# Computational Fluid Dynamics: The Finite-Volume Method

**David Apsley** 



# 1. Introduction

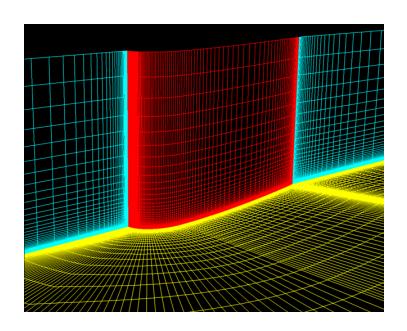


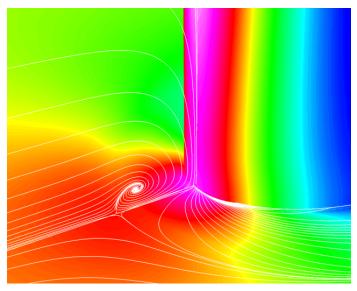
# What is ... Computational Fluid Dynamics?

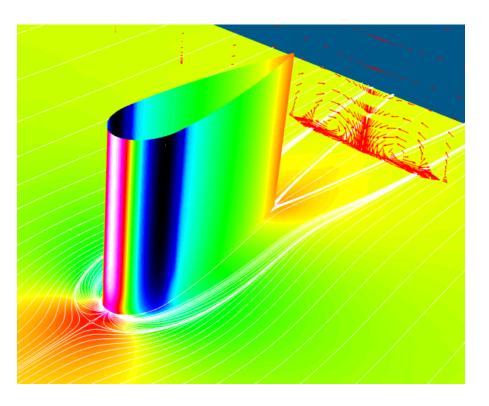
The use of computers and numerical methods to solve problems involving fluid flow



# **Aerodynamics**

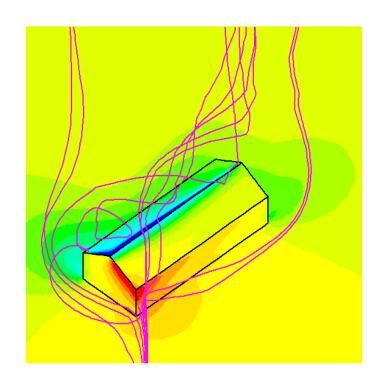


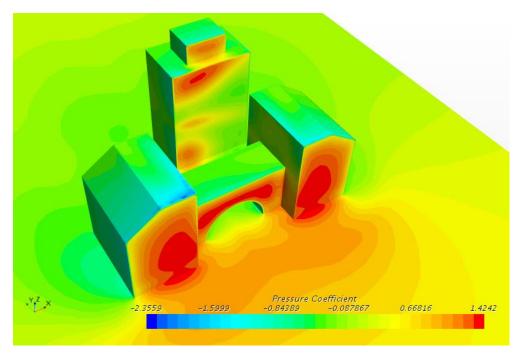






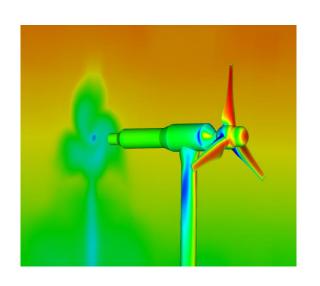
# **Wind Loading**

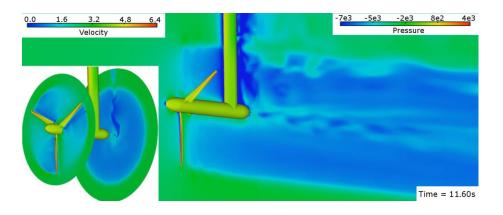


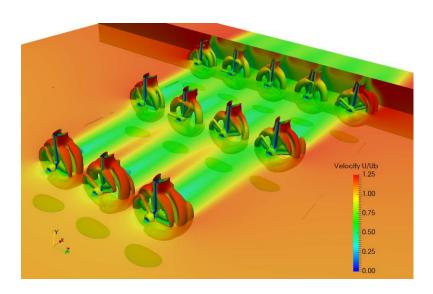


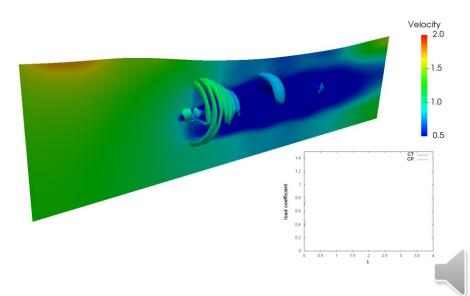


# **Turbine Technology**

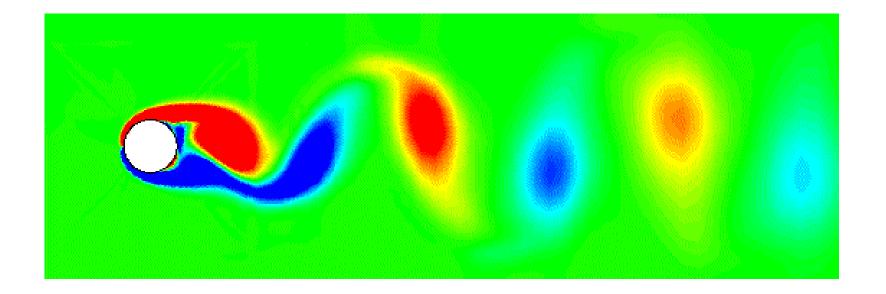






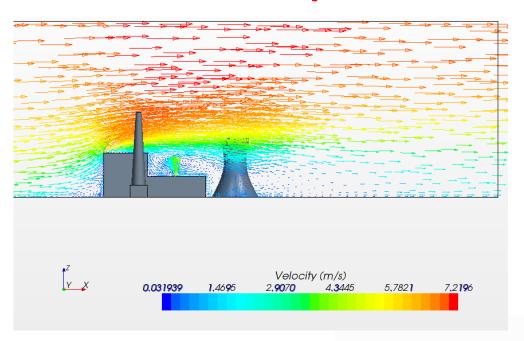


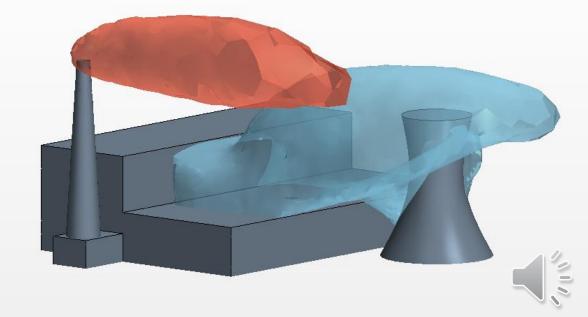
# **Vortex Shedding**



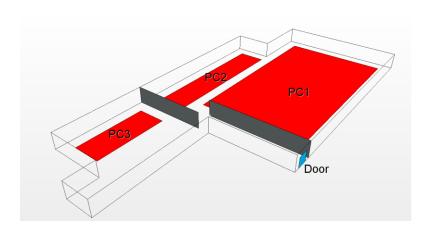


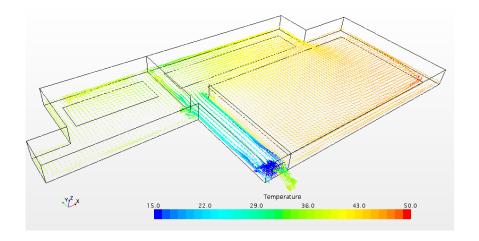
# **Dispersion of Pollution**

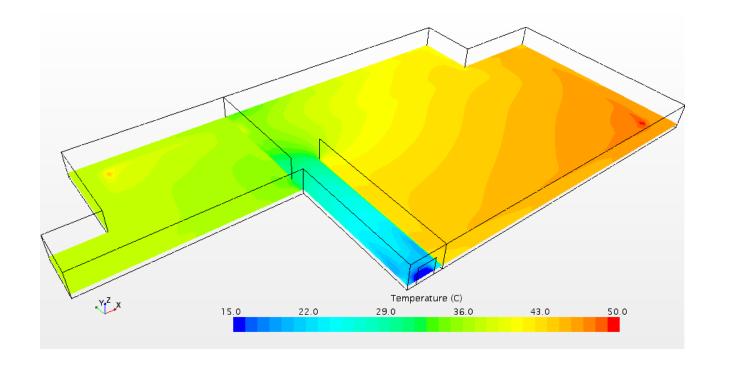




# **Ventilation**

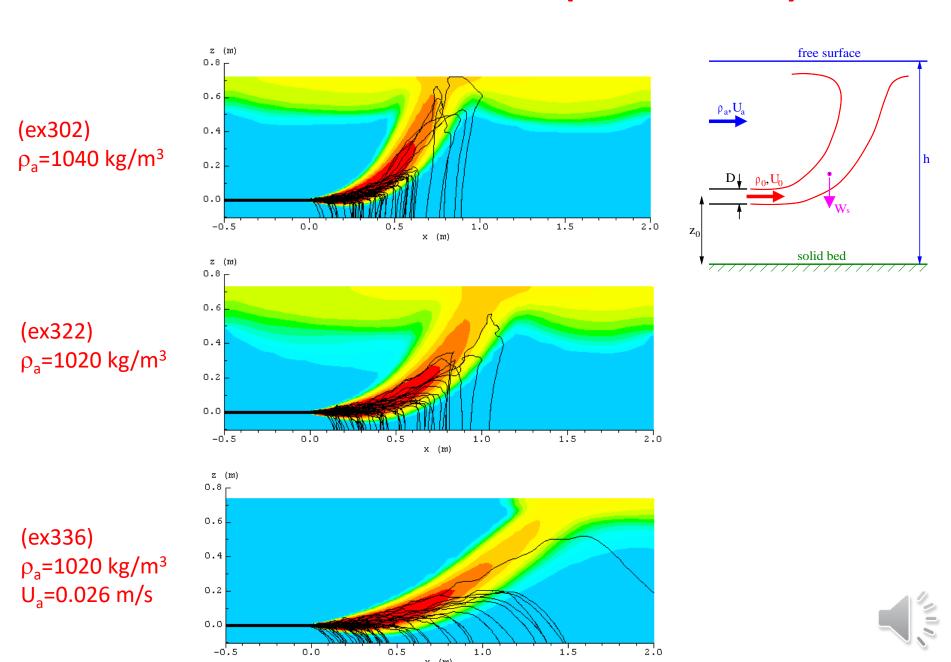




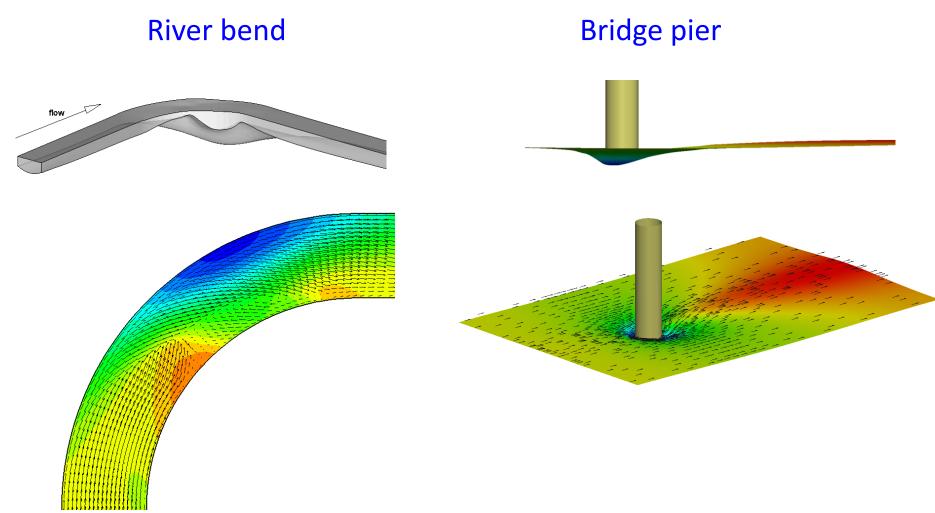




# Particle-Laden Plumes (Sea Outfalls)



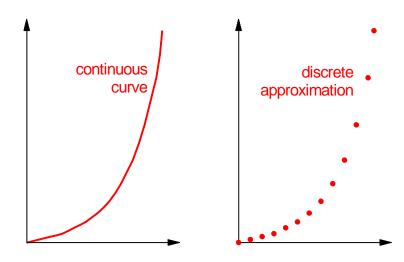
# **Sediment Scour**



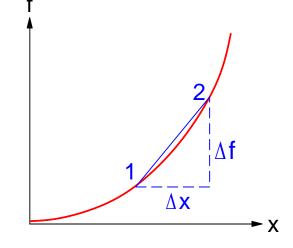


### **Discretisation**





#### **Equations:**



$$\frac{\mathrm{d}f}{\mathrm{d}x} \approx \frac{\Delta f}{\Delta x} = \frac{f_2 - f_1}{x_2 - x_1}$$



## **Basic Principles of CFD**

#### 1. Discretise space:

replace field variables  $(\rho, u, v, w, p, ...)$  by values at a finite number of **nodes** 

#### 2. Discretise equations:

**continuum** equations → **algebraic** equations

#### 3. Solve:

large system of simultaneous equations



# **Stages of a CFD Analysis**

#### Pre-processing:

- formulate problem (geometry, equations, boundary conditions)
- construct computational mesh

#### Solving:

- discretise
- solve

#### Post-processing:

- analyse
- visualise (graphs and plots)



## **Fluid-Flow Equations**

• Mass: change of mass = 0

Momentum: change of momentum = force × time

Energy: change of energy = work + heat

(Other constituents)

In fluid mechanics, these are normally expressed in rate form



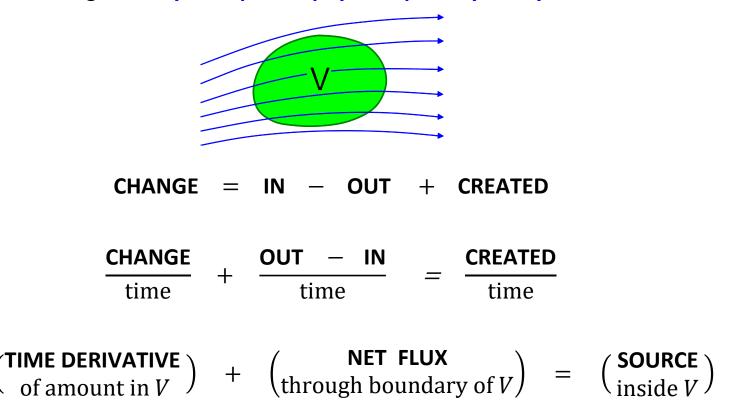
# **Form of Equations**

- Integral (control-volume)
- Differential



# Integral (Control-Volume) Approach

Consider the budget of any transported physical quantity in any control volume





## **Differential Equations For Fluid Flow**

- Derived by considering the rate of change at a point; i.e. using infinitesimal control volumes
- Discretisation gives a finite-difference method for CFD
- Several types:
  - fixed-point ("Eulerian"): conservative
  - moving with the flow ("Lagrangian"): non-conservative
  - derived variables; e.g. potential flow

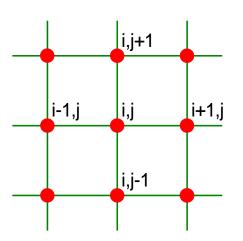


### **Main Methods for CFD**

#### Finite-difference:

discretise differential equations

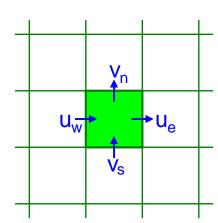
$$0 = \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \qquad \approx \frac{u_{i+1,j} - u_{i-1,j}}{2\Delta x} + \frac{v_{i,j+1} - v_{i,j-1}}{2\Delta y}$$



#### Finite-volume:

discretise control-volume equations

$$0 = \text{net mass outflow} = (\rho u A)_e - (\rho u A)_w + (\rho v A)_n - (\rho v A)_s$$



#### • Finite-element:

represent solution as a weighted sum of basis functions

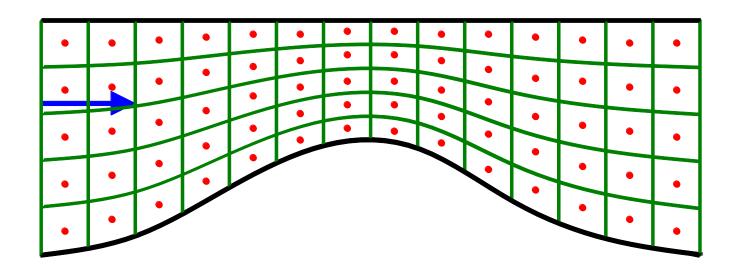
$$u(\mathbf{x}) = \sum u_{\alpha} S_{\alpha}(\mathbf{x})$$



## **Advantages of the Finite-Volume Method in CFD**

- Rigorously enforces conservation
- Flexible in terms of:
  - geometry
  - fluid phenomena
- Directly relatable to physical quantities





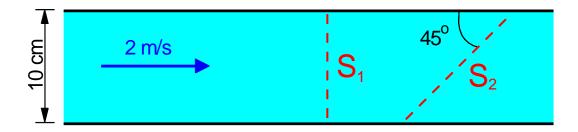


# **Examples**



## **Example Q1**

Water (density 1000 kg m<sup>-3</sup>) flows at 2 m s<sup>-1</sup> through a circular pipe of diameter 10 cm. What is the mass flux C across the surfaces  $S_1$  and  $S_2$ ?





Water (density 1000 kg m<sup>-3</sup>) flows at 2 m s<sup>-1</sup> through a circular pipe of diameter 10 cm. What is

the mass flux C across the surfaces  $S_1$  and  $S_2$ ?

$$S_1$$
: mass flux (C) =  $\rho uA$   
=  $1000 \times 2 \times \frac{\pi \times 0.1^2}{4}$   
=  $15.7 \text{ kg s}^{-1}$ 

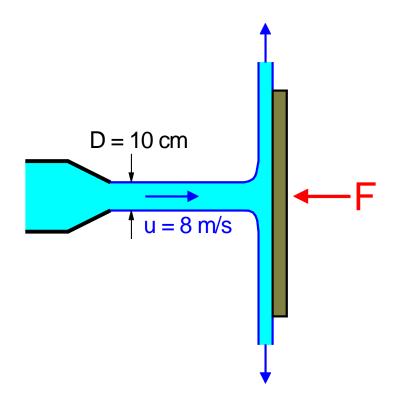
$$S_2$$
: the same!  $C = \rho(u\cos\theta)A$   
 $C = \rho u(A\cos\theta)$ 

In general: 
$$C = \rho \mathbf{u} \cdot \mathbf{A}$$



# **Example Q2**

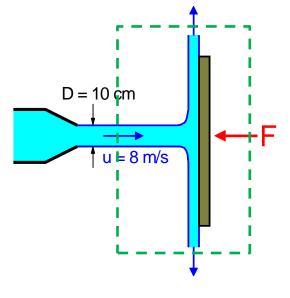
A water jet strikes normal to a fixed plate as shown. Compute the force F required to hold the plate fixed.





A water jet strikes normal to a fixed plate as shown. Compute the force  ${\it F}$  required to hold the

plate fixed.



force (on fluid) =  $(momentum flux)_{out} - (momentum flux)_{in}$ 

momentum flux = mass flux  $\times$  velocity

$$-F = 0 - (\rho UA)U$$

$$F = \rho U^2 A$$

$$= 1000 \times 8^2 \times \frac{\pi \times 0.1^2}{4}$$

$$= 503 \text{ N}$$



## **Example Q3**

An explosion releases 2 kg of a toxic gas into a room of dimensions 30 m  $\times$  8 m  $\times$  5 m. Assuming the room air to be well-mixed and to be vented at a speed of 0.5 m s<sup>-1</sup> through an aperture of area 6 m<sup>2</sup>, calculate:

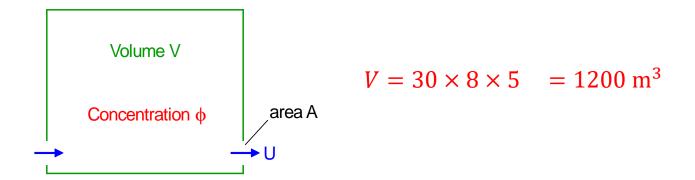
- (a) the initial concentration of gas in ppm by mass;
- (b) the time taken to reach a safe concentration of 1 ppm.

(Take the density of air as  $1.2 \text{ kg m}^{-3}$ .)



An explosion releases 2 kg of a toxic gas into a room of dimensions 30 m  $\times$  8 m  $\times$  5 m. Assuming the room air to be well-mixed and to be vented at a speed of 0.5 m s<sup>-1</sup> through an aperture of area 6 m<sup>2</sup>, calculate:

(a) the initial concentration of gas in ppm by mass;



mass of fluid  $\times$  concentration = mass of toxin

$$(\rho V)\phi_0 = 2 \text{ kg}$$

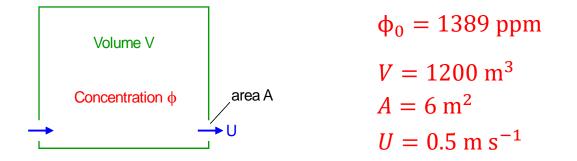
$$\phi_0 = \frac{2}{1.2 \times 1200} = 1.389 \times 10^{-3}$$

$$1389 \text{ ppm}$$



An explosion releases 2 kg of a toxic gas into a room of dimensions 30 m  $\times$  8 m  $\times$  5 m. Assuming the room air to be well-mixed and to be vented at a speed of 0.5 m s<sup>-1</sup> through an aperture of area 6 m<sup>2</sup>, calculate:

(b) the time taken to reach a safe concentration of 1 ppm.



Change in amount of toxin = amount in - amount out

Rate of change of amount of toxin = rate of entering - rate of leaving

$$\frac{d}{dt}(\rho V \phi) = 0 - (\rho u A) \phi$$

$$\frac{d\phi}{dt} = -\left(\frac{u A}{V}\right) \phi, \qquad \phi = \phi_0 \text{ at } t = 0$$

$$\frac{d\phi}{dt} = -\lambda \phi \qquad \lambda = \frac{u A}{V} = 0.0025 \text{ s}^{-1}$$

$$\phi = \phi_0 e^{-\lambda t}$$

$$\frac{d}{dt}(\rho V \phi) = 0 - (\rho u A) \phi$$

$$\frac{d\phi}{dt} = -\left(\frac{u A}{V}\right) \phi, \qquad \phi = \phi_0 \text{ at } t = 0$$

$$t = \frac{1}{\lambda} \ln \frac{\phi_0}{\phi} = \frac{1}{0.0025} \ln(1389) = 2895 \text{ s}$$

$$\approx 48 \text{ min}$$

