Symmetric Three-body Problem Notes

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1 Equations of Motion

From Ekeland page 59 (paperback edition):

Imagine two stars of equal mass, rotating around their common center of gravity. Newton's law asserts that this is possible, and that the orbit of the two stars will be a circle, along which they travel with equal speed, being exactly opposite to each other at all times. [...] A third body, with very small mass—a comet, for instance—moves along [the axis].

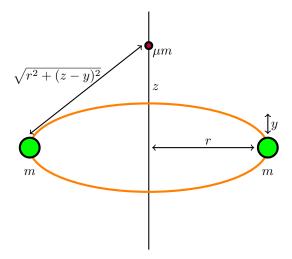


Figure 1: Two stars of the same mass m travel in a circular orbit along with a small comet on the centerline with an initial velocity along the z-axis. The system has cylindrical symmetry so that the plain containing the stars moves as a single unit and the comet does not leave the axis.

Gravitational potentials from the interactions of the 3 masses. The first term is the interaction of the stars while the second term is the sum of the interactions of the comet with each of the stars.

$$U_{mm} = \frac{-Gmm}{2r} \tag{1}$$

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$$U_{\mu m} = \frac{-2G\mu mm}{\sqrt{r^2 + (z - y)^2}}$$

$$(1)$$

(3)

Kinetic energy terms represent radial, circular and motion along the axis.

$$T = \frac{1}{2}\mu m\dot{z}^2 + m\dot{y}^2 + m\dot{r}^2 + m(r\dot{\theta})^2 \tag{4}$$

Lagrangian.

$$\mathcal{L} = T - V \tag{5}$$

$$= \frac{1}{2}\mu m\dot{z}^2 + m\dot{y}^2 + m\dot{r}^2 + m(r\dot{\theta})^2 + \frac{-Gmm}{2r} + \frac{-2G\mu mm}{\sqrt{r^2 + (z-y)^2}}$$
 (6)

Partial derivatives with respect to velocities and postitions.

$$\frac{\partial \mathcal{L}}{\partial \dot{z}} = \mu m \dot{z} \tag{7}$$

$$\frac{\partial \mathcal{L}}{\partial \dot{r}} = 2m\dot{r}$$
 (8)

$$\frac{\partial \mathcal{L}}{\partial \dot{y}} = 2m\dot{y} \tag{9}$$

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$$\frac{\partial \mathcal{L}}{\partial \dot{y}} = 2m \dot{y} \tag{9}$$

$$\frac{\partial \mathcal{L}}{\partial \dot{\theta}} = 2m r^2 \dot{\theta} \tag{10}$$

$$\frac{\partial \mathcal{L}}{\partial z} = \frac{2G\mu m^2(z-y)}{(r^2 + (z-y)^2)^{3/2}} \tag{11}$$

$$\frac{\partial \mathcal{L}}{\partial r} = 2mr\theta^2 - \frac{Gm^2}{r^2} - \frac{2G\mu m^2 r}{(r^2 + (z - y)^2)^{3/2}}$$
(12)

$$\frac{\partial \mathcal{L}}{\partial y} = -\frac{2G\mu m^2(z-y)}{(r^2 + (z-y)^2)^{3/2}}$$
(13)

$$\frac{\partial \mathcal{L}}{\partial \theta} = 0 \tag{14}$$

Equations of motion,

$$\ddot{z} = \frac{2Gm(z-y)}{(r^2 + (z-y)^2)^{3/2}} \tag{15}$$

$$\ddot{z} = \frac{2Gm(z-y)}{(r^2 + (z-y)^2)^{3/2}}$$

$$\ddot{y} = \frac{G\mu m(z-y)}{(r^2 + (z-y)^2)^{3/2}}$$

$$\ddot{r} = r\dot{\theta}^2 - Gm(\frac{1}{2r^2} + \frac{\mu r}{(r^2 + (z-y)^2)^{3/2}})$$
(15)

$$\ddot{r} = r\dot{\theta}^2 - Gm(\frac{1}{2r^2} + \frac{\mu r}{(r^2 + (z-y)^2)^{3/2}})$$
 (17)

$$\ddot{\theta} = \frac{-2\dot{r}\dot{\theta}}{r} \tag{18}$$

can be rewritten as a system of 1^{st} differential equations and integrated with a "Leap frog" integrator. This preserves phase space topology.

2 Sources and Licensing

For code and example output, please visit and fork or clone https://github. com/DrSkippy/Gravitational-Three-Body-Symmetric. If you find errorsor have comments, please email scott@drskippy.net. This work is licensed under a Creative Commons CC0 1.0 Universal (CC0 1.0) http://creativecommons. org/publicdomain/zero/1.0/.

References

[Ekeland1990] Ivar Ekeland. Mathematics and the Unexpected. http: //www.amazon.com/Mathematics-Unexpected-Ivar-Ekeland/dp/ 0226199908 1990.

[Drexel] The Leapfrog Integrator, http://einstein.drexel.edu/courses/ Comp_Phys/Integrators/leapfrog/.

