 8.048 6.996 6.096 5.325 4.663 4.094 3.605 3.183 2.820	1.356 +3 1.378 1.400 1.424 1.449 1.476 1.504 1.533 1.564 1.595	2.8 +5 3.2 3.5 3.9 4.3 4.8 5.2 5.7 6.2 6.8	11.51 11.16 10.80 10.44 10.08 9.72 9.36 9.01 8.66 8.32
7.00 +5 7.10 7.20 7.30 7.40 7.50 7.60 7.70 7.80 7.90	1000 1000 1000 1000 1000 1000 1000 100	1.863 0 1.863 1.863 1.863 1.863 1.863 1.863 1.863 1.863	3.149 -13 2.922 2.719 2.538 2.376 2.230 2.099 1.979 1.871 1.772	2.506 -14 2.234 1.998 1.793 1.615 1.460 1.324 1.205 1.100 1.008	1.627 +3 1.660 1.693 1.727 1.760 1.794 1.827 1.860 1.892 1.924	7.3 +5 7.9 8.5 9.1 9.7 1.0 +6 1.1 1.2 1.2	8.00 7.69 7.39 7.10 6.83 6.58 6.34 6.12 5.91 5.72
8.00 +5 8.10 8.20 8.30 8.40 8.50 8.60 8.70 8.80 8.90	1000 1000 1000 1000 1000 1000 1000 100	1.863 0 1.863 1.863 1.863 1.863 1.863 1.863 1.863 1.863	1.681 -13 1.598 1.522 1.451 1.385 1.324 1.267 1.214 1.164 1.117	9.272 -15 8.554 7.916 7.348 6.841 6.387 5.978 5.611 5.278 4.976	1.954 +3 1.984 2.012 2.039 2.065 2.090 2.113 2.135 2.155 2.175	1.4 +6 1.4 1.5 1.6 1.7 1.7 1.8 1.9 2.0 2.1	5.54 5.38 5.23 5.09 4.96 4.85 4.74 4.65 4.56 4.48
9.00 +5 9.10 9.20 9.30 9.40 9.50 9.60 9.70 9.80 9.90	1000 1000 1000 1000 1000 1000 1000 100	1.863 0 1.863 1.863 1.863 1.863 1.863 1.863 1.863 1.863 1.863	1.073 -13 1.032 9.923 -14 9.551 9.199 8.864 8.546 8.243 7.954 7.679	4.701 -15 4.451 4.221 4.010 3.815 3.635 3.468 3.313 3.168 3.033	2.193 +3 2.210 2.225 2.240 2.254 2.266 2.278 2.289 2.300 2.309	2.1 +6 2.2 2.3 2.4 2.5 2.6 2.7 2.8 2.9 3.0	4.40 4.34 4.28 4.22 4.17 4.12 4.08 4.04 4.00 3.97



### FIGURE 1 DIFFERENCE BETWEEN GEOMETRIC AND GEOPOTENTIAL ALTITUDES

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### FIGURE 2 DIFFERENCE BETWEEN GEOMETRIC AND GEOPOTENTIAL ALTITUDES

### APPENDIX A EQUATIONS DEFINING STANDARD ATMOSPHERE FOR ALTITUDES UP TO 80 km

#### A1. ADDITIONAL NOTATION AND UNITS

		SI	British
$H_i$	dividing altitude in temperature-altitude profile, $i = 0, 1, 2, \dots 7$	m	ft
$L_i$	temperature gradient between altitudes $H_i$ and $H_{i+1}$	K/m	K/ft
$p_i$	pressure at altitude $H_i$	$N/m^2$	lbf/ft <sup>2</sup>
$T_i$	temperature at altitude $H_i$	K	K

#### A2. EQUATIONS FOR STANDARD ATMOSPHERE

The atmosphere defined in Derivations 1 and 2 is divided into seven layers for altitudes up to H = 80 km. Within each of these the temperature varies linearly with altitude and is expressed as

$$T = T_i + L_i(H - H_i), \tag{A2.1}$$

for  $H_i \leq H < H_{i+1}$ , where i=0,1,2,...7. The temperature at altitude  $H_i$  is  $T_i$ , and  $L_i$  is the temperature gradient between  $H_i$  and  $H_{i+1}$ . For altitudes up to 80 km the constituent gases of the air retain their sea-level proportions and the mean molecular mass is constant at  $m_0$ . Using the hydrostatic equation, the perfect gas law and Equation (A2.1), two analytic expressions for pressure as a function of altitude are obtained for  $H_i \leq H \leq H_{i+1}$ . If  $L_i \neq 0$ ,

$$\frac{p}{p_i} = \left(\frac{T_i}{T}\right)^{(g_0 m_0 / \mathcal{R}L_i)},\tag{A2.2}$$

and if  $L_i = 0$ , then

$$\frac{p}{p_i} = \exp\left(-\frac{g_0 m_0 (H - H_i)}{\Re T_i}\right),\tag{A2.3}$$

where  $p_i$  is the pressure at altitude  $H_i$ .

Tables A2.1 (SI units) and A2.2 (British units) give the values of the constants required for solving Equations (A2.1) to (A2.3). If H < 0, then Equations (A2.1) and (A2.2) apply directly with i = 0.

Knowing T and p, other atmospheric data are calculated through the equations given in Table A2.3.

**TABLE A2.1 Constants for Basic Equations (SI Units)** 

i	<i>H<sub>i</sub></i> (m)	T <sub>i</sub> (K)	L <sub>i</sub> (K/m)	<i>p<sub>i</sub></i> (N/m <sup>2</sup> )	$g_0 m_0 / \mathcal{R} L_i$	$g_0 m_0 / \mathcal{R} T_i $ $(\mathbf{m}^{-1})$
0	0	288.15	$-6.5 \times 10^{-3}$	$1.013250 \times 10^5$	-5.255880	_
1	$11 \times 10^3$	216.65	0	$2.263204 \times 10^4$	-	$1.576885 \times 10^{-4}$
2	$20 \times 10^3$	216.65	$1.0 \times 10^{-3}$	$5.474879 \times 10^3$	34.16322	_
3	$32 \times 10^3$	228.65	$2.8\times10^{-3}$	$8.680160 \times 10^2$	12.20115	_
4	$47 \times 10^3$	270.65	0	$1.109058 \times 10^2$	-	$1.262266 \times 10^{-4}$
5	51 × 10 <sup>3</sup>	270.65	$-2.8 \times 10^{-3}$	6.693853 × 10	-12.20115	-
6	$71 \times 10^3$	214.65	$-2.0 \times 10^{-3}$	3.956392	-17.08161	-
7	$80 \times 10^3$	196.65	_	$8.862722 \times 10^{-1}$	_	_

**TABLE A2.2 Constants for Basic Equations (British Units)** 

i	H <sub>i</sub> (ft)	T <sub>i</sub> (K)	$L_i$ (K/ft)	$p_i$ (lbf/ft <sup>2</sup> )	$g_0 m_0 / \mathcal{R} L_i$	$g_0 m_0 / \mathcal{R} T_i $ $(\mathrm{ft}^{-1})$
0	0	288.15	$-1.9812 \times 10^{-3}$	$2.11622 \times 10^3$	-5.255880	_
1	$3.608924 \times 10^4$	216.65	0	$4.726807 \times 10^2$	-	$4.806346 \times 10^{-5}$
2	$6.561680 \times 10^4$	216.65	$3.048 \times 10^{-4}$	$1.143454 \times 10^2$	34.16322	_
3	$1.049869 \times 10^5$	228.65	$8.5344 \times 10^{-4}$	$1.812892 \times 10$	12.20115	_
4	$1.541995 \times 10^5$	270.65	0	2.316319	-	$3.847386 \times 10^{-5}$
5	$1.673228 \times 10^5$	270.65	$-8.5344 \times 10^{-4}$	1.398046	-12.20115	-
6	$2.329396 \times 10^5$	214.65	$-6.096 \times 10^{-4}$	$8.263130 \times 10^{-2}$	-17.08161	_
7	$2.624672 \times 10^5$	196.65	_	$1.851025 \times 10^{-2}$	_	_

Minor inconsistencies exist between Tables A2.1 and A2.2 due to unit conversions. Table A2.1 gives the definitive values for the atmosphere, and any inconsistencies are minimised by using the composite parameters  $g_0 m_0 / \mathcal{R} T_i$  and  $g_0 m_0 / \mathcal{R} L_i$ , rather than working with values of the individual parameters  $g_0$ ,  $\mathcal{R}$  and  $L_i$  in British units. For all practical purposes the values in the tables lead to equivalent atmospheric properties.

**TABLE A2.3 Formulae for Atmospheric Data** 

Quality	SI		British		
a	$\left(\frac{1.4\mathscr{R}T}{m_0}\right)^{1/2}$	m/s	m/s $ \left(\frac{1.4  \mathscr{R}  T}{m_0}\right)^{1/2} $		
ρ	$\frac{pm_0}{\mathcal{R}T}$	kg/m <sup>3</sup>	$\frac{pm_0}{\mathcal{R}T}$	slug/ft <sup>3</sup>	
μ	$\frac{1.458 \times 10^{-6} T^{3/2}}{T + 110.4}$	N s/m <sup>2</sup>	$\frac{3.04509 \times 10^{-8} T^{3/2}}{T + 110.4}$	lbf s/ft <sup>2</sup>	
v	<u>μ</u> ρ	m <sup>2</sup> /s	<u>μ</u> ρ	ft <sup>2</sup> /s	
$k^*$	$\frac{2.648151 \times 10^{-3} T^{3/2}}{T + 245.4 \times 10^{(-12/T)}}$	W/m K	$\frac{5.953280 \times 10^{-4} T^{3/2}}{T + 245.4 \times 10^{(-12/T)}}$	ft lbf/ft s K	

<sup>\*</sup> Following Derivation 1 which converts the equation for *k* given in Reference 7 from kg-cal/m s K units to W/m K units by using the relationship 1 cal (International Steam Table) = 4.1868 J, rather than Derivation 2 which quotes a conversion 1 cal (15°C) = 4.1858 J.

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### Properties of a standard atmosphere. ESDU 77021

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