

AEROTHON 2025

UNCREWED AIRCRAFT SYSTEM (UAS) DESIGN, BUILD AND FLY CONTEST



PHASE 1: DESIGN REPORT

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Introduction

1.1 Overview

In the face of natural and man-made disasters, rapid response and situational awareness are critical. Drones have emerged as powerful tools in disaster management, offering real-time aerial insights, access to hard-to-reach areas, and faster deployment compared to traditional methods. Whether locating survivors, assessing damage, or delivering essential supplies, drone technology enhances the efficiency and safety of relief operations. As disasters grow more complex and unpredictable, integrating drones into emergency response systems is no longer a luxury—it's a necessity.

1.2 Problem Statement and Mission Requirements

This year's AEROTHON is themed on *Surveillance and Disaster Management*. The problem statement is to build an *Uncrewed Aircraft System (UAS)* to be able to perform the mission requirements as per the rulebook. The mission requirements at a glance are as follows:

Mission - 1: Advanced Obstacle Navigation & Fragile Payload Delivery with Precision Placement – Manual Operation

Mission - 2: Autonomous Object Classification, Disaster Situation Identification & Payload Drop – Autonomous Operation

1.3 Scope of Report

The scope of this report is to provide a comprehensive understanding of the design rationale we have used while building this project. We have tried to provide the relevant calculations, figures, and analysis models to justify the materials/design/framework we've chosen to work with for our structural and system architectures.

Apart from that, this report is intended to also serve as an accessible guide catering to neophytes in UAV/UAS systems. We have tried our best to aim at providing clear context and insight that sort of demystifies drone development.

System Requirements & Design Objectives

2.1 Mission Profile

1. **Mission 1:** Advanced Obstacle Navigation & Fragile Payload Delivery with Precision Placement

This is a *Manual Operation*. In this mission, the drone must transport a fragile payload through a challenging course filled with static obstacles such as walls, barriers, and narrow passages. The primary objective is to navigate these obstacles with high precision while ensuring the payload remains undamaged.

Upon reaching the target zone, the drone must land carefully and place the fragile payload on the ground without causing any damage. After the successful placement, the drone must then return to the takeoff point or designated home base, ensuring safe and efficient navigation back through the course. The mission is complete once the payload is placed securely, and the drone successfully returns to the home base.

2. **Mission 2:** Autonomous Object Classification, Disaster Situation Identification & Payload Drop

This is an *Autonomous Operation*. In this mission, the drone will autonomously scan, classify, and assess objects within a predefined area using onboard sensors and algorithms. The objects will vary in shape, size, color, and structure, and may be partially obscured, presenting challenges for detection and classification. Once the objects are classified, the drone will identify potential disaster scenarios, such as flooding, fire, or damaged infrastructure, within the same area.

2.2 Key Performance Indicators

According to the above defined mission profiles, we have a few KPIs (*Key Performance Index*) to keep in mind.

- 1. Flight Endurance and Range
- 2. Payload Handling
- 3. Autonomous Capabilites
- 4. System Reliability
- 5. Design and Innovation

2.3 Constraints

The design and development of the UAV is subjected to several constraints as per the guidelines mentioned in the rulebook AEROTHON 2025. These include dimensional constraints, payload restrictions and strict autonomy requirements. The drone must perform all missions bound by these constraints and we have taken great time and care to articulate them down to ensure nothing is amiss.

1. Dimensional Constraints

Maximum Wingspan: 1.5 metres - the UAV must fit inside a 1.5m x 1.5m x 1.5m bounding box in assembled condition.

Maximum Takeoff Weight: 3.5kg including battery and payload.

2. Payload Constraints

- Payload: One fragile payload cube of 10cm x 10cm x 10cm weighing 150 200g.
- Payload must be released within a 3m x 3m target zone.

3. Flight Environment Constraints

- Missions are conducted in open outdoor airspace.
- Expect wind speeds upto 5m/s

4. Autonomy and Mission Constraints

- Mission 1: Manual flight only (no GPS or autopilot usage).
- Mission 2: Fully autonomous flight (no pilot intervention or RC use).
- All autonomous missions must avoid obstacles and make decisions based on onboard computation.

5. Power and Communication Constraints

- Must operate on battery only
- · No cellular or internet-based comms allowed
- Only 2.4 GHz or 5.8 GHz RF modules permitted

6. Safety and Compliance

- Must have a failsafe mode (e.g., return-to-home or emergency land)
- Must pass technical inspection before flying
- Compliance with DGCA drone guidelines (if relevant in test zones)

7. Operational Constraints

- The team must complete the flight within a 15-minute slot.
- Payload must be dropped in an area of 3m x 3m.

Conceptual Design Approach

- 3.1 Design Methodology and Material Selection
- 3.2 Trade-off Analysis
- 3.3 Schematic Diagram and CAD Model

Detailed Design Breakdown

4.1 Aerodynamics & Flight Performance

- 4.1.1 Lift, Drag and Stability Considerations
- 4.1.2 Theoretical Performance Predictions
- 4.2 Propulsion System
- 4.2.1 Motor, ESC, Propellor Selection & Battery

Motor: DYS D2836-7 1120KV Brushless Motor

The DYS D2836-7 1120KV Brushless Motor is our go-to motor for this project because of the following leverages it offers:

• KV Rating: 1120KV - KV generally means RPM per volt. In layman terms, in one volt, how many rotations does it make per minute = KV. In this case, 1120KV is mid-range, which means good thrust at moderate RPMs, and it works decently with larger propellors (9" - 11") which improves lift and efficiency, especially at low speeds. This is perfect for surveillance drones that require loitering and stability. A lower KV would force us to use bulky propellors, and a higher KV would drain the battery faster. 1120KV is a sweet spot between the two.



Figure 4.1: DYS D2836-7 1120kV BLDC

- Power & Efficiency: With a 3S or 4S LiPo, this motor produces 800g to 1100g of thrust, depending on the propeller used. It can pull 20–25A max, so it's efficient for mid-weight UAVs (in our case, it is around 1.5 2kg AUW (All Up Weight.)), so it's ideal for our choice.
- Build Quality & Availability DYS motors are known for durability and consistent performance, and since they are so weidely available, it's easier to find cheap spare parts, like mounts/ shafts/ bearings.

Detailed Specifications Table:

Motor KV (RPM/V):	1120
Motor Type:	Brushless Motor
Compatible LiPO Batteries:	2S to 4S
Weight (g):	70
Shaft Diameter (mm):	Φ4.0×49mm
Max. Power (W):	336
Maximum Thrust (gm):	1130
Compatible Prop (inch):	11×7 / 7×3
Required ESC (A):	40
Shipping Weight:	0.089 kg
Shipping Dimensions:	10 × 6 × 5 cm

ESC: ReadytoSky 40A 2-4S ESC

The ReadytoSky 40A 2-4S ESC is a good choice for surveillance drones, and our use case for the following reasons:

 Sufficient Current Handling - Our DYS D2836 motor draws about 20 – 25A at max. The ESC we're using can supply upto 40A, which is comfortably safe for our use case.



 2-4s LiPo Compatility - This works best with common drone battery setups like 3S - 4S. Another reason for this choice is that it matches our motor's voltage tolerance (D2836 runs fine upto 4S).

Figure 4.2: ReadytoSky 40A 2-4S ESC

- **BLHeli Firmware** Comes pre-flashed with BLHeli, and this offers a smooth throttle response; brake support (for fixed wing or quad hybrid designs).
- Thermal Performance and Build The ESC is rating to suport 40A, which means it can take bursty loads without frying good for maneuvers, sudden wind corrections, or takeoffs.

Propellor:

Battery: Orange 7.4V 5200 MAH

The Orange 7.4V 5200 MAH battery is optimum for our choice, for the following reasons:

Generally, 2s LiPo batteries are thermally stable. They account for less ESC or motor overheating, especially in long-endurance missions. The **discharge-rate** of this battery is **35C**, and so, the maximum Theoretical discharge current than can be supplied is



Figure 4.3: Orange 7.4V 5200 MAH

Max Continuous Current = Capacity (Ah)
$$\times$$
 C - rate = $5.2 \mathrm{Ah} \times 35$ = $182 A$

From the official datasheet of the battery, it is rated at **156A**, which is not what we're expecting. So we can assume that the actual discharge-rate is **30C**.

Now, the DYS D2836-7 motor draws on an average of 18-22A current on a 2s LiPo battery with 10x4.2 propellor. Let us assume that on full throttle, with a load also, it draws 25A current. If a single motor draws 25A, then total current drawn by quad motors, $25 \times 4 = 100A$. That is well enough below the total theoretical discharge current of 156A. In that case, if we're using the motor at full-throttle, and it draws 25A current, then

Discharge Load
$$=\frac{25}{156}=16.02\%$$
 of battery's safe limit

Discharge Time (hrs) =
$$\frac{5.2}{25 \times 4} = 0.052$$
 hrs = 3.12 minutes

This means the battery is well within safe thermal and chemical limits.

- 4.2.2 Powertrain Efficiency Calculations
- 4.3 Avionics and Control Systems
- 4.3.1 Flight controller, sensors, and navigation
- 4.3.2 Communication System
- 4.3.3 Autonomous Capabilites
- 4.4 Payload and Gripper Mechanism
- 4.4.1 Integration with the UAV
- 4.4.2 Impact on aerodynamics and weight
- 4.5 Ground Control Systems
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- 4.5.2 Mission Execution strategy

Safety, Compliance & Risk Assessment

- 5.1 Compliance with SAEINDIA/FAA guidelines
- 5.2 Risk analysis and mitigation strategies
- 5.3 Emergency procedures

Bill of Materials



Innovations and Future Scope