A REPORT

<u>ON</u>

MODELING, SIMULATION AND FABRICATION OF LATTICE BASED IMPLANT

SUBMITTED IN PARTIAL FULFILMENT FOR THE DEGREE OF **BACHELOR OF TECHNOLOGY**

IN

MECHANICAL ENGINEERING

BY

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GHAZIABAD

<u>ABSTRACT</u>

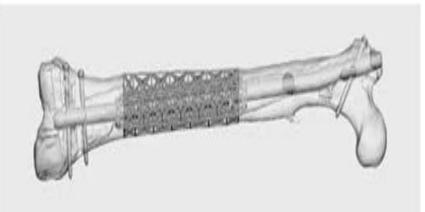
Orthopedic regenerative medicine is the latest trend in biomedical sector. Designing of bone scaffolds and implants is a challenging step followed by its fabrication and surgical implantation. Lattice structure based designs are desirable candidates for healing or replacing the damaged bones, as they provide larger surface area for osseointegration. Additive manufacturing has the potential to fabricate these complex lattice structures. In these work SolidWorks 2020 CAD software is used for modelling the hollow cubical lattice structure which are then scaled while maintaining the same overall size and volume. After designing Finite Element Analysis (FEA) is performed to investigate the compressive behaviors of the lattice structures using the Ansys FEA package. The compressive modulus of porous scaffolds for stainless steel and TI-6Al-4V are calculated and compared with that of compact bone.

INTRODUCTION

IMPLANT

- ➤ It is a medical device manufactured to replace a missing biological structure, support a damaged biological structure, or enhance an existing biological structure.
- ➤ Implants are required for large segmental damaged bones because they cannot heal themselves, as against the small damaged bones which have the ability of self healing.
- ➤ The main goal of implants is to help patients with disabilities to return to normal function for the longest possible duration. Further, implants can be used either to augment existing performance of the body or to replace missing tissues, organs or parts of the body.

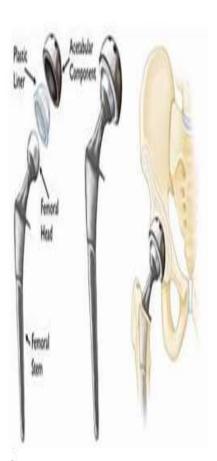




Dental Implant







Cranial Implant

Orthopedic Implant

Unit cell & Scaffold

The unit cell is the smallest part, that repeated regularly in three dimensions to creates the scaffold

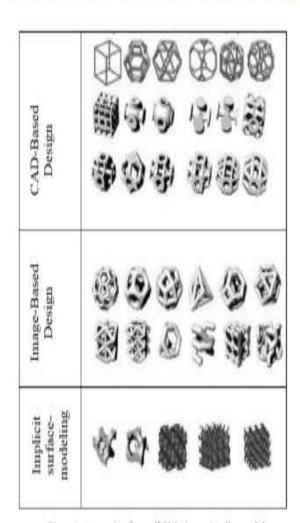
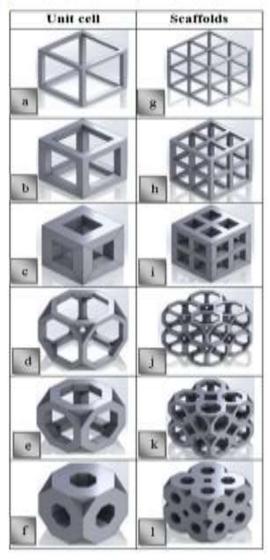


Figure 1. Approaches for scaffold design: unit cell types [1]



Use & Advantage of Scaffold

- Use to make a bridge between the gap of two ends of damaged bone.
- Porous scaffold is best for the growth (proliferation) and deposition (vascularization).
- Porous scaffold are also good in osseointegration(the direct structural and functional connection b/w living bone and surface of load bearing artificial implant.

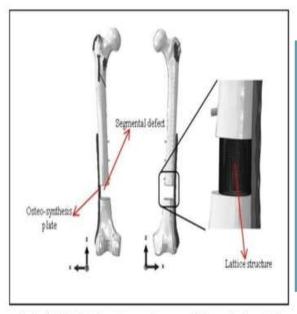
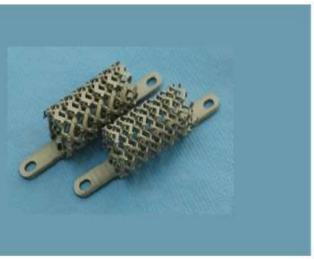


Fig. 1. Model of the femur bone and segmental defect with the applied osteo-synthesis plate and filled with lattice structure based implant [3].



Lattice Based Implant

Additive Manufacturing

- The process of joining materials to make objects from 3D model data, usually layer by layer.
- Commonly known as 3D printing.
- Manufacturing components with virtually no geometric limitations or tools.

Advantages of Additive Manufacturing

- It has the capability to fabricate external as well as internal architecture.
- Cost effective, low wastage, rapid manufacturing of parts or components that can be customized basis.

Techniques used to manufacture Porous Scaffold

- Fused Deposition Modeling
- Selective Laser Melting
- Selective Laser Sintering

- Electron Beam Melting
- Direct Laser Deposition

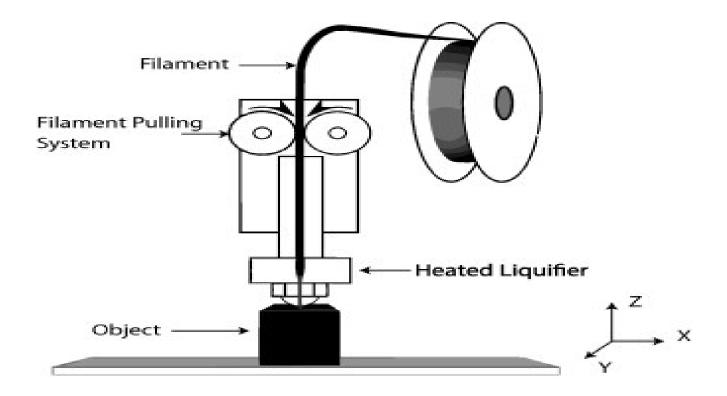
Fused Deposition Modeling (FDM)

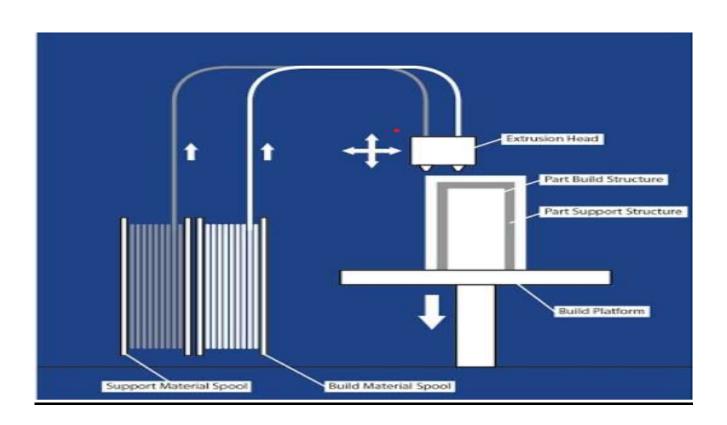
➤ FDM Printer use a thermoplastic filament, which is heated to its melting point and then extruded, layer by layer, to create a 3D object.

→ Polymers Printed by FDM

- PLA (Polylactic Acid)
- ABS (Acrylonitrile Butadiene Styrene)
- PET (Polyethylene terephthalate)
- Nylon
- TPU (Thermoplastic polyurethane)
- PU (polyurethane)







DESIGN & METHODLOGY

- Lattice based part designed in SolidWorks (CAD software)
 & import in Ansys (simulation software).
- Two rigid plates are modeled, assembled with lattice part.
- Surface to surface contact has been selected and the contact properties are taken as friction coefficient of 0.2 with normal behavior of hard contact.
- To simulate the boundary conditions, the lower plate is kept fixed and the upper plate is given a downward displacement of 0.01 mm.

FINITE ELEMENT ANALYSIS (FEA)

- ➤ Design geometry is a lot more complex; and the accuracy requirement is a lot higher. We need To understand the physical behaviours of a complex object (strength, heat transfer capability, fluid flow, etc.) To predict the performance and behaviour of the design; to calculate the safety margin; and to identify the weakness of the design accurately; and To identify the optimal design with confidence.
- FEA is used to simulate the compressive behavior of lattice structures.
- Compressive behavior of designed scaffold must be predicted before actual fabrication, to reduce the cost of experimentation and material.
- In this analysis, compressive behavior & strength of the lattice structures will be simulated.

Literature Review

| Sr No. | Paper Description | Remarks |
|-----------|---|---|
| 1. | Finite Element Modeling and Analysis of Implant Scaffolds By- T Bhardwaj, SP Singh, M Shukla. International Conference on, 2017 | Geometry of structure such that Elastic modulus is in the range of Elastic modulus of the cortical bone (3-30-GPa) By porous structure, surface area is increase. It helps to cell in growth & vascularization. |
| 2. | Lattice modeling and Finite Element Simulation for Additive Manufacturing of Porous Scaffolds By- T Bhardwaj, SP Singh, M Shukla International Conference on, 2017 | To avoid the stress shielding effect, low dense porous scaffolds are manufactured that provide cell attachment, mechanical stability and fluid perfusion. Effect of unit size scaling results in generating more surface area for same porosity that leads to more bone regeneration |

| Sr No. | Paper Description | Remarks |
|-----------|--|--|
| 3. | Additively manufactured porous metallic biomaterials By-Amir A Zadpoor - 2019 | Porous metallic biomaterials with topologically ordered unit cells have improving bone tissue regeneration and preventing implant-associated infections. Discussed how the huge (internal) surfaces of AM porous biomaterials and their pore space could be used respectively for surface biofunctionalization and accommodation of drug delivery vehicles so as to enhance their bone tissue regeneration performance and minimize the risk of implant-associated infections. We conclude with a general discussion and by suggesting some possible areas for future research. |

| Sr No. | Paper Description | Remarks |
|-----------|--|---|
| 4. | Additively manufactured functionally graded biodegradable porous zinc By - Y Li 1, P Pavanram 2, J Zhou 1, K Lietaert 3, F S L Bobbert 1, Yusuke Kubo 2, M A Leeflang 1, H Jahr 4, A A Zadpoor - 2020 | Two uniform AM porous Zn designs with diamond unit cell. Cylindrical specimens were fabricated from pure Zn powder by using a powder bed fusion technique, followed by a comprehensive study on their static and dynamic biodegradation behaviors, mechanical properties, permeability, and biocompatibility. Topological design, indeed, affected the biodegradation behavior of the specimens, as evidenced by 150% variations in biodegradation rate between the three different designs. Using topological design of AM porous Zn for controlling its mechanical properties and degradation behavior is thus clearly promising, thereby rendering flexibility to the material to meet a variety of clinical requirements. |

- 5. Direct Laser Deposition Additive Manufacturing
 of Ti-15Mo Alloy: Effect
 of Build Orientation
 Induced Surface
 Topography on Corrosion
 and Bioactivity
 - By- T Bhardwaj, M Shukla, NK Prasad, CP Paul

Published in- Metals & Materials, 2020 Springer

- Examined the higher content of refractory metals(Mo, Nb & Ta) in Ti alloy.
- But select Ti (15%)Mo due to better density as compare to other alloy composition.
- To optimize the DED-LAM process parameters for minimum dilution, RSM technique is used

- 6. Biomaterials & Scaffolds
 for tissue engineering
 Author of article Fregal
 J. O'Brien
- Biomaterials are generally categorized in three 1.-Ceramic, 2-Synthetic Polymer, 3 - Natural Polymer
- Scaffold should have a balance between mechanical properties and porous architecture which allows cell infiltration and vascularization & it is the key to success of any scaffold.
- Improvement in vascularization strategies is one of the area requiring the most extensive research in the field of tissue engineering.

CONCLUSION from Literature review

- ❖ There are different approach of generating lattice for AM, CAD, Image and implicit based. Implicit based lattice is preferred now a days as it reduces weight whilst maintaining optimal performance.
- Polymer based lattice have a balance between mechanical properties and porous architecture which allows cell infiltration and vascularization
- Design of AM porous Zn for controlling its mechanical properties and degradation behavior.
- ❖ FEA is performed for static load, pressure to get stress deformation curve, which will decrease the physical manufacturing cost .
- Increase in more surface area for same porosity increases the chances of bone regeneration.

Research Gap

➤ FEA has been performed to simulate only the compression behaviour of lattice structure but we shall also perform to simulate the tensile and fatigue behaviours of lattice structure.

- > Impact testing of polymer based lattice will also performed.
- ➤ We also measure the rate of decomposition of scaffold material which is easily consume by our body because due to excess decomposition of material causes adverse effect on body.

Research Objective

To design the lattice structure which have high porosity with desire mechanical properties.

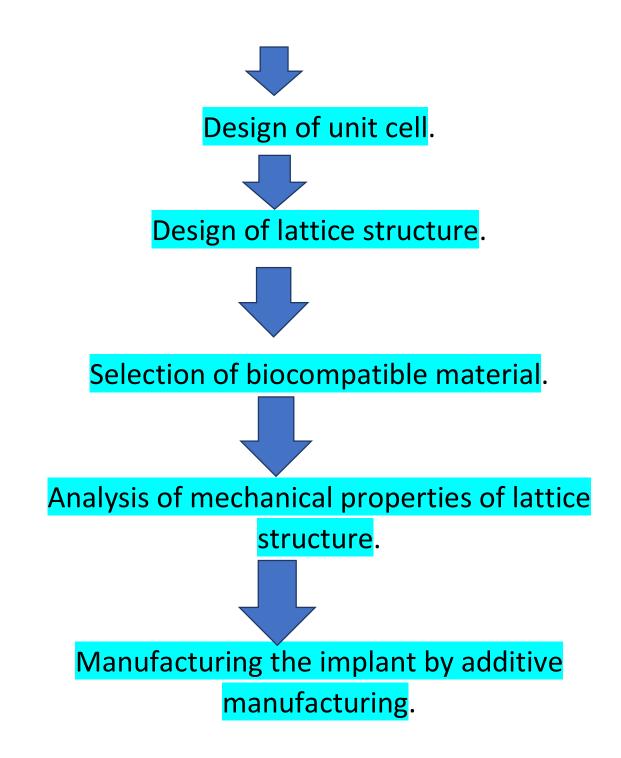
- The lattice structure in which the better interconnected network that helps in proliferation and vascularization.
- ➤ The scaffold is biodegradable it allow the body's own cells, over time, to eventually replace the implanted scaffold. Scaffold and constructs are not intended as permanent implants. The scaffold must therefore be biodegradable so to allow cell to produce their own extracellular matrix.

Research methodology

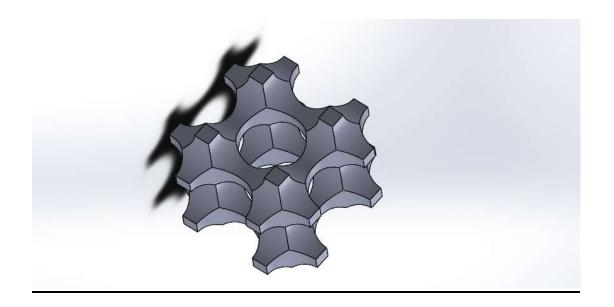
- Lattice based part designed in **SolidWorks**(CAD software) & import in **Ansys**(simulation software).
- > Two rigid plates are modeled, assembled with lattice part.
- ➤ Surface to surface contact has been selected and the contact properties are taken as friction coefficient of 0.2 with normal behavior of hard contact.
- ➤ To simulate the boundary conditions, the **lower plate** is kept **fixed** and **the upper plate** is given a downward **displacement** of **0.01 mm**.

Research Framework

Specify the need of polymer scaffold.



IMPLICIT LATTICE



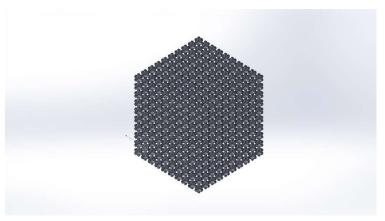
CUBIC LATTICE WITH CYLINDRICAL EXTRUDE CUT.

DIMENSION = 20*20*20.

POROSITY = 51.10% (internal radius = 0mm).

= 71.94% (internal radius =0.5mm).

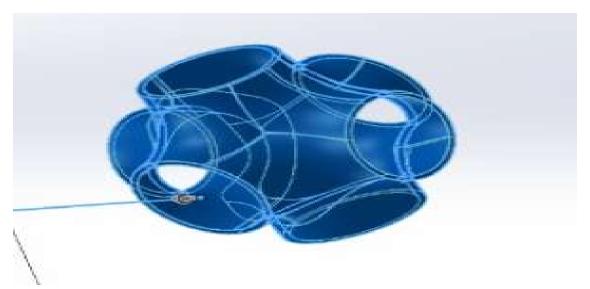
= 82.60% (internal radius =0.3mm).



LATTICE STRUCTURE

IMAGE LATTICE

UNIT CELL



Surface modelling with smooth circular opening from 6 faces of cube.

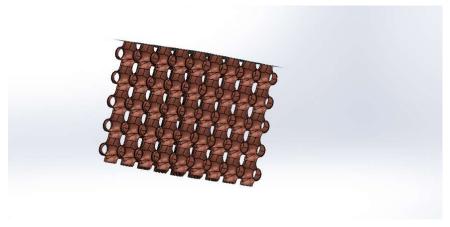
DIMENSION = 20*20*20.

POROSITY = 54.27% (THICKNESS=6MM).

= 66.94% (THICKNESS=7MM).

= 78.38% (THICKNESS=8MM).

= 87.75% (THICKNESS=9MM).



LATTICE STRUCTURE

Methodology

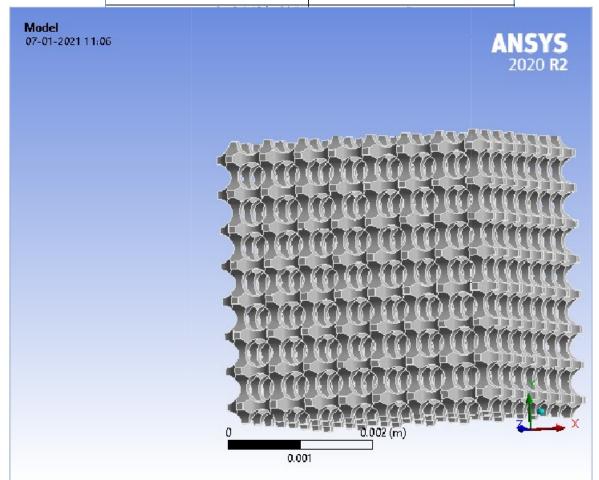
➤ Part designed in SolidWorks and imposed in Ansys in IGES file format.

- ➤ Ti6Al4V is assigned as a material (Elastic modulus 114GPa and Poission's ratio 0.34).
- ➤ Mess is generated of 0.034mm.
- Force (10N) is applied on one face while keeping the opposite face as the fixed support.
- ➤ Different solution tools like total deformation, Equivalent stress & Equivalent strain based on Von-misses criteria is used.
- > Required result is obtained.



Project*

| First Saved | Wednesday, January 6, 2021 |
|-----------------|----------------------------|
| Last Saved | Wednesday, January 6, 2021 |
| Product Version | 2020 R2 |



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 $nSYS\Solid$

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n Solution Information

n Results

ı Material Data

Titanium Alloy

Units

TARIE 1

| TABLE 1 | | |
|---------------------|--|--|
| Unit System | Metric (m, kg, N, s, V, A) Degrees rad/s Celsius | |
| • | | |
| | | |
| | | |
| Angle | Degrees | |
| - | | |
| | | |
| | | |
| Rotational Velocity | rad/s | |
| | | |
| | | |
| | | |
| Temperature | Celsius | |
| | | |
| | | |
| | | |

Model (A4)

Geometry

TABLE 2 Model (A4) > Geometry

| Object Name | Geometry | | |
|-----------------|--|--|--|
| State | Fully Defined | | |
| State | Fully Defined | | |
| Definition | | | |
| Source | C:\Users\divya\OneDrive\Documents\implicit structure compressive report_files\dp0\SYS\DM\SYS.scdoc | | |
| | | | |
| Туре | SpaceClaim | | |
| Length Unit | Meters | | |
| Element Control | Program Controlled | | |
| Display Style | Body Color | | |
| | Bounding Box | | |
| Length X | 4.e-003 m | | |
| Length Y | 4.e-003 m | | |
| Length Z | 4.e-003 m | | |
| | Properties | | |
| Volume | 1.3831e-008 m³ | | |
| Mass | 6.1256e-005 kg | | |
| Wass | 5.22555 555 N _D | | |

| Scale Factor Value | 1. | | |
|-----------------------------------|---------------------------|--|--|
| Statistics | | | |
| Bodies | 1 | | |
| Active Bodies | 1 | | |
| Nodes | 69547 | | |
| Elements | 31098 | | |
| Mesh Metric | None | | |
| | Update Options | | |
| Assign Default Material | No | | |
| | Basic Geometry Options | | |
| Solid Bodies | Yes | | |
| Surface Bodies | Yes | | |
| Line Bodies | Yes | | |
| Parameters | Independent | | |
| Parameter Key | | | |
| Attributes | Yes | | |
| Attribute Key | | | |
| Named Selections | Yes | | |
| Named Selection Key | | | |
| Material Properties | Yes | | |
| | Advanced Geometry Options | | |
| Use Associativity | | | |
| Coordinate Systems | | | |
| Coordinate System Key | | | |
| Reader Mode Saves Updated File | No | | |
| Use Instances | Yes | | |
| Smart CAD Update | | | |
| Compare Parts On Update | | | |
| Analysis Type | 3-D | | |
| Mixed Import Resolution | | | |
| Clean Bodies On Import | | | |
| Stitch Surfaces On Import | | | |
| Decompose Disjoint Geometry | | | |
| Enclosure and Symmetry Processing | Yes | | |
| | | | |

| Object Name | SYS\Solid |
|---------------------|-----------|
| State | Meshed |
| Graphics Properties | |
| Visible | Yes |
| Transparency | 1 |

| Definition | | | |
|------------------------|---------------------------|--|--|
| Suppressed | No | | |
| Stiffness Behavior | Flexible | | |
| Coordinate System | Default Coordinate System | | |
| Reference Temperature | By Environment | | |
| Treatment | None | | |
| | Material | | |
| Assignment | Titanium Alloy | | |
| Nonlinear Effects | Yes | | |
| Thermal Strain Effects | Yes | | |
| Во | unding Box | | |
| Length X | 4.e-003 m | | |
| Length Y | 4.e-003 m | | |
| Length Z | 4.e-003 m | | |
| F | Properties | | |
| Volume | 1.3831e-008 m³ | | |
| Mass | 6.1256e-005 kg | | |
| Centroid X | 1.5e-003 m | | |
| Centroid Y | 1.5e-003 m | | |
| Centroid Z | -1.5e-003 m | | |
| Moment of Inertia Ip1 | 1.6229e-010 kg·m² | | |
| Moment of Inertia Ip2 | 1.6229e-010 kg·m² | | |
| Moment of Inertia Ip3 | 1.6229e-010 kg·m² | | |
| | Statistics | | |
| Nodes | 69547 | | |
| Elements | 31098 | | |
| Mesh Metric | None | | |
| CAD Attributes | | | |
| PartTolerance: | 0.0000001 | | |
| Color:143.149.175 | | | |

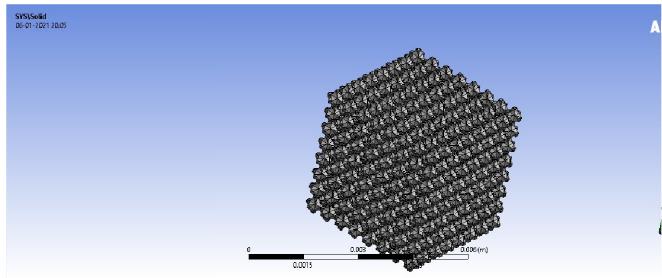


TABLE 3

Model (A4) > Geometry > Parts

FIGURE 1 Model (A4) > Geometry > SYS > Solid > Image

TABLE 4

| Model (A4) > Materials | | | |
|------------------------------------|---------------|--|--|
| Model (A4) > Materials Object Name | Materials | | |
| State | Fully Defined | | |
| Statistics | | | |
| | | | |
| Materials | 2 | | |
| Material Assignments | 0 | | |

U.UU13 FIGURE 2

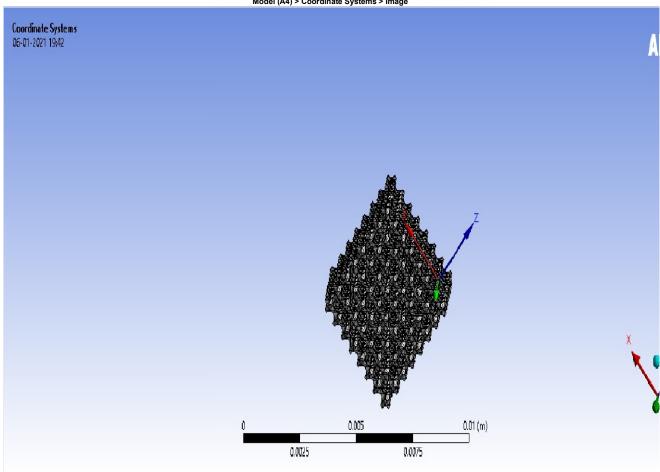
0.0040

als > Image

| Object Name | Global Coordinate System |
|----------------------|--------------------------|
| | |
| | |
| State | Fully Defined |
| State | rully befined |
| | |
| | |
| | Definition |
| | |
| | |
| | |
| Туре | Cartesian |
| | |
| | |
| Coordinate System ID | 0. |
| coordinate system is | 0. |
| | |
| | |
| | Origin |
| | |
| | |
| Origin X | 0. m |
| Oligiii X | 0.111 |
| | |
| | |
| Origin Y | 0. m |
| | |
| | |
| | |
| Origin Z | 0. m |
| | |
| | |
| Directional Vectors | |
| Directional vectors | |
| | |
| | |
| X Axis Data | [1.0.0.] |
| | |
| | |
| V Avia Data | [0.4.0.] |
| Y Axis Data | [0.1.0.] |
| | |
| | |
| Z Axis Data | [0.0.1.] |
| 2 AND Data | [5.5.1.] |
| | |
| | |

TABLE 5 Model (A4) > Coordinate Systems > Coordinate System

FIGURE 3 Model (A4) > Coordinate Systems > Image



Mesh

TABLE 6 Model (A4) > Mesh

| Object Name | Mesh |
|--------------------|----------------------|
| | |
| | |
| | |
| | |
| State | Solved |
| | |
| | |
| | |
| | |
| Display | |
| Бюршу | |
| | |
| | |
| | |
| Display Style | Use Geometry Setting |
| Display Style | ose deometry setting |
| | |
| | |
| | |
| Defaults | |
| Detaults | |
| | |
| | |
| | |
| | |
| Physics Preference | Mechanical |
| | |
| | |
| | |
| | |
| Element Order | Program Controlled |
| | |
| | |
| | |
| | |
| | |

| | Element Size | Default |
|----------------|--|-----------------------|
| | Sizing | |
| | Use Adaptive Sizing | Yes |
| | Resolution | Default (2) |
| | Mesh Defeaturing | Yes |
| | Defeature Size | Default |
| | Transition | Fast |
| | Span Angle Center | Coarse |
| | Initial Size Seed | Assembly |
| | Bounding Box Diagonal | 6.9282e-003 m |
| | Average Surface Area | 1.1781e-007 m² |
| | Minimum Edge Length | 1.e-004 m |
| FIGU | Quality | |
| RE 4 | Check Mesh Quality | Yes, Errors |
| | Error Limits | Aggressive Mechanical |
| | Target Quality | Default (0.050000) |
| | Smoothing | Medium |
| | Mesh Metric | None |
| | Inflation | |
| Model | Use Automatic Inflation | None |
| (A4) > Mesh | Inflation Option | Smooth Transition |
| VIESII | Transition Ratio | 0.272 |
| Image | Maximum Layers | 5 |
| | Growth Rate | 1.2 |
| | Inflation Algorithm | Pre |
| | View Advanced Options | No |
| | Advanced | |
| | Number of CPUs for Parallel Part Meshing | Program Controlled |
| | Straight Sided Elements | No |
| | Rigid Body Behavior | Dimensionally Reduced |
| | Triangle Surface Mesher | Program Controlled |
| | Topology Checking | Yes |
| | Pinch Tolerance | Please Define |
| | Generate Pinch on Refresh | No |
| | Statistics | |
| | Nodes | 69547 |
| | Elements | 31098 |

Static Structural (A5)

TABLE 7

| Model (A4) > Analysis | | | |
|-------------------------|------------------------|--|--|
| Object Name | Static Structural (A5) | | |
| State | Solved | | |
| Definition | | | |
| Physics Type | Structural | | |
| Analysis Type | Static Structural | | |
| Solver Target | Mechanical APDL | | |
| Options | | | |
| Environment Temperature | 22.°C | | |
| Generate Input Only | No | | |

TABLE 8
Model (A4) > Static Structural (A5) > Analysis Settings

Number Of Steps | 1.

| Current Stelen Name | Analysis _i Settings |
|--------------------------------|--|
| Step End Fiffle | គ្នីប្បឹy Defined |
| Auto Time Stepping | Step Controls Program Controlled |
| | Solver Controls |
| Solver Type | Program Controlled |
| Weak Springs | Off |
| Solver Pivot Checking | Program Controlled |
| Large Deflection | Off |
| Inertia Relief | Off |
| Quasi-Static Solution | Off |
| ., | Rotordynamics Controls |
| Coriolis Effect | Off |
| | Restart Controls |
| Generate Restart Points | Program Controlled |
| Retain Files After Full Solve | No No |
| Combine Restart Files | Program Controlled |
| Combine Nestare mes | Nonlinear Controls |
| Newton-Raphson Option | Program Controlled |
| Force Convergence | Program Controlled |
| Moment Convergence | Program Controlled |
| Displacement Convergence | Program Controlled |
| Rotation Convergence | Program Controlled |
| Line Search | Program Controlled |
| Stabilization | Program Controlled |
| Stabilization | Advanced |
| Inverse Option | No |
| Contact Split (DMP) | Off |
| contact spirt (DIVIF) | Output Controls |
| Stress | Yes |
| Surface Stress | No No |
| Back Stress | No |
| Strain | Yes |
| Contact Data | Yes |
| Nonlinear Data | No No |
| Nodal Forces | No No |
| Volume and Energy | Yes |
| Euler Angles | |
| General Miscellaneous | Yes No |
| Contact Miscellaneous | No No |
| Store Results At | All Time Points |
| l l | All Time Points Program Controlled |
| Result File Compression | |
| Colora Filos Discotore | Analysis Data Management |
| Solver Files Directory | C:\Users\divya\OneDrive\Documents\implicit structure compressive report_files\dp0\SYS\MECH |
| Future Analysis | None |
| Scratch Solver Files Directory | NI. |
| Save MAPDL db | No . |
| Contact Summary | Program Controlled |
| Delete Unneeded Files | Yes |
| Nonlinear Solution | No |
| Solver Units | Active System |
| Solver Unit System | mks |

| Object Name | Fixed Support | Force |
|-------------------|--------------------|--------------------------|
| State | | Fully Defined |
| | Scope | |
| Scoping Method | Geometry Selection | |
| Geometry | | 64 Faces |
| | Definition | |
| Туре | Fixed Support | Force |
| Suppressed | | No |
| Define By | | Components |
| Applied By | | Surface Effect |
| Coordinate System | | Global Coordinate System |
| X Component | | -10. N (ramped) |
| Y Component | | 0. N (ramped) |
| Z Component | | 0. N (ramped) |
| | | |

TABLE 9 Model (A4) > Static Structural (A5) > Loads

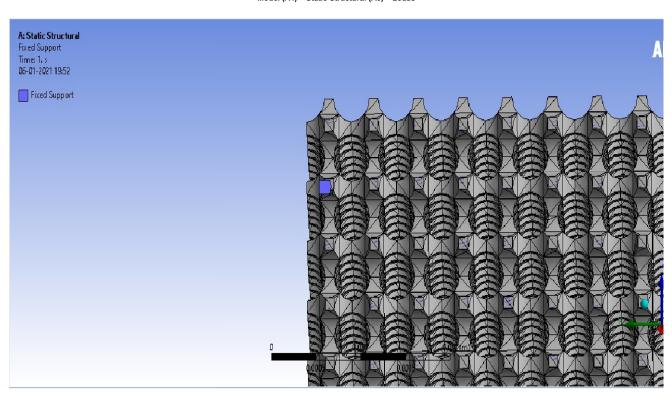
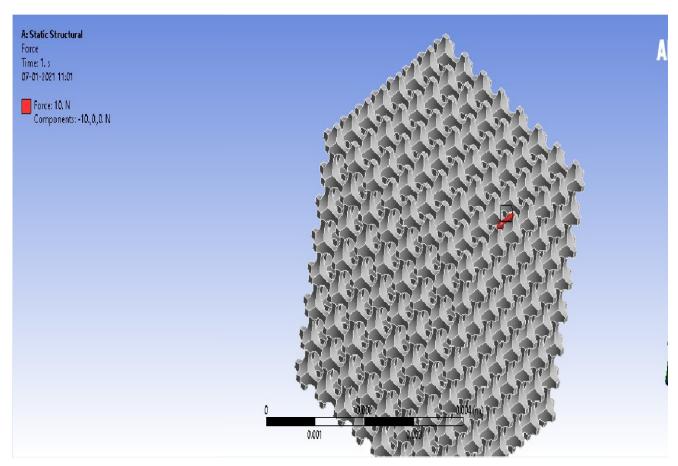


FIGURE 6 Model (A4) > Static Structural (A5) > Force

FIGURE 7 Model (A4) > Static Structural (A5) > Force > Image



Solution (A6)

| Object Name | Solution (A6) |
|--------------------------|---------------|
| State | Solved |
| Adaptive Mesh Refinement | |
| Max Refinement Loops | 1. |
| Refinement Depth | 2. |
| Information | |
| Status | Done |
| MAPDL Elapsed Time | 12. s |
| MAPDL Memory Used | 816. MB |
| MAPDL Result File Size | 21.125 MB |
| Post Processing | |
| Beam Section Results | No |
| On Demand Stress/Strain | No |
| | |

| Object Name | Solution Information |
|-------------|----------------------|
| | |

TABLE 10

| State | Solved |
|------------------------------|-------------------|
| | |
| Solution Information | |
| Solution Output | Solver Output |
| | |
| Newton-Raphson Residuals | 0 |
| | |
| Identify Element Violations | 0 |
| Update Interval | 2.5 s |
| Opuate interval | 2.33 |
| Display Points | All |
| | |
| FE Connection Visibility | |
| | |
| Activate Visibility | Yes |
| Display | All FE Connectors |
| | |
| Draw Connections Attached To | All Nodes |
| | |
| Line Color | Connection Type |
| Visible on Results | No |
| Visible of Results | 1.0 |
| Line Thickness | Single |
| | |
| Display Type | Lines |
| | |

TABLE 11 Model (A4) > Static Structural (A5) > Solution (A6) > Solution Information

TABLE 12

Model (A4) > Static Structural (A5) > Solution (A6) > Results

Total Deformation Equivalent Elastic Strain Equivalent Stress Object Name State Solved Scope Scoping Method Geometry Selection All Bodies Geometry Definition Total Deformation Equivalent Elastic Strain Туре Equivalent (von-Mises) Stress Time Ву Display Time Last Calculate Time History Yes Identifie No Suppressed Results 0. m 7.3698e-007 m/m 53173 Pa 2.639e-007 m 1.8159e-004 m/m 1.8789e+007 Pa Average 1.2969e-007 m 4.2338e-005 m/m 3.7462e+006 Pa SYS\Solid Minimum Occurs On Maximum Occurs On SYS\Solid Information Time 1. s Load Step 1 Substep Iteration Number Integration Point Results Display Option Averaged Average Across Bodies No

TABLE 13
Model (A4) > Static Structural (A5) > Solution (A6) > Total Deformation

| Model (A4) > Static Structural (A5) > Solution (A6) > Total Deformation | | | | |
|---|-------------|-------------|-------------|--|
| Time [s] | Minimum [m] | Maximum [m] | Average [m] | |
| | | | | |
| | | | | |
| | | | | |
| | | | | |
| | | | | |
| | | | | |
| 1. | 0. | 2.639e-007 | 1.2969e-007 | |
| | | | | |
| | | | | |
| | | | | |
| | | | | |
| | | | | |
| | | | | |

FIGURE 9
Model (A4) > Static Structural (A5) > Solution (A6) > Total Deformation > Image

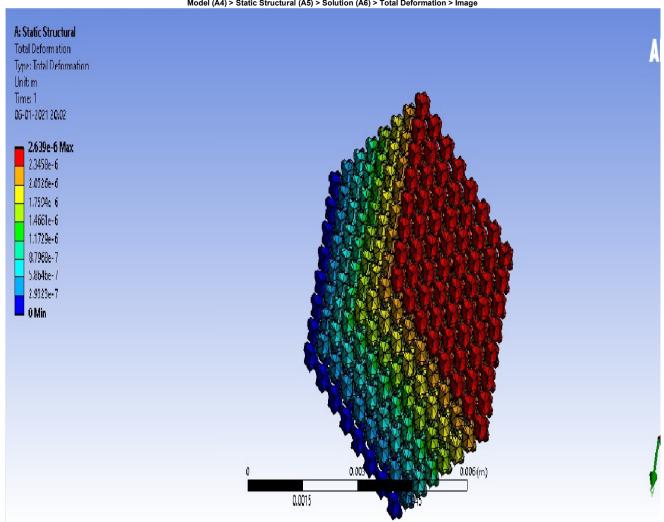


TABLE 14

| | IADEL IT | | | | |
|---|---------------|---------------|---------------|--|--|
| Model (A4) > Static Structural (A5) > Solution (A6) > Equivalent Elastic Strain | | | | | |
| Time [s] | Minimum [m/m] | Maximum [m/m] | Average [m/m] | | |
| 1. | 7.3698e-007 | 1.8159e-004 | 4.2338e-005 | | |

FIGURE 11
Model (A4) > Static Structural (A5) > Solution (A6) > Equivalent Elastic Strain > Image

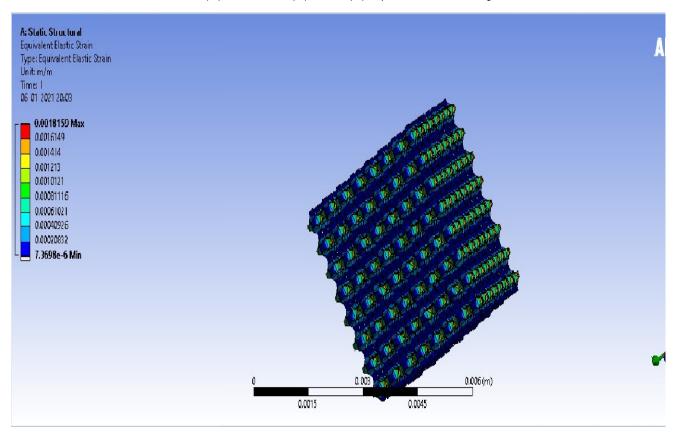
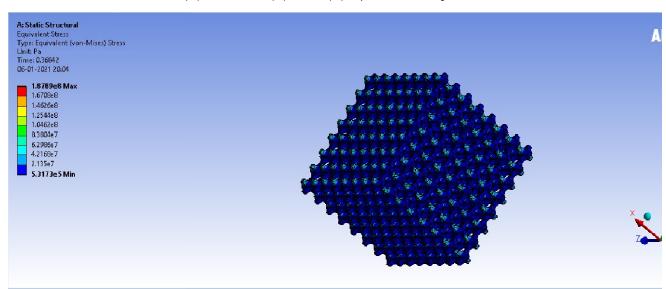


TABLE 15
Model (A4) > Static Structural (A5) > Solution (A6) > Equivalent Stress

| - | moder (A4) > Static Structural (A5) > Solution (A6) > Equivalent Stress | | | | |
|---|---|--------------|--------------|--------------|--|
| | Time [s] | Minimum [Pa] | Maximum [Pa] | Average [Pa] | |
| | | • • | , , | , , , | |
| | | | | | |
| | | | | | |
| | | | | | |
| | | | | | |
| | | | | | |
| Н | | 50470 | 4.0700 007 | 27452 225 | |
| | 1. | 53173 | 1.8789e+007 | 3.7462e+006 | |
| | | | | | |
| | | | | | |
| | | | | | |
| | | | | | |
| | | | | | |
| | | | | | |
| | | | | | |

FIGURE 13
Model (A4) > Static Structural (A5) > Solution (A6) > Equivalent Stress > Image



Material Data

Titanium Alloy

TABLE 16

| Titanium Alloy > Constants | |
|----------------------------------|------------------|
| Density | 4429 kg m^-3 |
| | |
| Coefficient of Thermal Expansion | 9.4e-006 C^-1 |
| | |
| Specific Heat | 522 J kg^-1 C^-1 |
| | |
| Thermal Conductivity | 21.9 W m^-1 C^-1 |
| | |
| Resistivity | 1.7e-006 ohm m |
| | |
| | |

TABLE 17

| Titanium Alloy > Color | | | |
|------------------------|-------|------|--|
| Red | Green | Blue | |
| | o.cc | Dide | |
| | | | |
| | | | |
| | | | |
| 88 | 72 | 117 | |
| | | | |
| | | | |
| | | | |
| | | | |

TABLE 18
Titanium Alloy > Compressive Ultimate Strength
Compressive Ultimate Strength Pa 0

TABLE 19
Titanium Alloy > Compressive Yield Strength
Compressive Yield Strength Pa

1.07e+009

TABLE 20
Titanium Alloy > Tensile Yield Strength
Tensile Yield Strength Pa

1.1e+009

TABLE 21

Titanium Alloy > Tensile Ultimate Strength

Tensile Ultimate Strength Pa 1.17e+009

TABLE 22
Titanium Alloy > Isotropic Secant Coefficient of Thermal Expansion

Zero-Thermal-Strain Reference Temperature C

22

TABLE 23
Titanium Allov > Isotropic Elasticity

| ritalium Alloy > Isotropic Elasticity | | | | | | | |
|---------------------------------------|-----------------|-----------------|------------------|---------------|--|--|--|
| Young's Modulus Pa | Poisson's Ratio | Bulk Modulus Pa | Shear Modulus Pa | Temperature C | | | |
| 1.04e+011 | 0.36 | 1.2381e+011 | 3.8235e+010 | | | | |

TABLE 24
Titanium Alloy > Isotropic Relative Permeability
Relative Permeability

References:-

Additively manufactured porous metallic biomaterials

Amir A Zadpoor 1

Affiliations

•PMID: 31701985

*DOI: 10.1039/c9tb00420c

Additively manufactured functionally graded biodegradable porous zinc

Y Li 1, P Pavanram 2, I Zhou 1, K Lietaert 2, F.S.L. Bobbert 2, Yusuke Kubo 2, M.A. Leeflang 2, H. Jahr 4, A.A. Zadpoor 2

Affiliations

PMID: 31993592

•DOI: 10.1039/c9bm01904a

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Y Li 1, P Pavanram 2, I Zhou 1, K Lletaert 3, F S L Bobbert 1, Yusuke Kubo 2, M A Leeflang 2, H Jahr 5, A A Zadpoor 2

Affiliations

PMID: 31993592

•DOI: 10.1039/c9bm01904a

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- . Biomaterials & Scaffolds for tissue engineering Author of article Fregal J. O'Brien