

# A bit of history...



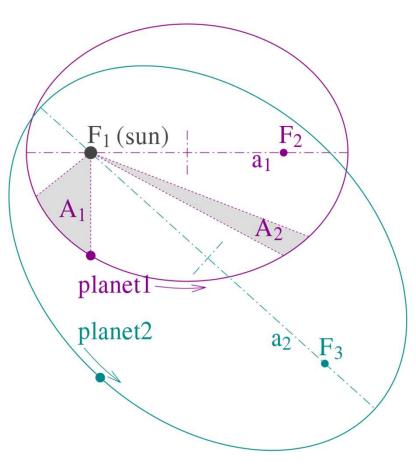


Issac Newton and Edmund Halley

Tycho Brahe and Johannes Kepler, Statue in Prague, Czech Republic

#### Rules of thumb

- Newton's laws of motion
- Newton's law of gravitation
- Kepler's laws of planetary motion
  - <u>First Law</u> The orbit of each planet is an ellipse, with the Sun at a focus
  - <u>Second Law</u> The line joining the planet to the Sun sweeps out equal areas in equal times.
  - <u>Third Law</u> The square of the period of a planet is proportional to the cube of its mean distance from the Sun.



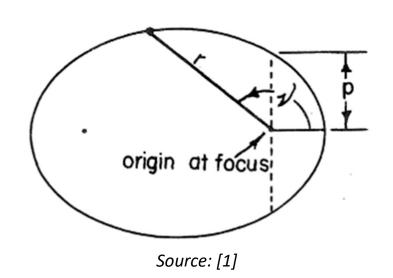
### Mathematical description of orbits

Equations of motion:  $\ddot{r} + \frac{\mu}{r^3} r = 0$ 

Analytical solution:  $r = \frac{h^2/\mu}{1 + (B/\mu)\cos(\nu)}$ 

General equation of a conic section:

$$r = \frac{p}{1 + e\cos(\nu)}$$



All orbits in a two-body system are conic sections!

## Geometry of an orbit

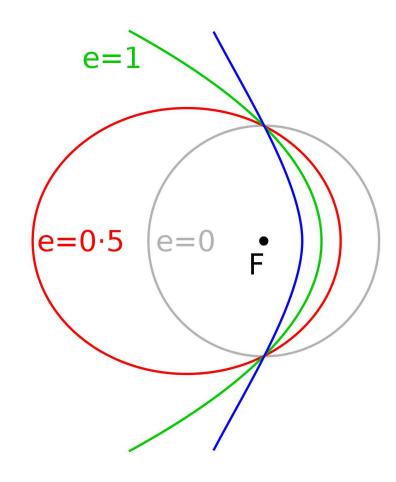
Relationship between Energy and geometry of orbit:

$$\mathcal{E} = -\frac{\mu}{2a}$$

$$e = \sqrt{1 + \frac{2\mathcal{E}h^2}{\mu^2}}$$

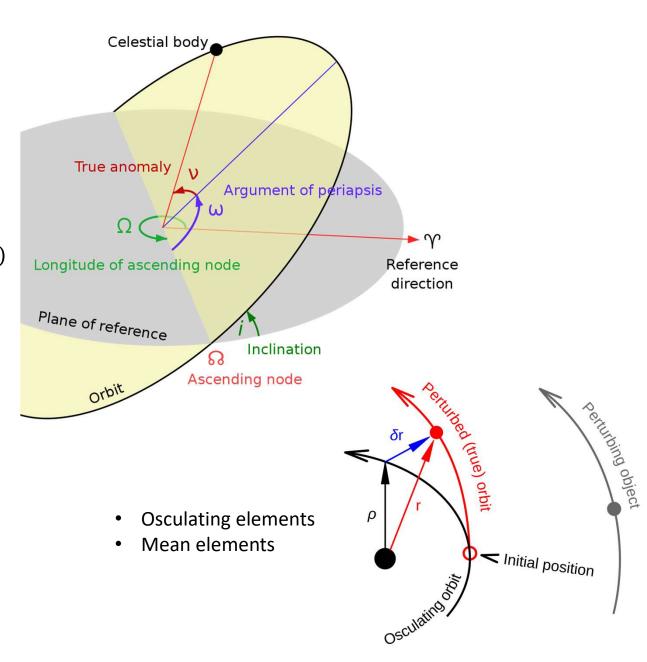
where  $\mathcal{E} = E/\mu$  is specific orbital energy

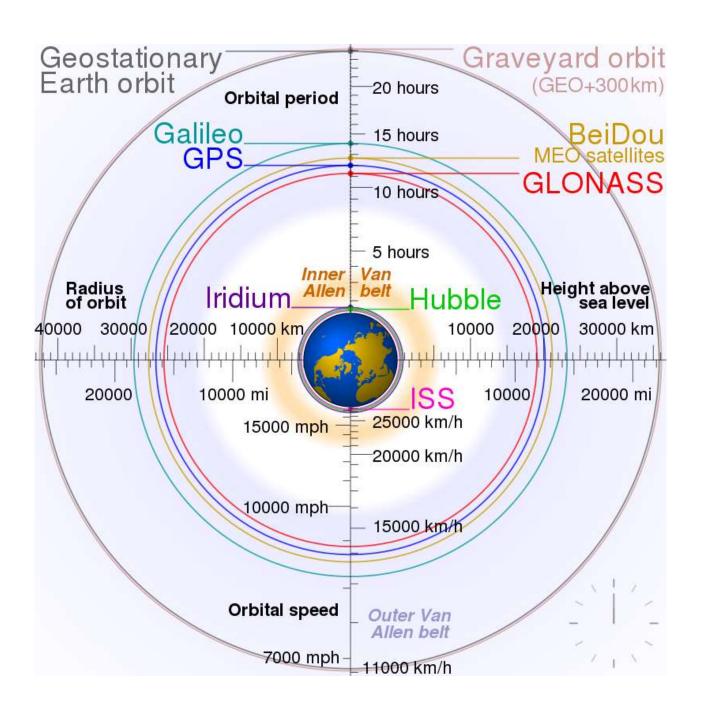
- Elliptical ( $0 \le e < 1, \varepsilon < 0$ )
- Parabolic (e = 1,  $\mathcal{E}$  = 0)
- Hyperbolic (e > 1,  $\varepsilon > 0$ )



#### Orbital elements

- Semi-major axis (a)
- Eccentricity (e)
- Inclination (i)
- Longitude of the ascending node  $(\Omega)$
- Argument of periapses  $(\omega)$
- True anomaly  $(\nu)$





# Achieving and Maintaining Orbit



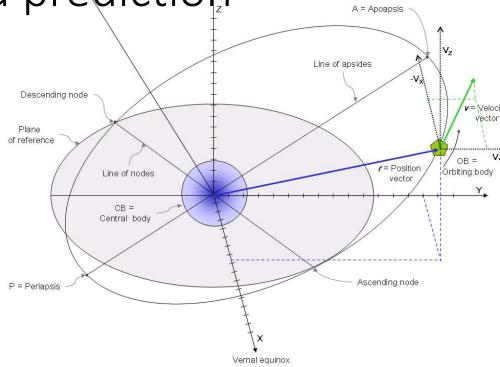


Credits: NASA

Orbit determination and prediction

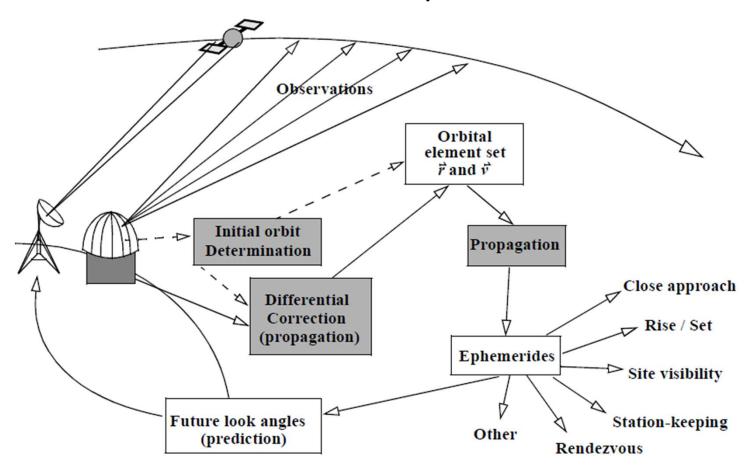


The Raisting Satellite Earth Station, Germany



Orbital state vectors of a satellite in orbit around earth

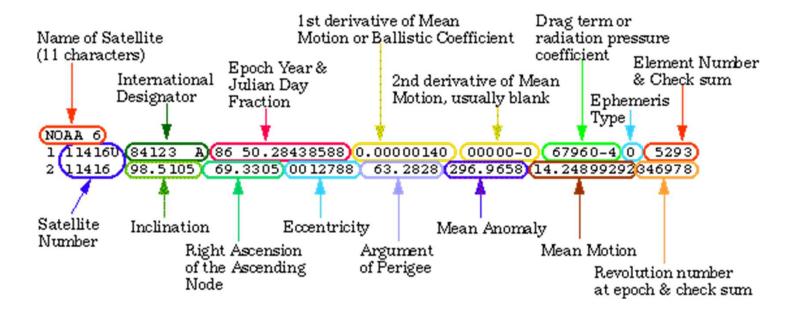
### Orbit determination and prediction



Credits: David A Vallado

#### Orbit determination and prediction

TLEs along with Simplified General Perturbations models (SGP, <u>SGP4</u>, SDP4, SGP8 and SDP8) are used for orbit prediction



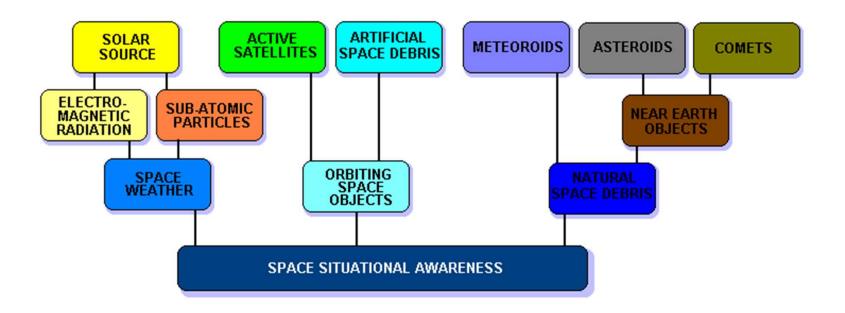
Credits: NASA

Intermission

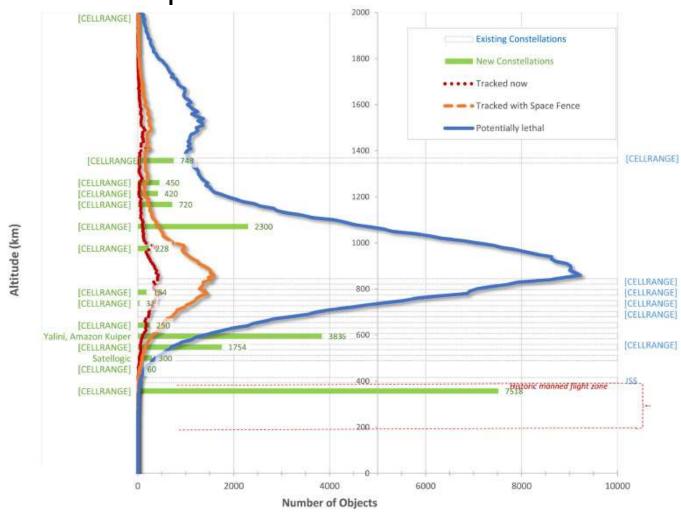
# Space Situational Awareness



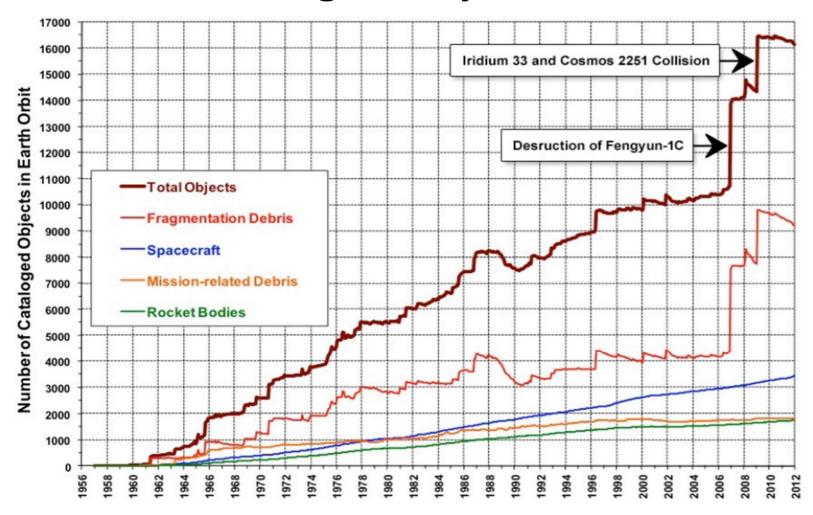
# Components of SSA



Why is SSA important?

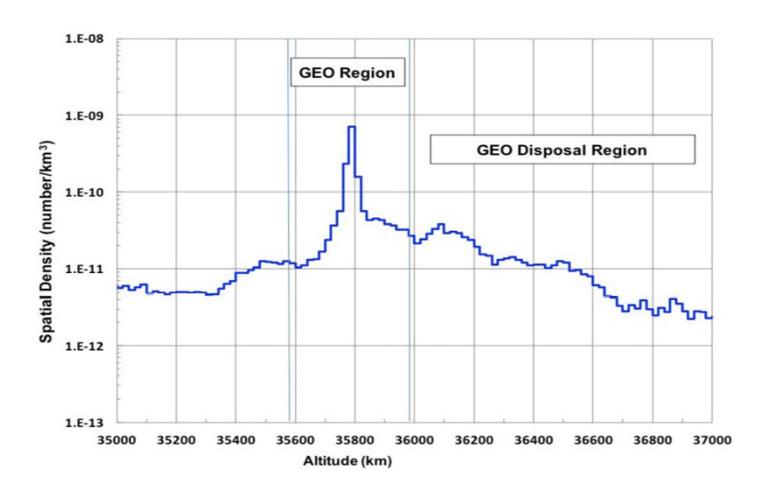


#### Number of Cataloged Objects in LEO



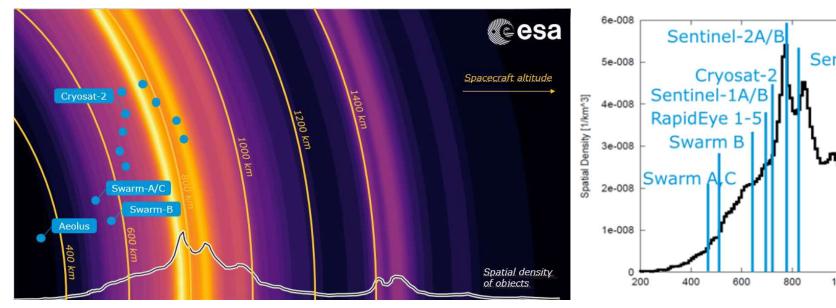
Source: NASA

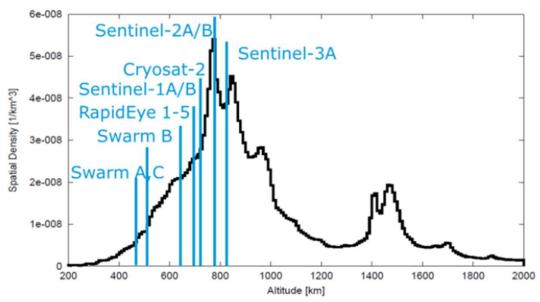
### Number of Cataloged Objects in GEO



Source: NASA

### The threat of orbital space debris

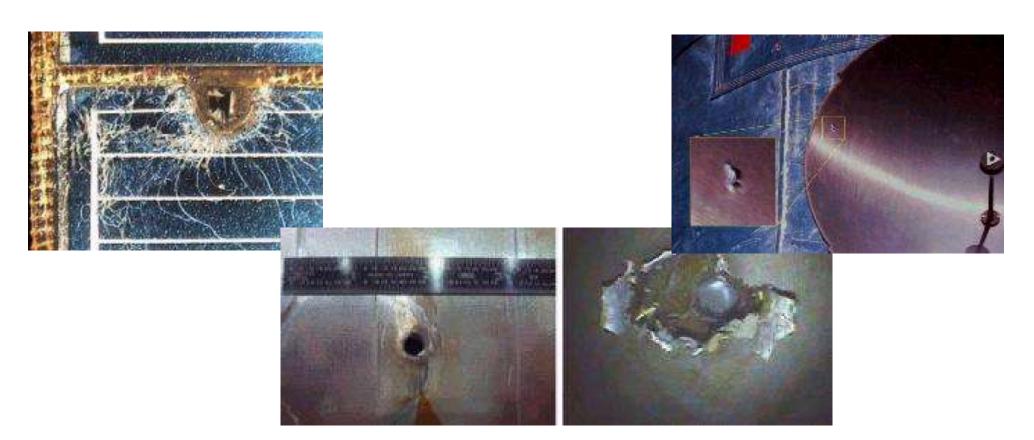




Spatial density of objects > 10 cm

Credits: ESA

# Debris damage



Credits: https://www.spaceacademy.net.au/watch/debris/gsd/gsd.htm

### Worst collisions in history

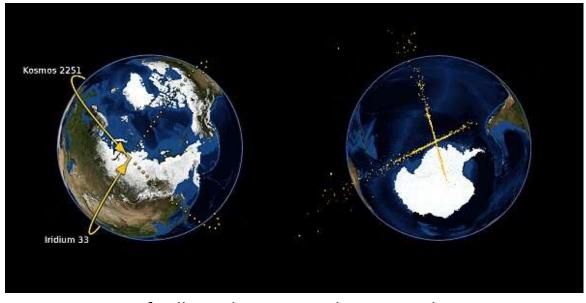
#### 2007 Chinese anti-satellite missile test



View of ISS Orbit (green band) and LEO satellites (green dots) along with debris ring (red) from Chinese ASAT Test

Credit: Dr. Thomas Kelso at CSSI (Center for Space Standards and Innovation)

#### 2009 Iridium - Cosmos collision



Point of collision between Iridium 33 and Kosmos 2251 alongside the debris field after 50 minutes

#### Collision Avoidance

#### Identify conjunctions

 by propagating objects' orbits into the future

#### Generate CDMs

 containing information regarding the conjunction

#### Propose maneuvers

 By analyzing CDMs that show high risk of collision

### Sustainability of Space

#### Debris Mitigation - Minimize creation of future debris

- Prevent in-orbit explosions and collisions
- Post-mission disposal

#### Space Traffic Management - Minimize effect of debris on spacecraft

- Collision avoidance
- Spacecraft shielding

#### Debris Removal - Remove existing debris from orbit

Dedicated space mission

### Further Reading

- 1. Fundamentals of Astrodynamics by Roger R. Bate, Donald D. Mueller and Jerry E. White
- 2. Fundamentals of Astrodynamics and Applications by David A. Vallado
- 3. An Overview Of Space Situational Awareness. <a href="https://www.spaceacademy.net.au/intell/ssa.htm">https://www.spaceacademy.net.au/intell/ssa.htm</a>
- 4. SSA Programme Overview. https://www.esa.int/Safety\_Security/SSA\_Programme\_overview\_
- 5. Space traffic management in the new space era. Muelhaupt et. al. The Journal of Space Safety Engineering (2019)

