

Turbofan Jet Engine Predictive Maintenance

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1. Problem statement
Objective

2. Data/EDA

3. Predictive model

4. Recommendation



Problem statement & Objective

Problem statement: Aircraft engine maintenance

- 1) Significant engine failure: Passengers' life, Reputation & Cost
i.e. B777-300ER Max passenger: 392* Cost: \$375.5 million**
- 2) Break-down replacement: \$30 million***
- 3) Maintenance frequency: Normally every 28 flight cycles
(Assume normal maintenance frequency)****

Objective (Assume B777-300ER GE90 engine)

Develop a predictive model for the engine life time (RUL)

Define engine maintenance criteria

*<https://www.boeing.com/commercial/777>

**<https://simpleflying.com/how-much-does-a-boeing-777-cost/>

#:~:text=The%20Boeing%20777%2D200ER%20costs,777%2D300ER%20costs%20%24375.5%20million

***<https://simpleflying.com/most-expensive-jet-engines/>

****https://www.aircraft-commerce.com/wp-content/uploads/aircraft-commerce-docs/Maintenance/2013/ISSUE87_MTCE_A.pdf

Remaining Useful Life (RUL)*

The estimated amount of time an asset has until it becomes unusable or requires replacement.
This is commonly used in predictive maintenance to determine when maintenance need to be performed on equipment.

For this project

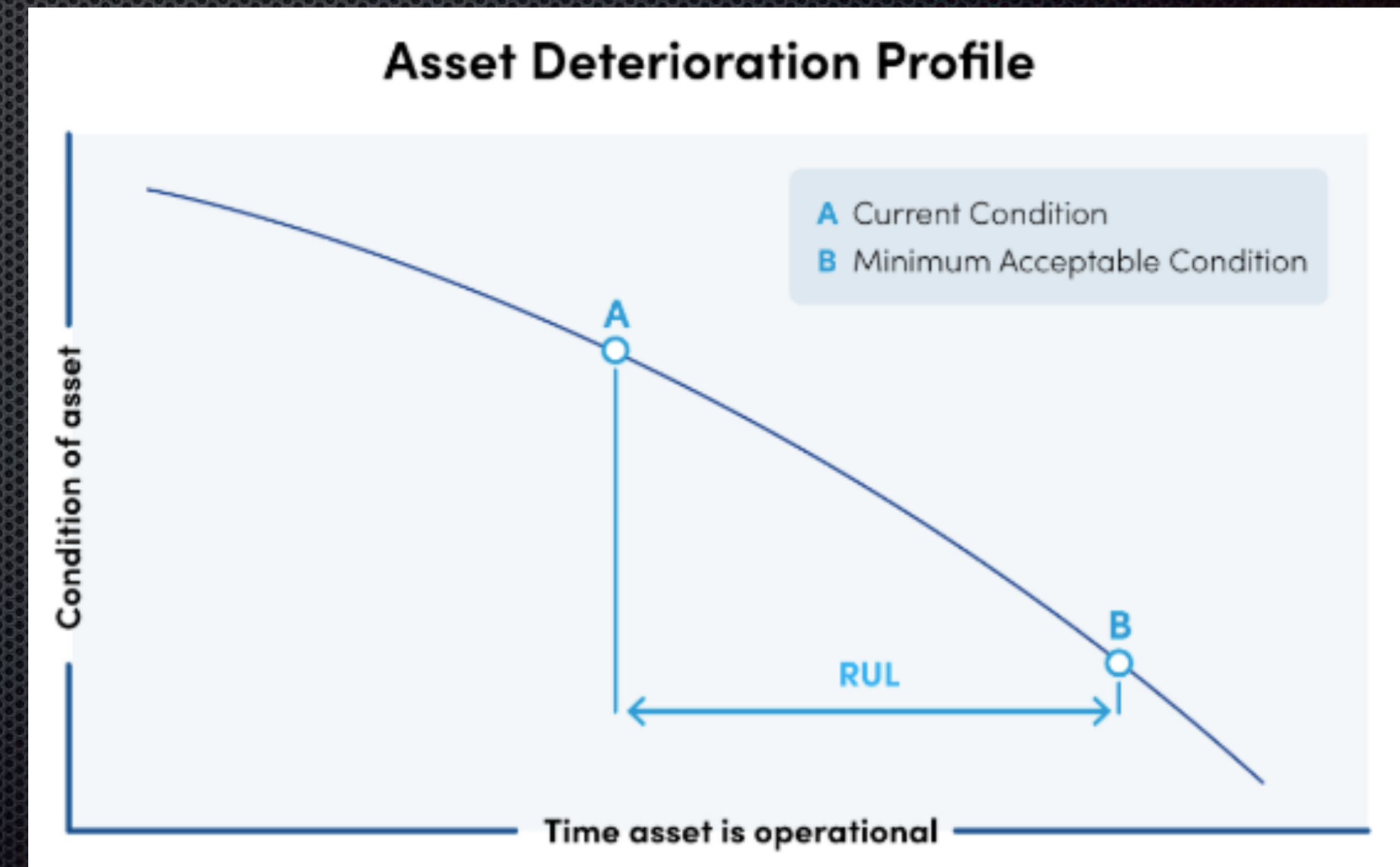
(Train data)

Maximum acceptance cycle - Subsequent cycle

(Test data)

Conditional data for certain time cycle and predict RUL rather than current condition is abnormal or not

*<https://fiixsoftware.com/maintenance-metrics/remaining-useful-life/>



Data

Data → Predictive Model → Recommendation

Feature selection/engineering

Data
extraction

EDA

Select features

Normalized by start value

Maximum RUL limit

Dataset



- ✦ Downloaded from Kaggle <https://www.kaggle.com/datasets/behrad3d/nasa-cmaps/data>

Ref: Reference: A. Saxena, K. Goebel, D. Simon, and N. Eklund, Damage Propagation Modeling for Aircraft Engine Run-to-Failure Simulation, in the Proceedings of the 1st International Conference on Prognostics and Health Management (PHM08), Denver CO, Oct 2008

- ✦ Train & Test data already separated (100 engines)
We will split train data into train & validation set

<i>Symbol</i>	<i>Description</i>	<i>Units</i>
engine	Engine number	—
cycle_time	Cycle number	—
operational_set_1	Operational set up #1	—
operational_set_2	Operational set up #2	—
operational_set_3	Operational set up #3	—
T2	Total temperature at fan inlet	°R
T24	Total temperature at LPC outlet	°R
T30	Total temperature at HPC outlet	°R
T50	Total temperature at LPT outlet	°R
P2	Pressure at fan inlet	psia
P15	Total pressure in bypass-duct	psia
P30	Total pressure at HPC outlet	psia
Nf	Physical fan speed	rpm
Nc	Physical core speed	rpm
epr	Engine Pressure ratio (P50/P2)	—
Ps30	Static pressure at HPC outlet	psia
phi	Ratio of fuel flow to Ps30	pps/psi
NRf	Corrected fan speed	rpm
NRc	Corrected core speed	rpm
BPR	Bypass Ratio	—
farB	Burner fuel-air ratio	—
htBleed	Bleed Enthalpy	—
Nf_dmd	Demanded fan speed	rpm
PCNfR_dmd	Demanded corrected fan speed	rpm
W31	HPT coolant bleed	lbm/s
W32	LPT coolant bleed	lbm/s

Exploratory Data analysis

No null value

```
Null value rate
engine      0.0
cycle_time  0.0
operational_set_1  0.0
operational_set_2  0.0
operational_set_3  0.0
T2          0.0
T24         0.0
T30         0.0
T50         0.0
P15         0.0
P30         0.0
Nf          0.0
Nc          0.0
Ps30       0.0
NRf         0.0
NRc         0.0
BPR         0.0
farB        0.0
htBleed     0.0
Nf_dmd      0.0
PCNfR_dmd   0.0
W31         0.0
W32         0.0
RUL         0.0
```

*No null cases
could differ from
actual case; i.e.
sensor fault

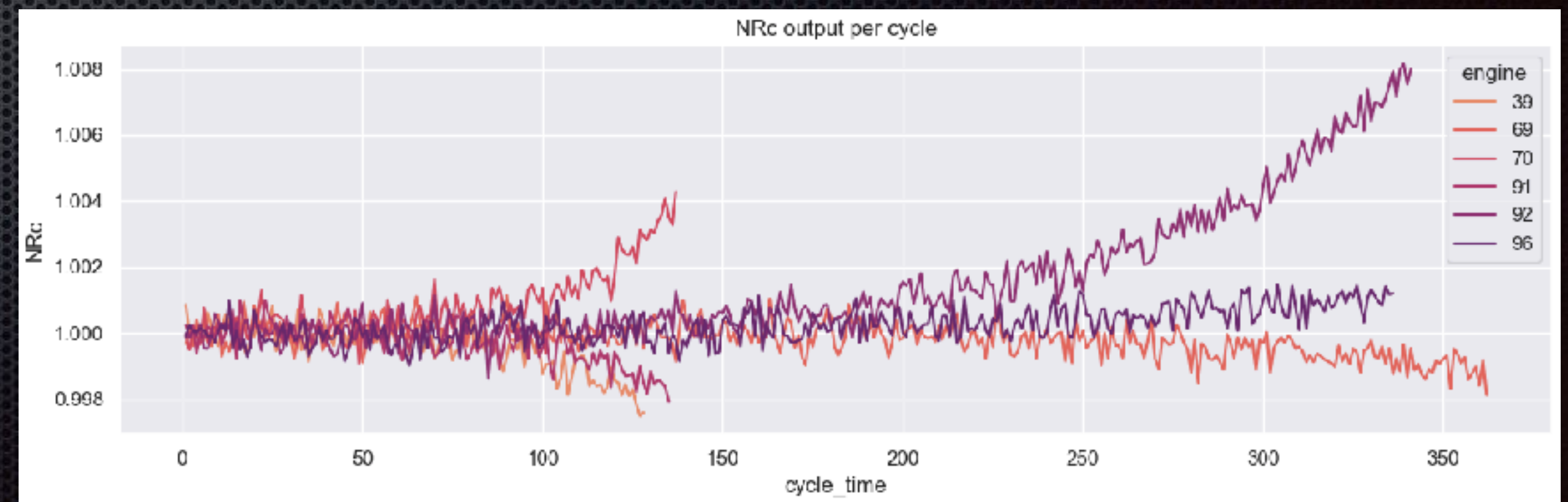
Remove columns

```
Unique value count
engine      100
cycle_time  362
operational_set_1  158
operational_set_2   13
operational_set_3    1
T2          1
T24         310
T30         3012
T50         4051
P2          1
P15         2
P30         513
Nf          53
Nc         6403
epr         1
Ps30        159
phi         427
NRf         56
NRc        6078
BPR         1918
farB        1
htBleed     13
Nf_dmd      1
PCNfR_dmd   1
W31        120
W32        4745
```

Each sensor output per cycle times



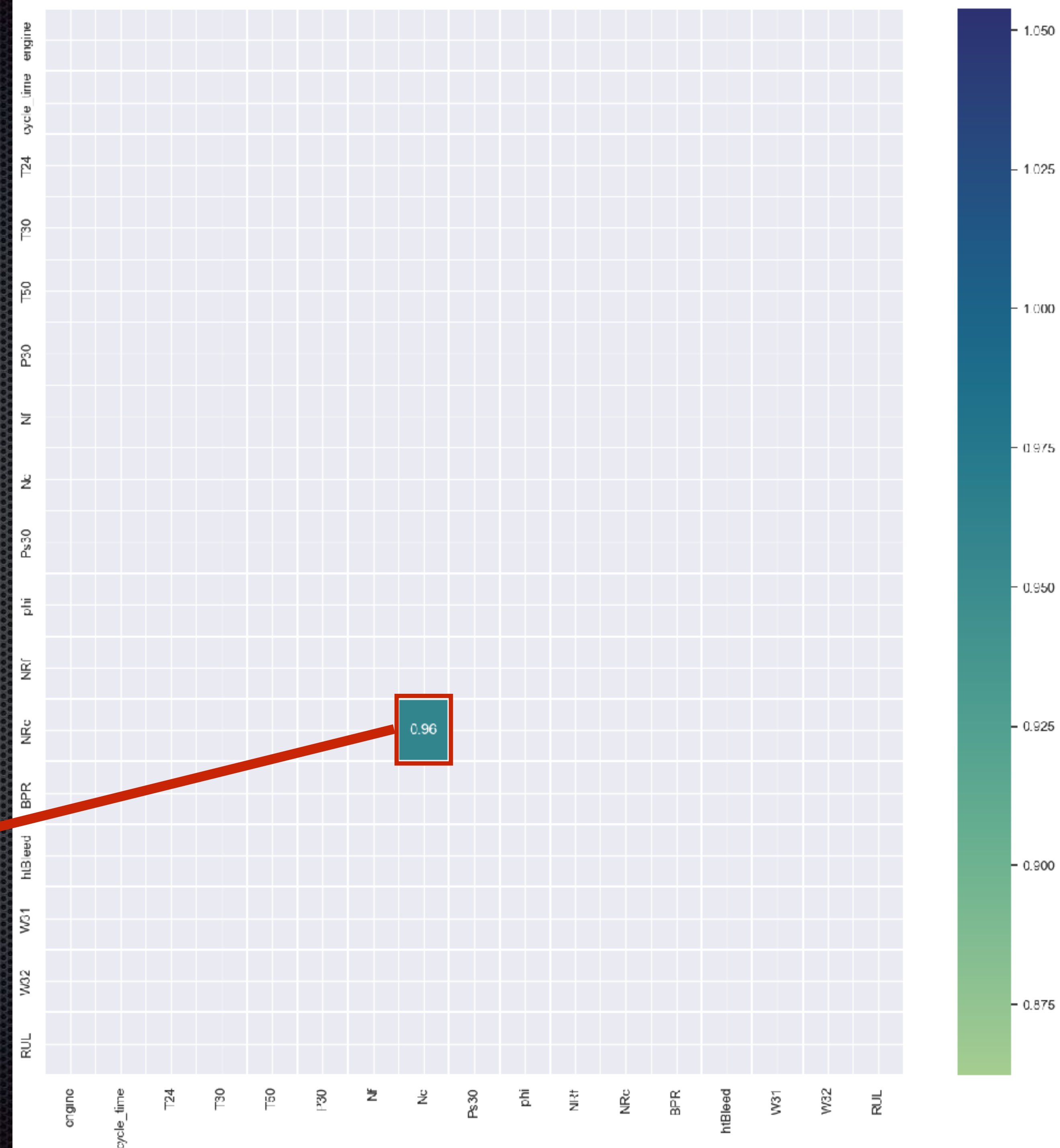
- 1) Signals have noise; Smoothing..?
→ Not implemented for this time (Assume real situation)
- 2) Different scale of each sensor: Normalization by start value



Reduce features

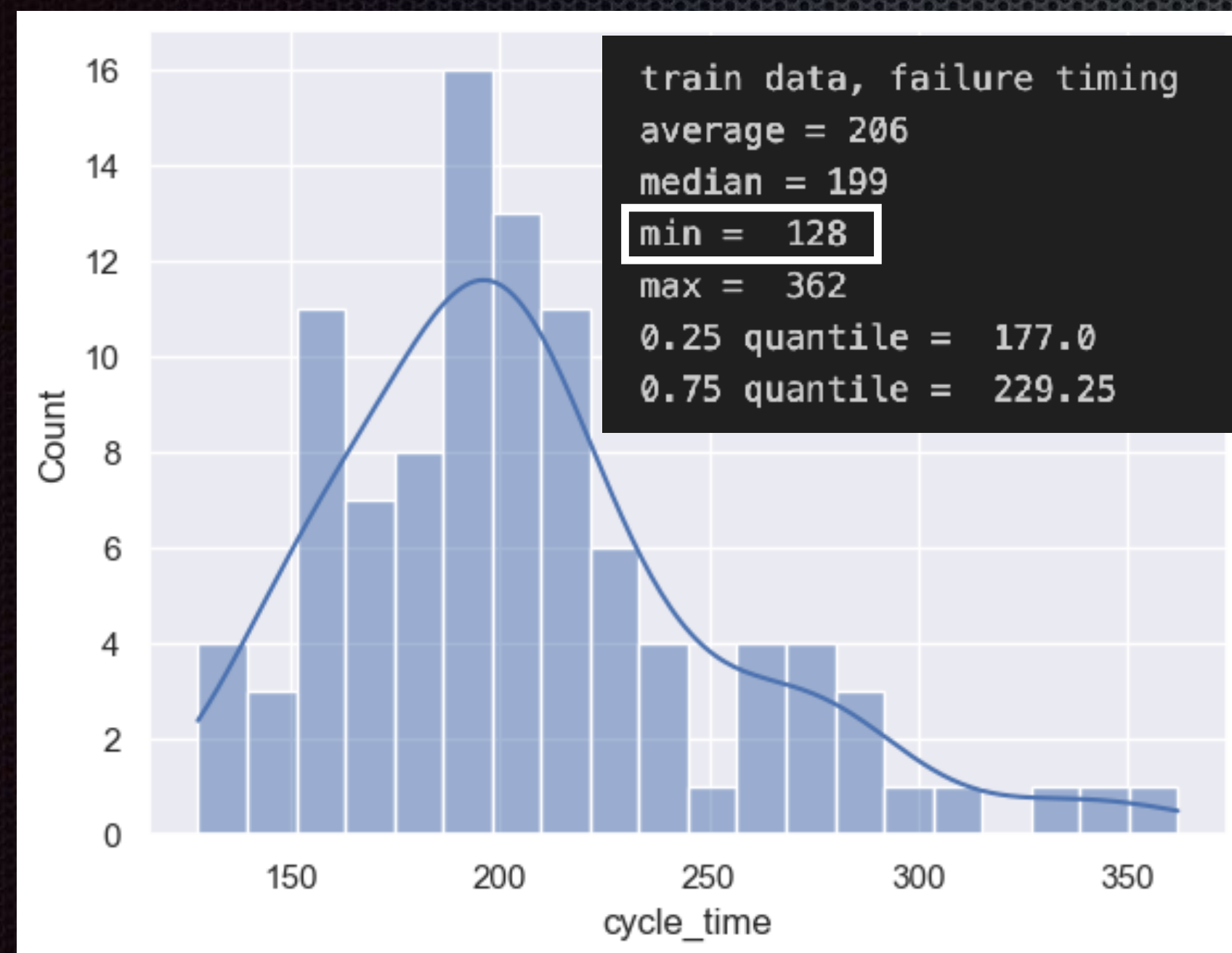
High correlated features are removed

Symbol	Description	Units
T2	Total temperature at fan inlet	°R
T24	Total temperature at LPC outlet	°R
T30	Total temperature at HPC outlet	°R
T50	Total temperature at LPT outlet	°R
P2	Pressure at fan inlet	psia
P45	Total pressure in bypass duct	psia
P30	Total pressure at HPC outlet	psia
Nf	Physical fan speed	rpm
Nc	Physical core speed	rpm
epi	Engine Pressure Ratio (P50/P2)	
Ps30	Static pressure at HPC outlet	psia
phi	Ratio of fuel flow to Ps30	pps/psi
NRf	Corrected fan speed	rpm
NRc	Corrected core speed	rpm
BPR	Bypass Ratio	—
farB	Bypass fuel/air ratio	
htBleed	Bleed Enthalpy	—
Nf_dmd	Demanded fan speed	rpm
PCHNP_dmd	Demanded corrected fan speed	rpm
W31	HPT coolant bleed	lbm/s
W32	LPT coolant bleed	lbm/s



Maximum RUL limit (Clipping)

- We are interested in RUL which requires maintenance
- No need to have precise prediction higher RUL



engine	cycle_time	T24	T30	T50	P30	Nf	Nc	Ps30	phi	NRf	BPR	htBleed	W31	W32	RUL	
0	1	1	641.82	1589.70	1400.60	554.36	2388.06	9046.19	47.47	521.66	2388.02	8.4195	392	39.06	23.4190	191
1	1	2	642.15	1591.82	1403.14	553.75	2388.04	9044.07	47.49	522.28	2388.07	8.4318	392	39.00	23.4236	190
2	1	3	642.35	1587.99	1404.20	554.26	2388.08	9052.94	47.27	522.42	2388.03	8.4178	390	38.95	23.3442	189
3	1													23.3739		188
4	1													23.4044		187

Set up upper RUL limit based on
historical minimum life time * 0.95
(safety net) = 121

engine	cycle_time	T24	T30	T50	P30	Nf	Nc	Ps30	phi	NRf	BPR	htBleed	W31	W32	RUL	
0	1													1.001257		121
1	1	2	0.999921	1.002587	1.001642	0.999376	0.999996	0.999393	1.004612	1.000257	1.000011	1.002784	1.000510	1.000410	1.001454	121
2	1	3	1.000232	1.000175	1.002399	1.000296	1.000013	1.000373	0.999958	1.000525	0.999994	1.001119	0.995406	0.999128	0.998059	121
3	1	4	1.000232	0.996900	1.000735	1.000639	1.000026	0.999991	0.996996	1.001367	1.000015	0.995220	1.000510	0.997332	0.999329	121
4	1	5	1.000263	0.996938	1.003841	0.999827	1.000005	1.000618	1.000169	1.000084	0.999998	1.002499	1.003063	0.997845	1.000633	121

Predictive model

Data → Predictive Model → Recommendation

Regression models

- 1) Linear
- 2) ElasticNet
- 3) SVR
- 4) Random Forest
- 5) LightGBM
- 6) XGBoost



- 1) Train data →
Train/Validation data split
- 2) Hyper parameter tuning:
GridSearchCV with Train data
- 3) Model selection: MSE
with Train/Validation data



Prediction with
test data



Maintenance
flag

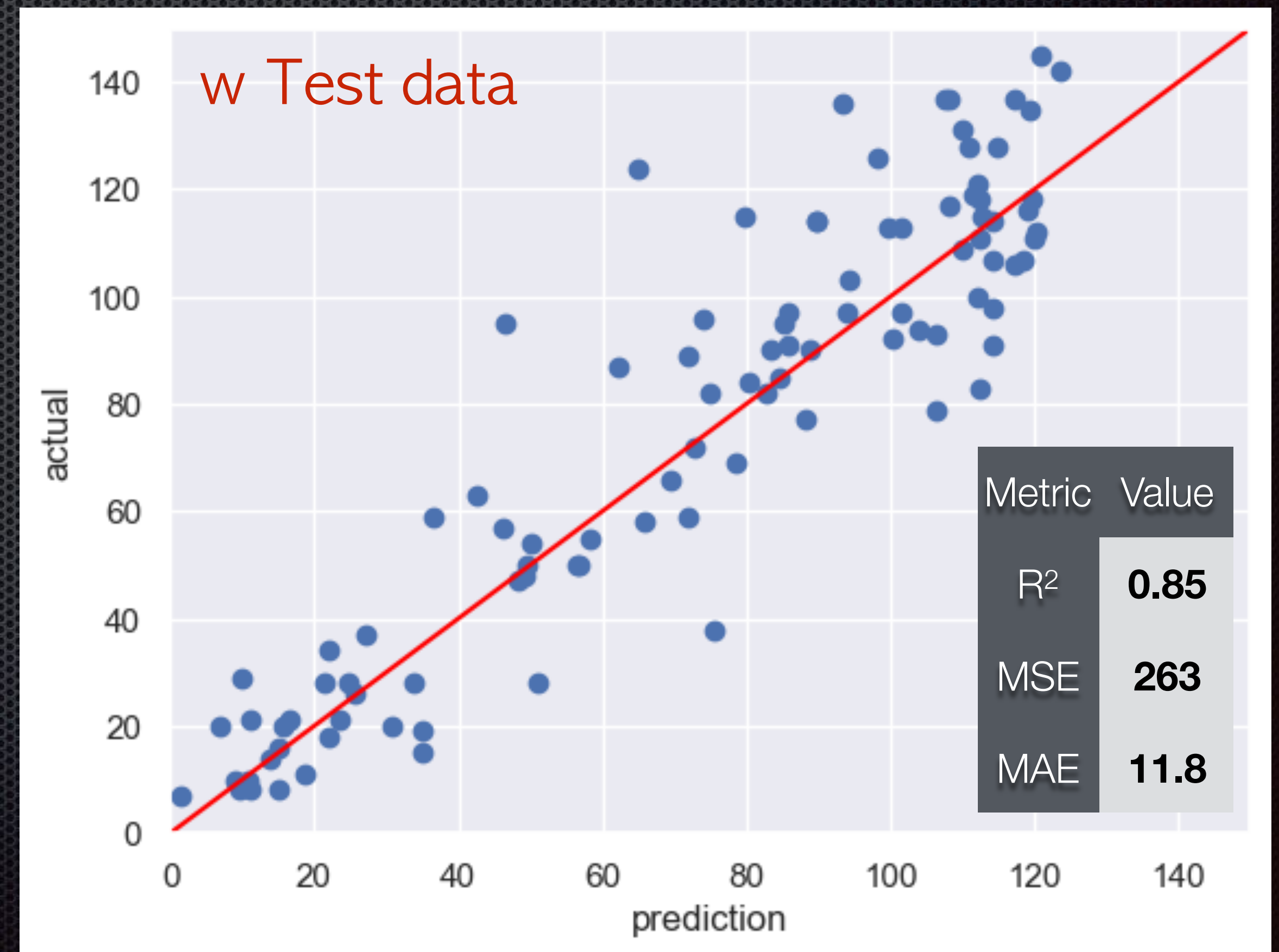
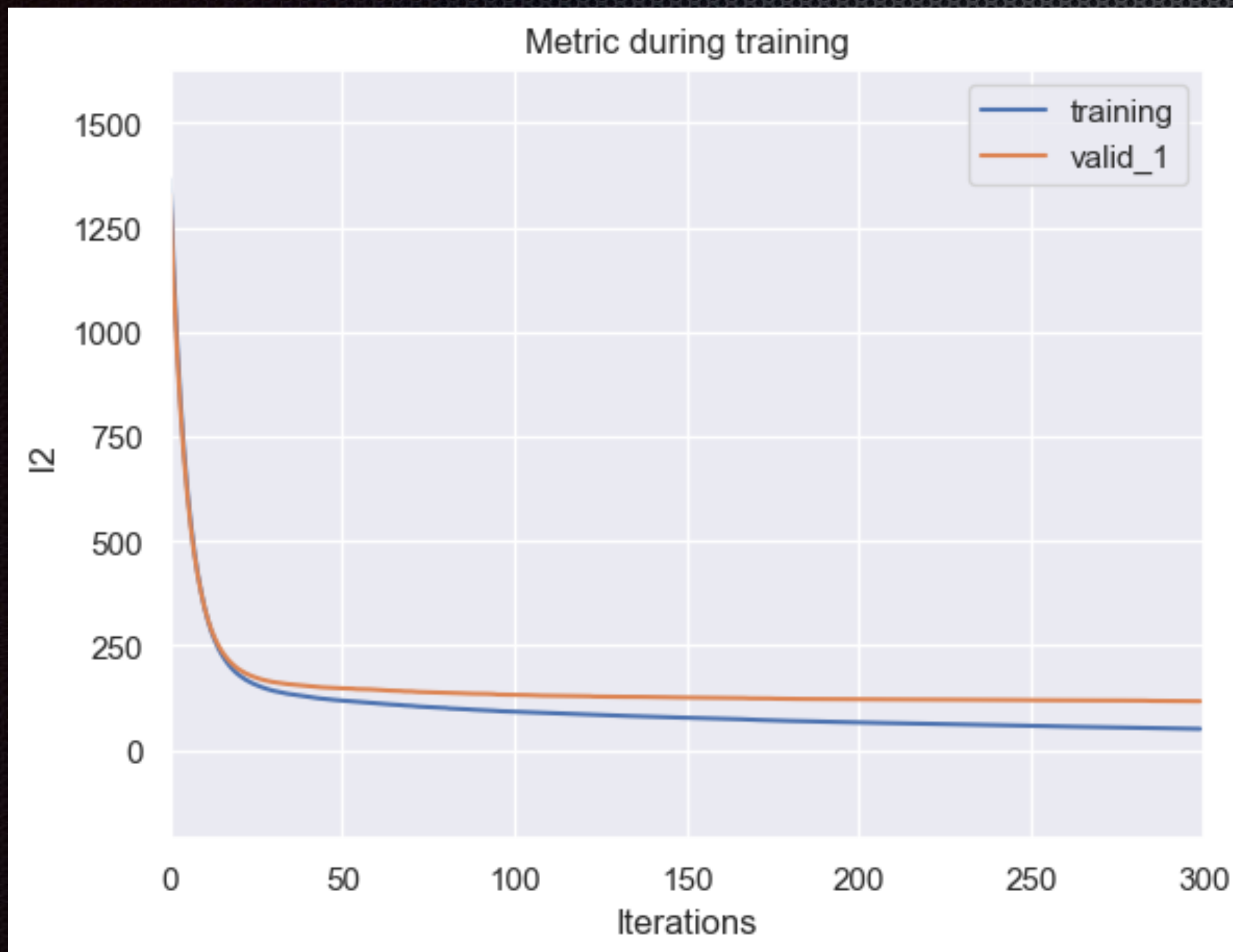
Model selection

- Hyper parameter tuning done by GridSearchCV

Model	Train (MSE)	Validation (MSE)	Execution time (Train, msec)	Execution time (Validation, msec)
Linear	313.0	296.8	6	2
ElasticNet	734.5	691.1	30	1
SVR	708.4	659.9	41629	3516
Random Forest	37.5	152.5	98706	133
LightGBM	51.0	117.2	3445	19
XGBoost	69.7	125.7	12636	8

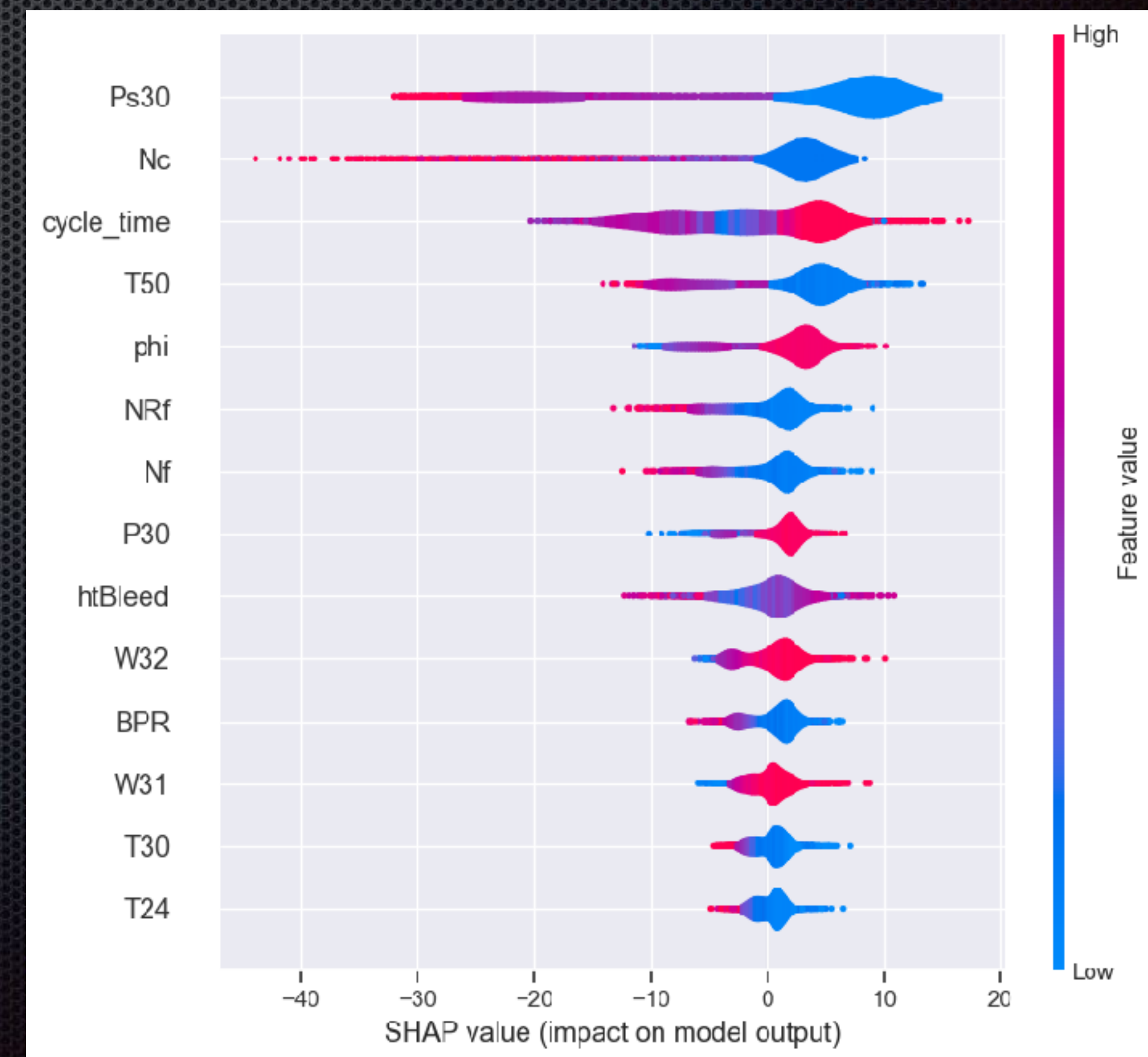
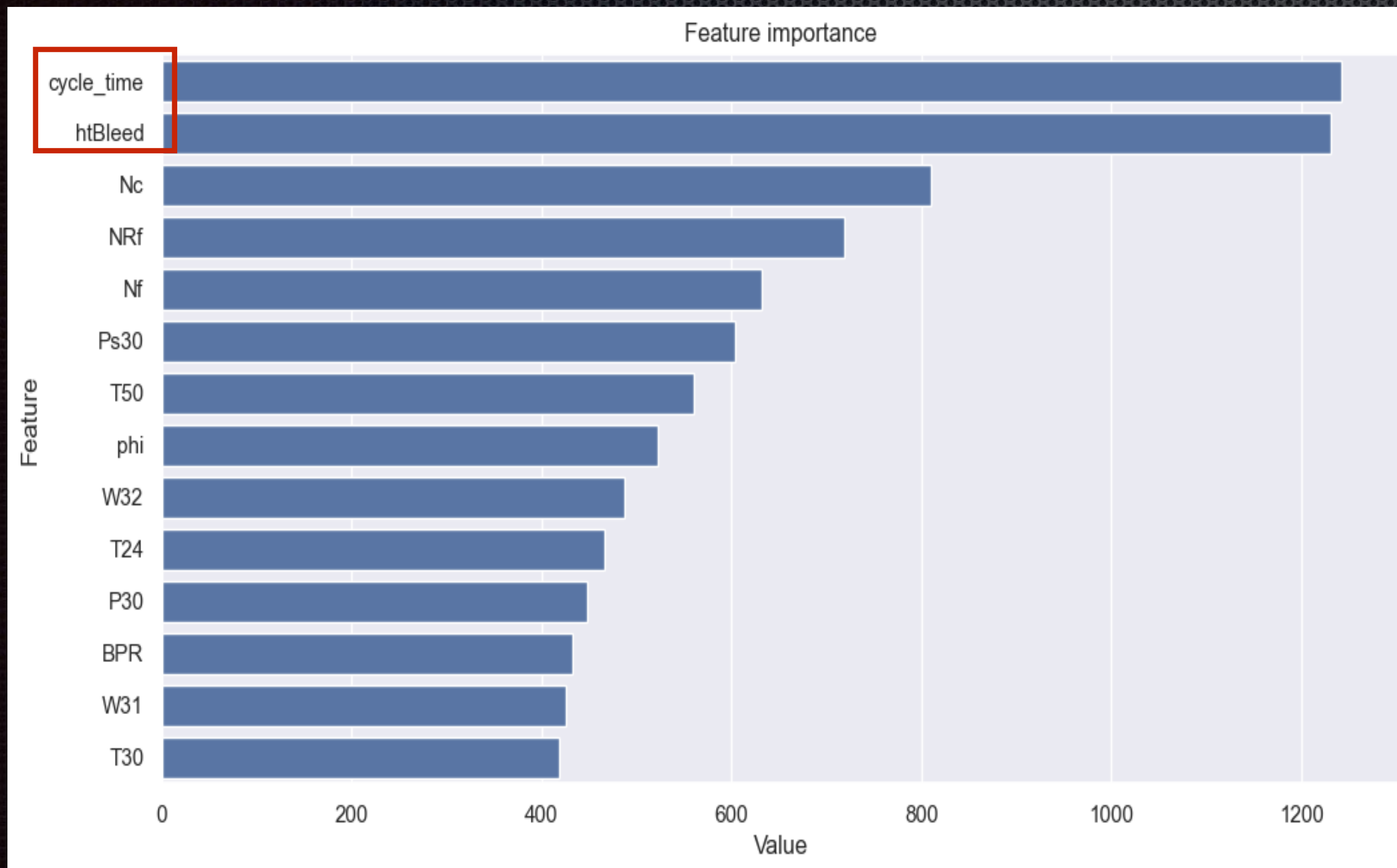
Model selection

- Hyper parameter tuning done by GridSearchCV



LGBM; feature importance

- htBleed (Bleed Enthalpy) & Cycle time are key features



Maintenance flag

- Lead time of engine maintenance: 3~6weeks*
 - 42 flight cycles (6 weeks) for the maintenance vs. current 28 flight cycles (Assume 1 flight/day)

- Mapping maintenance flag if RUL ≤ 42
 - Accuracy of model: 97%
 - Recall: 92.9%
 - Precision: 96.3%

		Actual	
		Need maintenance	Not yet
Pred	Need maintenance	26	1
	Not yet	2	71

* <https://www.avbuyer.com/articles/engine-maintenance-hub#:~:text=How%20long%20does%20aircraft%20engine, costs%20of%20aircraft%20engine%20maintenance.>

Recommendation



- Predictive model: LightGBM for regression of RUL
- Maintenance frequency: 28 flight cycles → 42 flight cycles (30% reduction/year)
 - Max. cost avoidance: ~\$700 million/year*

* # of day a year * # of engine x maintenance cost** / maintenance frequency * (Recall / Precision)

** \$1.4 million (assume C-checks, https://www.aircraft-commerce.com/wp-content/uploads/aircraft-commerce-docs/Maintenance/2013/ISSUE87_MTCE_A.pdf)

Limitation & Further improvement

Limitations

- 1) Sensor physical faults: Predictive model will not work
- 2) Maximum cycle time (121 cycles) assumed by historical records: It could vary by engines

Further improvement idea (Recall: 92.9%)

- 1) Apply smoothing on each sensor signal
- 2) Feature engineering; Add more features & PCA